MPC for Robot Manipulators With Integral Sliding Modes Generation

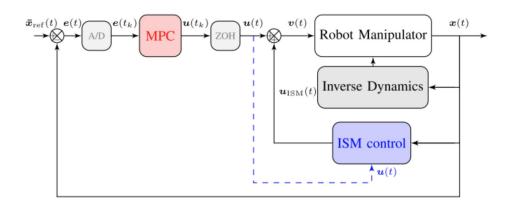


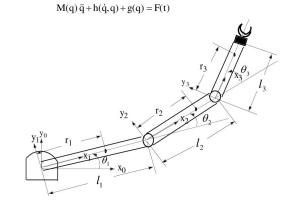
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Problem Statement

- MPC struggles with robot systems due to:
 - Nonlinear MIMO dynamics
 - Modelling uncertainties
 - High computational cost
- Proposed Solution:
 - Hierarchical control: Inverse Dynamics + ISM + MPC
 - ISM handles uncertainties first
 - Leads to a simplified MPC optimization
- Key Novelty:
 - ISM enforces sliding modes from initial time, creating uncertainty-free dynamics
 - Multi-rate control (fast ISM, slower MPC)
 - Significantly reduced computational complexity

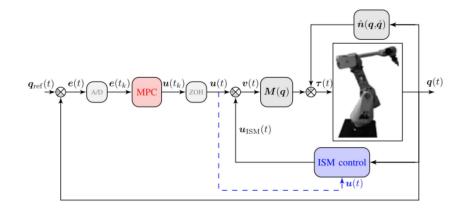




Controller Overview

Three Layer Control Architecture

- 1) Inner Loop: Inverse Dynamics Controller
 - a) Linearizes nonlinear MIMO system
 - b) Transforms to n decoupled double integrators
- 2) Middle Loop: Integral Sliding Mode (ISM)
 - a) Rejects matched uncertainties
 - b) Fast sampling rate (high frequency)
- 3) Outer Loop: MPC
 - a) Optimal trajectory tracking
 - b) Handles constraints
- 4) Combine Control Law:
 - a) $v = u_MPC + u_ISM$



The full coupled robot model:

$$\underbrace{M(q)}_{ ext{inertia matrix}} \ddot{q} + \underbrace{n(q,\dot{q})}_{ ext{Coriolis/centripetal gravity + friction}} = au$$

$$e = q_{\rm ref} - q$$

$I_2 = I_{zz2} + m_2 * (r_{z2}^2 + r_{y2}^2) + (m_3 + m_4 + m_5 + m_6) * a_2^2 ;$ $I_3 = -I_{zz2} + I_{yy2} + (m_3 + m_4 + m_3 + m_4) * a_2^2 ;$	Table A4. The expressions giving the elements of the kinetic energy matrix. (The Abbreviated Expressions have units of kg-m ² .)	$+I_{16} * \{C223 * C5 - S223 * C4 * S5\} + I_{21} * SC23 * CC4$ $+I_{20} * \{(1 + CC4) * SC23 * SS5 - \{1 - 2 * SS23\} * C4 * SC5\}$ $+I_{22} * \{(1 - 2 * SS23) * C5 - 2 * SC23 * C4 * S5\}\}$	$b_{224} = 2 * \{-I_{16} * C3 * S4 * S5 + I_{20} * SC4 * SS5 + I_{21} * SC4 - I_{22} * S4 * S5\};$	$b_{634} = b_{624}$. $b_{635} = b_{625}$. $b_{634} = 0$. $b_{645} = b_{456}$. $b_{646} = 0$.	
$m_2 * r_{e2}^2 - m_2 * r_{y2}^2$ ($+I_{10}*(1-2*SS23)+I_{11}*(1-2*SS2);$	$\approx -2.48 \times 10^{-5} \cdot C3 \cdot S4 \cdot S5$.	0643 = 0436 - 1046 - 1	
$I_4 = m_2 * \tau_{z2} * (d_2 + \tau_{z2}) + m_3 * a_2 * \tau_{z3} + (m_3 + m_4 + m_5 + m_6) * a_2 * (d_2 + d_3) ;$	$a_{11} = I_{m1} + I_1 + I_2 * CC2 + I_7 * SS23 + I_{10} * SC23 + I_{11} * SC2$ $+I_{20} * (SS5 * (SS23 * (1 + CC4) - 1) - 2 * SC23 * C4 * SC5)$ $+I_{21} * SS23 * CC4 + 2 * (I_5 * C2 * S23 + I_{12} * C2 * C23)$	$\approx -2.76 \cdot SC2 + 7.44 \times 10^{-1} \cdot C223 + 0.60 \cdot SC23 - 2.13 \times 10^{-2} \cdot (1 - 2 \cdot SS23)$.	$b_{125} = 2 * \{-I_{15} * S5 + I_{16} * (C3 * C4 * C5 - S3 * S5) + I_{10} * SS4 * SC5 + I_{12} * C4 * C5\};$	Table A6. The expressions for the terms of the centrifugal matrix. (The Abbreviated Expressions have units of kg-m ² .)	
$I_5 = -m_5 * a_2 * r_{y5} + (m_4 + m_5 + m_6) * a_2 * d_4 + m_4 * a_2 * r_{z4};$	$+I_{15} * (SS23 * C5 + SC23 * C4 * S5)$	$b_{115} = 2 * \{I_5 * C2 * C23 + I_7 * SC23 - I_{12} * C2 * S23\}$	$\approx -2.50 \times 10^{-5} * S5 + 2.48 \times 10^{-5} * (C3 * C4 * C5 - S3 * S5)$.	c ₁₁ = 0.	
$I_6 = I_{zz5} + m_5 * r_{y5}^2 + m_4 * a_5^2 + m_4 * (d_4 + r_{c4})^2 + I_{yy4} + m_5 * a_5^2 + m_5 * d_4^2 + I_{zz5} + m_6 * a_2^2 + m_6 * d_4^2$	$+I_{16} * C2 * (S23 * C5 + C23 * C4 * S5)$ $+I_{16} * S4 * S5 + I_{12} * (SC23 * C5 + CC23 * C4 * S5));$	$+I_{15} * (2 * SC23 * C5 + (1 - 2 * SS23) * C4 * S5)$ $+I_{16} * C2 * (C23 * C5 - S23 * C4 * S5) + I_{21} * SC23 * CC4$	$b_{226} = 0$, $b_{254} = b_{224}$,	$c_{13} = +14 \cdot C2 - 18 \cdot S23 - 19 \cdot S2 + 113 \cdot C23$	
$+m_6 \bullet r_{z6}^2 + I_{xx6}$ 1	≈ 2.57 + 1.38 • CC2 + 0.30 • SS23 + 7.44×10 ⁻¹ • C2 • S23 .	$+I_{20} * ((1 + CC4) * SC23 * SS5 - (1 - 2 * SS23) * C4 * SC5)$	$b_{255} = b_{225}$, $b_{256} = 0$,	$+I_{15} * S23 * S4 * S5 + I_{16} * C2 * S4 * S5$	
$I_7 = m_5 \cdot r_{y,5}^2 + I_{xx5} - I_{yy5} + m_4 \cdot r_{z4}^2 + 2 \cdot m_4 \cdot d_4 \cdot r_{z4}$ $+ (m_4 + m_5 + m_6) \cdot (d_4^2 - a_5^2) + I_{yy4} - I_{zz4} + I_{zz5}$ $-I_{yy5} + m_6 \cdot r_{z6}^2 - I_{zx6} + I_{zx6}^2$	$\begin{array}{lll} a_{12} = I_4 * S2 + I_6 * C23 + I_9 * C2 + I_{12} * S23 - I_{15} * C23 * S4 * S5 \\ + I_{16} * S2 * S4 * S5 + I_{18} * \{S23 * C4 * S5 - C23 * C5\} \end{array}$	$+I_{22} * (\{1-2*SS23\} * C5-2*SC23*C4*S5\})$ $+I_{10} * \{1-2*SS23\} ;$ $\approx 7.44 \times 10^{-1} * C2*C23 + 0.60*SC23$	$b_{245} = 2 * \{-I_{15} * S4 * C5 - I_{14} * S3 * S4 * C5\} -I_{17} * S4 + I_{20} * S4 * (1 - 2 * SS5); \approx 0.$	$+I_{15} * (C23 * C4 * S5 + S23 * C5) + I_{19} * C23 * SC4$ $+I_{29} * S4 * (C23 * C4 * C5 - S23 * SC5)$ $+I_{29} * C23 * S4 * S5 ;$	
$I_4 = -m_4 * (d_2 + d_5) * (d_4 + r_{c4}) - (m_5 + m_6) * (d_2 + d_5) * d_4$	$+I_{10} * S23 * SC4 + I_{20} * S4 * (S23 * C4 * CC5 + C23 * SC5)$ $+I_{22} * S23 * S4 * S5;$	$+2.20\times10^{-2} * G2 * S23 - 2.13\times10^{-2} * (1-2 * SS23)$.	$b_{246} = I_{25} * C4 * S5 ; \approx 0 .$	$\approx 6.90 \times 10^{-1} \cdot C2 + 1.34 \times 10^{-1} \cdot S23 - 2.38 \times 10^{-2} \cdot S2$.	
$m_3 * r_{y3} * r_{c3} + m_3 * (d_2 + d_3) * r_{y3}$;	$\approx 6.90 \times 10^{-1} * S2 - 1.34 \times 10^{-1} * C23 + 2.38 \times 10^{-2} * C2$,	$b_{114} = 2 * \{-I_{15} * SC23 * S4 * S5 - I_{16} * C2 * C23 * S4 * S5 + I_{18} * C4 * S5 - I_{92} * \{SS23 * SS5 * SC4 - SC23 * S4 * SC5\}$	$b_{254} = I_{95} * S4 * C5 ; \approx 0 .$	$c_{12} = 0.5 * b_{123}$,	
$I_0 = m_2 * r_{\sqrt{2}} * (d_2 + r_{/2});$	$a_{13} = I_5 * C23 + I_{13} * S23 - I_{15} * C23 * S4 * S5 + I_{19} * S23 * SC4$	-I ₂₂ * GC23 * S4 * S5 - I ₂₁ * SS23 * SC4 - SC23 * S4 * SC5)	$b_{312} = 0$. $b_{313} = 0$.	$c_{14} = -I_{15} \cdot S23 \cdot S4 \cdot S5 - I_{16} \cdot C2 \cdot S4 \cdot S5$	
$I_{10} = 2 * m_4 * a_5 * r_{24} + 2 * (m_4 + m_5 + m_6) * a_5 * d_4 ;$	$+I_{18} * (S23 * C4 * S5 - C23 * C5) + I_{29} * S23 * S4 * S5$ $+I_{29} * S4 * (S23 * C4 * CC5 + C23 * SC5);$	$\approx -2.50 \times 10^{-5} * SC23 * S4 * S5 + 8.60 \times 10^{-4} * C4 * S5$	$b_{314} = 2 * \{-I_{35} * C23 * C4 * S5 + I_{22} * S23 * C4 * S5 + I_{20} * (S23 * (CC5 * CC4 - 0.5) + C23 * C4 * SC5)\}$	$+I_{18} * C23 * C4 * S5 + I_{20} * S23 * S4 * SC5$	
$I_{11} = -2 \cdot m_2 \cdot r_{x2} \cdot r_{y2} ;$	= -1.34×10 ⁻¹ * C23 + -3.97×10 ⁻⁵ * S23 .	- 2.48×10 ⁻⁵ * C2 * C23 * S4 * S5 .	+I ₁₄ * S23 + I ₁₀ * S23 * (1 - (2 * SS4)) 1	$-I_{22} \cdot C23 \cdot S4 \cdot S5$; ≈ 0 .	
$I_{12} = (m_4 + m_5 + m_9) * a_2 * a_5;$	$a_{14} = I_{14} * C23 + I_{15} * S23 * C4 * S5 + I_{16} * C2 * C4 * S5$	$b_{115} = 2 * \{I_{20} * (SC5 * (CC4 * (1 - CC23) - CC23) - SC23 * C4 * (1 - 2 * SS5)\} - I_{15} * (SS23 * S5 - SC23 * C4 * C5) $	$\approx -2.50 \times 10^{-3} \bullet C23 \bullet C4 \bullet S5 \ + \ 1.64 \times 10^{-3} \bullet S23$	$c_{15} = -I_{18} \cdot S23 \cdot S4 \cdot S5 - I_{14} \cdot C2 \cdot S4 \cdot S5 ;$ $+I_{14} \cdot (S23 \cdot C5 + C23 \cdot C4 \cdot S5) - I_{23} \cdot C23 \cdot S4 \cdot S5$	
$I_{15} = (m_4 + m_5 + m_6) * a_5 * (d_2 + d_3);$ $I_{15} = I_{15} + I_{15} + I_{15} *$	$+I_{18} * C23 * S4 * S5 - I_{20} * (S23 * C4 * SC5 + C23 * SS5)$	$-I_{14} * C2 * (S23 * S5 - C23 * C4 * C5) + I_{18} * S4 * C5$	+ 0.30×10 ⁻³ • S23 • (1 - 2 • SS4) .	≈ 0.	
$I_{14} = I_{xx4} + I_{yy5} + I_{xx6}$; $I_{15} = m_6 * d_4 * r_{x6}$;	$+I_{22} \cdot C23 \cdot C4 \cdot S5$; ≈ 0 .	$+I_{22} * (CC23 * C4 * C5 - SC23 * S5)$; $\approx -2.50 \times 10^{-5} * (SS23 * S5 - SC23 * C4 * C5)$	$b_{345} = 2 * (-I_{15} * C23 * S4 * C5 + I_{22} * S23 * S4 * C5)$ $-I_{17} * C23 * S4$	$c_{26} = 0$, $c_{21} = -0.5 \cdot b_{112}$,	
$I_{16} = m_6 * a_2 * r_{16} ;$	$a_{15} = I_{15} * S23 * S4 * C5 + I_{16} * C2 * S4 * C5 + I_{17} * S23 * S4 + I_{18} * (S23 * S5 - C23 * C4 * C5) + I_{22} * C23 * S4 * C5;$	$-2.48 \times 10^{-3} \cdot C2 \cdot (S23 \cdot S5 - C23 \cdot C4 \cdot C5)$	$+I_{20} * S4 * (C23 * (1 - 2 * SS5) - 2 * S23 * C4 * SC5);$	$c_{22} = 0$. $c_{23} = 0.5 * b_{223}$.	
$I_{17} = I_{245} + I_{446} + m_6 \cdot r_{24}^2$;	≈ O.	+ 8.60×10 ⁻⁴ + S4 + C5.	$\approx -2.50 \times 10^{-5} * C23 * S4 * C5 - 6.42 \times 10^{-4} * C23 * S4$.	$c_{24} = -I_{15} * C4 * S5 - I_{16} * S3 * C4 * S5 + I_{20} * C4 * SC5 ;$	
$I_{18} = m_6 * (d_2 + d_3) * r_{c6} $	$a_{14} = I_{23} * (C23 * C5 - S23 * C4 * S5); \approx 0.$	$b_{116} = 0$.	$b_{316} = -b_{336}$. $b_{325} = 0$.	≈ 0.	
$I_{19} = I_{yy4} - I_{xx4} + I_{xx5} - I_{yy5} + m_6 * r_{x6}^2 + I_{xx6} - I_{xx6}$;	$a_{22} = I_{m2} + I_2 + I_6 + I_{20} \cdot SS4 \cdot SS5 + I_{21} \cdot SS4$	$b_{123} = 2 * \{-I_6 * S23 + I_{15} * C23 + I_{15} * S23 * S4 * S5\}$	$b_{224} = 2 * \{I_{20} * SC4 * SS5 + I_{21} * SC4 - I_{12} * S4 * S5\};$	$c_{23} = -I_{15} * C4 * S5 + I_{16} * (C3 * C5 - S3 * C4 * S5)$	
$I_{23} = I_{yy5} - I_{xx5} - m_6 \cdot r_{x6} + I_{xx6} - I_{xx6} $	$+2 * \{I_5 * S3 + I_{12} * C3 + I_{15} * C5 \}$	$+I_{18} * (C23 * C4 * S5 + S23 * C5) + I_{19} * C23 * SC4$ $+I_{20} * S4 * (C23 * C4 * CC5 - S23 * SC5)$	≈ 0 .	$+I_{22} * C5$; ≈ 0 .	
$I_{21} = I_{224} - I_{334} + I_{225} - I_{125} $	+I ₁₆ * (S3 * C5 + C3 * C4 * S5) + I ₂₂ * C4 * S5) ;	+J ₂₁ + C23 + S4 + S5) ;	4-DOF robot a	$c_{26} = 0$. $c_{31} = -0.5 * b_{113}$.	
I ₂₂ = m ₆ * a ₅ * r ₁₆ ;	nple: Lagrangian	n expension for a	4 POF TODOL G		
I23 = I246 Part II. Gravitional Constants	$+I_{22} * SS4 * SS5 + I_{21} * SS4 + 2 * (I_{15} * C5 + I_{22} * C4 * S5);$ $\approx .333 + 3.72 \times 10^{-1} * S3 - 1.10 \times 10^{-2} * C3 .$	$\begin{array}{lll} b_{124} &= -I_{16} * 2 * S23 * S4 * S5 + I_{19} * S23 * (1 - (2 * S54)) \\ &+ I_{20} * S23 * (1 - 2 * SS4 * CC5) - I_{14} * S23 \; ; &\approx 0 \; . \end{array}$	$b_{326} = 0$. $b_{334} = b_{324}$. $b_{335} = b_{325}$. $b_{336} \approx 0$.	$c_{54} = -I_{15} \cdot C4 \cdot S5 + I_{22} \cdot C4 \cdot SC5$; $\approx -1.25 \times 10^{-3} \cdot C4 \cdot S5$.	
$g_1 = -g * ((m_3 + m_4 + m_5 + m_6) * a_2 + m_2 * r_{x2});$	$a_{24} = -I_{15} \cdot S4 \cdot S5 - I_{16} \cdot S3 \cdot S4 \cdot S5 + I_{20} \cdot S4 \cdot SC5$;	$b_{125} = I_{17} * C23 * S4 + I_{18} * 2 * (S23 * C4 * C5 + C23 * S5)$ + $I_{20} * S4 * (C23 * (1 - 2 * S55) - S23 * C4 * 2 * SC5);$	$b_{345} = -I_{15} * 2 * S4 * C5 - I_{17} * S4 + I_{20} * S4 * (1 - 2 * SS5);$	$c_{55} = -I_{15} * C4 * S5 + I_{22} * C5;$ $\approx c_{34}$.	
$g_2 = g * (m_2 * r_{23} - (m_4 + m_5 + m_6) * d_4 - m_4 * r_{24}) ;$	≈ 0.	≈ 0.	$\approx -2.50 \times 10^{-5} * S4 * C5$.	$c_{56} = 0$, $c_{41} = -0.5 * b_{114}$, $c_{42} = -0.5 * b_{224}$	
$g_3 = g \cdot m_2 \cdot r_{y2}$;	$a_{23} = I_{13} * C4 * C5 + I_{16} * (C3 * S5 + S3 * C4 * C5)$	$b_{126} = -I_{23} * (S23 * C5 + C23 * C4 * S5);$ ≈ 0 .	$b_{346} = b_{246}$, $b_{256} = b_{234}$, $b_{412} = -b_{214}$, $b_{44} = 0$.	$c_{45} = 0.5 * b_{435}$, $c_{44} = 0$, $c_{43} = 0$,	
$g_4 = -g \cdot (m_4 + m_5 + m_6) \cdot a_3$;	$+I_{17} \bullet C4 + I_{22} \bullet S5$; ≈ 0 .	$b_{134} = b_{124}$, $b_{135} = b_{125}$, $b_{136} = b_{126}$.	$b_{415} = -b_{514}$. $b_{415} = -I_{20} * (S23 * C4 * (1 - 2 * SS5) + 2 * C23 * SC5)$	$c_{44} = 0$, $c_{51} = -0.5 * b_{115}$, $c_{52} = -0.5 * b_{225}$	
$g_5 = -g * m_6 * r_{26} ;$	$a_{26} = I_{23} * S4 * S5 \; ; \qquad \approx 0 \; .$	$b_{145} = 2 * \{I_{15} * S23 * C4 * C5 + I_{16} * C2 * C4 * C5 \}$	-I ₁₇ + S23 + C4 ;	$c_{55} = 0.5 * b_{525}$. $c_{54} = -0.5 * b_{445}$. $c_{56} = 0$.	
Table A3. Computed Values for the Constants Appearing	$a_{25} = I_{m3} + I_0 + I_{20} \cdot SS4 \cdot SS5 + I_{21} \cdot SS4$ $+2 \cdot \{I_{15} \cdot C5 + I_{22} \cdot C4 \cdot S5\}; \approx 1.16.$	$+I_{18} * C23 * S4 * C5 + I_{22} * C23 * C4 * C5\} + I_{17} * S23 * C4$	$\approx -6.42 \times 10^{-4} * 523 * C4$.	$c_{64} = 0$. $c_{61} = 0$. $c_{62} = 0$.	
in the Equations of Forces of Motion.	$a_{34} = -I_{15} * S4 * S5 + I_{20} * S4 * SC5$;	$-I_{20} * (S23 * C4 * (1 - 2 * SS5) + 2 * C23 * SC5) ;$ ≈ 0 .	$b_{426} = -b_{146}$, $b_{423} = -b_{324}$, $b_{434} = 0$.	$c_{63} = 0$. $c_{64} = 0$. $c_{63} = 0$.	
(Inertial constants have units of kilogram meters-squared)	≈ -1.25×10 ⁻⁵ • S4 • S5 .	$b_{146} = I_{23} * S23 * S4 * S51 \approx 0$.	$b_{425} = I_{17} * S4 + I_{20} * S4 * (1 - 2 * SS5);$	com = 0.	
$I_1 = 1.43 \pm 0.05$ $I_2 = 1.75 \pm 0.07$	$a_{55} = I_{15} * C4 * C5 + I_{17} * C4 + I_{22} * S5$;	$b_{146} = -I_{23} * 323 * 34 * 351 \approx 0$. $b_{136} = -I_{23} * (C23 * S5 + S23 * C4 * C5) ; \approx 0$.	≈ 6.42×10 ⁻⁴ • S4 +		
$I_2 = 1.38 \pm 0.05$ $I_4 = 6.90 \times 10^{-1} \pm 0.20 \times 10^{-1}$ $I_5 = 3.72 \times 10^{-1} \pm 0.31 \times 10^{-1}$ $I_6 = 3.33 \times 10^{-1} \pm 0.16 \times 10^{-1}$	$pprox 1.25 imes 10^{-3} imes C4 imes C5$.		$b_{424} = -b_{246}$. $b_{454} = 0$.	Table A7. Gravity Terms. [The Abbreviated Expressions have units of newton-meters.]	
$I_7 = 2.98 \times 10^{-1} \pm 0.29 \times 10^{-1}$ $I_8 = -1.34 \times 10^{-1} \pm 0.14 \times 10^{-1}$	$a_{56} = I_{23} * S4 * S5 ; \approx 0.$	$b_{212} = 0$, $b_{215} = 0$,	$b_{435} = b_{425}$. $b_{456} = -b_{346}$.	$\mathbf{g}_1 = 0$.	
$I_9 = 2.38 \times 10^{-2} \pm 1.20 \times 10^{-2}$ $I_{10} = -2.13 \times 10^{-2} \pm 0.22 \times 10^{-2}$ $I_{11} = -1.42 \times 10^{-2} \pm 0.70 \times 10^{-2}$ $I_{12} = -1.10 \times 10^{-2} \pm 0.11 \times 10^{-2}$	$a_{44} = I_{m4} + I_{14} - I_{20} * SS5;$ ≈ 0.20 .	$b_{2:4} = I_{14} * S23 + I_{19} * S23 * (1 - (2 * SS4))$ $+2 * (-I_{15} * C23 * C4 * S5 + I_{14} * S2 * C4 * S5$	$b_{445} = -I_{20} * 2 * SC5 ; \approx 0 .$	$g_2 = g1 * C2 + g2 * S23 + g3 * S2 + g4 * C23$;	
$I_{18} = -3.79 \times 10^{-5} \pm 0.90 \times 10^{-5}$ $I_{18} = 1.64 \times 10^{-5} \pm 0.07 \times 10^{-5}$	$a_{45} = 0$.	$+I_{20} * (S23 * (CC5 * CC4 - 0.5) + C23 * C4 * SC5)$	$b_{440} = 0$;	$g_2 = g1 \cdot G2 + g2 \cdot S23 + g3 \cdot S2 + g4 \cdot G23$; + $g5 \cdot (S23 \cdot G5 + G23 \cdot G4 \cdot S5)$	
$I_{15} = 1.25 \times 10^{-5} \pm 0.30 \times 10^{-5}$ $I_{16} = 1.24 \times 10^{-5} \pm 0.30 \times 10^{-8}$	$a_{46} = I_{23} * C5; \approx 0.$	$+I_{22} \cdot S23 \cdot C4 \cdot S5$: $\approx 1.64 \times 10^{-3} \cdot S23 - 2.50 \times 10^{-3} \cdot C23 \cdot C4 \cdot S5 +$	$b_{456} = -I_{23} * S5 \; ; \qquad \approx 0 \; .$	$\approx -37.2 \cdot C2 - 8.4 \cdot S23 + 1.02 \cdot S2$.	
$I_{17} = 6.42 \times 10^{-4} \pm 3.00 \times 10^{-4}$ $I_{18} = 4.51 \times 10^{-4} \pm 1.30 \times 10^{-4}$ $I_{10} = 3.00 \times 10^{-4} \pm 14.0 \times 10^{-4}$ $I_{20} = -2.02 \times 10^{-4} \pm 8.00 \times 10^{-4}$	$a_{55} = I_{m5} + I_{17} \; \ \ \approx 0.18 \; .$	$2.48 \times 10^{-3} * S2 * C4 * S5 + 0.30 \times 10^{-5} * S23 * (1 - (2 * SS4))$.	$b_{510} = -b_{215}$, $b_{512} = -b_{515}$, $b_{514} = -b_{415}$.	$g_5 = g2 * S23 + g4 * C23 + g5 * (S23 * C5 + C23 * C4 * S5);$	
$I_{21} = -1.00 \times 10^{-4} \pm 6.00 \times 10^{-4}$ $I_{22} = -5.80 \times 10^{-5} \pm 1.50 \times 10^{-5}$	$a_{56} = 0.$	$b_{215} \; = \; 2 * \{ -I_{15} * C23 * S4 * C5 + I_{21} * S23 * S4 * C5 \\$	$b_{515} = 0$, $b_{516} = -b_{156}$, $b_{525} = -b_{525}$.	$\approx -8.4 * S23 + 0.25 * C23$.	
$I_{23} = 4.00 \times 10^{-5} \pm 2.00 \times 10^{-5}$ $I_{m1} = 1.14 \pm 0.27$ $I_{m2} = 4.71 \pm 0.54$	$a_{66} = I_{m4} + I_{13}$; ≈ 0.19 .	$+I_{16} * S2 * S4 * C5\} - I_{17} * C23 * S4$ $+I_{20} * (C23 * S4 * (1 - 2 * SS5) - 2 * S23 * SC4 * SC5);$	$b_{524} = -b_{625}$, $b_{525} = 0$, $b_{526} = -b_{256}$.	$\mathbf{g}_{\bullet} = -g5 \cdot S23 \cdot S4 \cdot S5;$	
$I_{m3} = 8.27 \times 10^{-1} \pm 0.93 \times 10^{-1}$ $I_{m4} = 2.00 \times 10^{-1} \pm 0.16 \times 10^{-1}$	Table A5. The expressions giving the elements of the Coriolis matrix.	$\approx -2.50 \times 10^{-5} \cdot C23 \cdot S4 \cdot C5 + 2.48 \times 10^{-5} \cdot S2 \cdot S4 \cdot C5$	$b_{554} = b_{524}$, $b_{555} = 0$, $b_{556} = -b_{556}$.	$\approx 2.8 \times 10^{-2} \cdot S23 \cdot S4 \cdot S5$.	
$I_{m5} = 1.79 \times 10^{-1} \pm 0.14 \times 10^{-1}$ $I_{m6} = 1.93 \times 10^{-1} \pm 0.16 \times 10^{-1}$	(The Abbreviated Expressions have units of kg-m ² .)	- 6.42×10 ⁻⁴ • C23 • S4 .	$b_{545} = 0$, $b_{546} = -b_{456}$. $b_{556} = 0$,	$g_5 = g_5 * (C23 * S5 + S23 * C4 * C5) \ddagger$	
(Gravitational constants have units of newton meters)	h - 1 () - 2	$b_{216} = -b_{126}$.	$b_{612} = b_{126}$, $b_{615} = b_{156}$, $b_{614} = b_{146}$,	$\approx -2.8 \times 10^{-2} * (C23 * S5 + S23 * C4 * C5)$.	
$\begin{array}{llllllllllllllllllllllllllllllllllll$	$b_{112} = 2 * \{-I_5 * SC2 + I_5 * C223 + I_7 * SC23 - I_{12} * S223 + I_{15} * (2 * SC23 * C5 + (1 - 2 * SS23) * C4 * S5)$	$b_{323} = 2 * \{-I_{12} * S3 + I_5 * C3 + I_{16} * \{C3 * C5 - S3 * C4 * S5\}\};$	$b_{615} = b_{156}$, $b_{620} = 0$. $b_{623} = 0$.	[3]	
g ₅ = -2.82×10 ⁻² ± 0.56×10 ⁻²		$\approx 2.20 \times 10^{-2} \cdot S3 + 7.44 \times 10^{-1} \cdot C3$.	$b_{024} = b_{246}$, $b_{025} = b_{256}$, $b_{026} = 0$,	ان	
ranama manamanan ang mara ang ang ang			σ ₆₂₄ = σ ₂₄₆ , σ ₆₂₅ = σ ₂₅₆ . σ ₆₂₆ = 0 .		

Controller Formulation - ISM

Three Layer Control Architecture

1) Inner Loop: Inverse Dynamics Controller

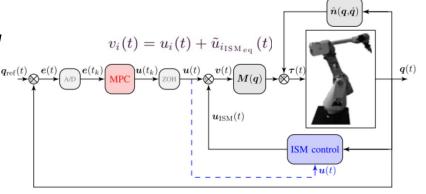
$$\ddot{q}=v-\eta(q,\dot{q})$$
 ,where $\eta(q,\dot{q})=-M^{-1}(q)(\hat{n}(q,\dot{q})-n(q,\dot{q}))$.

N-decoupled
$$\begin{cases} \dot{x}_{1_i}(t) = x_{2_i}(t) \\ \dot{x}_{2_i}(t) = v_i(t) - \eta_i(t) \end{cases}$$

2) Middle Loop: Integral Sliding Mode (ISM)

$$\tilde{u}_{i_{\text{ISM eq}}}(t) = \frac{1}{\mu_{i}} \int_{t_{0}}^{t} e^{-\frac{1}{\mu_{i}}(t-\zeta)} u_{i_{\text{ISM}}}(\zeta) d\zeta \qquad \qquad \begin{cases} \dot{x}_{1_{i}}(t) = x_{2_{i}}(t) \\ \dot{x}_{2_{i}}(t) = u_{i}(t) \end{cases}$$

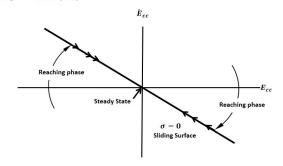
$$egin{aligned} \sigma_i(oldsymbol{x}_i(t)) &= \ & oldsymbol{S}_i\left(oldsymbol{x}_i(t) - oldsymbol{x}_i(t_0) - \int_t^t [x_{2_i}(\zeta),\,v_i(\zeta) - u_{i_{ ext{ISM}}}(\zeta)]^T d\zeta
ight) \qquad \dot{oldsymbol{x}}_i(t) = oldsymbol{A}_ioldsymbol{x}_i(t) + oldsymbol{B}_iu_i(t) \end{aligned}$$



Sliding Mode Control

$$\sigma_i = S_iigg[x_i(t) - x_i(t_0) - \int_{t_0}^tigg[egin{array}{c} x_{2i}(\zeta) \ u_i(\zeta) \ \end{bmatrix}\!d\zetaigg]\,, \qquad S_i = ig[c_i \ 1ig]$$

$$u_i^{ ext{ISM}} = -U_i^{ ext{max}} \; ext{sgn}(\sigma_i)$$



Sliding surface and sliding variable. [2]

Controller Formulation - MPC

Outer Loop: MPC

Recap:
$$\ddot{\boldsymbol{q}} = \boldsymbol{v} + \boldsymbol{M}^{-1}(\boldsymbol{q})(\hat{\boldsymbol{n}}(\boldsymbol{q},\dot{\boldsymbol{q}}) - \boldsymbol{n}(\boldsymbol{q},\dot{\boldsymbol{q}})) = \boldsymbol{v} - \boldsymbol{\eta}(\boldsymbol{q},\dot{\boldsymbol{q}}))$$

$$v_i(t) = u_i(t) + \tilde{u}_{i_{\text{ISM}}_{\text{eq}}}(t)$$

$$X_{k+1} = ilde{A}\,x_k + ilde{B}\,u_k, \qquad ilde{A} = egin{bmatrix} 1 & T \ 0 & 1 \end{bmatrix}, \; ilde{B} = egin{bmatrix} rac{T^2}{2} \ T \end{bmatrix},$$

Cost Function:

$$J(e_i(t_k), \mathbf{u}_{i_{[t_k, t_{k+N-1}|t_k]}}, N) =$$

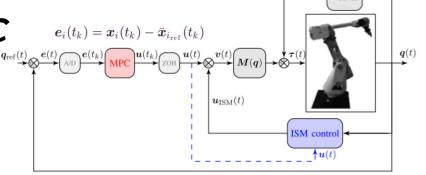
$$\sum_{i=0}^{N-1} \|\boldsymbol{e}_i(t_{k+j})\|_{\boldsymbol{Q}_i}^2 + \|u_i(t_{k+j})\|_{R_i}^2 + \|\boldsymbol{e}_i(t_{k+N})\|_{\boldsymbol{\Pi}_i}^2$$

Subject to:

$$x_i(t_{k+j}) \in \mathcal{X}$$

$$x_i(t_{k+N}) \in \mathcal{X}_f$$

$$||u_i(t_{k+j})|| \le v_{i_{\max}} - U_{i_{\max}}$$



Terminal Set and Terminal Control:

$$\mathcal{X}_{\mathrm{f}} := \{ \boldsymbol{x}_i \mid \| \boldsymbol{x} - \bar{\boldsymbol{x}}_{i_{\mathrm{out}}} \|_{\mathbf{H}}^2 \leq \rho \}, \quad \mathcal{X}_{\mathrm{f}} \subseteq \mathcal{X}$$

K_{LQ} is the control gain of an infinite horizon Linear Quadratic (LQ) controller with the same cost function

$$\kappa_{i_{\mathrm{f}}}(oldsymbol{e}_{i}(t_{k})) = oldsymbol{K}_{\mathrm{LQ}}oldsymbol{e}_{i}(t_{k})$$

$$\boldsymbol{x}_i(t_k) \in \mathcal{X}_{\mathrm{f}}$$

$$\|\kappa_{i_{\mathrm{f}}}(\boldsymbol{e}_{i}(t_{k}))\| \leq v_{i_{\mathrm{max}}} - U_{i_{\mathrm{max}}}$$

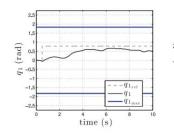
Solution to the Riccati Equation

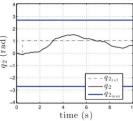
$$(\widetilde{\boldsymbol{A}}_i - \widetilde{\boldsymbol{B}}_i \boldsymbol{K}_{LQ})^T \boldsymbol{\Pi}_i (\widetilde{\boldsymbol{A}}_i - \widetilde{\boldsymbol{B}}_i \boldsymbol{K}_{LQ}) - \boldsymbol{\Pi}_i = -\boldsymbol{Q}_i - \boldsymbol{K}_{LQ}^T R_i \boldsymbol{K}_{LQ}$$

Paper Results

TABLE I STATE AND INPUT CONSTRAINTS FOR EACH JOINT

Joint i	$q_{i_{\max}}(\mathrm{rad})$	$\dot{q}_{i_{\mathrm{max}}}(\mathrm{rad}\;\mathrm{s}^{-1})$	$v_{i_{\max}}$ (rad/s ²)
1	1.83	2	145
2	2.71	3.5	250
3	3.49	6.3	350

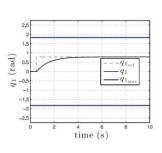


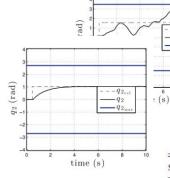


- 3-joint COMAU Smart3-S2 industrial robot manipulator
- Realistic uncertainty injection (identified from real robot data)
- Constraints: See Table I (position, velocity, acceleration limits)
- Target position: q ref = $[\pi/4, \pi/3, 2\pi/4]^T$, from q 0 = [0, 0, 0]
- Controller parameters:

Key Results:

- Ci = [10, 10, 10]
- ISM gains: [20, 35, 85]
- MPC: Q = diag(100,100), R = 0.1,
- **Terminal Weight** $\mathbf{\Pi}_i = \begin{bmatrix} 5213.4 & 165.8 \\ 165.8 & 221.3 \end{bmatrix}$ Horizon N=10
- Sampling: MPC = 20ms, ISM = 1ms





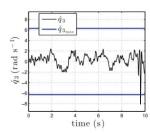
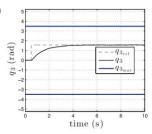
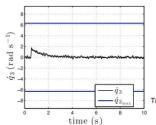


TABLE II PERFORMANCE INDEXES FOR EACH JOINT

- MPC/ISM reduces RMS error dramatically:
 - Joint 1: $0.2105 \rightarrow 0.0070 \text{ rad } (30 \times \text{decrease})$
 - Joint 3: $1.0458 \rightarrow 0.0152 \text{ rad (69} \times \text{decrease)}$
- Similar control effort but better performance
- Computational efficiency:
 - MPC: 18ms average execution time
 - ISM: only 29µs average execution time
 - Suitable for real-time implementation
- MPC alone violates constraints,
- MPC/ISM ensures constraint satisfaction



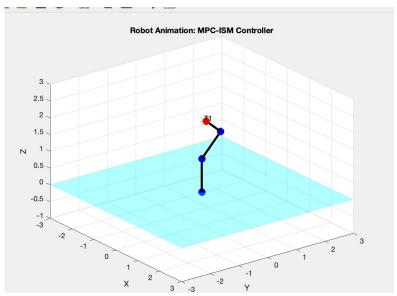


Joint i	$e_{\mathrm{RMS}_{i}}(\mathrm{rad})$	E_{c_i} (rad/s ²
1	0.2105	8.0303
2	0.3749	13.4839
3	1.0458	41.2172
1	0.0070	8.0600
2	0.0102	13.4217
3	0.0152	34.8530
	1 2 3	1 0.2105 2 0.3749 3 1.0458 1 0.0070 2 0.0102

Algo.	Mean	Min.	Max.	Std. dev.
MPC	0.018	0.017	0.71	0.011
ISM	2.9×10^{-5}	2.7×10^{-5}	0.0086	1.4×10^{-4}

Implementation on Matlab

- 1) Robot Joint Angle target: pi/2, pi/3, pi/4
- Joint trajectory from MPC and MPC-ISM controllers compared to reference cubic polynomial trajectory
- 3) Sinusoidal noise injected to see the benefits of the MPC-ISM Combined Controller Algorithm



Uncertainty Formulation and Injection

1.) Uncertainty formulation:

```
% Deterministic component based on position
pos_comp = 0.5 * max_eta * sin(q(i));

% Deterministic component based on velocity
vel_comp = 0.2 * max_eta * sign(q_dot(i)) * min(abs(q_dot(i)), 1);

% Random component (bounded)
rand_comp = 0.3 * max_eta * (2*rand() - 1);

% Combined uncertainty
eta(i) = pos_comp + vel_comp + rand_comp;
```

2.) Uncertainty injection:

Where M is the inertial matrix, n contains the nonlinear terms (friction, coriolis effect), and η is the uncertainty

$$\ddot{q} = M^{-1}(u - n + \eta)$$

```
% Compute acceleration: q_ddot = M^{-1}(\tau - n + \eta)| q_ddot = M \ (u - n + eta);
```

3.) This is done for all 3 joints.

Results

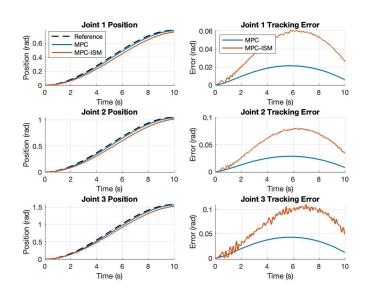
- The results were not what was expected based on the paper.
- MPC Controller was very well tuned to account for the noise and gave highly accurate results.
- 3) MPC-ISM performed slightly worse than MPC.
- 4) However, the difference seems negligible both controllers have an error within hundredths of a radian.

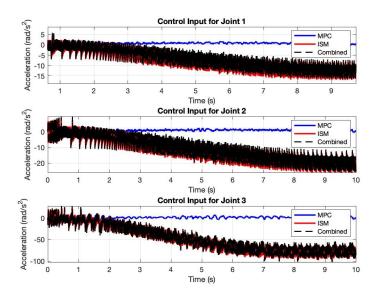
```
Controller: MPC
RMS Tracking Error (rad):
   Joint 1: 0.0156
   Joint 2: 0.0207
   Joint 3: 0.0311
Control Effort (rad/s^2):
   Joint 1: 0.0259
   Joint 2: 0.0345
   Joint 3: 0.0518
Steady-state Error (rad):
   Joint 1: 0.005942
   Joint 2: 0.007923
   Joint 3: 0.011884
```

```
Controller: MPC-ISM
RMS Tracking Error (rad):
    Joint 1: 0.0434
    Joint 2: 0.0571
    Joint 3: 0.0748
Control Effort (rad/s^2):
    Joint 1: 8.4462
    Joint 2: 13.5679
    Joint 3: 57.2287
Steady-state Error (rad):
    Joint 1: 0.026736
    Joint 2: 0.034350
    Joint 3: 0.050732
```

Results

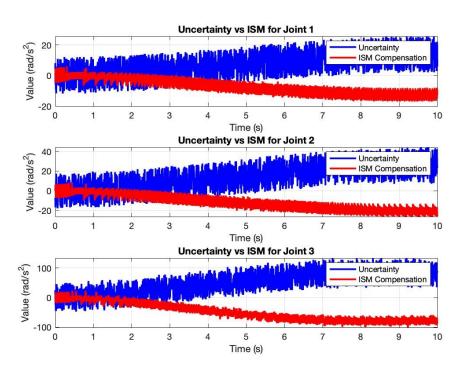
 We can see however, that the ISM Controller adjusts the control input for the joints to account for the uncertainty





Results

 As the uncertainty goes higher in magnitude, the ISM controllers counter action increases as well



Possible considerations

- 1) Uncertainty modeling The simulated uncertainty might have not been aggressive enough to impact MPC performance as the uncertainty injected by the authors
- 2) Hardware Differences: The COMAU Smart3-S2 robot in the paper may have different characteristics than our simulated model
 - a) The paper notes that their model of the COMAU Smart3-S2 on Matlab Simulink was based on data from real experiments.
 - b) However, we are simulating a perfect model on which the MPC acts upon. Due to this, the performance of MPC itself is highly accurate.
 - c) The performance difference between MPC and ISM is marginally different, and seems to be negligible due to this.

References

- [1] Incremona, Gian Paolo, et al. "MPC for robot manipulators with integral sliding modes generation." IEEE/ASME Transactions on Mechatronics, vol. 22, no. 3, June 2017, pp. 1299–1307, https://doi.org/10.1109/tmech.2017.2674701.
- [2] Adaptive chaos control of a humanoid robot arm: a fault-tolerant scheme -Scientific Figure on ResearchGate. Available from: https://www.researchgate.net/figure/Sliding-surface-and-sliding-variable_fig1370320084
- [3] <u>ECE 5463</u> taught at Ohio State University by Prof. Wei Zhang https://storage1.ucsd.edu/slides/CSE291Robo/L9 lagrangian.html#/title

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