# EE4302 Advanced Control Systems

Owen Leong owenleong@u.nus.edu

November 29, 2024

# 1 State Space Representation

$$\dot{x} = Fx + Gu \quad y = Hx + Ju$$

Consider  $\ddot{y} + a_1 \ddot{y} + a_2 \dot{y} + a_3 y = b_1 \ddot{u} + b_2 \dot{u} + b_3 u$ 

$$\frac{Y(s)}{U(s)} = \frac{b_1 s^2 + b_2 s + b_3}{s^3 + a_1 s^2 + a_2 s + a_3}$$

### 1.1 Method 1 (Controller Can. Form)

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ -a_3 & -a_2 & -a_1 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix} u$$
$$y = \begin{bmatrix} b_3 & b_2 & b_1 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix}$$

To get this formulation, take  $\frac{\xi(s)}{U(s)} = \frac{1}{s^3 + a_1 s^2 + a_2 s + a_3}$  so that  $\ddot{\xi} + a_1 \ddot{\xi} + a_2 \dot{\xi} + a_3 \xi = u(t)$  and  $y(t) = b_1 \ddot{\xi} + b_2 \dot{\xi} + b_3 \xi$  and start with  $x_1 = \xi, x_2 = \dot{\xi}, \dots$ 

# 1.2 Method 2 (Observer Canonical Form)

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \end{bmatrix} = \begin{bmatrix} -a_1 & 1 & 0 \\ -a_2 & 0 & 1 \\ -a_3 & 0 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix} + \begin{bmatrix} b_1 \\ b_2 \\ b_3 \end{bmatrix} u$$
$$y = \begin{bmatrix} 1 & 0 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix}$$

To get this formulation, let  $\iiint \ddot{y} = \iiint (-a_1\ddot{y} + b_1\ddot{w}) + \iiint (-a_2\dot{y} + b_2\dot{u}) + \iiint (-a_3y + b_3u)$  so that  $y(t) = \int (-a_1y + b_1u) + \iiint (-a_2y + b_2u) + \iiint (-a_3y + b_3u) = \int (-a_1y + b_1u + \int (-a_2y + b_2u + \int (-a_3y + b_3u)))$  and set  $x_1 = y, \dot{x}_1 = -a_1y + b_1u + x_2, x_2 = \int \dots$ 

### 1.3 State Transformations

Suppose 
$$\dot{p}=Fp+Gu,y=Hp+Ju$$
 and  $p=Tx,$  
$$\dot{x}=(T^{-1}FT)x+(T^{-1}G)u$$
 
$$y=(HT)x+Ju$$
 
$$T=\mathcal{C}_2\mathcal{C}_1^{-1}$$

### 1.4 Transfer Function

$$Y(s) = \left[H\left\{sI - F\right\}^{-1}G + J\right]U(s)$$

### 1.5 Direct Poles and Zeroes

Poles of the system are eigenvalues of F,

$$\det\{\lambda_i I - F\} = 0$$

Zeroes of the system are

$$\det\left[\begin{array}{c|c} sI - F & -G \\ \hline H & J \end{array}\right] = 0$$

# 2 State Feedback

Suppose we use state feedback u = -kx,

$$\dot{x} = Fx + Gu = [F - Gk]x$$

Then closed-loop poles are given by

$$\det\left[sI - \{F - Gk\}\right] = 0$$

Let our desired closed-loop poles be given by

$$\alpha_c(s) = s^n + \alpha_1 s^{n-1} + \alpha_2 s^{n-2} + \dots + \alpha_n$$

We can expand the determinant expression and equate to our desired closed-loop pole equation, or used closed form solution:

$$k = \begin{bmatrix} 0 & 0 & \dots & 0 & 1 \end{bmatrix} \mathcal{C}^{-1} \alpha_c(F)$$

$$\mathcal{C} = \begin{bmatrix} G & FG & F^2G & \dots & F^{n-1}G \end{bmatrix}$$

$$\alpha_c(F) = F^n + \alpha_1 F^{n-1} + \dots + \alpha_n I$$

Controllability condition is thus that  ${\mathfrak C}$  is non-singular.

#### 2.1 Selection of Pole Locations

Use standard tables or use LQR

# 3 Changing Set-Point

### 3.1 Include term in control U

$$u = -kx + k_s r, \quad y = Hx$$

$$\dot{x} = Fx + Gu = [F - Gk]x + Gk_s r$$

$$\frac{Y(s)}{R(s)} = H[sI - (F - Gk)]^{-1}Gk_s$$

$$\lim_{s \to 0} H[sI - (F - Gk)]^{-1}Gk_s = 1$$

$$k_s = \frac{1}{\left(H[sI - (F - Gk)]^{-1}G\right)\Big|_{s=0}}$$

### 3.2 Augmentation with Error Term

Let 
$$\dot{x}_I = y - r = Hx - r$$

$$\dot{\bar{x}} = \begin{bmatrix} \dot{x} \\ \dot{x}_I \end{bmatrix} = \begin{bmatrix} F & 0 \\ H & 0 \end{bmatrix} \begin{bmatrix} x \\ x_I \end{bmatrix} + \begin{bmatrix} G \\ 0 \end{bmatrix} u + \begin{bmatrix} 0 \\ -1 \end{bmatrix} r$$

$$= \bar{F}\bar{x} + \bar{G}\bar{u} + \begin{bmatrix} 0 \\ -1 \end{bmatrix} r, \quad \bar{u} = -\bar{k}\bar{x}$$

$$y = \begin{bmatrix} H & 0 \end{bmatrix} \begin{bmatrix} x \\ x_I \end{bmatrix} = \bar{H}\bar{x}$$

# 4 Nonlinear Systems

### 4.1 Root Locus of Nonlinear System

Useful if the nonlinearity has no dynamics (memoryless) and can be approximated as a gain that varies in the size of its input signal

### 4.2 Phase Plane Analysis

Equilibrium points are where system states can stay forever.

### 4.2.1 Sketching

To sketch the Phase portrait, solve differential equations for x as a function of t, or use

$$\frac{dx_2}{dx_1} = \frac{f_2(x_1, x_2)}{f_1(x_1, x_2)}$$

(method of isoclines)

A closed curve is a limit cycle.

### 4.2.2 Time from a phase portrait

- $\dot{x} = \frac{\Delta x}{\Delta t} \implies \Delta t \approx \frac{\Delta x}{\dot{x}}$  break into small  $\Delta x$  segments
- $\int dt = \int \frac{1}{x} dx$  so plot phase portrait of x and  $\frac{1}{x}$  then find area under curve

# 4.3 Describing Function Analysis

Equivalent gain, but frequency dependent. Output of nonlinear component can be approximated as

$$c(t) \approx a_1 \cos(\omega t) + b_1 \sin(\omega t) = M \sin(\omega t + \phi)$$

$$a_1 = \frac{1}{\pi} \int_{-\pi}^{\pi} c(t) \cos(\omega t) d(\omega t)$$

$$b_1 = \frac{1}{\pi} \int_{-\pi}^{\pi} c(t) \sin(\omega t) d(\omega t)$$

$$M = \sqrt{a_1^2 + b_1^2} \quad \phi = \arctan\left(\frac{a_1}{b_1}\right)$$

$$N(A, \omega) = \frac{M \angle \phi}{A \angle 0} - \frac{1}{A}(b_1 + ja_1)$$

Limit cycling is obtained when  $G(j\omega)N(A,\omega) = -1$ 

#### 4.4 Linearization

Select operating points, and at each operating point, make a linear approximation to plant dynamics and design a linear controller.

$$\begin{bmatrix} \dot{\Delta x}_1 \\ \dot{\Delta x}_2 \end{bmatrix} = \begin{bmatrix} \frac{\partial g}{\partial x_1} & \frac{\partial g}{\partial x_2} \\ \frac{\partial h}{\partial x_1} & \frac{\partial h}{\partial x_2} \end{bmatrix} \begin{bmatrix} \Delta x_1 \\ \Delta x_2 \end{bmatrix} + \begin{bmatrix} \frac{\partial g}{\partial u} \\ \frac{\partial h}{\partial u} \end{bmatrix} \Delta u = F \Delta x + G \Delta u$$

### 4.5 Sliding Control

Problem formulation is

$$\frac{d^{n}y}{dt^{n}} = f_{1}(y, \frac{dy}{dt}, ..., \frac{d^{n-1}y}{dt^{n-1}}) + g_{1}(y, \frac{dy}{dt}, ..., \frac{d^{n-1}y}{dt^{n-1}})u$$

$$x = \begin{bmatrix} \frac{d^{n-1}y}{dt^{n-1}} & ... & \frac{dy}{dt} & y \end{bmatrix}^{T}$$

$$\dot{x} = \begin{bmatrix} f_{1}(x) \\ x_{1} \\ ... \\ x_{n-1} \end{bmatrix} + \begin{bmatrix} g_{1}(x) \\ 0 \\ ... \\ 0 \end{bmatrix} u = f(x) + g(x)u$$

Define the switching surface

$$\sigma(x) = p^T x = 0$$

which is stable if  $P(s) = p_1 s^{n-1} + p_2 s^{n-2} + ... + p_n$  has all roots in the left-half plane.

Lyapunov function  $V(x) = \frac{\sigma^2(x)}{2}$ 

$$\begin{split} \frac{dV}{dt} &= \sigma(x)\dot{\sigma}(x) = \sigma(x)p^T\dot{x} = \sigma(x)(p^Tf(x) + p^Tg(x)u(t))\\ u(t) &= -\frac{p^Tf}{p^Tg} - \frac{\mu}{p^Tg}\mathrm{sign}(\sigma(x))\\ \frac{dV}{dt} &= -\mu\sigma(x)\mathrm{sign}(\sigma(x)) = -\mu|\sigma(x)| < 0 \end{split}$$

The switching surface is reached in approximately

$$t_{\rho} = \frac{\sigma_0}{\mu}$$

 $\operatorname{sat}(\sigma, \epsilon)$ , which rises linearly from -1 to 1 between  $[-\epsilon, \epsilon]$  can be used in place of sign to prevent chattering.

### 5 Extras

## 5.1 Laplace Transform

$$\mathcal{L}(s) = \int_{0^{-}}^{\infty} f(t)e^{-st}dt$$

f	$\mathcal{L}(f)$	
u(t)	$\frac{1}{s}$	unit step
$\delta(t)$	ĭ	delta function
$\sin(bt)$	$\frac{b}{s^2+b^2}$	sine
$\cos(bt)$	$\frac{s}{s^2+b^2}$	cosine
f'(t)	$sF(s) - f(0^-)$	differentiation
$\int_0^t f(\tau)d\tau$ $tf(t)$	$\frac{F(s)}{s}$	integration
tf(t)	-F'(s)	multiplying by t
$f(t-t_0)$	$e^{-st_0}F(s)$	time shifting
$e^{-at}f(t)$	F(s+a)	frequency shifting
f(at)	$\frac{1}{ a }F(\frac{s}{a})$	time scaling

#### 5.2 Mathematics Identities

- $\sin(A+B) = \sin A \cos B + \cos A \sin B$
- $\cos(A+B) = \cos A \cos B \sin A \sin B$
- $\tan(A + B) = \frac{\tan A + \tan B}{1 \tan A \tan B}$
- $\bullet \ \frac{d}{dx}\tan^{-1}(x) = \frac{1}{1+x^2}$