



OA-II Backplane Bus System Design

DR00001

Rev: A01
Jinzhi Cai
2019-07-19

Table of Contents

1	Introduction	2
1.1	Scope	2
1.2	Purpose	2
2	Revision History	2
3	BUS System Requirement	3
3.1	Hardware Requirement	3
3.2	Software Requirement	3
3.3	Bandwidth Calculation	3
4	Current Bus Analyze	5
4.1	I2C	5
4.2	SPI	5
4.3	UART	5
4.4	CAN	5
4.5	PCIe	6
4.6	RapidIO	6
4.7	SpaceWire	6
4.8	Interlaken	6
5	Recommend System Design	8
5.1	General Structure	8
5.1.1	Initialization	8
5.1.2	Launch Ready	8
5.1.3	During Flight	8
5.1.4	Post-Flight	9
5.1.5	Emergent Backup	9
5.1.6	Bus Data Summery	9
5.2	CAN + SpaceWire	10
5.3	CAN + RapidIO	11

1 Introduction

1.1 Scope

This document analyze the requirement for OA-II VEH system data transmission, and current bus technology in the field, come up with a system design to fullfill the need of OA-II VEH system.

1.2 Purpose

The goal for the OA-II backplane bus system is constructure a high speed, high compatibility, and high robustness backplane data transmission system.

2 Revision History

Rev#	Editor	Delta	Date
A01	Jinzhi Cai	Initialize	2019-7-19

Table 1: Summary of Revision History

3 BUS System Requirement

3.1 Hardware Requirement

Backplane Bus The bus need to support swappable module

Vibration-proof The bus need to have strong support to the module on the frame.

Size The size need to fit into the rocket.

Topology The hardware structure need to support out-of-order locating.

3.2 Software Requirement

Point to Point & Broadcast The bus need to support broadcast.

Bandwidth The bus need to support the max bandwidth.

Topology The bus need to allow change in software topology.

Real Time The bus need to support message priority level.

Various Speed The bus need to allow low end device connect into the system.

3.3 Bandwidth Calculation

Low Speed Payload Each low speed payload it sensing in 10kHz 16bit

- 4 high pressure sensors for propulsion system
- 2 low pressure sensors for pitot tube
- 4 high temperature sensors for propulsion system
- 4 low temperature sensors for electronics
- 4 low temperature sensors for batteries
- 2 low temperature sensor for ambient

$$\begin{aligned}
 4 + 2 + 4 + 4 + 4 + 2 &= 20 \text{ channels} \\
 10 \text{ kHz} &= 10000 \text{ Hz} \\
 16 \text{ bit} &= 2 \text{ byte} \\
 10000 \text{ Hz} \times 2 \text{ byte} &= 20000 \text{ byte/s} = 20 \text{ Kbyte/s} \\
 20 \text{ Kbyte/s} \times 20 &= 400 \text{ Kbyte/s}
 \end{aligned}$$

High Speed Payload

- 9 axis IMU
- GNSS
- 4x cameras

9 axis IMU in 10kHz is
 $9 \times 10000Hz \times 2byte = 180000byte/s = 180Kbyte/s$

GNSS module¹
 UTC launch time 4byte
 Latitude 4byte
 Longitude 4byte
 Height 4byte
 Direction+Ground speed 4byte
 $4byte \times 5 = 20byte$
 $10Hz \times 20byte = 200byte/s$

Camera, set the bitrate to 8Mbps²
 $8Mbps = 1Mbyte/s$
 $1Mbyte/s \times 4 = 4Mbyte/s$

Total bandwidth

$$(180Kbyte/s + 4Mbyte/s + 200byte/s + 400Kbyte/s) \times 2 \approx 10Mbyte/s$$

¹Did not include any fixing factor

²High bitrate is necessary for high vibration environment

4 Current Bus Analyze

4.1 I2C

I2C is a serial protocol for two-wire interface to connect low-speed devices like microcontrollers, EEPROMs, A/D and D/A converters, I/O interfaces and other similar peripherals in embedded systems. It was invented by Philips and now it is used by almost all major IC manufacturers.[?]

I2C is a great low speed communication bus, however it do not support hardware priority level and change software topology.

4.2 SPI

Serial Peripheral Interface (SPI) is an interface bus commonly used to send data between microcontrollers and small peripherals such as shift registers, sensors, and SD cards.[?]

Serial Peripheral Interface allow device to increase the bandwidth by increase the data clock rate. However, it also not support hardware priority level and change software topology.

4.3 UART

A universal asynchronous receiver-transmitter is a computer hardware device for asynchronous serial communication in which the data format and transmission speeds are configurable.[?]

UART bus do not require clock line to transmit data. It also have different bitrate allow device to change. However it is a point to point communication, so it need switch for more than two devices. It also too low to meet the bandwidth requirement.

4.4 CAN

A Controller Area Network (CAN bus) is a robust vehicle bus standard designed to allow microcontrollers and devices to communicate with each other in applications without a host computer.[?]

The CAN bus have hardware priority level and support 500kbps³ bandrate. It also allow group boardcast and point to point communication.

³About 62.5Kbyte/s

4.5 PCIe

PCI Express, officially abbreviated as PCIe or PCI-e, is a high-speed serial computer expansion bus standard, designed to replace the older PCI, PCI-X and AGP bus standards. It is the common motherboard interface for personal computers' graphics cards, hard drives, SSDs, Wi-Fi and Ethernet hardware connections.[?]

The PCI Express is a common use buses in personal computer. However, the topology of this bus is mostly tree structure. It will increase difficulty when a second master need to add into the system.

4.6 RapidIO

The RapidIO architecture is a high-performance packet-switched interconnect technology. RapidIO supports messaging, read/write and cache coherency semantics.[?]

The RapidIO is a high speed connection that support up to 5Gbps⁴ by a single lane. It also support multi-master structure. By using RapidIO switch, it could change the software topology. However, the rapidIO is a new bus technology that mainly use in DSP, high speed FPGA, and SoC. It require heavy hardware resource compare with the other kind of buses.

4.7 SpaceWire

SpaceWire is defined in the European Cooperation for Space Standardization standard ECSS-E-ST-50-12C (replaces ECSS-E50-12A). The SpaceWire standard was authored by Steve Parkes, University of Dundee with contributions from many individuals within the SpaceWire Working Group from European Space Agency (ESA), European Space Industry, Academia and NASA.[?]

The SpaceWire is use LVDS voltage standard which is a commonly use voltage standard in FPGA. The PHY for SpaceWire is relatively simple and require less resource for construct the PHY. The newest SpaceWire bus support 400Mbps for one lane⁵.

4.8 Interlaken

Interlaken was invented by Cisco Systems and Cortina Systems in 2006, optimized for high-bandwidth and reliable packet transfers. It builds on the channelization and per channel flow control features of SPI-4.2, while reducing the number of integrated circuit (chip) I/O pins by using high speed SerDes technology.[?]

Interlaken is a bus design for the replace the ethernet. It also support port division and flow

⁴About 625Mbyte/s

⁵About 50Mbyte/s

control. However, the Interlaken PHY do not support in most of the FPGA. It mean it will require extra chip for PHY. It have the similar speed with RapidIO⁶.

⁶About 625Mbyte/s

5 Recommend System Design

5.1 General Structure

Base on the requirement of the bus, the OA-II Bus System will include two part, the command bus which incharge of the initialization of another bus and the prioritize data transmission, the data bus which incharge the transmission of large amount of data and program. The job for those two bus will be different during each stage.⁷

5.1.1 Initialization

Command Bus **ON**
Data Bus **Configuration**

In the initialization stage, OA-II VEH COM MCU(Main Control Unit) will sense all the on board device and running a check to exam all the device is functioning correctly. During this stage, all the command and the reply will happen in the Command bus, and each device will start to configuring the data bus interface. The final checking message will send from the MCU via the command bus and reply by each device via data bus.

5.1.2 Launch Ready

Command Bus **ON**
Data Bus **ON**

After the initialization stage, the whole vehicle will be into launch ready stage. In this stage, all the ignition process will download to all the unit via command bus and waiting for synchronize ignition signal. All the sensor will running in full speed and deliver data into MCU to monitor vehicle status via the data bus.

5.1.3 During Flight

Command Bus **StandBy**
Data Bus **ON**

After the ignition, all the sensor will running in full speed and deliver data into MCU to monitor vehicle status and recording via the data bus. Command Bus will be in the standby mode to be ready to emergent messege to passby.

⁷More detail in this section at ES00007 OA-II Payload Bus Specifications

5.1.4 Post-Flight

Command Bus **StandBy**
Data Bus **OFF**

OA-II VEH COM MCU(Main Control Unit) After landing, the MCU will do a final checking to all existing units via the Command bus before turn all the sensor unit power off. The data bus will be off during the whole landing process.

5.1.5 Emergent Backup

Scenario 1 Command Bus Failure

When COM FRU detect a command bus failure, it will cutoff the control of the COM MCU and use data bus send out a Failure Alert to all the existing unit. From this moment, the data bus will only transmitting critical data and command. Other data will be power off or save to unit internal storage.

Scenario 2 Data Bus Failure

When COM FRU detect a data bus failure, it will use command bus to communicate with MCU and checking the MCU status. The mission will continue and all unit will try to switch to secondary data bus which will running in the lower speed.

5.1.6 Bus Data Summery

	From: COM	From: TAM	From: PAM
To: COM	X	Data Bus	Data Bus
To: TAM	Command Bus/Data Bus	X	Data Bus
To: PAM	Command Bus	Data Bus	X

Table 2: Communication route

Command Bus	Data Bus
Need realtime transmission	Not mission critical
Mission critical	high bitrate data
Small size data	Research data
Send from COM	Base station information

Table 3: Bus Data Catalog

5.2 CAN + SpaceWire

Command Bus CAN

Data Bus SpaceWire

In this plan, the command bus is using CAN. It provide prioritize data transmission for small amount data transmission. Up to 500kbps bandwidth provide enough bandwidth for command data transmission. The data bus is using the SpaceWire which provide enough bandwidth for high speed data transmission. It also include timestamp feature that will help for the future sensor fusion. The whole system is build base on two router, one is the main router that will use for the major data transmission and the other is backup router which can take over when the main router fail.

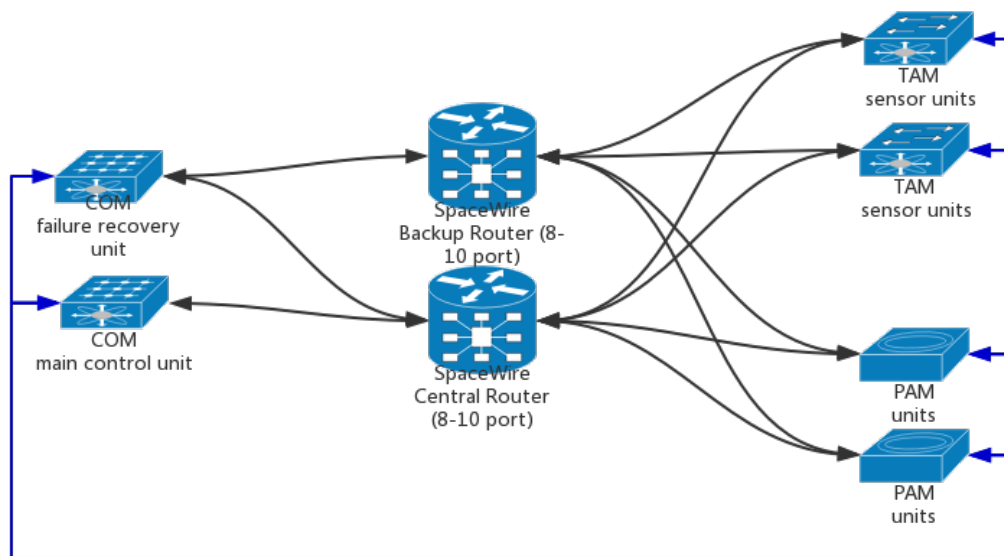


Figure 1: Physical Layer for DVP Interface

5.3 CAN + RapidIO

Command Bus CAN

Data Bus RapidIO

In this plan, CAN bus still is the main command bus due to its prioritize data transmission. In the data bus, RapidIO each lane can carry 10 times data rate compare with the SpaceWire. It can handle most future advance mission without any hardware upgrade. Some company also providing ASIC for rapidIO bus router.

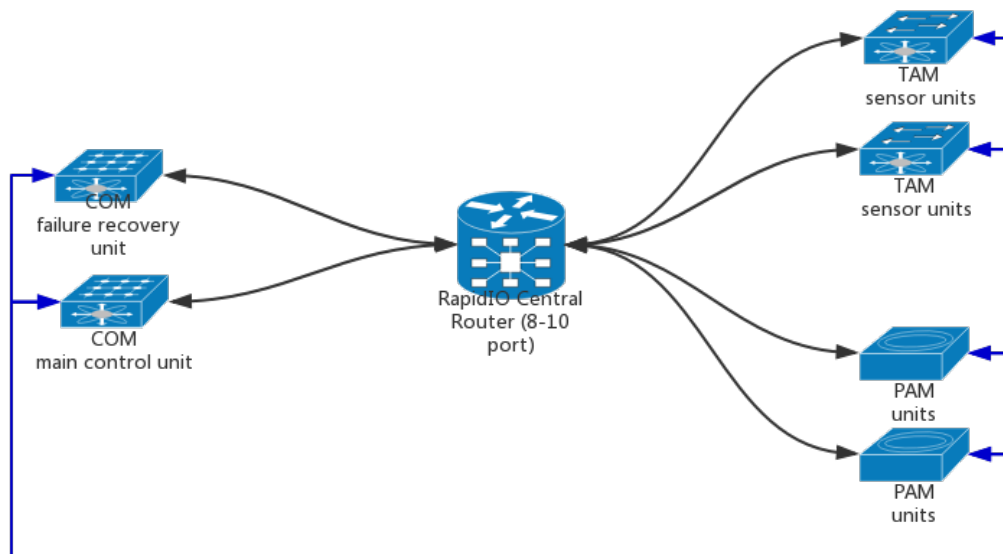


Figure 2: Physical Layer for DVP Interface