

# 521 M7280 – SATELLITE GEODESY

## SPRING SEMESTER 2014

### Lab No. 7

handed out Wednesday, April 23, 2014  
due Wednesday, April 30, 2014, 09:20 Name: \_\_\_\_\_

#### Satellite Orbit in 3-D Space (II) Creating your own SP3 files

1. From a GPS satellite almanac (satellite ID = your date of birth) that is given to you, please write a Matlab program which computes the satellite's orbit every 5 min for a 48-hour period.
  - a. List your results in both an initial frame and ECEF Cartesian frame with 7 columns (time,  $X_{in}$ ,  $Y_{in}$ ,  $Z_{in}$ ,  $x$ ,  $y$ ,  $z$ ).
  - b. Plot the satellite's position in two 3-D maps (one for inertial and the other for ECEF Cartesian).
  - c. What are the six Keplerian elements you use to compute the orbit?
  - d. What are the period and mean motion of your satellite?
2. Convert your ECEF Cartesian coordinates into the ECEF spherical frame.
  - a. List the results in a table with 4 columns (time, longitude, latitude, height).
  - b. Plot in a 2-D map for the longitude, latitude, and height values as functions of time.
3. Discuss your results.

Note: GPS almanac files (in the YUMA format) can be found through the following link:  
<http://www.navcen.uscg.gov/?pageName=gpsAlmanacs>

Use for  $GM = 398600.4418(\text{km}^3/\text{s}^2)$ ,  $\omega_e^* = 7292115.8553 \times 10^{-11}(\text{rad/s})$ ,  
 $\omega_e = 7292115 \times 10^{-11}(\text{rad/s})$ , and  $R = 6371.000000(\text{km})$ .

**Your (individual) final report should contain (use A4 papers):**

- this page as the cover sheet
- source code(s) and outputs; do not forget to add your name and lots of comment cards to the source listing (% .....
- input and output files from program [input/output values used and calculated], if any
- plots, including captions on axes, title, your name, LB#/HM#, course title, date (if any)
- derivation and description of formulas used, accompanied by figures where applicable
- evidence of computational accuracy
- discussion of results