CE 382 Reinforced Concrete Fundamentals

Pure Bending: Analysis of RC Sections

Given:

- $b_w = 300 \text{ mm}, d = 450 \text{ mm}, d' = 30 \text{ mm}$
- ► C16 → $f_{cd} = 11 \text{ MPa}$ & S420 → $f_{vd} = 365 \text{ MPa}$
- $A_s = 1580 \text{ mm}^2$ & $A'_s = 520 \text{ mm}^2$
- Find: Ultimate Moment M_r
- Assume both tension & compression steel have yielded

$$\varepsilon_{sy} = \frac{^{365}}{^{200000}} = 0.001825$$

$$k_1 c = \frac{(A_s - A_s') f_{yd}}{0.85 f_{cd} b_w} = \frac{(1580 - 520)365}{0.85 \times 11 \times 300} = 138 \text{ mm}$$

$$c = \frac{138}{0.85} = 162.3 \text{ mm}$$

$$\varepsilon_s' = 0.003 \frac{c - d'}{c} = 0.003 \frac{162.3 - 30}{162.3} = 0.00244 > \varepsilon_{sy}$$

$$\varepsilon_s = 0.003 \frac{d-c}{c} = 0.003 \frac{450-162.3}{162.3} = 0.00532 > \varepsilon_{sy} \checkmark$$

$$M_r = 0.85 f_{cd} b_w k_1 c \left(d - \frac{k_1 c}{2} \right) + A'_s f_{yd} (d - d')$$

$$= 0.85 \times 11 \times 300 \times 138 \left(450 - \frac{138}{2} \right) + 520 \times 365 (450 - 30)$$

$$= 227 \text{ kNm}$$

- same as Example 4 but $A'_s = 1200 \text{ mm}^2$
- Tension steel has surely yielded
- Assume compression steel has yielded

$$c = \frac{(1580 - 1200)365}{0.85 \times 0.85 \times 11 \times 300} = 58.4 \text{ mm}$$

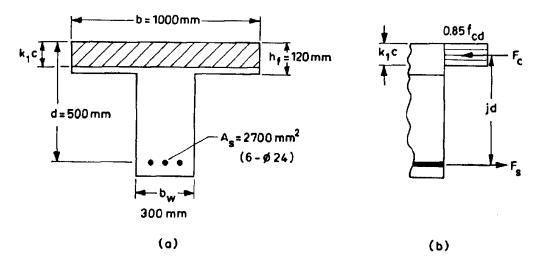
- > $\varepsilon_s' = \frac{58.4 30}{58.4} = 0.00146 < \varepsilon_{sy}$ × compression steel has not yielded
- Use general solution

$$\sigma_S' = \varepsilon_S' E_S \quad \& \quad \varepsilon_S' = 0.003 \frac{c - d'}{c} \quad \Rightarrow \quad \sigma_S' = 0.003 E_S \frac{c - d'}{c}$$

$$\sigma_s' = 600 \frac{c-30}{c}$$

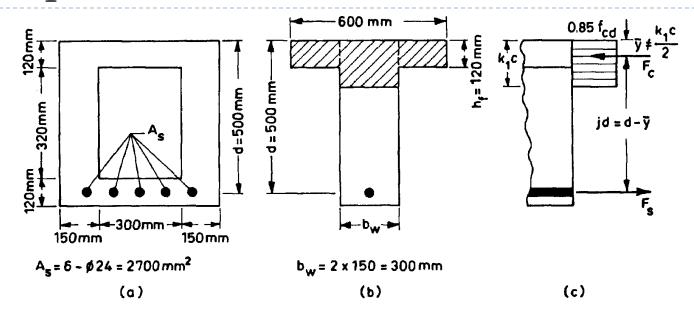
- Force equilibrium:
- $A_{s}f_{yd} 0.85f_{cd}b_{w}k_{1}c A'_{s}\sigma'_{s} = 0$ $1580 \times 365 0.85 \times 11 \times 300 \times 0.85c 1200 \times 600 \frac{c 30}{c} = 0$
- $c^2 + 60c 9060 = 0$
- c = 70 mm
- $\varepsilon_s' = 0.003 \frac{70-30}{70} = 0.00171$ \Rightarrow $\sigma_s' = 342 \text{ Mpa}$
- $M_r = 0.85 f_{cd} b_w k_1 c \left(d \frac{k_1 c}{2} \right) + A_s' \sigma_s' (d d')$
- $M_r = 243 \text{ kNm}$

Given:



- $b = 1000 \text{ mm}, \ b_w = 300 \text{ mm}, \ t = h_f = 120 \text{ mm}$
- $d = 500 \text{ mm}, A_s = 6024 = 2700 \text{ mm}^2$
- ► C20 → $f_{cd} = 13 \text{ MPa}$ S420 → $f_{yd} = 365 \text{ Mpa}$
- $M_r = ?$
- Assume $k_1c = t = 120 \text{ mm}$ & steel yielded

- $F_c = 0.85 f_{cd} b k_1 c = 0.85 \times 13 \times 1000 \times 120 = 1326 \text{ kN}$
- $F_s = A_s f_{vd} = 2700 \times 365 = 985 \text{ kN}$
- $F_c > F_s \implies k_1 c < t$ analyze as a rectangular section
- $k_1 c = \frac{A_s f_{yd}}{0.85 f_{cd} b} = \frac{2700 \times 365}{0.85 \times 13 \times 1000} = 89 \text{ mm}$
- $\varepsilon_s = 0.003 \frac{d-c}{c} = 0.003 \frac{500-89/0.85}{89/0.85} = 0.01133 > \varepsilon_{yd} = 0.001825$
- $M_r = A_s f_{yd} jd = 2700 \times 365 \times \left(500 \frac{89}{2}\right) = 449 \text{ kNm}$



- Convert the box section into a T-section
- Check if $k_1 c < t$

$$F_c = 0.85 \times 13 \times 600 \times 120 = 796 \text{ kN}$$

$$F_S = 2700 \times 365 = 985 \text{ kN}$$

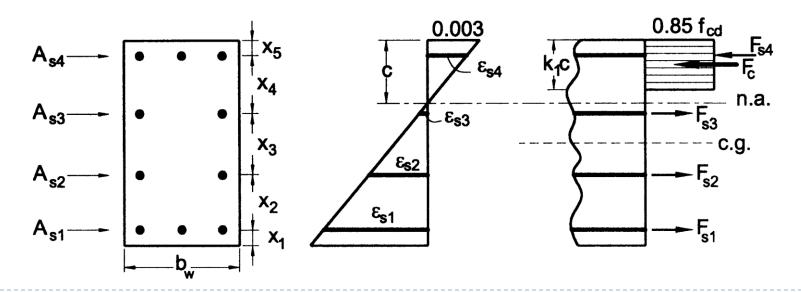
$$F_c < F_s$$

$$\to k_1 c > t$$

- $F_c = 0.85 \times 13[300 \times k_1 c + (600 300)120]$
- $= 3315k_1c + 397800$
- $F_s = 985500 \text{ N}$
- $F_c = F_s \Rightarrow k_1 c = 178 \text{ mm} \& c = 209 \text{ mm}$
- ► Cetroid: $\bar{x} = \frac{300 \times 178 \times \frac{178}{2} + 300 \times 120 \times \frac{120}{2}}{300 \times 178 + 300 \times 120} = 77 \text{ mm}$
- $ightarrow jd = d \bar{x} = 500 77 = 423 \text{ mm}$
- $M_r = A_s f_{vd} jd = 2700 \times 365 \times 423 = 417 \text{ kNm}$

Beams with several layers of steel

- Neutral axis depth c is unknown
- Which steel layer is under compression?
- Which steel layer is under tension?
- Steel layers under tension yielded?
- Steel layers under compression yielded?



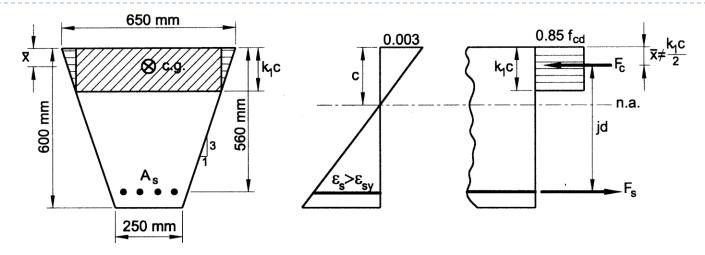
Beams with several layers of steel

Trial and error approach

- \triangleright Assume c
- lacktriangle Compatibility; from similar triangles compute steel strains $arepsilon_{si}$
- Compute $F_{si} = A_{si}\sigma_{si}$ $\sigma_{si} = \varepsilon_{si}E_s \le f_{yd}$
- Compute $F_c = 0.85 f_{cd} b_w k_1 c$
- Check $\sum F = 0$
 - ▶ if $\sum F \le 1 2\%$ \sum compressive forces → no further iteration
- Change c & repeat steps until equilibrium is established
 - ▶ \sum tension > \sum compression → increase c
 - ▶ \sum tension < \sum compression → decrease c
- Compute moment of forces about a convenient point (usually centroid)

Beams with Non-rectangular Cross-Section

- ▶ Trial & error procedure can be used
- When cross-section can be divided into rectangles and/or triangles, a closed solution can be possible



- $A_s = 1590 \text{ mm}^2$, C20 ($f_{cd} = 13 \text{ MPa}$), S420 ($f_{yd} = 365 \text{ MPa}$)
- $M_r = ?$
- ▶ Shaded area:
- $A_{cc} = \left(650 2\frac{1}{3}k_1c\right)k_1c + 2\frac{1}{2}k_1c\frac{1}{3}k_1c$
- $A_{cc} = 650k_1c \frac{(k_1c)^2}{3}$

$$F_c = 0.85 f_{cd} A_{cc} = 0.85 \times 13 \times \left(650 k_1 c - \frac{(k_1 c)^2}{3}\right)$$

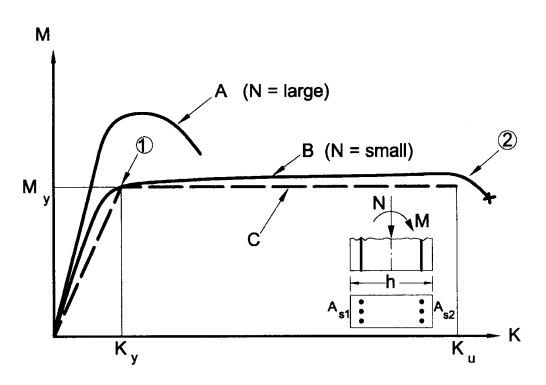
- $F_c = 7182.5k_1c 3.68(k_1c)^2$
- $F_s = A_s f_{vd} = 1590 \times 365 = 580350 \text{ N}$

$$F_c = F_s \quad \Rightarrow \quad (k_1 c)^2 - 1952k_1 c + 159000 = 0$$

- $k_1c = 85 \text{ mm}$
- $\text{Centroid:} \frac{\left(650 \frac{2}{3}85\right)85 \times \frac{85}{2} + 2\frac{1}{2}85\frac{85}{3}\frac{1}{3}85}{\left(650 \frac{2}{3}85\right)85 + 2\frac{1}{2}85\frac{85}{3}} = 42 \text{ mm}$
- $ightarrow jd = d \bar{x} = 518 \text{ mm}$
- $M_r = A_s f_{vd} jd = 300.6 \text{ kNm}$

Section Response: Moment – Curvature

- Provides Full Range Behavior
- Multiple steel layers, axial force+moment, complicated section geometry requires use of computers.



- $M \mathcal{K}$ curve is nonlinear
- curve changes significantly with the level of axial load
- After yielding of tension steel, $\mathcal{K}_{\mathcal{Y}}$, curvature increases without any increase in the moment
 - → Plastic Hinge

Moment – Curvature relationship

- Classical hinge: section rotates under zero moment
- Plastic hinge: rotation takes place under a constant moment
- Ductility: the capability of undergoing large deformations without a significant reduction in the strength.
- curvature ductility ratio = $\frac{\mathcal{K}_u}{\mathcal{K}_v}$
- displacement ductility

related to cross-sectional properties & σ - ϵ of materials

related to member properties

















