Model Order Reduction of Phased

ARRAY ANTENNAS

Even with the immense computing power nowadays available, the electromagnetic modeling of electrically large structures, for computational efficiency while keeping adequate accuracy, still remains a challenge. Techniques such as the *domain decomposition* [24, 25] may be used in order to split a large computational domain into smaller ones, reducing in this way the memory requirements and the overall computational time. The finite elements analysis of arbitrarily shaped radiating structures is usually achieved by the decoupling of the finite elements formulation, that is a system of differential equations stated in a weak form to obtain a linear system of equations, from the boundary integrals required to compute the radiated far fields from equivalent surface currents. Thus the finite elements domain is limited to the near fields region, truncating the finite elements domain such that the memory requirements become acceptable. Then the fields computed from a surface bounding the radiating elements allow the near-to-far transformation through the solution of the integral equations presented in chapter 1.

Gyon do not rally solve integral equations. You only evaluate air integral equations. You only evaluate air integral equations organs in non-local radiation boundary conditions like e.g. FEM-BEM coupling.

The Petrov-Galerkin projections [13] on the finite elements differential equations leads to a linear system of equations in which the unknowns are the coefficients for the chosen basis expansion. This linear system can be thought as a the input linear system of equations of a radiating model, and the spanning coefficients of the basis as a model *state vector*. Then, an output system of equations, based on the near-to-far operator introduced in chapter 2, modified in order to act on the state vector, completes the modeling of an analyzed radiating structure.

The full system obtained in such a way is composed by very large but sparse matrices, typically with hundred thousands and even hundred millions of unknowns.

The application of projection based model order reduction techniques, leading to full matrices but of several orders of magnitude lower, allows fast in-line stage computations. Furthermore, the approximations made in the far-fields operators through the DFT-truncation allow to decouple the look angle from the dimensions of the reduced for Jay.

Order model (ROM), such that the output matrix can remain of a very low order.

In a first part of this chapter, the general FEM formulation will be viewed, together with the construction of the output system matrices, in order to build the full radiation model. Then, the choice of the approximation spaces for the near fields will be discussed for the model order reduction, keeping particular attention on the error introduced. Finally, the effectiveness of a model order reduction applied to two different radiating structures will be shown, comparing the results to the full-model computations.

3.1 The FEM vector Helmholtz formulation

The most commonly employed formulation of the full-wave finite elements for solving the Maxwell's equations in a bounded domain is the vector Helmholtz equation for the electric field (1.1.31), recast in a weak form (*weighted residuals*) to achieve a linear equations system. The weak form is supplied by the testing of (1.1.31) with

Attention: (1.1.31) only holds for homogeneous meder, which is mornally not what you have to conside the following telembolty-quation:

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opportune basis, the *curl-conforming vector finite elements*¹. Consider the domain $\Omega \subset \mathbb{R}^3$ to be bounded by $\partial \Omega = \Gamma = \Gamma_E \cup \Gamma_H \cup \Gamma_{WG} \cup \Gamma_R$ as depicted in figure 3.1. Γ_E cor-

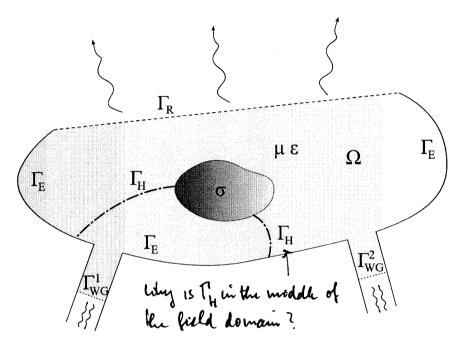


Figure 3.1: FEM domain of the radiation problem.

responds to Dirichlet conditions boundaries, Γ_H to Neumann conditions boundaries, Γ_{WG} to waveguides truncation boundaries, and Γ_R is an boundary through which the electromagnetic power flows, that is the radiation boundary. Consider now the discretized domain $\Omega_h = \bigcup_{n=1}^N \Omega_n \subset \Omega$, where the subdomains Ω_n (the tetrahedra for example) are such that $\Omega_n \cap \Omega_m = \{0\}$ for $n \neq m$. Ω_h is parameterized in by the maximum diameter h of the subdomains Ω_n , and we shall assume that as $h \to 0$, $\Omega_h \to \Omega$. With $\hat{\boldsymbol{n}}$ pointing outwardly from Ω , consider the *Sobolev* space [33] $\mathcal{W} = \mathcal{W}(\Omega_h) \subset \mathcal{H}(\text{curl};\Omega)$ of the *trial* functions \boldsymbol{w} , defined by the direct sum of two closed subspaces as follows

$$\mathcal{W} = \mathcal{W}_D \oplus \mathcal{W}_E,$$

$$\mathcal{W}_D := \{ \boldsymbol{w} \in \mathcal{W} | \hat{\boldsymbol{n}} \times \boldsymbol{w} = 0 \text{ on } \Gamma_E \} \subset \mathcal{H}(\operatorname{curl}; \Omega),$$

$$\mathcal{W}_E := \{ \boldsymbol{w} \in \mathcal{W} | \hat{\boldsymbol{n}} \times \boldsymbol{w} \neq 0 \text{ on } \Gamma_E \} \subset \mathcal{H}(\operatorname{curl}; \Omega, \Gamma_E). \tag{3.1.1}$$

¹ Initially introduced by Nédélec, then developed by the research community into the so-called *hierarchical finite element basis* for vector fields.

 \mathcal{W}_D is in fact the subspace of the imposed (annihilation of the tangential electric field). The total electric field is given by the approximation finite summations

$$E = \hat{E} + E_{D},$$

$$\hat{E} := \sum_{i} x_{i} \boldsymbol{w}_{i}, \quad \boldsymbol{w}_{i} \in \mathcal{W}_{E},$$

$$E_{D} := \sum_{j} x_{j} \boldsymbol{w}_{j}, \quad \boldsymbol{w}_{j} \in \mathcal{W}_{D}.$$

$$(3.1.2)$$

With Galerkin projections (the test functions are the same as the trial functions), it is possible to construct an algebraic linear system with as unknowns only the coefficients x_i . In effect, x_j are not problem unknowns as they are given by the boundary conditions.

Now, as Ω does not contain any impressed current sources, only induced currents persist and (1.1.31) can be rewritten as [32]

Think you have to explain the quantifier two
$$\nabla \times \frac{1}{\mu_r} \nabla \times E + j \, k_0 \zeta_0^2 \sigma \, E - k_0^2 \epsilon_r \, E = 0. \tag{3.1.3}$$

For notation disambiguation, notice that the formulation presented is a frequency-domain one, and the spectrum $\tilde{E}(r,\omega)$ has to be computed by means of a frequency sweep. Thus, consider $\tilde{E}(r,\omega)\Big|_{\omega=\omega_n}=E$ to be the electric field at a given angular frequency ω_n . The boundary conditions can be stated as

$$\hat{\boldsymbol{n}} \times (\boldsymbol{E} \times \hat{\boldsymbol{n}}) = \boldsymbol{E}_t, \quad \text{on } \Gamma_E,$$

$$\hat{\boldsymbol{n}} \times (\boldsymbol{H} \times \hat{\boldsymbol{n}}) = \boldsymbol{H}_t, \quad \text{on } \Gamma_H,$$

$$\hat{\boldsymbol{n}} \times (\boldsymbol{E} \times \hat{\boldsymbol{n}}) = Z_s \boldsymbol{H} \times \hat{\boldsymbol{n}}, \quad \text{on } \Gamma_R,$$
(3.1.4)

 Z_s being the wave impedance at the radiation boundary. We proceed with the Galerkin projections, obtaining [32]

$$\int_{\Omega} \nabla \times \boldsymbol{w}_{i} \cdot \frac{1}{\mu_{r}} \nabla \times \boldsymbol{E} \ d\Omega + j k_{0} \zeta_{0} \int_{\Omega} \boldsymbol{w}_{i} \cdot \sigma \boldsymbol{E} d\Omega +
j k_{0} \zeta_{0} \int_{\Gamma_{R}} \hat{\boldsymbol{n}} \times \boldsymbol{w}_{i} \cdot \frac{1}{Z_{s}} \hat{\boldsymbol{n}} \times \boldsymbol{E} d\Gamma - k_{0}^{2} \int_{\Omega} \boldsymbol{w}_{i} \cdot \epsilon_{r} \boldsymbol{E} d\Omega
= j k_{0} \zeta_{0} \int_{\Gamma_{WG}} \boldsymbol{w}_{i} \cdot \hat{\boldsymbol{n}} \times \boldsymbol{H}_{t} d\Gamma, \quad \forall \boldsymbol{w}_{i} \in \mathcal{W}_{E}.$$
(3.1.5)

On Γ_{WG} , if we consider the wave ports to be fed only by the dominant mode (lowest cutoff frequency), the following relation holds for the k^{th} port [31, 34]

$$\hat{\boldsymbol{n}} \times \boldsymbol{H}_{t}^{k} = \frac{1}{Z_{k}} \hat{\boldsymbol{n}} \times \hat{\boldsymbol{n}} \times \boldsymbol{E}_{t}^{k} - \frac{2}{Z_{k}} \hat{\boldsymbol{n}} \times \hat{\boldsymbol{n}} \times \boldsymbol{E}_{t}^{k \text{ inc}}, \quad \text{on } \Gamma_{WG}^{k}, \quad (3.1.6)$$

where $E_t^{k \, \mathrm{inc}}$ is the incident electric field and Z_k the modal impedance. Assuming the waveguides segments on which the wave ports are applied to be long enough such that the higher order modes are evanescent and once they encounter the electromagnetic structure's discontinuities (e.g. the junction to the antenna) their power contribution is several order of magnitude lower than the dominant mode, this so-called transfinite elements method [34] allows to feed a generic structure though waveguides and (3.1.5) becomes

$$\int_{\Omega} \nabla \times \boldsymbol{w}_{i} \cdot \frac{1}{\mu_{r}} \nabla \times \boldsymbol{E} \, d\Omega + j \, k_{0} \zeta_{0} \int_{\Omega} \boldsymbol{w}_{i} \cdot \sigma \boldsymbol{E} \, d\Omega - k_{0}^{2} \int_{\Omega} \boldsymbol{w}_{i} \cdot \epsilon_{r} \boldsymbol{E} \, d\Omega + j \, k_{0} \zeta_{0} \int_{\Gamma_{R}} \hat{\boldsymbol{n}} \times \boldsymbol{w}_{i} \cdot \frac{1}{Z_{s}} \hat{\boldsymbol{n}} \times \boldsymbol{E} \, d\Gamma + \sum_{m=1}^{M} \frac{j \, k_{0} \zeta_{0}}{Z_{m}} \int_{\Gamma_{WG}^{m}} \hat{\boldsymbol{n}} \times \boldsymbol{w}_{i} \cdot \hat{\boldsymbol{n}} \times \boldsymbol{E}_{t} \, d\Gamma \\
= \frac{j \, 2 \, k_{0} \zeta_{0}}{Z_{k}} \int_{\Gamma_{WG}^{k}} \hat{\boldsymbol{n}} \times \boldsymbol{w}_{i} \cdot \hat{\boldsymbol{n}} \times \boldsymbol{E}_{t}^{k \text{ inc}} \, d\Gamma, \quad \forall \boldsymbol{w}_{i} \in \mathcal{W}_{E} \tag{3.1.7}$$

where M is the total number of ports and $1 \le k \le M$. The present formulation, for the summation all over the ports entering into the computational domain, couples one port to the others, hence models the mutual couplings between the radiating elements. The condition on Γ_R is an impedance condition, the so-called *absorbing boundary condition* (ABC), which ideally annihilates the plane wave reflection on the boundary. Actually, the reflection coefficient is strongly dependent on the incidence angle of the plane wave, being null for perpendicular one. In order to keep sufficient validity in the computation of the fields in opened structures, the ABCs must be placed sufficiently far away from the radiating elements or scatterers such that incidence becomes less oblique. For example, in the commercial CAD HFSS², the recommendations are to place the radiation boundaries at least at $\lambda_{\min}/2$ away from the

 $^{^2}$ Ansoft HFSS $^{\text{TM}}$ (acquired by Ansys Inc.). Further informations available at http://www.ansys.com/products/electromagnetics/.

radiating sources, where λ_{min} is associated to highest frequency set in a frequency sweep.

Equation (3.1.7) leads, with the field expansion of (3.1.2) and setting the solving frequency ($k_0 = \omega \sqrt{\epsilon_0 \mu_0}$), to a linear system of the form [27, 31]

$$\mathbf{A} \mathbf{x} = \mathbf{b}, \tag{3.1.8}$$

where $\mathbf{A} \in \mathbb{C}^{N \times N}$ is a symmetric matrix, $\mathbf{x}, \mathbf{b} \in \mathbb{C}^N$ and N the number of unknowns or degrees of freedom (DoFs).

The system of (3.1.8) is such that, in the case we are analyzing an electrically large structure, N is in the order of hundred thousands to several millions. This is due to the fact that the size of the elements h has to be very low respectively to the wavelength in order to reduce the fields jumps from one element to a neighboring one. Typical values are comprised between $^{\lambda}/_3$ and $^{\lambda}/_{10}$ in order to keep sufficient accuracy while not exceeding in the computational resources demand. Also, when using higher order polynomials to approximate the fields, the number interpolating points that have incompletely may increase. Hence, the number of DoF depends on the chosen basis. Where the points is mislealing, fields are known to vary rapidly, e.g. next to fields singularities like the edges of a conducting plate, the size of the elements must be lowered to achieve better accuracy. As not mally the polate All these factors have to be taken into account for an effective design, and the final the field.

3.2 Projection-based model order reduction

Definition 2 Consider the single input, single output (SISO) full system

$$\mathbf{A}\mathbf{x} = \mathbf{b}u,$$

$$y = \mathbf{c}^H \mathbf{x}, \tag{3.2.9}$$

with $\mathbf{A} \in \mathbb{C}^{N \times N}$, $\mathbf{x}, \mathbf{b}, \mathbf{c} \in \mathbb{C}^{N}$, \mathbf{b} being the system input vector and \mathbf{c} the output vector. The projection-based reduced order model (ROM) is given by [32]

For this projection approach to make sense, you first have to introduce an affine parameter dependence! You have to say that (e.g.
$$A = \sum D_i(p) A_i$$
). Then projection leads to $X = \sum D_i(p) WA_i V$ is sure is true for b and controls: ROM has same parameterized structure as original model

$$y = \tilde{\mathbf{c}}^H \tilde{\mathbf{x}}, \tag{3.2.10}$$

with

$$\tilde{\mathbf{A}} := \mathbf{W}^H \mathbf{A}, \mathbf{V}$$

$$\tilde{\mathbf{b}} := \mathbf{W}^H \mathbf{b},$$

$$\tilde{\mathbf{c}} := \mathbf{V}^H \mathbf{c},$$
(3.2.11)

and **W**, **V** $\in \mathbb{C}^{N \times n}$, $n \ll N$.

approximated

The full unknowns vector is given by $\mathbf{x} \approx \mathbf{V}\tilde{\mathbf{x}}$, that is \mathbf{x} is spanned by the column space of \mathbf{V} (trial vectors) through the coefficients $\tilde{\mathbf{x}}$ computed from the solution of the ROM input system. Thus, \mathbf{V} constitute a set of global basis vectors for the full FEM system. \mathbf{W}^H applied to the column vectors of a matrix or to a vector is a testing operation with the space spanned by the columns of \mathbf{W} (see chapter 2).

The SISO system of (3.2.9) is actually, if no complex feeding structure with power dividers is included in the model, the model of a single antenna. We need, in order to handle more antennas, to redefine the full model into the following multiple inputs, multiple outputs (MIMO) form

$$\mathbf{A} \mathbf{x} = \mathbf{B} \mathbf{u},$$

$$\mathbf{y} = \mathbf{C}^{H} \mathbf{x}, \tag{3.2.12}$$

with $\mathbf{x} \in \mathbb{C}^N$, $\mathbf{u} \in \mathbb{C}^g$, $\mathbf{y} \in \mathbb{C}^h$, and $\mathbf{A} \in \mathbb{C}^{N \times N}$, $\mathbf{B} \in \mathbb{C}^{N \times g}$, $\mathbf{C} \in \mathbb{C}^{h \times N}$. Thus, the ROM of the present model is given by

$$\tilde{\mathbf{A}}\tilde{\mathbf{x}} = \tilde{\mathbf{B}}\mathbf{u},$$

$$\mathbf{y} = \tilde{\mathbf{C}}^H\tilde{\mathbf{x}}, \qquad (3.2.13)$$

with

$$\tilde{\mathbf{A}} := \mathbf{W}^H \mathbf{A} \mathbf{V},$$

$$\tilde{\mathbf{B}} := \mathbf{W}^H \mathbf{B},$$

$$\tilde{\mathbf{C}} := \mathbf{V}^H \mathbf{C},$$
(3.2.14)

and **W**, $\mathbf{V} \in \mathbb{C}^{N \times n}$, $n \ll N$.

3.3 Scan and look angles parameterized radiation model

The full system (3.2.12) represents our electromagnetic model of a phased array. Each of the columns of **B** represent a different right-hand side of the system, considering one by one each of the g excitation ports. In effect, with (3.1.7) the right-hand side is build for each k with $1 \le k \le M = g$, one by one while looking at the fields, that is the unknowns to find, produced by the excitations taken singularly. The superposition of the solutions, as the system is invariant (**A** remains the same all along this process), gives the fields produced by the ports excited contemporaneously. This is obvious if we think of the linearity property of the system. Now, as we are seeking for a beamsteering operation, and we know that it is achieved by relative phase shifts between the excitations given by (1.2.97), the input vector is $\mathbf{u} = \text{rowsp}(a_n \ e^{-jk(x_n \sin(\theta_x) + y_n \sin(\theta_y))}), 1 \le n \le g \text{ where } a_n \in \mathbb{C}$ is complex coefficient that can be devoted to pattern shaping [3, 4] or left to 1 if we desire only the beamsteering. We can rewrite the right-hand side of the input equation of (3.2.12) as the sum of coefficient-vector products

Ax =
$$a_1 e^{-jk(x_1\sin(\theta_x)+y_1\sin(\theta_y))} \mathbf{b}_1 + ... + a_g e^{-jk(x_g\sin(\theta_x)+y_g\sin(\theta_y))} \mathbf{b}_g$$
, (3.3.15) of the input vector needed by the projection approach.

emphasizing the parameterization of the system by the scan angle (θ_x, θ_y) . We have reduced this way the inputs to only two values, that is the scan angles in both x and y directions.

The output equation now has to be formulated in such a way that, from the solutions computed by the input equations, we derive the fields in several sampling points surrounding the array, then, applying the N2F operator we obtain the sought spherical components of the far fields detectors. The solutions to fields operator $\mathcal{F}(x)$ can be viewed as a large (as large as the unknowns) but sparse matrix that opportunely selects the solutions related to the elements in which the fields have to be sampled, then using them in the expansion (3.1.2), recovers the vector fields, electric and magnetic, in each selected point providing for example their Cartesian compo-

nents. Notice that, for Faraday's law (1.1.15) and the constitutive relation (1.1.21), the magnetic field is derived from the electric field by a curl operation. Then, from the near fields we can apply our N2F operator $\mathcal{L}(\cdot;\theta)$ (having opportunely included in the operator (1.3.109) the $\hat{n} \times \cdot$ operation on the fields to achieve the tangential components, that is the equivalent currents, and the Cartesian to spherical transformation of note 15) to the operator $\mathcal{F}(x)$, that is $\mathcal{L}(\mathcal{F}(x);\theta)$ to obtain the far fields detectors in the look angle direction θ for the chosen ϕ plane. As we have chosen to supply a DFT-truncation in order to reduce the dimensions of the output operator, the operator $\mathcal{L}(\cdot;\theta)$ is now given by the approximation (2.2.22), such that the rows are given by the c=2R+1 coefficients chosen in the expansion. We can thus write the output matrix \mathbf{C} as

$$\mathbf{C} = \mathbf{F}^H \,\hat{\mathbf{X}}^H \, \mathbf{t}^H = \mathbf{C}' \, \mathbf{t}^H, \qquad \mathbf{F} \in \mathbb{C}^{N \times p}, \, \hat{\mathbf{X}} \in \mathbb{C}^{p \times c}, \, \mathbf{C}' \in \mathbb{C}^{N \times c}$$
(3.3.16)

with $\mathbf{t} = \mathbf{t}(\theta) = \operatorname{colsp}\{1, \mathrm{e}^{j\theta}, \dots, \mathrm{e}^{jR\theta}, \mathrm{e}^{-jR\theta}, \dots, \mathrm{e}^{-j\theta}\} \in \mathbb{C}^c$ a row vector dependent on the look angle. p is the number of vector components for all the sampling points, that is 6 times the number of sampling points. To handle the θ and ϕ polarizations, it is necessary to provide two different DFT coefficients matrices \mathbf{X}_{ϕ} and \mathbf{X}_{θ} , computing separately the far field detectors in each polarization, here considered in the only output γ . The full system of the radiating model can be stated as

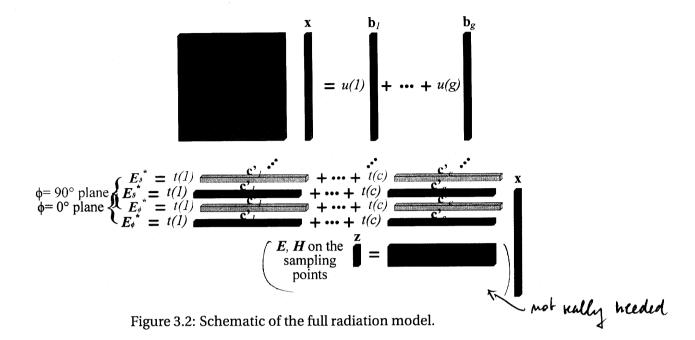
$$\mathbf{A} \mathbf{x} = a_1 e^{-jk(x_1 \sin(\theta_x) + y_1 \sin(\theta_y))} \mathbf{b}_1 + \dots + a_g e^{-jk(x_g \sin(\theta_x) + y_g \sin(\theta_y))} \mathbf{b}_g$$

$$y = \left(\operatorname{row}_1(\mathbf{C}'^H) + \dots + e^{jR\theta} \operatorname{row}_{R+1}(\mathbf{C}'^H) + \dots + e^{-jR\theta} \operatorname{row}_{R+2}(\mathbf{C}'^H) + \dots + e^{-j\theta} \operatorname{row}_{2R+1}(\mathbf{C}'^H) \right) \mathbf{x},$$

$$\mathbf{z} = \mathbf{F} \mathbf{x}. \tag{3.3.17}$$

The power radiated must be computed separately using (1.3.114), considering the near fields (z) obtained by F x, with F a solutions-to-near-fields operator. Notice that you can the total power radiated vary during the beamsteering operation, the mutual couplings varying as the excitations change, thus has to be computed for each scan angle selected. Finally, the patterns are derived using (1.3.112) and (1.3.113). A model radiated power schematic is illustrated in figure 3.2.

reflection coefficients you do not have to rely on the operator F. - I be you do not wally need to Fx



We proceed applying the model order reduction definition (3.2.13) to the SISO-system of (3.3.17):

$$\tilde{\mathbf{A}}\tilde{\mathbf{x}} = a_1 e^{-jk(x_1\sin(\theta_x)+y_1\sin(\theta_y))} \tilde{\mathbf{b}}_1 + \dots + a_g e^{-jk(x_g\sin(\theta_x)+y_g\sin(\theta_y))} \tilde{\mathbf{b}}_g$$

$$y = \left(\text{row}_1(\tilde{\mathbf{C}}'^H) + \dots + e^{jR\theta} \text{row}_{R+1}(\tilde{\mathbf{C}}'^H) + \dots + e^{-jR\theta} \text{row}_{R+2}(\tilde{\mathbf{C}}'^H) + \dots + e^{-j\theta} \text{row}_{2R+1}(\tilde{\mathbf{C}}'^H)\right) \tilde{\mathbf{x}},$$

$$\mathbf{z} = \tilde{\mathbf{F}} \mathbf{x}. \tag{3.3.18}$$

where the reduced order matrices of the system are $\tilde{\bf A} = {\bf W}^H {\bf A} {\bf V}$, $\tilde{\bf B} = {\bf W}^H {\bf B}$, $\tilde{\bf C}' = {\bf V}^H {\bf C}'$ or analogously, doing the transpose conjugate in both hand-sides, $\tilde{\bf C}'^H = {\bf C}'^H {\bf V}$, $\tilde{\bf F} = {\bf F} {\bf V}$, and ${\bf W}, {\bf V} \in \mathbb{C}^{N \times n}$, $n \ll N$, are respectively trial and test matrices we need to define in order to properly reduce de model.

3.4 Proper choice of the trial and test spaces

The model order reduction is mainly based on the fact that we approximate the solutions as $\mathbf{x} = \mathbf{V}\tilde{\mathbf{x}}$. If \mathbf{V} is an orthonormal basis, then $\tilde{\mathbf{x}} = \mathbf{V}^H\mathbf{x}$ and this means to test the solutions vector with an orthonormal basis, i.e. to provide a change of basis,

and the testing returns the coefficients $\tilde{\mathbf{x}}$ for the expansion in terms of this new basis. Now, to choose \mathbf{V} , we need to think of what we are testing. The solutions vector is straightly related to the fields by the expansion (3.1.2). They are in fact the coefficients of the expansion in terms of orthogonal shape functions that approximates the near fields. Hence, from the results obtained from chapter 2, section 2.1.3, we can express the near fields pictures to span the testing space in terms of solutions obtained for various scan angles.

Now, as the input system equation is linear, to apply scalar values u(i), $1 \le i \le g$ to the right-hand side vectors $\mathbf{b}_i \subset \mathbf{B}$ gives the same results as applying the same coefficients to the solutions vectors \mathbf{x}_i' relative to \mathbf{b}_i :

$$\mathbf{A} \mathbf{x} = \mathbf{B} \mathbf{u} \Rightarrow \sum_{i=1}^{g} \mathbf{A} \mathbf{x}_{i} = \sum_{i=1}^{g} \mathbf{b}_{i} u(i) \Rightarrow \sum_{i=1}^{g} \mathbf{x}_{i} = \sum_{i=1}^{g} \left(\mathbf{A}^{-1} \mathbf{b}_{i} \right) u(i) = \sum_{i=1}^{g} \mathbf{x}_{i}^{'} u(i),$$

$$\mathbf{y} = \mathbf{u}(i) = \mathbf{u}(i; \theta_{x}, \theta_{y}). \tag{3.4.19}$$

We can thus compute in a prior stage the solutions vectors \mathbf{x}_i' , $1 \le i \le g$ corresponding to equal phase for all the excitations, i.e. a broadside radiation configuration. Then, the remaining scan angle selection is done changing the vector \mathbf{u} values select-

ing the appropriate (θ_x, θ_y) in (1.2.97). He solutions to those superpositions of r.h.s. which belong Finally, we need to collect several solutions pictures to span the scan angle space.

Finally, we need to collect several *solutions pictures* to span the scan angle space. The dimension of the scan angle space, as it was shown in chapter 2, depends on the array structure: the relative spacing an the number of radiating elements are the principal parameters. To compute the orthonormal vectors \mathbf{V} , either the SVD (right singular vectors) or the QR factorization (with \mathbf{Q} a matrix of orthonormal column vectors) [20] can be used, depending on the number of solutions pictures collected. In effect the complexity of an SVD is in the order of $2mn^2 + 11n^3$ while the QR factorization requires $2mn^2$, where m here corresponds to the DoFs while n is the number of scan angle pictures selected. For large n, the QR has a lower cost than the SVD, however with the QR we pay the price that the factorization is not always accurate [20].

The trial space can be, in a first attempt, the same as the testing space, leading to a Galerkin projection. Thus if we chose $\mathbf{W} = \mathbf{V}$, we may expect a total matching

e I do not really understand which you are downg here and wley (We do not want) to compute the solution for even hight hand side to locate, since with 1-10.000 dements

the scan angles selected and for the remaining scan angles an approximated solution.

Thus we can define the trial and test spaces as

$$\operatorname{span}\left\{\bigcup_{i=1}^{n}\mathbf{x}_{i}(\theta_{x},\theta_{y})\right\} \subseteq \operatorname{colsp}\left\{\mathbf{V}\right\},\tag{3.4.20}$$

$$\operatorname{span}\left\{\bigcup_{i=1}^{n}\mathbf{x}_{i}(\theta_{x},\theta_{y})\right\} \subseteq \operatorname{colsp}\left\{\mathbf{W}\right\}. \tag{3.4.21}$$

Another possibility in the choice of **W** is, according to [36], to use as trial space the solutions provided by the *dual problem* or *adjoint problem*. These solutions are given by the dual problem input system equation

$$\mathbf{A}^{H} \mathbf{z} = \mathbf{C} = \mathbf{C}' \mathbf{t}^{H} \Rightarrow \sum_{i=1}^{h} \mathbf{z}_{i} = \sum_{i=1}^{h} \left((\mathbf{A}^{H})^{-1} \mathbf{c}'_{i} \right) t^{\dagger}(i) = \sum_{i=1}^{h} \mathbf{z}'_{i} t^{\dagger}(i),$$

$$t(i) = t(i; \theta, \phi_{\text{const}}). \tag{3.4.22}$$

In a similar manner as done for the test space, we compute the solutions excluding the look angle vector \mathbf{t} , leaving a flexibility in the look angle selection. An inconvenience resulting from the radiation model definition is that any matching we may achieve in the look angle with the dual problem approach is related to one cut-plane and one polarization. In effect, a matrix \mathbf{C}' is built for each polarization and for each cut-plane, and an expansion in the trial vectors derived from a look angle selection space for one polarization and one plane does not necessarily mean matching in the others. The scan and look angles selections spaces for the dual problem method can be defined as, respectively,

$$\operatorname{span}\left\{\bigcup_{i=1}^{n}\mathbf{x}_{i}(\theta_{x},\theta_{y})\right\} \subseteq \operatorname{colsp}\left\{\mathbf{V}\right\},\tag{3.4.23}$$

$$\operatorname{span}\left\{\bigcup_{i=1}^{n}\mathbf{z}_{i}(\theta,\phi)\right\} \subseteq \operatorname{colsp}\left\{\mathbf{W}\right\}. \tag{3.4.24}$$

3.5 Numerical examples

The following examples have been realized combining several computational steps. First, the array structure is designed with HFFS, choosing the mesh density (h) and

polynomial degree (*p*) parameters and individuating the resonant frequency of the single element. Then, the simulation is run with the internal solver, in order to provide data for comparison with our results. For proper comparison, the mesh and material data are extracted from the HFSS model, then the system matrices are built with the LTE³ proprietary code and loaded into Matlab. We now have the input system matrices **A** and **B**. A direct solver optimized for sparse matrices is used to solve the input system. Finally, with the N2F operator and the solutions-to-near-fields operator, together with the bounding surface informations, the entire full model is achieved and its reduced order version is computable. In this section, we will mainly discuss on the spanning spaces for projection, and the accuracy of the reduced order model respectively to the full model.

3.5.1 3 by 5 patch antennas array

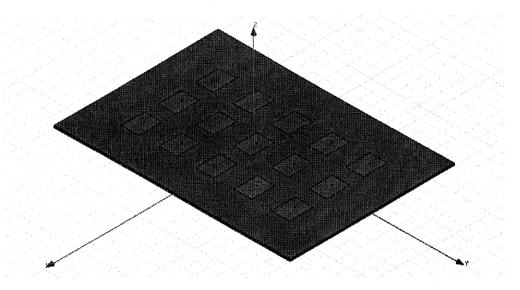


Figure 3.3: 3 by 5 patch antennas array fed by coaxial probes.

Figure 3.3 shows the an array of 3 by 5 patch antennas, equally spaced of $\lambda_0/2$, fed by 15 individual coaxial wave ports. The metal materials, that is the ground plane, the patches, an the coaxial lines and probes [3] are set to be *perfect electrical conductors*

³ Lehrstuhl für Teoretische Elektrotechnik, Universität des Saarlandes, http://www.lte.uni-saarland.de/.

(PEC). The coaxial dielectric, necessary to hold the core of the waveguide, is made of Teflon with $\epsilon_r = 2.1$ and its dielectric loss tangent is of $\delta = 0.001$. The dielectric substrate of the patches is of Rogers RT/duroid 5880, which has an $\epsilon_r = 2.2$ and a $\tan\delta=0.0009$. The patches have been designed in order to be resonant at 2.35 GHz. The triangulated mesh, performed by the h-adaptive solver of HFSS, for a minimum ΔS = 0.05 (scattering parameters at the ports) between the adaptive steps is shown in figures 3.4 and 3.5. The basis chosen are first order polynomials and the maximum tetrahedra dimension $h = \lambda/3$ (not $\lambda_0/3$, as in the dielectric media the wavelength is shorter). The total computation time have been of about 38 minutes on a dual core, 2 GHz machine, with a maximum physical memory requirement of 1.5 GB.

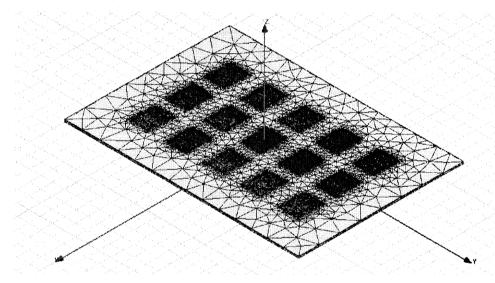


Figure 3.4: Resulting mesh of the 3 by 5 patch antennas array.

From these physical data, it has been possible to build the input system matrices of our model. Using the same interpolation basis, the resulting number of unknowns has been of 358 648. The total time required to build the full model is of about 45 minutes, and most of the time is used to build F and the solutions of the input system matrix. Before we proceed generating the output system matrices, we must select the sampling points within the FEM domain. The sampling resolution we have chosen is of $\approx \lambda_0/10$, with $\lambda_0 = 12.76$ cm at 2.35 GHz. As we have seen in section 1.3.4, such a sampling resolution provides an error of about $\varepsilon_{[\lambda/10]} \approx 2 \ 10^{-3}$, which is adequate for

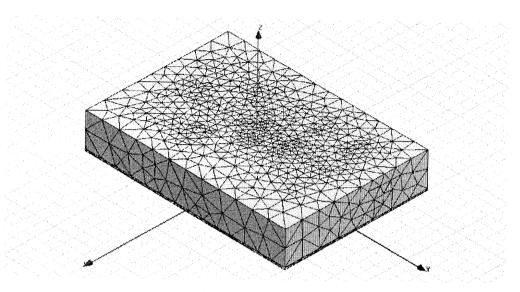


Figure 3.5: Resulting bounding box mesh of the 3 by 5 patch antennas array.

our patterns computations. Further errors are introduced by the sampling operation in a FEM domain. In effect, the field within the discretized domain suffers of discontinuities at the interfaces between the elements, especially if the fields that are approximated vary rapidly. The h-adaptive meshing have reduced this undesired behavior, but cannot eradicate it completely. Thus for better accuracy, a volume interpolation should be performed as proposed in [37], where the surface integrals are achieved through volume integrals within a thin layer that contains the fields sampling points. Also, the finite element method suffers of numerical dispersion in electrically large electromagnetic models [38], especially for high h (tetrahedra dimension) and low p (polynomial order). In our approach, we will consider these errors to be absent, remembering that they involve both the full and the reduced order models, and a comparison between these two models is unaffected by such errors. The near fields sampling grid is shown in figure 3.6. The ABCs are located $\approx 0.6 \lambda_0$ from the center of the patches and the sampling grid at $\approx \lambda_0/2$. The bottom face of the parallelepiped have been neglected, as the fields on the PEC ground plane are known to be null. The pattern computed by HFSS is shown in figure 3.7. The output system matrices, according to the results of figure 2.32, can be built with a DFT-truncation retaining only 21 coefficients. Hence, we obtain an output matrix $\mathbf{C}' \in \mathbb{C}^{358648 \times 21}$ for each po-

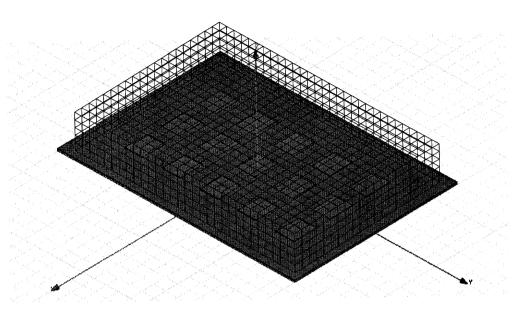


Figure 3.6: Sampling grid on the 3 by 5 patch antennas array.

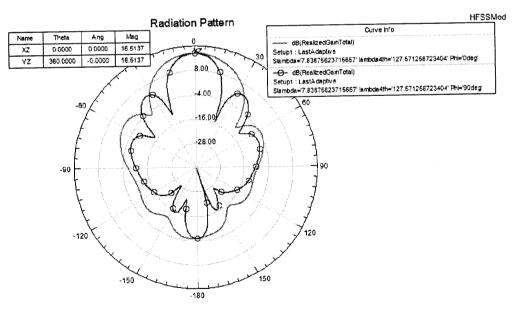


Figure 3.7: Broadside pattern of the 3 by 5 patch antennas array computed by HFSS.

Matrix	Memory [Bytes]
A	109 231 240
В	488
15 · x	86 075 520
XZ plane : $\mathbf{C}'_{\theta} + \mathbf{C}'_{\phi}$	9 037 568
YZ plane: $\mathbf{C}_{\theta}' + \mathbf{C}_{\phi}'$	9 037 568
Total	≈221MB

Table 3.1: Memory requirements for the full radiation model.

larization, and the planes of pattern computation will be the XZ and YZ cut planes. This completes our radiation model. The total memory employed, considering double precision, is detailed in table 3.1. As most of the matrices are sparse, the only real contributions are given by the FEM system matrix and the solution vectors which are not sparse. Thanks to the DFT truncation and to the few number of sampling points (1100), the output system matrix is very low memory consuming, and building a matrix of 5000 row vectors **t**, in order to compute the pattern on 5000 points within [0°, 360°) requires only 1.6 MB more. Notice that this matrix is the same for all the planes, thus has to be computed only once.

We can now build our reduced order model. Before we proceed with Galerkin projection method, we may discuss on the array configuration and how should be the near fields scanning space. According to the results of section 2.1.3, the rank for a scanning along the x direction, as we have 3 patches spaced of $\lambda_0/2$, should be of 3. Analogously, along the y direction, as we have 5 patches spaced of $\lambda_0/2$, the rank should be of 5. Thus, until we use 3 or 5 solutions pictures, for each cut-plane respectively, we expect the error to be non negligible.

To check the functionality of the ROM, we have performed a Galerkin projection selecting as scan angle the broadside direction. The resulting pattern is shown in figure 3.8. The total relative error introduced by the model order reduction is numerically null (3.395 10⁻¹³). There are some differences between the pattern computed by HFSS and ours. This is mainly due to the fact the our sampling points are distant form the radiation boundaries from which HFSS computes the far fields patterns imposing the ABCs. In effect, as we chose sampling points slightly distant from the radiation

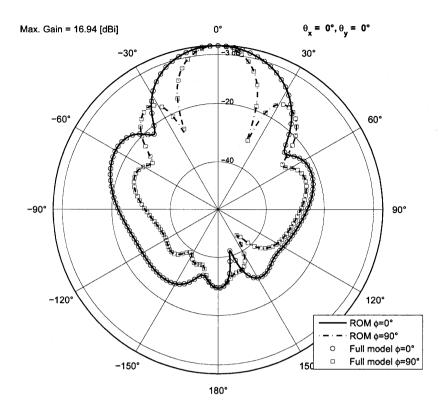


Figure 3.8: Total far fields pattern computed in the broadside direction by the full model and the ROM of the 3 by 5 patch antennas array.

boundaries (1mm close to the surface), we notice a higher similitude of our pattern with HFSS's one, as shown in figure 3.9. However, the main purpose of this thesis is to discuss the application of the projection-based model order reductions on phased arrays, with an emphasis on the reduced order dimension. Hence, we proceed working on the first near fields sampling points selection, without carrying on whether HFSS or our implementation is more accurate.

3.5.1.1 Galerkin projection method on cut planes

We have chosen the beamsteering angles to be along the y direction in which we have 5 groups of 3 antennas excited with a linear phase, each group having the same phase. The spanning scan angles are chosen equidistant in the range of $\theta_y \in [0^\circ, 90^\circ]$. Then the error is computed for 21 tested scan angles chosen in the same range, returning for each spanning space dimension the average relative error between the

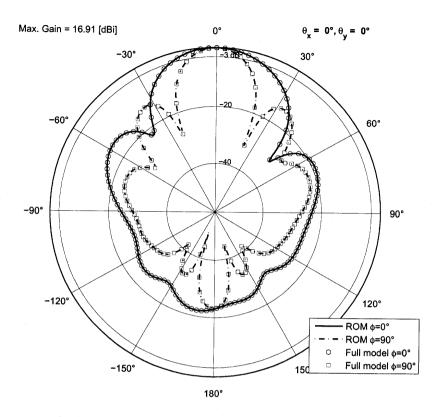


Figure 3.9: Broadside pattern computed from a surface 1mm next to the radiation boundaries.

full model and the MOR computed far fields detectors on both XZ and YZ planes. The results are shown in figure 3.10. The average error value computed with a scan angle space of dimension 4 is of 0.2189. This is still to high for accurate computations, and a steered patterns in that conditions are shown in figures 3.11 and 3.12. As we expected from conclusion 4, the pattern next to end fire is more accurate than that close to broadside. Furthermore, this behavior should be enhanced by the element factor allowing us to choose spanning angles much closer to the broadside direction. Now, we conclude discussing on the error we obtain with the ROM of order 5: it is null for all steering angles chosen along the y direction. This is the most interesting information achieved from the model order reduction of phased arrays. The final model is constituted of the following full (not sparse) matrices: $\tilde{\bf A} \in \mathbb{C}^{5\times5}$, $\tilde{\bf B} \in \mathbb{C}^{5\times15}$, $\tilde{\bf C}'_{\theta}$, $\tilde{\bf C}'_{\phi} \in \mathbb{C}^{21\times5}$ for each cut-plane, and $\tilde{\bf F} \in \mathbb{C}^{1100\times5}$, necessary in order to compute the radiated power from p=1100 near fields sampling points. 420 kB are occupied by this ROM, and we

by reflection coefficients is bunch Cheaper. Reflection coefficient is one entry in the solution vector x.

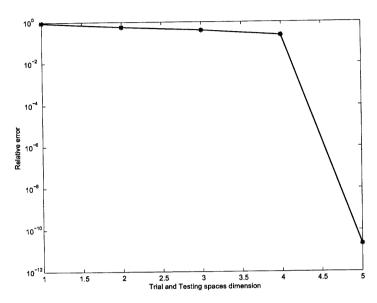


Figure 3.10: Relative error in the far fields detectors for the full model and those for the reduced model.

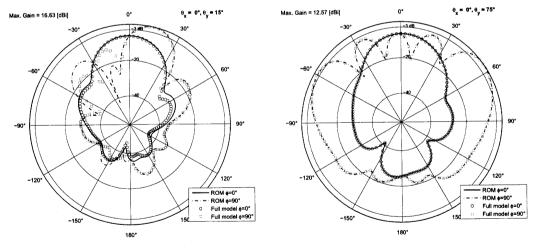


Figure 3.11: Steered pattern close to broad side computed from a ROM of order 4. Figure 3.12: Steered pattern close to end fire computed from a ROM of order 4.

only need to supply the inputs $\mathbf{u}(0^\circ, \theta_y)$ and the output trigonometric polynomials values $\mathbf{t}(\theta, \phi = 0^\circ, 90^\circ)$, the latter depending on the required pattern resolution (we used a $\Delta\theta$ of 0.036° building a matrix of about 1.6 MB in this case), to compute the cut-planes patterns. The error is now pending on the sampling resolution of $\lambda/10$, which has been considered to provide accurate patterns.

To demonstrate the benefit of the RDM you can perhaps add some nuntime comparisons between computation with original model and RDM.

3.5/1.2 Dual space projection method on cut planes

We proceed illustrating the effectiveness of the dual problem based projections. We need to compute further 21 solutions from one of the \mathbf{C}' matrices we dispose. We have chosen to match the look angles in the θ polarization, using the output matrix \mathbf{C}'_{θ} and a ROM of order 3, with matching look angles in 89.7297°, 179.8198° and 269.9099° and the test space is matched in 0°, 45° and 90°. As 11° is not an interpolation angle for both the test and trial spaces, we use that scan angle to compute the error. Figures 3.13 and 3.14 show that the pattern is matched (null error) in the look

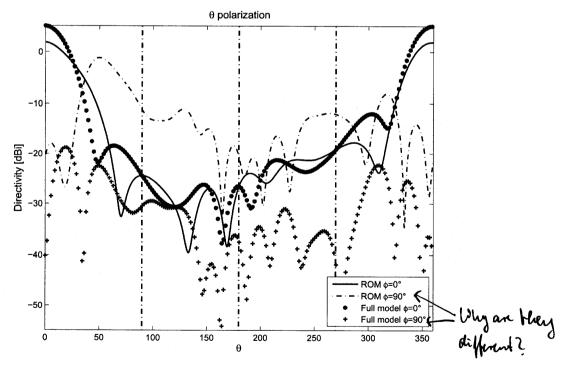


Figure 3.13: θ polarization pattern for the full model and those for the reduced model of order 3 with look angle trial space.

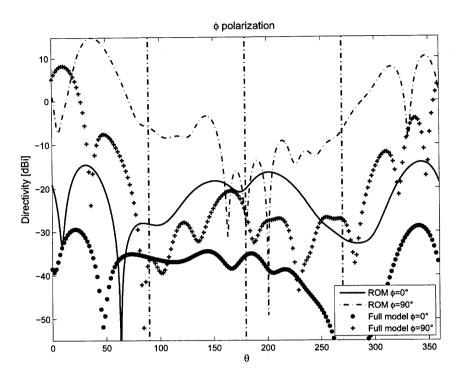


Figure 3.14: ϕ polarization pattern for the full model and those for the reduced model of order 3 with look angle trial space.

angles selected for the trial space in the XZ plane (dash-dotted vertical lines) only for the θ polarization, while the whole pattern error remains very high. We achieve only a local matching on the selected look angles and this is a result we could expect from the definition of the look angle. While the scan angle contains informations on the whole near fields on the bounding surface, the look angle, being the result of a near-to-far transformation, does not maintain any information on the near fields, hence on the total pattern. Furthermore, as the trial and the test spaces must have equal dimensions to allow the inversion of $\tilde{\bf A}$ during the solution of the reduced order input system, it will never be possible to cover the entire pattern matching the look angle space. Another non negligible negative point is associated to the fact that the number of coefficients retained with the DFT-truncation, considering all the polarizations and all the cut-planes, is typically higher than the number of radiating elements, leading in further computations that may not improve the accuracy. η_{IL}

computations that may not improve the accuracy. Remember: We do not want to solve for every radiating element, we only want to solve for those scan angles helded for the RDM construction.

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3.5.1.3 Galerkin projections for tridimensional pattern

It is our interest to see now whether the ROM behaves if we want to scan the whole tridimensional space. In a first attempt, we build the scan angle spaces from scan angles in both x and y directions, that is with 3 angles (θ_x , $\theta_y = 0^\circ$) and 5 angles in ($\theta_x = 0^\circ$, θ_y), with θ_x , $\theta_y \in [0^\circ, 90^\circ]$. The 8 selected angles, chosen properly in order to avoid the redundancy in the broadside direction, and the tested scan angles in the $\phi = 45^\circ$ are shown in figure 3.15. The resulting error, illustrated in figure 3.16, shows

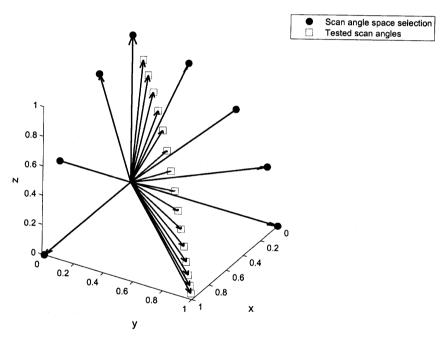


Figure 3.15: Scan angle selected for the projection space and tested angles (unit vectors of the directions).

that we cannot consider the oblique scan angles as a linear combination of the space spanned by scan angles in the x and y beamsteering directions: the error never drops below 10^{-2} .

From this results we understand that we need to select more vectors, with scan angles in the entire septentrional hemisphere. The first idea that comes to our minds is to employ a spiraling helicoidal trajectory to select the scan angles for the test space and an opposite one to test the quality of the space spanned. Those trajectories are

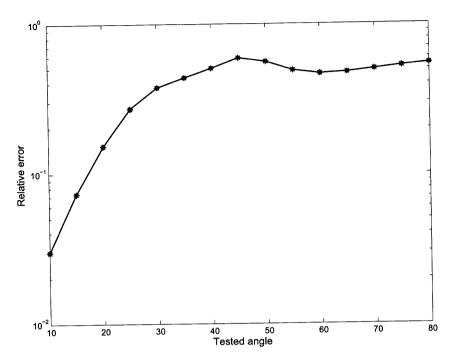


Figure 3.16: Error resulting from an oblique scan angle testing.

depicted in figure 3.17. We proceed computing the errors increasing the size of the

Selection trajectory Tested trajectory

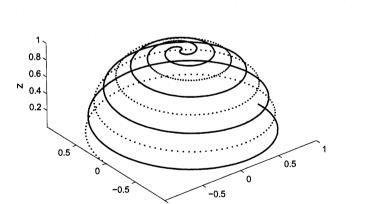


Figure 3.17: Scan angles selection trajectory and error test scan angles trajectory.

scan angle space, the results are shown in figure 3.18 First we notice that the error becomes numerically null at 15 scan angles on the selection trajectory, that is the

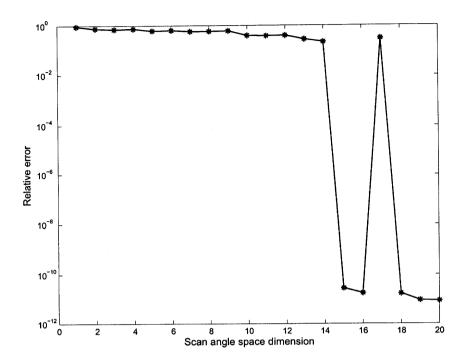
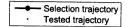


Figure 3.18: Average error on the test scan angles patterns with a space spanned with a spiraling helicoidal trajectory.

total number of array antennas. The error suddenly becomes non null with 17 scan angles. This is mainly due, as we can see in figure 3.19, to the spherical pyramid edge sampling condition we have achieved with 17 scan angles. The resulting vectors are linearly dependent, located only on x and y directions, hence the full rank of 15 is not attained. We can conclude that

Conclusion 7 The number of scan angles required in order to correctly approximate a tridimensional steering operation, all over an hemisphere, is of the number of radiating array elements. Furthermore, the scan angles must be chosen properly, in order to obtain a set of linearly independent vectors. These can be selected along a spiraling helicoidal trajectory over the septentrional hemisphere of a sphere aligned with the center of the array.

Finally, we can compute the radiation solid with the ROM of order 15. To scan the whole space, we have chosen cut-planes in the range $\phi \in [0^{\circ}, 180^{\circ})$ with a $\Delta \phi \approx 3^{\circ}$, while the sampled trigonometric polynomials matrix t covers the range $[0^{\circ}, 360^{\circ})$. The



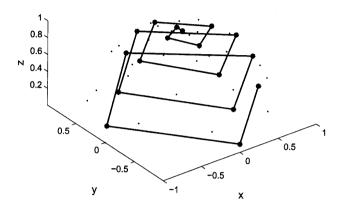


Figure 3.19: 17 scan angles selection trajectory and 31 error test scan angles trajectory.

total ROM occupies only 3.2 MB of memory, and each new scan angle is computed with the sampling resolution accuracy of \approx 2 10^{-3} in about 0.7 seconds on a dual core 2 GHz machine.

3.5.2 40 dual-polarized tapered slot antennas

To emphasize the importance of the relative spacing between the radiating elements, we now illustrate the results of the model order reduction applied to the array of in figure 3.21. It is a dual polarized array of 40 tapered slot antennas (TSAs), also called *Vivaldi* antennas, that is actually built superposing orthogonally two arrays (each ideally linearly polarized) of 4 by 5 of TSAs. The model has been designed according to [39]. Notice that in figure 3.21, the front x directed row of TSAs have some dielectrics parts hidden in order to show the feeding stripline, with radial stub termination. The radiating slots are tapered exponentially, while short circuited at the non radiating side. With the given geometrical parameters, the TSA achieves a bandwidth, in terms of *Voltage Standing Wave Ratio* (VSWR) below 2, of about 5.5 GHz when steering in the broadside direction, centered at 4 Ghz. We have chosen to build the radiation model at 3 GHz, for which the spacing between the TSAs, be-

X

Radiation solid Max. Gain = 15.8 [dBi]

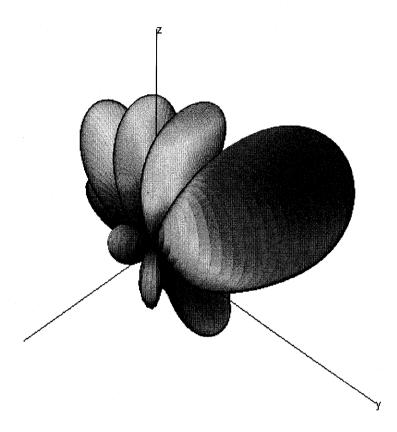


Figure 3.20: Radiation solid scanning in $(\theta_x = -10^\circ, \theta_y = 45^\circ)$ computed by the reduced order model.

ing of 2cm, is of $\approx \lambda_0/5$. The near fields sampling resolution is chosen to be of $\lambda_0/10$ and the number of coefficients retained in the DFT-truncation is of 51. To build the ROM with Galerkin projections, the steering direction chosen is x, where we have a maximum of 9 groups radiating elements, 5 being cross polarized. From the singular values of figure 3.22, we realize that the rank of the scan angle space in that direction is of 9 (for scan angles selected equidistantly in the range $\theta_x \in [0^\circ, 90^\circ]$) $\theta_y = 90^\circ$). The singular values are decreasing rapidly even before we attain the numerical rank. As we have seen in figure 2.1, the far fields detectors error was low even before the number of vectors arrived to the number of radiating elements. Thus we expect the

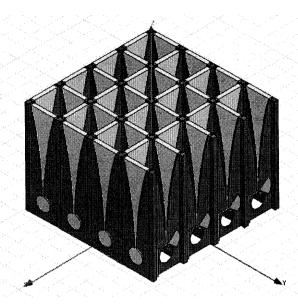


Figure 3.21: Array of tapered slot antennas equally spaced of $\approx \lambda_0/5$.

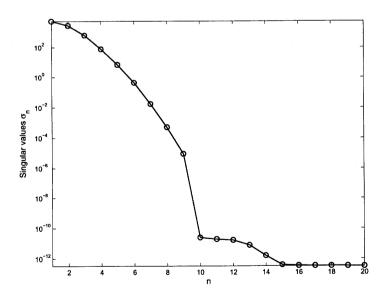


Figure 3.22: Singular values of the TSAs array steered in the x direction.

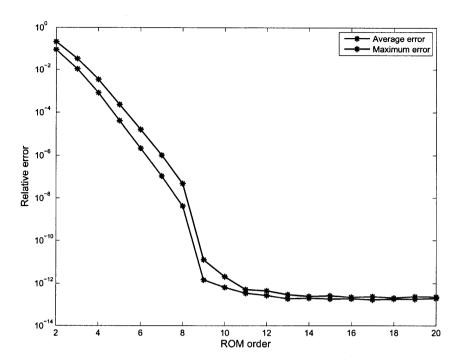


Figure 3.23: Error for several orders of the TSAs array ROM.

ROM to require an order lower than 9 while keeping the sampling resolution error. As we can see in figure 3.23, the maximum ROM approximation error, attained near the broadside direction (figure 3.24), is slightly lower than the near fields sampling resolution error once we use 5 spanning scan angles, and using 6 out of 9 renders the approximation error irrelevant to the radiation model. Figure 3.24 reports the error for several tested angles, the scan angles selected to span the projection space being of $\theta_x = 0^{\circ}$, 22.5°, 45°, 67.5°, 90°. This results in the fact we need much less than 81 spanning scan angles to scan the whole septentrional hemisphere. Figures 3.25 and 3.26 show that only 40 vectors (out of 81) are required to build an accurate ROM for tridimensional beamsteering of the tapered slot antennas array designed.

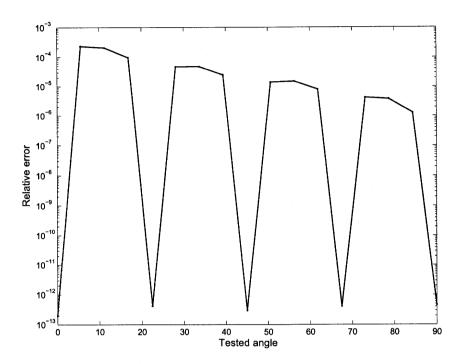


Figure 3.24: Error introduced by a ROM of order 5 for several tested angles.

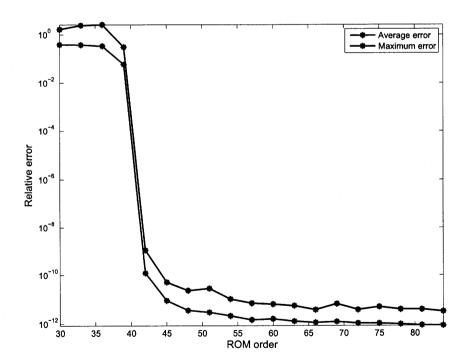


Figure 3.25: Error for several orders of ROM able to scan the whole tridimensional space.

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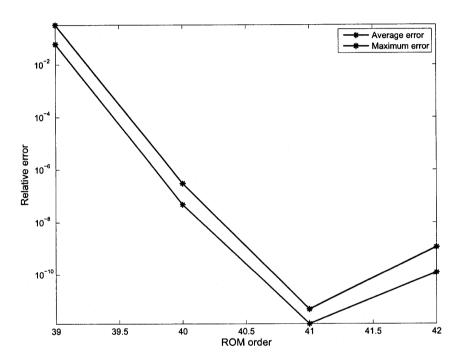


Figure 3.26: Error for 39, 40, 41 and 42 ROM orders.

CONCLUSIONS

The model order reduction techniques have demonstrated in the last two decades to be very promising in the full-wave analysis of electromagnetic structures. The most interesting issues arouse from the fact it is possible to achieve, with less computational requirements, the response of the electromagnetic structure in front of a parameter sweep [25], and even multiple parameters sweeps [32]. Well established methodologies for computing fast frequency sweeps are nowadays available. For example, the *Adaptive Lanczos-Padé Sweep* (ALPS) [35] is implemented in HFSS, confirming its robust formulation. In scattering problems [36], the incidence angles of plane waves have been parameterized in order to achieve the radar cross section of a scatterer through model order reduction.

The fundamental idea on which the projection-based model order reductions is based is the possibility to project a full order model onto a low dimensional parameterized subspace of appropriate global shape functions, keeping the necessary informations associated to that subspace. In this thesis, it has been shown that, parameterizing the input and output system vectors of the radiation model of a phased array in the scan and look angles then projecting the system onto an appropriately chosen scan angle subspace, highly efficient computations of the far fields pattern in front of a beam steering process can be achieved. The technique employed lead to reduced order models of dimensions comparable to the number of radiating elements that compose the array, where the original model, due to the high accuracy of the full-wave analysis, was several orders of magnitude higher.

According to [32], it is possible to include further polynomial parameterizations

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in finite element models of microwave structures. Taking into account the frequency le well multion lies. response of the radiating elements through, for example, an ALPS method, and in- This method is lead cluding variability of the dielectric and eventually ferroelectric materials properties by appropriate polynomial parameterization of the system matrix, may lead to multivariate reduced order models of phased arrays capable to give to the designer precious informations on the tolerances associated to physical realization.

The significant computational timings reduction, typically of several orders compared to the full model parameter sweep requirements [25], may allow fast optimization processes, either with deterministic algorithms or non-deterministic ones like the genetic algorithms, thus improving significantly the overall design process.