

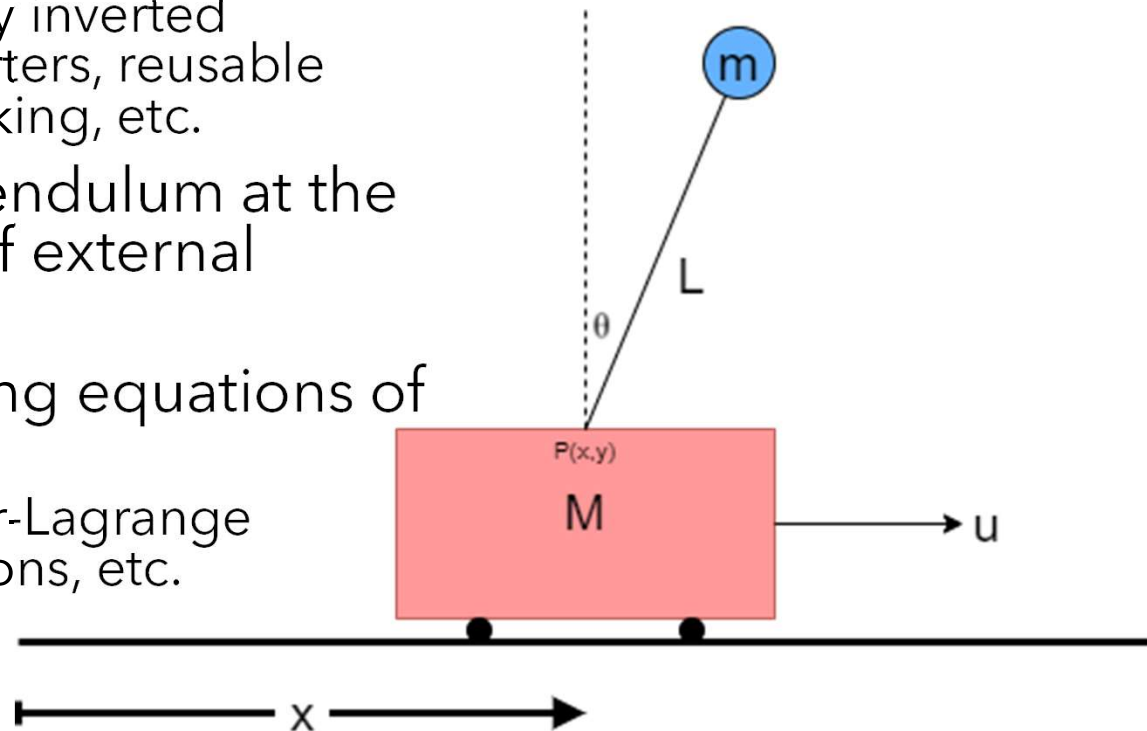


Inverted Pendulum on a Cart and Control

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Introduction

- Classic problem in control theory
 - Related problems: stationary inverted pendulum, Segway transporters, reusable rocket boosters, human walking, etc.
- Goal of system: balance pendulum at the upright position with use of external inputs/forces (\mathbf{u})
- Various methods for deriving equations of motion
 - Newtonian mechanics, Euler-Lagrange equation, Hamilton's equations, etc.



Introduction (cont.)

- Assumptions

- Massless and stiff rod
- Frictionless point of connection $P(x,y)$
- $P(x,y)$ is such that mass m is constrained to a single axis of rotation (circle of radius L)
- Uniform cart-mass M and point-mass m
- General model of real physical system

- Parameters: M = cart mass

m = pendulum/ball mass

L = length of rod

$P(x,y)$ = point of connection between cart and pendulum

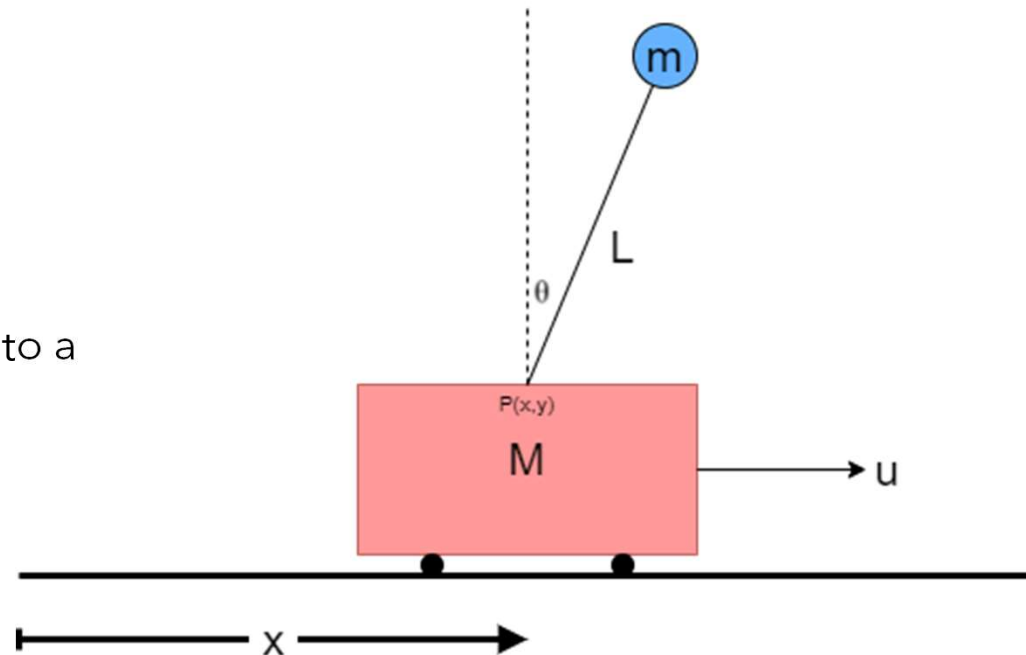
θ = angle of pendulum

\vec{u} = external control force

g = gravitational acceleration

b = friction/damping coefficient

$(x_G, y_G) = (x + L\sin\theta, L\cos\theta)$ = center of gravity of mass m



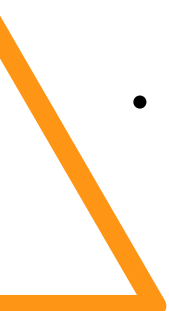
Introduction (cont.)





Physics Used for Derivation

- Newton's 2nd: sum of forces is equal to the product of mass and acceleration ($F=ma$)
- Newton's 3rd: for each force, there is an equal and opposite force
- Newton's 2nd for rotational motion: sum of torques is equal to the product of the moment of inertia and angular acceleration $ma = F \implies I_P \ddot{\theta} = \tau$
- Moment of inertia: measure of resistance to angular acceleration $I_p = \int r^2 dm$
- Torque/moments: product of force and perpendicular distance to point of rotation $\tau = (r_{\perp})(\vec{F})$
- Inertial/fictitious forces: force that appears to act on an object when viewed from an accelerating/rotating frame of reference



Derivation of Equations of Motion

Applying Newton's 2nd Law on to the cart mass M gives:

$$ma = F \implies M\ddot{x} = -H - b\dot{x} + \vec{u}$$

Applying Newton's 2nd Law on the pendulum mass in the horizontal direction gives us: $m\ddot{x}_G = H$. Plugging this in:

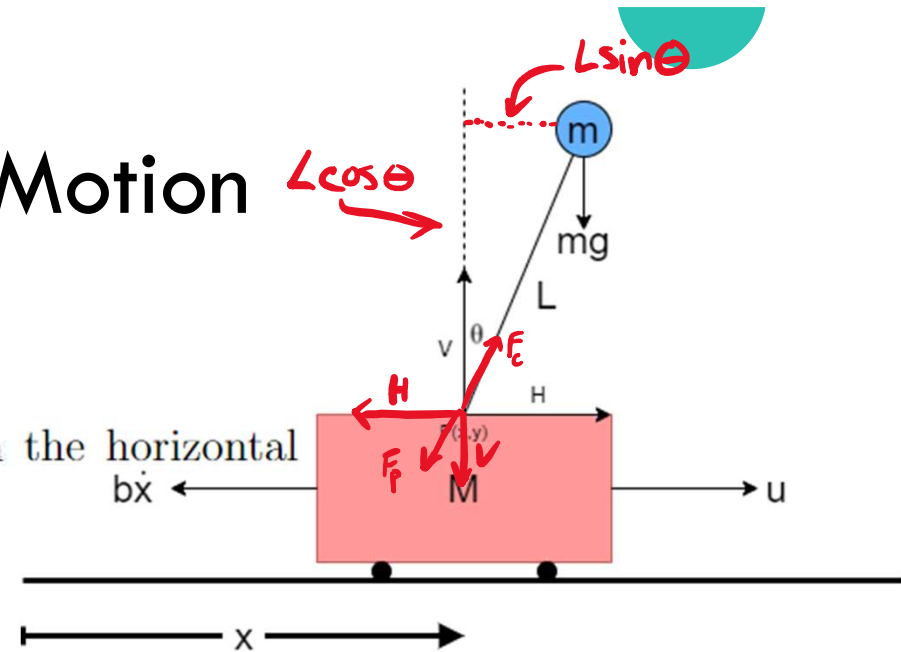
$$\implies M\ddot{x} = -m\ddot{x}_G - b\dot{x} + \vec{u}$$

Since $x_G = x + L\sin\theta$, then $\ddot{x}_G = \ddot{x} - L\omega^2\sin\theta + L\ddot{\theta}\cos\theta$. Plugging in:

$$\implies M\ddot{x} = -m(\ddot{x} - L\omega^2\sin\theta + L\ddot{\theta}\cos\theta) - b\dot{x} + \vec{u}$$

$$\iff M\ddot{x} + m(\ddot{x} - L\omega^2\sin\theta + L\ddot{\theta}\cos\theta) + b\dot{x} = \vec{u}$$

$$\iff (M + m)\ddot{x} - mL\omega^2\sin\theta + mL\ddot{\theta}\cos\theta + b\dot{x} = \vec{u} \quad (1)$$



Derivation of EoMs (cont.)

Applying Newton's 2nd to the fictitious force: $F_f = m\ddot{x}$

Applying analog of Newton's 2nd for rotational motion to the pendulum:

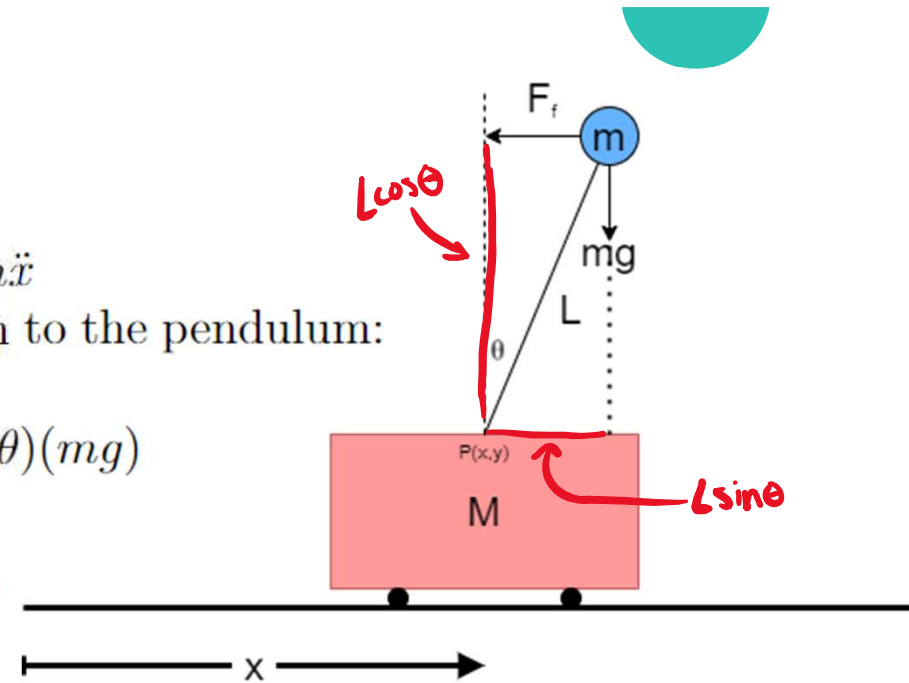
$$I_P \ddot{\theta} = \tau \implies mL^2 \ddot{\theta} = -(L \cos \theta)(m\ddot{x}) + (L \sin \theta)(mg)$$

$$\iff mL^2 \ddot{\theta} = -mL\ddot{x} \cos \theta + mgL \sin \theta$$

$$\iff mL^2 \ddot{\theta} + mL\ddot{x} \cos \theta - mgL \sin \theta = 0 \quad (2)$$

From equation 2, we can solve for $\ddot{\theta}$:

$$\implies \ddot{\theta} = \frac{mgL \sin \theta - mL\ddot{x} \cos \theta}{mL^2} \quad (3)$$



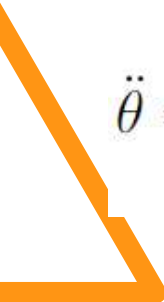


Derivation of EoMs (cont.)

Then, plugging equation 3 into equation 1 gives:

$$\ddot{x} = \frac{-m^2 L^2 g \cos\theta \sin\theta + m^2 L^3 \omega^2 \sin\theta - mL^2 b v + mL^2 \vec{u}}{mL^2(M + m(1 - \cos^2\theta))} \quad (4)$$

Then, plugging equation 4 into equation 3 gives:


$$\ddot{\theta} = \frac{(M + m)mgL \sin\theta - m^2 L^2 \omega^2 \cos\theta \sin\theta + mLbv \cos\theta - mL \cos\theta \vec{u}}{mL^2(M + m(1 - \cos^2\theta))} \quad (5)$$

System of EoMs

Let $\dot{x} = v$ such that $\ddot{x} = \dot{v} = (4)$. Let $\dot{\theta} = \omega$ such that $\ddot{\theta} = \dot{\omega} = (5)$. We obtain the system:

$$\dot{x} = v$$

$$\dot{v} = \frac{-m^2 L^2 g \cos \theta \sin \theta + m^2 L^3 \omega^2 \sin \theta - mL^2 b v + mL^2 \vec{u}}{mL^2 (M + m(1 - \cos^2 \theta))}$$

$$\dot{\theta} = \omega$$

$$\dot{\omega} = \frac{(M + m)mgL \sin \theta - m^2 L^2 \omega^2 \cos \theta \sin \theta + mLb v \cos \theta - mL \cos \theta \vec{u}}{mL^2 (M + m(1 - \cos^2 \theta))}$$

$$\bar{\mathbf{x}} = \begin{bmatrix} x \\ v \\ \theta \\ \omega \end{bmatrix}$$

The above system has fixed points: $v^* = 0$; $\theta^* = 0, \pi$; $\omega^* = 0$; x is free.

Basic Control Theory

- Consider a system of form: $\frac{d}{dt}\bar{\mathbf{x}} = \bar{\mathbf{f}}(\bar{\mathbf{x}}, \vec{u}) = \mathbf{A}\bar{\mathbf{x}} + \mathbf{B}\vec{u}$
- If the system is controllable, u can be rewritten as: $\vec{u} = -\mathbf{K}\bar{\mathbf{x}}$
- For controllability: $rank(\mathbf{C}) = n$ for $\mathbf{C} = [\mathbf{B} \quad \mathbf{AB} \quad \mathbf{A}^2\mathbf{B} \quad \dots \quad \mathbf{A}^{n-1}\mathbf{B}]$
- The system becomes: $\frac{d}{dt}\bar{\mathbf{x}} = \mathbf{A}\bar{\mathbf{x}} + \mathbf{B}\vec{u} = (\mathbf{A} - \mathbf{BK})\bar{\mathbf{x}}$
- Controllers often based on linearized dynamics
- Jacobian matrix of control system: $\mathbf{J}(\bar{\mathbf{f}}(\bar{\mathbf{x}}, \vec{u})) = \frac{d\bar{\mathbf{f}}}{d\bar{\mathbf{x}}} + \frac{d\bar{\mathbf{f}}}{d\vec{u}}$ such that $\bar{\mathbf{f}}(\bar{\mathbf{x}}, \vec{u}) = 0$

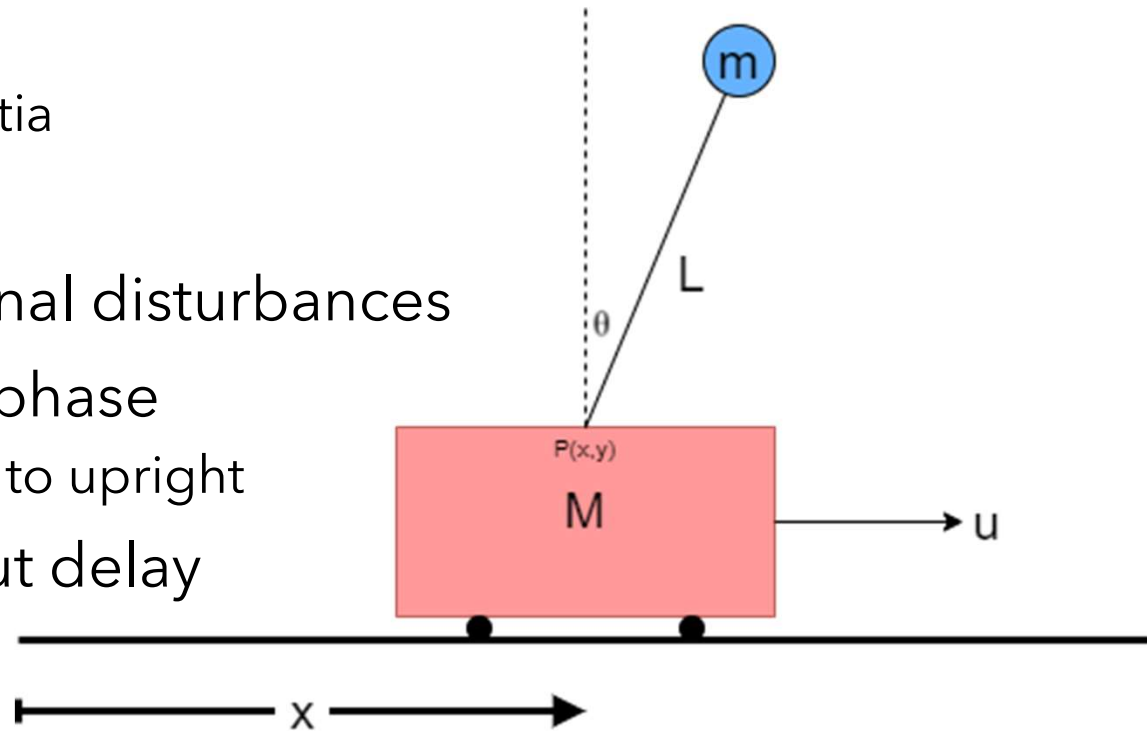
$$\bar{\mathbf{x}} = \begin{bmatrix} x \\ v \\ \theta \\ \omega \end{bmatrix}$$

$$\mathbf{J}(\bar{\mathbf{f}}(\bar{\mathbf{x}}, \vec{u})) = \begin{bmatrix} 0 & 1 & 0 & 0 \\ 0 & \frac{-k}{M} & \frac{kmg}{M} & 0 \\ 0 & 0 & 0 & 1 \\ 0 & \frac{-kb}{ML} & \frac{-kg(M+m)}{ML} & 0 \end{bmatrix} \bar{\mathbf{x}} + \begin{bmatrix} 0 \\ \frac{1}{M} \\ 0 \\ \frac{k}{ML} \end{bmatrix} \vec{u}$$

where $k = -1 \implies \theta = 0$ and $k = 1 \implies \theta = \pi$

Weaknesses of Model

- Massless rod does not reflect true center of gravity
 - Also, affects moment of inertia
- Hinge is not frictionless
- Does not account for external disturbances
- Does not have “swing-up” phase
 - Assumes pendulum is close to upright
- Assumes no controller input delay



Types of Control

- Passive control: no external input, typically low-cost
 - Ex: stop sign
- Active control: requires external input
 - Open-loop: no sensors, requires pre-programmed inputs
 - Ex: timed stoplight
 - Disturbance feedforward: uses prior knowledge to adjust system output
 - Ex: GPS rerouting
 - **Closed-loop feedback**: uses error-detecting sensors (**y**) that feed into a controller (**u**) that feeds back into the system
 - Ex: roadbed sensors for stoplight

Source: UW Book

