

Dassault Mirage 2000-5 Aerodynamic data built from vspaero; CG (7.72, 0, 0)M, 2021-04-10 10:41

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AeroDetail=MediumHigh, ExternalTanks, Flaps, Gear, GroundEffect, Mach, WakeIterations=3

Model summary

Dependent variable	Independent variables	Axis	Description
CFXB	alpha	DRAG	BASIC DRAG
CFXDED1L	alpha,beta,DED1L	DRAG	DRAG DUE TO ELEVON 1L
CFXDED1R	alpha,beta,DED1R	DRAG	DRAG DUE TO ELEVON 1R
CFXDED2L	alpha,beta,DED2L	DRAG	DRAG DUE TO ELEVON 2L
CFXDED2R	alpha,beta,DED2R	DRAG	DRAG DUE TO ELEVON 2R
CFXDSD1L	alpha	DRAG	DRAG DUE TO LE SLAT 1
CFXDSD2L	alpha	DRAG	DRAG DUE TO LE SLAT 2
CFXDSBL	alpha	DRAG	DRAG DUE TO LOWER SPEEDBRAKE DEFLECTION
CFXmn	mach,alpha	DRAG	DRAG DUE TO MACH
CFXDSBU	alpha	DRAG	DRAG DUE TO UPPER SPEEDBRAKE DEFLECTION
CFXGEAR	alpha	DRAG	DRAG INCREMENT DUE TO GEAR
CFXCTNK	alpha,beta	DRAG	DRAG INCREMENT DUE TO TANK(CENTRE)
CFXLTNK	alpha,beta	DRAG	DRAG INCREMENT DUE TO TANK(LEFT WING)
CFXRTNK	alpha,beta	DRAG	DRAG INCREMENT DUE TO TANK(RIGHT WING)
CFZB	alpha	LIFT	BASIC LIFT
CFZDED1L	alpha,beta,DED1L	LIFT	LIFT DUE TO ELEVON 1L
CFZDED1R	alpha,beta,DED1R	LIFT	LIFT DUE TO ELEVON 1R
CFZDE2L	alpha,beta,DED2L	LIFT	LIFT DUE TO ELEVON 2L
CFZDE2R	alpha,beta,DED2R	LIFT	LIFT DUE TO ELEVON 2R
CFZDSD1L	alpha	LIFT	LIFT DUE TO LE SLAT 1
CFZDSD2L	alpha	LIFT	LIFT DUE TO LE SLAT 2
CFZDEL	alpha	LIFT	LIFT DUE TO LOWER SPEEDBRAKE DEFLECTION
CFZmn	mach,alpha	LIFT	LIFT DUE TO MACH
CFZDSBU	alpha	LIFT	LIFT DUE TO UPPER SPEEDBRAKE DEFLECTION
CFZGEAR	alpha	LIFT	LIFT INCREMENT DUE TO GEAR
CFZCTNK	alpha,beta	LIFT	LIFT INCREMENT DUE TO TANK(CENTRE)
CFZLTNK	alpha,beta	LIFT	LIFT INCREMENT DUE TO TANK(LEFT WING)
CFZRTNK	alpha,beta	LIFT	LIFT INCREMENT DUE TO TANK(RIGHT WING)
CMM1	alpha	PITCH	BASIC PITCHING MOMENT
CMMQ	alpha	PITCH	PITCH DAMPING DERIVATIVE

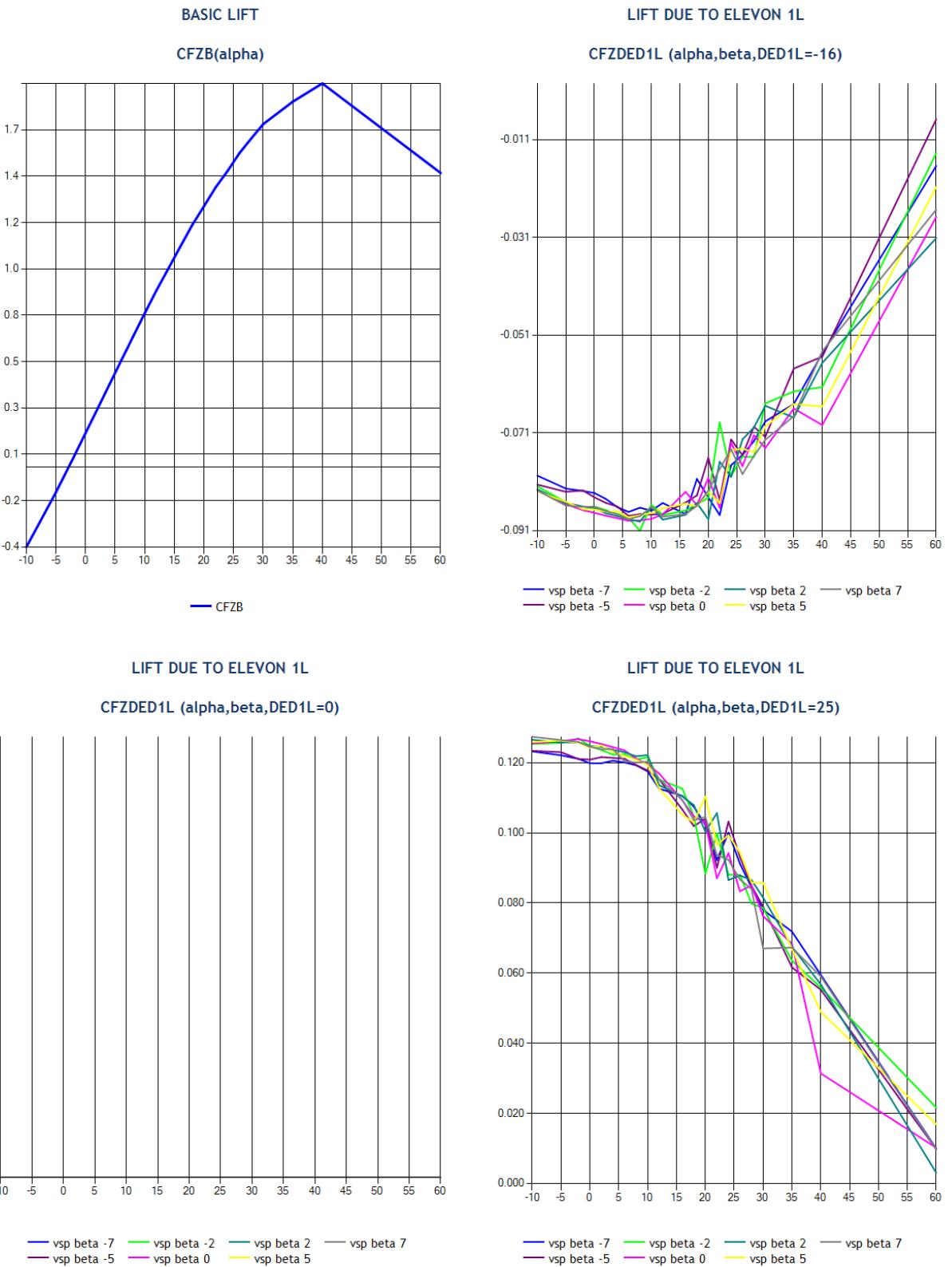
CMMmnw	mach,alpha	PITCH	PITCH DUE TO MACH
CMMDED1L	alpha,beta,DED1L	PITCH	PITCH MOMENT DUE TO ELEVON 1L
CMMDED1R	alpha,beta,DED1R	PITCH	PITCH MOMENT DUE TO ELEVON 1R
CMMDED2L	alpha,beta,DED2L	PITCH	PITCH MOMENT DUE TO ELEVON 2L
CMMDED2R	alpha,beta,DED2R	PITCH	PITCH MOMENT DUE TO ELEVON 2R
CMMDS1L	alpha	PITCH	PITCH MOMENT DUE TO LE SLAT 1
CMMDS2L	alpha	PITCH	PITCH MOMENT DUE TO LE SLAT 2
CMMDSBL	alpha	PITCH	PITCH MOMENT DUE TO LOWER SPEEDBRAKE DEFLECTION
CMMDSBU	alpha	PITCH	PITCH MOMENT DUE TO UPPER SPEEDBRAKE DEFLECTION
CMMGEAR	alpha	PITCH	PITCHING MOMENT INCREMENT DUE TO GEAR
CMMCTNK	alpha,beta	PITCH	PITCHING MOMENT INCREMENT DUE TO TANK(CENTRE)
CMLTNK	alpha,beta	PITCH	PITCHING MOMENT INCREMENT DUE TO TANK(LEFT WING)
CMMRTNK	alpha,beta	PITCH	PITCHING MOMENT INCREMENT DUE TO TANK(RIGHT WING)
CML1	alpha,beta	ROLL	BASIC ROLLING MOMENT
CMLP	alpha	ROLL	ROLL DAMPING DERIVATIVE
CMLmnw	mach,alpha	ROLL	ROLL DUE TO MACH
CMLDED1L	alpha,beta,DED1L	ROLL	ROLLING MOMENT DUE TO ELEVON 1L DEFLECTION
CMLDED1R	alpha,beta,DED1R	ROLL	ROLLING MOMENT DUE TO ELEVON 1R DEFLECTION
CMLDED2L	alpha,beta,DED2L	ROLL	ROLLING MOMENT DUE TO ELEVON 2L DEFLECTION
CMLDED2R	alpha,beta,DED2R	ROLL	ROLLING MOMENT DUE TO ELEVON 2R DEFLECTION
CMLRD	alpha,beta,DRD	ROLL	ROLLING MOMENT DUE TO RUDDER DEFLECTION
CMLR	alpha	ROLL	ROLLING MOMENT DUE TO YAW RATE
CMLGEAR	alpha	ROLL	ROLLING MOMENT INCREMENT DUE TO GEAR
CMLCTNK	alpha,beta	ROLL	ROLLING MOMENT INCREMENT DUE TO TANK(CENTRE)
CMLLTNK	alpha,beta	ROLL	ROLLING MOMENT INCREMENT DUE TO TANK(LEFT WING)
CMLRTNK	alpha,beta	ROLL	ROLLING MOMENT INCREMENT DUE TO TANK(RIGHT WING)
CFYB	alpha,beta	SIDE	BASIC SIDE FORCE
CFYDED1L	alpha,beta,DED1L	SIDE	SIDE FORCE DUE TO ELEVON 1L DEFLECTION
CFYDED1R	alpha,beta,DED1R	SIDE	SIDE FORCE DUE TO ELEVON 1R DEFLECTION
CFYDED2L	alpha,beta,DED2L	SIDE	SIDE FORCE DUE TO ELEVON 2L DEFLECTION
CFYDED2R	alpha,beta,DED2R	SIDE	SIDE FORCE DUE TO ELEVON 2R DEFLECTION
CFYP	alpha	SIDE	SIDE FORCE DUE TO ROLL RATE
CFYDRD	alpha,beta,DRD	SIDE	SIDE FORCE DUE TO RUDDER DEFLECTION
CFYR	alpha	SIDE	SIDE FORCE DUE TO YAW RATE
CFYGEAR	alpha	SIDE	SIDE FORCE INCREMENT DUE TO GEAR
CFYCTNK	alpha,beta	SIDE	SIDE FORCE INCREMENT DUE TO TANK(CENTRE)
CFYLTNK	alpha,beta	SIDE	SIDE FORCE INCREMENT DUE TO TANK(LEFT WING)
CFYRTNK	alpha,beta	SIDE	SIDE FORCE INCREMENT DUE TO TANK(RIGHT WING)

CMN1	alpha,beta	YAW	BASIC YAWING MOMENT
CMNR	alpha	YAW	YAW DAMPING DERIVATIVE
CMNDED1L	alpha,beta,DED1L	YAW	YAW MOMENT DUE TO ELEVON 1L
CMNDED1R	alpha,beta,DED1R	YAW	YAW MOMENT DUE TO ELEVON 1R
CMNDED2L	alpha,beta,DED2L	YAW	YAW MOMENT DUE TO ELEVON 2L
CMNDED2R	alpha,beta,DED2R	YAW	YAW MOMENT DUE TO ELEVON 2R
CMNP	alpha	YAW	YAWING MOMENT DUE TO ROLL RATE
CMNDRDr	alpha,beta,DRD	YAW	YAWING MOMENT DUE TO RUDDER DEFLECTION
CMNGEAR	alpha	YAW	YAWING MOMENT INCREMENT DUE TO GEAR
CMNCTNK	alpha,beta	YAW	YAWING MOMENT INCREMENT DUE TO TANK(CENTRE)
CMNLTNK	alpha,beta	YAW	YAWING MOMENT INCREMENT DUE TO TANK(LEFT WING)
CMNRTNK	alpha,beta	YAW	YAWING MOMENT INCREMENT DUE TO TANK(RIGHT WING)

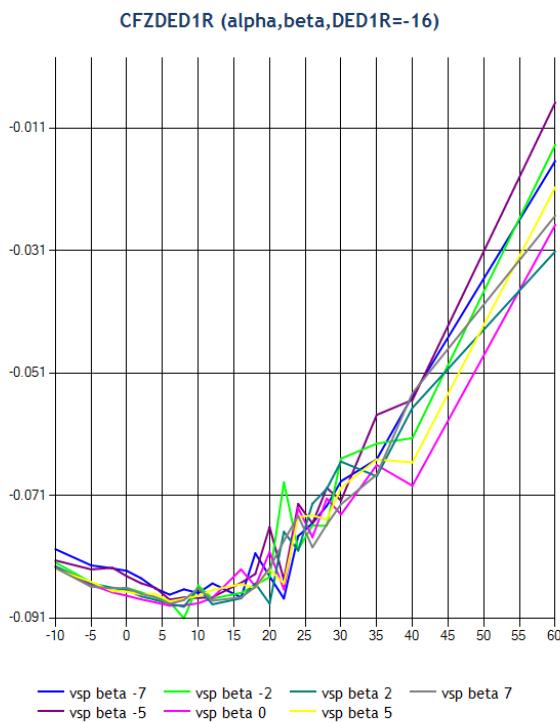
Coefficient Buildup

Axis	Buildup
DRAG	CFXDSD1L*DSD1L + CFXDSD2L*DSD2L + CFXDSBU*DSBU + CFXSBL*DSBL + CFXGEAR*gear + CFXCTNK*metrics/stores-centre-tank + CFXLTNK*metrics/stores-wing-tank-left + CFXRTNK*metrics/stores-wing-tank-right + CFXB + CFXDED1L + CFXDED1R + CFXDED2L + CFXDED2R + CFXmn
LIFT	CFZDSD1L*DSD1L + CFZDSD2L*DSD2L + CFZDSBU*DSBU + CFZDEL*DSBL + CFZGEAR*gear + CFZCTNK*metrics/stores-centre-tank + CFZLTNK*metrics/stores-wing-tank-left + CFZRTNK*metrics/stores-wing-tank-right + CFZB + CFZDED1L + CFZDED1R + CFZDE2L + CFZDE2R + CFZmn
PITCH	CMMDSD1L*DSD1L + CMMDSD2L*DSD2L + CMMDSBU*DSBU + CMMDSBL*DSBL + CMMGEAR*gear + CMMCTNK*metrics/stores-centre-tank + CMMLTNK*metrics/stores-wing-tank-left + CMMRTNK*metrics/stores-wing-tank-right + CMM1 + CMMQ*QB + CMMDED1L + CMMDED1R + CMMDED2L + CMMDED2R + CMMmnw
SIDE	CFYGEAR*gear + CFYCTNK*metrics/stores-centre-tank + CFYLTNK*metrics/stores-wing-tank-left + CFYRTNK*metrics/stores-wing-tank-right + CFYB + CFYDED1L + CFYDED1R + CFYDED2L + CFYDED2R + CFYDRD + CFYP*PB + CFYR*RB
ROLL	CMLGEAR*gear + CMLCTNK*metrics/stores-centre-tank + CMLLTNK*metrics/stores-wing-tank-left + CMLRTNK*metrics/stores-wing-tank-right + CML1 + CMLDED1L + CMLDED1R + CMLDED2L + CMLDED2R + CMLDRD + CMLP*PB + CMLR*RB + CMLmnw + (DLNB*BETA)
YAW	CMNGEAR*gear + CMNCTNK*metrics/stores-centre-tank + CMNLTNK*metrics/stores-wing-tank-left + CMNRTNK*metrics/stores-wing-tank-right + CMN1 + CMNDED1L + CMNDED1R + CMNDED2L + CMNDED2R + CMNDRDr + CMNP*PB + CMNR*RB + (DCNB*BETA)

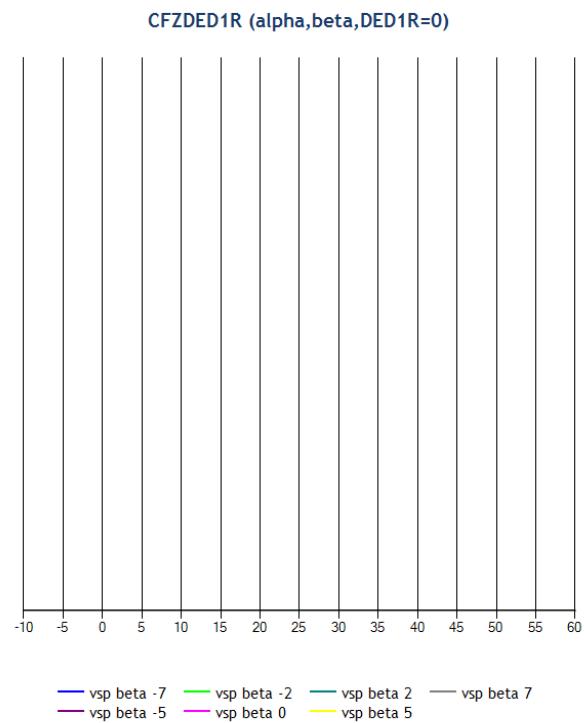
LIFT



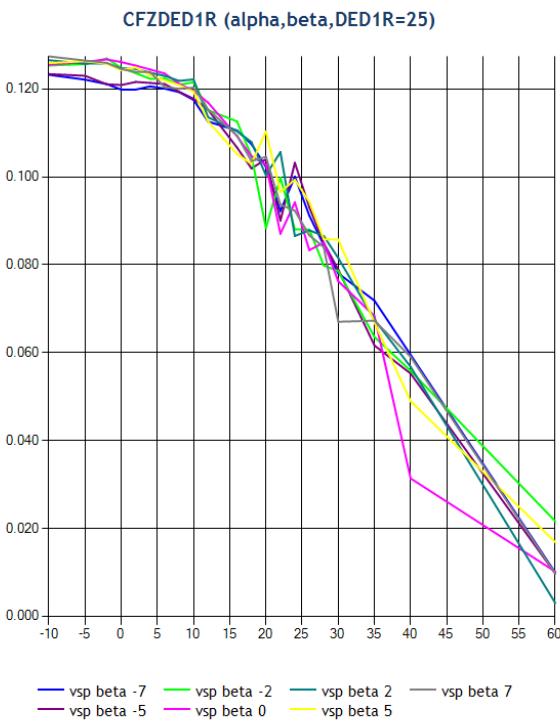
LIFT DUE TO ELEVON 1R



LIFT DUE TO ELEVON 1R



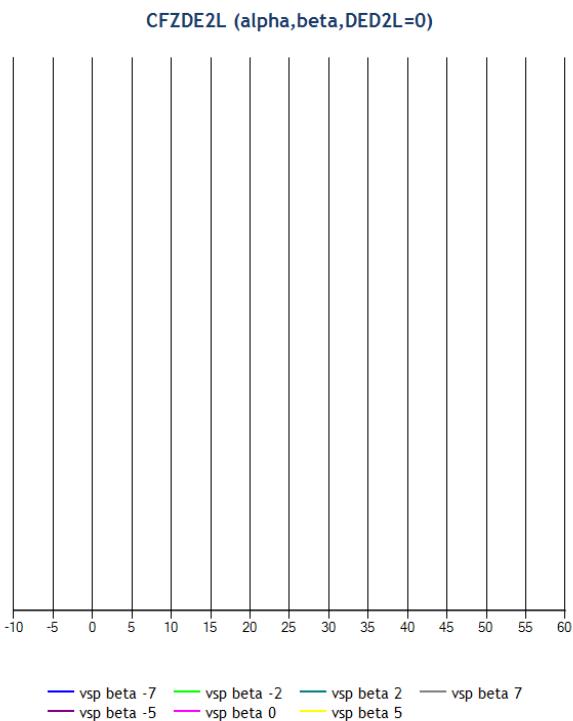
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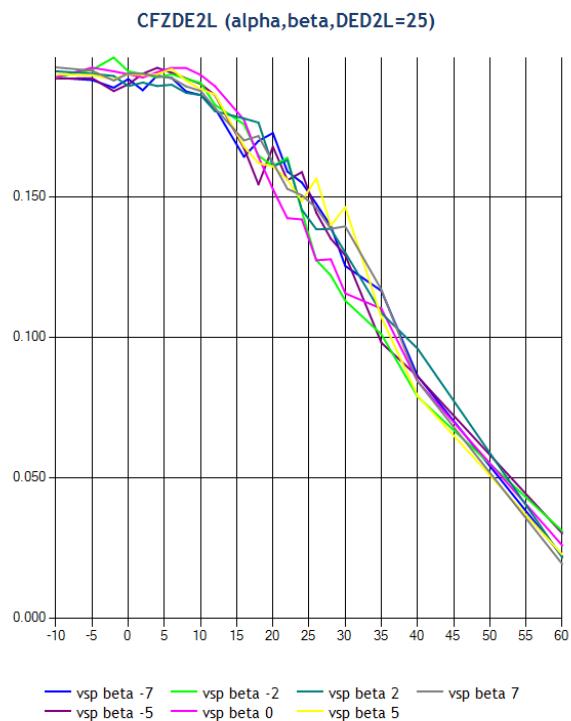
LIFT DUE TO ELEVON 2L



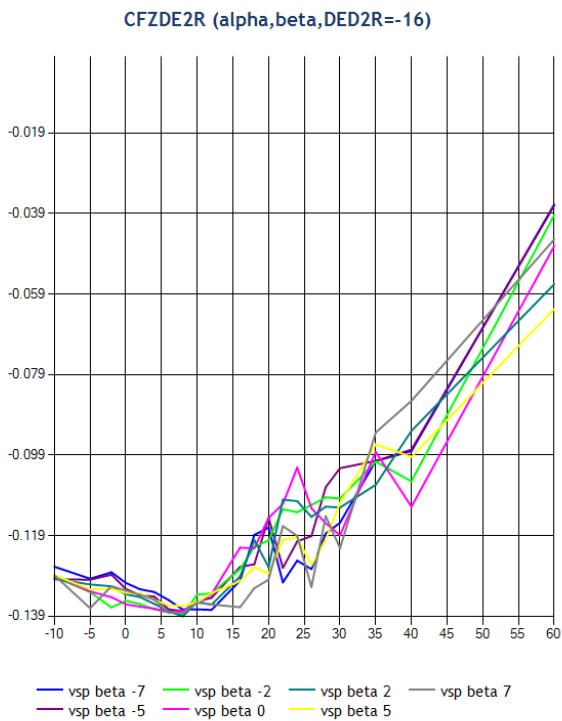
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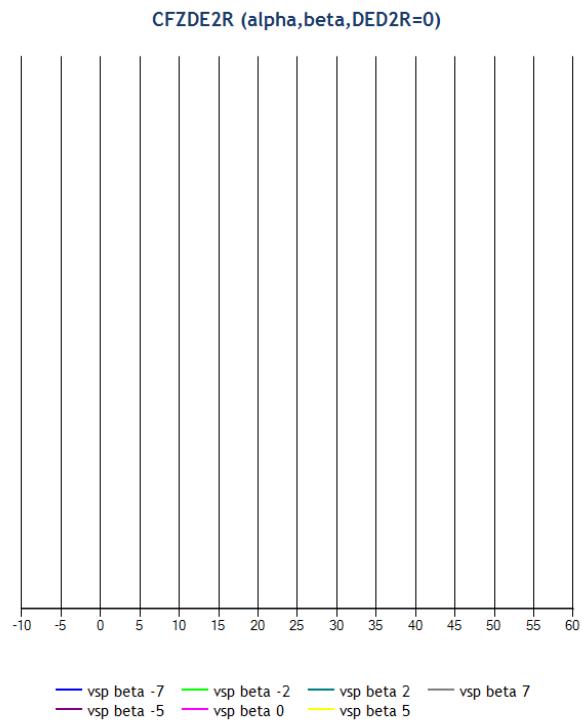
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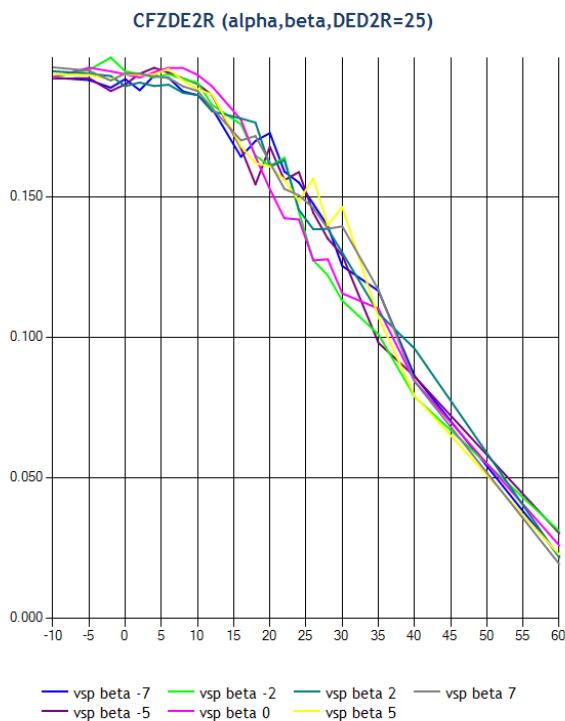
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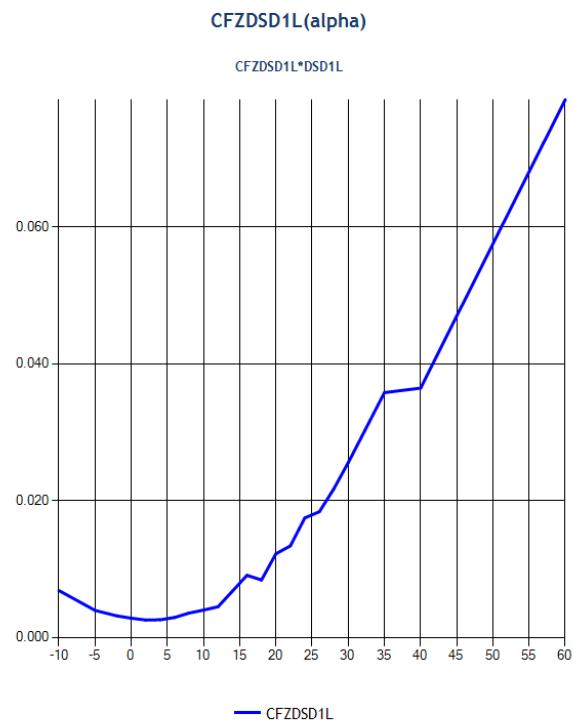
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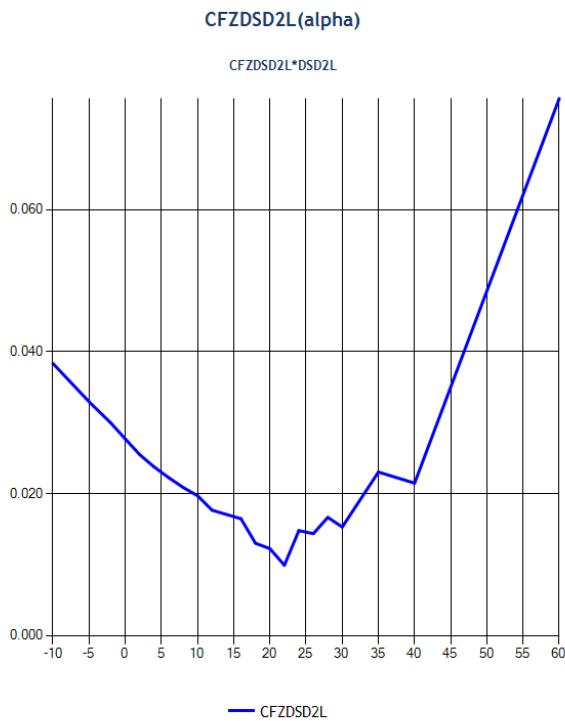
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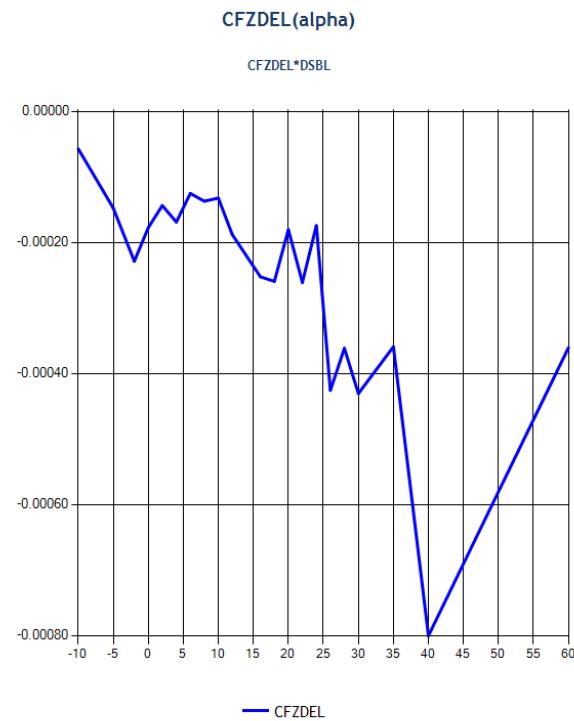
LIFT DUE TO LE SLAT 1



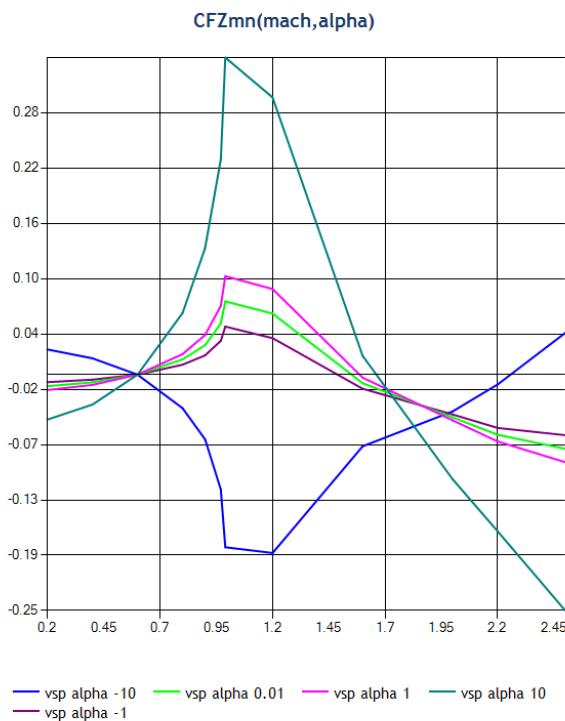
LIFT DUE TO LE SLAT 2



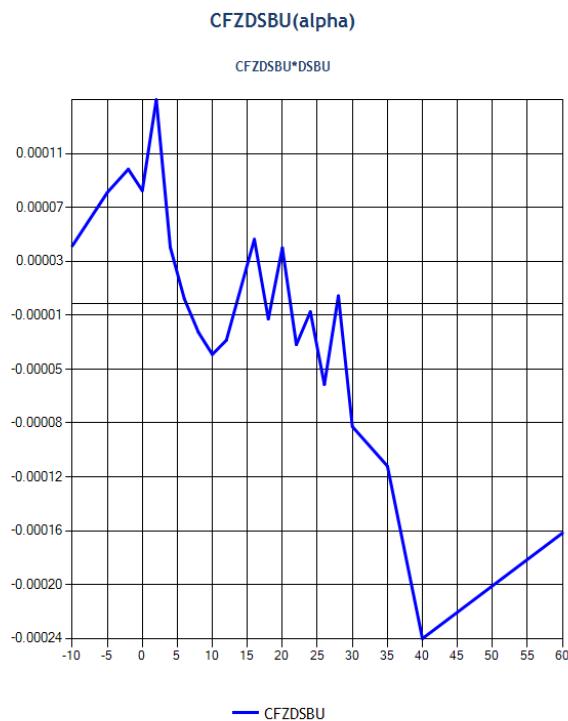
LIFT DUE TO LOWER SPEEDBRAKE DEFLECTION



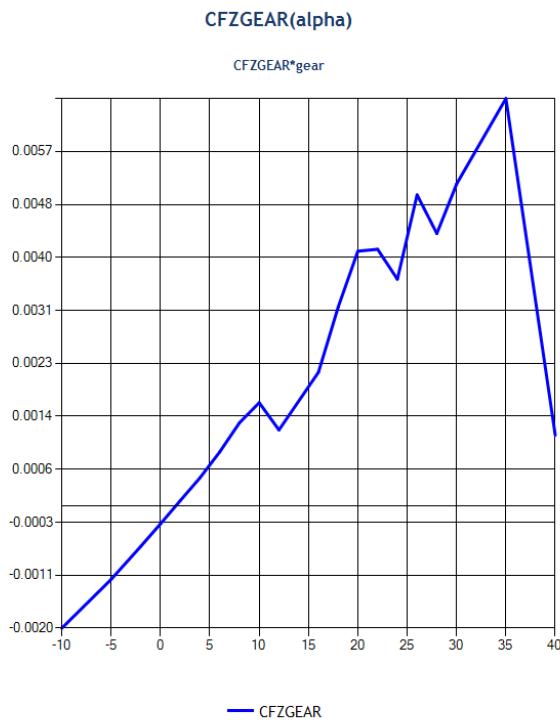
LIFT DUE TO MACH



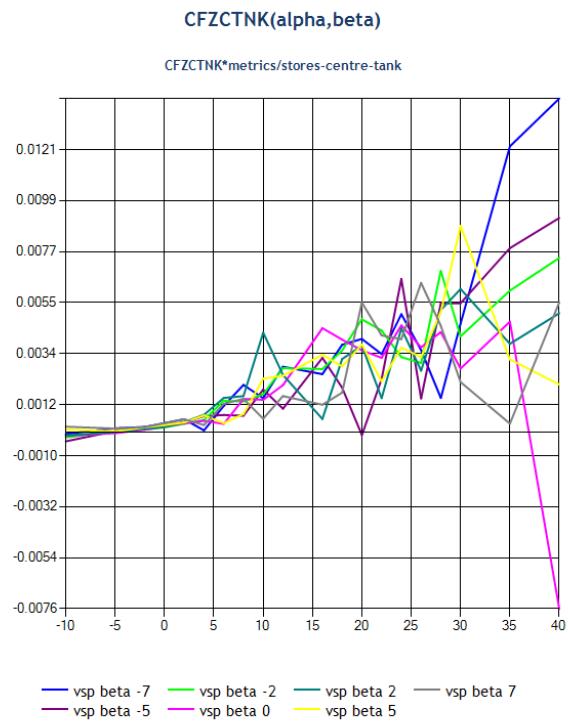
LIFT DUE TO UPPER SPEEDBRAKE DEFLECTION



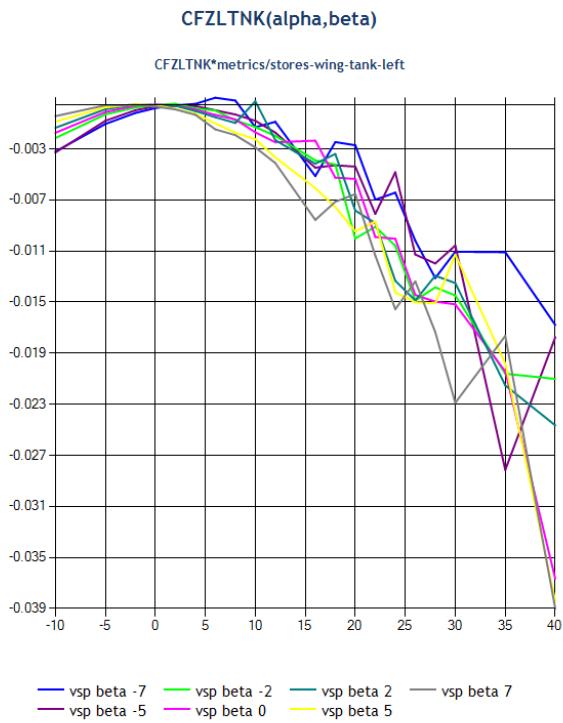
LIFT INCREMENT DUE TO GEAR



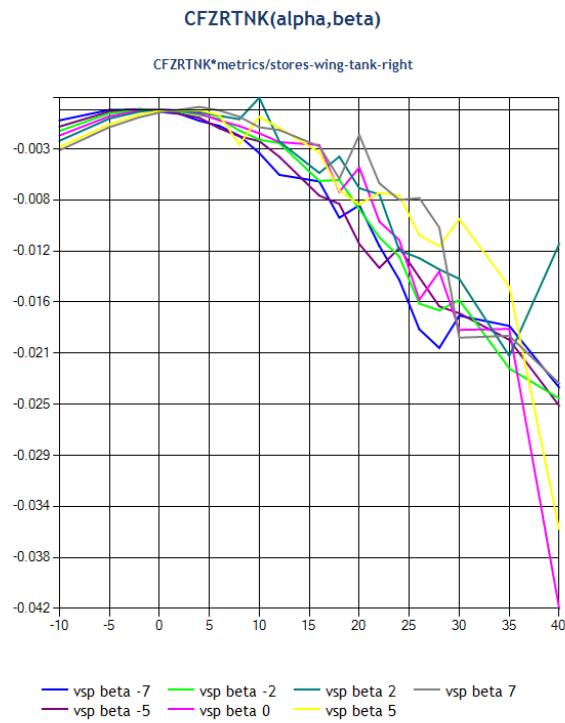
LIFT INCREMENT DUE TO TANK(CENTRE)



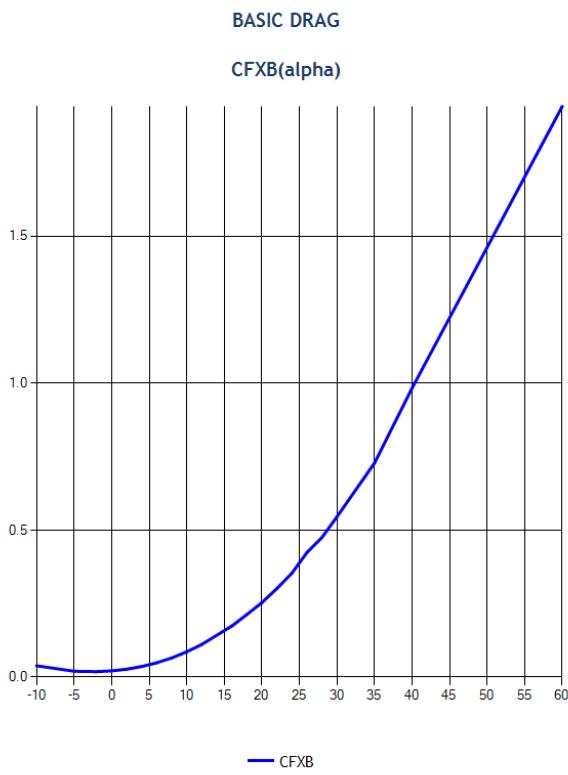
LIFT INCREMENT DUE TO TANK(LEFT WING)



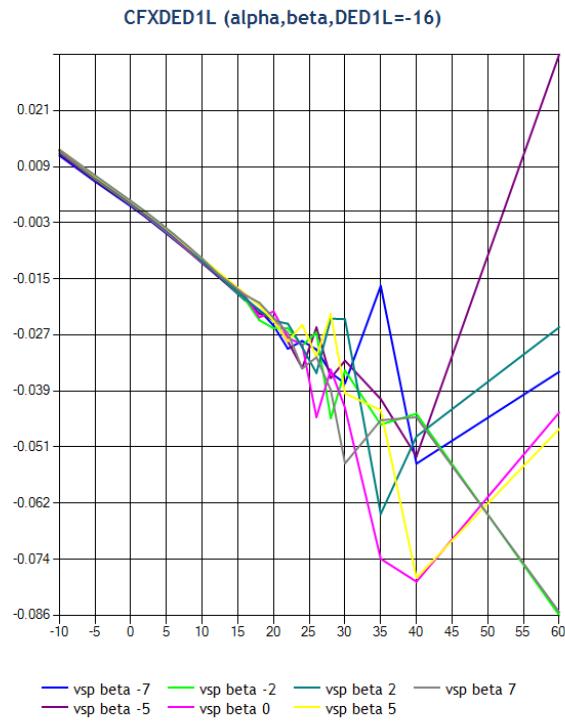
LIFT INCREMENT DUE TO TANK(RIGHT WING)



DRAG



DRAG DUE TO ELEVON 1L

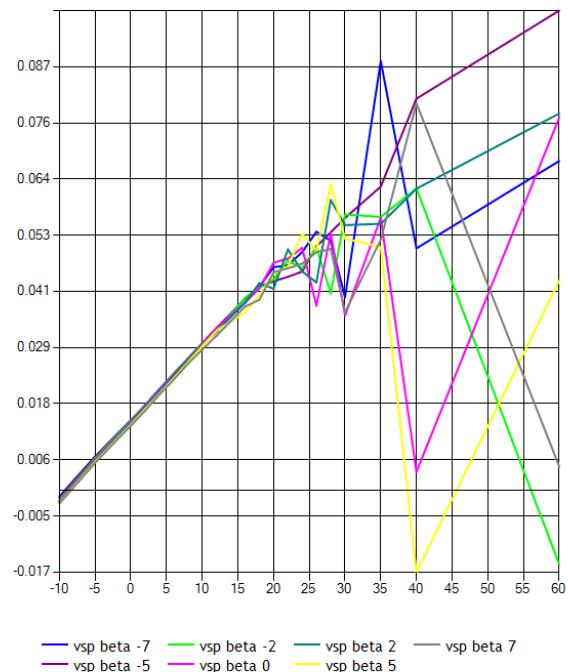
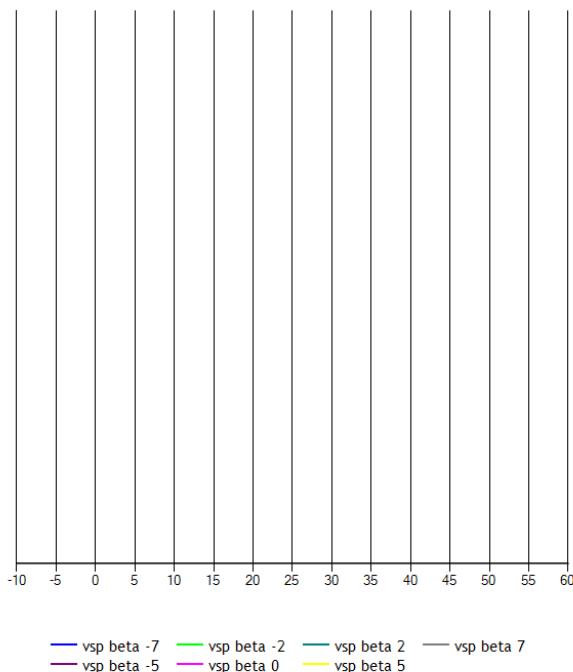


DRAG DUE TO ELEVON 1L

DRAG DUE TO ELEVON 1L

CFXDED1L (alpha,beta,DED1L=0)

CFXDED1L (alpha,beta,DED1L=25)

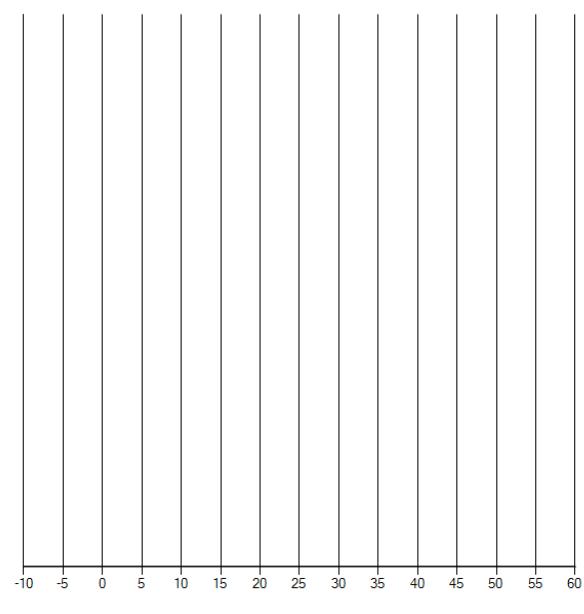
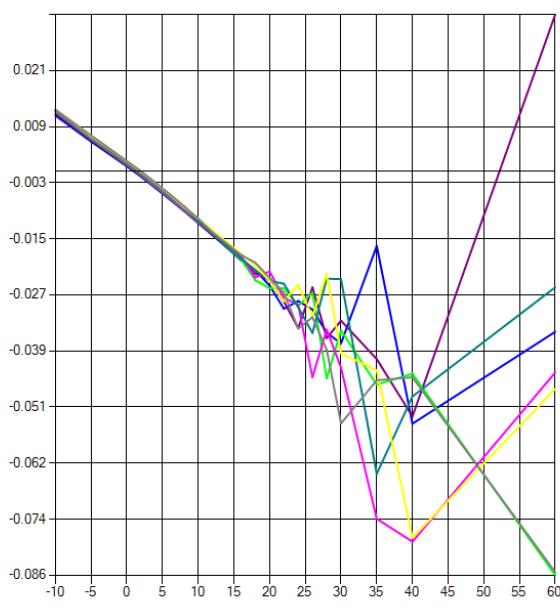


DRAG DUE TO ELEVON 1R

DRAG DUE TO ELEVON 1R

CFXDED1R (alpha,beta,DED1R=-16)

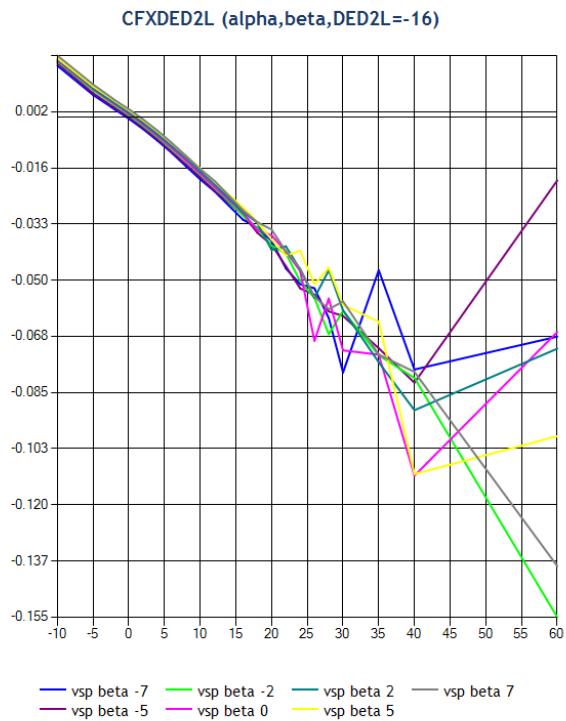
CFXDED1R (alpha,beta,DED1R=0)



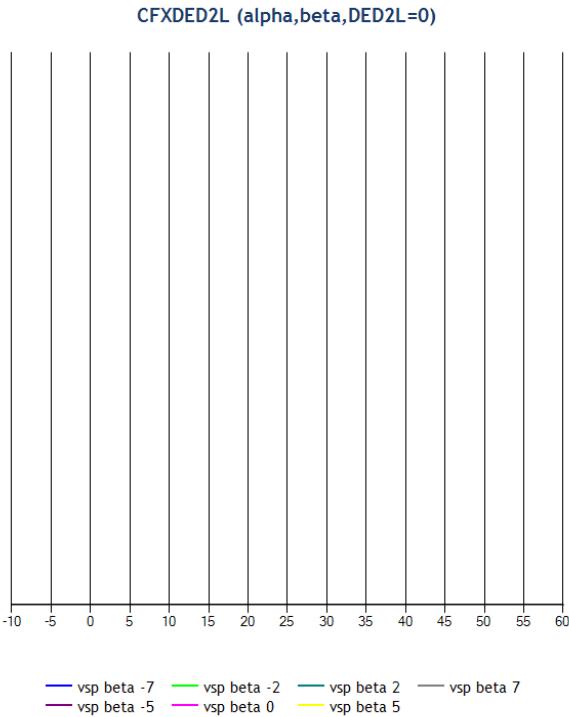
DRAG DUE TO ELEVON 1R



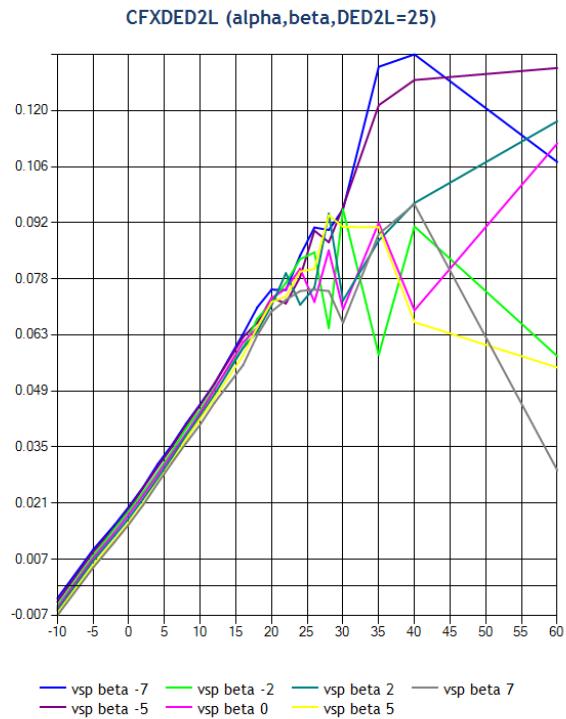
DRAG DUE TO ELEVON 2L



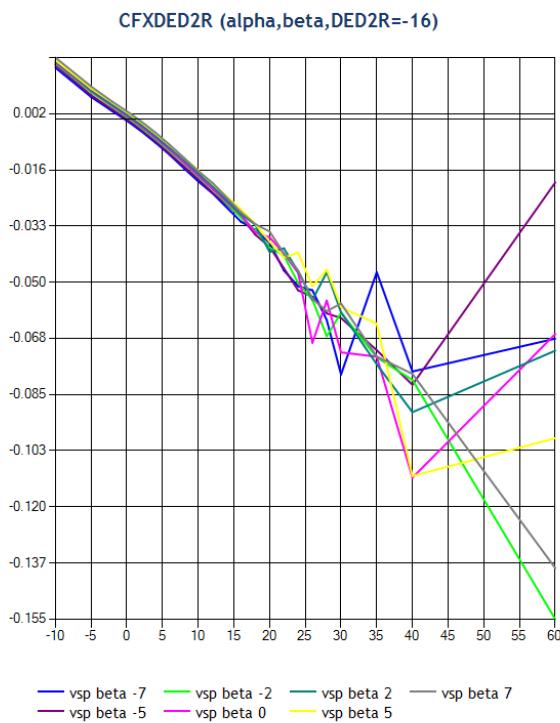
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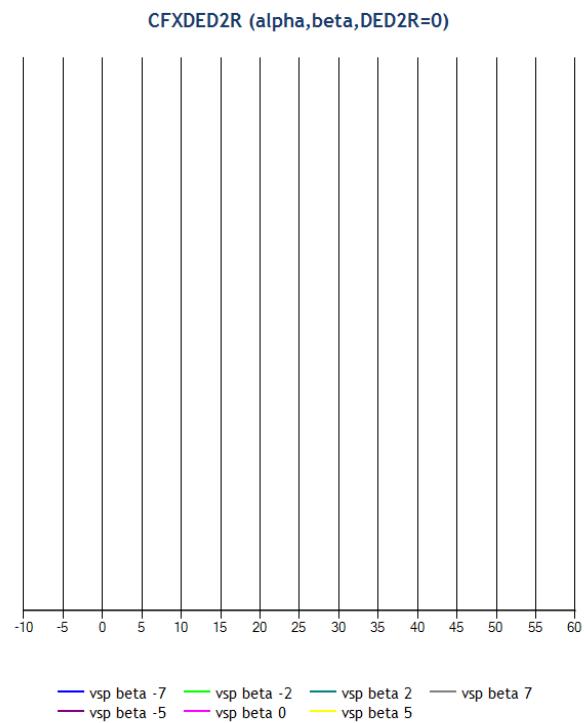
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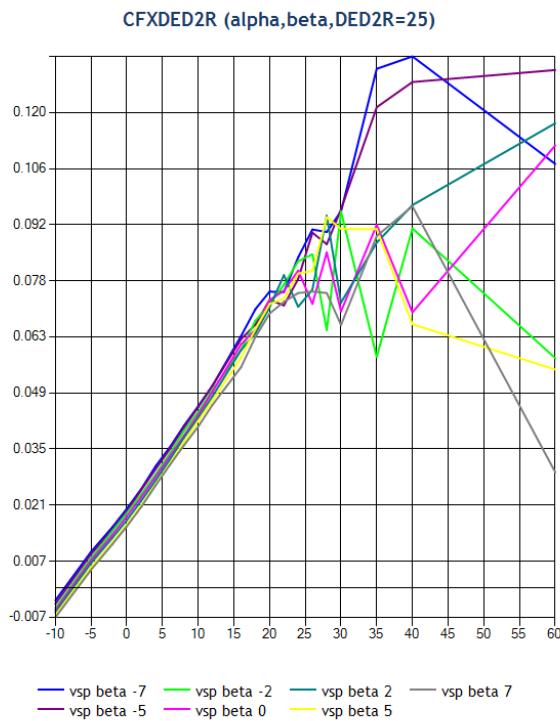
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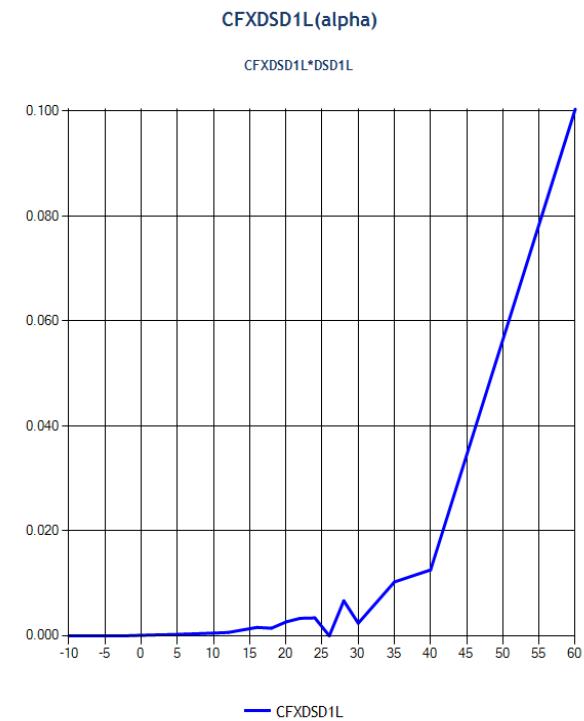
DRAG DUE TO ELEVON 2R



DRAG DUE TO ELEVON 2R



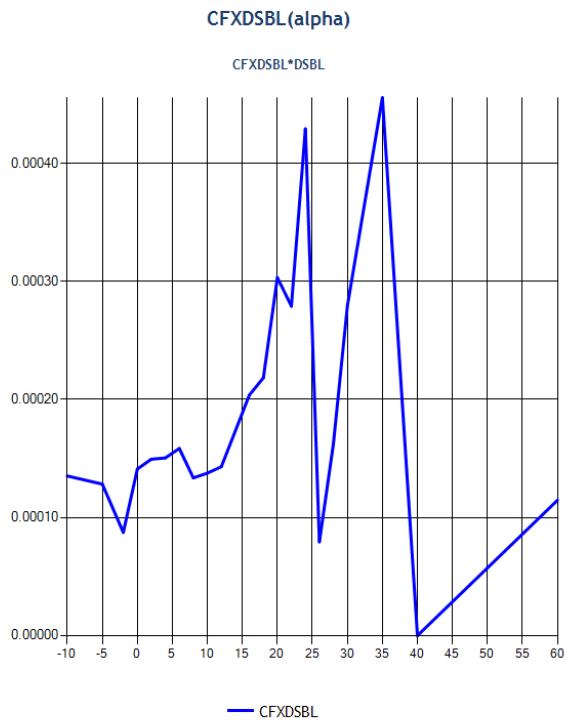
DRAG DUE TO LE SLAT 1



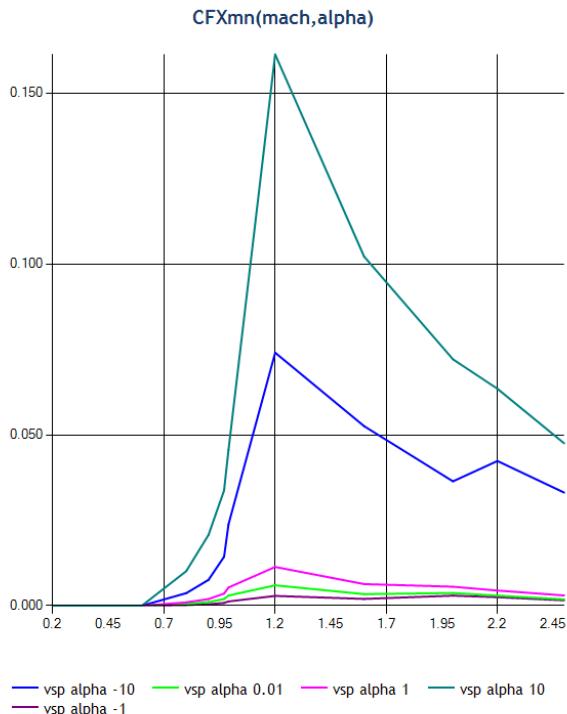
DRAG DUE TO LE SLAT 2



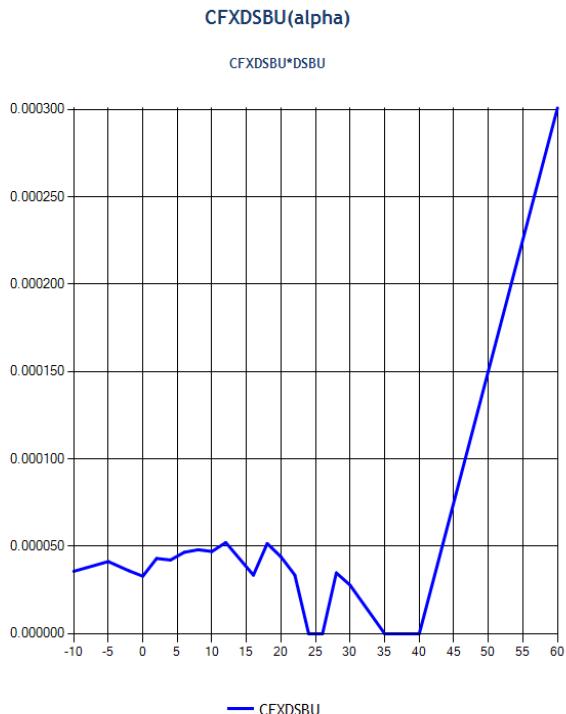
DRAG DUE TO LOWER SPEEDBRAKE DEFLECTION



DRAG DUE TO MACH



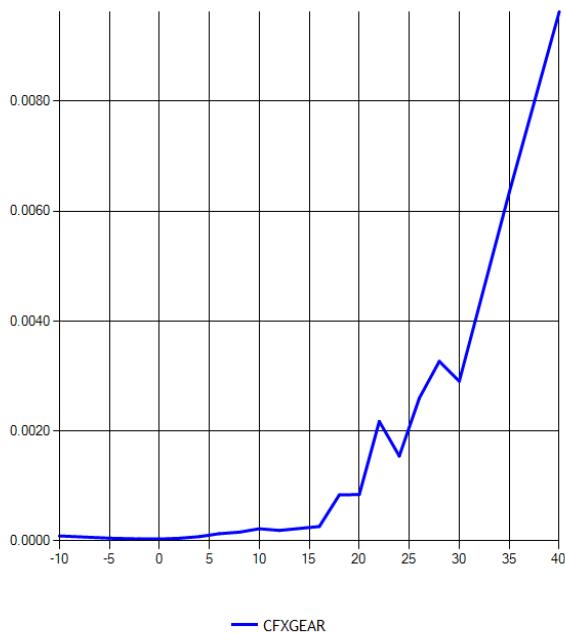
DRAG DUE TO UPPER SPEEDBRAKE DEFLECTION



DRAG INCREMENT DUE TO GEAR

CFXGEAR(alpha)

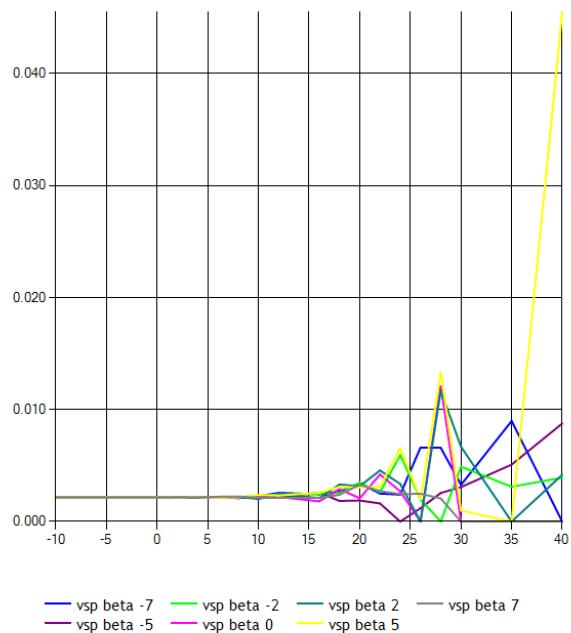
CFXGEAR*gear



DRAG INCREMENT DUE TO TANK(CENTRE)

CFXCTNK(alpha,beta)

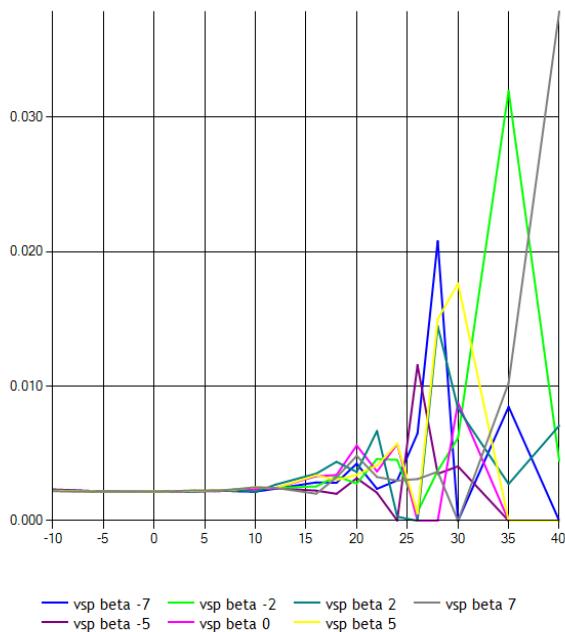
CFXCTNK*metrics/stores-centre-tank



DRAG INCREMENT DUE TO TANK(LEFT WING)

CFXLTNK(alpha,beta)

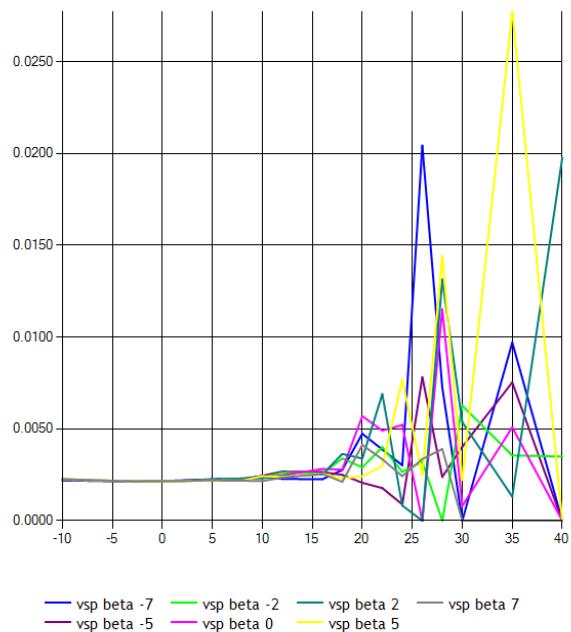
CFXLTNK*metrics/stores-wing-tank-left



DRAG INCREMENT DUE TO TANK(RIGHT WING)

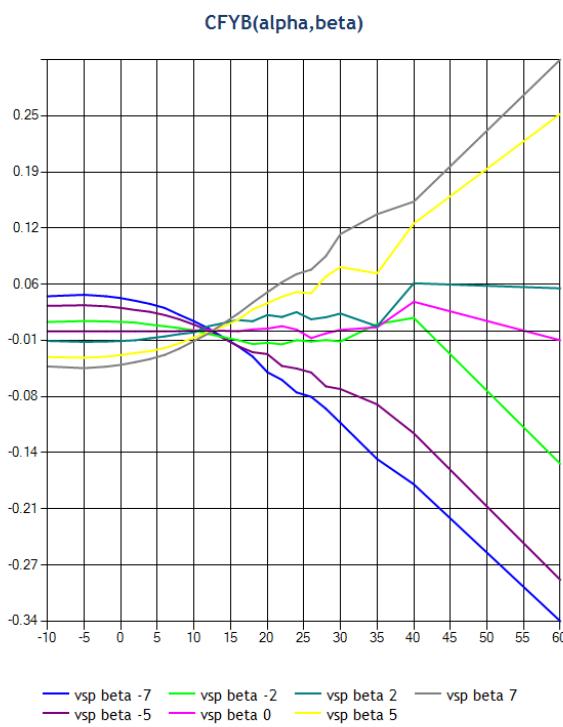
CFXRTNK(alpha,beta)

CFXRTNK*metrics/stores-wing-tank-right

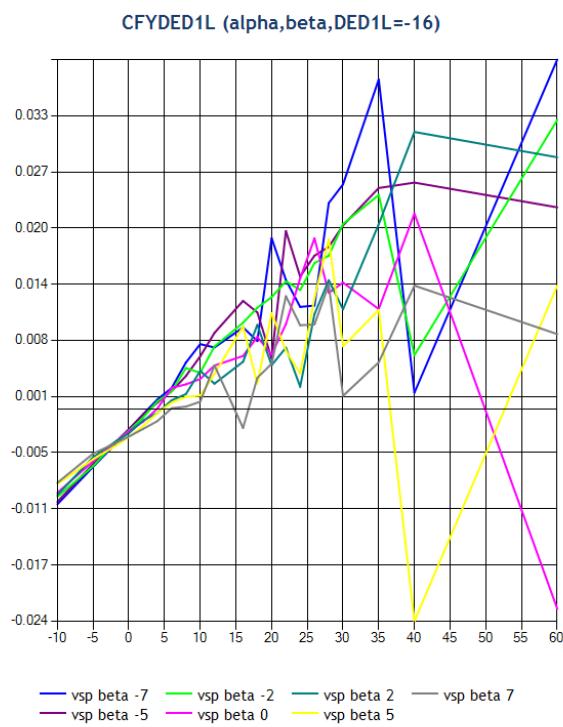


SIDE

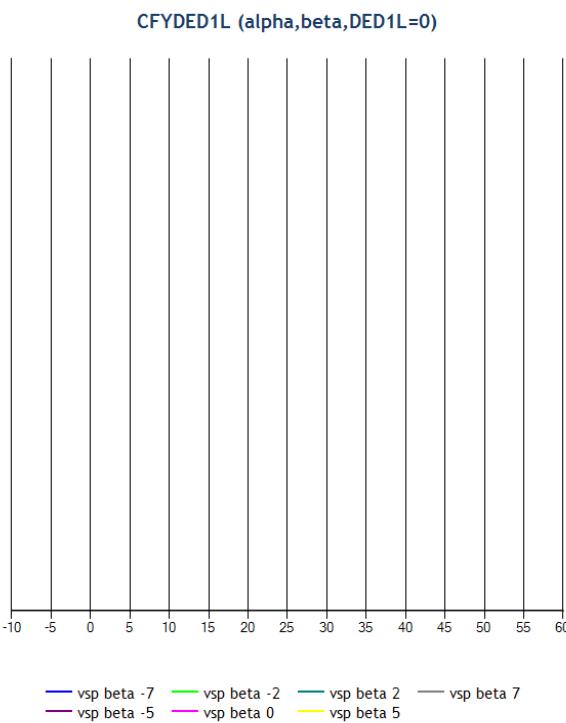
BASIC SIDE FORCE



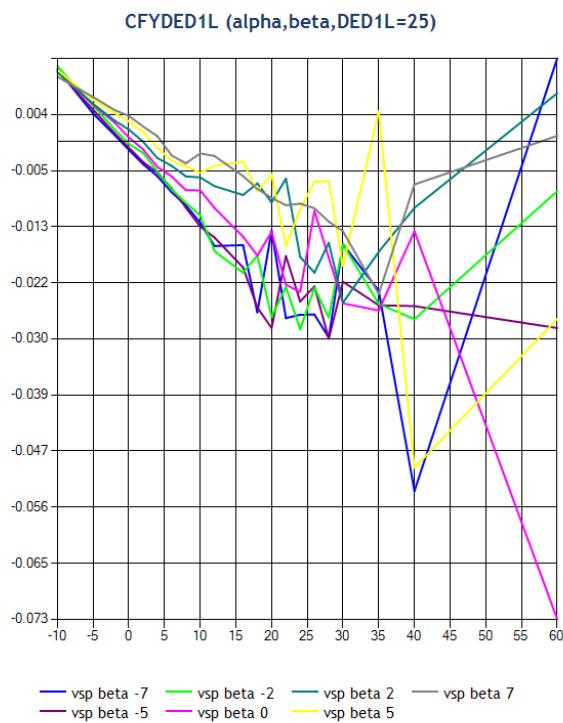
SIDE FORCE DUE TO ELEVON 1L DEFLECTION



SIDE FORCE DUE TO ELEVON 1L DEFLECTION

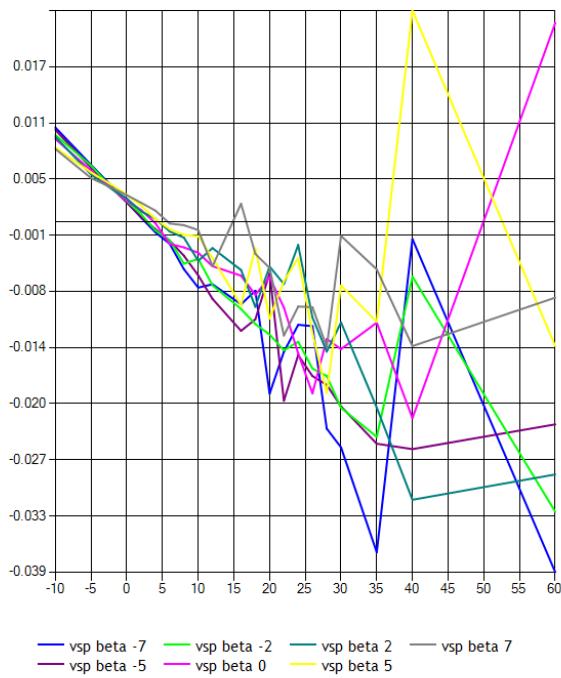


SIDE FORCE DUE TO ELEVON 1L DEFLECTION



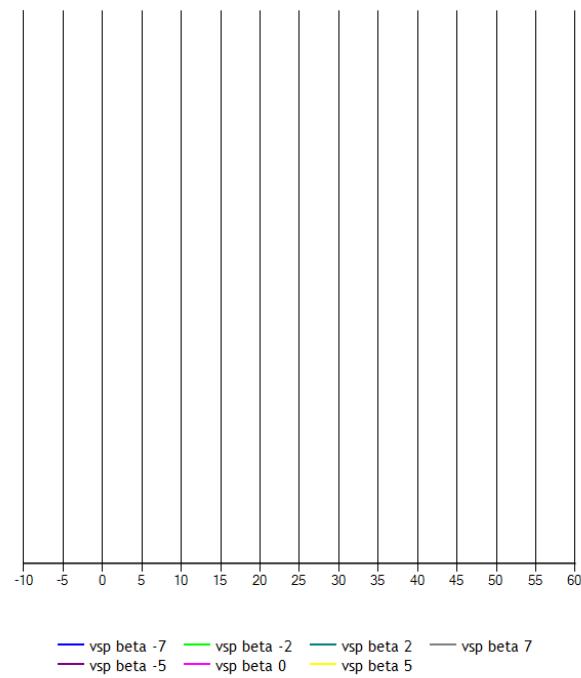
SIDE FORCE DUE TO ELEVON 1R DEFLECTION

CFYDED1R (alpha,beta,DED1R=-16)



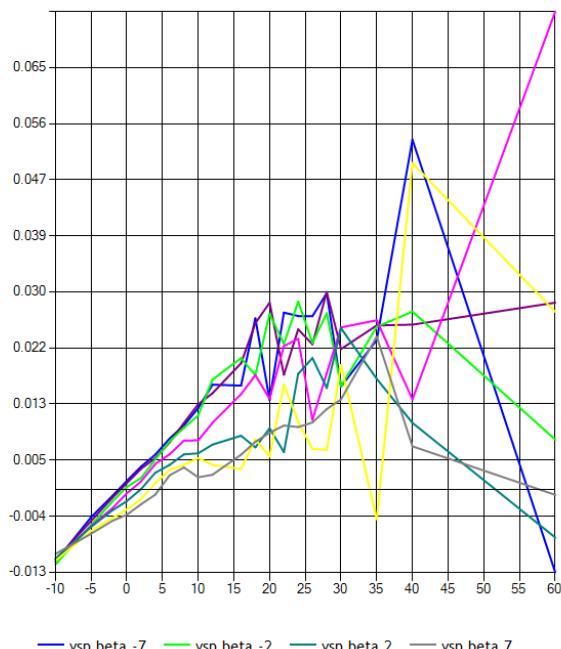
SIDE FORCE DUE TO ELEVON 1R DEFLECTION

CFYDED1R (alpha,beta,DED1R=0)



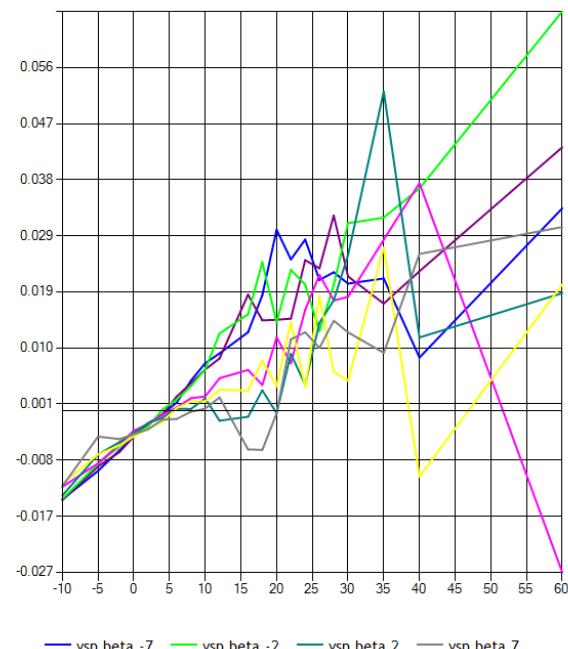
SIDE FORCE DUE TO ELEVON 1R DEFLECTION

CFYDED1R (alpha,beta,DED1R=25)



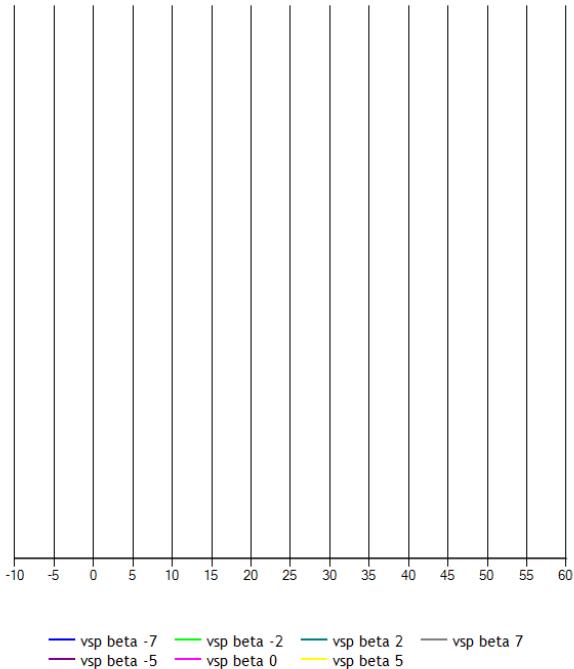
SIDE FORCE DUE TO ELEVON 2L DEFLECTION

CFYDED2L (alpha,beta,DED2L=-16)



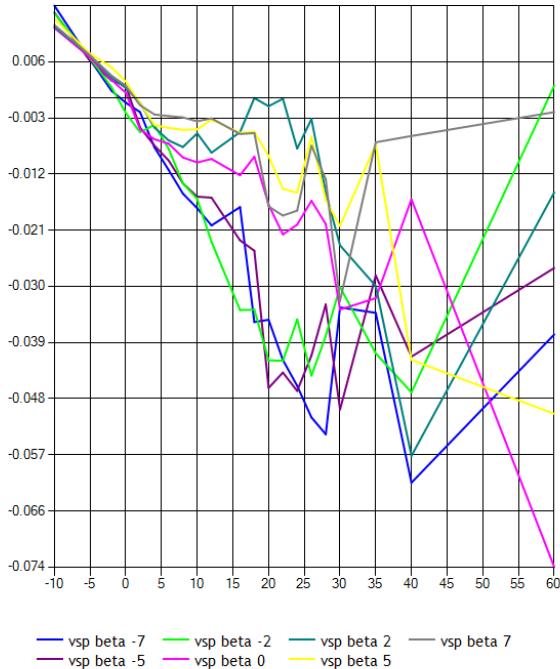
SIDE FORCE DUE TO ELEVON 2L DEFLECTION

CFYDED2L (alpha,beta,DED2L=0)



SIDE FORCE DUE TO ELEVON 2L DEFLECTION

CFYDED2L (alpha,beta,DED2L=25)



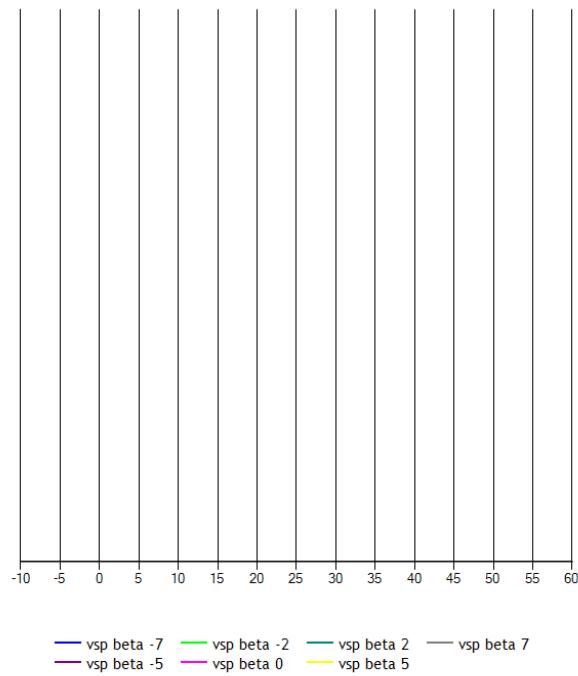
SIDE FORCE DUE TO ELEVON 2R DEFLECTION

CFYDED2R (alpha,beta,DED2R=-16)

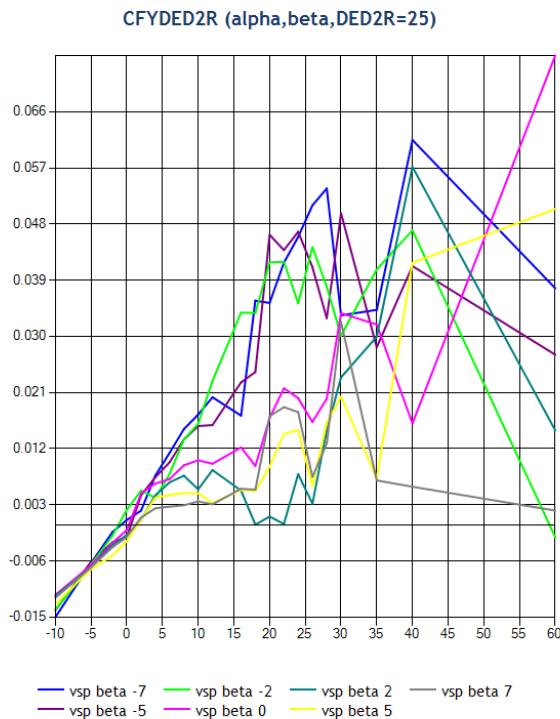


SIDE FORCE DUE TO ELEVON 2R DEFLECTION

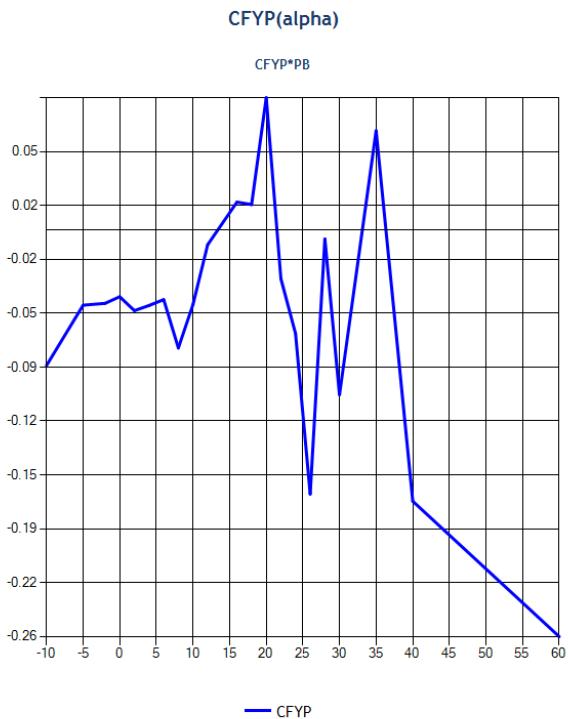
CFYDED2R (alpha,beta,DED2R=0)



SIDE FORCE DUE TO ELEVON 2R DEFLECTION



SIDE FORCE DUE TO ROLL RATE



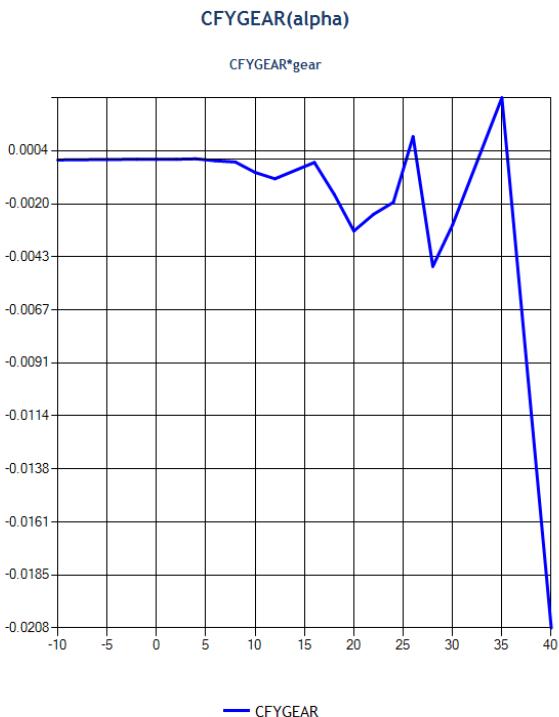
SIDE FORCE DUE TO RUDDER DEFLECTION



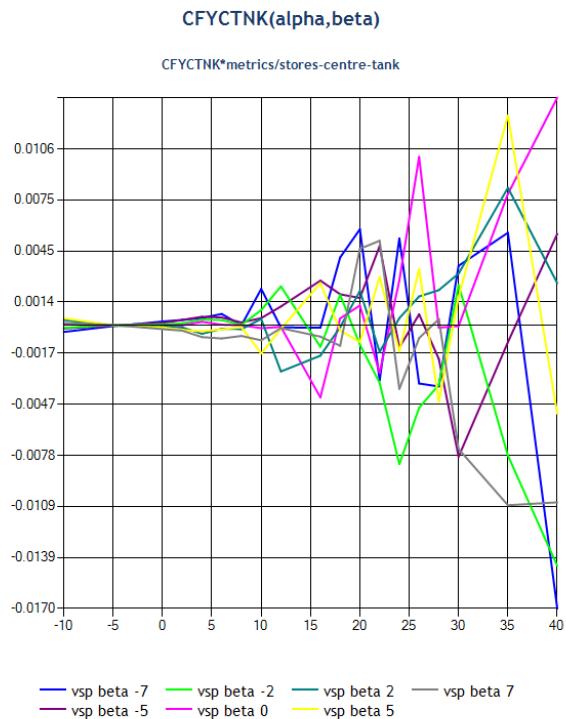
SIDE FORCE DUE TO YAW RATE



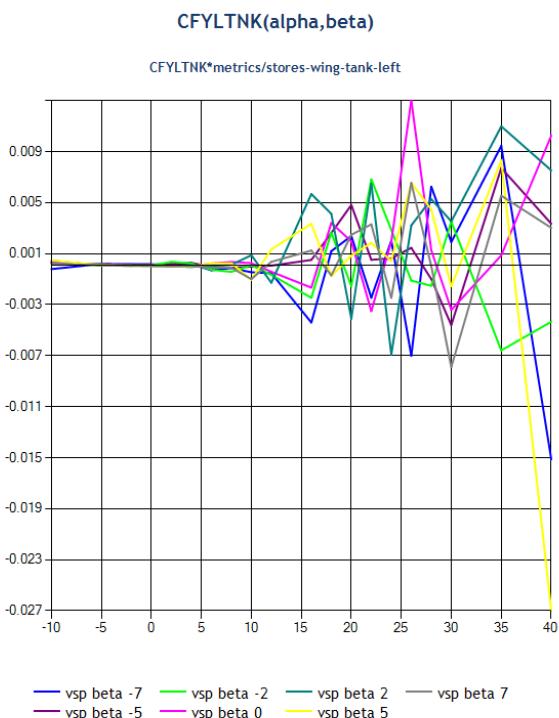
SIDE FORCE INCREMENT DUE TO GEAR



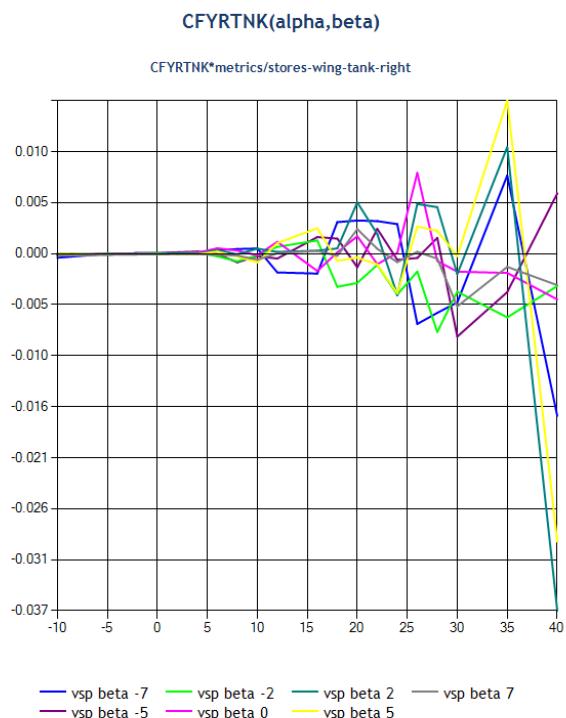
SIDE FORCE INCREMENT DUE TO TANK(CENTRE)



SIDE FORCE INCREMENT DUE TO TANK(LEFT WING)

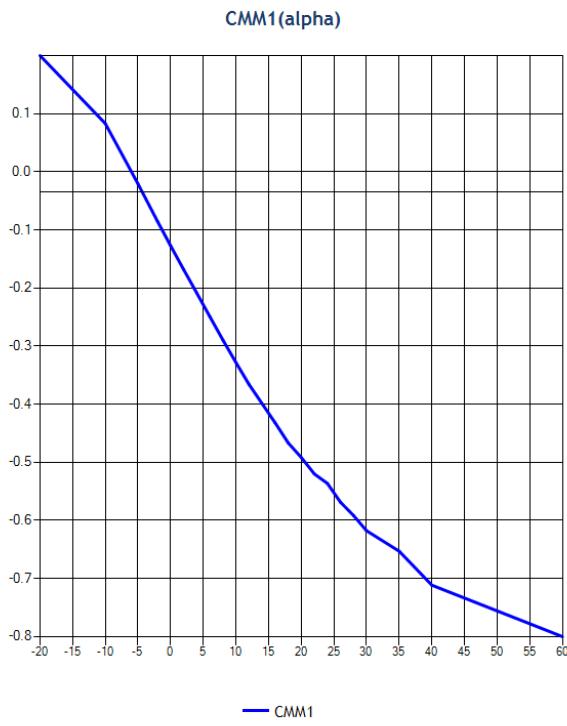


SIDE FORCE INCREMENT DUE TO TANK(RIGHT WING)

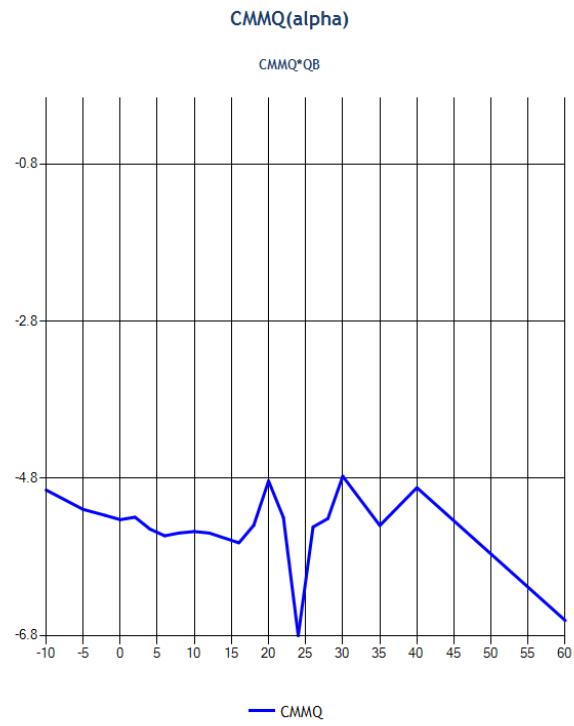


PITCH

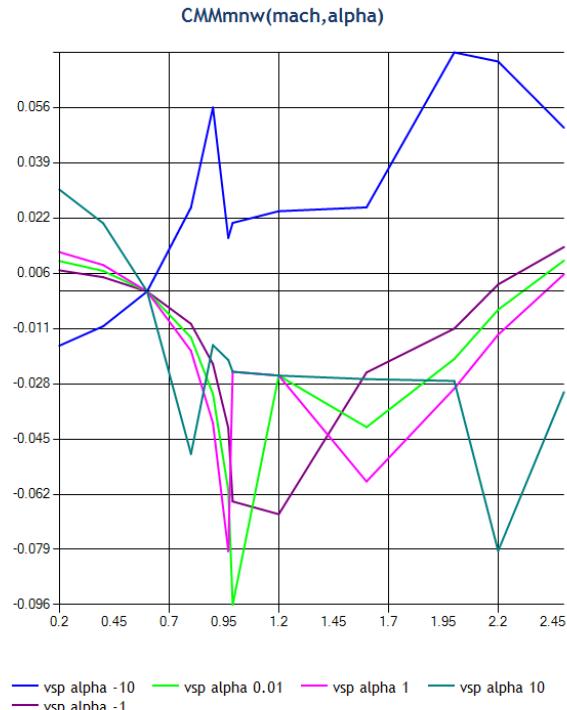
BASIC PITCHING MOMENT



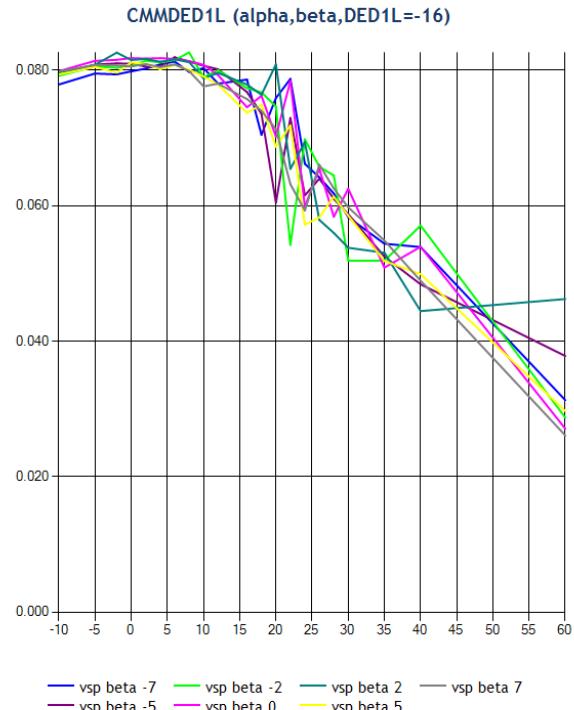
PITCH DAMPING DERIVATIVE



PITCH DUE TO MACH

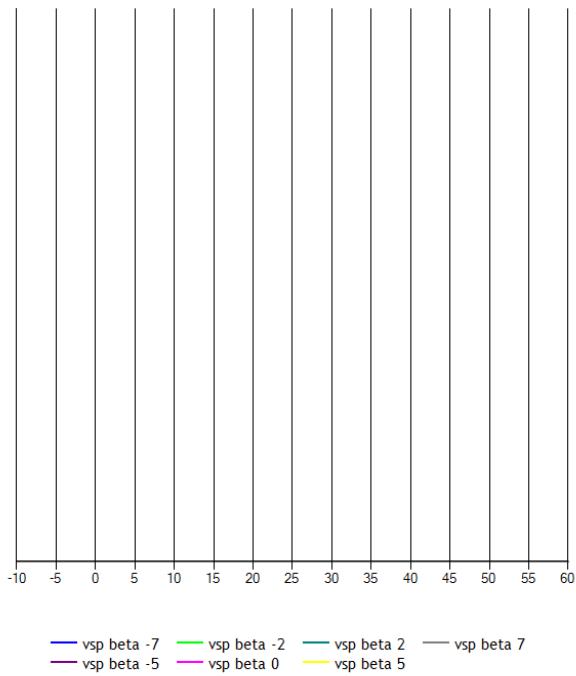


PITCH MOMENT DUE TO ELEVON 1L



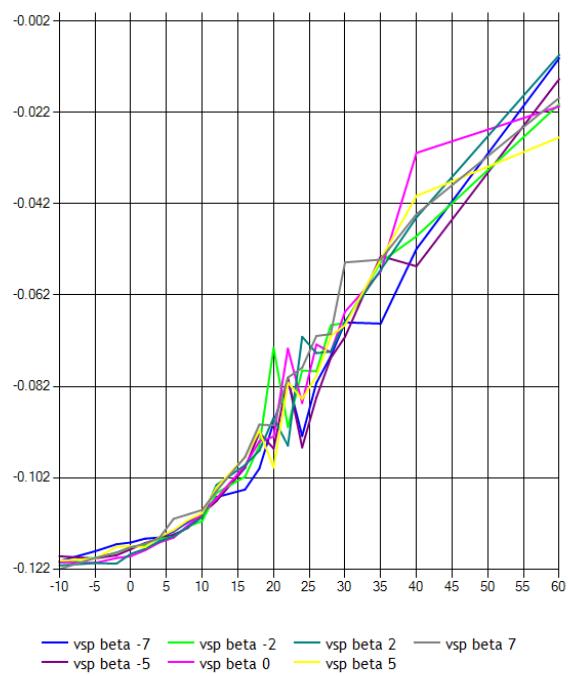
PITCH MOMENT DUE TO ELEVON 1L

CMMDED1L (alpha,beta,DED1L=0)



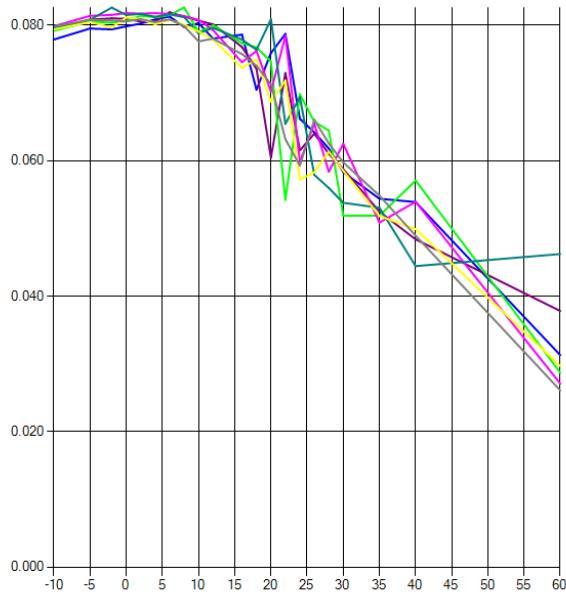
PITCH MOMENT DUE TO ELEVON 1L

CMMDED1L (alpha,beta,DED1L=25)



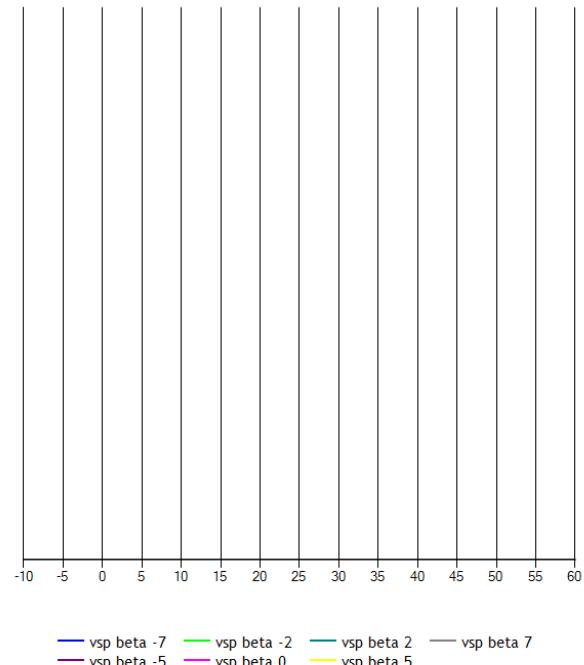
PITCH MOMENT DUE TO ELEVON 1R

CMMDED1R (alpha,beta,DED1R=-16)



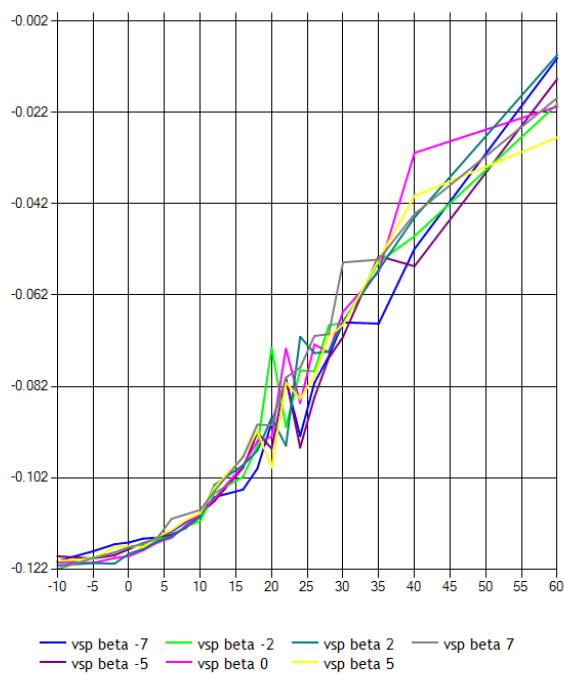
PITCH MOMENT DUE TO ELEVON 1R

CMMDED1R (alpha,beta,DED1R=0)



PITCH MOMENT DUE TO ELEVON 1R

CMMDED1R (alpha,beta,DED1R=25)



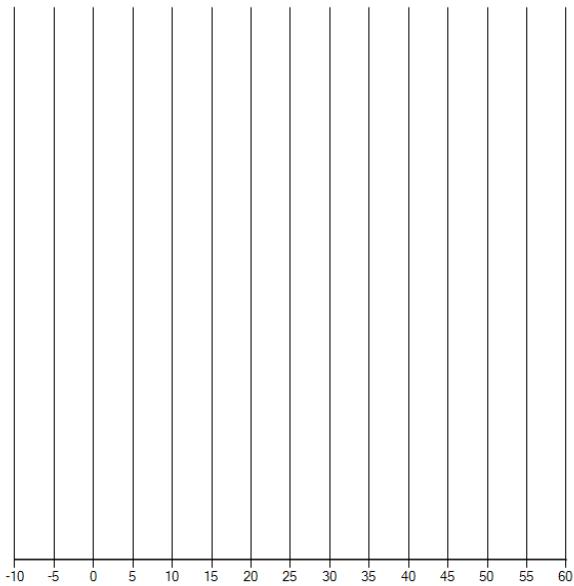
PITCH MOMENT DUE TO ELEVON 2L

CMMDED2L (alpha,beta,DED2L=-16)



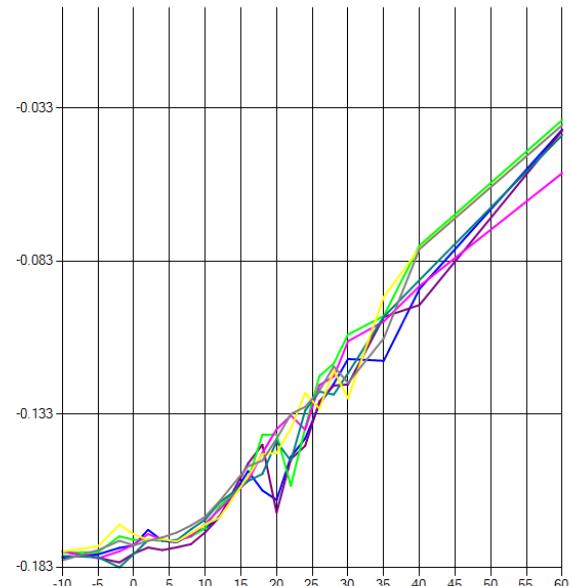
PITCH MOMENT DUE TO ELEVON 2L

CMMDED2L (alpha,beta,DED2L=0)

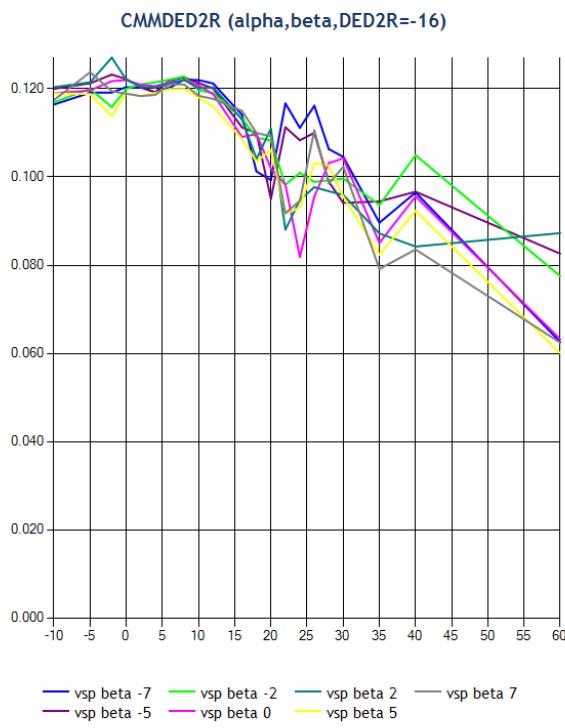


PITCH MOMENT DUE TO ELEVON 2L

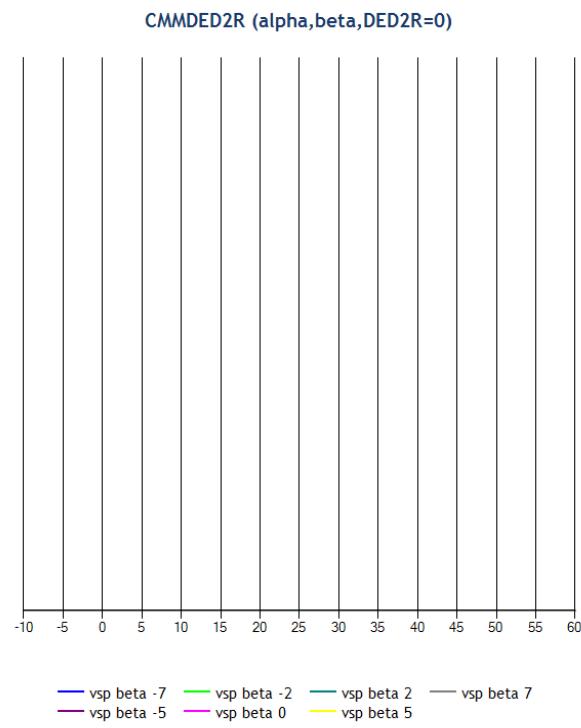
CMMDED2L (alpha,beta,DED2L=25)



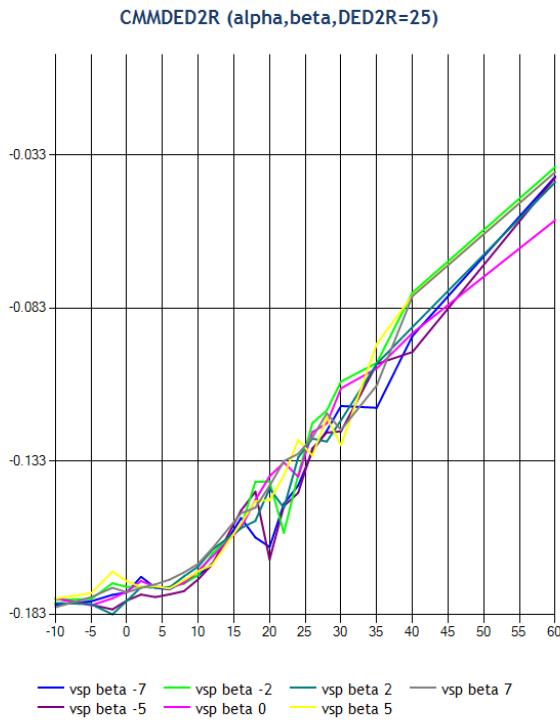
PITCH MOMENT DUE TO ELEVON 2R



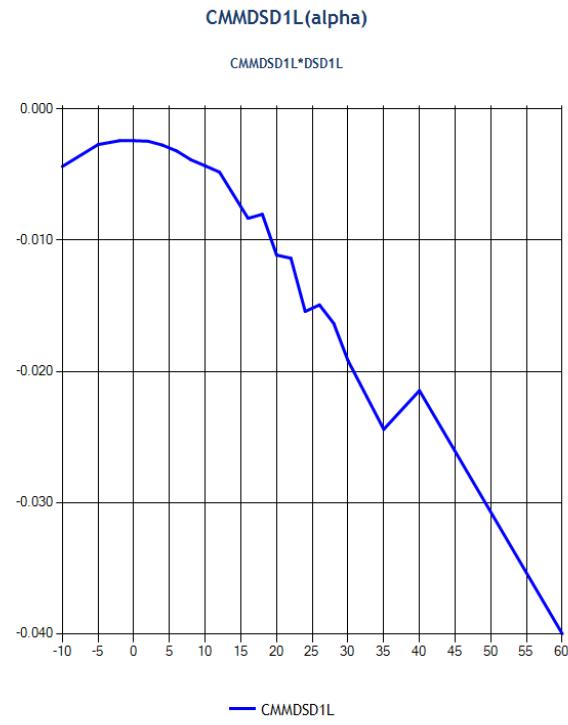
PITCH MOMENT DUE TO ELEVON 2R



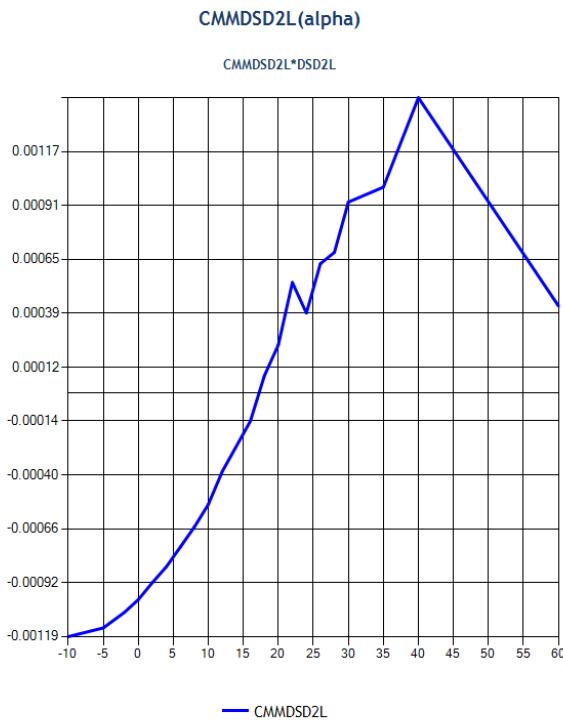
PITCH MOMENT DUE TO ELEVON 2R



PITCH MOMENT DUE TO LE SLAT 1



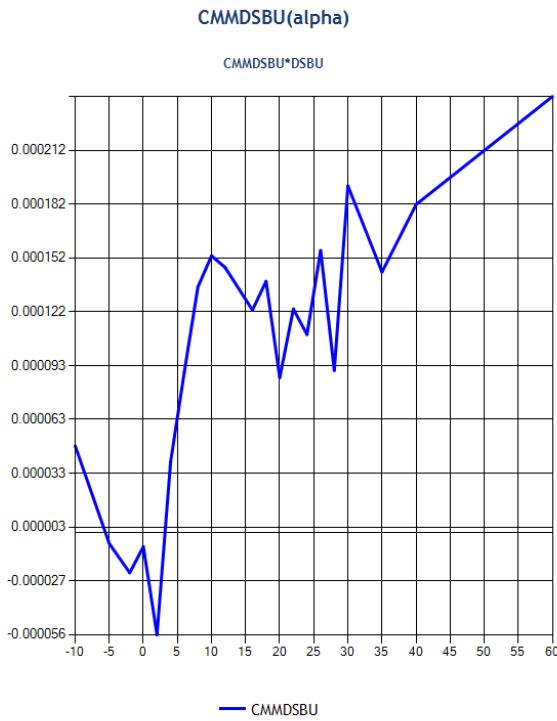
PITCH MOMENT DUE TO LE SLAT 2



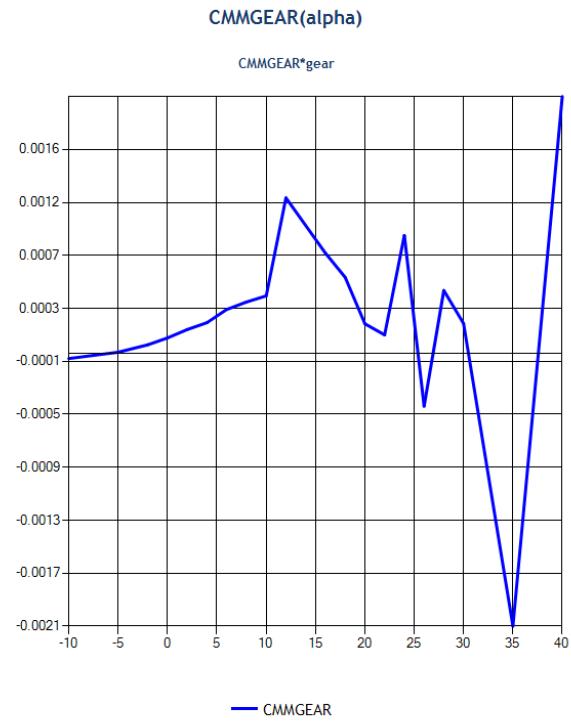
PITCH MOMENT DUE TO LOWER SPEEDBRAKE DEFLECTION



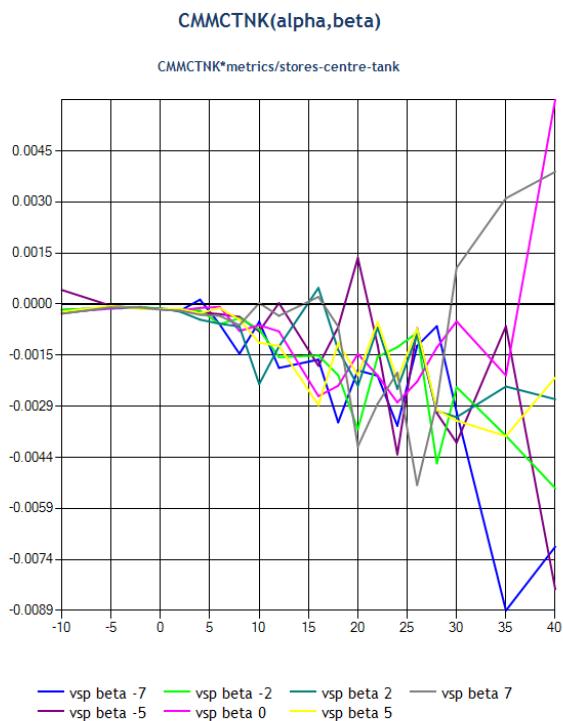
PITCH MOMENT DUE TO UPPER SPEEDBRAKE DEFLECTION



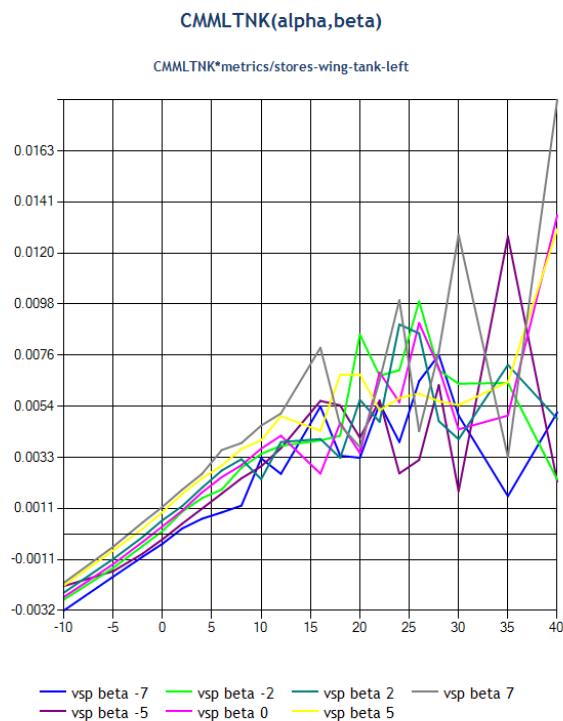
PITCHING MOMENT INCREMENT DUE TO GEAR



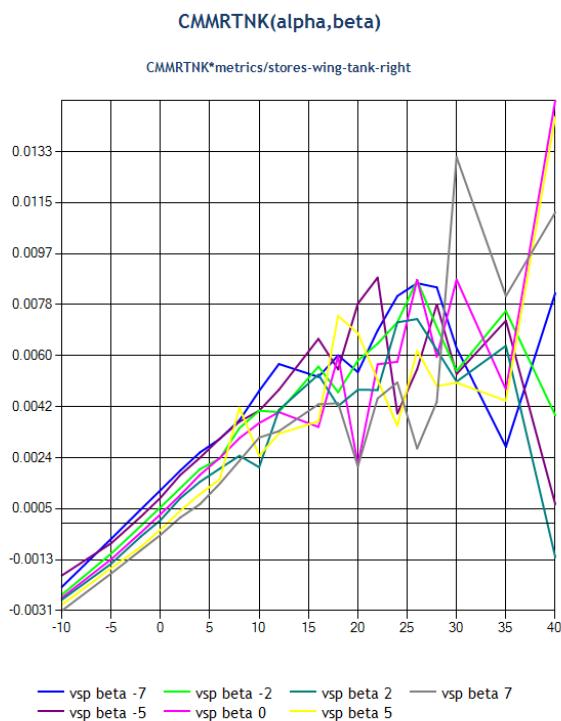
PITCHING MOMENT INCREMENT DUE TO TANK(CENTRE)



PITCHING MOMENT INCREMENT DUE TO TANK(LEFT WING)

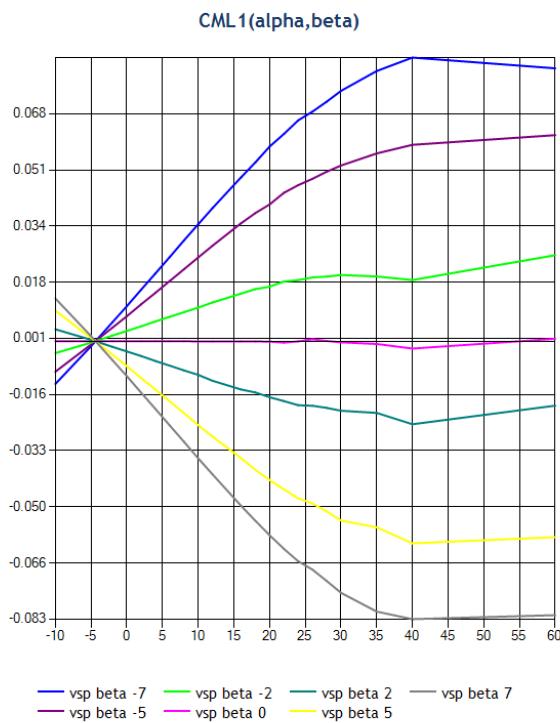


PITCHING MOMENT INCREMENT DUE TO TANK(RIGHT WING)



ROLL

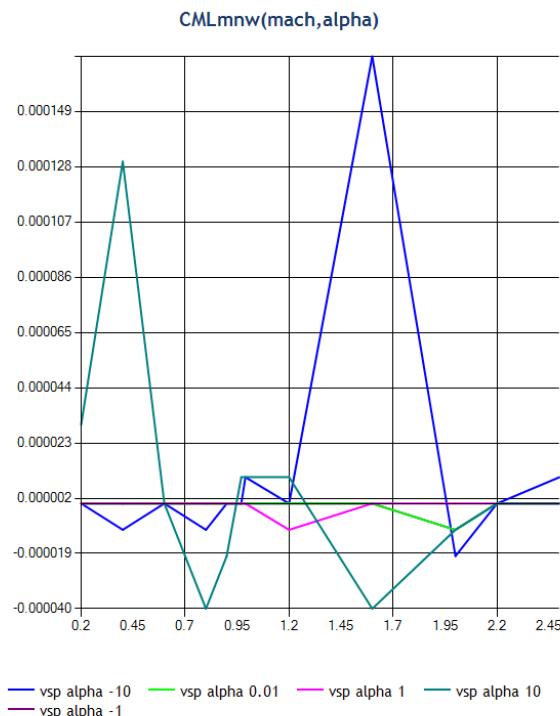
BASIC ROLLING MOMENT



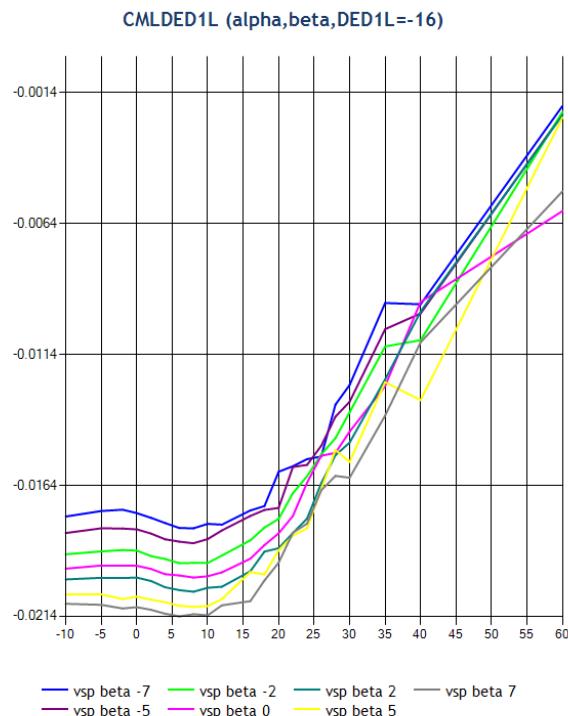
ROLL DAMPING DERIVATIVE



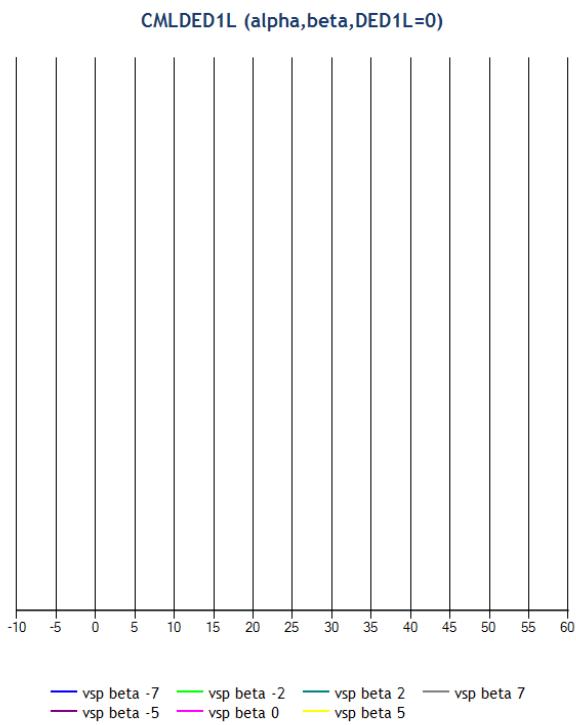
ROLL DUE TO MACH



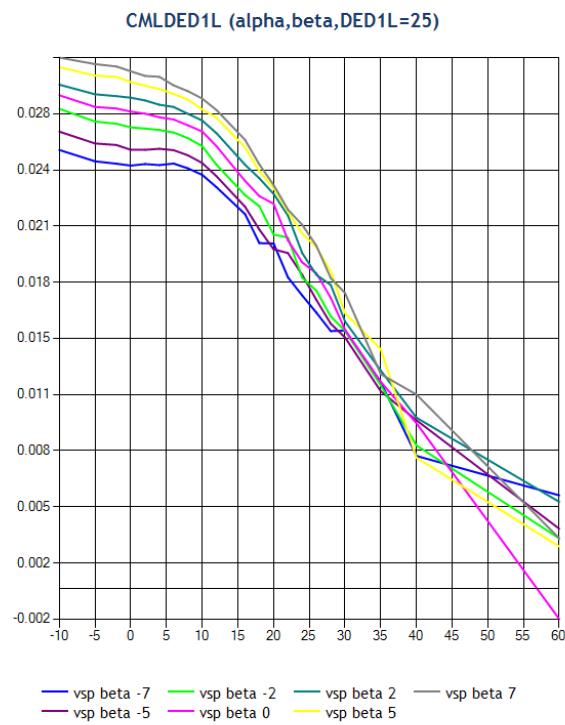
ROLLING MOMENT DUE TO ELEVON 1L DEFLECTION



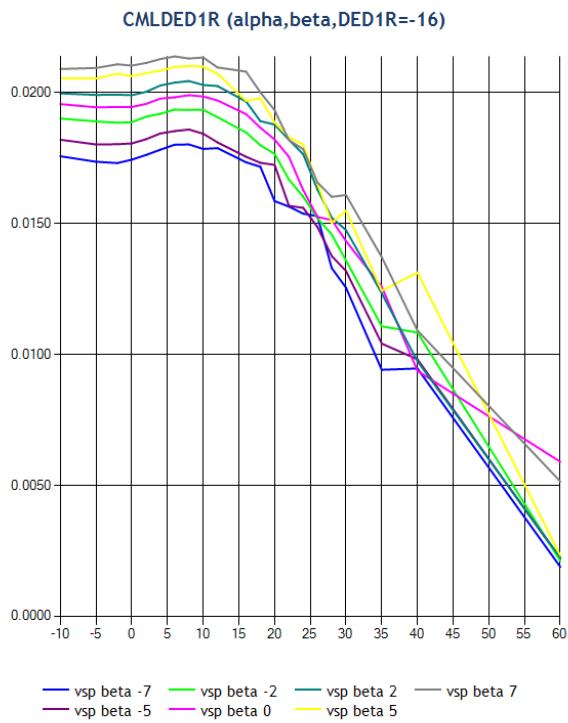
ROLLING MOMENT DUE TO ELEVON 1L DEFLECTION



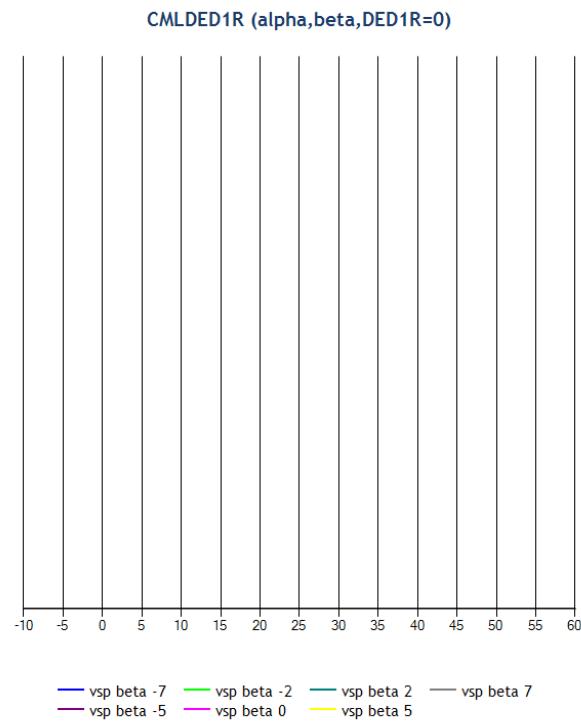
ROLLING MOMENT DUE TO ELEVON 1L DEFLECTION



ROLLING MOMENT DUE TO ELEVON 1R DEFLECTION

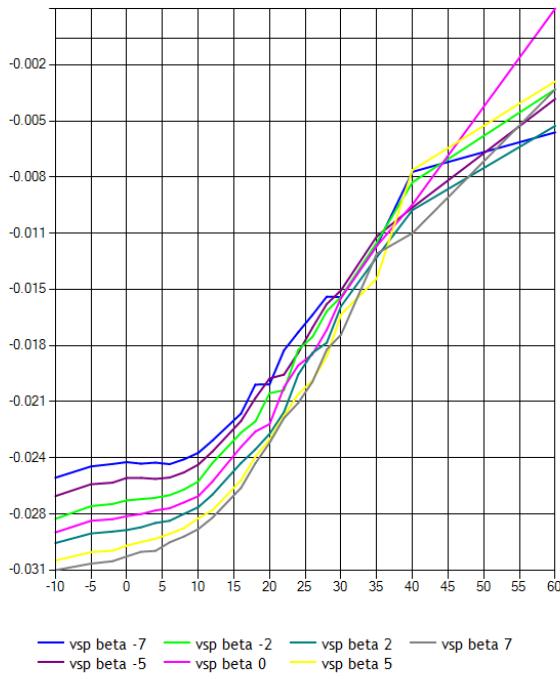


ROLLING MOMENT DUE TO ELEVON 1R DEFLECTION



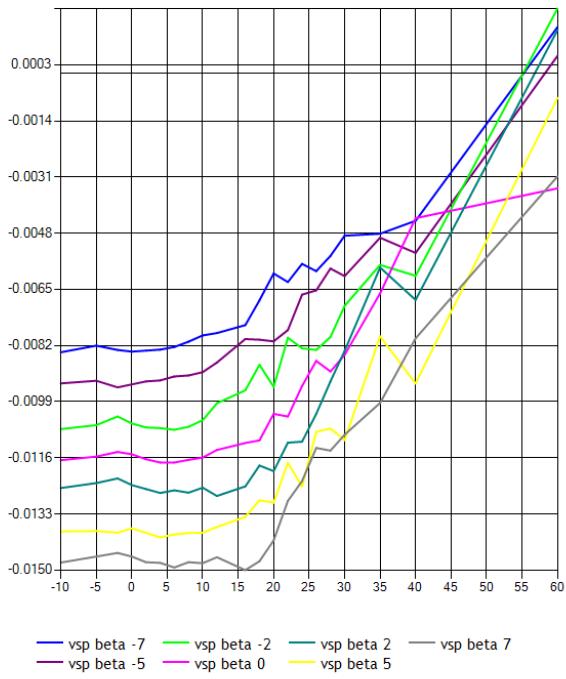
ROLLING MOMENT DUE TO ELEVON 1R DEFLECTION

CMLDED1R (alpha,beta,DED1R=25)



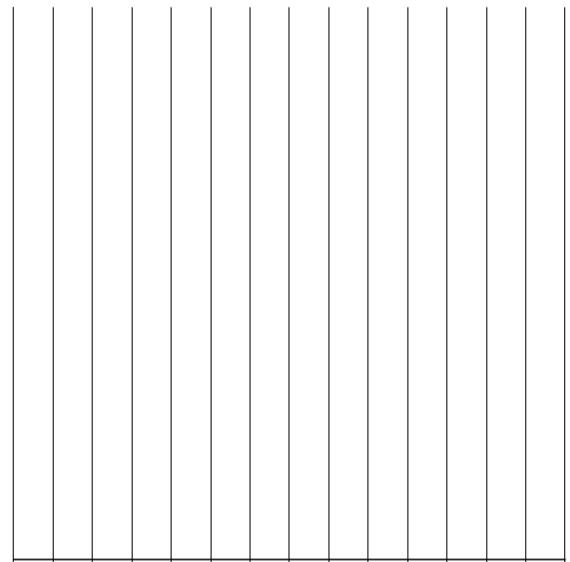
ROLLING MOMENT DUE TO ELEVON 2L DEFLECTION

CMLDED2L (alpha,beta,DED2L=-16)



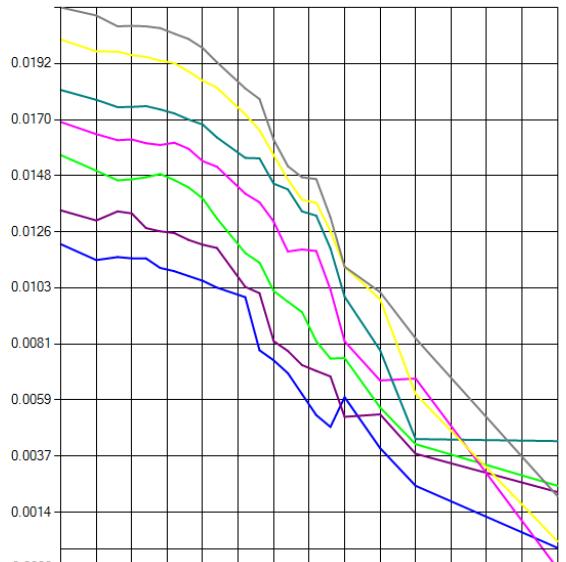
ROLLING MOMENT DUE TO ELEVON 2L DEFLECTION

CMLDED2L (alpha,beta,DED2L=0)



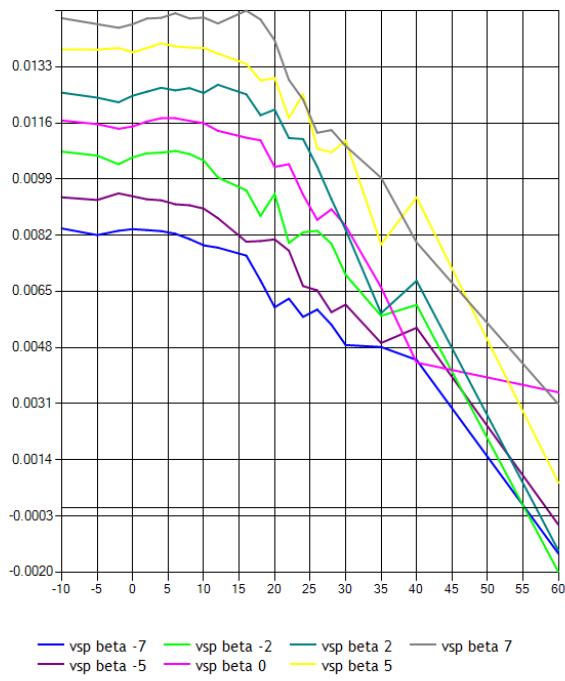
ROLLING MOMENT DUE TO ELEVON 2L DEFLECTION

CMLDED2L (alpha,beta,DED2L=25)



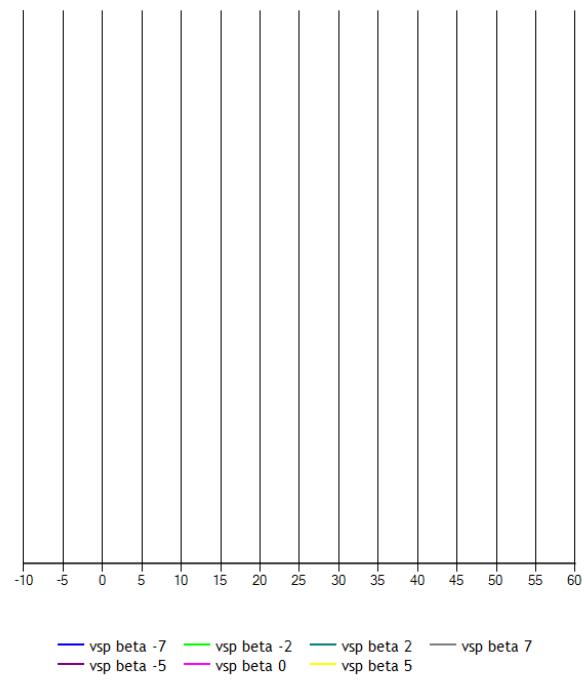
ROLLING MOMENT DUE TO ELEVON 2R DEFLECTION

CMLDED2R (alpha,beta,DED2R=-16)



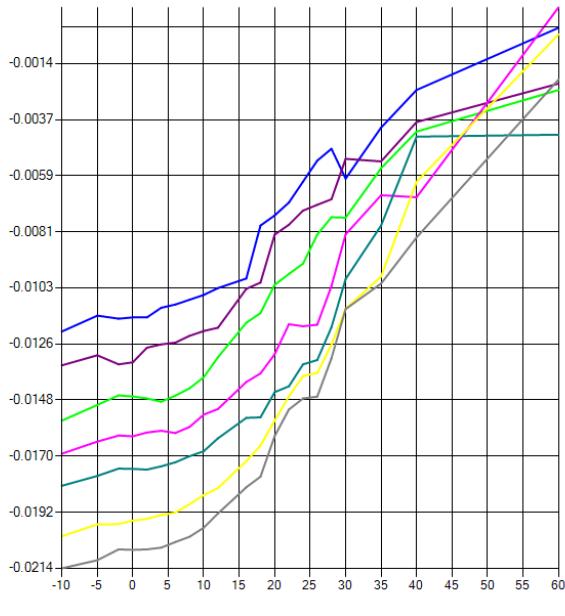
ROLLING MOMENT DUE TO ELEVON 2R DEFLECTION

CMLDED2R (alpha,beta,DED2R=0)



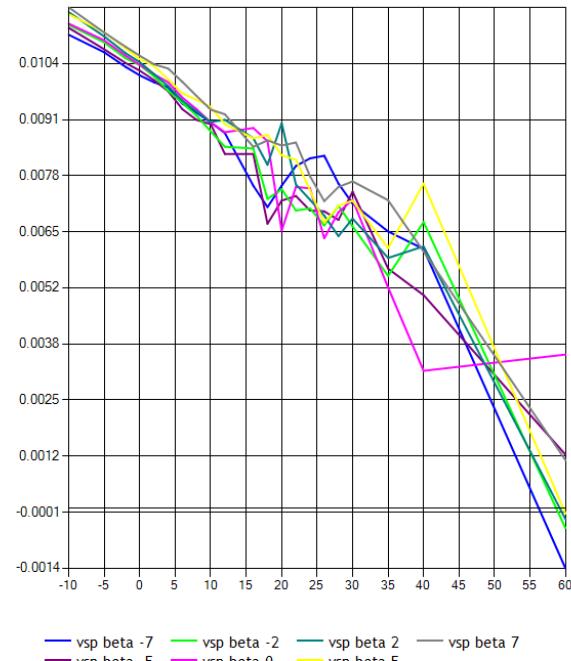
ROLLING MOMENT DUE TO ELEVON 2R DEFLECTION

CMLDED2R (alpha,beta,DED2R=25)

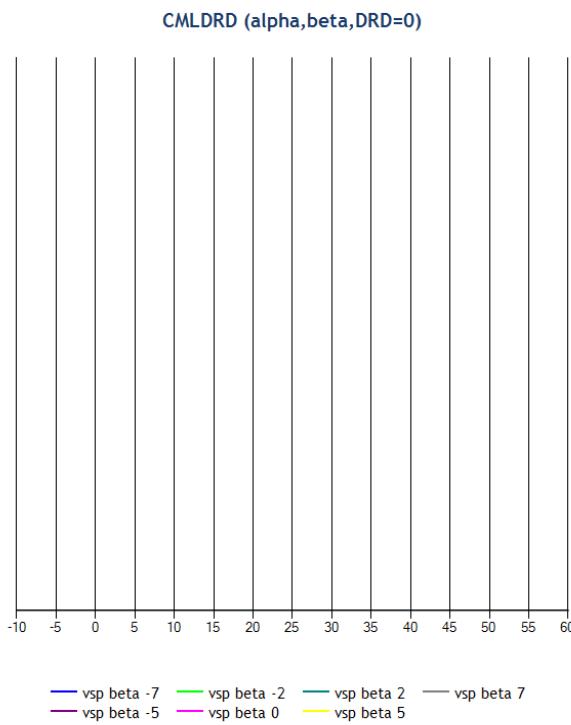


ROLLING MOMENT DUE TO RUDDER DEFLECTION

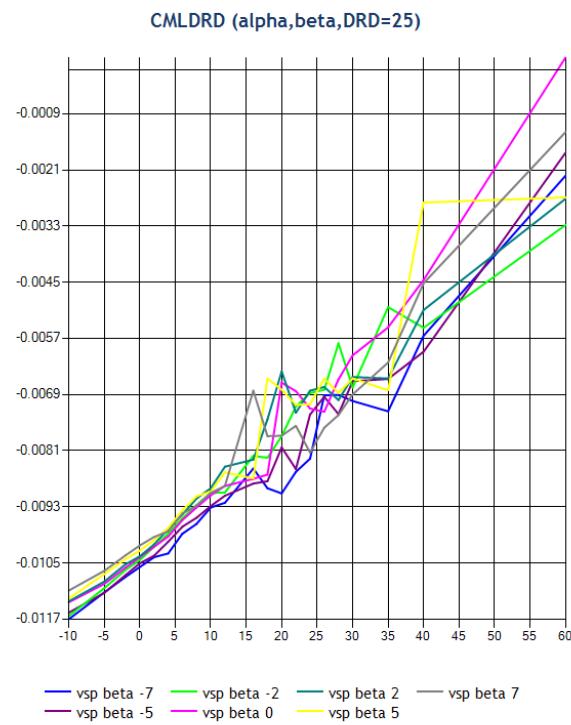
CMLDRD (alpha,beta,DRD=-25)



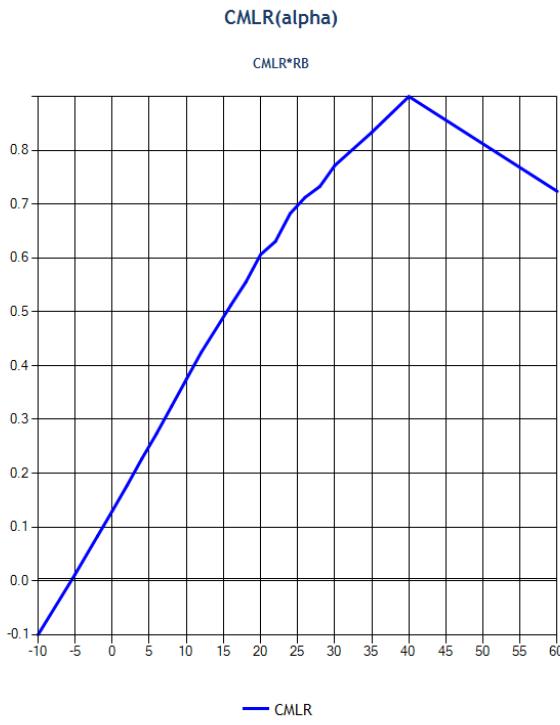
ROLLING MOMENT DUE TO RUDDER DEFLECTION



ROLLING MOMENT DUE TO RUDDER DEFLECTION



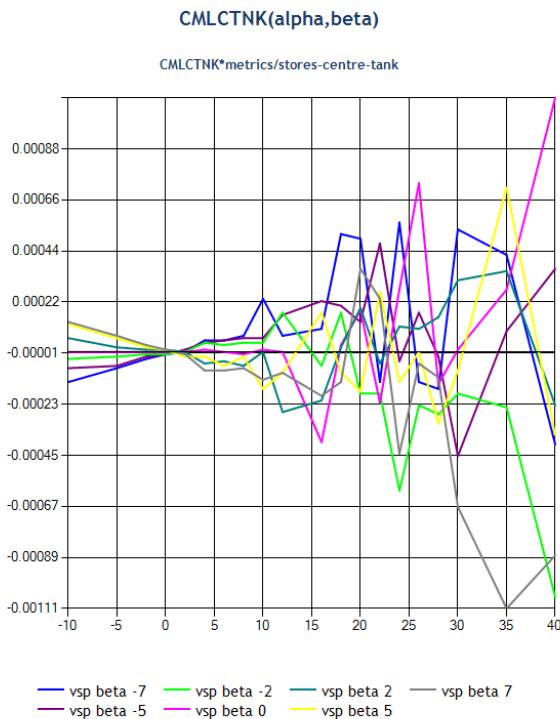
ROLLING MOMENT DUE TO YAW RATE



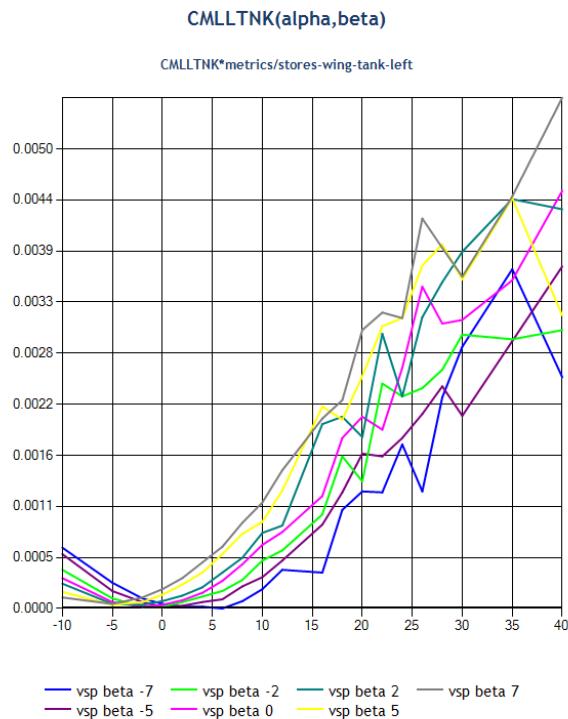
ROLLING MOMENT INCREMENT DUE TO GEAR



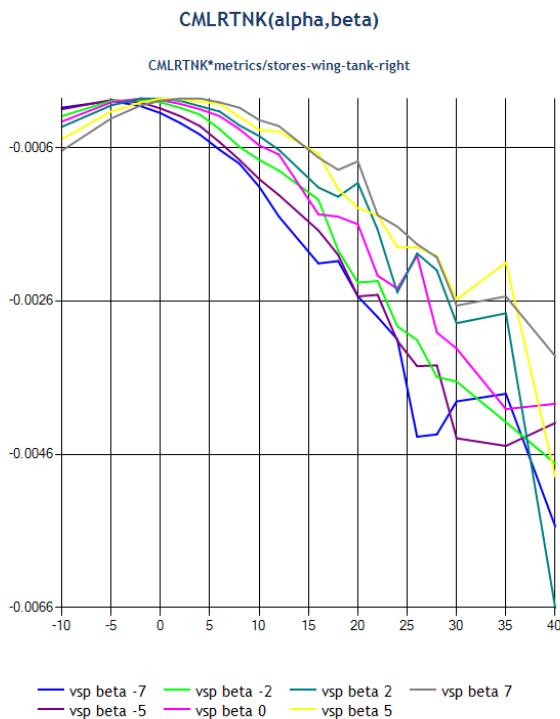
ROLLING MOMENT INCREMENT DUE TO TANK(CENTRE)



ROLLING MOMENT INCREMENT DUE TO TANK(LEFT WING)

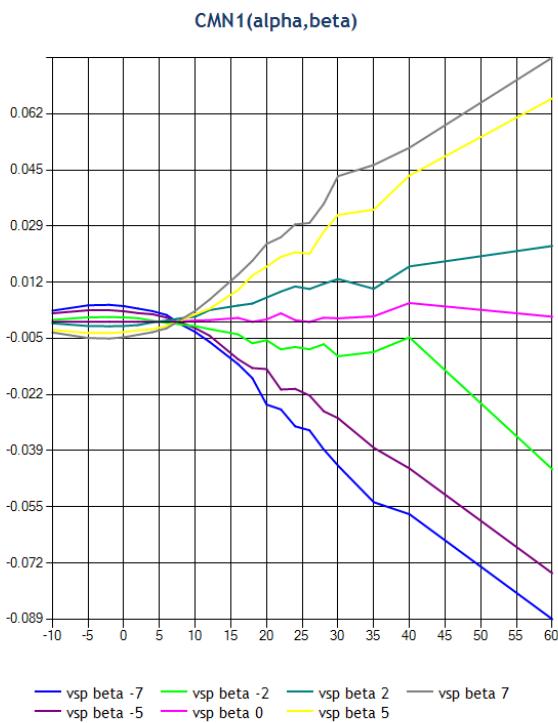


ROLLING MOMENT INCREMENT DUE TO TANK(RIGHT WING)

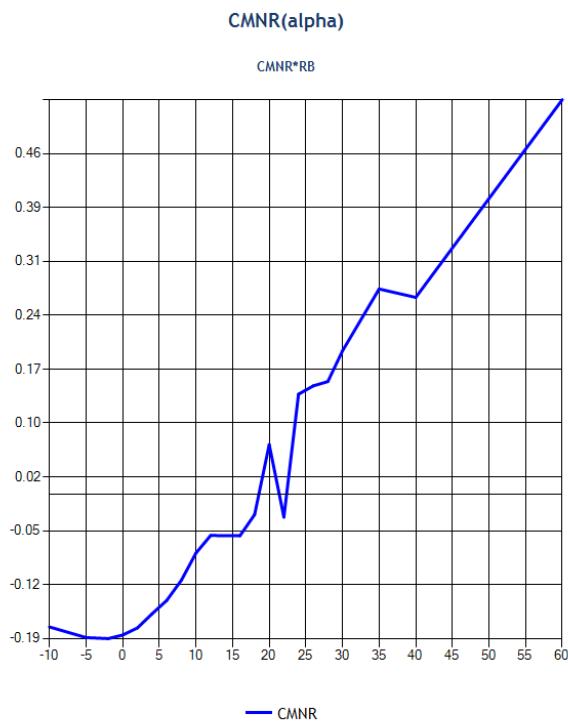


YAW

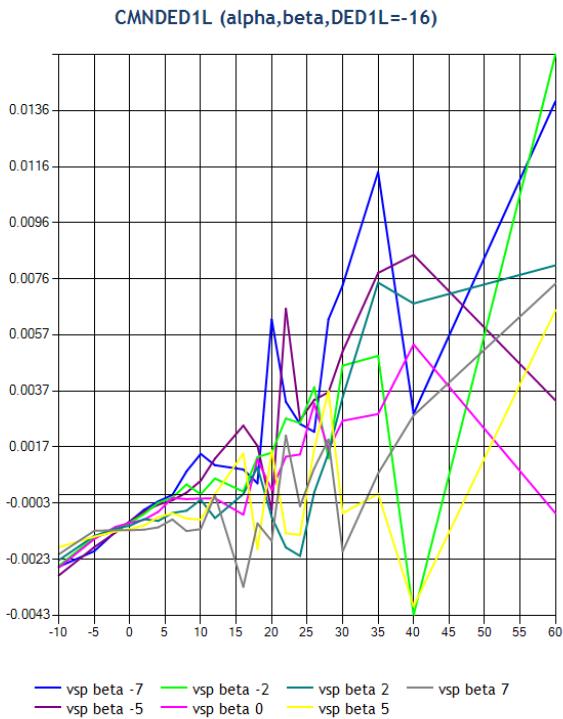
BASIC YAWING MOMENT



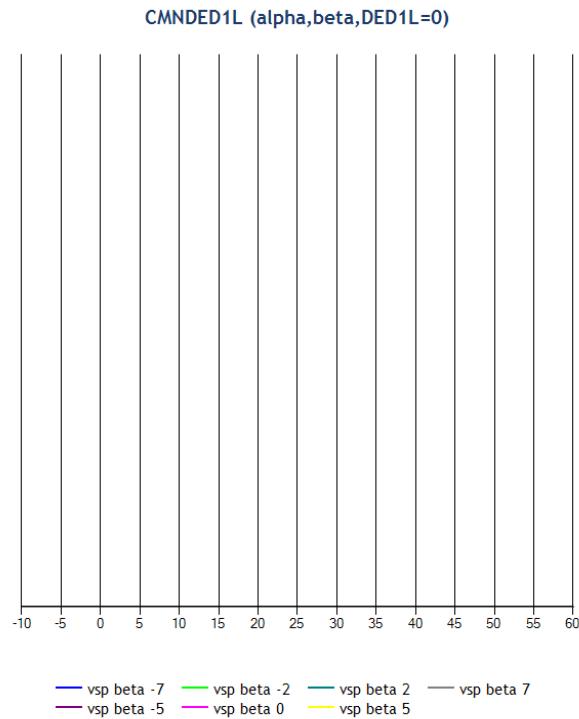
YAW DAMPING DERIVATIVE



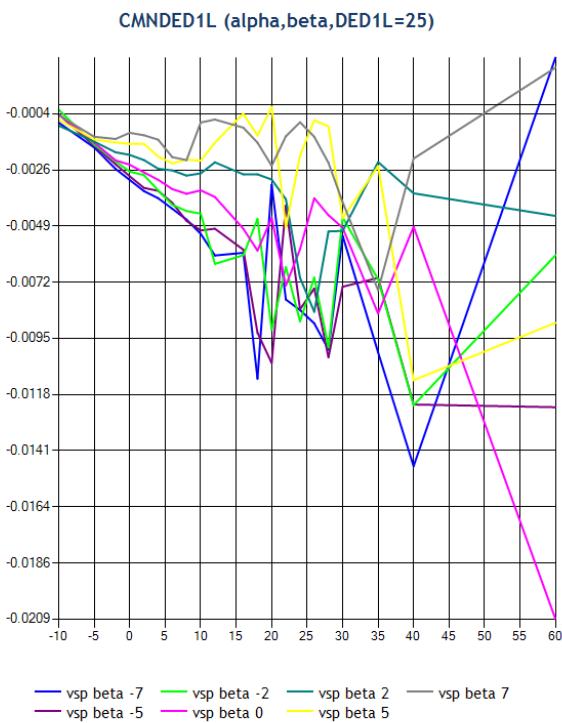
YAW MOMENT DUE TO ELEVON 1L



YAW MOMENT DUE TO ELEVON 1L



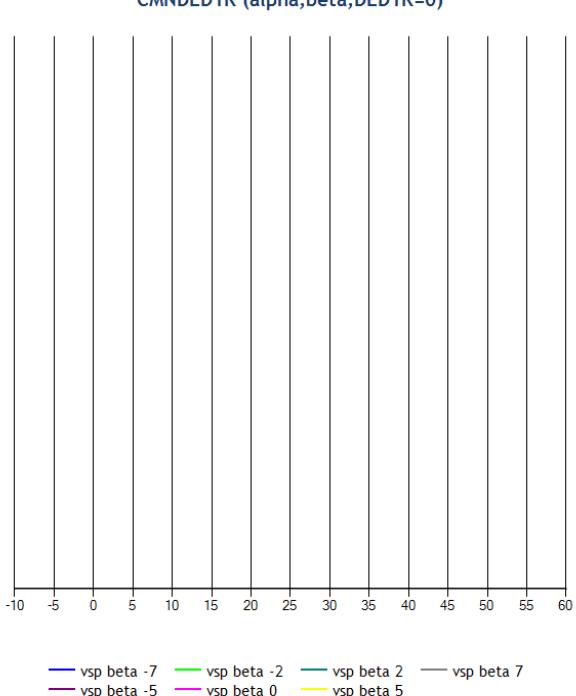
YAW MOMENT DUE TO ELEVON 1L



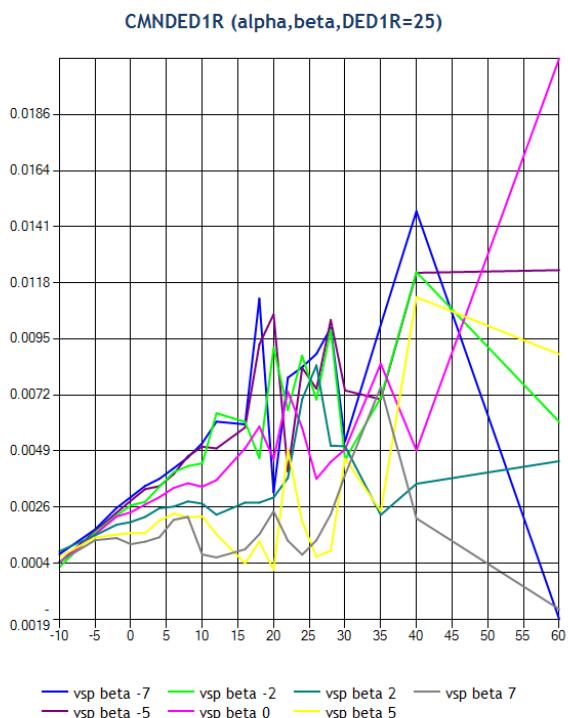
YAW MOMENT DUE TO ELEVON 1R



YAW MOMENT DUE TO ELEVON 1R

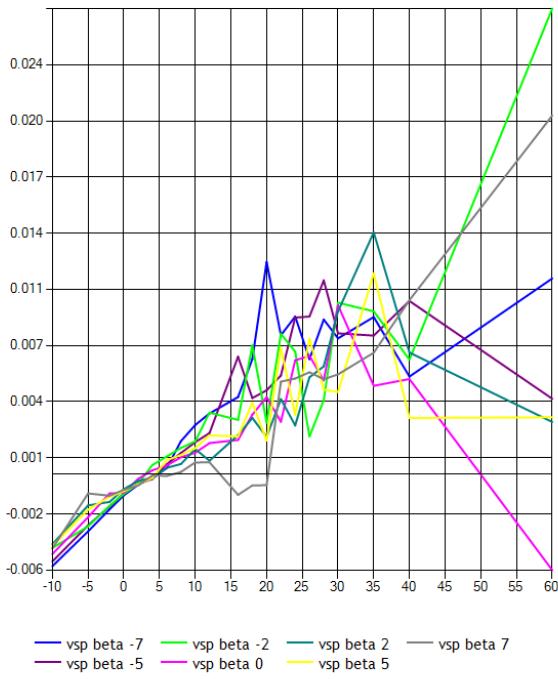


YAW MOMENT DUE TO ELEVON 1R



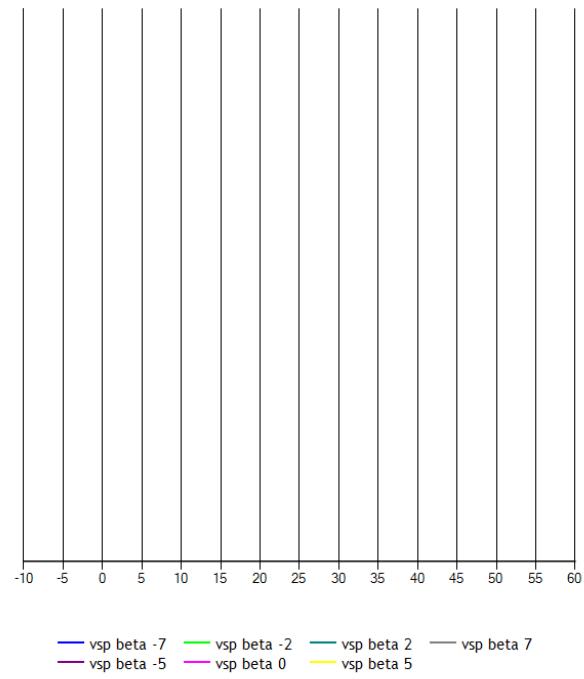
YAW MOMENT DUE TO ELEVON 2L

CMNDED2L (alpha,beta,DED2L=-16)



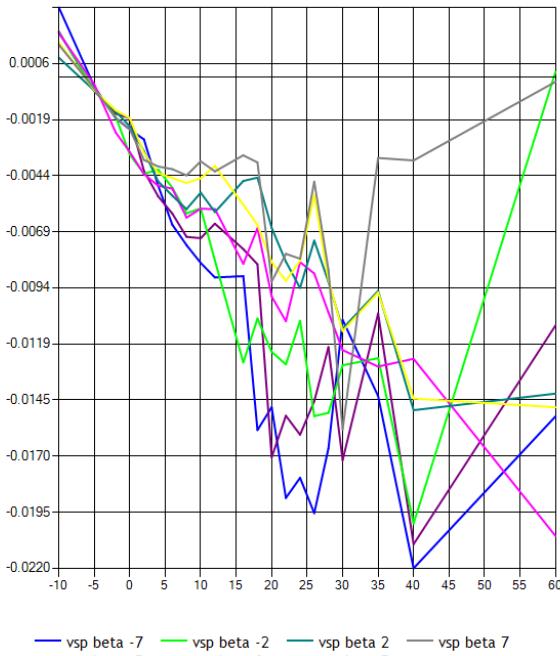
YAW MOMENT DUE TO ELEVON 2L

CMNDED2L (alpha,beta,DED2L=0)



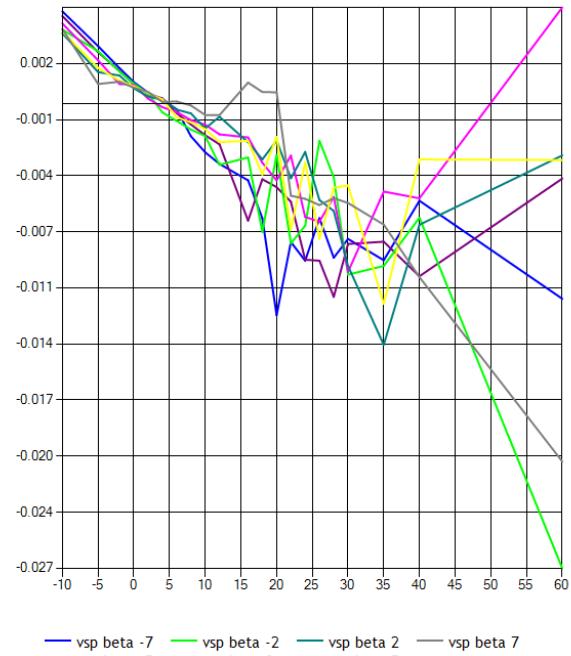
YAW MOMENT DUE TO ELEVON 2L

CMNDED2L (alpha,beta,DED2L=25)



YAW MOMENT DUE TO ELEVON 2R

CMNDED2R (alpha,beta,DED2R=-16)

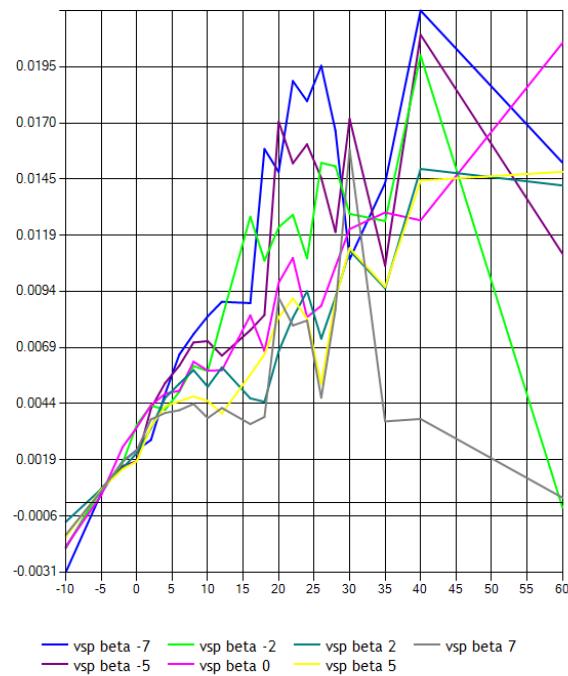
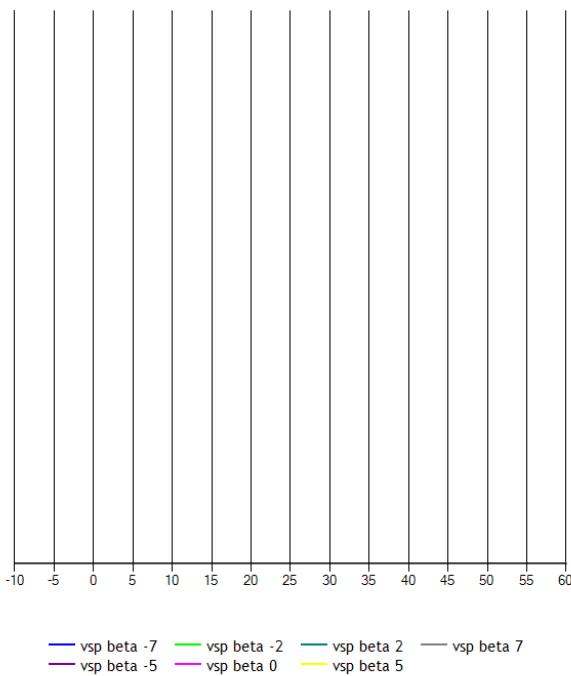


YAW MOMENT DUE TO ELEVON 2R

CMNDED2R (alpha,beta,DED2R=0)

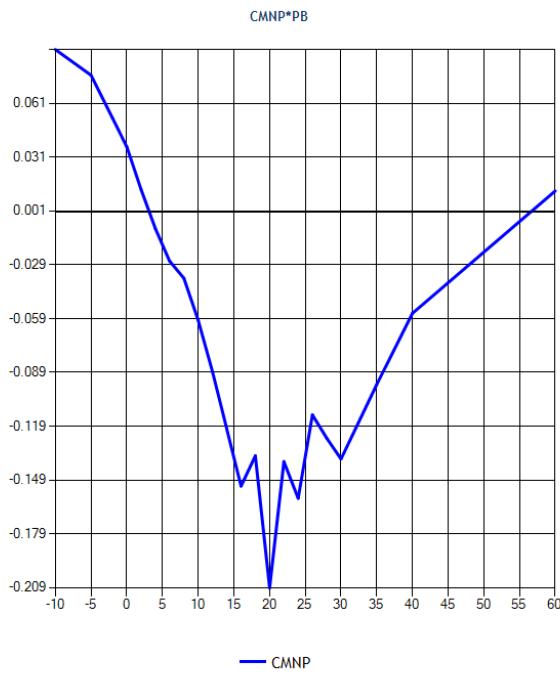
YAW MOMENT DUE TO ELEVON 2R

CMNDED2R (alpha,beta,DED2R=25)



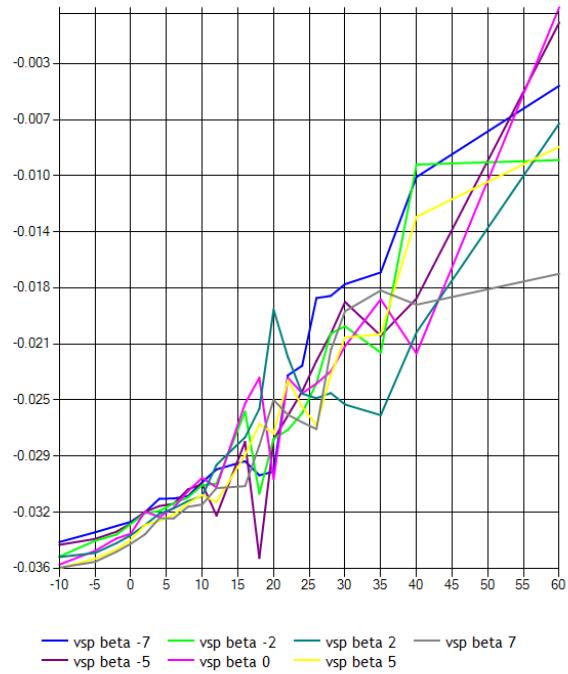
YAWING MOMENT DUE TO ROLL RATE

CMNP(alpha)

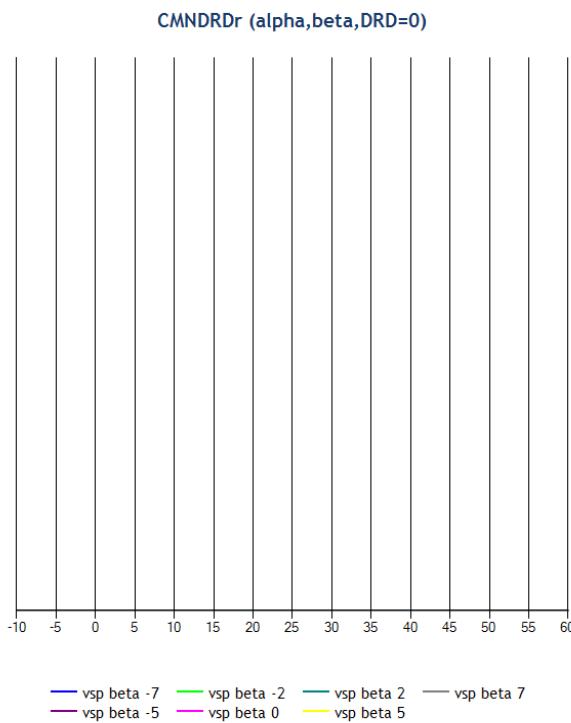


YAWING MOMENT DUE TO RUDDER DEFLECTION

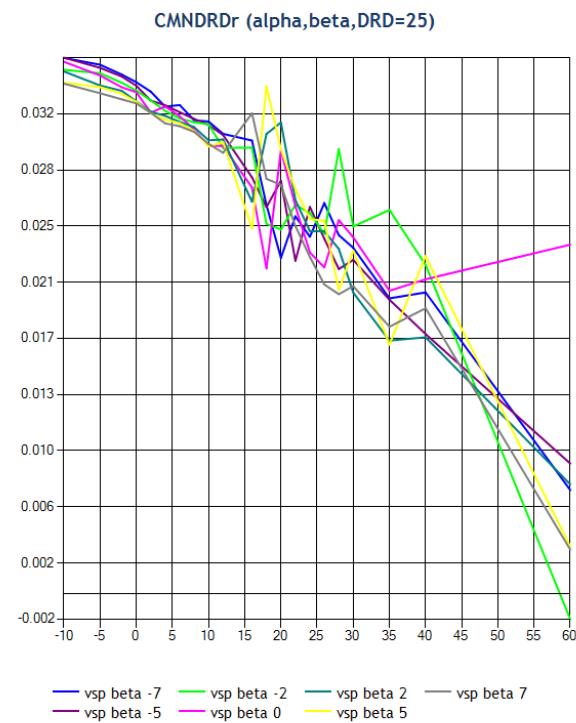
CMNDRDr (alpha,beta,DRD=-25)



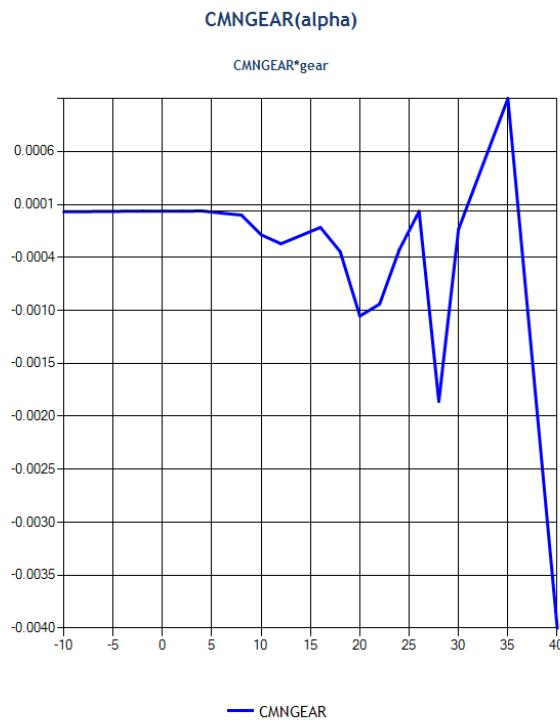
YAWING MOMENT DUE TO RUDDER DEFLECTION



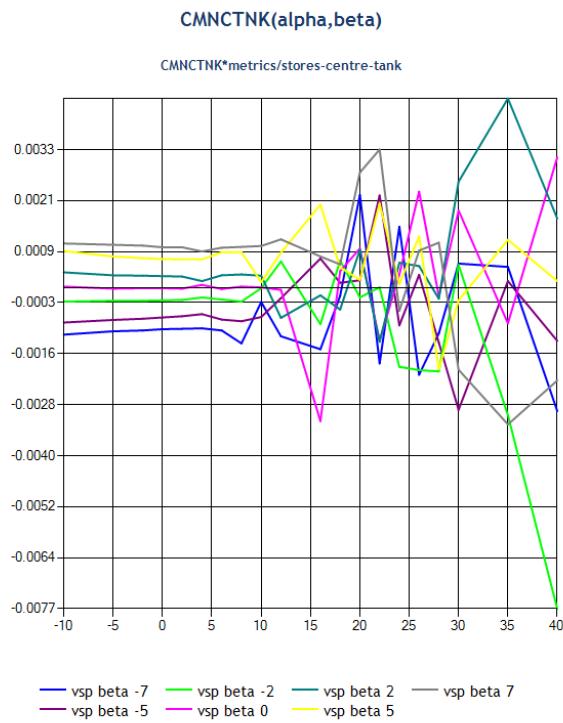
YAWING MOMENT DUE TO RUDDER DEFLECTION



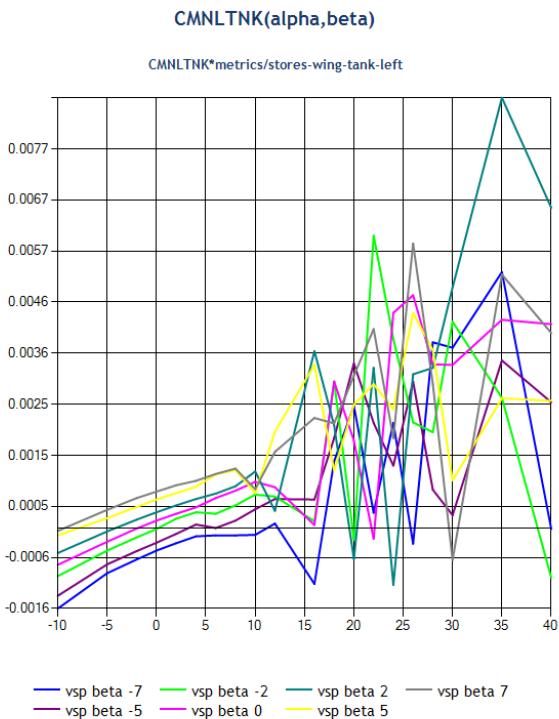
YAWING MOMENT INCREMENT DUE TO GEAR



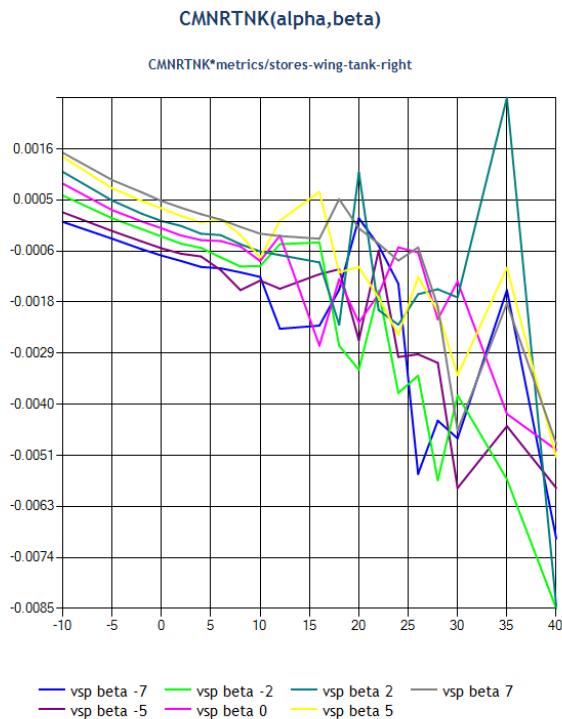
YAWING MOMENT INCREMENT DUE TO TANK(CENTRE)



YAWING MOMENT INCREMENT DUE TO TANK(LEFT WING)



YAWING MOMENT INCREMENT DUE TO TANK(RIGHT WING)



References

- Richard Harrison, rjh@zaretto.com: Mirage 2000-5 Aerodynamic data built from vspaero; AeroRP (7.72, 0, 0)M, ZDAT/AED/2017/09-08, September, 2017: <http://www.zaretto.com/sites/zaretto.com/files/Mirage2000-data-data/rjh-zaretto-Mirage2000-aerodynamic-data-vspaero.pdf>
- M. P. Bohn: AERODYNAMICS OF THE NEW GENERATION OF COMBAT AIRCRAFT WITH DELTA WINGS, NASA TM-75793, June 1978: <https://ntrs.nasa.gov/archive/nasa/casi.ntrs.nasa.gov/19800013819.pdf>

Aircraft Metrics

Element	X	Y	Z	Unit
Aerodynamic Reference Point (CoP)	7.72	0.00	0.00	M
Aircraft CG	7.72	0.00	0.00	M

Element	Unit
Wingspan	M
Wing Area	M2
Chord	M
CIMax	ND

Mass and balance

Element	Unit
Empty Weight	KG
I _{XX}	KG*M2
I _{YY}	KG*M2
I _{ZZ}	KG*M2
I _{XZ}	KG*M2

Element	X	Y	Z	Unit	Weight
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weight-0	5.31	-1.80	1.37	M	0 KG
weight-1	4.84	-3.28	-0.38	M	0 KG
weight-2	6.17	-2.36	-0.44	M	0 KG
weight-3	6.13	0.00	-0.54	M	0 KG
weight-4	6.17	2.36	-0.44	M	0 KG
weight-5	4.84	3.28	-0.38	M	0 KG
weight-6	5.31	1.80	1.37	M	0 KG
weight-7	7.41	-1.80	1.37	M	0 KG
weight-8	7.41	1.80	1.37	M	0 KG
Cannon ammo	3.50	0.00	0.00	IN	0 KG

Ground Reactions

Element	X	Y	Z	Unit	Index
NOSE_LG	4.82	0.00	-1.49	M	0
LEFT_MLG	7.92	-1.63	-1.51	M	1
RIGHT_MLG	7.92	1.63	-1.51	M	2
LEFT_WING_TIP	11.71	-4.53	-0.25	M	3
RIGHT_WING_TIP	11.71	4.53	-0.25	M	4
CANOPY	4.27	0.00	1.46	M	5
REAR_CANOPY	5.05	0.00	1.58	M	6
RADOME_FRONT	0.00	0.00	0.00	M	7
VERTICAL_TAIL_FRONT	13.06	0.00	3.63	M	8
VERTICAL_TAIL_REAR	13.72	0.00	3.54	M	9
REAR_BODY_LEFT	13.63	-0.50	0.53	M	10
REAR_BODY_RIGHT	13.63	0.50	0.53	M	11
LOWER_REAR_BODY	13.63	0.00	0.03	M	12
LOWER_MID_REAR_BODY	11.56	0.00	-0.32	M	13
REFUEL_PROBE	1.53	0.55	1.17	M	14
LEFT_STRAKE	5.21	-1.13	0.64	M	15
RIGHT_STRAKE	5.21	1.13	0.64	M	16
FRONT_LOWER_ANTENNA	2.35	0.00	-0.39	M	17
VSTAB_FRONT_ANTENNA	11.98	0.00	3.06	M	18
VSTAB_REAR_ANTENNA	13.74	0.00	2.98	M	19
CHUTE	13.83	0.00	1.21	M	20

Propulsion

Element	X	Y	Z	Unit	Feed
SNECMA_M53-P2	18.11	0.00	0.00	M	Feed line [0]

Tanks

Element	X	Y	Z	Unit	Capacity	Id	Priority	Standpipe
Feed line	1.53	0.55	1.17	M	50 KG	0	1	
forward fuselage internal tank	5.06	1.00	0.00	M	304 KG	1	3	30 KG
Right Wing Tank	8.56	4.00	-0.40	M	523 KG	2	3	30 KG
backward right feeder fuselage internal tank	8.56	0.70	0.00	M	592.6 KG	3	5	30 KG
backward left feeder fuselage internal tank	8.56	-0.70	0.00	M	592.6 KG	4	5	30 KG
forward fuselage internal tank	4.66	0.00	0.00	M	320 KG	5	3	30 KG
forward fuselage internal tank	4.66	-1.00	0.00	M	304 KG	6	3	30 KG
Left Wing Tank	8.56	-4.00	-0.40	M	523 KG	7	3	30 KG
External Tank Center RP522	8.56	0.00	1.00	M	990 KG	8	2	50 KG
External Tank Right RP542	8.56	4.00	1.00	M	1580 KG	9	2	50 KG
External Tank Left RP541	8.56	-4.00	1.00	M	1580 KG	10	2	50 KG
External Tank Right RP501	8.56	4.00	1.00	M	1343 KG	11	2	50 KG
External Tank Left RP502	8.56	-4.00	1.00	M	1343 KG	12	2	50 KG

Systems

Name
Mirage-2000-hydraulics
Mirage-2000-avionics
Mirage-2000-fadec
Mirage-2000-engines-Sneecma-M53
Mirage-2000-fcs
Mirage-2000-fuel

Independent variables

Name
aero/alpha-deg
aero/beta-deg
aero/pb
aero/qb
aero/rb
fcs/airbrake-lower
fcs/airbrake-upper
fcs/elevon-1L-pos-deg
fcs/elevon-1R-pos-deg
fcs/elevon-2L-pos-deg

fcs/elevon-2R-pos-deg

fcs/rudder-pos-deg

fcs/slat-1L-pos-deg

fcs/slat-2L-pos-deg

gear/gear-pos-norm

metrics/stores-centre-tank

metrics/stores-wing-tank-left

metrics/stores-wing-tank-right

velocities/mach