## CE 425 HOMEWORK-1 SOLUTIONS

91) Calculate the flexibility coefficients of the given fixed-fixed beam. Take rotations at point A and B as your redundant forces. EI is constant through the length of the beam.

sign convention: < (+ + +)>

$$RB = -RA = -1/L$$

$$M \longrightarrow M_1 = -1 + \times L$$

$$f_{II} = \frac{1}{EI} \int_{0}^{\infty} M_{1}^{2} dx = \frac{1}{EI} \int_{0}^{\infty} \left(\frac{x}{L} - I\right)^{2} dx = \frac{1}{EI} \int_{0}^{\infty} \left(\frac{x^{2}}{L^{2}} - \frac{2x}{L} + I\right) dx$$

$$= \frac{1}{EI} \int_{0}^{\infty} \left(\frac{x^{2}}{L^{2}} - \frac{2x}{L} + I\right) dx$$

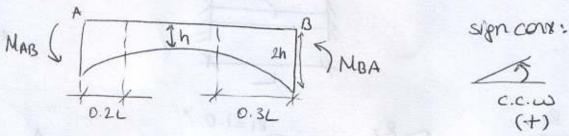
$$=\frac{1}{EI}\left(\frac{x^3}{3L^2}-\frac{2x^2}{2L}+x\right)\Big|_{0}=\frac{L}{3EI}$$

$$f_{22} = \frac{1}{EI} \int_{0}^{L} M_{2}^{2} dx = \frac{1}{EI} \int_{0}^{L} \frac{x^{2}}{L^{2}} dx = \frac{1}{EI} \frac{x^{3}}{3L^{2}} \Big|_{0}^{L} = \frac{L}{3EI}$$

$$f_{12} = f_{21} = \int_{0}^{L} \left(\frac{x}{L} - I\right) \left(\frac{x}{L}\right) dx = \int_{0}^{L} \left(\frac{x^{2}}{L^{2}} - \frac{x}{L}\right) dx = \frac{1}{EI} \left(\frac{x^{3}}{3L^{2}} - \frac{x^{2}}{2L}\right) dx$$

Flexibility Coefficients, 
$$f = \frac{1}{EI} \begin{bmatrix} L/3 & -L/6 \\ -L/6 & L/3 \end{bmatrix}$$

(92) The slope deflections equations for a parabolic haunch beam are given below Derive the element stiffness matrix for the given beam member (Do not consider axial deformators).



$$D_{1}=1$$

$$\sqrt{11.33} \frac{EL}{L^{2}}$$

$$D_{2}=\frac{1}{L}$$

$$D_{2}=\frac{1}{L^{2}}$$

$$D_{3}=\frac{12.24}{L^{2}}$$

$$MAB = 6.73EI (0+0+1.683.1) + FEMAR$$

$$= 11.33EI/L^{2}$$

$$= 11.33 \text{ EI} / L^{2}$$

$$MBA = \frac{7.68 \text{ EI}}{L} (0+0+1.598 \text{ ...} L) + FE/MBA$$

$$= 12.27 \text{ EI} / L^{2}$$

$$VA-L = (11.33+12.27)EI = 23.6EI$$
 $L^2$ 

$$V_{B} = -23.6 EI$$

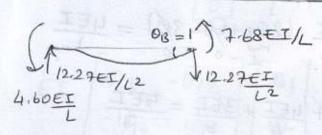
## case 2, D2 = 1, D1 = D3 = D4 = 0

$$k_{12} = 11.32 \frac{EI}{L^2}$$
 $k_{22} = 6.73 \frac{EI}{L}$ 
 $k_{32} = -11.32 \frac{EI}{L^2}$ 
 $k_{42} = 4.60 \frac{EI}{L}$ 

## Case 8, D3=1, D2=D1=D4=0

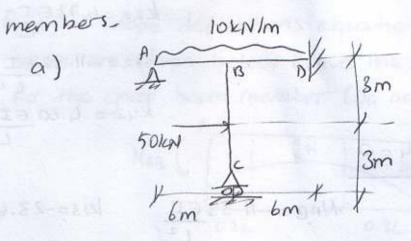
$$|C23 = -11.33 \frac{EI}{L^2}$$

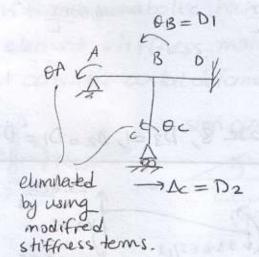
Case 4, 
$$D_4 = 1$$
,  $D_1 = D_2 = D_3 = 0$ 



$$\mathcal{K} = EI \begin{bmatrix} 23.6/L^3 & |1.32/L^2 & -23.6/L^3 & |228|_{L^2} \\ 6.73/L & -|1.33/L^2 & |4.60/L \\ \\ & & & & & & & & \\ & & & & & & \\ & & & & & \\ & & & & & & \\ & & & & & & \\ & & & & & & \\ & & & & & \\ & & & & & & \\ & & & & & & \\ & & & & & & \\ & & & & & & \\ & & & & & & \\ & & & & & & \\ & & & & & & \\ & & & & & & \\ & & & & & & \\ & & & & & \\ & & & & & & \\ & & & & & & \\ & & & & & & \\ & & & & & \\ & & & & & & \\ & & & & & & \\ & & & & & & \\ & & & & & & \\ & & & & & & \\ & & & & & & \\ & & & & & & \\ & & & & & & \\ & & & & & & \\ & & & & & \\ & & & & & & \\ & & & & & & \\ & & & & & \\ & & & & & \\ &$$

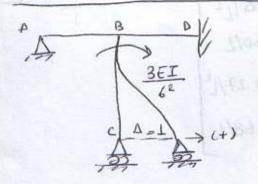
93) Analyze the given structures using general stiffness methor Calculate the support reactions and draw moment dragon Assume axial rigidity. EI is constant and some for all





2.D.O.F;

$$k_{11} = \frac{3EI}{6} + \frac{4EI}{6} + \frac{3EI}{6} = \frac{5EI}{3}$$
 $k_{21} = \left(\frac{3EI}{6}\right) \cdot \frac{1}{6} = -\frac{3EI}{36}$ 



$$MBC = \frac{3EI}{L} (\ThetaB - \cancel{D})$$

$$= -\frac{3EI}{6^2} = k2I$$

$$Vc = \frac{3EI}{63} = k_{22}$$

$$K = EI \begin{bmatrix} 5/3 & -3/36 \\ -3/36 & 3/216 \end{bmatrix} \qquad QA = \begin{bmatrix} 0 \\ 0 \end{bmatrix}, \text{ no directly opplied forces or moments}$$

$$To find QF;$$

$$lokn(my)$$

$$\frac{10 \text{ find } 9F}{4}$$

$$\frac{10 \text{ kN/m}}{8}$$

$$\frac{\omega L^2}{8} = \frac{10.6^2}{8}$$

$$\frac{10.62}{12}$$

$$\frac{WL^2}{8} = \frac{10.6^2}{8}$$

$$\left( \frac{1}{12} \right) \frac{10.6^2}{12}$$

combining, 
$$\frac{1}{50 \text{kN}} = \frac{1}{50.3.3(9)} = \frac{1}{50.3.3(9)} = \frac{1}{50.25 \text{km}}$$

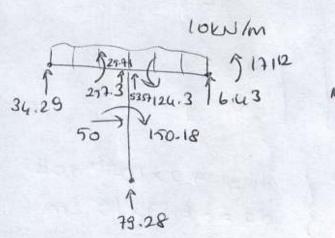
$$Q = QA - QF = \begin{bmatrix} 0 \\ 0 \end{bmatrix} - \begin{bmatrix} -71.25 \\ -15.63 \end{bmatrix} = \begin{bmatrix} 71.25 \\ 15.63 \end{bmatrix}$$

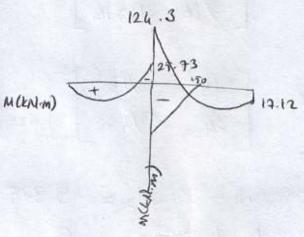
EI. 
$$\begin{bmatrix} 5/3 & -1/12 \\ -1/12 & 1/72 \end{bmatrix} \begin{bmatrix} D_1 \\ D_2 \end{bmatrix} = \begin{bmatrix} 71.25 \\ 15.63 \end{bmatrix} \Rightarrow D_1 = 141.45/EI$$

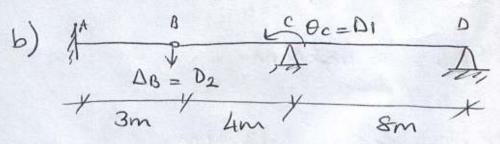
$$D_2 = 1974.09/EI$$

Considering FEM forces, back substitution gives moments offorces,

YC = 10.12 - 34.29-6.43 = 79.28KM



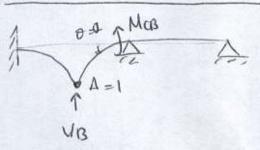




## case 1, D1=1, D2=0

MCB = 
$$\frac{3EI}{L}$$
 =  $\frac{3EI}{4}$  |  $\frac{3EI}{4}$  |  $\frac{3EI}{4}$  |  $\frac{3EI}{8}$  |  $\frac{3EI}{4}$  |  $\frac{3EI}{8}$  |  $\frac{3EI}{16}$  |  $\frac{3EI$ 

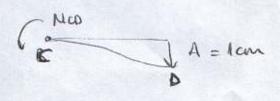
case 2, 
$$D_2 = 1$$
,  $D_1 = 0$ 



$$McB = \frac{3EI}{L} (\theta c - \frac{\Delta}{L})$$

$$= \frac{-3EI}{16} = k21$$

$$VB = \frac{3EI}{16} \cdot \frac{1}{4} + \frac{EI}{3} \cdot \frac{1}{3} = \frac{91EI}{576} = k21$$



$$Mco = \frac{3EI}{8} \cdot \left( + \frac{0.01}{8} \right) = \frac{3EI}{6000}$$

$$= \frac{3.20.000}{6000}$$

$$= 9.345 \text{ kN·m}$$

$$K = EI \begin{bmatrix} 9/8 & -3/16 \\ -3/16 & 91/576 \end{bmatrix}$$
  $A - AF = \begin{bmatrix} -9.375 \\ 0 \end{bmatrix}$ 

$$20.000 \begin{bmatrix} 9/8 & -3/16 \\ -3/16 & 91/576 \end{bmatrix} \begin{bmatrix} d_1 \\ d_2 \end{bmatrix} = \begin{bmatrix} -9.375 \\ 0 \end{bmatrix}$$

$$22700d_1 - 3750d_2 = -9.375$$

$$-3750d_1 + 3160d_2 = 0$$

$$d_1 = -5.2 \times 10^{-4} \text{ rad}$$

$$d_2 = -6.17 \times 10^{-4} \text{ m}$$

$$-3750 d1 + 3160 d2 = 0 } d2$$

$$MCB = \frac{3EI}{L} (\theta c - \frac{\Delta}{L}) = -9.48 kN.m$$

