

Examination Center DGAC

Examination Date _____

Name _____

Firstname _____

Birthday _____

- 1 For a frequency of 200 KHZ, what is the wavelength; (1.00 P.)
- [A] 1500km
 - [B] 150m
 - [C] 1500m
 - [D] 150km

- 2 An aircraft departs from position A ($04^{\circ}10' \text{ S } 178^{\circ}22' \text{ W}$) and flies northward following the meridian for 2950 NM. It then flies westward along the parallel of latitude for 382 NM to position B.
The coordinates of position B are? (1.00 P.)
- [A] $53^{\circ}20' \text{ N } 169^{\circ}22' \text{ W}$
 - [B] $45^{\circ}00' \text{ N } 172^{\circ}38' \text{ E}$
 - [C] $53^{\circ}20' \text{ N } 172^{\circ}38' \text{ E}$
 - [D] $45^{\circ}00' \text{ N } 169^{\circ}22' \text{ W}$
- 3 The angle between the true great-circle track and the true rhumb-line track joining the following points: A ($60^{\circ} \text{ S } 165^{\circ} \text{ W}$) B ($60^{\circ} \text{ S } 177^{\circ} \text{ E}$), at the place of departure A, is: (1.00 P.)
- [A] 5.2°
 - [B] 9°
 - [C] 15.6°
 - [D] 7.8°

- 4 The factors that determine the ranges available from the sky waves are: (1.00 P.)
- [A] Transmission power, diffraction, critical angle and angle of incidence.
 - [B] Transmission power, depth of penetration, critical angle and angle of incidence.
 - [C] Transmission power, conductivity of earth's surface and angle of incidence.
 - [D] Critical angle, refraction, angle of incidence and diffraction.

- 5 In aviation, the reflection on ionosphere layers phenomenon is used in the following frequencies: (1.00 P.)
- [A] HF
 - [B] VHF
 - [C] VLF
 - [D] UHF

- 6 The wavelength of a radio transmitted on frequency 121.95 MHz is: (1.00 P.)
- [A] 24.60 cm
 - [B] 2.46 m
 - [C] 24.60 m
 - [D] 2.46 cm

- 7 Skip distance in relation to the sky wave is; (1.00 P.)
- [A] distance between the receiver and the transmitter.
 - [B] distance between the transmitter and the point of first return of the sky wave on the surface.
 - [C] distance between the point of receiving the surface wave and the first return of the sky wave, where no reception is possible.
 - [D] distance between the point of receiving and the 1st sky wave return on the surface.

- 8 Given: Waypoint 1. $60^{\circ}\text{S } 030^{\circ}\text{W}$
 Waypoint 2. $60^{\circ}\text{S } 020^{\circ}\text{W}$
What will be the approximate latitude shown on the display unit of an inertial navigation system at longitude 025°W ? (1.00 P.)
- [A] $060^{\circ}11'\text{S}$
 - [B] $060^{\circ}00'\text{S}$
 - [C] $060^{\circ}06'\text{S}$
 - [D] $059^{\circ}49'\text{S}$
- 9 What is the time required to travel along the parallel of latitude 60°N between meridians 010°E and 030°W at a groundspeed of 480 kt? (1.00 P.)
- [A] 5 HR 00 MIN
 - [B] 2 HR 30 MIN
 - [C] 1 HR 45 MIN
 - [D] 1 HR 15 MIN

- 10 Night effect when using NDB may be minimized by; (1.00 P.)
- [A] use a low frequency NDB, operate at dusk and dawn, use a high powered NDB and utilise NDB's afar off from the aircraft.
 - [B] using a high frequency NDB, avoid operation at dusk and dawn, use a high powered NDB and utilise NDB's near the aircraft.
 - [C] use a high frequency NDB, operate at dusk and dawn, use a low powered NDB and utilise NDB's near the aircraft.
 - [D] use a low frequency NDB, avoid operations at dusk and dawn, use a high powered NDB and utilise NDB's near the aircraft.

- 11 If a failed RMI rose is stuck on 090° and the ADF pointer indicates 225° , the relative bearing to the station will be: (1.00 P.)
- [A] Impossible to read, due to the RMI failure.
 - [B] 135° .
 - [C] 315° .
 - [D] 225° .

- 12 An aircraft is on Heading 225° (M) ADF RMI reading 315° (M) variation 15° (W), Quadrantal error will be: (1.00 P.)
- [A] Zero.
 - [B] Maximum.
 - [C] proportional to sine heading times the signal strength.
 - [D] proportional to the tangent of the RMI reading.
- 13 An aircraft is maintaining a track outbound from an NDB with a constant relative bearing of 187° and a heading of 060° M. .To return to the NDB, the relative bearing to maintain is; (1.00 P.)
- [A] 247°
 - [B] 007°
 - [C] 067°
 - [D] 353°

- 14 An aircraft is following the 45°N parallel of latitude.
The track followed is a: (1.00 P.)
- [A] constant-drift track
 - [B] constant-heading track
 - [C] great circle
 - [D] rhumb line

- 15 An aircraft on heading 040° M has an ADF reading of 060° Relative. The alteration of heading to intercept the 120° Track inbound to the NDB at 50° in zero wind conditions is; (1.00 P.)
- [A] 40° Right.
 - [B] 50° Right.
 - [C] 30° Right.
 - [D] 20° Right.

- 16 Position A is located on the equator at longitude $130^{\circ}00'E$.
Position B is located 100 NM from A on a bearing of $225^{\circ}(T)$.
The coordinates of position B are: (1.00 P.)
- [A] $01^{\circ}11'S\ 131^{\circ}11'E$
 - [B] $01^{\circ}11'N\ 131^{\circ}11'E$
 - [C] $01^{\circ}11'S\ 128^{\circ}49'E$
 - [D] $01^{\circ}11'N\ 128^{\circ}49'E$

17 The ICAO NDB frequency band is; (1.00 P.)

- [A] 200 khz to 800 khz.
- [B] 200 khz to 1750 khz.
- [C] 200 khz to 112 Mhz.
- [D] 200 khz to 500 khz.

- 18 In order to fly from position A ($10^{\circ}00'N$, $030^{\circ}00'W$) to position B ($30^{\circ}00'N$, $050^{\circ}00'W$), maintaining a constant true course, it is necessary to fly: (1.00 P.)
- [A] a rhumb line track
 - [B] the constant average drift route
 - [C] the great-circle route
 - [D] a straight line plotted on a Lambert chart

- 19 The advantages of Single Side-Band Transmissions as compared with Double Side- Band Transmissions are: (1.00 P.)
- [A] Power is concentrated on frequency.
 - [B] Transmitter power is concentrated on three frequencies.
 - [C] Long range transmission and narrow band width.
 - [D] Narrow band width in the frequency band.

- 20 The purpose of Beat Frequency Oscillator (BFO) in ADF receiver is to; (1.00 P.)
- [A] Remove oscillations of the bearing indicator.
 - [B] Make the incoming signal audible .
 - [C] Remove noise from the incoming signal .
 - [D] Improve the strength of the incoming signal .

- 21 The rhumb line track between position A ($45^{\circ}00'N$, $010^{\circ}00'W$) and position B ($48^{\circ}30'N$, $015^{\circ}00'W$) is approximately: (1.00 P.)
- [A] 300
 - [B] 330
 - [C] 315
 - [D] 345

- 22 When there is a significant fault in the transmission of the ILS, the following will happen; (1.00 P.)
- [A] The needle moves to the centre and the flag appears in the appropriate window .
 - [B] No indication.
 - [C] The alarm flag showing 'ON' appears in the appropriate window.
 - [D] A flag alarm showing 'off' appears on the appropriate window.

23 The diameter of the Earth is approximately: (1.00 P.)

[A] 12 700 km

[B] 40 000 km

[C] 6 350 km

[D] 18 500 km

- 24 The Glide Path angle of an ILS is 3° . What is the elevation coverage? (1.00 P.)
- [A] 0.45° below the GP to 1.75° above the ILS reference point.
 - [B] 1.35° from the horizontal to 5.25° from the horizontal.
 - [C] 8° either side of the ILS reference point.
 - [D] 0.45° below GP to 1.75° above GP.

- 25 With reference to VOR the cone of confusion is; (1.00 P.)
- [A] an area around a VOR station where the signals radiate in a confused pattern.
 - [B] an area around a VOR station where the reading is zero.
 - [C] an area overhead a VOR station in form of a cone with a vertical angle of about 80° where the VOR indications are undetermined.
 - [D] an area overhead a VOR station in the form of an inverted cone which has 10 degrees either side of the vertical where the VOR indications are undetermined.

- 26 The nominal scale of a Lambert conformal conic chart is the: (1.00 P.)
- [A] mean scale between the parallels of the secant cone
 - [B] mean scale between pole and equator
 - [C] scale at the equator
 - [D] scale at the standard parallels

- 27 DME signals are affected by: (1.00 P.)
- [A] night effect and propagation errors.
 - [B] thunderstorm and propagation errors.
 - [C] none of the responses
 - [D] static and night effect.

- 28 The chart that is generally used for navigation in polar areas is based on a: (1.00 P.)
- [A] Stereographical projection
 - [B] Lambert conformal projection
 - [C] Gnomonic projection
 - [D] Direct Mercator projection
- 29 A Mercator chart has a scale at the equator = 1: 3 704 000. What is the scale at latitude 60° S? (1.00 P.)
- [A] 1: 185 200
 - [B] 1: 7 408 000
 - [C] 1: 3 208 000
 - [D] 1: 1 852 000

- 30 What circumstances would cause a warning shortly after take-off when using Ground Proximity Warning System (GPWS)? (1.00 P.)
- [A] Height loss between 50 ft - 700 ft.
 - [B] Height loss between 200 ft - 700 ft.
 - [C] Height loss between 100 ft - 500 ft.
 - [D] Height loss between 100 ft - 700 ft.

- 31 The distance measured between two points on a navigation map is 42 mm (millimetres). The scale of the chart is 1:1 600 000.
The actual distance between these two point is approximately: (1.00 P.)
- [A] 370.00 NM
 - [B] 36.30 NM
 - [C] 3.69 NM
 - [D] 67.20 NM

- 32 In a Doppler display in an aircraft the distance to go on the stage is usually more accurate than the distance left/right of the desired track because: (1.00 P.)
- [A] The heading input is the main source of error and it affects the across track accuracy.
 - [B] The Doppler computer is much more accurate when computing distance to go.
 - [C] none of the above.
 - [D] The land/sea switch enhances sensitivity and therefore ground speed input for the calculation of distance to G is very accurate.

- 33 The standard parallels of a Lambert's conical orthomorphic projection are $07^{\circ}40'N$ and $38^{\circ}20' N$.

The constant of the cone for this chart is: (1.00 P.)

[A] 0.39

[B] 0.60

[C] 0.92

[D] 0.42

- 34 The maximum range of a Radar is governed by; (1.00 P.)
- [A] Pulse interval.
 - [B] Pulse repetition frequency and pulse interval.
 - [C] Pulse repetition frequency.
 - [D] Pulse width.

- 35 On a Lambert conformal conic chart the convergence of the meridians: (1.00 P.)
- [A] is the same as earth convergency at the parallel of origin
 - [B] varies as the secant of the latitude
 - [C] is zero throughout the chart
 - [D] equals earth convergency at the standard parallels

- 36 With respect to Secondary Surveillance Radar (SSR), Mode C provides; (1.00 P.)
- [A] Radio failure when coupled with code A 7600.
 - [B] The range of the Aircraft.
 - [C] Automatic Pressure Attitude Information.
 - [D] Normal Identification of Aircraft.

- 37 State three uses of weather radar in addition to the cloud detection; (1.00 P.)
- [A] ground mapping, detection of Clear Air Turbulence (CAT) determination of approximate cloud heights.
 - [B] safe terrain clearance, determination of approximate cloud heights, ground mapping.
 - [C] indicate height, distance and amount of cloud.
 - [D] safe Terrain Clearance, ground Mapping, Air Traffic Control Separation Purposes.

- 38 A straight line drawn on a chart measures 4.63 cm and represents 150 NM.
The chart scale is: (1.00 P.)
- [A] 1: 1 000 000
 - [B] 1: 5 000 000
 - [C] 1: 3 000 000
 - [D] 1: 6 000 000

- 39 On a Polar Stereographic chart, the initial great circle course from A 70°N 060°W to B 70°N 060°E is approximately: (1.00 P.)
- [A] 330° (T)
 - [B] 210° (T)
 - [C] 030° (T)
 - [D] 150° (T)

- 40 On a Direct Mercator chart, a rhumb line appears as a: (1.00 P.)
- [A] small circle concave to the nearer pole
 - [B] spiral curve
 - [C] straight line
 - [D] curve convex to the nearer pole

- 41 A VOR is sited at position A ($45^{\circ}00'N$, $010^{\circ}00'E$). An aircraft is located at position B ($44^{\circ}00'N$, $010^{\circ}00'E$). Assuming that the magnetic variation at A is $10^{\circ}W$ and at B is $15^{\circ}W$, the aircraft is on VOR radial: (1.00 P.)
- [A] 190°
 - [B] 180°
 - [C] 195°
 - [D] 185°

- 42 If the Barometer reads 1010 hpa at ground level and 1008 hpa at height on top of a building, what is the height of the building? (1.00 P.)
- [A] 60 ft.
 - [B] 120 ft.
 - [C] 30 ft.
 - [D] 90 ft.
- 43 What happens to the capsule in an altimeter as the aircraft gains height? (1.00 P.)
- [A] The value of the static and atmospheric pressure equalise within and around the capsule.
 - [B] The value of the static pressure increases and the capsule contracts under the tension of atmospheric pressure.
 - [C] The value of static pressure decreases and the capsule expands under the tension of the spring.
 - [D] The value of the static pressure increases and the capsule contracts under the tension of the spring.

- 44 The great circle distance between position A ($59^{\circ}34.1'N$ $008^{\circ}08.4'E$) and B ($30^{\circ}25.9'N$ $171^{\circ}51.6'W$) is: (1.00 P.)
- [A] 2 700 NM
 - [B] 10 800 km
 - [C] 10 800 NM
 - [D] 5 400 NM

- 45 Under what conditions are altimeters, calibrated? (1.00 P.)
- [A] MSL pressure 1020 hpa, MSL Temp + 15°C Lapse rate 2° per 1 000 ft from MSL to 20,000 ft etc.
 - [B] MSL pressure 1013.0 hpa, MSL Temp 0°C lapse rate 1.98° per 1000 ft from MSL to 10,000 ft. etc.
 - [C] MSL pressure 1013.2 hpa MSL Temp + 15°C Lapse rate 1.98° per 1 000 ft from MSL to 36,000 etc.
 - [D] MSL pressure 1020 hpa, MSL temp 0°C Lapse rate 2° per 1000 ft from MSL to 55000 ft etc.

- 46 On a Lambert Conformal Conic chart great circles that are not meridians are:
(1.00 P.)
- [A] straight lines regardless of distance
 - [B] curves concave to the parallel of origin
 - [C] curves concave to the pole of projection
 - [D] straight lines within the standard parallels

- 47 What is the cause of pressure error in an altimeter? (1.00 P.)
- [A] Position of the pitot tube.
 - [B] Position of the Aircraft in flight.
 - [C] Position of the altimeter.
 - [D] Position of the static vent.

- 48 On a direct Mercator projection, at latitude 45° North, a certain length represents 70 NM.

At latitude 30° North, the same length represents approximately: (1.00 P.)

- [A] 70 NM
- [B] 86 NM
- [C] 57 NM
- [D] 81 NM

- 49 When flying from an area of high pressure to an area of low pressure, what indication does the altimeter give in relation to the height AGL? (1.00 P.)
- [A] Height change not indicated.
 - [B] Under-reads.
 - [C] Over-reads
 - [D] Zero.

- 50 Given:
 Position A 45°N , $?\text{E}$
 Position B 45°N , $45^{\circ}15'\text{E}$
 Distance A-B = 280 NM
 B is to the East of A
 Required: longitude of position A? (1.00 P.)
- [A] $49^{\circ}57'\text{E}$
[B] $51^{\circ}51'\text{E}$
[C] $38^{\circ}39'\text{E}$
[D] $40^{\circ}33'\text{E}$

- 51 On a polar stereographic projection chart showing the South Pole, a straight line joins position A ($70^{\circ}\text{S } 065^{\circ}\text{E}$) to position B ($70^{\circ}\text{S } 025^{\circ}\text{W}$).
The true course on departure from position A is approximately: (1.00 P.)
- [A] 315°
 - [B] 225°
 - [C] 135°
 - [D] 250°

- 52 On a direct Mercator projection, the distance measured between two meridians spaced 5° apart at latitude 60°N is 8 cm.

The scale of this chart at latitude 60°N is approximately: (1.00 P.)

- [A] 1: 3 500 000
- [B] 1: 6 000 000
- [C] 1: 7 000 000
- [D] 1: 4 750 000

- 53 On a Mercator chart, a great circle is represented as a: (1.00 P.)
- [A] Curve concave to the meridians.
 - [B] Curve concave to the Equator.
 - [C] Straight line.
 - [D] Curve concave to the nearer pole.

- 54 Two positions plotted on a polar stereographic chart, A ($80^{\circ}\text{N } 000^{\circ}$) and B ($70^{\circ}\text{N } 102^{\circ}\text{W}$) are joined by a straight line whose highest latitude is reached at 035°W . At point B, the true course is: (1.00 P.)
- [A] 305°
 - [B] 247°
 - [C] 023°
 - [D] 203°

- 55 What is the meaning of the word "wave length" in relation to radio signals? (1.00 P.)
- [A] The maximum displacement or value a radio signal attains from its mean position during a cycle.
 - [B] The physical distance traveled by the radio wave during one complete cycle of transmission.
 - [C] The number of cycles occurring in one second expressed in Hertz.
 - [D] One complete series of values or one complete process.

- 56 How does the scale vary in a Direct Mercator chart? (1.00 P.)
- [A] The scale increases south of the Equator and decreases north of the Equator.
 - [B] The scale is constant.
 - [C] The scale decreases with increasing distance from the Equator.
 - [D] The scale increases with increasing distance from the Equator.

- 57 The frequency of a radio wave is : (1.00 P.)
- [A] the distance travelled during the transmission of one cycle
 - [B] the rate of change of a wave
 - [C] the number of cycles in one second
 - [D] one complete change of direction of current

- 58 Given:
 Magnetic heading 311°
 Drift angle 10° left
 Relative bearing of NDB 270°
 What is the magnetic bearing of the NDB measured from the aircraft? (1.00 P.)
- [A] 211°
 - [B] 180°
 - [C] 208°
 - [D] 221°

- 59 With reference to basic radio theory, the amplitude of a radio wave is: (1.00 P.)
- [A] one complete change of direction:
 - [B] the peak value of the current in either direction
 - [C] the longitudinal displacement of a sine wave
 - [D] the number of cycles in one second

- 60 A DME station is located 1000 feet above MSL.
An aircraft flying at FL 370 in ISA conditions which is 15 NM away from the DME station, will have a DME reading of: (1.00 P.)
- [A] 16 NM
 - [B] 14 NM
 - [C] 17 NM
 - [D] 15 NM

- 61 The frequency which corresponds to a wavelength of 12 cm is ; (1.00 P.)
- [A] 2 500 MHz
 - [B] 250 GHz
 - [C] 2 500 kHz
 - [D] 2 500 Hz

- 62 Given the following:
True track: 192°
Magnetic variation: 7°E
Drift angle: 5° left
What is the magnetic heading required to maintain the given track? (1.00 P.)
- [A] 190°
 - [B] 194°
 - [C] 180°
 - [D] 204°

- 63 The frequency which corresponds to a wavelength of 1500 metres is (1.00 P.)
- [A] 400 Hz
 - [B] 200 KHz
 - [C] 2 000 GHz
 - [D] 400 MHz

- 64 Given the following:
Magnetic heading: 060°
Magnetic variation: 8°W
Drift angle: 4° right
What is the true track? (1.00 P.)
- [A] 072°
 - [B] 048°
 - [C] 056°
 - [D] 064°

65 If the transmission frequency is 75 MHz, the wavelength is (1.00 P.)

- [A] 4 km
- [B] 800 m
- [C] 8 cm
- [D] 4 metres

- 66 An aircraft is following a true track of 048° at a constant TAS of 210 kt.
The wind velocity is $350^\circ / 30$ kt.
The GS and drift angle are: (1.00 P.)
- [A] 192 kt, 7° left
 - [B] 200 kt, 3.5° right
 - [C] 192 kt, 7° right
 - [D] 225 kt, 7° left

- 67 If the frequency of a radio is 1 439 kHz, the wavelength is (1.00 P.)
- [A] 2.0847 cm
 - [B] 208.47 metres
 - [C] 208.47 cm
 - [D] 0.20847 metres

- 68 Given:
 FL 350,
 Mach 0.80,
 OAT -55°C.
 Calculate the values for TAS and local speed of sound (LSS)? (1.00 P.)
- [A] 461 kt , LSS 576 kt
 - [B] 237 kt, LSS 296 kt
 - [C] 461 kt , LSS 296 kt
 - [D] 490 kt, LSS 461 kt

- 69 Given:
Magnetic heading = 255°
VAR = 40° W
GS = 375 kt
W/V = 235° (T) / 120 kt
Calculate the drift angle? (1.00 P.)
- [A] 3° left
 - [B] 6° right
 - [C] 6° left
 - [D] 9° left

- 70 The wavelength of a radio wave transmission is: (1.00 P.)
- [A] one complete change of direction of current
 - [B] the lateral displacement of a wave
 - [C] the distance travelled during the transmission of one cycle
 - [D] the number of cycles in one second

- 71 Given:
 True Heading = 180°
 TAS = 500 kt
 W/V 225° / 100 kt
 Calculate the GS? (1.00 P.)
- [A] 535 kt
[B] 450 kt
[C] 435 kt
[D] 600 kt
-
- 72 Given:
 True heading = 310°
 TAS = 200 kt
 GS = 176 kt
 Drift angle 7° right.
 Calculate the W/V? (1.00 P.)
- [A] 360° / 33 kt
[B] 270° / 33 kt
[C] 180° / 33 kt
[D] 090° / 33 kt

- 73 Which of the following will give the most accurate calculation of aircraft ground speed? (1.00 P.)
- [A] A DME station sited across the flight route
 - [B] A VOR station sited on the flight route
 - [C] A DME station sited on the flight route
 - [D] An ADF sited on the flight route

- 74 Polarisation is the term used to describe the plane of oscillation of the; (1.00 P.)
- [A] the magnetic field of an electromagnetic wave
 - [B] electrical field of an electromagnetic wave
 - [C] the electrical and magnetic field of an electromagnetic wave
 - [D] none of the above

- 75 If an aeroplane was to circle around the Earth following parallel 60°N at a ground speed of 480 kt. In order to circle around the Earth along the equator in the same amount of time, it should fly at a ground speed of: (1.00 P.)
- [A] 240 kt
 - [B] 550 kt
 - [C] 480 kt
 - [D] 960 kt

- 76 Polarisation is the term used to describe the plane of oscillation of the; (1.00 P.)
- [A] electrical field of an electromagnetic wave
 - [B] the electrical and magnetic field of an electromagnetic wave
 - [C] the magnetic field of an electromagnetic wave
 - [D] none of the above

- 77 Given:
True Heading = 090°
TAS = 180 kt
GS = 180 kt
Drift 5° right
Calculate the W/V? (1.00 P.)
- [A] $005^\circ / 15$ kt
[B] $190^\circ / 15$ kt
[C] $355^\circ / 15$ kt
[D] $185^\circ / 15$ kt

- 78 Given:
True Heading = 090°
TAS = 200 kt
W/V = $220^\circ / 30$ kt.
Calculate the GS? (1.00 P.)
- [A] 220 kt
[B] 230 kt
[C] 180 kt
[D] 200 kt

79 Radio waves travel at the speed of light which is taken to be constant at (1.00 P.)

[A] $3 \times 120\text{nm/s}$

[B] $3 \times 10^{10} \text{ km/sec}$

[C] $3 \times 10^8 \text{ m/sec}$

[D] $3 \times 10^5 \text{ cm/sec}$

80 Radio waves travel at the speed of light which is taken to be constant at (1.00 P.)

[A] $3 \times 10^8 \text{ m/sec}$

[B] $3 \times 10^{10} \text{ km/sec}$

[C] $3 \times 120\text{nm/s}$

[D] $3 \times 10^5 \text{ cm/sec}$

- 81 An aeroplane is flying at TAS 180 kt on a track of 090°. The W/V is 045° / 50kt. How far can the aeroplane fly out from its base and return in one hour? (1.00 P.)
- [A] 85 NM
 - [B] 176 NM
 - [C] 88 NM
 - [D] 56 NM
- 82 The following information is displayed on an Inertial Navigation System:
GS 520 kt,
True HDG 090°,
Drift angle 5° right,
TAS 480 kt.
SAT (static air temperature) -51°C.
The W/V being experienced is: (1.00 P.)
- [A] 225° / 60 kt
 - [B] 220° / 60 kt
 - [C] 325° / 60 kt
 - [D] 320° / 60 kt

- 83 The frequency corresponding to a wavelength of 750 metres is (1.00 P.)
- [A] 400 KHz
 - [B] 400 MHz
 - [C] 400 GHz
 - [D] 400 Hz

- 84 The reported surface wind from the Control Tower is 240°/35 kt. Runway 30 (300°).

What is the cross-wind component? (1.00 P.)

- [A] 27 kt
- [B] 24 kt
- [C] 21 kt
- [D] 30 kt

- 85 An aircraft passes position A ($60^{\circ}00'N$ $120^{\circ}00'W$) on route to position B ($60^{\circ}00'N$ $140^{\circ}30'W$).
What is the great circle track on departure from A? (1.00 P.)
- [A] 279°
 - [B] 261°
 - [C] 270°
 - [D] 288°
- 86 A great circle track joins position A ($59^{\circ}S$ $141^{\circ}W$) and B ($61^{\circ}S$ $148^{\circ}W$).
What is the difference between the great circle track at A and B? (1.00 P.)
- [A] It decreases by 6°
 - [B] It increases by 3°
 - [C] It increases by 6°
 - [D] It decreases by 3°
- 87 What is the longitude of a position 6 NM to the east of $58^{\circ}42'N$ $094^{\circ}00'W$? (1.00 P.)
- [A] $093^{\circ}53.1'W$
 - [B] $094^{\circ}12.0'W$
 - [C] $093^{\circ}54.0'W$
 - [D] $093^{\circ}48.5'W$

- 88 If the frequency of a radar set is 13 500 MHz, the wavelength is (1.00 P.)
- [A] 2.22 metres
 - [B] 2.22 cm
 - [C] 0.22 cm
 - [D] 0.22 metres

- 89 An aircraft is maintaining a 5.2% gradient is at 7 NM from the runway, on a flat terrain; its height is approximately: (1.00 P.)
- [A] 2210 FT
 - [B] 680 FT
 - [C] 3640 FT
 - [D] 1890 FT

- 90 The amplitude modulation and the colour of an outer marker (OM) is: (1.00 P.)
- [A] 3000 Hz, blue
 - [B] 400 Hz, amber
 - [C] 1300 Hz, blue
 - [D] 400 Hz, blue

- 91 An RMI indicates aircraft heading and bearing. To convert the RMI bearings of NDBs and VORs to true bearings the correct combination for the application of magnetic variation is: (1.00 P.)
- [A] NDB: aircraft position
VOR: aircraft position
 - [B] NDB: beacon position
VOR: beacon position
 - [C] NDB: aircraft position
VOR: beacon position
 - [D] NDB: beacon position
VOR: aircraft position

- 92 Radio intelligence and information are relayed from a transmitter to a receiver by;
(1.00 P.)
- [A] phase difference
 - [B] polarisation
 - [C] line-of-sight
 - [D] modulation

- 93 An aircraft is flying on the true track 090° towards a VOR station located near the equator where the magnetic variation is 15°E . The variation at the aircraft position is 8°E .
The aircraft is on VOR radial: (1.00 P.)
- [A] 285°
 - [B] 255°
 - [C] 278°
 - [D] 262°
- 94 Given:
Magnetic heading 280°
VOR radial 090°
What bearing should be selected on the omni-bearing selector in order to centralise the VOR deviation needle with a "TO" indication? (1.00 P.)
- [A] 280°
 - [B] 270°
 - [C] 100°
 - [D] 090°

- 95 Amplitude modulation is; (1.00 P.)
- [A] All of the options
 - [B] varying the amplitude of the audio frequency in accordance with the change in amplitude of the carrier, keeping the frequency of the carrier constant
 - [C] varying the frequency of the carrier in accordance with the change in the amplitude of the audio, keeping the amplitude of the carrier constant
 - [D] varying the amplitude of the carrier wave in accordance with the change in amplitude of the audio modulating signal keeping the carrier frequency constant

- 96 A VOR is sited at position $58^{\circ}00'N$ $073^{\circ}00'W$ where the magnetic variation equals $32^{\circ}W$.
An aircraft is located at position $56^{\circ}00'N$ $073^{\circ}00'W$ where the magnetic variation equals $28^{\circ}W$.
The aircraft is on VOR radial: (1.00 P.)
- [A] 212
 - [B] 360
 - [C] 208
 - [D] 180
- 97 In order to plot a bearing from a VOR station, a pilot needs to know the magnetic variation: (1.00 P.)
- [A] at the half-way point between the aircraft and the station
 - [B] at the aircraft location
 - [C] at both the VOR and aircraft
 - [D] at the VOR

98 Amplitude modulation is; (1.00 P.)

- [A] All of the options
- [B] varying the amplitude of the carrier wave in accordance with the change in amplitude of the audio modulating signal keeping the carrier frequency constant
- [C] varying the amplitude of the audio frequency in accordance with the change in amplitude of the carrier, keeping the frequency of the carrier constant
- [D] varying the frequency of the carrier in accordance with the change in the amplitude of the audio, keeping the amplitude of the carrier constant

- 99 An aeroplane flies over position A which is due North of a VOR station sited at position B.
The magnetic variation at A is 18°W , and at B is 10°W .
What radial from B is the aircraft on? (1.00 P.)
- [A] 018°
 - [B] 342°
 - [C] 350°
 - [D] 010°

- 100 Two advantages of single side band transmissions are; (1.00 P.)
- [A] narrower bandwidth and transmitting power concentrated in two frequencies instead of three
 - [B] narrower bandwidth and transmitting power concentrated in three frequencies instead of two
 - [C] broader bandwidth and transmitting power concentrated in two frequencies instead of three
 - [D] none of the above

- 101 An aircraft DME receiver does not lock on to its own transmissions reflected from the ground because: (1.00 P.)
- [A] DME transmits twin pulses
 - [B] the pulse recurrence rates are varied
 - [C] they are not on the receiver frequency
 - [D] DME uses the UHF band
- 102 The DME (Distance Measuring Equipment) operates within the following frequencies: (1.00 P.)
- [A] 329 to 335 MHz
 - [B] 962 to 1213 kHz.
 - [C] 962 to 1213 MHz
 - [D] 108 to 118 MHz
- 103 A DME is located at MSL.
An aircraft passing vertically above the station at flight level FL 360 will obtain a DME range of approximately: (1.00 P.)
- [A] 8 NM
 - [B] 11 NM
 - [C] 6 NM
 - [D] 7 NM
- 104 During a flight at FL 210, a pilot does not receive any DME distance indication from a DME station located approximately 220 NM away.
The reason for this is that the: (1.00 P.)
- [A] range of a DME system is always less than 200 NM
 - [B] altitude is too high
 - [C] aeroplane is below the 'line of sight' altitude
 - [D] aeroplane is circling around the station

- 105 When considering factors affecting radio wave propagation it can be said that;
(1.00 P.)
- [A] as frequency is increased ionospheric attenuation increases
 - [B] as frequency is increased surface attenuation decreases
 - [C] as frequency is increased surface attenuation increases
 - [D] as frequency is increased ionospheric attenuation decreases

- 106 What is the approximate angular coverage of reliable navigation information for a 3° ILS glide path out to a minimum distance of 10 NM? (1.00 P.)
- [A] 1.35° above the horizontal to 5.25° above the horizontal and 8° each side of the localiser centreline
 - [B] 3° above and below the glide path and 10° each side of the localiser centreline
 - [C] 0.45° above the horizontal to 1.75° above the glide path and 8° each side of the localiser centreline
 - [D] 0.7° above and below the glide path and 2.5° each side of the localiser centreline

- 107 An aircraft is descending down a 6% slope whilst maintaining a G/S of 300 kt. The rate of descent of the aircraft is approximately: (1.00 P.)
- [A] 1800 FT/MIN
 - [B] 3600 FT/MIN
 - [C] 900 FT/MIN
 - [D] 10800 FT/MIN

- 108 Skip distance is the distance between; (1.00 P.)
- [A] successive sky wave touchdown points
 - [B] the transmitter and the first point of sky wave touchdown
 - [C] the end of the ground wave and the first point of touchdown
 - [D] the distance of a wave from the surface to the ozonosphere

- 109 When Mode C is selected on the aircraft SSR transponder the additional information transmitted is: (1.00 P.)
- [A] altitude based on regional QNH
 - [B] flight level based on 1013.25 hPa
 - [C] height based on QFE
 - [D] aircraft height based on sub-scale setting

- 110 Assuming sufficient transmission power, the maximum range of a ground radar with a pulse repetition frequency of 450 pulses per second is: (Given: velocity of light is 300 000 km/second) (1.00 P.)
- [A] 1333 km
 - [B] 333 km
 - [C] 150 km
 - [D] 666 km

- 111 Skip distance is the distance between; (1.00 P.)
- [A] the distance of a wave from the surface to the ozonosphere
 - [B] the transmitter and the first point of sky wave touchdown
 - [C] successive sky wave touchdown points
 - [D] the end of the ground wave and the first point of touchdown

- 112 The duration of civil twilight is the time: (1.00 P.)
- [A] agreed by the international aeronautical authorities which is 12 minutes
 - [B] between sunset and when the centre of the sun is 12° below the true horizon
 - [C] between sunset and when the centre of the sun is 6° below the true horizon
 - [D] needed by the sun to move from the apparent height of 0° to the apparent height of 6°
- 113 On the 27th of February, at 52°S and 040°E , the sunrise is at 0243 UTC.
On the same day, at 52°S and 035°W , the sunrise is at: (1.00 P.)
- [A] 0243 UTC
 - [B] 0523 UTC
 - [C] 0743 UTC
 - [D] 2143 UTC

- 114 An aeroplane flies from A ($59^{\circ}\text{S } 142^{\circ}\text{W}$) to B ($61^{\circ}\text{S } 148^{\circ}\text{W}$) with a TAS of 480 kt.
The autopilot is engaged and coupled with an Inertial Navigation System in which AB track is active.
On route AB, the true track: (1.00 P.)
- [A] increases by 5°
 - [B] varies by 4°
 - [C] varies by 10°
 - [D] decreases by 6°

- 115 When using HF communications at night the best frequency is one which is;
(1.00 P.)
- [A] twice the day frequency
 - [B] the same as the frequency for day operation
 - [C] thrice the day frequency
 - [D] half the day frequency

- 116 The rhumb-line distance between points A ($60^{\circ}00'N$ $002^{\circ}30'E$) and B ($60^{\circ}00'N$ $007^{\circ}30'W$) is: (1.00 P.)
- [A] 600 NM
 - [B] 450 NM
 - [C] 150 NM
 - [D] 300 NM

- 117 An aircraft is over position HO (55°30'N 060°15'W), where YYR VOR (53°30'N 060°15'W) can be received. The magnetic variation is 31°W at HO and 28°W at YYR.

What is the radial from YYR? (1.00 P.)

- [A] 028°
- [B] 332°
- [C] 031°
- [D] 208°

- 118 When using HF communications at night the best frequency is one which is;
(1.00 P.)
- [A] twice the day frequency
 - [B] half the day frequency
 - [C] the same as the frequency for day operation
 - [D] thrice the day frequency

- 119 Given:
TAS = 485 kt,
OAT = ISA +10°C,
FL 410.
Calculate the Mach Number? (1.00 P.)
- [A] 0.87
 - [B] 0.90
 - [C] 0.825
 - [D] 0.85

- 120 An aircraft at FL100 should be able to communicate with a VHF ground station at 100 ft amsl at an approximate maximum range of (1.00 P.)
- [A] 137.5 nm
 - [B] 25 nm
 - [C] 112.5 nm
 - [D] 123.2 nm

121 060-001.jpg

Assume a North polar stereographic chart whose grid is aligned with the Greenwich meridian.

An aircraft flies from the geographic North pole for a distance of 480 NM along the 110°E meridian, then follows a grid track of 154° for a distance of 300 NM.

Its position is now approximately: (1.00 P.)

Siehe Anlage 1

[A] $78^{\circ}45'\text{N } 087^{\circ}\text{E}$

[B] $70^{\circ}15'\text{N } 080^{\circ}\text{E}$

[C] $80^{\circ}00'\text{N } 080^{\circ}\text{E}$

[D] $79^{\circ}15'\text{N } 074^{\circ}\text{E}$

122 Given:

A polar stereographic chart whose grid is aligned with the zero meridian.

Grid track 344° ,

Longitude $115^{\circ}00'\text{W}$,

Calculate the true course? (1.00 P.)

[A] 049°

[B] 279°

[C] 099°

[D] 229°

- 123 The ionosphere is split into three distinct layers during day time which are; (1.00 P.)
- [A] E layer, Appleton layer, F layer
 - [B] D layer, Appleton layer, F layer
 - [C] E layer, Kennelley Heaviside layer, D layer
 - [D] D layer, Kennelley Heaviside layer , Appleton layer

- 124 060-002.jpg
1300 UTC DR position $37^{\circ}30'N$ $021^{\circ}30'W$ alter heading
PORT SANTO NDB ($33^{\circ}03'N$ $016^{\circ}23'W$)
TAS 450 kt,
Forecast W/V $360^{\circ}/30kt$.
Calculate the ETA at PORT SANTO NDB? (1.00 P.)

Siehe Anlage 2

- [A] 1344
- [B] 1341
- [C] 1354
- [D] 1348

- 125 For a distance of 1860 NM between Q and R, a ground speed "out" of 385 kt, a ground speed "back" of 465 kt and an endurance of 8 HR (excluding reserves) the distance from Q to the point of safe return (PSR) is: (1.00 P.)
- [A] 1532 NM
 - [B] 1865 NM
 - [C] 1685 NM
 - [D] 930 NM

- 126 The aircraft aerial(s) used to determine the direction of an NDB beacon is/are (1.00 P.)
- [A] sense aerial
 - [B] loop and sense aerals
 - [C] omni-directional aerals
 - [D] loop aerial

- 127 Two points A and B are 1000 NM apart. TAS = 490 kt.
On the flight between A and B the equivalent headwind is -20 kt.
On the return leg between B and A, the equivalent headwind is +40 kt.
What distance from A, along the route A to B, is the Point of Equal Time (PET)?
(1.00 P.)
- [A] 455 NM
 - [B] 470 NM
 - [C] 530 NM
 - [D] 500 NM
- 128 Given:
AD = Air distance
GD = Ground distance
TAS = True Airspeed
GS = Groundspeed
Which of the following is the correct formula to calculate ground distance (GD)
gone? (1.00 P.)
- [A] $GD = (AD - TAS)/TAS$
 - [B] $GD = TAS/(GS \times AD)$
 - [C] $GD = (AD \times GS)/TAS$
 - [D] $GD = AD \times (GS - TAS)/GS$

- 129 An aircraft is flying with the aid of an inertial navigation system (INS) connected to the autopilot. The following two points have been entered in the INS computer:
WPT 1: 60°N 030°W
WPT 2: 60°N 020°W
When 025°W is passed the latitude shown on the display unit of the inertial navigation system will be: (1.00 P.)
- [A] 60°11.0'N
 - [B] 60°00.0'N
 - [C] 59°49.0'N
 - [D] 60°05.7'N

- 130 The drift of the azimuth gyro on an inertial unit induces an error in the position given by this unit. "t" being the elapsed time.
The total error is: (1.00 P.)
- [A] proportional to $t/2$
 - [B] sinusoidal
 - [C] proportional to the square of time, t^2
 - [D] proportional to t

- 131 When determining the direction of an NDB station, the 180° ambiguity is resolved by using a; (1.00 P.)
- [A] loop aerial
 - [B] sense aerial and then a loop aerial
 - [C] guyed aerial
 - [D] sense aerial

- 132 With reference to inertial navigation systems, a TAS input is: (1.00 P.)
- [A] required for Polar navigation
 - [B] not required
 - [C] required for rhumb line navigation
 - [D] required to provide a W/V read out
- 133 The platform of an inertial navigation system (INS) is maintained at right angles to the local vertical by applying corrections for the effects of: (1.00 P.)
- [A] movement in the yawing plane, secondary precession and pendulous oscillation
 - [B] aircraft manoeuvres, earth rotation, transport wander and coriolis
 - [C] vertical velocities, earth precession, centrifugal forces and transport drift
 - [D] gyroscopic inertia, earth rotation and real drift

- 134 An aircraft travels from point A to point B, using the autopilot connected to the aircraft's inertial system. The coordinates of A ($45^{\circ}\text{S } 010^{\circ}\text{W}$) and B ($45^{\circ}\text{S } 030^{\circ}\text{W}$) have been entered.
The true course of the aircraft on its arrival at B, to the nearest degree, is: (1.00 P.)
- [A] 263°
 - [B] 277°
 - [C] 284°
 - [D] 270°

- 135 NDB bearings displayed on an ADF are relative bearings with reference to; (1.00 P.)
- [A] true North
 - [B] magnetic North
 - [C] aircraft track
 - [D] aircraft heading

- 136 Some inertial reference and navigation systems are known as "strapdown". This means that: (1.00 P.)
- [A] gyros and accelerometers are mounted on a stabilised platform in the aircraft
 - [B] the gyroscopes and accelerometers become part of the unit's fixture to the aircraft structure
 - [C] only the gyros, and not the accelerometers, become part of the unit's fixture to the aircraft structure
 - [D] gyros and accelerometers need satellite information input to obtain a vertical reference

- 137 As the INS position of the departure aerodrome, coordinates 35°32.7'N 139°46.3'W are input instead of 35°32.7'N 139°46.3'E. When the aircraft subsequently passes point 52°N 180°W, the longitude value shown on the INS will be: (1.00 P.)
- [A] 080° 27.4'W
 - [B] 080° 27.4'E
 - [C] 099° 32.6'W
 - [D] 099° 32.6'E

- 138 In order to maintain an accurate vertical using a pendulous system, an aircraft inertial platform incorporates a device: (1.00 P.)
- [A] without damping and a period of 84.4 MIN
 - [B] without damping and a period of 84.4 SEC
 - [C] with damping and a period of 84.4 SEC
 - [D] with damping and a period of 84.4 MIN

- 139 The cardoid polar diagram resulting from the resolution of the 180° ambiguity has; (1.00 P.)
- [A] two nulls and one maximum, giving the direction of the station
 - [B] two maxima and non nulls
 - [C] one null giving the direction of the station
 - [D] four nulls

- 140 What is the approximate maximum theoretical range at which an aircraft at FL130 could receive information from a VDF facility which is sited 1024 FT above MSL? (1.00 P.)
- [A] 220 NM
 - [B] 150 NM
 - [C] 180 NM
 - [D] 120 NM
- 141 In ISA conditions, what is the maximum theoretical range at which an aircraft at FL80 can expect to obtain bearings from a ground VDF facility sited 325 ft above MSL ? (1.00 P.)
- [A] 158 NM
 - [B] 114 NM
 - [C] 107 NM
 - [D] 134 NM

- 142 The aerals used when an ADF operates on "Automatic Direction Finding" are;
(1.00 P.)
- [A] loop and sense aerals together
 - [B] loop aerial only
 - [C] sense aerial only
 - [D] omni-directional aerals

- 143 A radio beacon has an operational range of 10 NM. By what factor should the transmitter power be increased in order to achieve an operational range of 20 NM? (1.00 P.)
- [A] Four
 - [B] Eight
 - [C] Two
 - [D] Six
- 144 'Night Effect' which causes loss of signal and fading, resulting in bearing errors from NDB transmissions, is due to: (1.00 P.)
- [A] skywave distortion of the null position and is maximum at dawn and dusk
 - [B] static activity increasing at night particularly in the lower frequency band
 - [C] the effect of the Aurora Borealis
 - [D] interference from other transmissions and is maximum at dusk when east of the NDB
- 145 Quadrantal errors associated with aircraft Automatic Direction Finding (ADF) equipment are caused by: (1.00 P.)
- [A] signal bending caused by electrical interference from aircraft wiring
 - [B] skywave/groundwave contamination
 - [C] misalignment of the loop aerial
 - [D] signal bending by the aircraft metallic surfaces
- 146 Errors caused by the effect of coastal refraction on bearings at lower altitudes are maximum when the NDB is: (1.00 P.)
- [A] near the coast and the bearing crosses the coast at an acute angle
 - [B] near the coast and the bearing crosses the coast at right angles
 - [C] inland and the bearing crosses the coast at an acute angle
 - [D] inland and the bearing crosses the coast at right angles

- 147 The four factors affecting the accuracy of NDB's are; (1.00 P.)
- [A] quadrantal error, mountain effect, night effect and static interference
 - [B] quadrantal error, night effect, pilotage error and aggregate error
 - [C] thunderstorm effect, night effect, slant range error, and station interference
 - [D] all of the options.

148 The principle used in VOR bearing measurement is: (1.00 P.)

- [A] beat frequency discrimination
- [B] phase comparison
- [C] difference in depth of modulation
- [D] envelope matching

149 Which frequency band is used by VOR transmissions? (1.00 P.)

- [A] SHF
- [B] VHF
- [C] HF
- [D] UHF

- 150 Precautions to reduce the effect of night effect on ADF bearings are to; (1.00 P.)
- [A] use the transmitter/receiver intermittently by switching "on" and "off"
 - [B] choose the nearest station and only use the ADF within one hour of sunrise and sunset
 - [C] avoid use the ADF within one hour of sunrise and sunset and use a weaker beacon to eliminate skywaves
 - [D] use a more powerfull beacon and bearings taken well within the surface wave range

- 151 Transmissions from VOR facilities may be adversely affected by: (1.00 P.)
- [A] uneven propagation over irregular ground surfaces
 - [B] night effect
 - [C] quadrantal error
 - [D] static interference
- 152 If VOR bearing information is used beyond the published protection range, errors could be caused by: (1.00 P.)
- [A] interference from other transmitters
 - [B] noise from precipitation static exceeding the signal strength of the transmitter
 - [C] sky wave interference from the same transmitter
 - [D] sky wave interference from distant transmitters on the same frequency
- 153 An aircraft is 100 NM from a VOR facility. Assuming no error when using a deviation indicator where 1 dot = 2° deviation, how many dots deviation from the centre line of the instrument will represent the limits of the airway boundary? (Assume that the airway is 10 NM wide) (1.00 P.)
- [A] 3.0
 - [B] 4.5
 - [C] 6.0
 - [D] 1.5
- 154 An airway 10 NM wide is to be defined by two VORs each having a resultant bearing accuracy of plus or minus 5.5° . In order to ensure accurate track guidance within the airway limits the maximum distance apart for the transmitter is approximately: (1.00 P.)
- [A] 210 NM
 - [B] 105 NM
 - [C] 50 NM
 - [D] 165 NM

- 155 The time between two relative bearings of 075 and 090 is 7 minutes 45 seconds and groundspeed is 130 kts. The time and distance to the beacon is; (1.00 P.)
- [A] 67.2 minutes - 31 nm
 - [B] 31 minutes - 67.2 nm
 - [C] 150.7 minutes - 50.5 nm
 - [D] 116.1 minutes - 251.6 nm

- 156 An aircraft is required to approach a VOR via the 104° radial. Which of the following settings should be made on the VOR/ILS deviation indicator? (1.00 P.)
- [A] 104° with the FROM flag showing
 - [B] 284° with the TO flag showing
 - [C] 104° with the TO flag showing
 - [D] 284° with the FROM flag showing

- 157 The time between two relative bearings of 075 and 090 is 7 minutes 45 seconds and groundspeed is 130 kts. The time and distance to the beacon is; (1.00 P.)
- [A] 116.1 minutes - 251.6 nm
 - [B] 67.2 minutes - 31 nm
 - [C] 31 minutes - 67.2 nm
 - [D] 150.7 minutes - 50.5 nm

- 158 An aircraft, on a heading of 180°M is on a bearing of 270°M from a VOR. The bearing you should select on the OMNI bearing selector to centralise the VOR/ILS left/right deviation needle is: (1.00 P.)
- [A] 090°
 - [B] 270°
 - [C] 360°
 - [D] 180°
- 159 An aircraft is required to approach a VOR station via the 244° radial. In order to obtain correct sense indications the deviation indicator should be set to: (1.00 P.)
- [A] 244° with the TO flag showing
 - [B] 064° with the TO flag showing
 - [C] 244° with the FROM flag showing
 - [D] 064° with the FROM flag showing
- 160 What is the maximum theoretical range that an aircraft at FL150 can receive signals from a VOR situated 609 feet above MSL? (1.00 P.)
- [A] 220 NM
 - [B] 147 NM
 - [C] 156 NM
 - [D] 184 NM

- 161 An aircraft heading 200°M receives a bearing of 190R from an NDB. ATC instructs the pilot to intercept the 250 QDR outbound from the NDB at 30°. The intercept heading and the relative bearing at intercept are; (1.00 P.)
- [A] 280°M/150
 - [B] 280°M/330
 - [C] 220°M/210
 - [D] 220°M/160

- 162 A typical frequency employed in Distance Measuring Equipment (DME) is: (1.00 P.)
- [A] 1000 MHz
 - [B] 100 MHz
 - [C] 10 MHz
 - [D] 100 GHz
-
- 163 Distance Measuring Equipment (DME) operates in the: (1.00 P.)
- [A] VHF band and uses the principle of phase comparison
 - [B] UHF band and uses one frequency
 - [C] SHF band and uses frequency modulation techniques
 - [D] UHF band and uses two frequencies

- 164 An aircraft on a constant heading with 8° right drift is tracking parallel to and 5 nm left of the centre line of an airway. The ADF reading of an NDB on the centre line 42 nm ahead is; (1.00 P.)
- [A] 286°R
 - [B] 011°R
 - [C] 359°R
 - [D] 015°R

- 165 For a conventional DME facility 'Beacon Saturation' will occur whenever the number of aircraft interrogations exceeds: (1.00 P.)
- [A] 100
 - [B] 80
 - [C] 60
 - [D] 200

166 An aircraft on a constant heading with 8° right drift is tracking parallel to and 5 nm left of the centre line of an airway. The ADF reading of an NDB on the centre line 42 nm ahead is; (1.00 P.)

[A] 015°R

[B] 011°R

[C] 359°R

[D] 286°R

- 167 The aircraft DME receiver is able to accept replies to its own transmissions and reject replies to other aircraft interrogations because: (1.00 P.)
- [A] aircraft interrogation signals and transponder responses are 63 MHz removed from each other
 - [B] pulse pairs are amplitude modulated with the aircraft registration
 - [C] the time interval between pulse pairs is unique to that particular aircraft
 - [D] transmission frequencies are 63 MHz different for each aircraft
- 168 The aircraft DME receiver cannot lock on to interrogation signals reflected from the ground because: (1.00 P.)
- [A] reflections are subject to doppler frequency shift
 - [B] DME transmits twin pulses
 - [C] DME pulse recurrence rates are varied
 - [D] aircraft transmitter and DME ground station are transmitting on different frequencies

- 169 An aircraft is tracking 065°T , variation 22°W and drift 9° right with an ADF indication of 237°R . The aircraft heading and indication on an RMI are; (1.00 P.)
- [A] heading $078^{\circ}\text{M}/315^{\circ}$
 - [B] heading $088^{\circ}\text{M}/326^{\circ}$
 - [C] heading $096^{\circ}\text{M}/333^{\circ}$
 - [D] heading $034^{\circ}\text{M}/271^{\circ}$

- 170 The design requirements for DME stipulate that, at a range of 100 NM, the maximum systematic error should not exceed: (1.00 P.)
- [A] + or - 0.25 NM
 - [B] + or - 1.5 NM
 - [C] + or - 1.25 NM
 - [D] + or - 3 NM
- 171 In which situation will speed indications on an airborne Distance Measuring Equipment (DME) most closely represent the groundspeed of an aircraft flying at FL400? (1.00 P.)
- [A] When overhead the station, with no change of heading at transit
 - [B] When passing abeam the station and within 5 NM of it
 - [C] When tracking directly towards the station at a range of 100 NM or more
 - [D] When tracking directly away from the station at a range of 10 NM

- 172 In order to maintain a track of 165°T away from an NDB (variation 16°W) with a drift 8° left, the ADF reading required is; (1.00 P.)
- [A] 188° relative
 - [B] 352° relative
 - [C] 172° relative
 - [D] 259° relative

- 173 The time taken for the transmission of an interrogation pulse by a Distance Measuring Equipment (DME) to travel to the ground transponder and return to the airborne receiver was 2000 micro-second, including time delay. The slant range from the ground transponder was: (1.00 P.)
- [A] 330 NM
 - [B] 296 NM
 - [C] 158 NM
 - [D] 186 NM

- 174 The reason why pre take-off holding areas are sometimes further from the active runway when ILS Category 2 and 3 landing procedures are in progress than during good weather operations is: (1.00 P.)
- [A] heavy precipitation may disturb guidance signals
 - [B] aircraft manoeuvring near the runway may disturb guidance signals
 - [C] to increase aircraft separation in very reduced visibility conditions
 - [D] to increase distance from the runway during offset approach operations
- 175 Which of the following correctly describes the Instrument Landing System (ILS) localiser radiation pattern? (1.00 P.)
- [A] Two overlapping lobes on the same VHF carrier frequency
 - [B] A pencil beam comprising a series of smaller beams each carrying a different modulation
 - [C] Two overlapping lobes on different radio carrier frequencies but with the same modulation
 - [D] Two overlapping lobes on the same UHF carrier frequency

176 The principle of operation of a VOR is bearing measurement by; (1.00 P.)

- [A] modulation comparison
- [B] phase comparison
- [C] polarisation comparison
- [D] de-modulation comparison

- 177 An aircraft tracking to intercept the Instrument Landing System (ILS) localiser inbound on the approach side, outside the published ILS coverage angle: (1.00 P.)
- [A] only glide path information is available
 - [B] may receive false course indications
 - [C] will receive signals without identification coding
 - [D] can expect signals to give correct indications
- 178 The MIDDLE MARKER of an Instrument Landing System (ILS) facility is identified audibly and visually by a series of: (1.00 P.)
- [A] two dashes per second and a blue light flashing
 - [B] dots and a white light flashing
 - [C] dashes and an amber light flashing
 - [D] alternate dots and dashes and an amber light flashing
- 179 The OUTER MARKER of an Instrument Landing System (ILS) facility transmits on a frequency of: (1.00 P.)
- [A] 200 MHz and is modulated by alternate dot/dash in morse
 - [B] 75 MHz and is modulated by alternate dot/dash in morse
 - [C] 75 MHz and is modulated by morse at two dashes per second
 - [D] 300 MHz and is modulated by morse at two dashes per second

- 180 The two modulations from which bearing information is obtained in a VOR are (1.00 P.)
- [A] reference signal frequency modulated and directional signal apparent amplitude modulated
 - [B] reference signal apparent amplitude modulated and directional signal frequency modulated
 - [C] both frequency modulated at the transmitter
 - [D] none of the options are correct

- 181 What approximate rate of descent is required in order to maintain a 3° glide path at a groundspeed of 120 kt? (1.00 P.)
- [A] 550 FT/MIN
 - [B] 950 FT/MIN
 - [C] 800 FT/MIN
 - [D] 600 FT/MIN

- 182 The outer marker of an ILS with a 3° glide slope is located 4.6 NM from the threshold. Assuming a glide slope height of 50 FT above the threshold, the approximate height of an aircraft passing the outer marker is: (1.00 P.)
- [A] 1350 FT
 - [B] 1300 FT
 - [C] 1450 FT
 - [D] 1400 FT

- 183 Airborne weather radar systems use a wavelength of approximately 3 cm in order to: (1.00 P.)
- [A] obtain optimum use of the Cosecant squared beam
 - [B] transmit at a higher pulse repetition frequency for extended range
 - [C] detect the smaller cloud formations as well as large
 - [D] detect the larger water droplets

184 The VOR frequency range is; (1.00 P.)

[A] 108 - 112 MHz

[B] 108 - 136 MHz

[C] 108 - 118 MHz

[D] 106 - 136 MHz

185 The frequency which corresponds to a wavelength of 12 cm is: (1.00 P.)

- [A] 2500 kHz.
- [B] 3600 MHz.
- [C] 360 MHz.
- [D] 2500 MHz.

186 The frequency which corresponds to a wavelength of 12 cm is: (1.00 P.)

- [A] 360 MHz.
- [B] 2500 kHz.
- [C] 3600 MHz.
- [D] 2500 MHz.

- 187 The ISO-ECHO facility of an airborne weather radar is provided in order to: (1.00 P.)
- [A] detect areas of possible severe turbulence in cloud
 - [B] inhibit unwanted ground returns
 - [C] give an indication of cloud tops
 - [D] extend the mapping range

- 188 The wavelength of a radio signal transmitted at the frequency of 75 MHz is: (1.00 P.)
- [A] 7.5 m.
 - [B] 75 m.
 - [C] 40 m.
 - [D] 4 m.

- 189 In the MAPPING MODE the airborne weather radar utilises a: (1.00 P.)
- [A] pencil beam effective from zero to 150 NM
 - [B] fan shaped beam effective up to a range of 150 NM
 - [C] fan shaped beam effective up to a maximum of 50 NM to 60 NM range
 - [D] pencil beam to a maximum range of 60 NM

- 190 The wavelength of a radio signal transmitted at the frequency 118.7 MHz is: (1.00 P.)
- [A] 25.3 cm.
 - [B] 2.53 m.
 - [C] 25.3 m.
 - [D] 2.53 cm.

- 191 Which of the following cloud types is most readily detected by airborne weather radar when using the 'weather beam'? (1.00 P.)
- [A] cumulus
 - [B] cirrocumulus
 - [C] altostratus
 - [D] stratus

- 192 Why is a secondary radar display screen free of storm clutter? (1.00 P.)
- [A] The principle of 'echo' return is not used in secondary radar
 - [B] The frequencies employed are too high to give returns from moisture sources
 - [C] A moving target indicator facility suppresses the display of static or near static returns
 - [D] The frequencies employed are too low to give returns from moisture sources

193 If the directional signal leads the reference signal by 30° the magnetic bearing to the VOR station will be (1.00 P.)

[A] 30°

[B] 210°

[C] 150°

[D] 330°

- 194 In order to indicate radio failure the aircraft SSR transponder should be selected to code: (1.00 P.)
- [A] 7700
 - [B] 7000
 - [C] 7600
 - [D] 7500
- 195 In order to indicate unlawful interference with the planned operation of the flight, the aircraft Secondary Surveillance Radar (SSR) transponder should be selected to: (1.00 P.)
- [A] 7000
 - [B] 7700
 - [C] 7600
 - [D] 7500

- 196 With an omni-selector of 090° on a VOR indicator and the directional phase lagging the reference phase by 280°, the VOR indicator will show (1.00 P.)
- [A] fly right - FROM
 - [B] fly left - TO
 - [C] fly right -- FROM
 - [D] fly right - TO

- 197 A ground feature appears 30° to the left of the centre line of the CRT of an airborne weather radar. If the heading of the aircraft is 355° (M) and the magnetic variation is 15° East, the true bearing of the aircraft from the feature is: (1.00 P.)
- [A] 130°
 - [B] 160°
 - [C] 220°
 - [D] 310°

- 198 An apparent increase in the transmitted frequency in doppler which is proportional to the transmitter velocity will occur when: (1.00 P.)
- [A] the receiver moves towards the transmitter
 - [B] both transmitter and receiver move towards each other
 - [C] the transmitter moves towards the receiver
 - [D] the transmitter moves away from the receiver

- 199 Which is the highest latitude listed below at which the sun will rise above the horizon and set every day? (1.00 P.)
- [A] 68°
 - [B] 66°
 - [C] 62°
 - [D] 72°

200 060-003.jpg

The UTC of sunrise on 6 December at WINNIPEG (Canada) ($49^{\circ}50'N$ $097^{\circ}30'W$) is: (1.00 P.)

Siehe Anlage 3

[A] 1413

[B] 2230

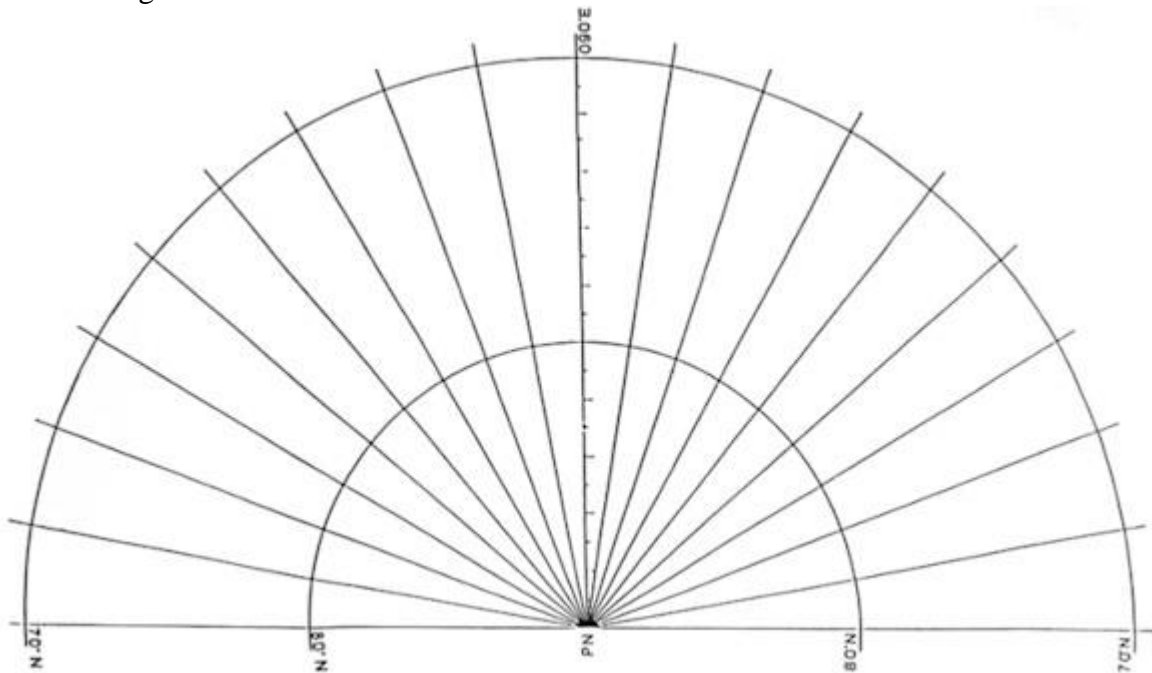
[C] 0113

[D] 0930

Anlagen zu den Aufgaben

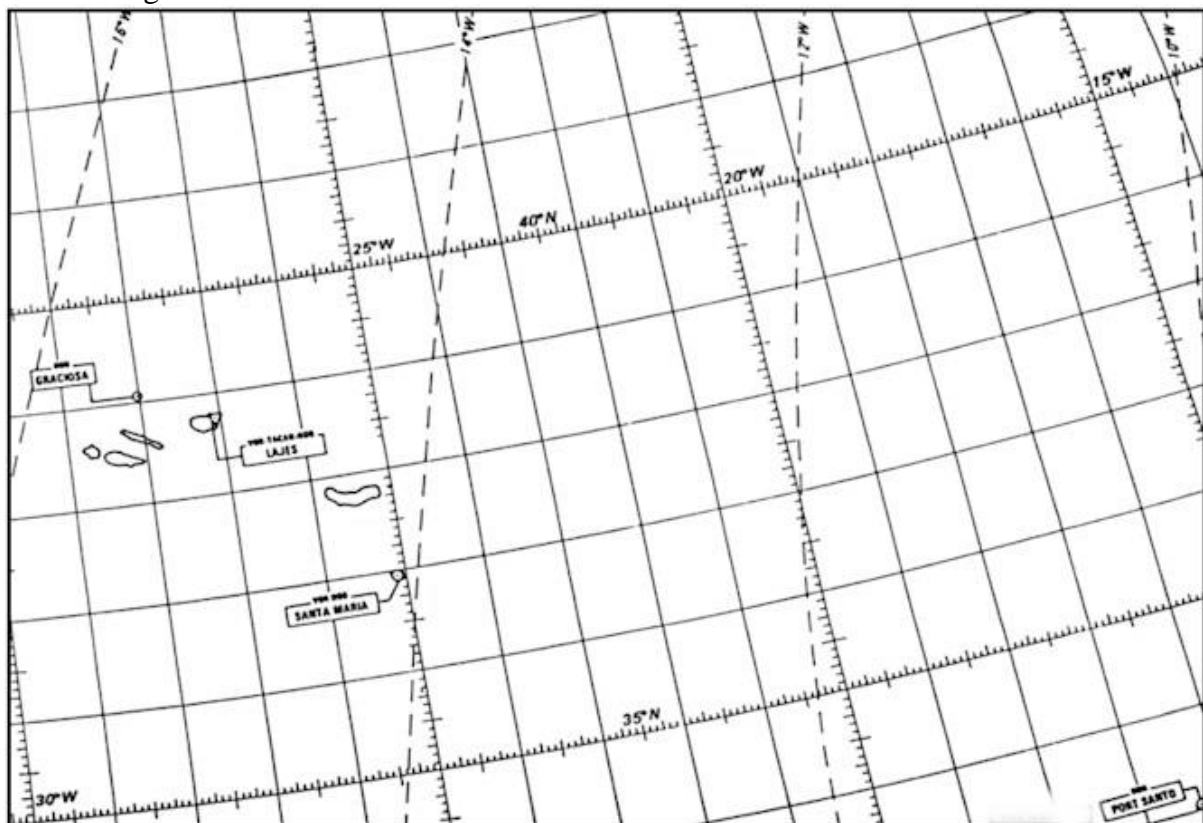
Anlage 1 zu Aufgabe 121

Titel: Anlage 1



Anlage 2 zu Aufgabe 124

Titel: Anlage 1



Anlagen zu den Aufgaben

Anlage 3 zu Aufgabe 200

Titel: Anlage 1

A86

SUNRISE

Lat.	November				December												Jan.
	19	22	25	28	1	4	7	10	13	16	19	22	25	28	31	3	
N 72	h m	h m	h m	h m	h m	h m	h m	h m	h m	h m	h m	h m	h m	h m	h m	h m	
70	10 14	10 40	11 19														
68	09 30	09 45	10 00	10 16	10 33	10 52	11 14										
66	09 01	09 12	09 23	09 34	09 45	09 55	10 05	10 14	10 22	10 28	10 32	10 35	10 35	10 33	10 30	10 25	
64	08 39	08 48	08 57	09 06	09 15	09 23	09 30	09 36	09 42	09 47	09 50	09 52	09 53	09 52	09 51	09 48	
62	21	29	37	08 45	08 52	08 59	09 05	09 10	09 15	09 19	22	24	24	24	23	22	
N 60	08 07	08 14	08 21	08 28	08 34	08 40	08 45	08 50	08 54	08 58	09 00	09 02	09 03	09 03	09 03	09 01	
58	07 54	08 01	08 07	13	19	24	29	33	37	40	08 43	08 45	08 46	08 46	08 46	08 45	
56	43	07 49	07 55	08 01	08 06	08 11	15	19	23	26	28	30	31	32	32	31	
54	34	40	45	07 50	07 55	07 59	08 03	08 07	08 10	13	15	17	19	19	19	19	
52	26	31	36	40	45	49	07 53	07 56	07 59	08 02	08 04	08 06	08 07	08 08	08 08	08 08	
N 50	07 18	07 23	07 28	07 32	07 36	07 40	07 43	07 47	07 50	07 52	07 54	07 56	07 57	07 58	07 59	07 58	
45	07 02	07 06	07 10	07 14	17	21	24	27	29	32	34	35	37	38	38	38	
40	06 49	06 53	06 56	06 59	07 02	07 05	07 08	07 10	07 13	15	17	18	20	21	22	22	
35	38	41	44	46	06 49	06 52	06 54	06 57	06 59	07 01	07 03	07 04	07 06	07 07	07 08	07 08	
30	28	30	33	35	38	40	43	45	47	06 49	06 50	06 52	06 53	06 54	06 55	06 56	
N 20	06 11	06 13	06 15	06 16	06 18	06 20	06 22	06 24	06 26	06 28	06 29	06 31	06 32	06 33	06 35	06 36	
N 10	05 56	05 57	05 58	06 00	06 01	06 03	06 04	06 06	06 07	06 09	06 10	06 12	06 13	06 15	06 16	17	
0	42	42	43	05 44	05 45	05 46	05 48	05 49	05 50	05 52	05 53	05 55	05 56	05 58	05 59	06 01	
S 10	28	28	28	28	29	30	31	32	33	34	35	37	39	40	42	05 43	
20	05 12	05 12	05 11	05 11	05 12	05 12	05 13	05 13	05 14	05 15	05 17	05 18	05 20	05 21	23	25	
S 30	04 54	04 53	04 52	04 52	04 51	04 51	04 51	04 52	04 52	04 53	04 55	04 56	04 58	04 59	05 01	05 03	
35	44	42	41	40	39	39	39	39	40	41	42	43	45	46	04 48	04 51	
40	32	30	28	27	26	25	25	25	25	26	27	28	29	31	34	36	
45	18	04 15	04 13	04 11	04 09	04 08	04 07	04 07	04 07	04 08	04 08	04 10	04 11	04 13	04 16	04 19	
50	04 00	03 57	03 54	03 51	03 49	03 47	03 46	03 45	03 45	03 46	03 47	03 49	03 51	03 54	03 57	03 59	
S 52	03 52	03 48	03 45	03 42	03 39	03 37	03 36	03 35	03 34	03 34	03 35	03 36	03 38	03 40	03 43	03 46	
54	43	39	35	31	29	26	24	23	22	22	23	24	26	28	31	34	
56	32	28	23	19	16	03 13	03 11	03 09	03 09	03 08	03 09	03 10	03 12	03 14	17	21	
58	21	15	03 10	03 06	03 02	02 58	02 55	02 53	02 52	02 52	02 52	02 53	02 55	02 57	03 01	03 05	
S 60	03 06	03 00	02 54	02 49	02 44	02 40	02 37	02 34	02 32	02 31	02 31	02 32	02 34	02 37	02 41	02 45	

SUNSET

Lat.	November				December												Jan.
	19	22	25	28	1	4	7	10	13	16	19	22	25	28	31	3	
N 72	h m	h m	h m	h m	h m	h m	h m	h m	h m	h m	h m	h m	h m	h m	h m	h m	
70	13 15	12 51	12 13	—	—	—	—	—	—	—	—	—	—	—	—	—	
68	14 00	13 47	13 33	13 19	13 04	12 47	12 27	—	—	—	—	—	—	—	—	—	
66	29	14 19	14 10	14 01	13 52	13 44	13 37	13 30	13 26	13 22	13 21	13 21	13 24	13 29	13 35	13 43	
64	14 51	14 43	14 36	29	14 22	14 17	14 12	14 08	14 05	14 04	14 03	14 04	14 06	14 10	14 15	14 21	
62	15 09	15 02	14 56	14 50	14 45	14 41	37	34	32	31	32	33	35	38	14 42	14 47	
N 60	15 23	15 18	15 12	15 08	15 03	15 00	14 57	14 53	14 54	14 53	14 53	14 55	14 57	14 59	15 03	15 07	
58	36	31	26	22	19	16	15 13	15 12	15 11	15 10	15 11	15 12	15 14	15 17	20	24	
56	46	42	38	35	32	29	27	26	25	25	26	27	29	31	34	38	
54	15 56	15 52	48	45	43	41	39	38	38	38	39	40	42	44	47	15 50	
52	16 04	16 01	15 58	15 55	15 53	15 51	50	49	48	49	49	50	51	52	15 55	16 01	
N 50	16 12	16 09	16 06	16 03	16 01	16 00	15 59	15 58	15 58	15 58	15 59	16 01	16 02	16 05	16 07	16 10	
45	28	26	24	22	21	20	16 19	16 19	16 19	16 19	16 20	21	23	25	28	30	
40	41	39	38	37	36	35	35	35	35	36	37	38	40	42	44	16 47	
35	16 53	16 51	16 50	16 49	16 49	16 48	16 48	16 48	16 49	16 50	16 51	16 52	16 54	16 56	16 58	17 00	
30	17 03	17 01	17 01	17 00	17 00	17 00	17 00	17 01	17 01	17 02	17 04	17 05	17 07	17 09	17 10	13	
N 20	17 20	17 19	17 19	17 19	17 19	17 20	17 20	17 21	17 22	17 23	17 25	17 26	17 28	17 29	17 31	17 33	
N 10	35	35	35	36	37	37	38	39	41	42	17 43	17 45	17 46	17 48	17 50	17 51	
0	17 49	17 50	17 51	17 52	17 53	17 54	17 55	17 56	17 58	17 59	18 01	18 02	18 04	18 05	18 07	18 08	
S 10	18 03	18 05	18 06	18 07	18 09	18 10	18 12	18 14	18 15	18 17	18	20	21	23	24	25	
20	19	21	23	25	26	28	30	32	34	36	37	18 39	18 40	18 41	18 43	18 44	
S 30	18 37	18 39	18 42	18 44	18 47	18 49	18 51	18 54	18 56	18 58	18 59	19 01	19 02	19 03	19 04	19 05	
35	47	18 50	18 53	18 56	18 59	19 02	19 04	19 06	19 09	19 11	19 12	14	15	16	17	18	
40	18 59	19 03	19 06	19 09	19 13	16	18	21	23	26	27	29	30	31	32	32	
45	19 14	18	22	25	29	32	36	19 39	19 41	19 44	19 45	19 47	19 48	19 49	19 50	19 50	
50	31	36	41	45	50	19 54	19 57	20 01	20 03	20 06	20 08	20 10	20 11	20 12	20 12	20 11	
S 52	19 40	19 45	19 50	19 55	19 59	20 04	20 07	20 11	20 14	20 17	20 19	20 21	20 22	20 22	20 22	20 22	
54	19 49	19 55	20 00	20 05	20 10	15	19	23	26	29	31	33	34	34	34	34	
56	20 00	20 06	12	18	23	28	33	37	40	20 43	20 45	20 47	20 48	20 49	20 48	20 47	
58	12	19	25	32	38	20 43	20 48	20 53	20 57	21 00	21 02	21 04	21 05	21 05	21 05	21 03	
S 60	20 26	20 34	20 41	20 48	20 55	21 01	21 07	21 12	21 17	21 20	21 23	21 25	21 25	21 25	21 24	21 22	

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62.	A	B	C	D
65.	A	B	C	D
3.	A	B	C	D
6.	A	B	C	D
9.	A	B	C	D
12.	A	B	C	D
15.	A	B	C	D
18.	A	B	C	D
21.	A	B	C	D
24.	A	B	C	D
27.	A	B	C	D
30.	A	B	C	D
33.	A	B	C	D
36.	A	B	C	D
39.	A	B	C	D
42.	A	B	C	D
45.	A	B	C	D
48.	A	B	C	D
51.	A	B	C	D
54.	A	B	C	D
57.	A	B	C	D
60.	A	B	C	D
63.	A	B	C	D
66.	A	B	C	D

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67.	A	B	C	D	68.	A	B	C	D	69.	A	B	C	D
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85.	A	B	C	D	86.	A	B	C	D	87.	A	B	C	D
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94.	A	B	C	D	95.	A	B	C	D	96.	A	B	C	D
97.	A	B	C	D	98.	A	B	C	D	99.	A	B	C	D
100.	A	B	C	D	101.	A	B	C	D	102.	A	B	C	D
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106.	A	B	C	D	107.	A	B	C	D	108.	A	B	C	D
109.	A	B	C	D	110.	A	B	C	D	111.	A	B	C	D
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115.	A	B	C	D	116.	A	B	C	D	117.	A	B	C	D
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121.	A	B	C	D	122.	A	B	C	D	123.	A	B	C	D
124.	A	B	C	D	125.	A	B	C	D	126.	A	B	C	D
127.	A	B	C	D	128.	A	B	C	D	129.	A	B	C	D
130.	A	B	C	D	131.	A	B	C	D	132.	A	B	C	D

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184 .	A	B	C	D
187 .	A	B	C	D
190 .	A	B	C	D
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167 .	A	B	C	D
170 .	A	B	C	D
173 .	A	B	C	D
176 .	A	B	C	D
179 .	A	B	C	D
182 .	A	B	C	D
185 .	A	B	C	D
188 .	A	B	C	D
191 .	A	B	C	D
194 .	A	B	C	D
197 .	A	B	C	D
135 .	A	B	C	D
138 .	A	B	C	D
141 .	A	B	C	D
144 .	A	B	C	D
147 .	A	B	C	D
150 .	A	B	C	D
153 .	A	B	C	D
156 .	A	B	C	D
159 .	A	B	C	D
162 .	A	B	C	D
165 .	A	B	C	D
168 .	A	B	C	D
171 .	A	B	C	D
174 .	A	B	C	D
177 .	A	B	C	D
180 .	A	B	C	D
183 .	A	B	C	D
186 .	A	B	C	D
189 .	A	B	C	D
192 .	A	B	C	D
195 .	A	B	C	D
198 .	A	B	C	D

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199	A	B	C	D		200	A	B	C	D
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LÖSUNGSBOGEN

Prüf.-Nr.:

Prüfungsdatum:

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25.	A			
28.	A			
31.		B		
34.			C	
37.		B		
40.			C	
43.			C	
46.		B		
49.		B		
52.	A			
55.		B		
58.				D
61.	A			
64.			C	
2.		B		
5.	A			
8.			C	
11.		B		
14.				D
17.		B		
20.		B		
23.	A			
26.				D
29.				D
32.	A			
35.	A			
38.				D
41.	A			
44.				D
47.				D
50.			C	
53.		B		
56.				D
59.		B		
62.	A			
65.				D
3.				D
6.		B		
9.		B		
12.	A			
15.	A			
18.	A			
21.			C	
24.		B		
27.			C	
30.	A			
33.	A			
36.			C	
39.			C	
42.	A			
45.			C	
48.		B		
51.		B		
54.				D
57.			C	
60.	A			
63.		B		
66.			C	

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LÖSUNGSBOGEN

Prüf.-Nr.:

Prüfungsdatum:

67.		B		
70.			C	
73.			C	
76.	A			
79.			C	
82.				D
85.	A			
88.		B		
91.			C	
94.		B		
97.				D
100.	A			
103.			C	
106.	A			
109.		B		
112.			C	
115.				D
118.		B		
121.			C	
124.				D
127.			C	
130.				D
68.	A			
71.			C	
74.		B		
77.	A			
80.	A			
83.	A			
86.			C	
89.	A			
92.				D
95.				D
98.		B		
101.			C	
104.			C	
107.	A			
110.		B		
113.			C	
116.				D
119.			C	
122.				D
125.			C	
128.			C	
131.				D
69.			C	
72.		B		
75.				D
78.	A			
81.	A			
84.				D
87.				D
90.				D
93.		B		
96.	A			
99.				D
102.			C	
105.			C	
108.		B		
111.		B		
114.	A			
117.	A			
120.	A			
123.				D
126.		B		
129.				D
132.				D

Nur für den internen Gebrauch

LÖSUNGSBOGEN

Prüf.-Nr.:

Prüfungsdatum:

133	.		B		
136	.		B		
139	.			C	
142	.	A			
145	.				D
148	.		B		
151	.	A			
154	.		B		
157	.			C	
160	.				D
163	.				D
166	.	A			
169	.	A			
172	.			C	
175	.	A			
178	.				D
181	.				D
184	.			C	
187	.	A			
190	.		B		
193	.			C	
196	.				D
134	.		B		
137	.			C	
140	.			C	
143	.	A			
146	.			C	
149	.		B		
152	.	A			
155	.		B		
158	.	A			
161	.	A			
164	.				D
167	.			C	
170	.		B		
173	.			C	
176	.		B		
179	.			C	
182	.			C	
185	.				D
188	.				D
191	.	A			
194	.			C	
197	.		B		
135	.				D
138	.				D
141	.				D
144	.	A			
147	.	A			
150	.				D
153	.				D
156	.		B		
159	.		B		
162	.	A			
165	.	A			
168	.				D
171	.			C	
174	.		B		
177	.		B		
180	.	A			
183	.				D
186	.				D
189	.			C	
192	.	A			
195	.				D
198	.			C	

Nur für den internen Gebrauch	
LÖSUNGSBOGEN	Prüf.-Nr.:
Prüfungsdatum:	

199		B					200	A				
.							.					