

MAE 263F HW5*

Deformation of a Clamped Thin Beam Using a Plate Model

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Abstract—In this homework, you I simulate the static deformation of a thin, rectangular beam under its own weight using the *discrete plate model* and compare your numerical results against the classical Euler-Bernoulli beam theory prediction.

I. PROBLEM STATEMENT

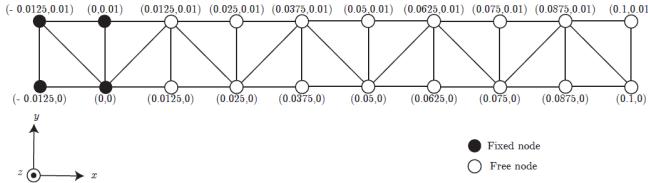


Fig. 1: Beam Mesh

The objective is to simulate the static deformation of the beam across time until the numerical solver converges. I will then report the steady displacement and include a plot of the tip displacement as a function of time.

Beam Geometry

| | |
|-----------|-----------------------|
| Length | $l = 0.1 \text{ m}$ |
| Width | $w = 0.01 \text{ m}$ |
| Thickness | $h = 0.002 \text{ m}$ |

Material Properties

| | |
|-----------------|-----------------------------------|
| Young's Modulus | $Y = 1.07 \times 10^7 \text{ Pa}$ |
| Density | $\rho = 1000 \text{ kg/m}^3$ |
| Gravity | $g = 9.81 \text{ m/s}^2$ |

Section Properties

| | |
|-----------------------|-----------------------|
| Cross-sectional Area | $A = wh$ |
| Second Moment of Area | $I = \frac{wh^3}{12}$ |
| Distributed Load | $q = \rho Ag$ |

Mesh. Use the mesh shown in the figure above. The length of the free portion of the plate is $l = 0.1 \text{ m}$. The four leftmost nodes are fixed to enforce the clamped boundary condition. Gravity acts in the negative z -direction.

Boundary Condition. The left edge of the plate (at $x = 0$) is fully clamped: all displacement and rotation components are fixed.

Plate Model Simulation. Model the domain $l \times w$ as a thin plate of thickness h . Compute the static deformation by solving the equilibrium equations until the configuration reaches steady state. Let the tip displacement at the centerline of the free edge $x = l$ be

$$d_{\text{plate}}(t) = z_{\text{tip}}(t) - z_{\text{tip}}(t = 0).$$

II. COMPARISON WITH EULER–BERNOULLI BEAM THEORY

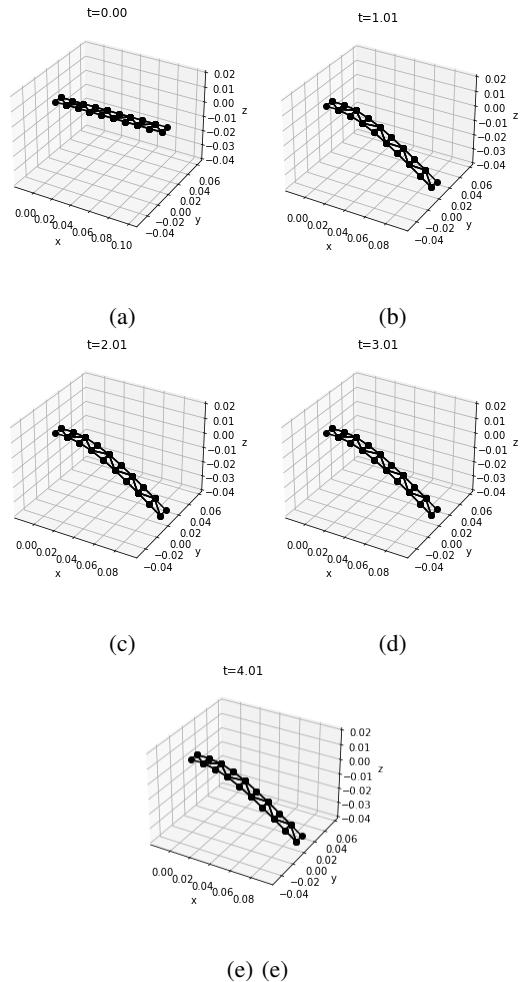


Fig. 2: Five snapshots of the helix at (a) $t = 0 \text{ s}$, (b) $t = 1.01 \text{ s}$, (c) $t = 2.01 \text{ s}$, (d) $t = 3.01 \text{ s}$, (e) $t = 4.01 \text{ s}$.

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A. Theory

For a cantilever beam under uniform load q (N/m), the Euler–Bernoulli tip displacement is $\delta_{EB} = \frac{ql^4}{8YI}$. Theoretical prediction δ_{EB} : $\delta_{EB} = \frac{ql^4}{8YI} = -0.03675m$

B. Simulation

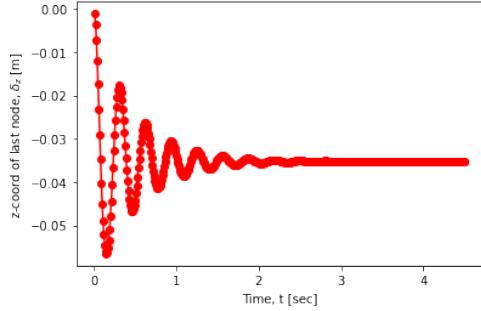


Fig. 3: δ_{plate} vs. time t

Convergence criteria:

```
if ((timeStep*dt) % 0.5 == 0) and (ctime != 0):
    endZ_1secondbefore = endZ[timeStep -int(1.0/dt)]
    endZ_diff = np.abs(endZ[timeStep] - endZ_1secondbefore)
    endZ_percent_diff = (endZ_diff/np.abs(endZ[timeStep])) * 100
    if endZ_percent_diff < 0.1:
        print("End Z position has stabilized.")
        final_timeStep = timeStep
        break
```

The program makes comparisons to a time point 0.5 seconds in the past every 0.5 seconds. If the values are within 0.1% of each other, convergence is reached.

Numerical prediction δ_{plate} : -0.035207m

C. Comparing Theory and Simulation

$$\begin{aligned}
\text{Normalized difference} &= \left| \frac{\delta_{plate} - \delta_{EB}}{\delta_{EB}} \right| \\
&= \left| \frac{-0.035207 \text{ m} - (-0.03675 \text{ m})}{-0.03675 \text{ m}} \right| \\
&\approx 0.042 \approx 4.2\%
\end{aligned}$$

The resultant end deflection value produced by numerical simulation appears to be appreciably similar to the theoretical prediction.

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