

"Autonomous Indoor Drone - Round 3"

Team Name : Orbit

Institute Name: KIET Group of Institutions, Ghaziabad

Team member details

Team Name	Orbit					
Institute Name	KIET Group of Institutions, Ghaziabad					
Team Members >	1 (Leader)	2	3			
Name	Md. Aanis Noor	Pravesh Narayan Soni	Prateek Gupta			
Batch	2022	2022	2022			
Area of expertise	+Project Planning +Software testing/debugging +Computer Vision	+Embedded firmware +Hardware +Drone tech	+Al training +Software coding +Simulation			
+ Participated in Dronathon (by Uttarakhand Govt.) + Stood 2 nd in DTMF controlled robot picking race by TechShiksha coaching institute. + Built a DTMF-call-only controlled UAV drone for college Technical Fest from scratch. + Worked/built 3+ UAV drones.		 + Made 10+ motherboards (ARM) & some MEMS modules as well. + Participated in Dronathon (by Uttarakhand Govt.) + Worked on RTOS firmwares for ARM microcontrollers. + Made 10+ UAV drones. + Made own working flight controller from scratch and its software as well. + Conducted a drone race in college. Repository: https://github.com/Elvez?tab=repositories	+ Made smart glasses as an independent project + Worked on 'my parliament' app for Innovation Centre robot management + Participated in Smart India Hackathon			

Drone specifications

Frame model / Material Mini 290 (Carbon Fiber) Frame wheelbase (mm) 290 Frame arm size (mm) 125 x 25 (L x W) Multi-copter type Hexacopter ESC category 15A Motor rating 2300kV BLDC Propeller rating 6" 6030 Flight Controller DJI Naza M-lite Battery 3300mAh LiPO battery 3S Claw servo M-995 10kg Servo Compass Module HMC5883L Master microcontroller STM32F103C8 Camera for obstacle detection Pixy2cam IR Sensor for obstacle avoidance Sharp GP2D12 RxTx Radiolink R12DS pair WiFi Tx for Camera **ESP826**

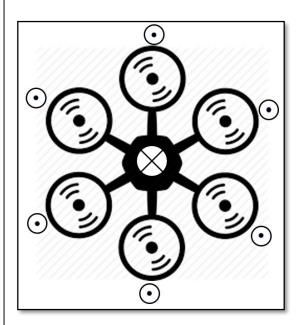
MS5611

Barometer / Altimeter

Payload Calculation

- Thrust vector max 1024gm for each motor
- Payload vector 6kg capacity (As centered as possible)

- Overall weight of drone with components (excluding payload) = 1,434gm
- Maximum thrust capacity left = 4,566gm
- By applying environmental filter for NTP and wind = 4,300gm
- Finally applying flying filter of (6.018:1) according to NTP and clear weather assumption = 2,300gm
- Hence for comfortable flying, the capacity of the drone is 2,300gm payload weight



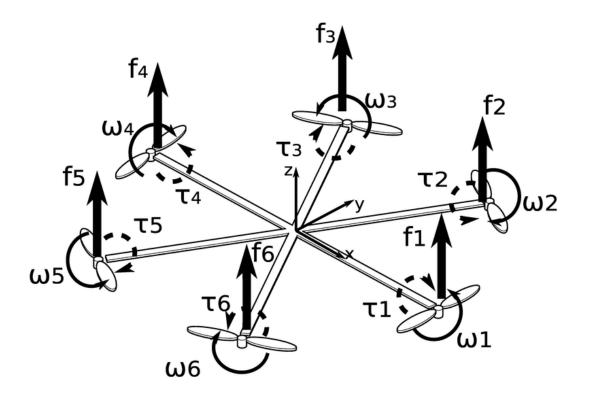


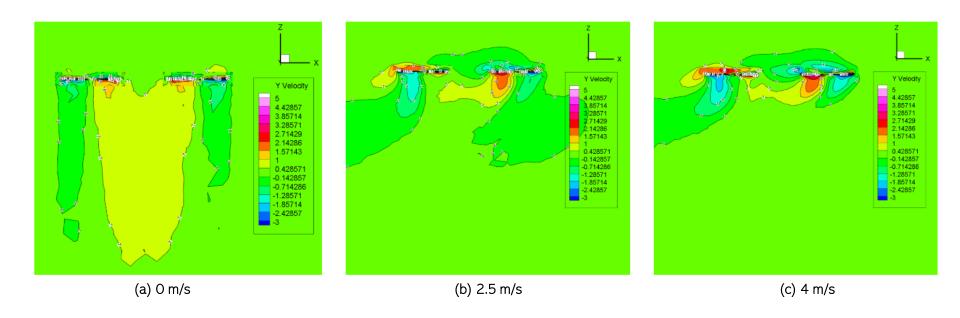
Diagram : <u>Various mathematical components of the hexacopter UAV</u>

 $\omega-\text{Angular Momentum}$

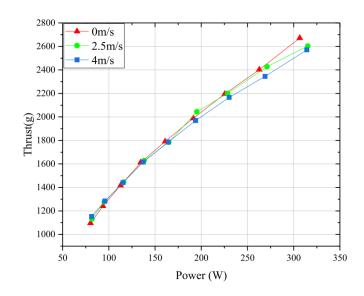
 $\tau \ - Torque$

F - Thrust

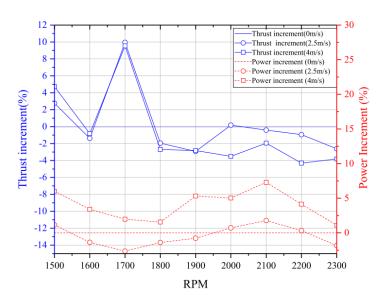
The velocity contour between the rotors with the influence of horizontal airflow is shown:



From an earlier made hexacopter, these were the results of experimental analysis using attached sensors :

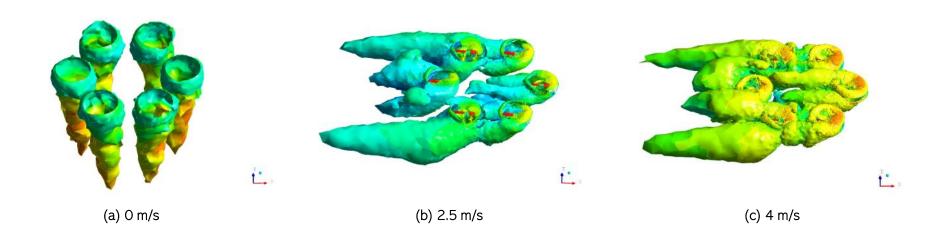


Graph(1): Thrust and power consumption with different wind speeds.

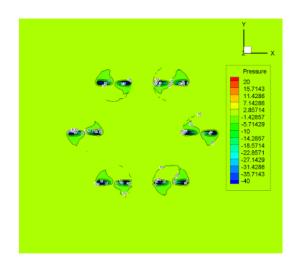


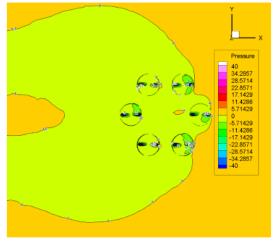
Graph (2): Thrust and power increment at different speeds.

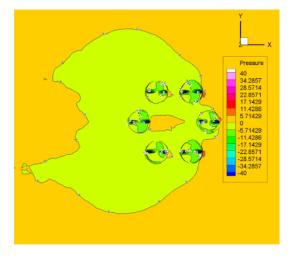
Vortex distribution of Hexacopter UAV at different horizontal speeds



Pressure distribution on the plane upper 10 mm of the center of the Hexacopter UAV

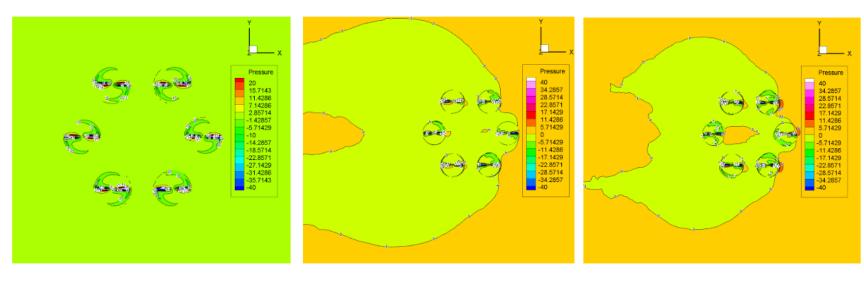






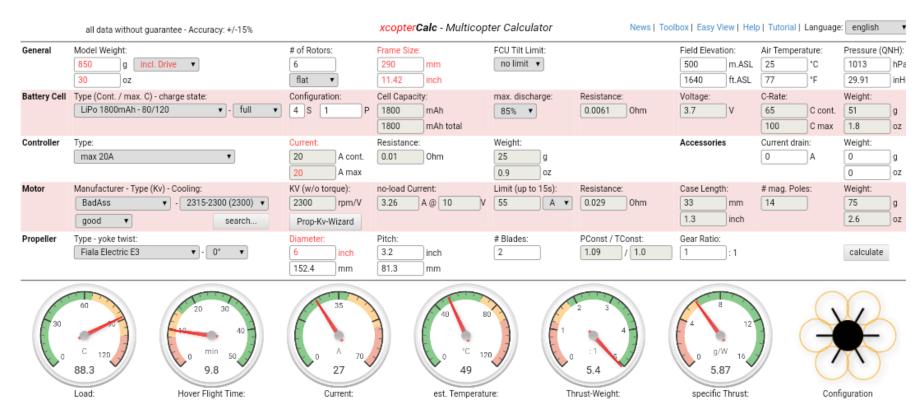
(a) 0 m/s (b) 2.5 m/s (c) 4 m/s

Pressure distribution on the center plane of the Hex-rotor UAV



(a) 0 m/s (b) 2.5 m/s (c) 4 m/s

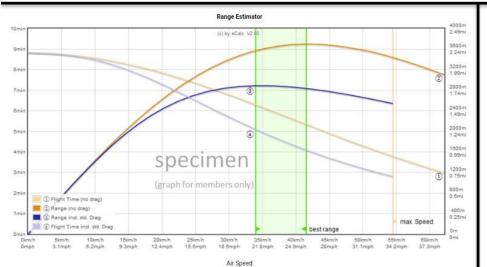
Performance Statistics

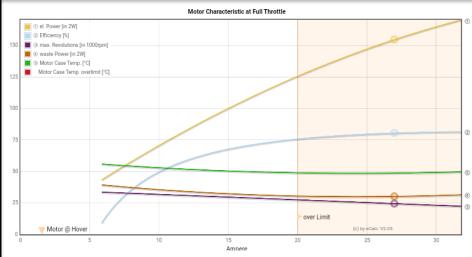


Computed using xCopter: https://www.ecalc.ch/xcoptercalc.php?ecalc&lang=en

(continued...)

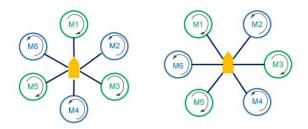
Battery		Motor @ Optimum	Efficiency	Motor @ Maximum		Motor @ Hover		Total Drive		Multicopter	
Load:	88.33 C	Current:	28.36 A	Current:	26.50 A	Current:	1.56 A	Drive Weight:	884 g	All-up Weight:	850 g
Voltage:	11.81 V	Voltage:	11.25 V	Voltage:	11.54 V	Voltage:	15.43 V		31.2 oz		30 oz
Rated Voltage:	14.80 V	Revolutions*:	23827 rpm	Revolutions*:	24630 rpm	Revolutions*:	8800 rpm	Thrust-Weight:	5.4 : 1	add. Payload:	3103 g
Energy:	26.64 Wh	electric Power:	319.1 W	electric Power:	305.9 W	Throttle (log):	17 %	Current @ Hover:	9.38 A		109.5 oz
Total Capacity:	1800 mAh	mech. Power:	258.3 W	mech. Power:	245.7 W	Throttle (linear):	26 %	P(in) @ Hover:	147.1 W	max Tilt:	78 °
Used Capacity:	1530 mAh	Efficiency:	80.9 %	Power-Weight:	2159.3 W/kg	electric Power:	24.1 W	P(out) @ Hover:	95.6 W	max. Speed:	122 km/h
min. Flight Time:	0.6 min				979.4 W/lb	mech. Power:	15.9 W	Efficiency @ Hover:	65.0 %		75.8 mph
Mixed Flight Time:	4.2 min			Efficiency:	80.3 %	Power-Weight:	173.1 W/kg	Current @ max:	159.00 A	est. Range:	- m
Hover Flight Time:	9.8 min			est. Temperature:	49 °C		78.5 W/lb	P(in) @ max:	2494.4 W		- mi
Weight:	204 g				120 °F	Efficiency:	66.1 %	P(out) @ max:	1474.2 W	est, rate of climb:	22.6 m/s
	7.2 oz					est. Temperature:	28 °C	Efficiency @ max:	59.1 %		4449 ft/min
				Wattmeter readings			82 °F			Total Disc Area:	10.94 dm ²
				Current:	159 A	specific Thrust:	5.87 g/W				169.57 in ²
				Voltage:	11.81 V	-	0.21 oz/W			with Rotor fail:	3
				Power:	1877.8 W						•



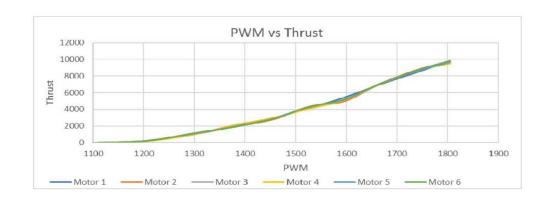


Structural & Stability Analysis (with & without payload)

Hexacopter configuration -

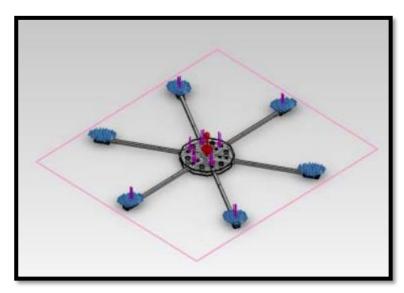


thrust =
$$\frac{total\ weight\ X\ 2}{number\ of\ motors}$$

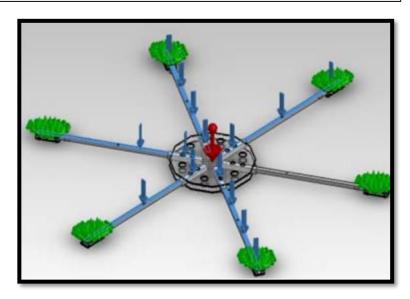


Note: We showcase a hexacopter made earlier with more realistic assumptions and simulation.

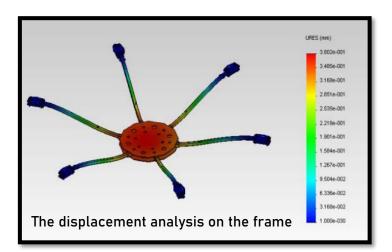
- The Hexacopter frame consist of ABS material for its strength and lightweight characteristics.
- Furthermore, a loading simulation of this frame will be done by analyzing the static and dynamic materials. The force that is given as the load on this simulation is 50 N (more than required for better understanding of stress).

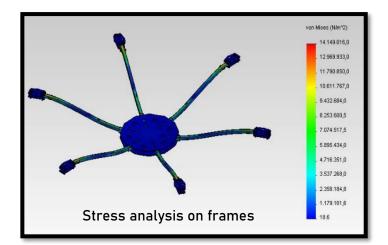


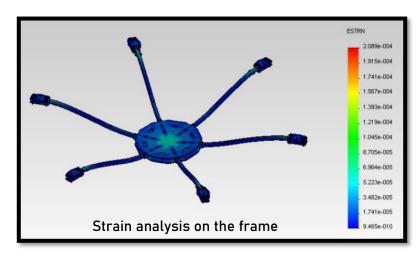
Hexacopter ABS frame design



Placement of styles on the frame







CAD File with Payload & Drone Components

CAD File - https://drive.google.com/file/d/1TjKUJIND-tie4K79vyENH-ALdE3wVsSZ/view?usp=sharing

CAD File (gripper) - https://drive.google.com/file/d/10pdvWDvvScgcSHJvw-XqJ5o8-uEJ5rBZ/view?usp=sharing

CAD zip (with part files)- https://drive.google.com/file/d/1H6qdqrtN59HDinxw74zl31VKuelBhMW4/view?usp=sharing





Autonomous Flight Algorithm Details

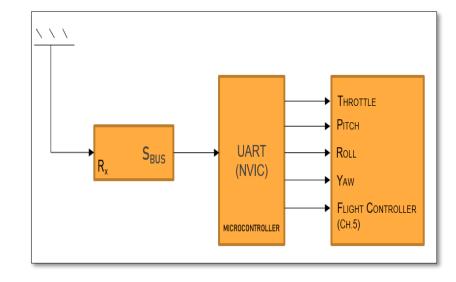
Basic working principle:

A typical UAV drone is controlled mainly using the Flight Controller with manual PWM inputs.

But what we do is - Instead of connecting the receiver to the flight controller, we connect it to a micro-controller first and then use the global interrupt on the chip to drive the PWM connections on the flight controller by *TIM1_CH1*, channel 2, 3, 4 and *TIM2* (if we need).

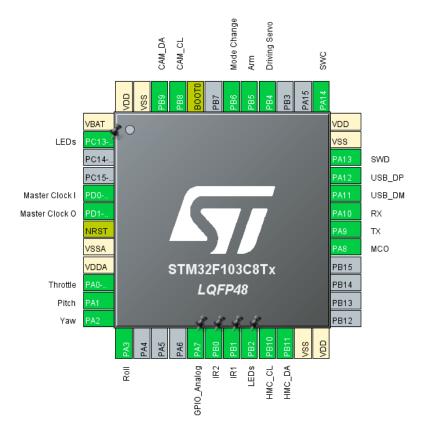
- 'TIM is the TIMER peripheral of the microcontroller.
- We call this process "Receiver Hijack Protocol (RHP)"

Technically, instead of manually providing PWM via Tx, we embed the algorithm into the micro-controller (master) which then instructs the Flight Controller (slave) as dummy Rx/Tx.



(Refer to image on next page)

Video explaining RHP - https://drive.google.com/file/d/1nrVtzBP0b7A1DopLgrXx-qB41DZE8kKL/view?usp=sharing



Reference: STM32 Microcontroller Pin Layout

i) Take off and landing logic

Nutshell:

- The height for the given task is constant and pre-defined. We use an altimetric/barometric sensor to fix the height from the point of flight for take-off. Further to this, the flight and frame detection algorithms kick in.
- The final point of landing would be given a QR Code (printed image) which will be read by drone's Al-camera module; this determines the point of landing. The UAV will go stable approximately above this point, and the throttle will be lowered down until the UAV drone is on ground.



(continued...)

Take off Logic



Landing Logic



- · Associate a QR code to landing function
- if (numberOfGates == None) : GoTo Step (1)



- Lurk & Look for a matching QR Code (through camera module)
- if (foundQrCode == 1): GoTo Step (2)



- Open servo to drop item (after throttle
- Lower down throttle until UAV touches

Extended: "Drop Mechanism & Landing"

```
repeat_process(); //until signature = 0
//payload
  if (QrRead == wantedNumber)
     { writeServo(fullAngle); //dropPayload }
  Land();
  Disarm();
```

ii) Flight algorithm

Nutshell:

- For the process of moving the drone automated, we first create a dataset by manually controlling it via PWM inputs of a transmitter; and then writing it into a file.
- We make functions for every movement of the drone via these experimental inputs.
- Finally, we apply the frame detection algorithm to automate drone movement in the environment.



Use Transmitter to move drone in complex geometries

Write PWM data via interrupted UART (into a text file)

Apply frame detection logic to automate

Drone movement as required

(continued...)

The dataset generation process:

- We use the transmitter to move the drone in complex geometries like turning, circling, quarter circling, etc.
- As the R_x gives data to the interrupted UART –

```
HAL_UART_Receive_IT (&huart1, (uint16_t)pwm, 2, 10);

//&huart1 is a TypeDef to the HAL (Hardware Application Layer) peripheral of UART
//16-bit SBUS PWM data is received in "pwm", two bytes are received with a timeout of 10ms

{

WriteToFile(pwm, File.txt);
}
```

- Through this process, we assume that we've generated these **four** basic functions (used for flight):
 - moveForward(time); moveBackwards(time);
 - **2.** moveLeft(time); moveRight(time);
 - 3. turnClockwise(angle);
 - 4. turnAnticlockwise(angle);



Memory Allocated: 5kB

Scheduler: Pre-emptive

Semaphores:
1 semaphore of 5 count

Number of tasks: 5

RTOS Block for Dataset Generation

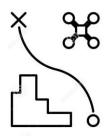
Dataset completion time: 3 days

After we have the dataset, we proceed to make the code that allows us to move the drone as we simply like, according to the requirements.

iii) Trip management

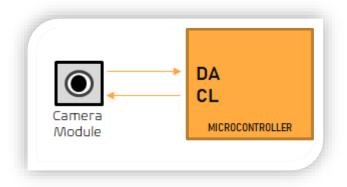
After successful take off, we go through the following steps for trip management:

- 1. Gathering data from the Al-Cam
- 2. Correction Algorithm
- 3. Find center of the corresponding gate & Alignment Algorithm
- 4. IR PID loop to avoid collision while passing through gates



(continued...)

A. Gathering data from the Al-Cam



* Address - 0xAE

The 16-bit ASCII return packet from pixy2cam contains all the info we need.

* Packet - { Rank; Signature; Vector X,Y; Dimension(Px); Age...}

* We only need the first 4 elements of the packet, so we move on to the correction algorithm...

B. Correction Algorithm

```
(PID loop 1)
```

To extract only the first 4 elements (pseudo code) -

```
do {
    camData[count] = camData[count] << 8;  //I2C data read and store
    camData[count+1] = camData[count+1] << 8;  //I2C data read and store
}
while {
    camData[count+3]
}
```

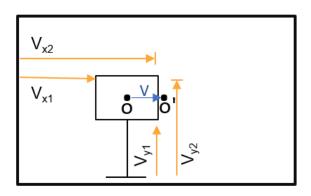
The execution of above will leave us with only the first 4 elements which are required (from 8 elements)

Input data from the pixy2 AI camera module

Bit shift 8 bits (of 16 bits) to retain required data only

Apply logic to the acquired data for further processes

C. Find center of the corresponding gate & alignment



The vector X, Y read from I2C gives 4 vectors;

$$L = |V_{x1} - V_{x2}|$$

 $B = |V_{y1} - V_{y2}|$

0 => Center of Gate

O' => Center of Screen (known by default)

For 0,

$$0_{x} = [(V_{x2} - V_{x1})/2] + V_{x1};$$

$$0_{y} = [(V_{y2} - V_{y1})/2] + V_{y1};$$

$$0 = (0_{x}, 0_{y})$$

Now drawing a vector from 0 to 0' through vector register (0xC20158) in pixy2

V => Vector between 0 and 0'

Through the process below, the drone is centered according to the gate in front of it.

```
while (L:B == designatedRatio)
    PID(); //using vector V plotted earlier
            error = LV<sub>c</sub>;
            P = K_p * (L_v - L_c) //L_v is needed length of vector (0)
                        //L_{vc} is the current length
            while (L_{vc} < 3) //very small error
               I = K_I * (L_{vc}) + I; //Integral
                D = K_D * (L_{vc} - L_{vc1}) //currentError-previousError
            PID = P+I+D; //actively uses move functions
All we need to do after this is
{ moveFowrward(time) }
                                         //time=> delayUntilSignature = Signature-1
                                         //signature getting reduced means there's one less gate in sight
```

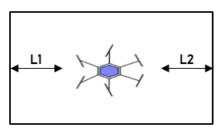
After this process, IR PID kicks in.

D. IR PID loop to avoid collision while passing through gates

While the drone goes inside the gates, the InfraRed sensor module gives data for PID correction of the drone to avoid any collision within the frame.

Interrupt of semaphore when drone is inside the gate:

The IR sensor reads along values...



We know that, $Length_{gate}$ = 1000mm $Height_{gate}$ = 500mm $Diameter_{drone}$ = 290mm

So, 1000mm - 290mm = 710mm Hence,

L1 = L2 = 710/2 = 355mm (for best results of centering inside gate)

(continued...)

IR PID loop:

```
P = K_p * (355 - L_1);
while (L_1 < 3 || L_1 < 5)
   | = (K_i * L_1) + |;
D = K_D * (L_1 - L_{1prev}); //current-previous error
PID = P+I+D; //Actively uses more functions
```

Frame Detection Algorithm Details

By knowledge, L:B ratio is same for all gates & provided colour is '*RAL 1023*' This makes us not worry about other similar looking gates.

- We grab image data as explained in Slide 17
- We centre the drone as explained in Slide 18/19
- We adjust drone inside frame using IR PID as explained in Slide 20/21

All this was done after the frame was detected.

For frame detection, we use pixy2cam (camera) Al module available after-market.

Provided frame dimensions were fixed. In figures:

 $Length_{gate} = 1000mm$ $Height_{gate} = 500mm$

- The Length: Height ratio is fixed as 2:1 (or 2.00 as float)
- Given frame colour RAL1023



pixy2cam module



"We use this ratio and colour 'RAL1023' to train the AI camera module for detecting frames on-the-go using in-built operations."

Rank : Assigned number to multiple same objects in sight or off sight.

- Gates in sight are given ranks by the AI cam. We call them as signature.
- The drone will keep looking for gates unless no gate is in sight.

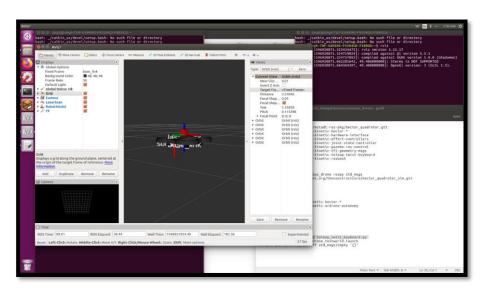
• When the frame are no more in sight and (signature == 0):

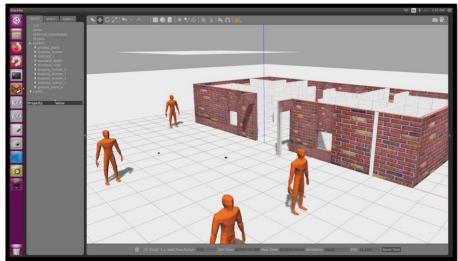
Execute(landing_procedure)

Reference: https://www.youtube.com/watch?v=EcCbEWiyiQY

Gazebo Simulation & Assumptions

Gazebo Simulation Resources: https://drive.google.com/drive/folders/1cQmLsdZa3bA02r0xepxtQAlwnBhihTJf?usp=sharing





We've provided the link to resources (in use and development).

Unfortunately for us, Gazebo simulation was incomplete due to unfortunate events with our teammate's family. We were not able to meet this requirement within the deadline.





Component Sources, Price & total cost

Compass/Magnetomemter HMC5883L	₹370/-	https://robu.in/product/gy-273-hmc5883I-3-axis-module-magnetic-field-sensor/
Barometer MS5611	₹900/-	https://robu.in/product/gy-63-ms5611-01ba03-high-precision-pressure-sensor-height-sensor-module/
IR Sensor x2 (sharp GP2Y0A41SK0F)	₹1200/- (600x2)	https://robu.in/product/sharp-ir-distance-measuring-sensor-unit-4-30-cm-cable/
Motors 2300kv x6	₹4500/- (750x6)	https://robu.in/product/rs2205-2300kv-cw-brushless-motor-fpv-racing-quad-motor-fpv-multicopter-qav250-qav300/
6" Propeller Pairs (CW+CCW) x 3	₹900/-	https://robu.in (or local market)
1500 mAh LiPO battery Orane	₹2800/-	https://robu.in/product/orange-1500mah-4s-100c200c-lithium-polymer-battery-pack-lipo/
15A ESC x6	₹3000/- (500x6)	https://www.roboelements.com/product/30a-esc-brushless-bidc-motor-electronic-speed-controller-for-quadcopter/?gclid=Cj0KCQjwvb75BRD1ARIsAP6LcqsZredyqG3bi0cnp8qZr1oxrylL_Wqly5hLrXrpgqNPt85X168MTLkaAiiSEALw_wcB
Carbon Fibre Drone Printing	₹4000/-	https://robu.in
Pixy2 Camera	₹7300/-	https://robu.in/product/pixy-1-0-smart-vision-sensor-object-tracking-camera/
Master Microcontroller	₹500/-	https://robu.in/product/stm32f103c8t6-minimum-system-board-microcomputer-stm32-arm-core-board/
10kg Servo Motor	₹500/-	https://robu.in/product/towerpro-mg945-digital-high-speed-servo-motor-standard-quality/
flight controller	- Own -	-
Receiver and Transmitter	- Own -	-
WiFi Tx for camera	₹2000/-	https://robu.in
Reflective balls for motion capture training (x5)	- Own -	
TOTAL APPROXIMATE COST -	32-35k INR	

_____Component Details

Component	Specifications
IR Sensor	3–80cm accuracy and range
Al Camera	640x480 @60Hz
Claw servo	10kg drive
Motors	2300kv, 0.000434kt
Frame Material	Carbon Fibre (Twill weave)
Flight Controller	8Ch Naza mLite
Radio	R12DS FHDSS 12 channel
Battery	1500mAh LiPO 4s 40C
ESC	15A with large decouple
Propellers	6" ABS 6030
Master MCU	ARM Cortex M3 72mHz

Execution plan with timelines

- Drone build time = 1-2 days
- Drone motion Al training and RHP training = 3-4 days
- PID tuning for IR correction = 3 days
- Gate detection training = 7 days
- PID tuning for gate detection and obstacle avoiding = 5 days
- Autonomous mission training and logging = 4 days
- GCS and GUI creation = 3 days



We need a total of <u>25-28 days</u> for full completion of the product, once we get a hands on for all the modules and parts for it.



Current Progress

- ✓ Minimal viable firmware ready
- ✓ Action plan ready
- ✓ Hardware assembly as soon as the college re-opens
- ✓ Al training logic ready



References

https://www.ecalc.ch/xcoptercalc.php?ecalc&lang=en

https://ieeexplore.ieee.org/document/7849648

https://www.researchgate.net/

https://www.udemy.com/course/robotics-with-ros-autonomous-drone-with-path-planning-slam/

https://www.semanticscholar.org/

https://www.simscale.com/

https://www.wired.com/2014/05/modeling-the-thrust-from-a-quadcopter/

https://www.instructables.com/id/Design-Build-and-Improve-a-Quadcopter/

https://airshaper.com/