

EE 547 (PMP): Homework 1

Assigned: Thursday January 8, 2015. Due: Wednesday January 14, 2015.

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Problem 1 Consider a MinSeg model as below. The geometric parameters and some constants are labeled on the figure. A system of equations is derived as (1) to describe the movement and angle of rotation of MinSeg body.

- a) Let's focus on the translation of MinSeg wheel (x) and rotation of MinSeg pendulum (α). Please linearize (1) around equilibrium point $x^{eq} = \alpha^{eq} = 0$. (Hint: high order terms shall be ignored.)

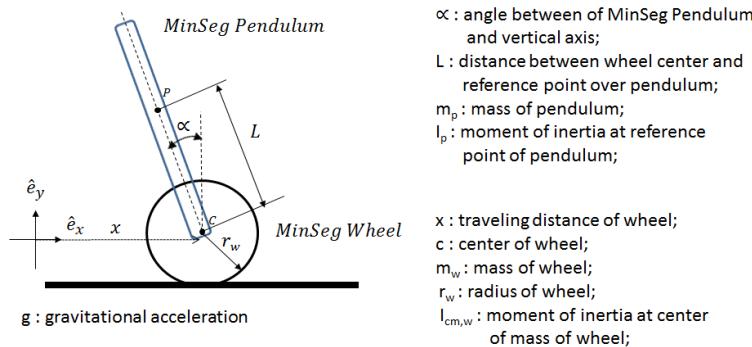


Figure 1 MinSeg system

$$\begin{bmatrix} -(I_p + m_p L^2) & m_p L \cos \alpha \\ m_p L r_w^2 \cos \alpha & -(I_{cm,w} + m_w r_w^2 + m_p r_w^2) \end{bmatrix} \begin{bmatrix} \ddot{x} \\ \ddot{\alpha} \end{bmatrix} = \begin{bmatrix} T_m - m_p L g \sin \alpha \\ T_m r_w + m_p L r_w^2 \dot{\alpha}^2 \cos \alpha \end{bmatrix} \quad (1)$$

where T_m is the torque from the DC motor, which is not explicitly demonstrated in Figure 1.

- b) A torque equation is given as (2). Please rewrite the derived linear model with (2).

$$T_m = \frac{k_t}{R}V + \frac{k_t k_b}{R r_w} \dot{x} + \frac{k_t k_b}{R} \dot{\alpha} \quad (2)$$

where V is applied voltage, R is the resistance of DC motor, k_t is torque constant and k_b is back-EMF constant.

- c) If state variables are defined as $X = [\alpha \dot{\alpha} x \dot{x}]^T$ and input variable is $u = V$, please derive state space matrices A and B. Now we are interested in $Y = [\alpha x]^T$ and there is no disturbance over the system. Please derive matrices C and D.

HW hint:

$$\begin{bmatrix} \ddot{\alpha} \\ \ddot{x} \end{bmatrix} = \begin{bmatrix} -(I_p + m_p L^2) & m_p L \cos \alpha \\ m_p L r_w^2 \cos \alpha & -(I_{cm,w} + m_w r_w^2 + m_p r_w^2) \end{bmatrix}^{-1} \begin{bmatrix} T_m - m_p L g \sin \alpha \\ T_m r_w + m_p L r_w^2 \dot{\alpha}^2 \cos \alpha \end{bmatrix} = \begin{bmatrix} f_2(\alpha, \dot{\alpha}, x, \dot{x}, V) \\ f_4(\alpha, \dot{\alpha}, x, \dot{x}, V) \end{bmatrix} \quad (3)$$

$$\dot{X} = \frac{\partial}{\partial t} \begin{bmatrix} \alpha \\ \dot{\alpha} \\ x \\ \dot{x} \end{bmatrix} = \begin{bmatrix} f_1(\alpha, \dot{\alpha}, x, \dot{x}, V) \\ f_2(\alpha, \dot{\alpha}, x, \dot{x}, V) \\ f_3(\alpha, \dot{\alpha}, x, \dot{x}, V) \\ f_4(\alpha, \dot{\alpha}, x, \dot{x}, V) \end{bmatrix} \quad (4)$$

Problem 1 solution:

- a) Around equilibrium point $x^{eq} = \alpha^{eq} = 0$, we can assume

$$\cos \alpha \sim 1; \sin \alpha \sim \alpha; \dot{\alpha}^2 \sim 0$$

Then, (1) can be simplified as (5).

$$\begin{bmatrix} -(I_p + m_p L^2) & m_p L \\ m_p L r_w^2 & -(I_{cm,w} + m_w r_w^2 + m_p r_w^2) \end{bmatrix} \begin{bmatrix} \ddot{\alpha} \\ \ddot{x} \end{bmatrix} = \begin{bmatrix} T_m - m_p L g \alpha \\ T_m r_w \end{bmatrix} \quad (5)$$

(5) can be rewritten as (6).

$$\begin{bmatrix} -(I_p + m_p L^2) & m_p L \\ m_p L & -\left(\frac{I_{cm,w}}{r_w^2} + m_w + m_p\right) \end{bmatrix} \begin{bmatrix} \ddot{\alpha} \\ \ddot{x} \end{bmatrix} = \begin{bmatrix} T_m - m_p L g \alpha \\ \frac{T_m}{r_w} \end{bmatrix} \quad (6)$$

$$\begin{bmatrix} \ddot{\alpha} \\ \ddot{x} \end{bmatrix} = \begin{bmatrix} -(I_p + m_p L^2) & m_p L \\ m_p L & -\left(\frac{I_{cm,w}}{r_w^2} + m_w + m_p\right) \end{bmatrix}^{-1} \begin{bmatrix} T_m - m_p L g \alpha \\ \frac{T_m}{r_w} \end{bmatrix} \quad (7)$$

- b) Simply substitute (2) into (6), the equations become

$$\begin{bmatrix} \ddot{\alpha} \\ \ddot{x} \end{bmatrix} = \begin{bmatrix} -(I_p + m_p L^2) & m_p L \\ m_p L & -\left(\frac{I_{cm,w}}{r_w^2} + m_w + m_p\right) \end{bmatrix}^{-1} \begin{bmatrix} \frac{k_t}{R} V + \frac{k_t k_b}{R r_w} \dot{x} + \frac{k_t k_b}{R} \dot{\alpha} - m_p L g \alpha \\ \frac{k_t}{R r_w} V + \frac{k_t k_b}{R r_w^2} \dot{x} + \frac{k_t k_b}{R r_w} \dot{\alpha} \end{bmatrix} \quad (8)$$

c) Since state variable is defined as $X = [\alpha \dot{\alpha} x \dot{x}]^T$, we could write the state-space representation as

$$\dot{X} = \begin{bmatrix} \dot{\alpha} \\ \ddot{\alpha} \\ \dot{x} \\ \ddot{x} \end{bmatrix} = [A] \begin{bmatrix} \alpha \\ \dot{\alpha} \\ x \\ \dot{x} \end{bmatrix} + [B]V \quad (9)$$

The first row and third row of matrix A are simple row vectors $[0 \ 1 \ 0 \ 0]$ and $[0 \ 0 \ 0 \ 1]$, respectively. To have second row and fourth row, we have to solve (8) by taking the inversion of coefficient matrix. The final matrix A and matrix B are shown in (10) and (11).

$$A = \begin{bmatrix} 0 & 1 & 0 & 0 \\ \frac{gLm_p(i_{cmw} + (m_p + m_w)r_w^2)}{i_{cmw}(i_p + L^2m_p) + (L^2m_p m_w + i_p(m_p + m_w))r_w^2} & -\frac{k_b k_t(i_{cmw} + r_w(m_w r_w + m_p(L + r_w)))}{R(i_{cmw}(i_p + L^2m_p) + (L^2m_p m_w + i_p(m_p + m_w))r_w^2)} & 0 & -\frac{k_b k_t(i_{cmw} + r_w(m_w r_w + m_p(L + r_w)))}{Rr_w(i_{cmw}(i_p + L^2m_p) + (L^2m_p m_w + i_p(m_p + m_w))r_w^2)} \\ 0 & 0 & 0 & 0 \\ \frac{gL^2m_p^2r_w^2}{i_{cmw}(i_p + L^2m_p) + (L^2m_p m_w + i_p(m_p + m_w))r_w^2} & -\frac{k_b k_t r_w(i_p + Lm_p(L + r_w))}{R(i_{cmw}(i_p + L^2m_p) + (L^2m_p m_w + i_p(m_p + m_w))r_w^2)} & 0 & -\frac{k_b k_t(i_p + Lm_p(L + r_w))}{R(i_{cmw}(i_p + L^2m_p) + (L^2m_p m_w + i_p(m_p + m_w))r_w^2)} \end{bmatrix} \quad (10)$$

$$B = \begin{bmatrix} 0 \\ -\frac{k_t(i_{cmw} + r_w(m_w r_w + m_p(L + r_w)))}{R(i_{cmw}(i_p + L^2m_p) + (L^2m_p m_w + i_p(m_p + m_w))r_w^2)} \\ 0 \\ -\frac{k_t r_w(i_p + Lm_p(L + r_w))}{R(i_{cmw}(i_p + L^2m_p) + (L^2m_p m_w + i_p(m_p + m_w))r_w^2)} \end{bmatrix} \quad (11)$$

$$C = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \\ 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix} \quad (12)$$

$$D = \begin{bmatrix} 0 \\ 0 \\ 0 \\ 0 \end{bmatrix} \quad (13)$$

MATLAB reference code:

```
%% MATLAB reference code for EE547 homework1 Witner 2015
%% Developer: HRLin
% Description: This is an example to show MATLAB's way to evaluate
% A B C D matrices through jacobian function.
clear all;
close all;
clc;
%% Definition and Initialization
% g: gravity constant
% t: time
% x: position of wheel center
% alpha: rotation angle
% L: distance between wheel center and reference point over pendulum
% mp: mass of pendulum
% Ip: moment of inertia of pendulum
% mw: mass of wheel
% rw: wheel radius of wheel
% Icmw: moment of inertia at wheel center of mass
% Tm: torque from DC motor
% V: applied voltage
% kt: torque constant
% kb: back EMF constant
% R: resistance of DC motor
% xi: state variables

syms g t x xdot alpha alphadot L mp mw rw Icmw Tm V kt kb R x1 x2 x3 x4

C1 = [ -(Ip + mp*L^2), mp*L*cos(alpha);
       mp*L*rw^2*cos(alpha), -(Icmw+mw*rw^2+mp*rw^2) ];

C2 = [ Tm-mp*L*g*sin(alpha);
       Tm*rw+mp*L*rw^2*alphadot^2*cos(alpha) ];

%Tm = kt/R*V+kt*kb/R/rw*xdot+kt*kb/R*alphadot;
%%
% state variable vector = [alpha alphadot x xdot] = [x1 x2 x3 x4]';
% time derivative of state variable vector =
% [alphadot alpha_doubledot xdot x_doubledot] = [f1 f2 f3 f4]
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C3 = inv(C1)*C2;
C3 = subs(C3, 'Tm', kt/R*V+kt*kb/R/rw*xdot+kt*kb/R*alphadot);

f1 = alphadot;
f2 = C3(1,1);
f3 = xdot;
f4 = C3(2,1);

f = [f1;f2;f3;f4];
xvec = [alpha alphadot x xdot]; % xvec is state variable, not position.
u = v;

JA = jacobian(f,xvec);
JB = jacobian(f,u);
A = subs(JA,xvec, [0 0 0 0]); % Two of equilibrium values are also zeros.
B = subs(JB,xvec, [0 0 0 0]);
C = eye(size(A));
D = zeros(size(C,1),size(B,2));

```