

Class 1

Lecture 1

Fundamental Concepts

Terminology

Block Diagrams

State Space Representation of a Linear System

Lecture 2

Linearization

Mechanical Systems & Examples

Lecture 1

1.0

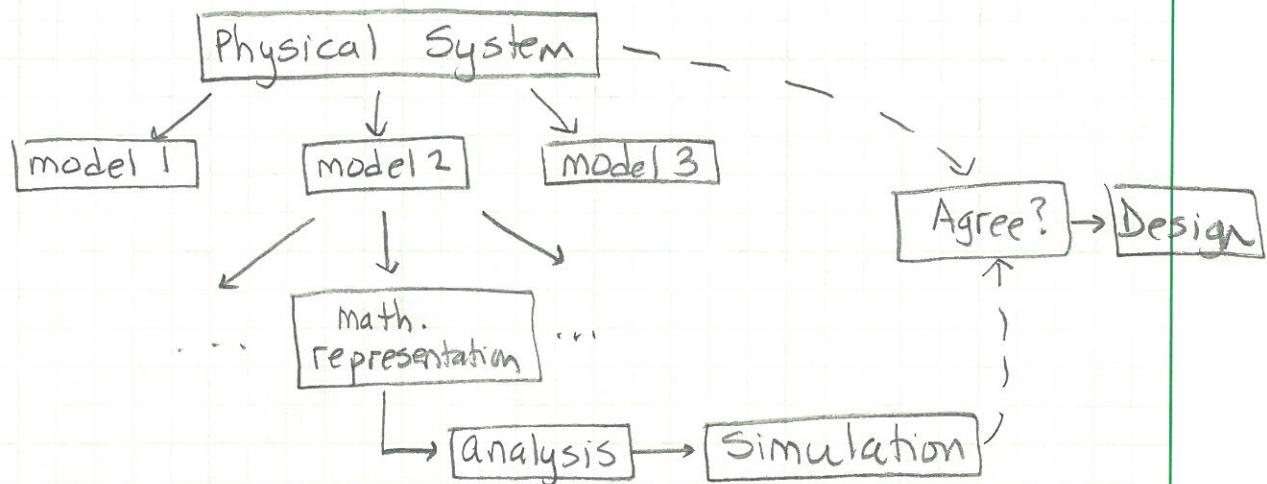
Outline

Fundamental Concepts

Linear Systems - continuous & discrete time

Terminology / Definitions

Block Diagrams & Interconnections

Fundamental Concepts

Physical System: e.g., satellite, robot, vehicle, circuit, airplane

Model: e.g., model satellite as a particle, rigid body or elastic body

★ Selecting a model is a difficult but important step!
- close to physical system but simple enough to analyze analytically

Mathematical Representation (of the model):

e.g., use KVL, KCL, Newton's Laws, Euler-Lagrange Equations to write equations.

Analysis: quantitative - response of system to certain inputs
vs $u \rightarrow \boxed{\quad} \rightarrow y$

qualitative - general properties such as
controllability
observability
stability
robustness

The Study of Systems: consists of four parts

Modeling (we assume these are given)
Setting up mathematical equations
Analysis
Design

Definition: Linear Systems Theory is the study of the properties, capabilities & limitations of representations of linear models of physical systems.

Example: RLC Circuit consumes but doesn't produce energy R, L, C

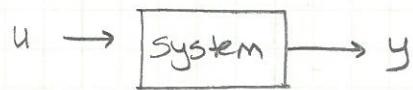
Model #1: Model passive elements as LTI elements

Model #2: take into account nonlinearities of elements.

⇒ Each gives own state equation for the system

Linear Systems (Continuous Time)

Every linear system can be described as the input-output



$$y(t) = \int_0^t G(t, \tau) u(\tau) d\tau$$

will see later
 G is impulse response
matrix of system

If system has "finite" number of state variables, then

LTI

$$\begin{cases} \text{state equation} & \dot{x}(t) = A(t)x(t) + B(t)u(t) & x \in \mathbb{R}^n, u \in \mathbb{R}^k \\ \text{output equation} & y(t) = C(t)x(t) + D(t)u(t) & y \in \mathbb{R}^m \end{cases}$$

$u: [0, \infty) \rightarrow \mathbb{R}^k$ input
 $x: [0, \infty) \rightarrow \mathbb{R}^n$ state
 $y: [0, \infty) \rightarrow \mathbb{R}^m$ output

These equations are referred to as "State-Space"

If system is Time-Invariant,

LTI

$$\begin{cases} y(t) = \int_0^t G(t-\tau)u(\tau)d\tau \\ \dot{x}(t) = Ax(t) + Bu(t) \\ y(t) = Cx(t) + Du(t) \end{cases}$$

time domain shorthand $\left\{ \begin{array}{l} \dot{x} = Ax + Bu \\ y = Cx + Du \end{array} \right.$

Recalling Laplace Transform, we also get

$$y(s) = \hat{G}(s)\hat{u}(s) \quad \left. \begin{array}{l} \text{frequency} \\ \text{domain} \end{array} \right\}$$

will see later
 $\hat{G}(s)$ is the transfer function

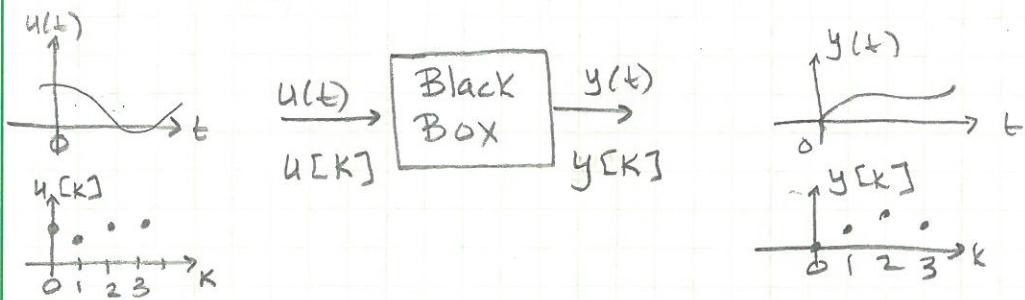
Linear Systems (Discrete Time)

$$x(t+1) = A(t)x(t) + B(t)u(t) \quad x \in \mathbb{R}^n, u \in \mathbb{R}^k$$

$$y(t) = C(t)x(t) + D(t)u(t) \quad y \in \mathbb{R}^m$$

Domain of signals on $\mathbb{N} = \{0, 1, 2, \dots\}$

Shorthand: $\left\{ \begin{array}{l} x^{+} = Ax + Bu \\ y = Cx + Du \end{array} \right.$

Terminology

input \longrightarrow unique response = output

SISO - single-input, single-output

MIMO - multi-input, multi-output

SIMO - single-input, multi-output

Continuous Time system: $u(t) \rightarrow y(t)$ $t \in \mathbb{R}$

Discrete Time system: $u[k] \rightarrow y[k]$ k Integer

Memoryless system: $y(t_0)$ depends only on $u(t_0)$

Independent of input applied before or after t_0

System with Memory: $y(t_0)$ depends on $u(t)$ for

$t < t_0$, $t = t_0$ and $t > t_0$
i.e., past current future inputs

Causal System: $y(t_0)$ depends on $u(t)$ for

$t < t_0$ & $t = t_0$ ONLY
past current

* No physical system can be non-causal, i.e., predict future inputs \rightarrow causality is necessary condition in order to build/implement system in real world

Difficulty: $u(t)$ for $t = -\infty$ to $t = t_0$ affects $y(t_0)$

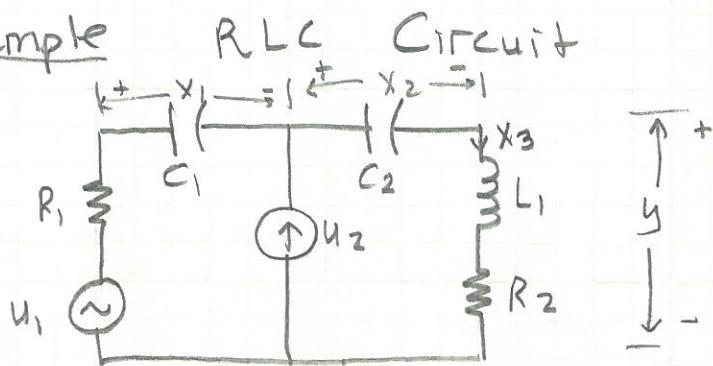
BUT - tracking $u(t)$ from $-\infty$ to t_0 is hard if not impossible

Solution: Concept of a state

Terminology (continued)

Definition: The state $x(t_0)$ of a system at time t_0 is the information at t_0 , that together with the input $u(t)$ for $t \geq t_0$, determines uniquely the output $y(t)$ for $t \geq t_0$.

- \Rightarrow if we know $x(t_0)$, no need to know $u(t)$ applied before t_0 to determine $y(t)$ after t_0 ,
- \Rightarrow state summarizes effect of past inputs on future outputs.

Example

$$x(t) = \begin{bmatrix} x_1(t) \\ x_2(t) \\ x_3(t) \end{bmatrix} \quad \text{state variables}$$

If we know
 $x_1(t_0)$ voltage
 $x_2(t_0)$ voltage
 $x_3(t_0)$ current

Then $+ u(t) t \geq t_0$
we can uniquely determine
 $y(t) t \geq t_0$

Initial state = initial conditions

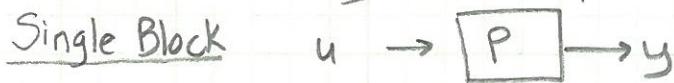
$$\left. \begin{array}{l} x(t_0) \\ u(t), t \geq t_0 \end{array} \right\} \rightarrow y(t) t \geq t_0$$

Lumped System: # state variables is finite

Distributed System: ∞ number of state variables
e.g., transmission line

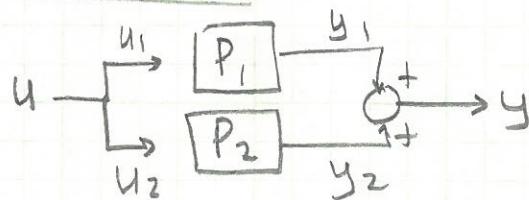
Block Diagrams & Interconnections

Compact way to represent system



Let $P_1 : \begin{cases} \dot{x}_1 = A_1 x_1 + B_1 u_1 & x \in \mathbb{R}^{n_1}, u \in \mathbb{R}^{k_1} \\ y_1 = C_1 x_1 + D_1 u_1 & y \in \mathbb{R}^{m_1} \end{cases}$
 LTI Systems $P_2 : \begin{cases} \dot{x}_2 = A_2 x_2 + B_2 u_2 & x \in \mathbb{R}^{n_2}, u \in \mathbb{R}^{k_2} \\ y_2 = C_2 x_2 + D_2 u_2 & y \in \mathbb{R}^{m_2} \end{cases}$

Parallel Interconnection



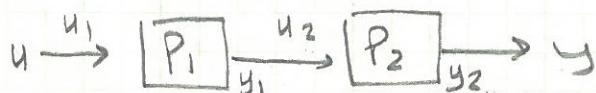
$$u = u_1 = u_2$$

$$y = y_1 + y_2$$

$$\dot{x} = \begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} A_1 & 0 \\ 0 & A_2 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} B_1 \\ B_2 \end{bmatrix} u \quad x \in \mathbb{R}^{n_1+n_2}$$

$$y = [C_1 \ C_2] \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + (D_1 + D_2) u$$

Cascade Interconnection



$$\dot{x} = \begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} A_1 & 0 \\ B_2 C_1 & A_2 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} B_1 \\ B_2 D_1 \end{bmatrix} u \quad x \in \mathbb{R}^{n_1+n_2}$$

$$y = [D_2 C_1 \ C_2] \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + D_2 D_1 u$$

ICP Derive This (5 minutes)

Solution: $y = y_2 = C_2 x_2 + D_2 u_2$

$$u_2 = y_1 = C_1 x_1 + D_1 u_1$$

$$\begin{aligned} \Rightarrow y &= C_2 x_2 + D_2 (C_1 x_1 + D_1 u_1) = D_2 C_1 x_1 + C_2 x_2 + D_2 D_1 u_1 \\ &= [D_2 C_1 \ C_2] \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + D_2 D_1 u_1 \end{aligned}$$

$$u = u_1 \Rightarrow y = [D_2 C_1 \ C_2] \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + D_2 D_1 u \quad \checkmark$$

Interconnections (continued)

ICP (continued)

$$\dot{x}_1 = A_1 x_1 + B_1 u_1$$

$$u_1 = u \Rightarrow \dot{x}_1 = A_1 x_1 + B_1 u \quad \checkmark$$

$$\dot{x}_2 = A_2 x_2 + B_2 u_2$$

$$u_2 = y_1 = C_1 x_1 + D_1 u_1$$

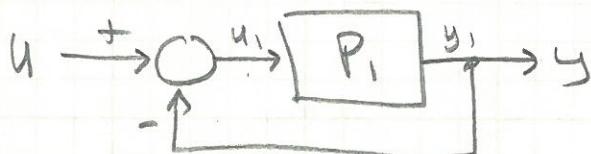
$$\Rightarrow \dot{x}_2 = A_2 x_2 + B_2 (C_1 x_1 + D_1 u_1)$$

$$= B_2 C_1 x_1 + A_2 x_2 + B_2 D_1 u_1$$

$$u_1 = u \Rightarrow \dot{x}_2 = B_2 C_1 x_1 + A_2 x_2 + B_2 D_1 u \quad \checkmark$$

$$\Rightarrow \begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} A_1 & 0 \\ B_2 C_1 & A_2 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} B_1 \\ B_2 D_1 \end{bmatrix} u \quad \checkmark$$

Negative Feedback Interconnection



$$u_1 = u - y_1$$

$$y = y_1$$

(ICP)

Derive LTI Equations (10 min)

Solution:

$$y = y_1 = C_1 x_1 + D_1 u_1$$

$$u_1 = u - y_1 = u - y$$

$$\Rightarrow y = C_1 x_1 + D_1 (u - y) = C_1 x_1 + D_1 u - D_1 y$$

$$y + D_1 y = C_1 x_1 + D_1 u$$

$$(I + D_1) y = C_1 x_1 + D_1 u$$

$$y = (I + D_1)^{-1} C_1 x_1 + (I + D_1)^{-1} D_1 u \quad \checkmark$$

Interconnections (continued)

ICP (continued)

$$\dot{x}_1 = A_1 x_1 + B_1 u_1$$

$$u_1 = u - y_1 = u - y \Rightarrow$$

$$\Rightarrow \dot{x}_1 = A_1 x_1 + B_1 (u - y)$$

$$= A_1 x_1 + B_1 u - B_1 y$$

$$y = (I + D_1)^{-1} C_1 x_1 + (I + D_1)^{-1} D_1 u$$

$$\Rightarrow \dot{x}_1 = A_1 x_1 + B_1 u - B_1 (I + D_1)^{-1} C_1 x_1 - B_1 (I + D_1)^{-1} D_1 u$$

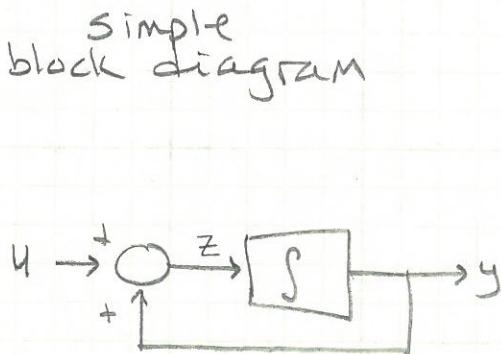
$$= (A_1 - B_1 (I + D_1)^{-1} C_1) x_1 + B_1 (I - (I + D_1)^{-1} D_1) u$$

System Decomposition

LTI complex equation \rightarrow simple block diagram

Example Integrator

$$\begin{cases} \dot{x} = x + u \\ y = x \end{cases}$$



$$\text{input: } z = \dot{x} = x + u = y + u$$

Example LTI

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} 1 & 1 \\ 0 & 3 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 1 \\ 5 \end{bmatrix} u \Rightarrow P_3: \begin{cases} \dot{x}_2 = 3x_2 + 5u \\ y_2 = x_2 \end{cases}$$

$$y = [1 \ 0] \begin{bmatrix} x_1 \\ x_2 \end{bmatrix}$$

$$P_4: \begin{cases} \dot{x}_1 = x_1 + x_2 + u = x_1 + z \\ y_1 = y_1 = x_1 \end{cases}$$

Represent as 2 cascade systems



$$\text{Let } y_2 = x_2$$

$$z = y_2 + u = x_2 + u$$

$$\Rightarrow P_3: u \rightarrow y_2$$

$$P_4: z \rightarrow y$$

two
systems

Lecture 2

Outline

Linearization

Example: Nonlinear system \rightarrow LTI System

Mechanical Systems

Examples

Goal : Write mathematical equation

Linearize and/or approximations

State-space form (LTI System)

Laws of Motion

Linearization

Nonlinear system: $\begin{cases} \dot{x}(t) = f(x(t), u(t)) \\ y(t) = g(x(t), u(t)) \end{cases}$

Some NL systems can be approximated by linear equations

Linearization around an Equilibrium Point

Define: $(x^{eq}, u^{eq}) \in \mathbb{R}^n \times \mathbb{R}^k$ is an equilibrium point

of the NL system above if $f(x^{eq}, u^{eq}) = 0$

i.e., $u(t) = u^{eq}$

$x(t) = x^{eq}$

$y(t) = y^{eq} = g(x^{eq}, u^{eq})$

} Solution to NL system equations

Suppose we perturb the input and initial conditions:

$$x(0) = x^{eq} + \delta x^{eq}$$

$$u(t) = u^{eq} + \delta u(t)$$

where $(\delta x^{eq})^i$ and $(\delta u(t))^i$ $i > 0$ are very small compared to δx^{eq} and $\delta u(t)$.

How much are $x(t)$ and $y(t)$ perturbed by $\delta u(\cdot)$ & δx^{eq} ?

$$\text{Let } \delta x(t) = x(t) - x^{eq}$$

$$\delta y(t) = y(t) - y^{eq}$$

Then

$$\delta \dot{x} = \dot{x} = f(x, u) = f(x^{eq} + \delta x, u^{eq} + \delta u)$$

$$\delta y = g(x, u) - y^{eq} = g(x^{eq} + \delta x, u^{eq} + \delta u) - g(x^{eq}, u^{eq})$$

Expand $f(\cdot)$ and $g(\cdot)$ as Taylor Series around (x^{eq}, u^{eq})

$$\delta \dot{x} = f(x^{eq}, u^{eq}) + \left. \frac{\partial f}{\partial x} \right|_{x^{eq}, u^{eq}} \delta x + \left. \frac{\partial f}{\partial u} \right|_{x^{eq}, u^{eq}} \delta u + \text{H.O.T.}$$

$$\delta y = g(x^{eq}, u^{eq}) + \left. \frac{\partial g}{\partial x} \right|_{x^{eq}, u^{eq}} \delta x + \left. \frac{\partial g}{\partial u} \right|_{x^{eq}, u^{eq}} \delta u + \text{H.O.T.}$$

Linearization (continued)

where $\left. \frac{\partial f}{\partial x} \right|_{x^{eq}, u^{eq}} = \left[\left(\frac{\partial f_i(x, u)}{\partial x_j} \right)_{ij} \right]_{x^{eq}, u^{eq}} \in \mathbb{R}^{n \times n}$ "Jacobian"
 $n \times n$ Matrix

$$\left. \frac{\partial f}{\partial u} \right|_{x^{eq}, u^{eq}} = \left[\left(\frac{\partial f_i(x, u)}{\partial u_j} \right)_{ij} \right]_{x^{eq}, u^{eq}} \in \mathbb{R}^{n \times k}$$

expanding,

$$\left. \frac{\partial f}{\partial x} \right|_{x^{eq}, u^{eq}} = \begin{bmatrix} \frac{\partial f_1}{\partial x_1} & \frac{\partial f_1}{\partial x_2} & \dots & \frac{\partial f_1}{\partial x_n} \\ \vdots & & & \\ \frac{\partial f_n}{\partial x_1} & \frac{\partial f_n}{\partial x_2} & \dots & \frac{\partial f_n}{\partial x_n} \end{bmatrix}_{x^{eq}, u^{eq}} = A(t) \in \mathbb{R}^{n \times n}$$

$$\left. \frac{\partial f}{\partial u} \right|_{x^{eq}, u^{eq}} = \begin{bmatrix} \frac{\partial f_1}{\partial u_1} & \frac{\partial f_1}{\partial u_2} & \dots & \frac{\partial f_1}{\partial u_n} \\ \vdots & & & \\ \frac{\partial f_n}{\partial u_1} & \frac{\partial f_n}{\partial u_2} & \dots & \frac{\partial f_n}{\partial u_n} \end{bmatrix}_{x^{eq}, u^{eq}} = B(t) \in \mathbb{R}^{n \times k}$$

The linearized equations around the equilibrium point (x^{eq}, u^{eq}) are then

$$\dot{x}(t) = A(t) x(t) + B(t) u(t)$$

and similarly,

$$\dot{y}(t) = C(t) x(t) + D(t) u(t)$$

} Linearized Equations

* See p. 14 text for Linearization around a trajectory

- Similar process

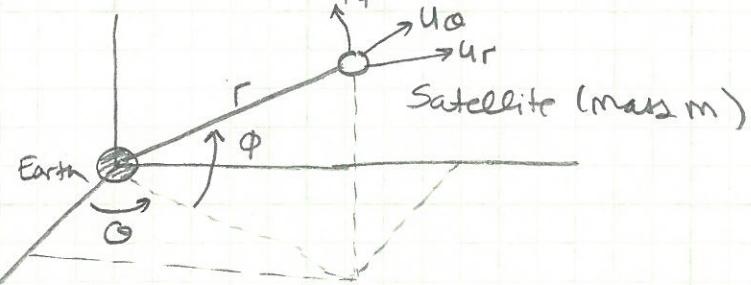
Note $\dot{x} = \begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \\ \vdots \\ \dot{x}_n \end{bmatrix} := \begin{bmatrix} f_1 \\ f_2 \\ f_3 \\ \vdots \\ f_n \end{bmatrix}$

* do $\frac{\partial f_i}{\partial x_j}$ above on all $i=1, \dots, n$ and $j=1, \dots, n$ to get $n \times n$ Jacobian matrix

i.e., $x_1(t) = f_1(x(t), u(t))$ nonlinear function

Example (Linearization)

Communication satellite orbiting around Earth



$$x(t) = \begin{bmatrix} r(t) \\ \dot{r}(t) \\ \theta(t) \\ \dot{\theta}(t) \\ \phi(t) \\ \dot{\phi}(t) \end{bmatrix}$$

state

$$u(t) = \begin{bmatrix} u_r(t) \\ u_\theta(t) \\ u_\phi(t) \end{bmatrix}$$

orthogonal
thrusts = controller

$$y(t) = \begin{bmatrix} r(t) \\ \theta(t) \\ \phi(t) \end{bmatrix}$$

altitude

$$\dot{x}(t) = f(x, u) = \begin{bmatrix} \dot{r} \\ \dot{\theta} \\ \dot{\phi} \\ \ddot{r} - r\dot{\theta}^2 \cos^2\phi + r\dot{\phi}^2 - K/r^2 + u_r/m \\ -2\dot{r}\dot{\theta}/r + 2\dot{\theta}\dot{\phi}\sin\phi/\cos\phi + u_\theta/mr\cos\phi \\ -\dot{\theta}^2 \cos\phi \sin\phi - 2\dot{r}\dot{\phi}/r + u_\phi/mr \end{bmatrix}$$

A solution corresponds to the satellite being in a circular equatorial orbit:

$$x_0(t) = [r_0 \ 0 \ \omega_0 t \ \omega_0 \ 0 \ 0]'$$

$$u_0(t) = 0$$

where radius r_0 & angular velocity ω_0 are s.t. $r_0^3\omega_0^2 = K$ constant

Satellite will deviate from (x_0, u_0) due to disturbances.

Of interest to consider linearized equations around (x_0, u_0) .

Use Jacobians to find linear equations

$$\text{By definition } y = Cx = \left[\begin{array}{cccc|cc} 1 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 \end{array} \right] x$$

Example (continued)

$$B = \left. \frac{\partial f}{\partial u} \right|_{X_0, U_0} = \begin{bmatrix} \frac{\partial f_1}{\partial u_1} & \frac{\partial f_1}{\partial v_2} & \frac{\partial f_1}{\partial u_3} \\ \frac{\partial f_2}{\partial u_1} & \frac{\partial f_2}{\partial v_2} & \frac{\partial f_2}{\partial v_3} \\ \vdots & & \\ \frac{\partial f_b}{\partial u_1} & \frac{\partial f_b}{\partial u_2} & \frac{\partial f_b}{\partial u_3} \end{bmatrix}_{X_0, U_0} = \begin{bmatrix} 0 & 0 & 0 \\ 0 & 0 & 0 \\ 0 & 0 & 0 \\ \hline 0 & 0 & 0 \\ 0 & 0 & 0 \end{bmatrix}_{X_0, U_0}$$

$$A = \left. \frac{\partial f}{\partial x} \right|_{X_0, U_0} = \begin{bmatrix} \frac{\partial f_1}{\partial x_1} & \frac{\partial f_1}{\partial x_2} & \cdots & \frac{\partial f_1}{\partial x_6} \\ \vdots & & & \\ \frac{\partial f_b}{\partial x_1} & \frac{\partial f_b}{\partial x_2} & \cdots & \frac{\partial f_b}{\partial x_6} \end{bmatrix}_{X_0, U_0}$$

$$= \begin{bmatrix} 0 & 1 & 0 & 0 & 0 & 0 \\ 3w_0^2 & 0 & 0 & 2w_0\Gamma_0 & 0 & 0 \\ 0 & 0 & 0 & 1 & 0 & 0 \\ 0 & -2w_0/\Gamma_0 & 0 & 0 & 0 & 0 \\ \hline 0 & 0 & 0 & 0 & 0 & 1 \\ 0 & 0 & 0 & 0 & -w_0^2 & 0 \end{bmatrix}$$

Calculations

$$\left. \frac{\partial f_2}{\partial x_1} \right|_{X_0, U_0} = \left. \dot{\theta}^3 \cos^2 \phi + \phi^2 + 2K/\Gamma_0 \right|_{X_0, U_0} = w_0^2 \cos \phi + \frac{2\Gamma_0^3 w_0^2}{\Gamma_0^3} = 3w_0^2$$

$$\dot{\theta} = w_0, \phi = 0, \Gamma = \Gamma_0, K = \Gamma_0^3 w_0^2$$

$$\left. \frac{\partial f_2}{\partial x_4} \right|_{X_0, U_0} = \left. \frac{\partial f_2}{\partial \dot{\theta}} \right|_{X_0, U_0} = 2\Gamma \dot{\theta} \cos^2 \phi \Big|_{X_0, U_0} = 2\Gamma_0 w_0$$

(ICP)

$$\text{Derive } \left. \frac{\partial f_4}{\partial x_2} \right|_{X_0, U_0} = \left. \frac{\partial f_4}{\partial \dot{\theta}} \right|_{X_0, U_0} = -\frac{2\dot{\theta}}{\Gamma} \Big|_{X_0, U_0} = -\frac{2w_0}{\Gamma_0}$$

$$\left. \frac{\partial f_6}{\partial x_5} \right|_{X_0, U_0} = \left. \frac{\partial f_6}{\partial \dot{\theta}} \right|_{X_0, U_0} = -\dot{\theta}^2 \frac{\partial}{\partial \dot{\theta}} (\cos \phi \sin \phi) \Big|_{X_0, U_0} = -w_0^2$$

Example (continued)

$$\begin{cases} x(t) = x_0(t) + \delta x(t) \\ u(t) = u_0(t) + \delta u(t) \\ y(t) = y_0 + \delta y(t) \end{cases}$$

Small perturbations

⇒ Linearized equations

$$\begin{cases} \dot{\delta x} = A \delta x + B \delta u \\ \dot{\delta y} = C \delta x \end{cases} \quad \text{with } A, B, C \text{ as derived above}$$

Note - can be decoupled into 2 subsystems

$$\{r, \dot{r}, \theta, \dot{\theta}\} \text{ and } \{\phi, \dot{\phi}\}$$

Mechanical Systems

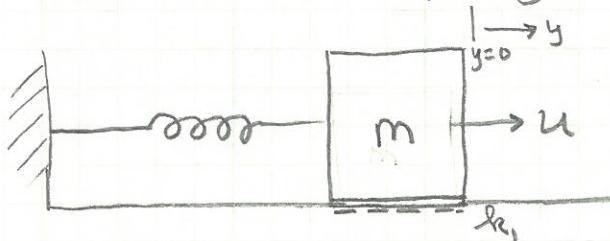
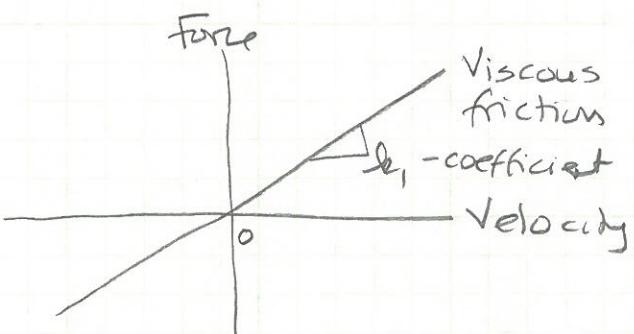
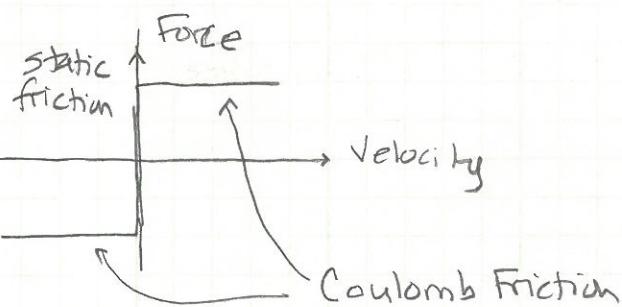
$$M(q) \ddot{q} + B(q, \dot{q}) \dot{q} + G(q) = F$$

equation
of
motion

where

 $q \in \mathbb{R}^k$ generalized coordinates vector $M(q)$ $k \times k$ positive-definite matrix called the mass matrixNote A symmetric matrix M is PD if

$$x^T M x > 0 \quad \forall x \neq 0$$

 $F \in \mathbb{R}^k$ applied forces / torques $G(q) \in \mathbb{R}^k$ conservative forces (e.g., gravity, spring forces) $B(q, \dot{q}) \in \mathbb{R}^{k \times k}$ centrifugal / Coriolis / friction matrixExample 1 (Mass-Spring) (Chen 2.5)output = y = displacement
from equilibrium

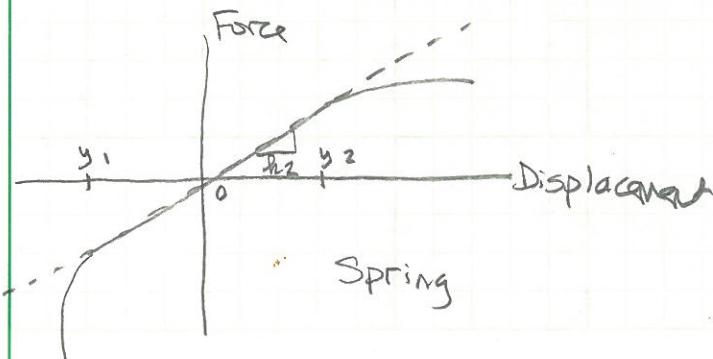
$$\frac{dy}{dt} = \text{velocity}$$

static + Coulomb friction are NL
 \Rightarrow disregard

$$\text{Viscous friction} = \delta_1 y(t)$$

Spring displacement $\in (y_1, y_2)$
is approx. linear

$$\text{Spring force} = \delta_2 y$$



Example 1 (continued)

Model mechanical system as linear under linearization & simplification;

Newton's Law: $F = ma$

applied force, u , must overcome viscous friction & spring force & also accelerate the block

$$m\ddot{y} = u - k_1 \dot{y} - k_2 y$$

$\uparrow \quad \uparrow \quad \uparrow \quad \uparrow$
 $M(q) \ddot{q} \quad F \quad B(q, \dot{q})\dot{q} \quad G(q)$

Apply Laplace transform & assume zero i.c.

$$m s^2 \hat{y}(s) = \hat{u}(s) - k_1 s \hat{y}(s) - k_2 \hat{y}(s)$$

$$\Rightarrow \hat{y}(s) = \underbrace{\frac{1}{ms^2 + k_1 s + k_2}}_{\hat{G}(s) \text{ transfer function}} \hat{u}(s)$$

State Space Equations:

$$\text{Let } x_1 = y$$

$$x_2 = \dot{y}$$

$$\text{Then } \dot{x}_1 = x_2$$

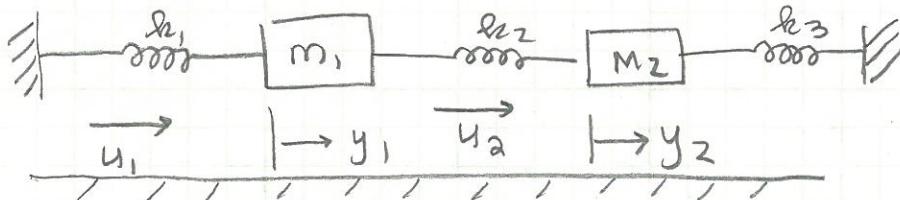
$$\dot{x}_2 = \frac{1}{m} (u - k_1 x_2 - k_2 x_1)$$

$$\begin{bmatrix} \dot{x}_1(t) \\ \dot{x}_2(t) \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ -k_2/m & -k_1/m \end{bmatrix} \begin{bmatrix} x_1(t) \\ x_2(t) \end{bmatrix} + \begin{bmatrix} 0 \\ m \end{bmatrix} u(t)$$

$$y(t) = [1 \quad 0] \begin{bmatrix} x_1(t) \\ x_2(t) \end{bmatrix}$$

Example 2 (Spring-Mass System)

(Chap 2.5)



Vibration example

Assume - no friction between blocks & floor

Applied force u_1 must overcome spring forces
& accelerate block

$$\Rightarrow u_1 - k_1 y_1 - k_2(y_1 - y_2) = m_1 \ddot{y}_1 \\ \text{or} \\ m_1 \ddot{y}_1 + (k_1 + k_2) y_1 - k_2 y_2 = u_1$$

For 2nd block,

$$m_2 \ddot{y}_2 - k_2 y_1 + (k_1 + k_2) y_2 = u_2$$

$$\Rightarrow \begin{bmatrix} m_1 & 0 \\ 0 & m_2 \end{bmatrix} \begin{bmatrix} \ddot{y}_1 \\ \ddot{y}_2 \end{bmatrix} + \begin{bmatrix} k_1 + k_2 & -k_2 \\ -k_2 & k_1 + k_2 \end{bmatrix} \begin{bmatrix} y_1 \\ y_2 \end{bmatrix} = \begin{bmatrix} u_1 \\ u_2 \end{bmatrix}$$

"normal form"
Standard vibration
equation

$$\text{Define } x_1 = y_1 \quad x_3 = y_2$$

$$x_2 = \dot{y}_1 \quad x_4 = \dot{y}_2$$

state
variables

$$\Rightarrow \begin{bmatrix} \ddot{x}_1 \\ \ddot{x}_2 \\ \ddot{x}_3 \\ \ddot{x}_4 \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 & 0 \\ -(k_1 + k_2)/m_1 & 0 & k_2/m_1 & 0 \\ 0 & 0 & 0 & 1 \\ k_2/m_2 & 0 & -(k_1 + k_2)/m_1 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \\ x_4 \end{bmatrix} + \begin{bmatrix} 0 & 0 \\ 0 & 0 \\ 0 & 0 \\ 1 & 0 \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \end{bmatrix}$$

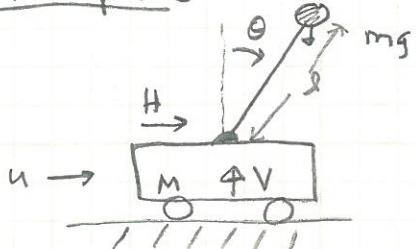
$$y := \begin{bmatrix} y_1 \\ y_2 \end{bmatrix} = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 \end{bmatrix} x$$

$$\text{Laplace Transform: } \underset{\text{(zero i.c.)}}{m_1 s^2 \hat{y}_1(s) + (k_1 + k_2) \hat{y}_1(s) - k_2 \hat{y}_2(s)} = \hat{u}_1(s) \\ m_2 s^2 \hat{y}_2(s) - k_2 \hat{y}_1(s) + (k_1 + k_2) \hat{y}_2(s) = \hat{u}_2(s)$$

$$\Rightarrow \begin{bmatrix} \hat{y}_1(s) \\ \hat{y}_2(s) \end{bmatrix} = \begin{bmatrix} \frac{m_2 s^2 + k_1 + k_2}{d(s)} & \frac{k_2/d(s)}{d(s)} \\ \frac{k_2/d(s)}{d(s)} & \frac{m_2 s^2 + k_1 + k_2}{d(s)} \end{bmatrix} \begin{bmatrix} \hat{u}_1(s) \\ \hat{u}_2(s) \end{bmatrix}$$

$$d(s) = (m_1 s^2 + k_1 + k_2)(m_2 s^2 + k_1 + k_2) - k_2^2$$

Example 3 (Inverted Pendulum on a Cart) (Chap 2.5)



Assume cart & pendulum move in one plane.

Disregard friction, mass of stick, gust of wind

Problem: maintain pendulum at vertical position

Simple model for a space vehicle at take-off

Let H = horizontal force } exerted by cart on pendulum
 V = vertical force }

Apply Newton's Law to linear movements:

$$\begin{cases} M\ddot{y} = u - H \\ H = m \frac{d^2}{dt^2} (y + l \sin \theta) = m\ddot{y} + ml\ddot{\theta} \cos \theta - ml\dot{\theta}^2 \sin \theta \\ mg - V = m \frac{d^2}{dt^2} (l \cos \theta) = ml(-\ddot{\theta} \sin \theta - \dot{\theta}^2 \cos \theta) \end{cases}$$

Apply Newton's Law to rotational movement of pendulum around hinge:

$$mgl \sin \theta = ml\ddot{\theta} \cdot l + m\ddot{y}l \cos \theta$$

Nonlinear equations.

Assume $\theta, \dot{\theta}$ small (want to keep pendulum vertical)

$$\Rightarrow \sin \theta = \theta$$

$$\cos \theta = 1$$

Drop H.O.T., e.g., $\theta^2, \dot{\theta}^2, \theta\dot{\theta}, \ddot{\theta}\dot{\theta}$

$$\Rightarrow \begin{cases} H = m\ddot{y} + ml\ddot{\theta} \\ V = mg \\ M\ddot{y} = u - m\ddot{y} - ml\ddot{\theta} \end{cases} \quad \begin{array}{l} \text{linear movement} \\ \text{rotational movement} \end{array}$$

$$\Rightarrow \text{Rewrite} \quad \begin{cases} M\ddot{y} = u - mg\dot{\theta} \\ ml\ddot{\theta} = (M+m)g\dot{\theta} - u \end{cases} \quad \begin{array}{l} \text{Linearized Equations} \end{array}$$

Example 3 (Continued)

Laplace Transform (zero i.c.)

$$Ms^2 \hat{y}(s) = \hat{u}(s) - mg \hat{\theta}(s) \quad \text{eqt 1}$$

$$Mls^2 \hat{\theta}(s) = (M+m)g \hat{\theta}(s) - \hat{u}(s) \quad \text{eqt 2}$$

Transfer Functions

$$u \rightarrow y : \quad y = \frac{u - mg\theta}{Ms^2} \quad \text{from eq. 1}$$

$$\theta = \frac{-u}{Mls^2 - (M+m)g} \quad \text{from eq 2}$$

$$\text{eq. 2} \rightarrow 1 \quad y = u - mg \left(\frac{-u}{Mls^2 - (M+m)g} \right)$$

$$= \frac{u}{Ms^2} \left(1 + \frac{mg}{Mls^2 - (M+m)g} \right)$$

$$= \frac{u}{Ms^2} \left(\frac{Mls^2 - Mg}{Mls^2 - (M+m)g} \right)$$

$$\hat{y}(s) = \frac{ls^2 - g}{s^2(Mls^2 - (M+m)g)} \cdot \hat{u}(s)$$

u → θ :

$$\theta = \frac{-u}{Mls^2 - (M+m)g}$$

$$\hat{\theta}(s) = \frac{-1}{Mls^2 - (M+m)g} \cdot \hat{u}(s)$$

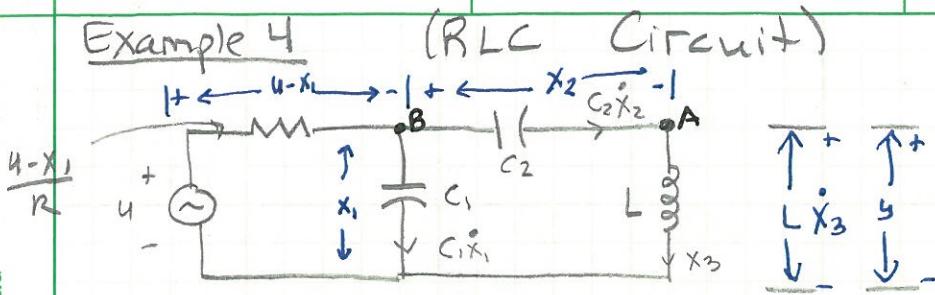
State Space

$$\begin{aligned} \text{Let } x_1 &= y \\ x_2 &= \dot{y} \end{aligned} \quad \begin{aligned} x_3 &= \theta \\ x_4 &= \dot{\theta} \end{aligned}$$

$$\Rightarrow \begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \\ \dot{x}_4 \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & -mg/M \\ 0 & 0 & 0 & 1 \\ 0 & 0 & (M+m)g/M & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \\ x_4 \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ 0 \\ -1/M \end{bmatrix} u$$

for small $\theta, \dot{\theta}$

Example 4 (RLC Circuit) (Chen 2.5)



Set States as capacitor voltages & inductor current:

$$x_1 = C_1 \text{ capacitor voltage}$$

$$x_2 = C_2 \text{ capacitor voltage}$$

$$x_3 = \text{inductor current}$$

with polarities
as shown

$$\Rightarrow \text{currents are } C_1 \dot{x}_1 \text{ & } C_2 \dot{x}_2$$

$$\text{voltage is } L \dot{x}_3$$

$$\Rightarrow \text{voltage across resistor } R \text{ is } u - x_1$$

$$V = IR \Rightarrow \text{current through } R \text{ is } \frac{u - x_1}{R}$$

Kirchoff's Current Law (KCL)

Sum of currents flowing into a node
= sum flowing out

KCL @ node A:

$$\frac{u - x_1}{R} = C_1 \dot{x}_1 + \underbrace{C_2 \dot{x}_2}_{= x_3} = C_1 \dot{x}_1 + x_3$$

$$\Rightarrow \begin{cases} \dot{x}_1 = \frac{x_1}{RC_1} - \frac{x_3}{C_1} + \frac{u}{RC_1} \\ \dot{x}_2 = \frac{x_3}{C_2} \end{cases}$$

Kirchoff's Voltage Law (KVL)

Sum of voltages around any closed circuit is zero

KVL on right-hand-side loop

$$L \dot{x}_3 = x_1 - x_2$$

$$\Rightarrow \dot{x}_3 = \frac{x_1}{L} - \frac{x_2}{L}$$

Example 4 (continued)

Output $y = L\dot{x}_3 = x_1 - x_2$

State space form:

$$\dot{x} = \begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \end{bmatrix} = \begin{bmatrix} -1/R_C_1 & 0 & -1/C_1 \\ 0 & 0 & 1/C_2 \\ 1/L & -1/L & 0 \end{bmatrix} x + \begin{bmatrix} 1/R_C_1 \\ 0 \\ 0 \end{bmatrix} u$$

$$y = [1 \quad -1 \quad 0] x + 0 \cdot u$$

Review

(see Handout #1)

Physical System → Model → Mathematical Equations
 given given derive
 "representation of system"

State Space Representation

$$\begin{cases} \dot{x} = Ax + Bu \\ y = Cx + Du \end{cases}$$

Linearization around an equilibrium point

Given NL system

$$\begin{cases} \dot{x}(t) = f(x(t), u(t)) & x \in \mathbb{R}^n, u \in \mathbb{R}^k \\ y(t) = g(x(t), u(t)) \end{cases}$$

s.t.

$$\dot{x}(t) = \begin{bmatrix} \dot{x}_1(t) \\ \dot{x}_2(t) \\ \vdots \\ \dot{x}_n(t) \end{bmatrix} = \begin{bmatrix} f_1(x(t), u(t)) \\ f_2(x(t), u(t)) \\ \vdots \\ f_n(x(t), u(t)) \end{bmatrix}$$

Given x^{eq}, u^{eq} at $f(x^{eq}, u^{eq}) = 0$ equilibrium point

Derive Jacobian matrices

$$A = \left. \frac{\partial f}{\partial x} \right|_{x^{eq}, u^{eq}} = \begin{bmatrix} \frac{\partial f_1}{\partial x_1} & \frac{\partial f_1}{\partial x_2} & \cdots & \frac{\partial f_1}{\partial x_n} \\ \vdots & \vdots & & \vdots \\ \frac{\partial f_n}{\partial x_1} & \cdots & \cdots & \frac{\partial f_n}{\partial x_n} \end{bmatrix} \Big|_{x^{eq}, u^{eq}}$$

$n \times n$ matrix

$$B = \left. \frac{\partial f}{\partial u} \right|_{x^{eq}, u^{eq}}, \quad C = \left. \frac{\partial g}{\partial x} \right|_{x^{eq}, u^{eq}}, \quad D = \left. \frac{\partial g}{\partial u} \right|_{x^{eq}, u^{eq}}$$

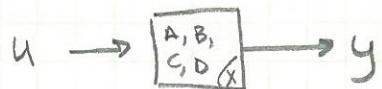
To get linear approximation of NL system around x^{eq}, u^{eq} :

$$\begin{cases} \dot{s}_x = A s_x + B s_u \\ y = C s_x + D s_u \end{cases}$$

Concept of State

(Kailath 2.2.3)

$$\begin{cases} \dot{x}(t) = Ax(t) + Bu(t) & x \in \mathbb{R}^n \\ y(t) = Cx(t) + Du(t) & y \in \mathbb{R}^m \end{cases}$$



$x(t)$ describes evolution of basic internal variables

If we know $x(t_0)$ for any given $t=t_0$,

Then given $u(t)$ for $t \geq t_0$, we can calculate all present & future values of $y(t)$

Note, if we were given $u(t)$ for all time $-\infty < t$ we could calculate $y(t) \forall t$

BUT We may only start looking at the system at $t=0$, and we may not know the input $u(t)$ for $-\infty < t < t_0$

So $x(t_0)$ gives sufficient information to be able to calculate the future response $t \geq t_0$ to a new input $u(t), t \geq t_0$, without worrying about what $u(t)$ was in the past.

Note - more than one past input can lead to $x(t_0)$, so $x(t_0)$ is a minimal sufficient statistic for the system, containing just enough information to calculate future responses

State vector gives knowledge of state/condition of system