

Lecture 10

Controllability

Lecture 11

Observability

Lecture 12

Decomposition

Usage - Necessary concepts to study
State feedback controllers
State observers

Motivation

Controllability: Satellite Attitude Control

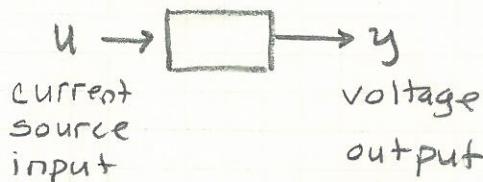
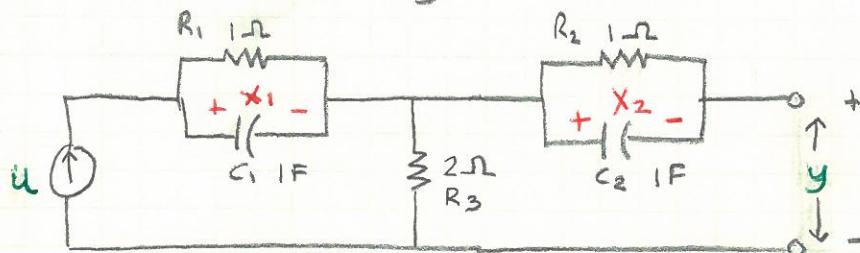
Satellite must change its orientation from one configuration (initial state) to another state in a certain amount of time.

Observability: Internal Combustion Engine

It is difficult/impossible to measure internal pressures & temperatures (states of the system). We can determine the states by observing the inputs & outputs over time.

Motivation

Consider the following network



States: x_1 - voltage across 1F capacitor C_1
 x_2 - voltage across 1F capacitor C_2

We notice:

- 1) The input current u has no effect on x_2 because of the open circuit across y
- 2) The current passing through the 2Ω resistor R_3 always $= u \Rightarrow$ the response excited by the initial state of x_1 will not appear in y .

We say:

- 1) The input u cannot control x_2
- 2) The initial state of x_1 cannot be observed from the output y

Concepts for the internal structure of linear systems:

- 1) Controllability - whether or not the state of a state-space equation can be controlled from the input.
- 2) Observability - whether or not the initial state can be observed from the output.

Lecture 10

Outline

Controllability of a Dynamical System

Controllability of a Linear System

Controllability of LTI Systems

Controllability Indices

Goal Steer the state from the input

Controllability of a Dynamical System

Consider the dynamical system $\mathcal{D} = \{\mathcal{U}, \Sigma, \mathcal{Y}, \mathcal{S}, \Gamma\}$

Recall \mathcal{U} = input set = $\{u(t), t \in T \subset \mathbb{R}\}$

Σ = state set = $\{x(t), t \in T\}$

\mathcal{Y} = output set = $\{y(t), t \in T\}$

\mathcal{S} = state transition function, $S(t_1, t_0, x_0, u) = x(t_1)$

"state at t_1 reached from x_0 at t_0 as a result of u "

Γ = read out function, $\Gamma(t, x(t), u(t))$

"response of \mathcal{D} at t when it is in state $x(t)$ at t and the input is $u(t)$ "

Definition

Given $t_0, t_1 \in T, t_1 > t_0$

\mathcal{D} is said to be completely controllable on $[t_0, t_1]$ "CC"

iff $\forall x_0, x_1 \in \Sigma, \exists u_{[t_0, t_1]} \in \mathcal{U}$ that drives
 (x_0, t_0) to (x_1, t_1)

$$x_1 = S(t_1, t_0, x_0, u)$$

Comments

- 1) The output can be disregarded because it does not play a role in controllability
- 2) The trajectory $x(t)$ takes from x_0 to x_1 is not specified
- 3) There is no constraint placed on the input e.g., the magnitude of $u(t)$ can be as large as desired

Special Cases

$$\mathcal{D} = \{ u, \Sigma, Y, s, r \}$$

1) $x_1 = 0$

\mathcal{D} is completely controllable to the origin on $[t_0, t_1]$

if $\forall x_0 \in \Sigma, \exists u_{[t_0, t_1]}$ such that

$$0 = s(t_1, t_0, x_0, u)$$

u transfers $x(t_0) = x_0$ to $x(t_1) = x_1 = 0$

2) $x_0 = 0$

\mathcal{D} is completely reachable from the origin on $[t_0, t_1]$

if $\forall x_1 \in \Sigma, \exists u_{[t_0, t_1]}$ such that

$$x_1 = s(t_1, t_0, 0, u)$$

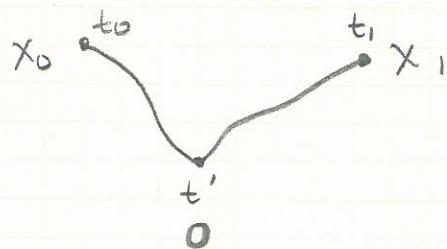
u transfers $x(t_0) = x_0 = 0$ to $x(t_1) = x_1$

Thus, we have

a) CC \Rightarrow CC to 0 on $[t_0, t_1]$

b) CC \Rightarrow CR from 0 on $[t_0, t_1]$

c) CC to 0 on $[t_0, t'] \quad \left. \begin{array}{l} \\ \text{CR from 0 on } [t', t_1] \end{array} \right\} \Rightarrow$ CC on $[t_0, t_1]$



Controllability of a Linear System

Consider the Linear system $R = [A(\cdot), B(\cdot), C(\cdot), D(\cdot)]$

$$\text{LTV} \begin{cases} \dot{x}(t) = A(t)x(t) + B(t)u(t) \\ y(t) = C(t)x(t) + D(t)u(t) \\ x(t_0) = x_0 \end{cases}$$

Theorem CC on $[t_0, t_1]$ iff $\text{cc to } 0$ on $[t_0, t_1]$
 iff cr from 0 on $[t_0, t_1]$

Proof $\text{CC} \Leftrightarrow \text{cc to } 0$

$\text{cc to } 0$ on $[t_0, t_1] \Leftrightarrow \exists \xi_0, \exists u$ s.t.

$$\begin{aligned} 0 &= s(t_1, t_0, \xi_0, u) \\ &= \Xi(t_1, t_0)\xi_0 + \int_{t_0}^{t_1} \Xi(t_1, \tau) B(\tau) u(\tau) d\tau \end{aligned}$$

We need to show $\exists x_0, x_1, \exists \tilde{u}$ s.t.

$$\begin{aligned} x_1 &= s(t_1, t_0, x_0, \tilde{u}) \\ &= \Xi(t_1, t_0)x_0 + \int_{t_0}^{t_1} \Xi(t_1, \tau) B(\tau) \tilde{u}(\tau) d\tau \end{aligned}$$

But

$$0 = x_1 - x_0 = \underbrace{(\Xi(t_1, t_0)x_0 - x_1)}_{\text{if } = \Xi(t_1, t_0)\xi_0, \text{ we are set}} + \int_{t_0}^{t_1} \Xi(t_1, \tau) B(\tau) \tilde{u}(\tau) d\tau$$

Pick ξ_0 as follows:

$$\Xi(t_1, t_0) \underbrace{(x_0 - \Xi^{-1}(t_1, t_0)x_1)}_{\xi_0} = \Xi(t_1, t_0)\xi_0$$

Pick $\tilde{u} = u$ to finish proof.

Note: We needed $\Xi(t, t_1)$ to be nonsingular

Proof \Leftrightarrow CR from 0

CR from 0 on $[t_0, t_1]$ $\Leftrightarrow \exists \xi_1, \exists \bar{u}$ st

$$\xi_1 = S(t_1, t_0, 0, \bar{u})$$

$$= \int_{t_0}^{t_1} \Xi(t_1, \tau) B(\tau) \bar{u}(\tau) d\tau$$

We need to show $\exists x_0, x_1, \exists u$ st

$$x_1 = S(t_1, t_0, x_0, u)$$

$$= \Xi(t_1, t_0) x_0 + \int_{t_0}^{t_1} \Xi(t_1, \tau) B(\tau) u(\tau) d\tau$$

Pick $\xi_1 = x_1 - \Xi(t_1, t_0) x_0$

$$u = \bar{u}$$

Then

$$\xi_1 = x_1 - \Xi(t_1, t_0) x_0 = \int_{t_0}^{t_1} \Xi(t_1, \tau) B(\tau) \bar{u}(\tau) d\tau$$

$$x_1 = \Xi(t_1, t_0) x_0 + \int_{t_0}^{t_1} \Xi(t_1, \tau) B(\tau) u(\tau) d\tau$$

Which proves the Theorem.

Remark Since for linear systems

$\text{CC} \Leftrightarrow \text{CR}$ from 0 on $[t_0, t_1]$

We only need to consider the ZSR of the system

$$x(t_0) = 0$$

$$x(t_1) = x_1 = \int_{t_0}^{t_1} \Xi(t_1, \tau) B(\tau) u(\tau) d\tau \quad \text{ZSR}$$

Definition

$$L_c : U_{[t_0, t_1]} \longrightarrow \int_{t_0}^{t_1} \Xi(t_1, \tau) B(\tau) u(\tau) d\tau \quad \text{ZSR}$$

The range space (or image) of L_c is defined as

the set of all $x_1 \in \mathbb{R}^n$ for which $x_1 = L_c u$ has a

solution $u \in \mathbb{R}^k$:

$$R(L_c) = \{x_1 \in \mathbb{R}^n : \exists u \in \mathbb{R}^k, x_1 = L_c u\}$$

x_1 is the ZSR

$$\dim R(L_c) = \text{rank}(L_c)$$

Remark Then we have

- 1) x is cr from 0 on $[t_0, t_1] \Leftrightarrow x \in R(L_c)$
- 2) $R(L_c) :=$ set of reachable states from 0 on $[t_0, t_1]$
- 3) cc on $[t_0, t_1] \Leftrightarrow R(L_c) = \mathbb{R}^n$

Definition Controllability Gramian

$$W_c(t_0, t_1) = \int_{t_0}^{t_1} \mathbb{E}(t, \tau) B(\tau) B'(\tau) \mathbb{E}'(t_1, \tau) d\tau \in \mathbb{R}^{n \times n}$$

This is an $n \times n$ matrix

Theorem Given $W_c(t_0, t_1)$ as above,

$$\begin{aligned} R(L_c) = \mathbb{R}^n &\Leftrightarrow W_c(t_0, t_1) \text{ is nonsingular} \\ &\Leftrightarrow \text{cc on } [t_0, t_1] \end{aligned}$$

Construct Input

Assume system is cc on $[t_0, t_1]$

How to find $u(t)$?

We have $W_c(t_0, t_1)$ nonsingular

$$x(t_1) = x_1 = \mathbb{E}(t_1, t_0) x_0 + \int_{t_0}^{t_1} \mathbb{E}(t_1, \tau) B(\tau) u(\tau) d\tau$$

Claim

$$u(t) = B'(t) \mathbb{E}'(t_1, t) W_c^{-1}(t_0, t_1) (x_1 - \mathbb{E}(t_1, t_0) x_0)$$

control input

Proof - by substitution. ($u(t)$ into $x(t_1)$)

Definition

The null space (or Kernel) of L_c is defined as
the set of all $u \in \mathbb{R}^k$ st $L_c(u) = 0$

$$N(L_c) = \{u \in \mathbb{R}^k : L_c(u) = 0\}$$

$$\dim N(L_c) = \text{nullity of } L_c$$

* Note $u(t)$ above is unique - prove by showing $N(L_c) = \{0\}$

Controllability of LTI Systems $R = [A, B, C, D]$

Theorem The following are equivalent:

- 1) (A, B) $A_{n \times n}$, $B_{n \times k}$ is controllable
- 2) The $n \times n$ matrix

$$W_c(t) = \int_0^t e^{A\tau} BB' e^{A'\tau} d\tau = \int_0^t e^{A(t-\tau)} BB' e^{A'(t-\tau)} d\tau$$

is nonsingular $\forall t > 0$ (Thus positive definite)

- 3) The $n \times nk$ matrix

$$\mathcal{C} = [B \ AB \ A^2B \ \dots \ A^{n-1}B]$$

Controllability Matrix

has rank n (full row rank)

- 4) The $n \times (n+k)$ matrix

$$[A - \lambda I \mid B]$$

has full row rank + eigenvalue λ of A

- 5) If all eigenvalues of A have negative real parts
 $\text{then } \exists!$ positive definite solution, W_c , to

the Lyapunov equation:

$$AW_c + W_c A' = -BB'$$

The solution can be expressed as

$$W_c = \int_0^\infty e^{A\tau} BB' e^{A'\tau} d\tau$$

Controllability Gramian

LTI $\left\{ \begin{array}{l} \dot{x} = Ax + Bu \\ y = Cx + Du \\ x(t_0) = x_0 \end{array} \right.$

Theorem

x is cr from 0 on $[t_0, t_1]$, i.e., $x \in R(L_c)$ for

$$L_c : U[t_0, t_1] \mapsto \int_{t_0}^{t_1} \mathbb{B}(t, \tau) B u(\tau) d\tau \quad ZSR$$

$$\Leftrightarrow x \in R(C) = \{x \in \mathbb{R}^n : \exists v \in \mathbb{R}^{n \times 1}, x = Cv\}$$

$$\text{where } C = [B \ AB \ A^2B \ \dots \ A^{n-1}B]$$

Proof (\Rightarrow)

Assume $x \in R(L_c)$

$$\text{Then } \exists u[t_0, t_1] \text{ s.t. } x = L_c u = \int_{t_0}^{t_1} e^{A(t_1-\tau)} B u(\tau) d\tau$$

$$\text{Recall } e^{At} = \sum_{k=0}^{\infty} \frac{A^k t^k}{k!}$$

Recall Cayley Hamilton Theorem gives

$$0 = \Delta(s) = \det(sI - A) = s^n + \alpha_1 s^{n-1} + \dots + \alpha_n$$

$$0 = \Delta(A) = A^n + \alpha_1 A^{n-1} + \dots + \alpha_n I$$

$$\Rightarrow (i) A^n = -\alpha_1 A^{n-1} - \alpha_2 A^{n-2} - \dots - \alpha_n I$$

$$(ii) A^k = -\gamma_1 A^{k-1} - \gamma_2 A^{k-2} - \dots - \gamma_k I \quad k \leq n$$

$$(iii) f(A) = \beta_1 A^{n-1} + \beta_2 A^{n-2} + \dots + \beta_n I$$

for any polynomial of A

characteristic
polynomial
of A

useful
formula

Thus for $f(A) = e^{At}$, $e^{At} = \beta_1 A^{n-1} + \beta_2 A^{n-2} + \dots + \beta_n I$

Thus

$$x = \sum_{i=0}^{n-1} A^i B \underbrace{\int_{t_0}^{t_1} \beta_i(t_1-\tau) u(\tau) d\tau}_{\text{a vector}}$$

Thus x is a linear combination of $B, AB, \dots, A^{n-1}B$

Thus $x \in R(C)$

$\Rightarrow R(C)$ is the set of reachable states.

Constructing the Input $u(t)$

Claim: the input

$$u(t) = -B' e^{A'(t_1-t)} W_c^{-1}(t_1) [e^{At_1} x_0 - x_1]$$

"minimum energy control"

will transfer $x(0) = x_0$ to

$$x(t_1) = x_1 = e^{At_1} x_0 + \int_0^{t_1} e^{A(t_1-\tau)} B u(\tau) d\tau$$

Proof: by substitution

$$\begin{aligned} x(t_1) &= e^{At_1} x_0 - \underbrace{\int_0^{t_1} e^{A(t_1-\tau)} B B' e^{A'(t_1-\tau)} d\tau}_{W_c(t_1)} \underbrace{W_c^{-1}(t_1) [e^{At_1} x_0 - x_1]}_{\text{not a function of } t} \\ &= e^{At_1} x_0 - W_c(t_1) W_c^{-1}(t_1) [e^{At_1} x_0 - x_1] \\ &= e^{At_1} x_0 - I [e^{At_1} x_0 - x_1] \\ &= e^{At_1} x_0 - e^{At_1} x_0 + x_1 \\ &= x_1 \end{aligned}$$

Remark

If W_c nonsingular, W_c^{-1} exists and $u(t)$ transfers $x_0 \rightarrow x_1 \Rightarrow (A, B)$ is controllable

Remark

Control "energy" $J = \frac{1}{2} \int_{t_0}^{t_1} \|u(t)\|^2 dt$

Minimum energy control = $u(t)$ that expends minimal energy

$u: x_0 \rightarrow x_1$
to t_1

$$\int_{t_0}^{t_1} \bar{u}^*(t) \bar{u}(t) dt = \int_{t_0}^{t_1} u^*(t) u(t) dt$$

for any other input $\bar{u}(t): x_0 \rightarrow x_1$
to t_1

Application: Minimize fuel consumption on satellite launch

Example 1

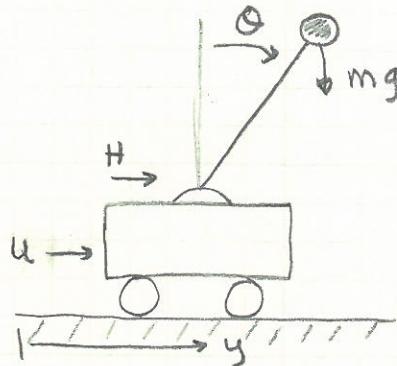
$$\dot{x} = \begin{bmatrix} 0 & 1 & 0 & 0 \\ 0 & 0 & -1 & 0 \\ 0 & 0 & 0 & -1 \\ 0 & 0 & 5 & 0 \end{bmatrix} x + \begin{bmatrix} 0 \\ 1 \\ 0 \\ -2 \end{bmatrix} u$$

$$y = [1 \ 0 \ 0 \ 0] x$$

$$C = [B \ AB \ A^2B \ A^3B] \quad n=4$$

$$= \begin{bmatrix} 0 & 1 & 0 & 2 \\ -1 & 0 & 2 & 0 \\ 0 & -2 & 0 & -10 \\ -2 & 0 & -10 & 0 \end{bmatrix}$$

$$\text{rank}(C) = 4 \Rightarrow \text{cc}$$



Inverted Pendulum

\approx balancing
broom
on palm

Example 2

$$\dot{x} = \begin{bmatrix} -0.5 & 0 \\ 0 & -1 \end{bmatrix} x + \begin{bmatrix} 0.5 \\ 1 \end{bmatrix} u$$

$$\text{If } x_1(0) = 10 \text{ and } x_2(0) = -1,$$

Can we apply a force to bring the platform to equilibrium in 2 seconds?

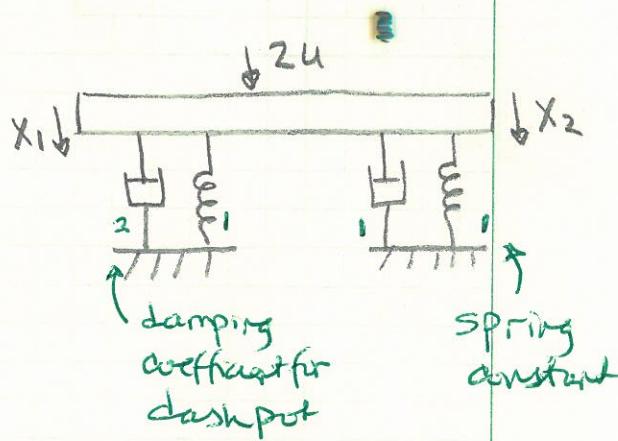
$$\text{rank}([B \ AB]) = \text{rank}(C) = \text{rank}\left(\begin{bmatrix} 0.5 & -0.25 \\ 0 & -1 \end{bmatrix}\right) = 2 = n$$

Thus cc. Thus $\exists u : x(0) \rightarrow 0$ in 2 seconds.

$$\begin{aligned} W_C(2) &= \int_0^2 e^{AT} BB' e^{A'T} dt \\ &= \int_0^2 \begin{bmatrix} e^{-T/2} & 0 \\ 0 & e^{-T} \end{bmatrix} \begin{bmatrix} 0.5 \\ 1 \end{bmatrix} \begin{bmatrix} 0.5 & 1 \\ 0 & 1 \end{bmatrix} \begin{bmatrix} e^{-T/2} & 0 \\ 0 & e^{-T} \end{bmatrix} dt \\ &= \begin{bmatrix} 0.2162 & 0.3167 \\ 0.3167 & 0.4908 \end{bmatrix} \end{aligned}$$

$$\begin{aligned} u(t) &= -B' e^{A'(2-t)} W_C^{-1}(2) [e^{2A} x_0 - x_1] \\ &= -[0.5 \ 1] \begin{bmatrix} e^{-(2-t)/2} & 0 \\ 0 & e^{-(2-t)} \end{bmatrix} W_C^{-1}(2) \begin{bmatrix} e^{-1} & 0 \\ 0 & e^{-2} \end{bmatrix} \begin{bmatrix} 10 \\ -1 \end{bmatrix} \\ &= -58.82 e^{t/2} + 27.96 e^{-t} \quad t \in [0, 2] \end{aligned}$$

plot in Matlab



Controllability Indices

$$\mathcal{C} = [B \ AB \ A^2B \ \dots \ A^{n-1}B]$$

Controllability Matrix

Case 1: $B \ n \times 1$, $A \ n \times n \Rightarrow \mathcal{C} \ n \times n$

If $\text{rank}(\mathcal{C}) = n$, \mathcal{C}^{-1} exists, (A, B) controllable

\Rightarrow all columns of \mathcal{C} are linearly independent

\Rightarrow can use columns as a basis for $T\mathbb{R}^n$ useful fact
in control
design

Case 2: $B \ n \times k$, $A \ n \times n$ Multivariable System

$$\mathcal{C} = [b_1 \ b_2 \ \dots \ b_k \mid Ab_1 \ Ab_2 \ \dots \ Ab_k \mid \dots \mid A^{n-1}b_1 \ A^{n-1}b_2 \ \dots \ A^{n-1}b_k]$$

$n \times nk$ matrix

There are many ways to choose n lin. indep. columns

$\text{rank}(\mathcal{C}) = n \Rightarrow n$ lin. indep. columns

Method I: Search \mathcal{C} from left to right to

find n linearly independent columns

$$\mathcal{C}_j = [B \mid AB \mid \dots \mid A^j B]$$

rank: $p(\mathcal{C}_0) \leq p(\mathcal{C}_1) \leq \dots \leq p(\mathcal{C}_p) \leq \dots \leq p(\mathcal{C}_{n-1}) = p(\mathcal{C})$
 $= n$ if controllable

Define μ to be the integer s.t.

$$p(\mathcal{C}_0) < p(\mathcal{C}_1) < \dots < p(\mathcal{C}_{\mu-1}) \underset{\text{yellow}}{=} p(\mathcal{C}_\mu) = \dots = p(\mathcal{C}) = n$$

μ is called the controllability index

Rearrange n lin. indep. columns of \mathcal{C}

$$[b_1 \ Ab_1 \ \dots \ A^{\mu_1-1}b_1 \mid b_2 \ Ab_2 \ \dots \ A^{\mu_2-1}b_2 \mid \dots \mid b_\mu \ Ab_\mu \ \dots \ A^{\mu_{\mu}-1}b_\mu]$$

n columns

$\{\mu_1, \mu_2, \dots, \mu_k\}$ are called the controllability indices of (A, B)

Note $\sum \mu_i = n$

$$\mu = \max_i \mu_i$$

Theorem $\{\mu_i\}$ $i=1, \dots, k$ is invariant under any equivalence transformation and any reordering of the columns of B .

Method II: Rearrange C as follows

$$C = \underbrace{[b_1, Ab_1, \dots, A^{n-1}b_1]}_{n \times n_p} \quad \underbrace{[b_2, Ab_2, \dots, A^{n-1}b_2]}_{\bar{\mu}_1 \text{ LI columns}} \quad \dots \quad \underbrace{[b_k, Ab_k, \dots, A^{n-1}b_k]}_{\bar{\mu}_k \text{ additional LI columns}}$$

and search left to right for \underline{n} LI columns.

$$\Rightarrow \sum \bar{\mu}_i = n \text{ if controllable}$$

$\bar{\mu}_i$ depends on ordering of $\{b_1, b_2, \dots, b_k\}$

$$\bar{\mu} = \max_i \bar{\mu}_i = \mu$$

Example

$$A = \begin{bmatrix} 2 & 0 & 1 \\ 0 & 2 & 0 \\ 0 & 0 & 1 \end{bmatrix}$$

$$AB = \begin{bmatrix} 0 & 1 \\ 2 & 0 \\ 1 & 1 \end{bmatrix}$$

$$B = \begin{bmatrix} b_1 & b_2 \\ 0 & 0 \\ 1 & 0 \\ 0 & 1 \end{bmatrix}$$

$$A^2B = \begin{bmatrix} -1 & 3 \\ 4 & 0 \\ 3 & 1 \end{bmatrix}$$

$$n=3$$

$$k=2$$

$$x \in \mathbb{R}^3$$

$$u \in \mathbb{R}^2$$

Method II:

$$C = \left[\begin{array}{cc|cc|cc} 0 & 0 & 0 & 1 & 1 & 3 \\ -1 & 0 & 2 & 0 & 4 & 0 \\ 0 & -1 & 1 & -1 & 3 & 1 \\ \hline b_1 & b_2 & Ab_1 & Ab_2 & A^2b_1 & A^2b_2 \end{array} \right]$$

$$\mu = 2$$

= # blocks needed to find $n=3$ LI columns

$$\text{Rearrange } C = \left[\begin{array}{c|cc} 0 & 0 & 1 \\ 1 & 0 & 0 \\ 0 & 1 & 0 \\ \hline b_1 & b_2 & Ab_2 \end{array} \right] \quad \mu_1 = 1 \quad \mu_1 + \mu_2 = 3$$

$$M_2 = 2 \quad \mu = \max(\mu_1, \mu_2) = 2$$

Method II:

$$C = \left[\begin{array}{ccc|ccc} 0 & 0 & 1 & 0 & 1 & 3 \\ -1 & 2 & 4 & 0 & 0 & 0 \\ 0 & 1 & 3 & 1 & 1 & 1 \\ \hline b_1 & Ab_1 & A^2b_1 & b_2 & Ab_2 & A^2b_2 \end{array} \right] \quad \bar{\mu}_1 = 3$$

$$\bar{\mu}_2 = 0$$

$$\bar{\mu} = \max(\bar{\mu}_1, \bar{\mu}_2) = 3 > \mu = 2$$

Theorem

The controllability property is invariant under any equivalence transform.

Proof

Consider (A, B) with controllability matrix

$$\mathcal{C} = [B \ AB \ A^2B \ \dots \ A^{n-1}B]$$

And its equivalent pair (\bar{A}, \bar{B}) with

$$\bar{A} = PAP^{-1}$$

$$\bar{B} = PB$$

The controllability matrix of (\bar{A}, \bar{B}) is

$$\begin{aligned}\bar{\mathcal{C}} &= [\bar{B} \ \bar{AB} \ \dots \ \bar{A}^{n-1}\bar{B}] \\ &= [PB \ PAP^{-1}PB \ \dots \ PA^{n-1}P^{-1}PB] \\ &= [PB \ PAB \ \dots \ PA^{n-1}B] \\ &= P[B \ AB \ \dots \ A^{n-1}B] \\ &= P\mathcal{C}\end{aligned}\quad \downarrow (MN)^{-1} = N^{-1}M^{-1}$$

Because P is nonsingular, $\rho(\mathcal{C}) = \rho(\bar{\mathcal{C}})$

This proves the theorem. □

Theorem

(A, B) is controllable iff the matrix

$$\mathcal{C}_{n \times k+1} = [B \ AB \ \dots \ A^{n-k}B]$$

where $\rho(B) = k$, has rank n .

\Rightarrow don't need to check $n \times n+k$ matrix \mathcal{C}

Lecture 11

11.0

Outline

Observability of a Dynamical System

Observability of a Linear System

Observability of LTI Systems

Observability of a system

Goal) estimate the state from the output

Observability of a Dynamical System

$$\mathcal{D} = \{U, Z, Y, S, F\}$$

$$t_0, t_1 \in T \subset \mathbb{R} \quad t_1 > t_0$$

Definition

\mathcal{D} is said to be completely observable on $[t_0, t_1]$ "CO"

iff given any $U[t_0, t_1]$ and corresponding $y[t_0, t_1]$,
the initial state $x_0 = x(t_0)$ is uniquely determined.

Comment

Once x_0 is found, knowing S , we can find $x(t)$
for any time $t \in [t_0, t_1]$ uniquely:

$$x(t) = S(t, t_0, x_0, U[t_0, t_1])$$

Observability of a Linear System

$$R = [A(\cdot), B(\cdot), C(\cdot), D(\cdot)]$$

LTV $\begin{cases} \dot{x}(t) = A(t)x(t) + B(t)u(t) \\ y(t) = C(t)x(t) + D(t)u(t) \\ x(t_0) = x_0 \end{cases}$

Theorem x_0 on $[t_0, t_1]$ \Leftrightarrow ZIR x_0 on $[t_0, t_1]$

Proof

$$y(t_1) = C(t_1)\mathbb{E}(t_1, t_0)x_0 + \int_{t_0}^{t_1} C(t_1)\mathbb{E}(t, \tau)B(\tau)u(\tau)d\tau$$

y and u are known. x_0 is unknown.

Thus, we can write

$$C(t_1)\mathbb{E}(t_1, t_0)x_0 = \bar{y}(t_1) \quad \text{ZIR}$$

Where

$$\bar{y}(t_1) = y(t_1) - \int_{t_0}^{t_1} C(t_1)\mathbb{E}(t_1, \tau)B(\tau)u(\tau)d\tau \quad \text{Known}$$

Note - the input doesn't affect observability

Thus, x_0 can be uniquely determined from the ZIR

Definition

$$L_0 : x_0 \mapsto C(\cdot)\mathbb{E}(\cdot, t_0)x_0 \quad \text{ZIR}$$

Remarks

1) x_0 on $[t_0, t_1] \Leftrightarrow N(L_0) = \{0\}$ Nullespace

2) $N(L_0)$:= set of unobservable x_0

Recall $N(L_0) = \{x_0 \in \mathbb{R}^n : L_0 x_0 = 0\}$

DefinitionObservabilityGramian

$$W_o(t_0, t_1) = \int_{t_0}^{t_1} E'(t, t_0) C'(t) C(t) E(t, t_0) dt$$

This is an $n \times n$ matrix

Theorem Given $W_o(t_0, t_1)$ as above,

$C \circ$ on $[t_0, t_1] \Leftrightarrow W_o(t_0, t_1)$ is nonsingular

Construct Initial State

Assume system is $C \circ$ on $[t_0, t_1]$.

How to find $x(t_0) = x_0$?

We have $W_o(t_0, t_1)$ nonsingular

Claim

$$x_0 = W_o^{-1}(t_0, t_1) \int_{t_0}^{t_1} E'(t, t_0) C'(t) y(t) dt$$

is the unique initial state

Observability of LTI Systems

$$R = [A, B, C, D]$$

LTI $\left\{ \begin{array}{l} \dot{x} = Ax + Bu \\ y = Cx + Du \\ x(t_0) = x_0 \end{array} \right.$

$x \in \mathbb{R}^n$
 $y \in \mathbb{R}^m$
 $u \in \mathbb{R}^k$

Theorem The LTI system is observable iff

$$W_o(t) = \int_0^t e^{A^\top \tau} C' C e^{A\tau} d\tau$$

is nonsingular $\forall t > 0$. Observability Gramian

Theorem Duality

$$(A, B) \text{ is controllable} \Leftrightarrow (A', B') \text{ is observable}$$

Proof

$$(A, B) \text{ controllable} \Leftrightarrow W_c(t) = \int_0^t e^{At} B B' e^{A^\top t} dt$$

nonsingular || identical

$$(A', B') \text{ observable} \Leftrightarrow W_o(t) = \int_0^t e^{A^\top t} B B' e^{At} dt$$

nonsingular

Theorem The following are equivalent:

1) (A, C) is observable $A_{n \times n}, C_{m \times n}$

2) The $n \times n$ matrix

$$W_0(t) = \int_0^t e^{A^\top \tau} C^\top C e^{A\tau} d\tau$$

is nonsingular $\forall t > 0$

3) The $nm \times n$ matrix

$$\Theta = \begin{bmatrix} C \\ CA \\ \vdots \\ CA^{n-1} \end{bmatrix}$$

Observability Matrix

has rank n (full column rank)

4) The $(n+m) \times n$ matrix

$$\begin{bmatrix} A - \lambda I \\ C \end{bmatrix}$$

has full column rank \forall eigenvalue λ of A

5) If all eigenvalues of A have negative real parts
 \leftarrow Thus Asymptotically Stable
 then \exists positive definite solution W_0 to the
 Lyapunov Equation

$$A^\top W_0 + W_0 A = -C^\top C$$

The solution can be expressed as

$$W_0 = \int_0^\infty e^{A^\top \tau} C^\top C e^{A\tau} d\tau$$

Observability Gramian

Theorem

The observability property is invariant under
 any equivalence transformation.

Lecture 12

Outline

Decomposition of State Equations

- Separation of Controllable Part
- Separation of Observable Part
- Kalman Decomposition

Goal: Establish a relationship between
the state-space description and
the transfer function matrix $\hat{G}(s)$

Motivation: Suppose system is not cc (\bar{c}),
i.e., $R(\bar{c}) \neq \mathbb{R}^n$. We will show that
we can then change coordinates
(states) to decompose the system
states into reachable, unreachable,
observable & unobservable parts

Separation of Controllable Part

Recall the equivalence transformation:

$$\textcircled{1} \begin{cases} \dot{x} = Ax + Bu \\ y = Cx + Du \end{cases} \quad \Rightarrow \quad \textcircled{2} \begin{cases} \dot{\bar{x}} = \bar{A}\bar{x} + \bar{B}u \\ y = \bar{C}\bar{x} + \bar{D}u \end{cases}$$

P nonsingular $\bar{A} = PAP^{-1}$ $\bar{C} = CP^{-1}$
 $\bar{B} = PB$ $\bar{D} = D$

$x \in \mathbb{R}^n$
 $y \in \mathbb{R}^m$
 $u \in \mathbb{R}^k$

Remarks

- 1) (1) is equivalent to (2)
- 2) All properties of (1), including stability, controllability & observability are preserved in (2)
- 3) We also have

$$\bar{C} = PC \quad \text{and} \quad \bar{O} = CO^{-1}$$

Controllability
Matrix Observability
Matrix

Theorem For LTI System (1) with

$$\text{rank}(C) = p(C) = p([B \ AB \ \dots \ A^{n-1}B]) = n, < n$$

Meaning $\dim R(C) = n, < n$

Meaning (A, B) and (\bar{A}, \bar{B}) are not controllable

We form the $n \times n$ matrix

$$P^{-1} = [q_1 \ q_2 \ \dots \ q_n \ \dots \ q_n]$$

Where the first n , columns are any n , linearly independent columns of C , and the remaining columns can be arbitrarily chosen as long as P is nonsingular.

Theorem (continued)

Then the equivalence transformation $\bar{x} = Px$

or $x = P^{-1}\bar{x}$ will transform the LTI system ① into

$$\begin{cases} \begin{bmatrix} \dot{\bar{x}}_c \\ \dot{\bar{x}}_{\bar{c}} \end{bmatrix} = \begin{bmatrix} \bar{A}_c & \bar{A}_{12} \\ 0 & \bar{A}_{\bar{c}} \end{bmatrix} \begin{bmatrix} \bar{x}_c \\ \bar{x}_{\bar{c}} \end{bmatrix} + \begin{bmatrix} \bar{B}_c \\ 0 \end{bmatrix} u \\ y = \begin{bmatrix} \bar{C}_c & \bar{C}_{\bar{c}} \end{bmatrix} \begin{bmatrix} \bar{x}_c \\ \bar{x}_{\bar{c}} \end{bmatrix} + Du \end{cases}$$

where $\bar{A}_c \in \mathbb{R}^{n_c \times n_c}$

$$\bar{A}_{\bar{c}} \in (\mathbb{R}^{n - n_c}) \times (\mathbb{R}^{n - n_c})$$

Furthermore, the n_c -dimensional subsystem

$$\textcircled{2} \quad \begin{cases} \dot{\bar{x}}_c = \bar{A}_c \bar{x}_c + \bar{B}_c u \\ \bar{y} = \bar{C}_c \bar{x}_c + Du \end{cases}$$

is controllable and has the same transfer function matrix as the original system ①

Proof

i) Show ② and ① have same transfer function

$$\hat{G}_1(s) = C(sI - A)^{-1}B + D \stackrel{?}{=} \hat{G}_2(s) = \bar{C}_c(sI - \bar{A}_c)^{-1}\bar{B}_c + D$$

Substitute

$$A = P^{-1}\bar{A}P, \quad B = P^{-1}\bar{B}, \quad C = \bar{C}P$$

$$\begin{aligned} C(sI - A)^{-1}B &= \bar{C}P(sI - P^{-1}\bar{A}P)^{-1}P^{-1}B \\ &= \bar{C}(PsP^{-1} - \bar{A})^{-1}B \\ &= \bar{C}(sI - \bar{A})^{-1}B \\ &= \begin{bmatrix} \bar{C}_c & \bar{C}_{\bar{c}} \end{bmatrix} \begin{bmatrix} sI - \bar{A}_c & -\bar{A}_{12} \\ 0 & sI - \bar{A}_{\bar{c}} \end{bmatrix}^{-1} \begin{bmatrix} \bar{B}_c \\ 0 \end{bmatrix} \\ &= \begin{bmatrix} \bar{C}_c & \bar{C}_{\bar{c}} \end{bmatrix} \begin{bmatrix} (sI - \bar{A}_c)^{-1} & X \\ 0 & X \end{bmatrix} \begin{bmatrix} \bar{B}_c \\ 0 \end{bmatrix} \\ &= \bar{C}_c(sI - \bar{A}_c)^{-1}\bar{B}_c \\ &= \hat{G}_2(s) \end{aligned}$$

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Proof (continued)

2) Show ② is controllable

$$\bar{C} = [\bar{B}_c \quad \bar{A}_c \bar{B}_c \quad \dots \quad \bar{A}_c^{n-1} \bar{B}_c \quad \dots \quad \bar{A}_c^{n-1} \bar{B}_c]$$

Need to show $\text{rank } \rho(\bar{C}) = n_1$

$$C = [B \quad AB \quad \dots \quad A^{n-1}B] \quad \text{from original system ①}$$

$$PC = [PB \quad PAB \quad \dots \quad PA^{n-1}B]$$

$$= [\bar{B} \quad \bar{A}\bar{B} \quad \dots \quad \bar{A}^{n-1}\bar{B}]$$

$$= \left[\begin{array}{c|c|c|c} \bar{B}_c & \bar{A}_c \bar{B}_c & \dots & \bar{A}_c^{n-1} \bar{B}_c \\ \hline 0 & 0 & \dots & 0 \end{array} \right] \begin{matrix} n_1 \text{ rows} \\ n-n_1 \text{ rows} \end{matrix}$$

$$\rho(PC) = n_1$$

$$\rho(\bar{C}) = n_1 \quad \text{because } P \text{ is invertible}$$

Thus for

$$P\bar{C}_1 = [\bar{B}_c \quad \bar{A}_c \bar{B}_c \quad \dots \quad \bar{A}_c^{n-1} \bar{B}_c]$$

$$\rho(\bar{C}_1) = n_1$$

Which means the system ② is controllable. \blacksquare

Example (Chap 6.8)

$$\textcircled{1} \begin{cases} \dot{x} = \begin{bmatrix} 1 & 1 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 1 \end{bmatrix}x + \begin{bmatrix} 0 & 1 \\ 1 & 0 \\ 0 & 1 \end{bmatrix}u \\ y = [1 \ 1 \ 1]x \end{cases}$$

$n=3$
 $m=1$
 $k=2$

$$p(B) = 2$$

Therefore, we can use the matrix

$$b_2 = [B \ AB] \quad n-p(B) = 3-2=1$$

instead of the controllability matrix

$$b = [B \ AB \ A^2B]$$

to check for controllability

$$p(b_2) = p([B \ AB]) = p\left(\begin{bmatrix} 0 & 1 & 1 & 1 \\ 1 & 0 & 1 & 0 \\ 0 & 1 & 0 & 1 \end{bmatrix}\right) = 2 < 3$$

Thus, the state equation is not controllable.

Choose $P^{-1} = Q = \begin{bmatrix} 0 & 1 & 1 \\ 1 & 0 & 0 \\ 0 & 1 & 0 \end{bmatrix}$

$\underbrace{n_1=2}_{\text{LI columns of } e_2}$ choose to have full rank

Let $\bar{x} = Px$, Then:

$$\bar{A} = PAP^{-1} = \begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ 1 & 0 & -1 \end{bmatrix} \begin{bmatrix} 1 & 1 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 1 \end{bmatrix} \begin{bmatrix} 0 & 1 & 1 \\ 1 & 0 & 0 \\ 0 & 1 & 0 \end{bmatrix}$$

$$= \begin{bmatrix} 1 & 0 & 1 & 0 \\ -1 & 1 & 0 & 0 \\ 0 & 0 & 1 & 1 \end{bmatrix} = \begin{bmatrix} \bar{A}_c & \bar{A}_{12} \\ 0 & \bar{A}_{22} \end{bmatrix}$$

$$\bar{B} = PB = \begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ 1 & 0 & -1 \end{bmatrix} \begin{bmatrix} 0 & 1 \\ 1 & 0 \\ 0 & 1 \end{bmatrix} = \begin{bmatrix} 1 & 0 \\ 0 & 1 \\ 0 & 0 \end{bmatrix} = \begin{bmatrix} \bar{B}_c \\ 0 \end{bmatrix}$$

$$\bar{C} = CP^{-1} = [1 \ 1 \ 1] \begin{bmatrix} 0 & 1 & 1 \\ 1 & 0 & 0 \\ 0 & 1 & 0 \end{bmatrix} = [1 \ 2 \ 1] = [\bar{C}_c \ | \ \bar{C}_2]$$

Example (continued)

Thus the original system ① can be reduced to

$$\begin{cases} \dot{\bar{x}}_c = \bar{A}_c \bar{x}_c + \bar{B}_c u \\ \bar{y} = \bar{C}_c \bar{x}_c \end{cases}$$

or

$$\begin{cases} \dot{\bar{x}}_c = \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix} \bar{x}_c + \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix} u \\ y = \begin{bmatrix} 1 & 2 \end{bmatrix} \bar{x}_c \end{cases}$$

This equation is controllable and
has the same transfer function matrix $\hat{G}(s)$
as system ① has.

Separation of Observable Part (LTI)

$$\begin{cases} \dot{x} = Ax + Bu \\ y = Cx + Du \end{cases} \quad \begin{array}{l} x \in \mathbb{R}^n \\ y \in \mathbb{R}^m \\ u \in \mathbb{R}^k \end{array}$$

Theorem For the LTI system (1) with

$$P(\Theta) = P \begin{bmatrix} C \\ CA \\ \vdots \\ CA^{n-1} \end{bmatrix} = n_2 < n$$

Meaning (A, C) is not observable

We form the $n \times n$ matrix

$$P = \begin{bmatrix} P_1 \\ \vdots \\ P_{n_2} \\ \vdots \\ P_n \end{bmatrix}$$

Where the first n_2 rows are any n_2 linearly independent rows of Θ , and the remaining rows can be chosen arbitrarily as long as P is nonsingular.

Then the equivalence transformation $\bar{x} = Px$

will transform (1) into

$$\begin{cases} \begin{bmatrix} \dot{\bar{x}}_0 \\ \vdots \\ \dot{\bar{x}}_{n_2} \end{bmatrix} = \begin{bmatrix} \bar{A}_0 & 0 \\ \bar{A}_{21} & \bar{A}_{\bar{0}} \end{bmatrix} \begin{bmatrix} \bar{x}_0 \\ \vdots \\ \bar{x}_{n_2} \end{bmatrix} + \begin{bmatrix} \bar{B}_0 \\ \vdots \\ \bar{B}_{n_2} \end{bmatrix} u & \bar{A}_0 \text{ } n_2 \times n_2 \\ y = \begin{bmatrix} \bar{C}_0 & 0 \end{bmatrix} \begin{bmatrix} \bar{x}_0 \\ \vdots \\ \bar{x}_{n_2} \end{bmatrix} + Du & \bar{A}_{\bar{0}} \text{ } (n-n_2) \times (n-n_2) \end{cases}$$

And the n_2 -dimensional subsystem

$$(2) \begin{cases} \dot{\bar{x}}_0 = \bar{A}_0 \bar{x}_0 + \bar{B}_0 u \\ y = \bar{C}_0 \bar{x}_0 + Du \end{cases}$$

is observable and has the same transfer function matrix as the original system (1).

Kalman Decomposition Theorem (LTI)

Every state space equation can be transformed by an equivalence transformation into the following Kalman Canonical Form:

$$\begin{bmatrix} \dot{\bar{x}}_{c0} \\ \dot{\bar{x}}_{c\bar{0}} \\ \dot{\bar{x}}_{\bar{c}0} \\ \dot{\bar{x}}_{\bar{c}\bar{0}} \end{bmatrix} = \begin{bmatrix} \bar{A}_{c0} & 0 & \bar{A}_{13} & 0 \\ \bar{A}_{21} & \bar{A}_{c\bar{0}} & \bar{A}_{23} & \bar{A}_{24} \\ 0 & 0 & \bar{A}_{\bar{c}0} & 0 \\ 0 & 0 & \bar{A}_{43} & \bar{A}_{\bar{c}\bar{0}} \end{bmatrix} \begin{bmatrix} \bar{x}_{c0} \\ \bar{x}_{c\bar{0}} \\ \bar{x}_{\bar{c}0} \\ \bar{x}_{\bar{c}\bar{0}} \end{bmatrix} + \begin{bmatrix} \bar{B}_{c0} \\ \bar{B}_{c\bar{0}} \\ 0 \\ 0 \end{bmatrix} u$$

$$y = \begin{bmatrix} \bar{C}_{c0} & 0 & \bar{C}_{\bar{c}0} & 0 \end{bmatrix} \bar{x} + D u$$

where

\bar{x}_{c0} = controllable & observable states

$\bar{x}_{c\bar{0}}$ = controllable & unobservable states

$\bar{x}_{\bar{c}0}$ = uncontrollable & observable states

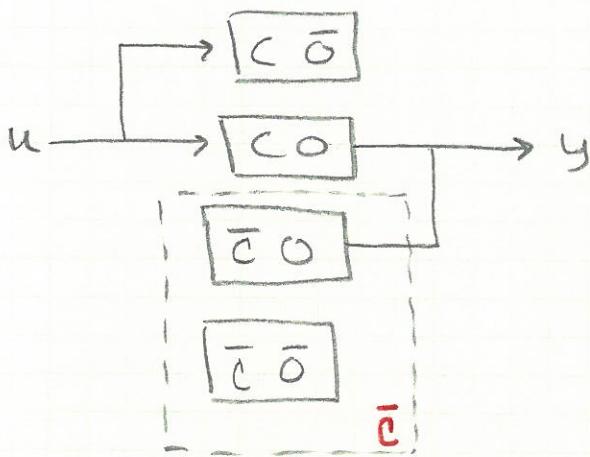
$\bar{x}_{\bar{c}\bar{0}}$ = uncontrollable & unobservable states

Furthermore, the state equation is zero-state equivalent (meaning ZSR same) to the controllable & observable state equation

$$\begin{cases} \dot{\bar{x}}_{c0} = \bar{A}_{c0} \bar{x}_{c0} + \bar{B}_{c0} u \\ y = \bar{C}_{c0} \bar{x}_{c0} + D u \end{cases}$$

And has the transfer function matrix

$$\hat{G}(s) = \bar{C}_{c0} (sI - \bar{A}_{c0})^{-1} \bar{B}_{c0} + D$$

Remark

- 1) The transfer function matrix X only depends on the $C + O$ part of the equation
 $\Rightarrow CO$ is only part connected to the input & output
- 2) The transfer function, or I/O, description of the system and the state-space description are not necessarily equivalent.
- 3) Thus, the I/O description of a system is sometimes insufficient to describe a system b/c \bar{C} & \bar{O} parts don't appear in the transfer function description