

Progress Update 03

Embry Riddle Aeronautical University - College of Business and Engineering

Team APPA



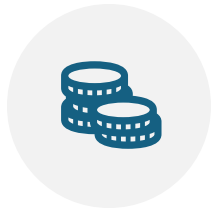
Agenda



AIRCRAFT DESIGN



SEATING
CONFIGURATIONS



UPDATED COST
ANALYSIS



ROUTE ANALYSIS



LIFT & DRAG
MODELS



STABILITY
ANALYSIS

Design Foundations



Fixed-wing regional aircraft



Up to 12 passengers



Highlight hybrid-electric technology



Reduced emissions and noise



Optimized cost for short-haul flights

Aircraft Regulations

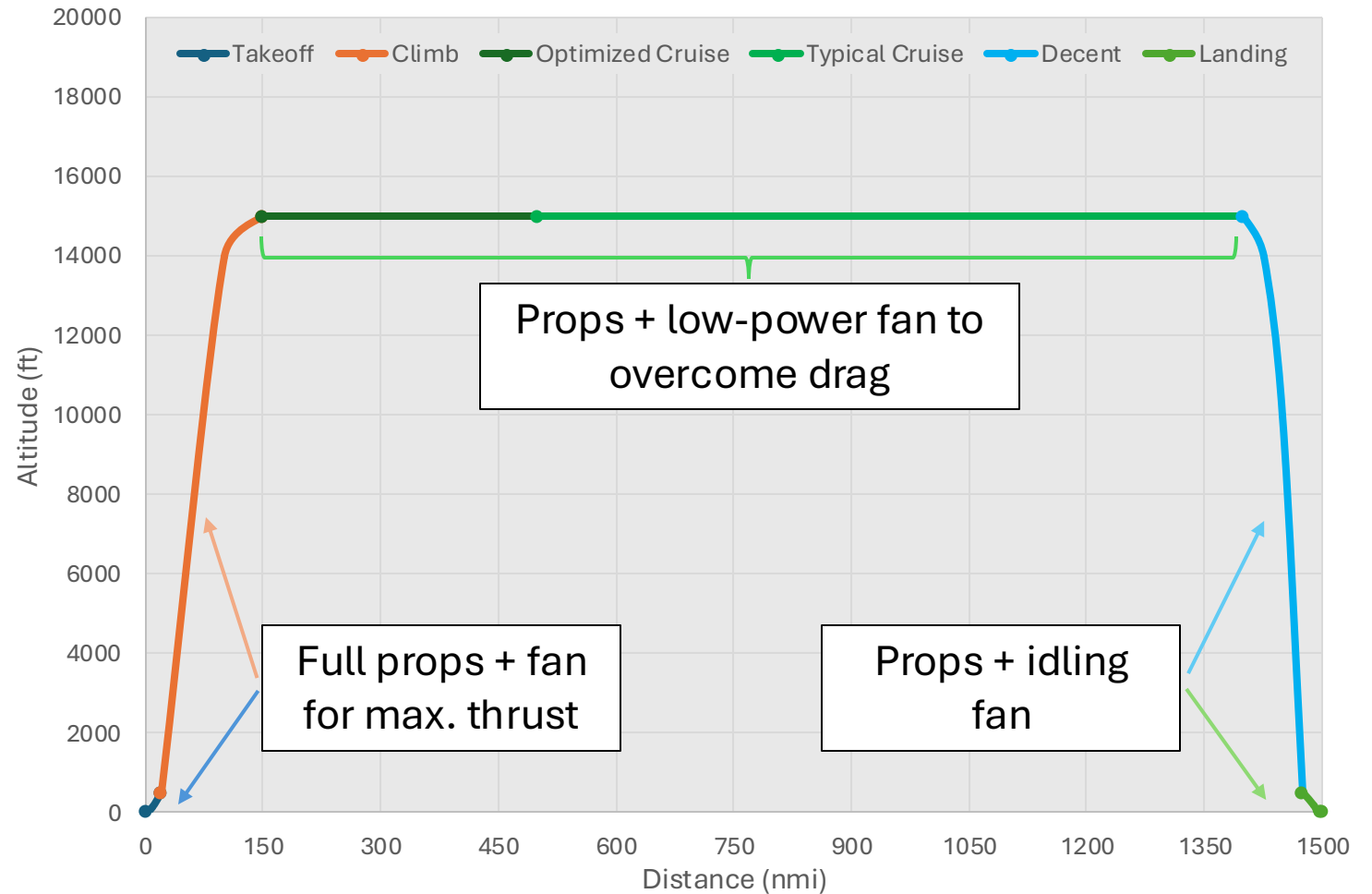


The aircraft falls under FAA 14 CFR Pt. 23, Level 4, High Speed

| Customer Requirements | | 14 CFR Part 23 Regulations | |
|-----------------------------|-------------------------------------|--|-------------|
| Passengers | 12 | Level 4 Passenger Range | 10 to 19 |
| Range | 1,500 nmi Optimized for 500 nmi | MTOW | ≤19,000 lbs |
| Technical Innovation | Hybrid-Electric Propulsion System | High Speed Category | >250 knots |
| Reduce Environmental Impact | Fuel Efficiency Noise Reductions | Service Ceiling (Oxygen and pressurization required) | 20,000 ft |

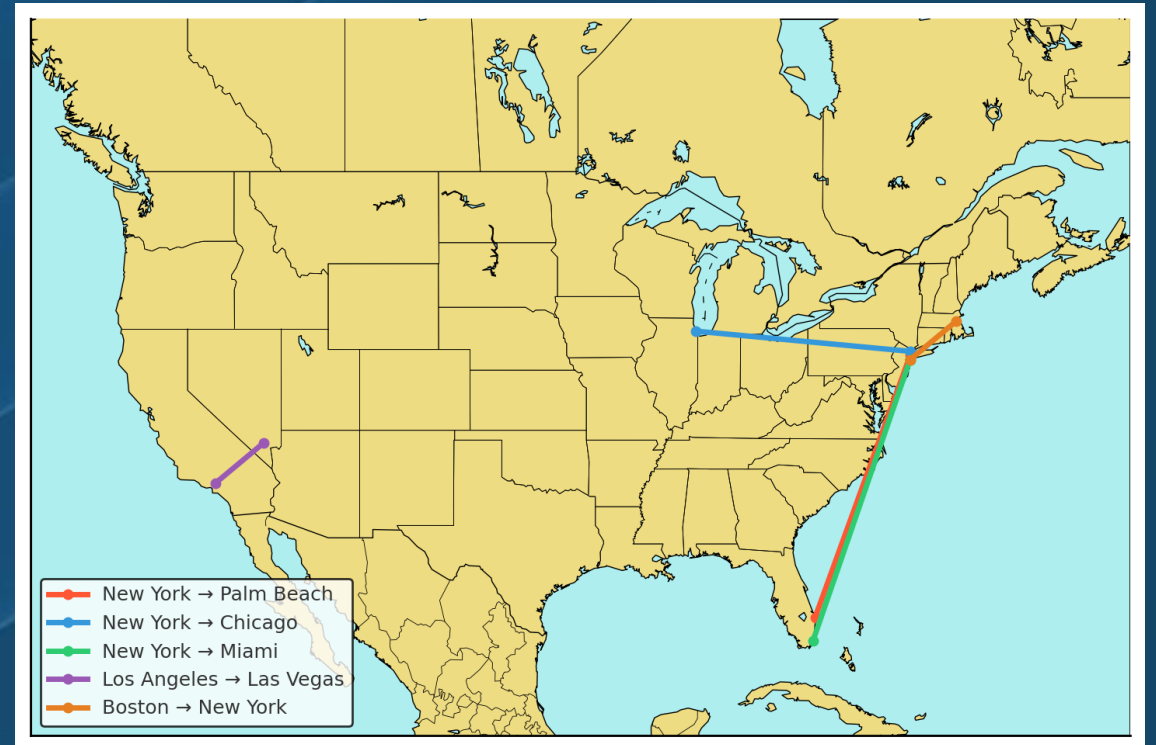
Mission Profile

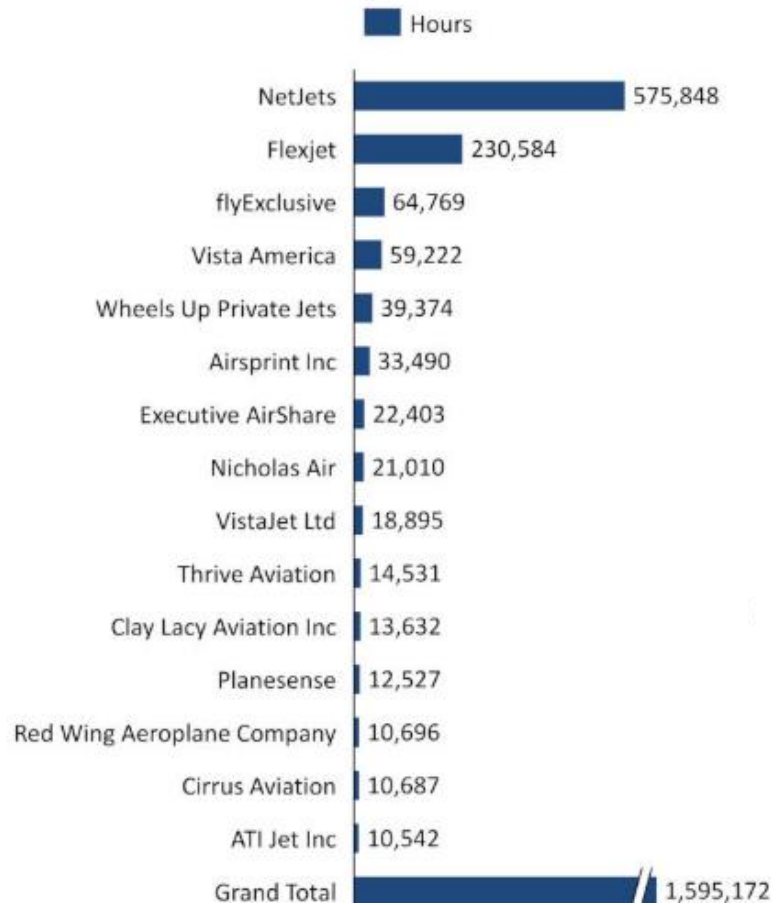
- Cruise altitude of 15,000 ft
 - Reduces climb/decent time
 - Improves fuel & propellor efficiency
- Utilizes entire hybrid system efficiently



Markets for America

- New York → Palm Beach ~ 900 NM
- New York → Chicago ~ 650 NM
- New York → Miami ~ 1,080 NM
- Los Angeles → Las Vegas ~ 202 NM
- Boston → New York ~ 190 NM





VISTA JET

NETJETS®

Market Clientele

- Fractional Owners/Operators
- Possess substantial upfront capital investment
- Demonstrated track record of safety
- Consistent market demand
- As of January 31st, fractional operators collectively operate 3,886 aircraft
- Identified market gap for shorter-range aircraft catering to regional travel needs

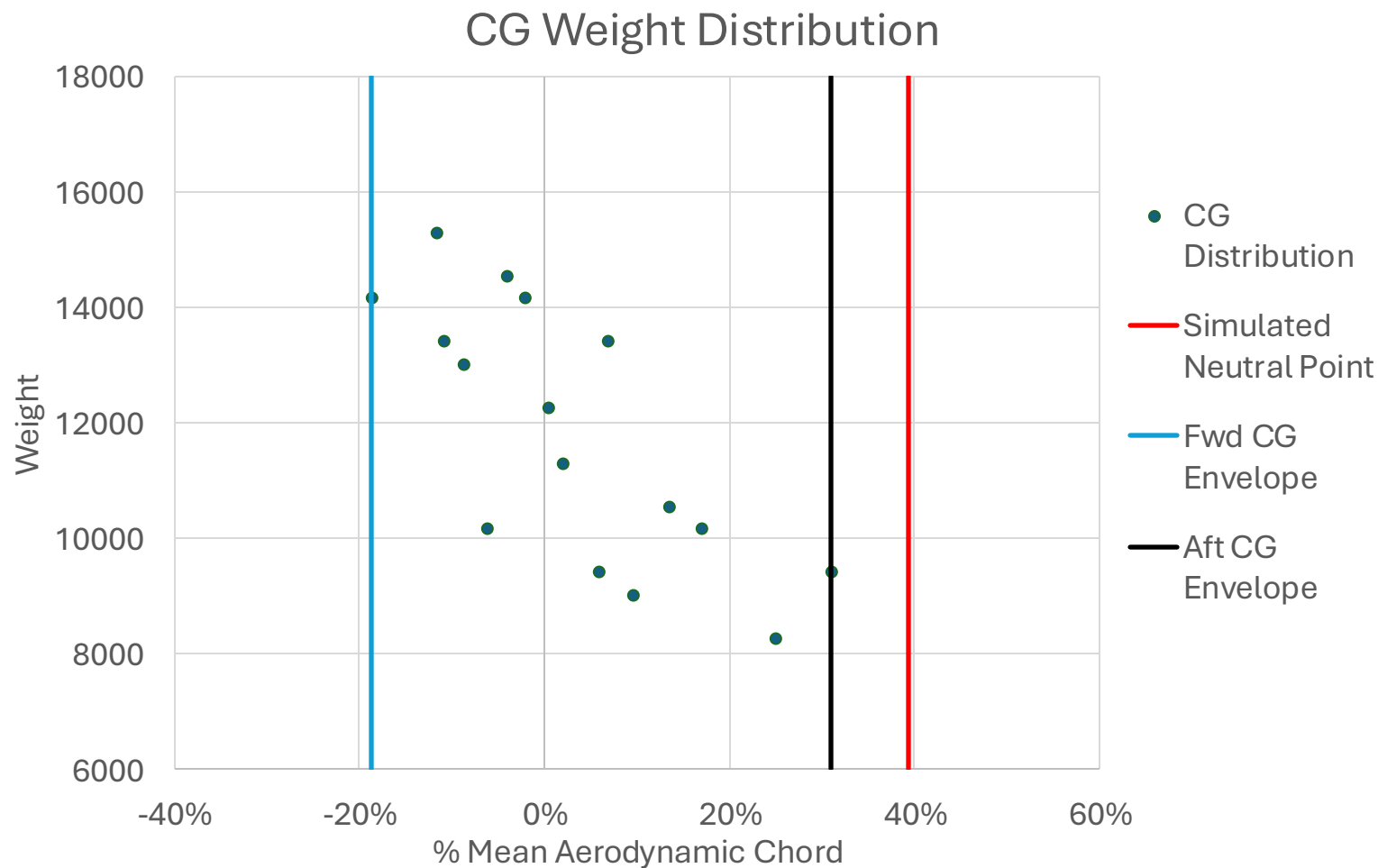
flyexclusive

FLEXJET

WHEELS UP

Weight Distribution & CG Envelope

| Type | Weight (lb) |
|------------|-------------|
| Structural | 4400 |
| Crew | 570 |
| Payload | 750 |
| | 2280 |
| | 1000 |
| Engines | 2300 |
| Fuel | 4000 |
| MTOW | 15,300 |



Weight Distribution & CG Envelope

| Loading Cloud for CG estimation | | | | | | | | | | | | | | | |
|---------------------------------|--------------|------|------|------|-------------|-----|------------|------|--------------|-----|------|-----|-------|--------|-------|
| Combination | Empty weight | | Crew | | Front Seats | | Back Seats | | Baggage Area | | Fuel | | Total | | |
| ID | We | XCGe | We | XCGe | W | arm | W | arm | W | arm | W | arm | W | moment | XCG |
| 1 | 6700 | 32.3 | 570 | 8 | 1140 | 17 | 1140 | 32.2 | 1750 | 15 | 4000 | 23 | 15300 | 395308 | 25.84 |
| 2 | 6700 | 32.3 | 570 | 8 | 1140 | 17 | 1140 | 32.2 | 1750 | 15 | 0 | 23 | 11300 | 303308 | 26.84 |
| 3 | 6700 | 32.3 | 570 | 8 | 1140 | 17 | 0 | 32.2 | 1750 | 15 | 4000 | 23 | 14160 | 358600 | 25.32 |
| 4 | 6700 | 32.3 | 570 | 8 | 1140 | 17 | 0 | 32.2 | 1750 | 15 | 0 | 23 | 10160 | 266600 | 26.24 |
| 5 | 6700 | 32.3 | 570 | 8 | 1140 | 17 | 0 | 32.2 | 1000 | 15 | 4000 | 23 | 13410 | 347350 | 25.90 |
| 6 | 6700 | 32.3 | 570 | 8 | 1140 | 17 | 0 | 32.2 | 1000 | 15 | 0 | 23 | 9410 | 255350 | 27.14 |
| 7 | 6700 | 32.3 | 570 | 8 | 1140 | 17 | 1140 | 32.2 | 1000 | 15 | 4000 | 23 | 14550 | 384058 | 26.40 |
| 8 | 6700 | 32.3 | 570 | 8 | 1140 | 17 | 1140 | 32.2 | 1000 | 15 | 0 | 23 | 10550 | 292058 | 27.68 |
| 9 | 6700 | 32.3 | 570 | 8 | 0 | 17 | 1140 | 32.2 | 1750 | 15 | 4000 | 23 | 14160 | 375928 | 26.55 |
| 10 | 6700 | 32.3 | 570 | 8 | 0 | 17 | 1140 | 32.2 | 1750 | 15 | 0 | 23 | 10160 | 283928 | 27.95 |
| 11 | 6700 | 32.3 | 570 | 8 | 0 | 17 | 1140 | 32.2 | 1000 | 15 | 4000 | 23 | 13410 | 364678 | 27.19 |
| 12 | 6700 | 32.3 | 570 | 8 | 0 | 17 | 1140 | 32.2 | 1000 | 15 | 0 | 23 | 9410 | 272678 | 28.98 |
| 13 | 6700 | 32.3 | 570 | 8 | 0 | 17 | 0 | 32.2 | 1750 | 15 | 4000 | 23 | 13020 | 339220 | 26.05 |
| 14 | 6700 | 32.3 | 570 | 8 | 0 | 17 | 0 | 32.2 | 1750 | 15 | 0 | 23 | 9020 | 247220 | 27.41 |
| 15 | 6700 | 32.3 | 570 | 8 | 0 | 17 | 0 | 32.2 | 1000 | 15 | 4000 | 23 | 12270 | 327970 | 26.73 |
| 16 | 6700 | 32.3 | 570 | 8 | 0 | 17 | 0 | 32.2 | 1000 | 15 | 0 | 23 | 8270 | 235970 | 28.53 |

Lift Model

- Based on Gudmundsson's Wing and Lift Enhancement.
 - Utilizing NACA 4415 airfoil for the wing, and Clark Y fixed slot airfoil for the LE and its single slotted fowler flap configuration for its TE.

| Weight | V_c | S_w | α_c | α_{ZL} | C_{lc} | C_{Lc} | $C_{L\alpha}$ | C_{L0} | C_L | Total Lift |
|------------|----------------------------------|-----------------------|------------|---------------|----------|----------|---------------|-----------|-------|------------|
| 15,300 lb. | $422 \frac{\text{ft}}{\text{s}}$ | 405.7 ft ² | -1.68° | -4.53° | 0.30 | 0.255 | 0.09 /deg | 0.41 /deg | 0.297 | 16,077 lb. |

Drag Model

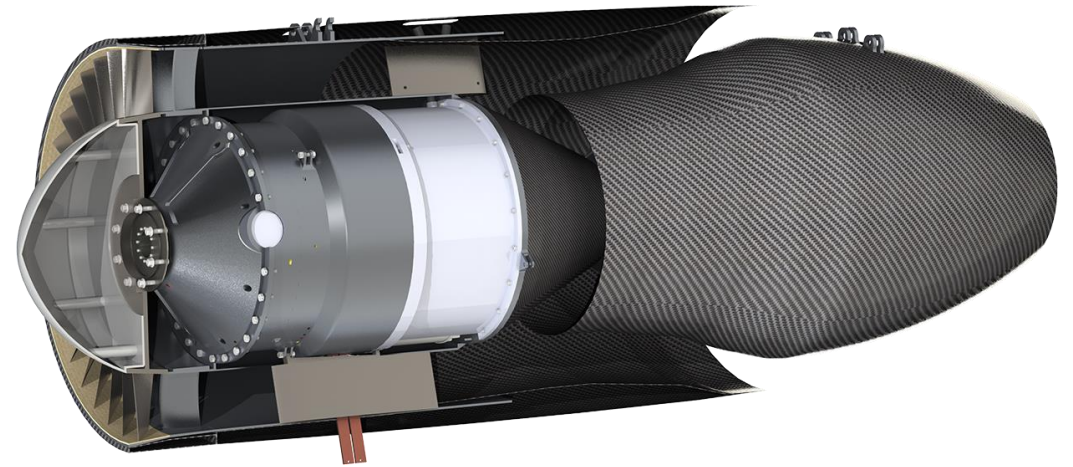
- Based on Gudmundsson's Aircraft Drag Analysis
- Cruise conditions

| | S_{ref} (ft ²) | S_{wet} (ft ²) | C_f | C_{Df} | $C_{D\text{min}}$ | C_{Di} | C_D | Drag (lb) |
|----------|-------------------------------------|-------------------------------------|----------|----------|-------------------|----------|----------|-----------|
| Wing | 405.7 | 868.2 | 1.92E-03 | 4.11E-03 | 7.33E-03 | 4.23E-03 | 3.47E-02 | 1875.5 |
| Tail | 352.9 | 727.1 | 1.88E-03 | 3.87E-03 | 6.25E-03 | | | |
| Fuselage | 40.72 | 588.1 | 1.12E-03 | 1.62E-02 | 6.78E-03 | | | |

Propulsion

- **VerdeGo Aero VH-5**
 - Hybrid-electric turbofan
 - Can generate power in series and/or parallel
 - Generates forward thrust and electrical power during cruise
 - Entry Service Date: 2031
- **Electric Puller-Props**
 - Driven by two MAGIDRIVE 500 kW motors
 - Powered by the Amprius High Power Battery
 - Two 105" Diameter McCauley 4-Bladed Propellers
 - Will be able to be feathered if not in use ($e=0.8$)
 - Utilizing larger props as they have a higher efficiency than smaller props

Will Generate a Combined Thrust of around 5600lbs at SSL



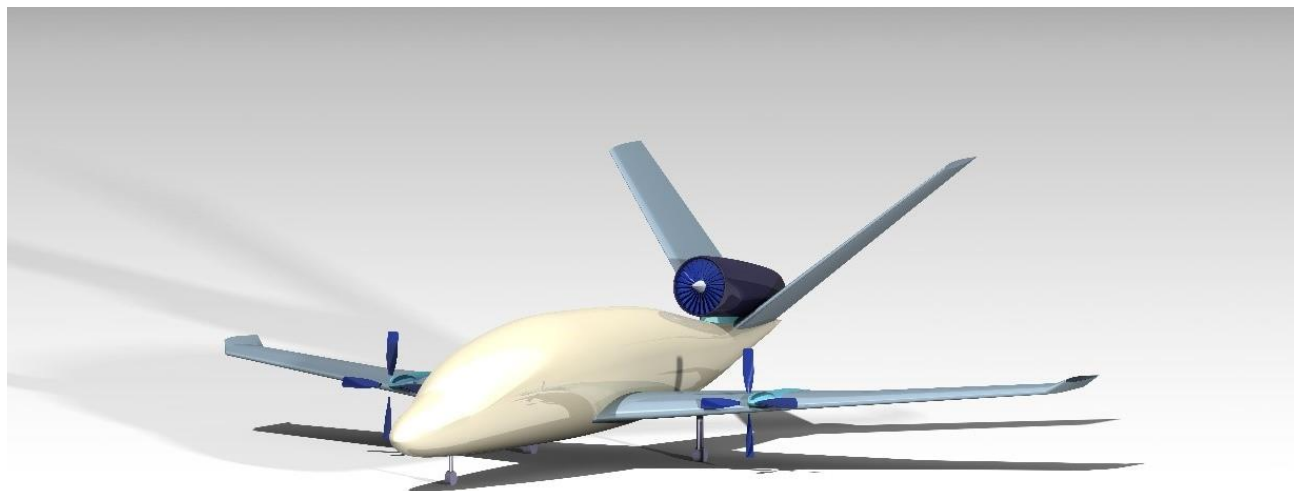
Performance

- Targeted Final Goals vs current calculations.

| | Current Calculation | Targeted Goal |
|----------------------------------|---------------------|---------------|
| T/W | 0.37 | 0.3-0.4 |
| L/D | 8.6 | 10-12 |
| MTOW | 15,300 lb | 15,000 lb |
| Combined Max thrust at cruise | 3,663 lbf | 4,000 lbf |
| Ground Roll | 3,100 ft | 3,000 ft |
| Thrust Required at cruise | 1,407 lbs | 1,000 lbs |

Aircraft Model

- Finalized seating and sizing
- CATIA model and drawings in progress



| Parameter | Value | Units |
|----------------------|-------|-----------------|
| Wing Span | 58 | ft ² |
| Wing Area | 405.7 | ft ² |
| Fuselage Length | 52 | ft |
| Fuselage Max Width | 8.16 | ft |
| Projected HT Span | 21 | ft |
| V-Tail Area (Actual) | 353 | ft ² |

Initial VSP Stability

| Description | Variable | Value | units |
|------------------------|-----------------|--------|-------|
| Neutral Point | X_{neu} | 29.573 | ft |
| Static Margin | SM | .2415 | - |
| Steady-State Roll Rate | $p_{s.s.}$ | 11.57 | deg/s |
| Max Lift Coefficient | $C_{L,max}$ | 1.8 | - |
| Stall Speed | V_s | 79.4 | kts |
| Trim AoA | α_{trim} | 1.5 | deg |
| Ruddervator Deflection | δ_v | 4.405 | deg |
| “Elevator” Deflection | δ_e | 3.115 | deg |

Initial VSP Stability

| Mode | Natural Frequency (w_n) | Damping Ratio (ζ) |
|------------------|-----------------------------|---------------------------|
| Short Period | 8.6e-5 | 1.4e5 |
| Phugoid | 0.26 | 0.0133 |
| Dutch Roll | .000197 | 425.8 |
| Spiral Stability | Spirally Stable | |

Cost Analysis continued



**Certification
Cost:**
**\$146,000,000
for 500 AC**



**Development
Cost:**
**\$9,918,000 For
500 AC**



**Manufacture/
Production
Cost:**
**\$304,590,000
for 500 AC**

\$610,000 per AC



**Break Even
Point: 427
units.**

Total Fixed Cost:
\$146,051,000

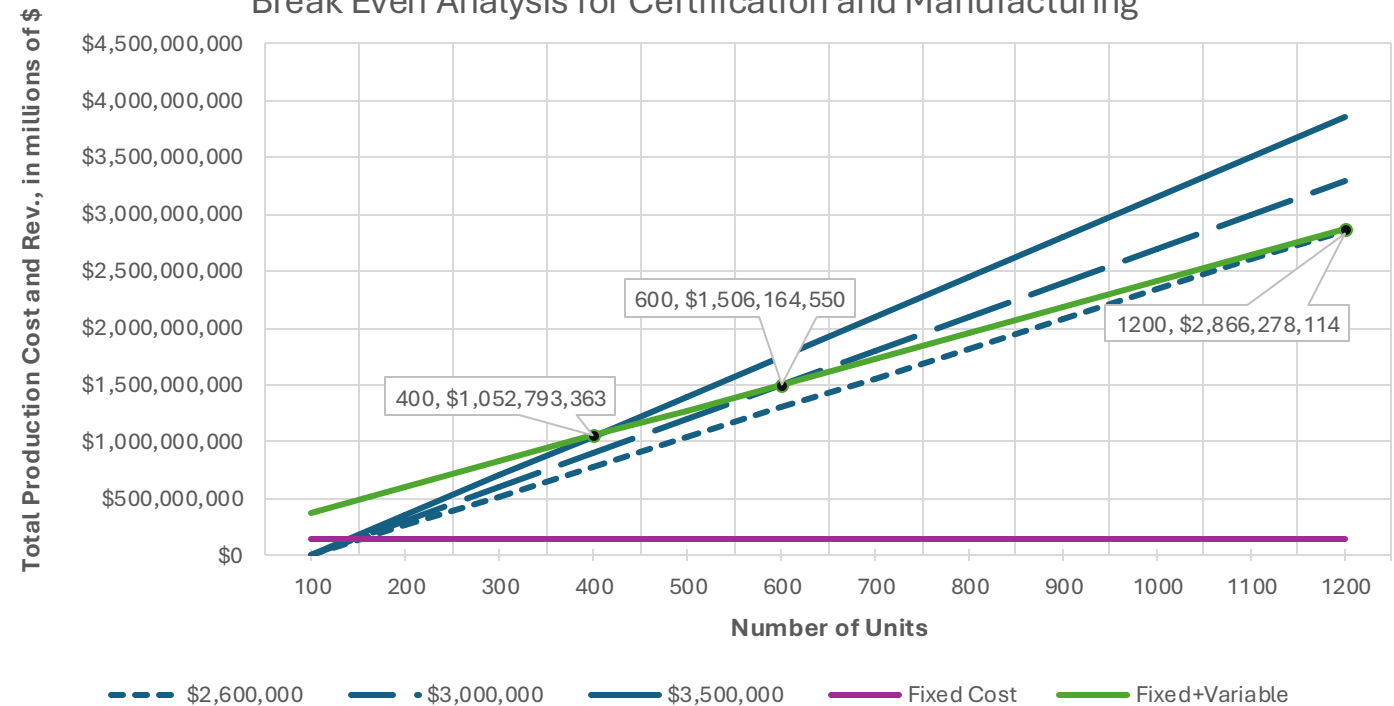
Unit Variable Cost:
\$2,267,000



**Operational:
\$545,000 per
year**

\$990 per flight hour

Break Even Analysis for Certification and Manufacturing



Cost Analysis continued



Targeting regional airlines & charter markets.



Maintenance cost advantages over legacy aircraft.

Cost analysis – Train vs Air

- Planes offer significantly faster long-distance travel (e.g., Paris to Rome in 2 hours vs. 11+ hours by train)
- Allows more time at the destination – maximize vacation or work time
- Aircraft can provide a premium cabin experience with better seating, service, and amenities
- In-flight entertainment, Wi-Fi, and refreshments
- Business-class or upgraded seating = better comfort than most train options
- American tourists who prefer air travel over trains



Understanding Marketing Expenses

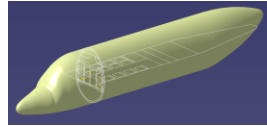
| Company | SG&A | Revenue | Selling & Marketing Expenses | % of SG&A | % of Revenue |
|---------------------------------|---------|-----------|------------------------------|-----------|--------------|
| Bombardier Inc | \$478 | \$8,665 | | | |
| Cirrus Aircraft | \$117 | \$475 | \$57 | 49% | 12% |
| Embraer | \$508.6 | \$6,396.0 | \$309.7 | 61% | 5% |

Marketing Budget

Budgeting (in millions)

| | | |
|---------------------------|--------|--------|
| Branding | 10.0% | \$1.80 |
| Digital Campaign | 25.0% | \$4.50 |
| Traditional Advertisement | 12.5% | \$2.25 |
| Events & Trade Shows | 20.0% | \$3.60 |
| Customer Engagement | 15.0% | \$2.70 |
| Partnerships | 17.0% | \$3.06 |
| Miscellaneous | 0.5% | \$0.09 |
| | 100.0% | \$18 |

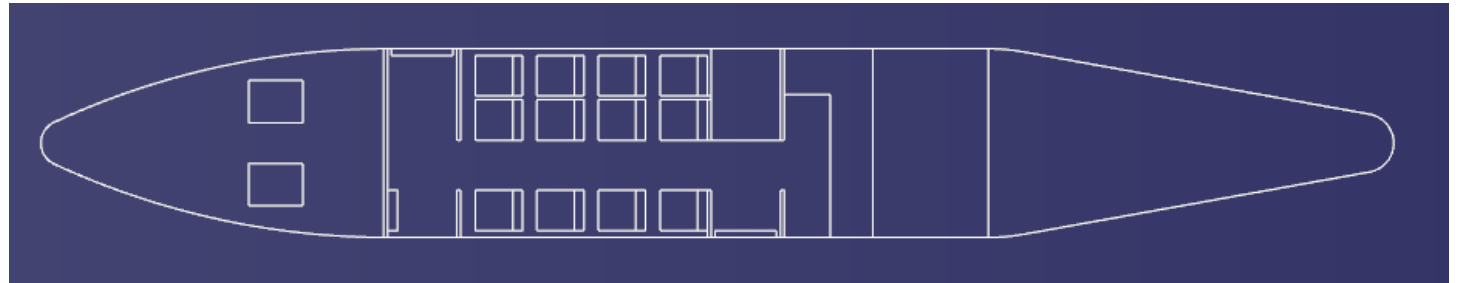
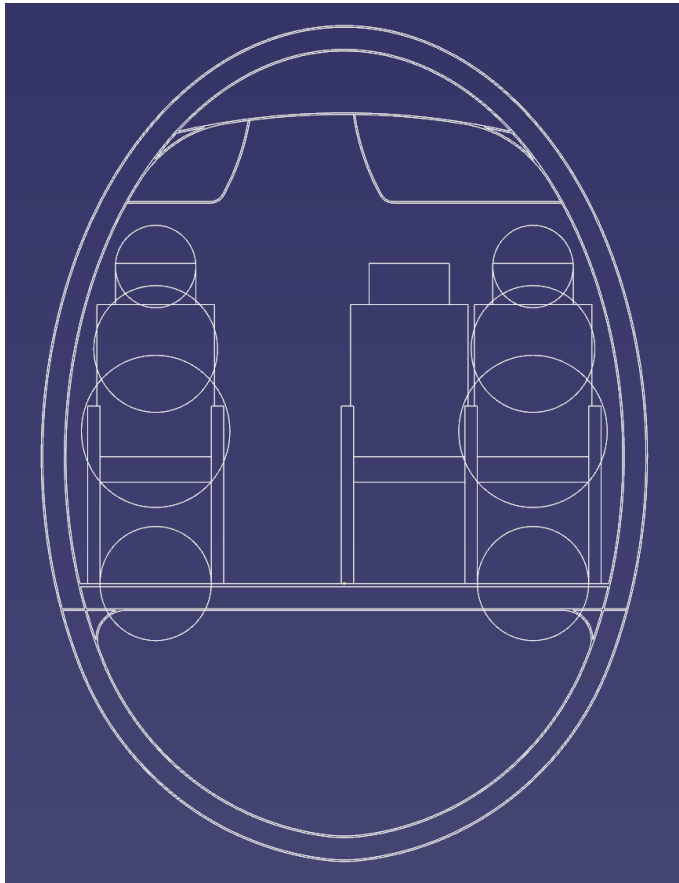
CATIA Layout



Cabin Height – 74"

Cabin Width – 90.5"

- Lavatory + Galley in Rear
- Cargo Area



- 2 Pilots
- Jump Seat for Crew
- 12 PAX

Seat Width – 19"

Seat Pitch – 32"

Aisle Width – 19"

Seating in Economy style

❖ Seat models

- Geven Essenza
- Recaro SL3710
- Haeco Vector Light

❖ Lightweight seats

❖ Features

- 27-29" pitch
- Anti corrosion material
- Entertainment capabilities





Seating in Business

❖ Seat models

- Geven Comoda
- Collins Air Rest Recliner
- Haeco Eclipse

❖ Features

- 60 inch recline
- Lightweight
- In seat tray table

❖ Option for club seating



Divan

- ❖ Built in bench style seating
- ❖ Equipped with latest safety features
- ❖ Can unfold into beds on certain models
- ❖ Provide storage capabilities

Development Plan

- Next Steps:
 - Refine
 - Stability and Control
 - Performance
 - Aerodynamics
 - Costs
 - Finalize Parameters
 - Simulations

