

# Cribsheet

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## 1 Transport equations

- **Material derivative:**

$$\frac{D\Phi}{Dt} = \frac{\partial\Phi}{\partial t} + v_i \frac{\partial\Phi}{\partial x_i}$$

With  $v_i$  (in Einstein notation form) the speed of displacement of the Lagrangian frame, following a streamline.

### 1.1 Flow

- **Mass balance:** (in units of  $\text{kg.m}^{-3}.\text{s}^{-1}$ )

$$\frac{\partial\rho}{\partial t} + \nabla \cdot (\rho \mathbf{v}) = 0$$

- **Momentum balance:** (in units of  $\text{kg.m}^{-2}.\text{s}^{-2}$ )

$$\rho \frac{Dv_i}{Dt} = \rho g_i - \frac{\partial p}{\partial x_i} + \frac{\partial \tau_{ij}}{\partial x_j}$$

$$\underbrace{\rho \frac{Dv_i}{Dt}}_{\text{Amount of acceleration}/m^3} = \underbrace{\rho g_i}_{\text{Volume forces}/m^3} - \underbrace{\frac{\partial p}{\partial x_i}}_{\text{Pressure forces}/m^3} + \underbrace{\frac{\partial \tau_{ij}}{\partial x_j}}_{\text{Viscous forces}/m^3}$$

- If the fluid is **Eulerian** (or ideal), viscous effects are neglected:

$$\rho \frac{Dv_i}{Dt} = \rho g_i - \frac{\partial p}{\partial x_i}$$

- If the fluid is **Newtonian** (linear *and* isotropic relationship between viscous stress and the strain rate):

$$\tau_{ij} = 2\mu d_{ij} - \underbrace{\frac{2}{3}\mu(\nabla \cdot \mathbf{v})\delta_{ij}}_{= 0 \text{ if incompressible}}$$

- If the fluid is **Newtonian and incompressible**, the viscous stress tensor can be further simplified:

$$\underbrace{\tau_{ij}}_{\text{Viscous stress tensor}} = \underbrace{2\mu}_{\text{Viscosity}} \cdot \underbrace{d_{ij}}_{\text{Strain rate tensor}}$$

with the strain rate tensor defined as: ( $i^*j$  matrix,  $3 \times 3$  in  $3D$ )

$$d_{ij} = \frac{1}{2} \left( \frac{\partial v_i}{\partial x_j} + \frac{\partial v_j}{\partial x_i} \right)$$

- **Navier-Stokes equations:** Momentum balance equations for *Newtonian* fluids:

- If  $\mu = cst$ :

$$\underbrace{\frac{\partial \tau_{ij}}{\partial x_j}}_{\text{Viscous forces}/m^3} = \mu \frac{\partial^2 u_i}{\partial x_j^2} + \frac{1}{3} \mu \frac{\partial}{\partial x_i} \left( \frac{\partial u_k}{\partial x_k} \right)$$

with

$$\frac{\partial}{\partial x_j} (\nabla \cdot \mathbf{v} \delta_{ij}) = \frac{\partial}{\partial x_j} \frac{\partial v_k}{\partial x_k} \delta_{ij} = \frac{\partial}{\partial x_i} \frac{\partial v_k}{\partial x_k}$$

- If  $\mu = cst$  **and** incompressible:

$$\underbrace{\frac{\partial \tau_{ij}}{\partial x_j}}_{\text{Viscous forces}/m^3} = \mu \frac{\partial^2 u_i}{\partial x_j^2}$$

## 1.2 Mass transfer

- **Mass (or species) transport equation**

$$\frac{\partial}{\partial t} (\rho Y_k) + \nabla \cdot \mathcal{J}_k = \dot{w}_k$$

$\underbrace{\frac{\partial}{\partial t} (\rho Y_k)}_{\text{non-stationary}} + \underbrace{\nabla \cdot (\rho Y_k \mathbf{v})}_{\text{convection}} = \underbrace{-\nabla \cdot \mathbf{J}_k}_{\text{diffusion}} + \underbrace{\dot{w}_k}_{\text{production}}$
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where  $\rho$  is the density of the fluid,  $Y_k$  is the mass fraction of species  $k$  in the fluid,  $\mathbf{v}$  the velocity vector of the fluid,  $\mathcal{J}_k$  the mass flux vector of species  $k$ ,  $\mathbf{J}_k$  the diffusive mass flux vector of species  $k$ ,  $\dot{w}_k$  the production rate of species  $k$ .

- **Fick's law:** (mass diffusion)

$$\mathbf{J}_k = -D_k \nabla C_k = -\rho D_k \nabla Y_k$$

We can then further develop the previous equation by introducing Fick's law and Einstein convention:

$$\rho \left( \frac{\partial Y_k}{\partial t} + \frac{\partial Y_k v_j}{\partial x_j} \right) = \rho D_k \frac{\partial^2 Y_k}{\partial x_j \partial x_j} + \dot{w}_k$$

## 2 Dimensionless numbers

- **Prandtl's number:** (with  $D_T$ : thermal diffusivity).

$$Pr = \frac{\nu}{D_T} = \frac{\text{momentum transport}}{\text{thermal transport}}$$

- **Schmidt's number:**

$$Sc = \frac{\nu}{D_J} = \frac{\text{momentum transport}}{\text{species or mass transport}}$$

- **Lewis' number:** Thermo-diffusive effects.

$$Le = \frac{D_T}{D_J} = \frac{Sc}{Pr} = \frac{\text{diffusive mass transfer}}{\text{heat transfer}}$$

- **Progress variable:** (with  $c=0$  in unburnt gases and  $c=1$  in burnt gases)

$$c = \frac{T - T_1}{T_2 - T_1} = \frac{Y_F - Y_{F_1}}{Y_{F_2} - Y_{F_1}}$$

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