

AM2540
Strength of Materials Lab
Code **G**

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Part 1

Tension Test: Mechanical Behavior of Mild Steel

1 Aim

To find the mechanical properties of mild steel.

2 Theory

Steel, an iron-carbon alloy, is extensively used for a wide variety of engineering applications. In design of steel components or structures it is important to quantify its mechanical properties. In a simple tension test, typically a round tensile bar is subject to tensile deformation and the force-deflection curve till failure is obtained. Using the experimental data several mechanical properties of the material can be determined.

Engineering stress in a tensile bar (S): Measure of the intensity of the internal force (Stress)

$$S = \frac{F}{A_0}$$

where F is the axial force and A_0 is the original cross-sectional area.

Engineering strain (e): Measure of deformation

$$e = \frac{\Delta l}{l_0}$$

where Δl is the change in length and l_0 is the original length.

True stress (σ): Axial force per unit current area

$$\sigma = \frac{F}{A} = S(1 + e)$$

True stress is the stress calculated using the instantaneous area of the bar.

True strain (ϵ): calculated using the instantaneous gauge length

$$\epsilon = \int_{l_0}^l \frac{dl}{l} = \ln \left(\frac{l}{l_0} \right) = \ln(1 + e)$$

True strain is the strain calculated using the instantaneous length.

3 Procedure

1. Find the dimensions of the round tensile bar (length of the uniform CS, average diameter).
2. Get familiarized with the machine and the devices for measurement of force and deflection. Install the specimen in the machine avoiding any significant load developing during the installation.
3. While one end is fixed, provide displacement at the other end obtained by hydraulic pressure.
4. Observe the plastic deformation, necking and finally fracture during the process of loading.
5. Record the force and the corresponding deflection. Besides these data points also record the points where any sudden change in data pattern occurs.
6. In the data table note at what value of load you first notice the localization of deformation (formation of a necked region).
7. After failure, record the characteristic features of the failure surface.

4 Observations

Pre-Test Data

- Length of the uniform cross-section: $l_0 = 29.4869$ mm
- Diameters at three different locations: $D_1 = 8.21$ mm, $D_2 = 8.26$ mm, $D_3 = 8.28$ mm
- Average diameter, $D = \frac{D_1 + D_2 + D_3}{3} = 8.25$ mm

Test Data

The following table contains a small sample of the test data obtained from the experiment. The complete data is plotted in the following graph.

Sl. No.	Applied Extension (mm)	Axial Force (N)	Engg. Stress, S (N/mm ²)	Engg. Strain, e (mm/mm)	True Stress, σ (N/mm ²)	True Strain, ϵ (mm/mm)
1	0.0	0.0	0.0	0.0	0.0	0.0
2	0.000230789	1e-10	1.8707e-12	7.8268e-06	1.8707e-12	7.8268e-06
3	0.000749588	17.039	0.3187	2.5421e-05	0.3187	2.5420e-05
4	0.001289368	32.8541	0.6146	4.3726e-05	0.6146	4.3725e-05
5	0.001289368	51.4189	0.9619	4.3726e-05	0.9619	4.3725e-05
6	0.000976563	67.5203	1.2631	3.3118e-05	1.2631	3.3117e-05
7	0.001024246	84.9083	1.5884	3.4735e-05	1.5884	3.4735e-05
8	0.001264572	100.7553	1.8848	4.2885e-05	1.8849	4.2884e-05
9	0.001205444	117.0793	2.1902	4.0880e-05	2.1902	4.0879e-05
10	0.001256943	134.3413	2.5131	4.2627e-05	2.5132	4.2626e-05
11	0.001607895	150.2513	2.8107	5.4529e-05	2.8108	5.4527e-05
12	0.001718521	166.3203	3.1113	5.8280e-05	3.1115	5.8279e-05
13	0.00148201	182.0723	3.4060	5.0259e-05	3.4061	5.0258e-05
14	0.001871109	199.5723	3.7334	6.3455e-05	3.7336	6.3453e-05
15	0.002210617	215.3243	4.0281	7.4969e-05	4.0283	7.4966e-05
16	0.0027771	231.7433	4.3352	9.4180e-05	4.3356	9.4176e-05
17	0.002473831	246.8583	4.6179	8.3895e-05	4.6183	8.3892e-05
18	0.002426147	263.2773	4.9251	8.2278e-05	4.9255	8.2275e-05
19	0.002429962	278.0433	5.2013	8.2408e-05	5.2017	8.2404e-05
20	0.002567291	294.1923	5.5034	8.7065e-05	5.5039	8.7061e-05

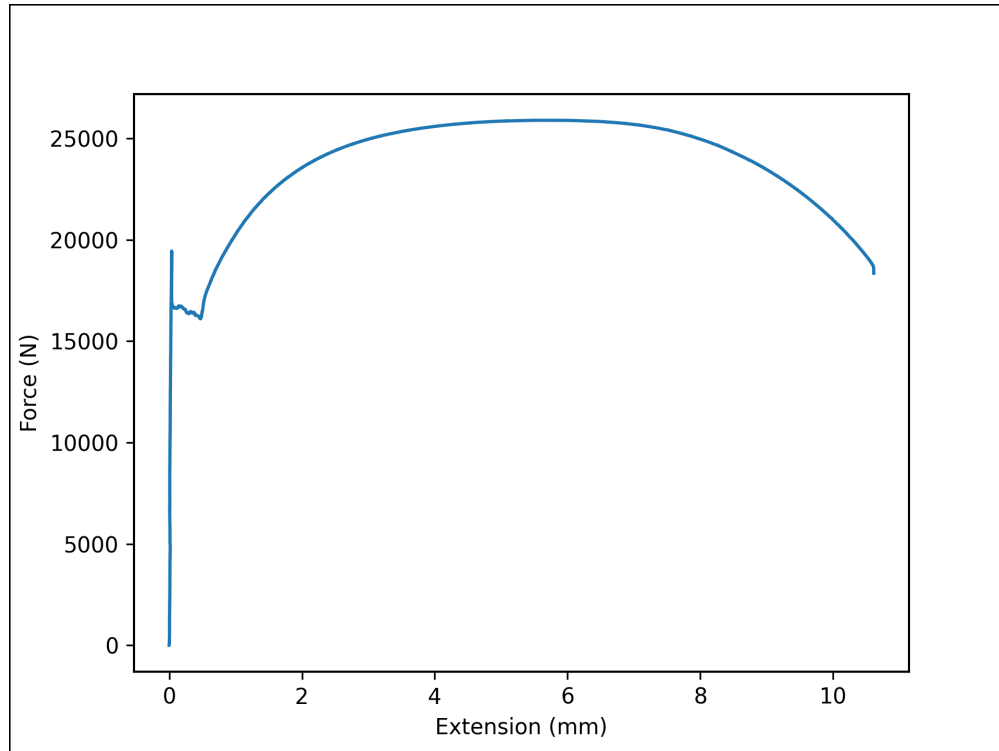


Figure 1.1: Force vs Extension Curve

Post-Test Data

- Failure Load: $F_f = 18357.6243$ N
- Final Diameter: $D_f = 5.16$ mm
- Final length of the initially uniform CS: $l_f = 29.4869 + 10.6134 = 40.1003$ mm
- Failure surface profile: *The type of fracture was a cup and cone fracture in which the failure surface was oriented at an angle of 45° to the axis of the bar.*

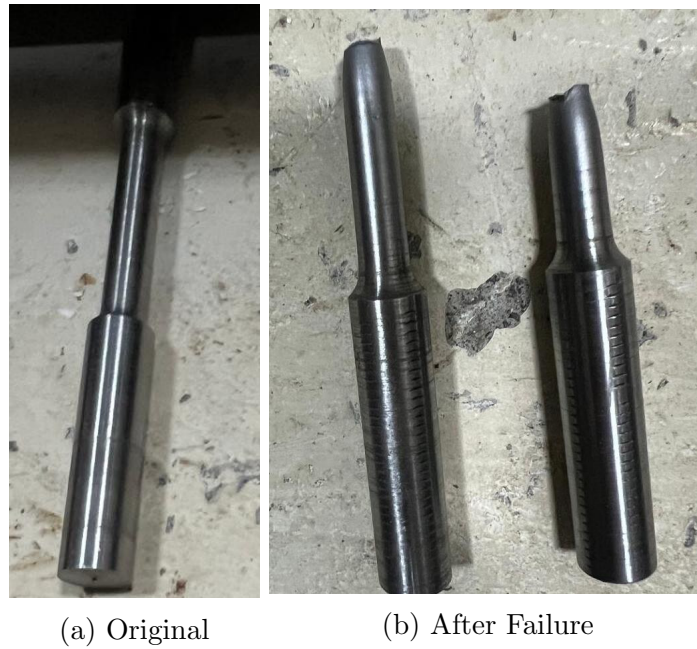


Figure 1.2: Mild Steel Bar

Plots and Values

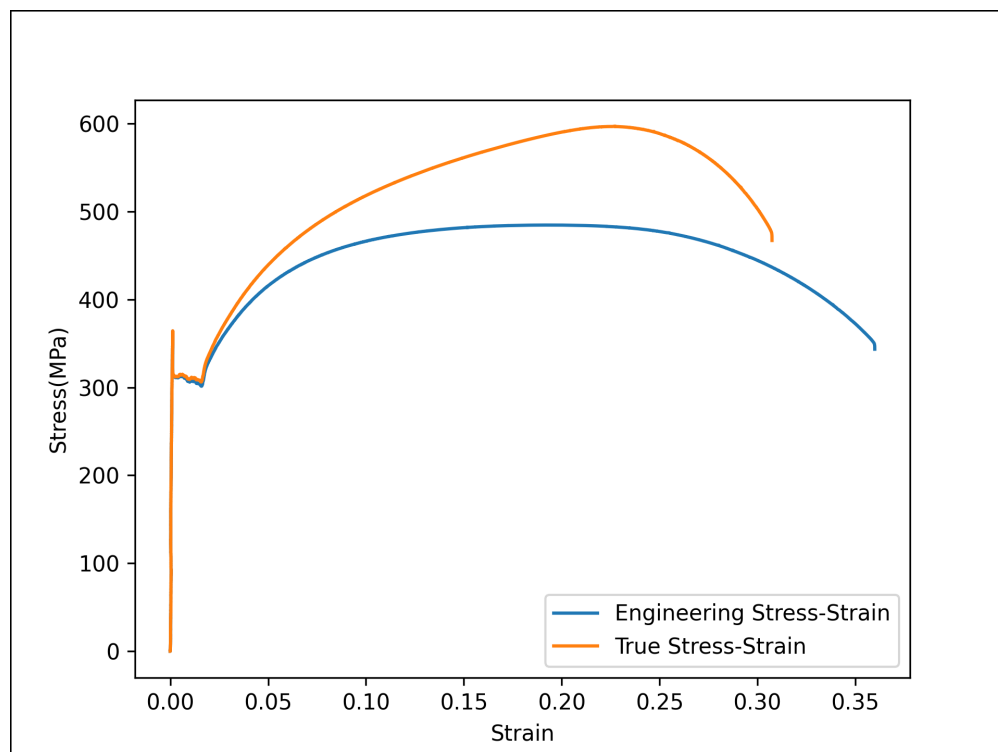


Figure 1.3: Engineering/True Stress-Strain Curve

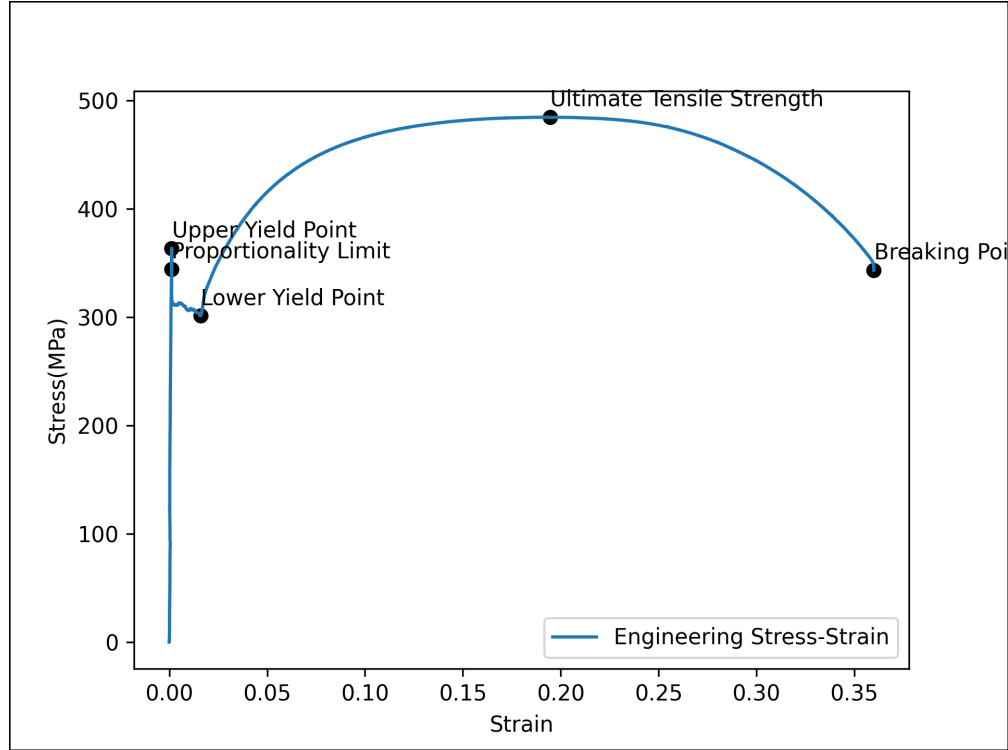


Figure 1.4: Various Points on the Stress-Strain Curve

Proportionality Limit (S_{pl}) ≈ 300 MPa

Tangent Modulus (E_{tan}) = 315.1938 GPa

Yield Strength (S_y) = 312.5066 MPa

Ultimate Tensile Strength (S_{uts}) = 484.5433 MPa

Percentage elongation, ductility = $\frac{L_f - L_0}{L_0} \times 100 = \frac{40.1003 - 29.4869}{29.4869} \times 100 = 35.99\%$

Percentage area reduction = $\frac{A_0 - A_f}{A_0} \times 100 = \frac{\pi \times D^2 - \pi \times D_f^2}{\pi \times D^2} \times 100 = 60.88\%$

5 Discuss

1. Why is steel a popular structural material?

Answer: Steel is good at handling tensile stresses. Concrete is the most used construction element but it is bad at handling tensile stresses, therefore it is reinforced with steel to prevent failure due to tensile stresses. Also the thermal expansion coefficient of steel is similar to that of concrete.

2. If the experiment was in load control (i.e., the force was increased incrementally rather than the displacement), how would the stress-strain curve look like in comparison to what you have obtained. Show schematically in a diagram.

Answer: We see in displacement control that there are multiple values of strain for a single value of stress. In load control, the stress-strain curve would be a straight line

with a slope equal to the Young's modulus i.e the same as displacement control up to the Yield Point.

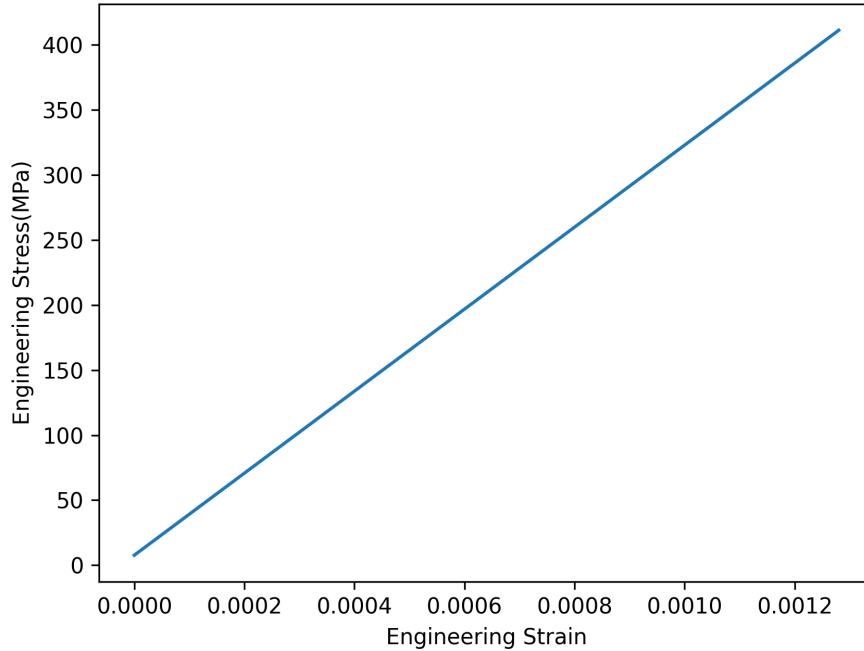
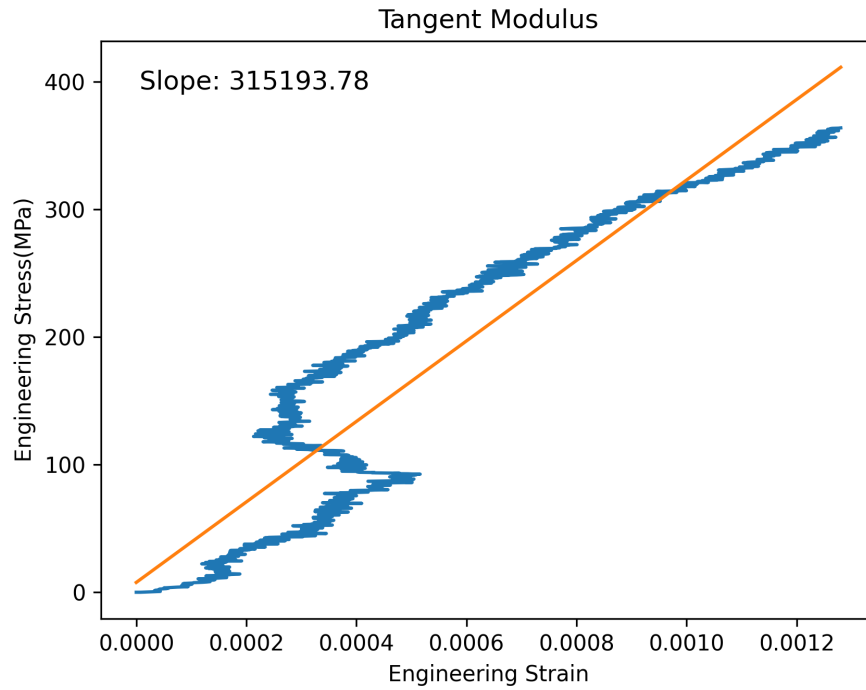


Figure 1.5: Load Controlled Stress-Strain Curve

3. Is the tangent modulus that you obtain near to the quoted values of steel (210 GPa)? If not what you think may be the reason for it.

Answer: The tangent modulus obtained by performing linear regression on the data till the yield point (since it was hard to deduct the Proportionality Limit) is **315.2 GPa**. This is considerably higher than the quoted value of 210 GPa. The reason for this discrepancy could be due to the fact that the Proportionality Limit was not accurately determined and the data was not perfectly linear. Other error contributors could be the non-uniformity of the bar, the presence of impurities etc.



4. Up to what values of strain there is less than 5% difference between true stress and engineering stress data?

Answer: The difference between true stress and engineering stress is less than 5% up to a strain of approximately 0.05. Excerpt from the data:

Engg. Stress, S (N/mm ²)	Engineering Strain, ϵ (mm/mm)	True Stress, σ (N/mm ²)	% Difference b/w S and σ
415.4689237478027	0.049999084339147216	8309.530649194463	4.999908433914707
415.51045310375224	0.049996608663508205	8310.7727546134	4.999660866350818
415.5349591651458	0.05000837660113474	8309.307108275874	5.000837660113477
415.56376781747116	0.05003788122861339	8304.983296931394	5.003788122861329
415.60922562601047	0.05004001777060322	8305.53713092737	5.004001777060328

Part 2

Failure Planes in Brittle and Ductile Materials

1 Aim

To find the differences between the critical planes of brittle and ductile materials.

2 Theory

Depending on their ability to absorb energy before fracture, materials can be divided into two broad categories: brittle or ductile. Brittle materials such as concrete, ceramics when subject to loads behave elastically and when critical conditions are reached they fail without any significant permanent deformation i.e., they don't exhibit any plasticity before failure. Even steels with higher carbon content fail in a brittle manner. In contrast, ductile material such as most aluminum alloys and low carbon steels have extensive plastic deformation before failure. The failure mechanisms are different and therefore different theories are applicable.

Maximum Principal Stress Theory

This failure theory applicable for brittle materials states that the failure occurs when the maximum principal stress in the material reaches a critical value. The principal plane with the maximum tensile stress is predicted to be the failure plane.

Maximum Shear Stress Failure Theory

This failure theory applicable for ductile material states that the failure occurs when the maximum shear stress in the material reaches a critical value. The maximum shear stress plane is the predicted failure plane.

3 Procedure

1. Take three chalk bars and subject them to axial, torsion and 3 point bend load manually.
2. Observe the failure planes and their location along the length.

4 Plots and Values



Figure 2.1: Torsional Loading(left) and Axia Loading(right)

Material	Axial Load	Torsional Load
Mild Steel	45°to the axis near the edge but here the material's shape makes it near the middle (Cup & Cone)	Perpendicular to the axis (90°) near the middle
Chalk	Perpendicular to the axis, near the edge (90°)	45°to the axis near the middle (Helicoidal)

Table 2.1: Location/Axis of Failure Planes

5 Discuss

1. **The difference between the failure planes you observe in the tensile test of mild steel and the chalk bar.**

Answer: Mild Steel is a ductile material and chalk is a brittle material. Their failure planes have different characteristics. The failure plane in the tensile test of mild

steel is inclined 45° to the axis of the bar, while the failure plane in the chalk bar is perpendicular to the axis of the bar.

2. **In your opinion where should the failure occur across the length in each case of loading? Why?**

Answer: In the case of axial loading, the failure should occur near the middle of the bar as the stress is maximum there. In case of torsional loading also the failure should occur near the middle of the bar as the stress is maximum there.

3. **If the mild steel specimen was loaded in torsion what will be the inclination of the failure plane with the length axis. Justify.**

Answer: The failure plane in the case of torsional loading of mild steel will be perpendicular to the axis of the bar. This is because the maximum shear stress is perpendicular to the axis of the bar.