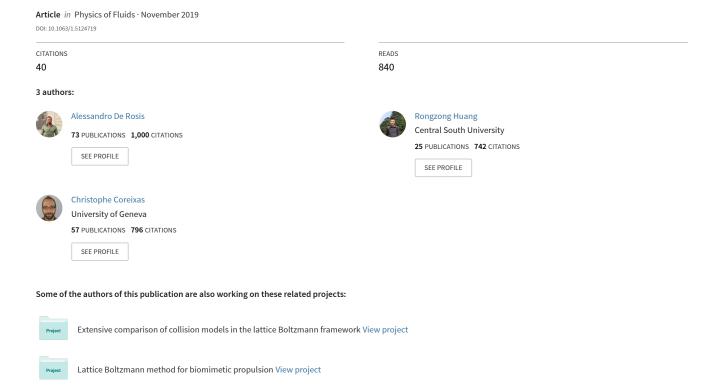
# Universal formulation of central-moments-based lattice Boltzmann method with external forcing for the simulation of multiphysics phenomena



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# **ABSTRACT**

The cascaded or central-moments-based lattice Boltzmann method (CM-LBM) is a robust alternative to the more conventional Bhatnagar-Gross-Krook-LBM for the simulation of high-Reynolds number flows. Unfortunately, its original formulation makes its extension to a broader range of physics quite difficult. In addition, it relies on CMs that are derived in an ad hoc manner, i.e., by mimicking those of the Maxwell-Boltzmann distribution to ensure their Galilean invariance a posteriori. This work aims at tackling both issues by deriving Galilean invariant CMs in a systematic and a priori manner, thanks to the Hermite polynomial expansion framework. More specifically, the proposed formalism fully takes advantage of the D3Q27 discretization by relying on the corresponding set of 27 Hermite polynomials (up to the sixth-order) for the derivation of both the discrete equilibrium state and the forcing term in an a priori manner. Furthermore, while keeping the numerical properties of the original CM-LBM, this work leads to a compact and simple algorithm, representing a universal methodology based on CMs and external forcing within the lattice Boltzmann framework. To support these statements, mathematical derivations and a comparative study with four other forcing schemes are provided. The universal nature of the proposed methodology is eventually proved through the simulation of single phase, multiphase (using both pseudopotential and color-gradient formulations), and magnetohydrodynamic flows.

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# I. INTRODUCTION

Nowadays, the lattice Boltzmann method (LBM) is a consolidated approach to simulate fluid flows. <sup>1–10</sup> The basic idea is to model the fluid through populations (or distributions) of fictitious particles moving along the links of a fixed Cartesian lattice. The space and time evolution of the distributions is predicted by solving the lattice Boltzmann equation that involves a collide-and-stream process, where the collision stage retains the flow physics. Then, a proper treatment of the collision process is instrumental to obtain accurate predictions of the fluid dynamics.

The impressive popularity of the LBM mainly stems from the intrinsic simplicity of the Bhatnagar-Gross-Krook (BGK) collision operator. 11 In short, all the populations are forced to relax (with a common unique rate) toward a discrete equilibrium state derived by applying a Gauss-Hermite quadrature to the continuous Maxwellian

distribution.<sup>12</sup> Unfortunately, it is well known to be prone to numerical instability in the low-viscosity regime, thus becoming unsuitable for the prediction of turbulent flows.

About 20 years ago, entropic LBMs were introduced as a way to tackle this stability issue. <sup>13–15</sup> The derivation of such LBMs is based on the H-theorem, <sup>16</sup> and it mainly consists in solving a minimization problem, at each grid node and for each time step, in order to ensure the growth over time of a predefined discrete entropy function. This methodology has led to great improvements in terms of numerical stability, especially in under-resolved conditions, which was later explained by its subgrid scale properties. <sup>17</sup> To find a better trade-off between stability and CPU time, analytical approximations of the minimization problem were recently proposed. <sup>18,19</sup> In the end, both methodologies were shown to be successful at simulating a large panel of very different phenomena. <sup>20–24</sup>

Another popular alternative is represented by the multiplerelaxation-time (MRT) LBM, which suggests to perform the collision in a space of raw moments (RMs). 25-28 While the secondorder ones are related to the flow physics and relax with a frequency directly linked to the fluid kinematic viscosity, higherorder moments are related to nonhydrodynamic modes, their relaxation rates can be considered as free parameters. The MRT formulation allows us to directly tune the relaxation frequencies related to these high-order contributions, eventually leading to an improved stability of the resulting LBM. In addition, the original MRT-LBM further increases the dissipation of acoustic waves, hence reducing their impact on the surrounding fluid dynamics.31,32 Usually, both BGK and MRT LBMs relax to equilibrium states derived through a second-order truncated Taylor expansion in the local Mach number of the above-mentioned continuous Maxwellian distribution. As a consequence, these are Galilean invariant only up to the second-order meaning that velocity-dependent errors of at least third-order are present in the macroscopic equations derived from the Chapman-Enskog expansion.33 Unfortunately, the lack of Galilean invariance at higher-orders leads to numerical instability in the low viscous regime.34

To cope with this problem, Geier et al.36 proposed LBMs based on the concept of central moments (CMs), the latter being obtained by shifting the lattice directions by the local fluid velocity. In actual practice, it is more convenient to operate in terms of RMs, thus requiring a transformation between the two types of moments. This is achieved by using binomial formulas that generate central moments of a certain order as a function of lower-order RMs only.<sup>37</sup> Consequently, the collision in the CM space shows a pyramidal hierarchical topology, where the postcollision state of CMs is constructed starting from the lowest order and then proceeding in an ascending sequence, hence the name "cascaded" lattice Boltzmann method (CLBM). Undoubtedly, the CLBM drastically outperforms both BGK and MRT in terms of stability.3 point can partially be explained by the positive hyperviscosity naturally introduced by CM-LBMs, and that can further be adjusted, through fine tuning of relaxation times related to high-order CMs, in order to find the best compromise between stability and accuracy.

An important step in the design of the CLBM is the match between continuous and discrete central moments, which enforces the Galilean invariance at all orders according to the considered velocity discretization. In other words, CMs are forced to relax to the equilibrium state of the continuous Maxwellian distribution, rather than to its discrete counterpart. Consequently, this methodology assumes that equilibrium CMs are unchanged by the velocity discretization of the Boltzmann equation, which might not be true depending on both equilibrium CMs of interest and on the considered lattice of discrete velocities. In addition, the choice of the set of moments toward which equilibrium CMs are assumed to relax is not known in an *a priori* manner in the above methodology. This prevents any *systematic* derivation of CM-LBMs, especially for high-order lattices whose discrete velocities propagate toward the second layer of neighboring nodes (or even farther).

Recently, we approached central moments from a different viewpoint. 59-62 Given a certain lattice, our methodology consists of building a transformation matrix allowing us to move from the space of populations to one of the central moments and vice versa. The resultant algorithm loses the above-mentioned pyramidal cascaded structure, and as a consequence, it can be interpreted as a "noncascaded" way to apply the collision step in the CM space. 63 In addition, this methodology is based on CMs computed from the polynomial expansion of populations instead of those derived from the continuous ones. Thus, one does not need anymore to choose which continuous CMs should be kept for the collision step. Consequently, the proposed approach allows the derivation of CM-LBMs for any lattice of discrete velocities in a straightforward manner. This was thoroughly demonstrated by successfully recovering different sets of governing equations with this approach, hence allowing the simulation of a rich variety of physics problems such as shallow waters, <sup>64</sup> magnetohydrodynamic, <sup>65</sup> and multicomponent flows<sup>66</sup> among others.

Nonetheless, it is important to understand that the above attempt to derive a CM-based scheme in the D3Q27 space was relying on the classical second-order truncated equilibrium state. 60 Even if the stability of the algorithm was greatly improved, it involved a huge computational cost.<sup>48</sup> This is mainly due to the use of an improper (or incomplete) equilibrium state, which eventually leads to a non-negligible number of nonzero velocity-dependent equilibrium CMs. Indeed, as pointed out by Malaspinas<sup>67</sup> and Coreixas et al., 35,68 the full potential of the D3Q27 velocity space can only be achieved by employing the correct (or complete) equilibrium populations accounting for the whole set of 27 Hermite polynomials. By including these high-order polynomials, the methodology outlined in Ref. 60 was proven to lead to Galilean invariant CMs that were used for the derivation of a force-free D3Q27-CM-LBM.<sup>69</sup> This formulation was further shown to recover the behavior of the original cascaded LBM by only relying on the correct set of Hermite polynomials. Since this family of polynomials is tightly linked to the velocity discretization of the Boltzmann <sup>70,71</sup> the correct set of Hermite polynomials is known in an a priori way, hence ensuring its consistency for any kind of

When it comes to external or internal forces, it is well known that they are ubiquitous in many fluid systems. Indeed, gravity, Coriolis, or Lorentz forces are just few examples of the fields that a fluid might undergo. It is then of paramount importance to properly take into account these forces in the lattice Boltzmann framework. To meet this need, a seminal contribution has been proposed by

Guo et al. 72 in 2002, where the authors analyzed the discrete lattice effects on the forcing term in the BGK-LBM. Later, Guo and Zheng<sup>73</sup> extended the approach to the MRT-LBM. Concerning the adoption of CMs, Premnath and Banerjee derived forcing schemes based on the cascaded approach for two- and three-dimensional simulations.<sup>54,55</sup> In particular, they approximated the forcing term according to the formula in the pioneering work by He et al. 7 In 2017, this forcing term has been adopted by Fei and Luo<sup>47</sup> within a framework where equilibrium CMs were computed from the continuous Maxwellian distributions. The same force treatment has been adopted in Ref. 61, where equilibrium CMs were derived from the classical second-order-truncated expansion. Recently, it has been demonstrated that the forcing term can be written as a Hermite polynomial expansion up to the fourth-order in the D2Q9 space,<sup>75</sup> where the well-known formula by Guo et al.<sup>72</sup> is recovered if Hermite polynomials of order higher than two are neglected.

In this paper, we derive a three-dimensional CM-based model with external forcing in an *a priori* manner for the simulation of multiphysics flows. Following the idea originally proposed by Shan *et al.* <sup>12</sup> and further adopted for the D2Q9 space by Huang *et al.*, <sup>75</sup> the forcing term is written as an expansion of Hermite polynomials. By adopting the very same set of Hermite polynomials for both the equilibrium state and the forcing term, it is thoroughly shown that the corresponding CMs become Galilean invariant, i.e., all remaining error terms depending on the mean flow vanish. As a consequence, the number of nonzero CMs of the forcing term drops to nine, which drastically simplifies its implementation.

The rest of this paper is organized as follows: The adopted methodology is first devised in Sec. II, and the impact of high-order Hermite polynomials on CMs of the forcing term is thoroughly quantified. In Sec. III, several numerical experiments of increasing complexity highlight the consistency, accuracy, and universality of the proposed model. A comparative study with four other forcing schemes is also provided to further evaluate the numerical properties of the present approach as well as the impact of the truncation order of the Hermite polynomial expansion for both the equilibrium and the forcing term. Some conclusions are finally drawn in Sec. IV. For the sake of completeness, details regarding Hermite polynomials, the D2Q9-CM-LBM with external forcing, the lattice Boltzmann method for the magnetic field, and the CM-based color-gradient method are reported in Appendixes A-D, respectively. Moreover, it is recalled in Appendix E that the CMbased LBM can be interpreted as a general MRT LBM, where RMs are obtained as a particular case of the approach outlined in Sec. II.

# II. LATTICE BOLTZMANN SCHEMES WITH HIGHER-ORDER HERMITE POLYNOMIALS

Let us consider an Eulerian basis  $\mathbf{x} = [x, y, z]$  and the D3Q27 space. The lattice Boltzmann equation predicts the space and time evolution of the particle distribution functions  $|f_i\rangle = [f_0, \ldots, f_i, \ldots, f_{26}]^{\mathsf{T}}$  that collide and stream along the generic link  $i = 0, \ldots, 26$  with the discrete velocities  $\mathbf{c}_i = [|c_{ix}\rangle, |c_{iy}\rangle, |c_{iz}\rangle]$  defined as

Let us employ the symbols  $|\cdot\rangle$  and  $\top$  to denote a column vector and the transpose operator, respectively. Within the BGK approximation, <sup>11</sup> the numerical discretization of the lattice Boltzmann equation with external forcing reads as follows:

$$f_{i}(\mathbf{x} + \mathbf{c}_{i}, t + 1) = f_{i}(\mathbf{x}, t) + \omega \left[ f_{i}^{eq}(\mathbf{x}, t) - f_{i}(\mathbf{x}, t) \right] + \left( 1 - \frac{\omega}{2} \right) \mathcal{F}_{i}(\mathbf{x}, t), \tag{2}$$

where the lattice unit system was assumed. As usual, this numerical scheme can be divided into two parts, i.e., collision, <sup>72</sup>

$$f_i^*(\mathbf{x},t) = f_i(\mathbf{x},t) + \omega \left[ f_i^{eq}(\mathbf{x},t) - f_i(\mathbf{x},t) \right] + \left( 1 - \frac{\omega}{2} \right) \mathcal{F}_i(\mathbf{x},t)$$
(3)

and streaming,

$$f_i(\mathbf{x} + \mathbf{c}_i, t + 1) = f_i^*(\mathbf{x}, t), \tag{4}$$

where the superscript \* denotes postcollision quantities here and henceforth. If not otherwise stated, the dependence on the space and

the time will be tacitly assumed in the rest of this paper. The term  $\mathcal{F}_i$  accounts for external body forces  $\mathbf{F} = [F_x, F_y, F_z]$  and it will be discussed later. The fluid density  $\rho$  and velocity  $\mathbf{u} = [u_x, u_y, u_z]$  are computed as<sup>72</sup>

$$\rho = \sum_{i} f_{i},\tag{5}$$

$$\rho \boldsymbol{u} = \sum_{i} f_{i} \boldsymbol{c}_{i} + \frac{\boldsymbol{F}}{2},\tag{6}$$

respectively. The second term of the right-hand side of Eq. (3) is the BGK collision operator, which forces all the populations to relax to a discrete local equilibrium  $f_i^{eq}$  with the same rate  $\omega = \left(v/c_s^2 + 1/2\right)^{-1}$ , with v being the fluid kinematic viscosity. Following the works by Malaspinas and Coreixas et al.,  $^{35,37,68}$  the full potential of the D3Q27 lattice can only be achieved by correctly expanding the equilibrium distribution onto the complete basis composed of 27 Hermite polynomials

$$f_{i}^{eq} = w_{i}\rho \left\{ 1 + \frac{c_{i} \cdot u}{c_{s}^{2}} + \frac{1}{2c_{s}^{4}} \left[ \mathcal{H}_{ixx}^{(2)} u_{x}^{2} + \mathcal{H}_{iyy}^{(2)} u_{y}^{2} + \mathcal{H}_{izz}^{(2)} u_{z}^{2} + 2 \left( \mathcal{H}_{ixy}^{(2)} u_{x} u_{y} + \mathcal{H}_{ixz}^{(2)} u_{x} u_{z} + \mathcal{H}_{iyz}^{(2)} u_{y} u_{z} \right) \right]$$

$$+ \frac{1}{2c_{s}^{6}} \left[ \mathcal{H}_{ixxy}^{(3)} u_{x}^{2} u_{y} + \mathcal{H}_{ixxz}^{(3)} u_{x}^{2} u_{z} + \mathcal{H}_{ixyy}^{(3)} u_{x} u_{y}^{2} + \mathcal{H}_{ixzz}^{(3)} u_{x} u_{z}^{2} + \mathcal{H}_{iyzz}^{(3)} u_{y} u_{z}^{2} + \mathcal{H}_{iyyz}^{(3)} u_{y}^{2} u_{z} + 2 \mathcal{H}_{ixyz}^{(3)} u_{x} u_{y} u_{z} \right]$$

$$+ \frac{1}{4c_{s}^{8}} \left[ \mathcal{H}_{ixxyy}^{(4)} u_{x}^{2} u_{y}^{2} + \mathcal{H}_{ixxzz}^{(4)} u_{x}^{2} u_{z}^{2} + \mathcal{H}_{iyyzz}^{(4)} u_{y}^{2} u_{z}^{2} + 2 \left( \mathcal{H}_{ixyyz}^{(4)} u_{x} u_{y} u_{z}^{2} + \mathcal{H}_{ixyyz}^{(4)} u_{x} u_{y}^{2} u_{z} + \mathcal{H}_{ixxyz}^{(4)} u_{x}^{2} u_{y}^{2} u_{z} \right) \right]$$

$$+ \frac{1}{4c_{s}^{10}} \left[ \mathcal{H}_{ixxyzz}^{(5)} u_{x}^{2} u_{y} u_{z}^{2} + \mathcal{H}_{ixxyyz}^{(5)} u_{x}^{2} u_{y}^{2} u_{z} + \mathcal{H}_{ixxyyzz}^{(5)} u_{x}^{2} u_{y}^{2} u_{z}^{2} \right] + \frac{1}{8c_{s}^{12}} \mathcal{H}_{ixxyyzz}^{(6)} u_{x}^{2} u_{y}^{2} u_{z}^{2} \right\},$$

$$(7)$$

where the weights are  $w_0 = 8/27$ ,  $w_1..._6 = 2/27$ ,  $w_7..._{18} = 1/54$ ,  $w_{19..._{26}} = 1/216$  and  $c_s = 1/\sqrt{3}$  is the lattice sound speed.  $\mathcal{H}_i^{(n)}$  denotes the *n*th-order Hermite polynomial tensor. Coefficients before these tensors are  $(n_x!n_y!n_z!c_s^{2n})^{-1}$ , where  $n_x$ ,  $n_y$ , and  $n_z$  are the number of occurrences of x, y, and z, respectively. Full expressions of the Hermite polynomials are compiled in Appendix A.

Before going any further, it is important to understand that in the present context, the sole purpose of the extended equilibrium state in Eq. (7) is to derive Galilean invariant equilibrium and forcing CMs. Hence, only these CMs are required for the implementation of the D3Q27-CM-LBM with or without external forcing. In addition, one can note that Eq. (7) recovers the classical second-order truncated equilibrium if  $\mathcal{H}_i^{(3)}$ ,  $\mathcal{H}_i^{(4)}$ ,  $\mathcal{H}_i^{(5)}$ , and  $\mathcal{H}_i^{(6)}$  are neglected. In addition, one can show that the better macroscopic behavior recovered by sophisticated collision models such as cumulant and recursive regularized 55,67,68 based LBMs is only due to this extended equilibrium. Hence, it is sufficient to rely on Eq. (7) if one wants to get rid of nondiagonal velocity-dependent errors in the viscous stress tensor, whereas one might want to change the collision model in order to increase the numerical stability.  $^{37}$ 

Regarding the forcing term, in the continuous Boltzmann equation, it is expressed as  $\mathcal{F} = -\mathbf{F} \cdot \nabla_{\mathbf{c}} f$ , where  $\nabla_{\mathbf{c}}$  is the derivative with respect to the mesoscopic velocity  $\mathbf{c}$ . Based on the

Chapman-Enskog expansion, the Navier-Stokes-Fourier equations can be recovered if the force term is approximated as  $-\mathbf{F} \cdot \nabla_c f^{eq}$  since its nonequilibrium contribution only impacts high-order physics (Burnett, Super Burnett, etc.). Then, it is possible to write a Hermite expansion of this forcing term based on the expansion of the equilibrium state 12

$$\mathcal{F} = -\mathbf{F} \cdot \nabla_{\mathbf{c}} f^{eq}$$

$$= -\mathbf{F} \cdot \nabla_{\mathbf{c}} \left( w(\mathbf{c}) \sum_{n=0}^{\infty} \frac{1}{n!} \mathbf{a}_{eq}^{(n)} \cdot \mathcal{H}^{(n)} \right)$$

$$= -\mathbf{F} \cdot \sum_{n=0}^{\infty} \frac{1}{n!} \mathbf{a}_{eq}^{(n)} \cdot \nabla_{\mathbf{c}} \left( w(\mathbf{c}) \mathcal{H}^{(n)} \right)$$

$$= \omega(\mathbf{c}) \sum_{n=1}^{\infty} \frac{1}{n!} \rho \left[ \mathbf{F} \mathbf{u}^{(n-1)} \right] \cdot \mathcal{H}^{(n)}, \tag{8}$$

where both Rodrigues' formula and  $\mathbf{a}_{eq}^{(n-1)} = \rho \mathbf{u}^{(n-1)}$  have been used.<sup>67</sup>  $\mathbf{u}^{(n-1)}$  is the velocity tensor of rank (n-1). The square bracket  $[\cdot]$  stands for cyclic permutations, e.g., for n=3,

$$[Fuu] = Fuu + uFu + uuF.$$

By adopting the extended equilibrium state related to the D3Q27 lattice [see Eq. (7)], the expansion of the *complete* forcing term reads as

$$\mathcal{F}_{i} = w_{i}\rho \left\{ \frac{F \cdot c_{i}}{c_{s}^{2}} + \frac{1}{2c_{s}^{4}} \left[ \mathcal{H}_{ixx}^{(2)}(2u_{x}F_{x}) + \mathcal{H}_{iyy}^{(2)}(2u_{y}F_{y}) + \mathcal{H}_{izz}^{(2)}(2u_{z}F_{z}) + 2\mathcal{H}_{ixy}^{(2)}(u_{x}F_{y} + u_{y}F_{x}) + 2\mathcal{H}_{ixz}^{(2)}(u_{x}F_{z} + u_{z}F_{x}) \right. \\
\left. + 2\mathcal{H}_{iyz}^{(2)}(u_{y}F_{z} + u_{z}F_{y}) \right] + \frac{1}{2c_{s}^{6}} \left[ \mathcal{H}_{ixxy}^{(3)}(2u_{x}u_{y}F_{x} + u_{x}^{2}F_{y}) + \mathcal{H}_{ixxz}^{(3)}(2u_{x}u_{z}F_{x} + u_{x}^{2}F_{z}) + \mathcal{H}_{ixyy}^{(3)}(2u_{x}u_{y}F_{y} + u_{y}^{2}F_{x}) \right. \\
\left. + \mathcal{H}_{ixzz}^{(3)}(2u_{x}u_{z}F_{z} + u_{z}^{2}F_{x}) + \mathcal{H}_{iyzz}^{(3)}(2u_{y}u_{z}F_{z} + u_{z}^{2}F_{y}) + \mathcal{H}_{iyyz}^{(3)}(2u_{y}u_{z}F_{y} + u_{y}^{2}F_{z}) + 2\mathcal{H}_{ixyz}^{(3)}(u_{x}u_{y}F_{z} + u_{x}F_{y}u_{z} \right. \\
\left. + F_{x}u_{y}u_{z} \right] + \frac{1}{4c_{s}^{6}} \left[ \mathcal{H}_{ixxyy}^{(4)}(2u_{x}^{2}u_{y}F_{y} + 2u_{x}u_{y}^{2}F_{x}) + \mathcal{H}_{ixxzz}^{(4)}(2u_{x}^{2}u_{z}F_{z} + 2u_{x}u_{z}^{2}F_{x}) + \mathcal{H}_{iyyzz}^{(4)}(2u_{y}^{2}u_{z}F_{z} + 2u_{y}u_{z}^{2}F_{y}) \right. \\
\left. + 2\mathcal{H}_{ixyzz}^{(4)}(2u_{x}u_{y}u_{z}F_{z} + u_{x}F_{y}u_{z}^{2} + F_{x}u_{y}u_{z}^{2}) + 2\mathcal{H}_{ixyyz}^{(5)}(2u_{x}u_{y}u_{z}F_{y} + u_{x}u_{y}^{2}F_{z} + F_{x}u_{y}^{2}u_{z}) \right. \\
\left. + 2\mathcal{H}_{ixxyz}^{(5)}(2u_{x}u_{y}u_{z}F_{x} + u_{y}u_{x}^{2}F_{z} + u_{x}^{2}u_{z}F_{y}) + \mathcal{H}_{ixyyzz}^{(5)}(2u_{x}u_{y}^{2}u_{z}F_{x} + 2u_{x}u_{y}u_{z}F_{y} + u_{x}^{2}u_{y}^{2}F_{z}) \right. \\
\left. + \mathcal{H}_{ixxyzz}^{(5)}(2u_{x}u_{y}u_{z}F_{x} + u_{y}^{2}u_{z}^{2}F_{y} + 2u_{x}^{2}u_{y}u_{z}F_{z}) + \mathcal{H}_{ixyyzz}^{(5)}(2u_{x}u_{y}^{2}u_{z}F_{x} + 2u_{x}u_{y}u_{z}F_{y} + 2u_{x}u_{y}^{2}u_{z}F_{z}) \right] \\
\left. + \frac{1}{8c_{z}^{12}}\mathcal{H}_{ixxyyzz}^{(6)}(2u_{x}u_{y}^{2}u_{z}^{2}F_{x} + 2u_{x}^{2}u_{y}u_{z}^{2}F_{y} + 2u_{x}^{2}u_{y}^{2}u_{z}F_{z}) \right\},$$

$$(9)$$

where  $1/c_s^{2n}$  terms appear due to the use of *discrete* Hermite polynomials  $\mathcal{H}_i^{(n)}$ . Once again, the full form in Eq. (9) is only used to derive Galilean invariant CMs of the forcing term. As a consequence, only these CMs are required for the implementation of the D3Q27-CM-LBM with external forcing, while the extended equilibrium state only serves a theoretical purpose in the present context. Indeed, by following the methodology proposed by Coreixas  $et\ al.$ ,  $^{37}$  it is possible to extend the above forcing term to any collision model, and this is illustrated in Appendix E for the RM-based MRT collision model. Regarding the forcing term itself, one can note that the popular formula by Guo  $et\ al.$  is recovered if terms proportional to  $\mathcal{H}_i^{(3)}$ ,  $\mathcal{H}_i^{(4)}$ ,  $\mathcal{H}_i^{(5)}$ , and  $\mathcal{H}_i^{(6)}$  are neglected.

Now, let us focus on central moments. The pillar to design

Now, let us focus on central moments. The pillar to design any CM-based collision operator is to shift the lattice directions by the local fluid velocity.<sup>36</sup> Therefore, it is possible to define  $\bar{c}_i = [|\bar{c}_{ix}\rangle, |\bar{c}_{iy}\rangle, |\bar{c}_{iz}\rangle]$ , where

$$|\bar{c}_{ix}\rangle = |c_{ix} - u_x\rangle, \quad |\bar{c}_{iy}\rangle = |c_{iy} - u_y\rangle, \quad |\bar{c}_{iz}\rangle = |c_{iz} - u_z\rangle.$$
 (10)

One possibility is to adopt the basis proposed in Ref. 60, which is  $\overline{T} = [\overline{T}_0, \dots, \overline{T}_i, \dots, \overline{T}_{26}]$  with

$$\begin{split} |\overline{T}_{0}\rangle &= |1, \ldots, 1\rangle, \\ |\overline{T}_{1}\rangle &= |\overline{c}_{ix}\rangle, \\ |\overline{T}_{2}\rangle &= |\overline{c}_{iy}\rangle, \\ |\overline{T}_{3}\rangle &= |\overline{c}_{iz}\rangle, \\ |\overline{T}_{4}\rangle &= |\overline{c}_{ix}\overline{c}_{iy}\rangle, \\ |\overline{T}_{5}\rangle &= |\overline{c}_{ix}\overline{c}_{iy}\rangle, \\ |\overline{T}_{5}\rangle &= |\overline{c}_{ix}\overline{c}_{iz}\rangle, \\ |\overline{T}_{6}\rangle &= |\overline{c}_{iy}\overline{c}_{iz}\rangle, \\ |\overline{T}_{7}\rangle &= |\overline{c}_{ix}^{2} - \overline{c}_{iy}^{2}\rangle, \\ |\overline{T}_{8}\rangle &= |\overline{c}_{ix}^{2} - \overline{c}_{iz}^{2}\rangle, \\ |\overline{T}_{9}\rangle &= |\overline{c}_{ix}^{2} + \overline{c}_{iy}^{2} + \overline{c}_{iz}^{2}\rangle, \\ |\overline{T}_{10}\rangle &= |\overline{c}_{ix}\overline{c}_{iy}^{2} + \overline{c}_{ix}\overline{c}_{iz}^{2}\rangle, \\ |\overline{T}_{11}\rangle &= |\overline{c}_{ix}^{2}\overline{c}_{iy} + \overline{c}_{iy}\overline{c}_{iz}^{2}\rangle, \\ |\overline{T}_{12}\rangle &= |\overline{c}_{ix}^{2}\overline{c}_{iy} + \overline{c}_{iy}\overline{c}_{iz}^{2}\rangle, \\ |\overline{T}_{13}\rangle &= |\overline{c}_{ix}\overline{c}_{iy}^{2} - \overline{c}_{ix}\overline{c}_{iz}^{2}\rangle, \\ |\overline{T}_{14}\rangle &= |\overline{c}_{ix}^{2}\overline{c}_{iy} - \overline{c}_{iy}\overline{c}_{iz}^{2}\rangle, \\ |\overline{T}_{15}\rangle &= |\overline{c}_{ix}^{2}\overline{c}_{iy} - \overline{c}_{iy}\overline{c}_{iz}\rangle, \\ |\overline{T}_{17}\rangle &= |\overline{c}_{ix}^{2}\overline{c}_{iy}^{2} + \overline{c}_{ix}^{2}\overline{c}_{iz}^{2} + \overline{c}_{iy}^{2}\overline{c}_{iz}^{2}\rangle, \\ |\overline{T}_{17}\rangle &= |\overline{c}_{ix}^{2}\overline{c}_{iy}^{2} + \overline{c}_{ix}^{2}\overline{c}_{iz}^{2} - \overline{c}_{iy}^{2}\overline{c}_{iz}^{2}\rangle, \\ |\overline{T}_{19}\rangle &= |\overline{c}_{ix}^{2}\overline{c}_{iy}^{2} - \overline{c}_{ix}^{2}\overline{c}_{iz}^{2}\rangle, \\ |\overline{T}_{20}\rangle &= |\overline{c}_{ix}^{2}\overline{c}_{iy}\overline{c}_{iz}\rangle, \\ |\overline{T}_{21}\rangle &= |\overline{c}_{ix}\overline{c}_{iy}\overline{c}_{iz}\rangle, \\ |\overline{T}_{22}\rangle &= |\overline{c}_{ix}\overline{c}_{iy}\overline{c}_{iz}^{2}\rangle, \\ |\overline{T}_{22}\rangle &= |\overline{c}_{ix}\overline{c}_{iy}\overline{c}_{iz}^{2}\rangle, \\ \end{aligned}$$

$$|\overline{T}_{23}\rangle = |\overline{c}_{ix}\overline{c}_{iy}^2\overline{c}_{iz}^2\rangle,$$

$$|\overline{T}_{24}\rangle = |\overline{c}_{ix}^2\overline{c}_{iy}\overline{c}_{iz}^2\rangle,$$

$$|\overline{T}_{25}\rangle = |\overline{c}_{ix}^2\overline{c}_{iy}^2\overline{c}_{iz}\rangle,$$

$$|\overline{T}_{26}\rangle = |\overline{c}_{ix}^2\overline{c}_{iy}^2\overline{c}_{iz}\rangle.$$
(11)

Let us first recall important results concerning the force-free CM-LBM. Precollision, equilibrium, and postcollision CMs are defined as

$$|k_i\rangle = [k_0, \dots, k_i, \dots, k_{26}]^{\mathsf{T}},$$

$$|k_i^{eq}\rangle = [k_0^{eq}, \dots, k_i^{eq}, \dots, k_{26}^{eq}]^{\mathsf{T}},$$

$$|k_i^{\star}\rangle = [k_0^{\star}, \dots, k_i^{\star}, \dots, k_{26}^{\star}]^{\mathsf{T}},$$

$$(12)$$

respectively. The first two quantities are evaluated by applying the transformation matrix  $\overline{\mathcal{T}}$  to the corresponding distributions, i.e.,

$$|k_i\rangle = \overline{\mathcal{T}}^{\mathsf{T}}|f_i\rangle, \quad |k_i^{eq}\rangle = \overline{\mathcal{T}}^{\mathsf{T}}|f_i^{eq}\rangle,$$
 (13)

where  $|f_i^{eq}\rangle = [f_0^{eq}, \dots, f_i^{eq}, \dots, f_{26}^{eq}]^{\mathsf{T}}$ . By adopting a Hermite polynomial expansion of the equilibrium state up to  $\mathcal{H}_i^{(6)}$  in Eq. (7), equilibrium CMs read as follows:

$$k_0^{eq} = \rho, \quad k_9^{eq} = 3\rho c_s^2, \quad k_{17}^{eq} = \rho c_s^2,$$

$$k_{18}^{eq} = \rho c_s^4, \quad k_{26}^{eq} = \rho c_s^6,$$
(14)

with  $k_{1...8}^{eq} = k_{10...16}^{eq} = k_{19...25}^{eq} = 0$ . Notably, only five equilibrium CMs assume nonzero values, and they have the same form as those of the continuous Maxwellian. In other words, Galilean invariant equilibrium CMs are *naturally* obtained when the full set of Hermite polynomials is considered, as originally demonstrated in Ref. 69. This is in agreement with the fact that all collision models are based on the extended equilibrium in Eq. (7) when high-order velocity terms are not neglected. Hence, all collision models should recover the same level of Galilean invariance, even if their corresponding equilibrium moments might depend on velocity terms, which might have an impact on their numerical stability.<sup>37</sup>

The postcollision CMs can be written as

$$|k_i^{\star}\rangle = (\mathbf{I} - \mathbf{\Lambda})\overline{\mathcal{T}}^{\mathsf{T}}|f_i\rangle + \mathbf{\Lambda}\overline{\mathcal{T}}^{\mathsf{T}}|f_i^{eq}\rangle$$
 (15)

or

$$|k_i^{\star}\rangle = (\mathbf{I} - \mathbf{\Lambda})|k_i\rangle + \mathbf{\Lambda}|k_i^{eq}\rangle,$$
 (16)

where **I** is the  $27 \times 27$  unit tensor and

$$\Lambda = \text{diag}[1, 1, 1, 1, \omega, \omega, \omega, \omega, \omega, 1, ..., 1]$$
 (17)

is the  $27 \times 27$  relaxation matrix adopted in this work. The latter allows us to increase the numerical stability through (1) the overdissipation of acoustic waves  $^{76}$  and (2) the equilibration of high-order

moments. 30,77 The postcollision state of CMs then becomes

$$k_{0}^{*} = \rho,$$

$$k_{4}^{*} = (1 - \omega)k_{4},$$

$$k_{5}^{*} = (1 - \omega)k_{5},$$

$$k_{6}^{*} = (1 - \omega)k_{6},$$

$$k_{7}^{*} = (1 - \omega)k_{7},$$

$$k_{8}^{*} = (1 - \omega)k_{8},$$

$$k_{9}^{*} = 3\rho c_{s}^{2},$$

$$k_{17}^{*} = \rho c_{s}^{2},$$

$$k_{18}^{*} = \rho c_{s}^{4},$$

$$k_{26}^{*} = \rho c_{s}^{6},$$

$$(18)$$

where

$$k_{4} = \sum_{i} f_{i} \bar{c}_{ix} \bar{c}_{iy}, \quad k_{5} = \sum_{i} f_{i} \bar{c}_{ix} \bar{c}_{iz}, \quad k_{6} = \sum_{i} f_{i} \bar{c}_{iy} \bar{c}_{iz},$$

$$k_{7} = \sum_{i} f_{i} (\bar{c}_{ix}^{2} - \bar{c}_{iy}^{2}), \quad k_{8} = \sum_{i} f_{i} (\bar{c}_{ix}^{2} - \bar{c}_{iz}^{2}),$$
(19)

and  $k_{1...3}^{\star} = k_{10...16}^{\star} = k_{19...25}^{\star} = 0$ . Eventually, the postcollision populations are reconstructed as

$$R_{1} = F_{x},$$

$$R_{2} = F_{y},$$

$$R_{3} = F_{z},$$

$$R_{10} = 2F_{x}c_{x}^{2} - F_{x}u_{y}^{2} - 2F_{y}u_{x}u_{y} - F_{x}u_{z}^{2} - 2F_{z}u_{x}u_{z},$$

$$R_{11} = 2F_{y}c_{x}^{2} - F_{y}u_{x}^{2} - 2F_{y}u_{x}u_{y} - F_{x}u_{z}^{2} - 2F_{z}u_{y}u_{z},$$

$$R_{11} = 2F_{y}c_{x}^{2} - F_{z}u_{x}^{2} - 2F_{x}u_{y}u_{x} - F_{y}u_{z}^{2} - 2F_{z}u_{y}u_{z},$$

$$R_{12} = 2F_{z}c_{y}^{2} - F_{z}u_{x}^{2} - 2F_{y}u_{x}u_{y} + F_{x}u_{z}^{2} + 2F_{z}u_{x}u_{z},$$

$$R_{13} = -F_{x}u_{y}^{2} - 2F_{y}u_{x}u_{y} + F_{y}u_{z}^{2} + 2F_{z}u_{y}u_{z},$$

$$R_{14} = -F_{y}u_{x}^{2} - 2F_{x}u_{y}u_{x} + F_{y}u_{z}^{2} + 2F_{z}u_{y}u_{z},$$

$$R_{15} = -F_{z}u_{x}^{2} - 2F_{x}u_{z}u_{x} + F_{z}u_{y}^{2} + 2F_{y}u_{z}u_{y},$$

$$R_{16} = -F_{z}u_{x}u_{y} - F_{y}u_{x}u_{z} - F_{x}u_{y}u_{z},$$

$$R_{17} = 4F_{y}u_{x}^{2}u_{y} + 4F_{z}u_{x}^{2}u_{z} + 4F_{x}u_{x}u_{y}^{2} + 4F_{x}u_{x}u_{z}^{2} + 4F_{z}u_{y}^{2}u_{z} - 4F_{y}u_{y}u_{z}^{2},$$

$$R_{18} = 4F_{y}u_{x}^{2}u_{y} + 4F_{z}u_{x}^{2}u_{z} + 4F_{x}u_{x}u_{y}^{2} + 4F_{z}u_{x}u_{z}^{2} - 4F_{z}u_{y}^{2}u_{z} - 4F_{y}u_{y}u_{z}^{2},$$

$$R_{19} = 4u_{x}(F_{x}u_{y}^{2} + F_{y}u_{x}u_{y} - F_{x}u_{z}^{2} - F_{z}u_{x}u_{z}),$$

$$R_{20} = 2u_{x}(F_{z}u_{x}u_{y} + F_{y}u_{x}u_{z} + F_{x}u_{y}u_{z}),$$

$$R_{21} = 2u_{y}(F_{z}u_{x}u_{y} + F_{y}u_{x}u_{z} + F_{x}u_{y}u_{z}),$$

$$R_{22} = 2u_{z}(2F_{z}u_{x}u_{y} + F_{y}u_{x}u_{z} + F_{x}u_{y}u_{z}),$$

$$R_{23} = F_{x}^{4} - 3F_{x}u_{y}^{2}u_{z}^{2} - 6F_{z}u_{x}u_{y}^{2}u_{z} - F_{x}u_{y}^{2}/3 - 6F_{y}u_{x}u_{y}u_{z}^{2} - 2F_{y}u_{x}u_{y}/3 - F_{x}u_{z}^{2}/3 - 2F_{z}u_{x}u_{z}/3,$$

$$R_{24} = F_{y}c_{x}^{4} - 3F_{x}u_{y}^{2}u_{z}^{2} - 6F_{y}u_{z}u_{x}^{2}u_{z} - F_{y}u_{x}^{2}/3 - 6F_{x}u_{x}u_{x}u_{y}^{2} - 2F_{x}u_{x}u_{x}/3 - F_{y}u_{z}^{2}/3 - 2F_{z}u_{y}u_{z}/3,$$

$$R_{25} = F_{z}c_{x}^{4} - 3F_{x}u_{x}^{2}u_{y}^{2} - 6F_{y}u_{z}u_{x}^{2}u_{y} - F_{z}u_{x}^{2}/3 - 6F_{x}u_{x}u_{x}u_{y}^{2} - 2F_{x}u_{x}u_{x}/3 - F_{y}u_{z}^{2}/3 - 2F_{z}u_{y}u_{z}/3,$$

$$R_{26} = 8(F_{z}u_{x}^{2}u_{y}^{2}u_{x}^{2} + F_{y}u_{x}^{2}u_{y}^{2} + 4F_{y}u_{y}^{2}u_{y}^{2}) + 4G_{y}^{2}(F_$$

$$|f_i^{\star}\rangle = \left(\overline{\mathcal{T}}^{\mathsf{T}}\right)^{-1}|k_i^{\star}\rangle,\tag{20}$$

with  $|f_i^{\star}\rangle = [f_0^{\star}, \dots, f_i^{\star}, \dots, f_{26}^{\star}]^{\mathsf{T}}$ , before being streamed [see Eq. (4)]. One can note that the computation of only five precollision CMs and ten postcollision CMs is required in this work. Hence, not only the two previous choices [equilibrium in Eq. (7) and relaxation matrix in Eq. (17)] lead to a robust and accurate CM-LBM, <sup>69</sup> but they also lead to a very concise formulation.

In the presence of an external force, the postcollision CMs can be rewritten as

$$|k_i^{\star}\rangle = (\mathbf{I} - \mathbf{\Lambda})\overline{\mathcal{T}}^{\mathsf{T}}|f_i\rangle + \mathbf{\Lambda}\overline{\mathcal{T}}^{\mathsf{T}}|f_i^{eq}\rangle + (\mathbf{I} - \frac{\mathbf{\Lambda}}{2})\overline{\mathcal{T}}^{\mathsf{T}}|\mathcal{F}_i\rangle$$
 (21)

or

$$|k_i^{\star}\rangle = (\mathbf{I} - \mathbf{\Lambda})|k_i\rangle + \mathbf{\Lambda}|k_i^{eq}\rangle + \left(\mathbf{I} - \frac{\mathbf{\Lambda}}{2}\right)|R_i\rangle,$$
 (22)

where the CMs of the forcing term are

$$|R_i\rangle = \overline{\mathcal{T}}^{\mathsf{T}}|\mathcal{F}_i\rangle.$$
 (23)

Now, let us proceed by assuming the presence of Hermite polynomials of progressively higher order. If only  $\mathcal{H}_i^{(1)}$  and  $\mathcal{H}_i^{(2)}$  are considered, Eq. (8) leads to the following CMs for the forcing term:

(24)

with  $R_0 = R_{4...9} = 0$ . One can note that most of these CMs are not Galilean invariant since they contain velocity-dependent terms. By further taking into account the third-order Hermite polynomials, terms proportional to the square of the velocity vanish, and the corresponding CMs become

$$R_{1} = F_{x},$$

$$R_{2} = F_{y},$$

$$R_{3} = F_{z},$$

$$R_{10} = 2F_{x}c_{s}^{2},$$

$$R_{11} = 2F_{y}c_{s}^{2},$$

$$R_{12} = 2F_{z}c_{s}^{2},$$

$$R_{17} = -2F_{y}u_{x}^{2}u_{y} - 2F_{z}u_{x}^{2}u_{z} - 2F_{x}u_{x}u_{y}^{2} - 2F_{x}u_{x}u_{z}^{2}$$

$$- 2F_{z}u_{y}^{2}u_{z} - 2F_{y}u_{y}u_{z}^{2},$$

$$R_{18} = -2F_{y}u_{x}^{2}u_{y} - 2F_{z}u_{x}^{2}u_{z} - 2F_{x}u_{x}u_{y}^{2} - 2F_{x}u_{x}u_{z}^{2}$$

$$+ 2F_{z}u_{y}^{2}u_{z} + 2F_{y}u_{y}u_{z}^{2},$$

$$R_{19} = -2u_{x}(F_{x}u_{y}^{2} + F_{y}u_{x}u_{y} - F_{x}u_{z}^{2} - F_{z}u_{x}u_{z}),$$

$$R_{20} = -u_{x}(F_{z}u_{x}u_{y} + F_{y}u_{x}u_{z} + 2F_{x}u_{y}u_{z}),$$

$$R_{21} = -u_{y}(F_{z}u_{x}u_{y} + 2F_{y}u_{x}u_{z} + F_{x}u_{y}u_{z}),$$

$$R_{22} = -u_{z}(2F_{z}u_{x}u_{y} + F_{y}u_{x}u_{z} + F_{x}u_{y}u_{z}),$$

$$R_{23} = F_{x}c_{s}^{4} + 3F_{x}u_{y}^{2}u_{z}^{2} + 6F_{z}u_{x}u_{y}^{2}u_{z} + 6F_{y}u_{x}u_{y}u_{z}^{2},$$

$$R_{24} = F_{y}c_{s}^{4} + 3F_{y}u_{x}^{2}u_{z}^{2} + 6F_{z}u_{y}u_{x}^{2}u_{z} + 6F_{x}u_{y}u_{x}u_{z}^{2},$$

$$R_{25} = F_{z}c_{s}^{4} + 3F_{z}u_{x}^{2}u_{y}^{2} + 6F_{y}u_{z}u_{x}^{2}u_{y} + 6F_{x}u_{z}u_{x}u_{y}^{2},$$

$$R_{26} = -12(F_{z}u_{x}^{2}u_{y}^{2}u_{z} - F_{y}u_{x}^{2}u_{y}u_{z}^{2} - F_{x}u_{x}u_{y}^{2}u_{z}^{2})$$

$$+ 2c_{s}^{2}(F_{y}u_{x}^{2}u_{y} - F_{z}u_{x}^{2}u_{z} - F_{x}u_{x}u_{y}^{2} - F_{x}u_{x}u_{z}^{2}$$

$$- F_{z}u_{y}^{2}u_{z} - F_{y}u_{y}u_{z}^{2}),$$

with now  $R_0 = R_{4...9} = R_{13...16} = 0$ . If  $\mathcal{H}_i^{(4)}$  polynomials are further considered in the definition of the forcing term in Eq. (8), we obtain

$$R_{1} = F_{x},$$

$$R_{2} = F_{y},$$

$$R_{3} = F_{z},$$

$$R_{10} = 2F_{x}c_{s}^{2},$$

$$R_{11} = 2F_{y}c_{s}^{2},$$

$$R_{12} = 2F_{z}c_{s}^{2},$$

$$R_{23} = F_{x}c_{s}^{4} - F_{x}u_{y}^{2}u_{z}^{2} - 2F_{z}u_{x}u_{y}^{2}u_{z} - 2F_{y}u_{x}u_{y}u_{z}^{2},$$

$$R_{24} = F_{y}c_{s}^{4} - F_{y}u_{x}^{2}u_{z}^{2} - 2F_{z}u_{y}u_{x}^{2}u_{z} - 2F_{x}u_{y}u_{x}u_{z}^{2},$$

$$R_{25} = F_{z}c_{s}^{4} - F_{z}u_{x}^{2}u_{y}^{2} - 2F_{y}u_{z}u_{x}^{2}u_{y} - 2F_{x}u_{z}u_{x}u_{y}^{2},$$

$$R_{26} = 8u_{x}u_{y}u_{z}(F_{z}u_{x}u_{y} + F_{y}u_{x}u_{z} + F_{x}u_{y}u_{z}),$$
(26)

with 
$$R_0 = R_{4...9} = R_{13...22} = 0$$
. By adding  $\mathcal{H}_i^{(5)}$ , CMs are
$$R_1 = F_x,$$

$$R_2 = F_y,$$

$$R_3 = F_z,$$

$$R_{10} = 2F_x c_s^2,$$

$$R_{11} = 2F_y c_s^2,$$

$$R_{12} = 2F_z c_s^2,$$

$$R_{23} = F_x c_s^4,$$

$$R_{24} = F_y c_s^4,$$

$$R_{25} = F_z c_s^4,$$

$$R_{26} = -2u_x u_y u_z (F_z u_x u_y + F_y u_x u_z + F_x u_y u_z).$$
(27)

Finally, by matching the Hermite polynomial expansion of the forcing term to one of the extended equilibrium states in Eq. (7), the following expressions are derived:

$$R_{1} = F_{x},$$

$$R_{2} = F_{y},$$

$$R_{3} = F_{z},$$

$$R_{10} = 2F_{x}c_{s}^{2},$$

$$R_{11} = 2F_{y}c_{s}^{2},$$

$$R_{12} = 2F_{z}c_{s}^{2},$$

$$R_{23} = F_{x}c_{s}^{4},$$

$$R_{24} = F_{y}c_{s}^{4},$$

$$R_{25} = F_{z}c_{s}^{4},$$
(28)

with  $R_0 = R_{4...9} = R_{13...22} = R_{26} = 0$  and where all the velocity-dependent terms have vanished, eventually leading to a fully Galilean invariant forcing term.

Before going any further, let us point out some interesting properties of the present approach. First, the same expressions of  $|\mathcal{R}_i\rangle$  in Eq. (28) can be achieved when CMs are computed from the continuous Maxwellian distribution.<sup>48</sup> This is consistent with a recent paper by De Rosis and Luo, <sup>69</sup> where it is shown that the CMs of the discrete equilibrium distribution reduce to those of the continuous counterpart when the full set of Hermite polynomials is considered. Hence, the proposed approach consistently leads to Galilean invariant forcing terms in accordance with the velocity discretization of interest. This methodology can then be extended to any kind of lattices in a straightforward and a priori manner, i.e., without having to rely on the CMs of the Maxwellian equilibrium state. Second, as an alternative to the present Hermite expansion of the forcing term, the latter could have been expressed by the formula  $F \cdot (c_i - u) f_i^{eq} / (\rho c_s^2)$  in Ref. 74. However, we find that, in this case, the CMs of the force show a dependence on the third power of the velocity. This is due to the fact that the forcing term  $-F \cdot \nabla_{c_i} f^{eq}$  is only approximated as  $-F \cdot \frac{c_i - u}{c_s^2} f^{eq}$  in the pioneering work by He et al. 74 This is consistent with the analysis presented in Ref. 75 for the D2Q9 space, and such a problem is avoided in the present work, thanks to the extended Hermite polynomial expansion.

Eventually, postcollision CMs accounting for the sixth-order Hermite polynomials in both the equilibrium populations and forcing term are

$$k_{0}^{*} = \rho,$$

$$k_{1}^{*} = F_{x}/2,$$

$$k_{2}^{*} = F_{y}/2,$$

$$k_{3}^{*} = F_{z}/2,$$

$$k_{4}^{*} = (1 - \omega)k_{4},$$

$$k_{5}^{*} = (1 - \omega)k_{5},$$

$$k_{6}^{*} = (1 - \omega)k_{6},$$

$$k_{7}^{*} = (1 - \omega)k_{8},$$

$$k_{9}^{*} = 3\rho c_{s}^{2},$$

$$k_{10}^{*} = F_{x}c_{s}^{2},$$

$$k_{11}^{*} = F_{y}c_{s}^{2},$$

$$k_{12}^{*} = F_{z}c_{s}^{2},$$

$$k_{13}^{*} = \rho c_{s}^{4},$$

$$k_{23}^{*} = F_{x}c_{s}^{4}/2,$$

$$k_{24}^{*} = F_{y}c_{s}^{4}/2,$$

$$k_{25}^{*} = F_{z}c_{s}^{4}/2,$$

$$k_{26}^{*} = \rho c_{s}^{6},$$

$$(29)$$

with  $k_{13...16}^{\star} = k_{19...22}^{\star} = 0$ . Once again, only five precollision CMs need to be computed in this work. Nevertheless, the inclusion of the forcing term now requires the computation of nine more postcollision CMs, and this brings the number of nonzero postcollision CMs to 19 instead of ten for the force-free case in Eq. (18). It is important to note that by adopting central-Hermite moments (CHMs) instead of the more conventional CMs, it might even be possible to reduce the number of nonzero terms to nine and six for CHM-LBMs with and without the forcing term, respectively. This is explained by the fact that the CHM framework is the only one that makes  $c_s$ -dependent terms vanish, while the cumulant based approach still contains some of them.<sup>37</sup> The behavior of CHM-LBMs is currently being investigated, and the corresponding results will be presented in a forthcoming paper.

To conclude, a time iteration of the present algorithm can be summarized as follows:

- 1. Computation of macroscopic variables by Eq. (6);
- 2. Evaluation of precollision CMs  $k_{4...8}$  by Eq. (19);
- 3. Calculation of postcollision CMs as in Eq. (29);
- 4. Reconstruction of postcollision populations by Eq. (20);
- 5. Streaming [see Eq. (4)].

For the sake of completeness, a script is provided to allow the reader to perform all the symbolic manipulations required for the implementation of the present approach (supplementary material).

### **III. NUMERICAL TESTS**

In this section, the accuracy, stability, and universality of the present approach are discussed using several numerical test cases. Unless differently specified, the density field is initialized as  $\rho(\mathbf{x},t=0)=\rho_0$ , with  $\rho_0=1$ , and the fluid is initially at rest, i.e.,  $\mathbf{u}(\mathbf{x},t=0)=0$ . Velocity boundary conditions are assigned through the regularized technique. The Six test cases of increasing complexity are investigated: (1) four-rolls mill, (2) Hartmann flow, (3) two-dimensional Orszag-Tang vortex, (4) three-dimensional Orszag-Tang vortex, (5) static droplet, and (6) the Rayleigh-Taylor instability.

The first three tests are two-dimensional (i.e., only one point is used for the spatial discretization in the z direction), while the last one is a fully three-dimensional case. In the first scenario, the effect of a force driving the flow is simulated. In the second, third, and fourth ones, an electrically conductive fluid is considered and the proposed treatment of the forcing scheme is applied to the Lorentz force. A self-consistent internal force is considered in order to simulate a two-component flow in the fifth scenario. Finally, the flexibility of the present approach to deal with more sophisticated LB models is shown through simulations of the Rayleigh-Taylor instability by means of the color-gradient method.

In all the tests, a diffusive scaling is adopted for the computation of the time step  $\Delta t$ , i.e.,  $\Delta t \propto (\Delta x)^2$  with  $\Delta x$  being the space step. Hence, the focus will be put on the incompressible behavior of the proposed methodology, which is a common approximation for the simulation of low-Mach number flows. In addition, all quantities are expressed in lattice units hereafter.

To further evaluate the robustness and the accuracy of the present method (referred to as  $M_1$ ), a comparative study with four other forcing schemes is conducted:

- M<sub>2</sub>: a recent effort by Fei and Luo,<sup>47</sup> where equilibrium CMs are derived from the continuous Maxwellian distribution (equivalent to the six-order truncated expansion of the discrete equilibrium state), and CMs of the forcing terms are computed from the approximation by He *et al.*;<sup>74</sup>
- M<sub>3</sub>: the methodology discussed in Ref. 61, where only a second-order truncated expansion is used for the discrete equilibrium populations, while the forcing scheme is the same as for M<sub>2</sub> (i.e., Ref. 74);
- M<sub>4</sub>: a model using the sixth-order truncated expansion for the discrete equilibrium populations and a second-order force treatment (i.e., Ref. 72);
- M<sub>5</sub>: a model using the second-order truncated expansion for both the discrete equilibrium populations and the forcing term (i.e., Ref. 72).

A synthetic overview of the considered models is given in Table I.

Comparisons between methods  $M_1$ ,  $M_4$ , and  $M_5$  are proposed to better understand the contribution of each high-order modification (equilibrium or forcing term) in the context of the Hermite polynomial expansion.  $M_2$  and  $M_4$ , as well as  $M_3$  and  $M_5$ , further highlight the impact of the forcing scheme since they are based on the same equilibrium CMs. On the contrary, comparing  $M_2$  and  $M_3$  (or  $M_4$  and  $M_5$ ) gives more information about the influence of equilibrium CMs.

**TABLE I.** Synthetic view of the schemes adopted in our comparative study, which differ for the choice of the equilibrium distribution and forcing term treatment.

Model	Equilibrium distribution	Forcing treatment
$\overline{M_1}$	Discrete, 6th order	Full Hermite expansion
$M_2$	Continuous	He et al. <sup>74</sup>
$M_3$	Discrete, 2nd order	He et al. <sup>74</sup>
$M_4$	Discrete, 6th order	Guo et al. <sup>72</sup>
$M_5$	Discrete, 2nd order	Guo et al. <sup>72</sup>

### A. Two-dimensional four-rolls mill

Let us consider a square periodic box of size  $[x, y] \in [0:N]^2$ , with N being the number of points in each spatial direction. This test simulates the effect of four rotating cylinders that are replaced by a constant body force driving the flow. The latter reads as

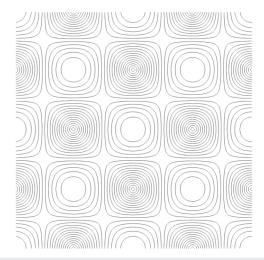
$$F(x) = 2vu_0\psi^2[\sin(\psi x)\sin(\psi y),\cos(\psi x)\cos(\psi y)]. \tag{30}$$

This test is known as four-rolls mill<sup>79</sup> and admits a steady state. When the latter is reached, the following velocity field is recovered:

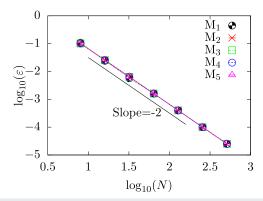
$$\mathbf{u}(\mathbf{x}) = u_0[\sin(\psi \mathbf{x})\sin(\psi \mathbf{y}),\cos(\psi \mathbf{x})\cos(\psi \mathbf{y})],\tag{31}$$

where  $u_0$  is the mean flow velocity and  $\psi = 2\pi/N$ . A contour plot of the velocity magnitude is depicted in Fig. 1, exhibiting the pattern that is typical of this kind of flow.

In this study, a Reynolds number Re =  $u_0N/v$  of 100 is considered. To assert the accuracy of the present numerical scheme, a convergence analysis is carried out by varying N as N = [8, 16, 32, 64, 128, 256, 512]. Let us collect the velocity field from Eq. (31) and the one from our numerical experiments in the vectors  $\mathbf{q}_{an}$  and  $\mathbf{q}_{num}$ , respectively. By doing so, it is then possible to compute the  $L_2$ -norm of the relative error  $\varepsilon$  between the present findings and



**FIG. 1.** Four-rolls mill: contour lines of the velocity magnitude. The plot is obtained by generating ten equally spaced intervals ranging from 0 to  $u_0$ .



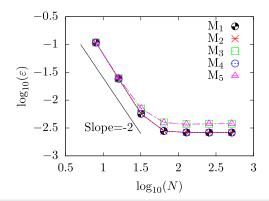
**FIG. 2.** Four-rolls mill at  $u_0 = 0.001$ : results of a convergence analysis by adopting different lattice Boltzmann models. The black continuous line corresponds to the expected second-order accuracy.

the analytical solution as

$$\varepsilon = \frac{\|\mathbf{q}_{an} - \mathbf{q}_{num}\|}{\|\mathbf{q}_{an}\|}.$$
 (32)

Note that we consider the numerical solution when the steady state is reached. This is achieved when the relative discrepancy in the velocity magnitude between two successive time steps is lower than a certain threshold ( $10^{-12}$  in this case). As shown in Fig. 2, the expected second-order accuracy property of the present method ( $M_1$ ) is successfully recovered, as an excellent convergence rate equal to 1.998 is found. Moreover, all the considered models show the same accuracy. This result can simply be explained by the fact that discrepancies between these models are proportional to powers of the velocity.

By increasing the mean velocity from  $u_0=0.001$  to 0.1, these discrepancies can be exerted, as illustrated in Fig. 3. Generally speaking, models with the highest equilibrium (M<sub>1</sub>, M<sub>2</sub>, and M<sub>4</sub>) show the best accuracy. An optimal convergence rate ~2 is achieved until N<32 and then the curves reach a plateau. Interestingly, M<sub>3</sub> and M<sub>5</sub> lose the second-order-accurate property of the LBE for a coarser



**FIG. 3**. Four-rolls mill at  $u_0 = 0.1$ : results of a convergence analysis by adopting different lattice Boltzmann models. The black continuous line corresponds to the expected second-order accuracy. The detrimental effect played by a poor equilibrium representation affecting  $M_3$  and  $M_5$  is clearly visible.

**TABLE II.** Four-rolls mill at  $u_0 = 0.1$ : base-10 logarithms of the relative errors for all the models and grid resolutions.  $M_1$  and  $M_2$  show the best performance.

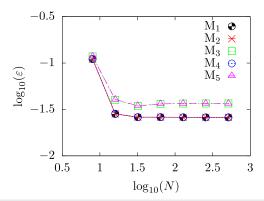
N	$M_1$ $M_2$		$M_3$	$M_4$	$M_5$	
8	-0.9655	-0.9655	-0.9637	-0.9654	-0.9636	
16	-1.6153	-1.6153	-1.6000	-1.6152	-1.5999	
32	-2.2445	-2.2445	-2.1513	-2.2442	-2.1510	
64	-2.5534	-2.5534	-2.4010	-2.5531	-2.4007	
128	-2.5797	-2.5797	-2.4319	-2.5795	-2.4317	
256	-2.5818	-2.5818	-2.4244	-2.5817	-2.4243	
512	-2.5820	-2.5820	-2.4230	-2.5820	-2.4230	

grid resolution, i.e., N < 16, showing lower accuracy for any larger value of N. We address this behavior to the poorer description of the equilibrium distribution. In fact,  $M_3$  and  $M_5$  are derived by using the classical expression of  $f^{eq}$  with Hermite polynomials up to the second order (see Table I). These findings are confirmed in Table II, where  $\log_{10}(\varepsilon)$  is reported for all the models and grid resolutions. Further insights are given in Fig. 4, where results from a convergence analysis at  $u_0 = 0.3$  are depicted. Again, the highest accuracy is obtained by adopting  $M_1$ ,  $M_2$ , and  $M_4$ . However, we note that the overall accuracy tends to deteriorate as the plateau is reached for lower resolutions with all considered models. This can be attributed to velocity-dependent error terms in the viscous stress tensor, which are drastically reduced by adopting the correct equilibrium state.<sup>37</sup> This is further highlighted in Fig. 5, where results are reported for  $M_1$  with  $u_0 = 0.001$ , 0.1, and 0.3.

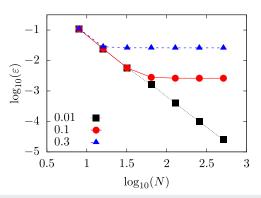
# B. Two-dimensional Hartmann flow

Let us now consider the flow of an electrically conductive fluid of magnetic resistivity  $\eta = v$ , which is subjected to a magnetic field  $\boldsymbol{b}$ . The interest reader shall refer to Appendix C for further details about the computation of the magnetic field  $\boldsymbol{b}$  in the lattice Boltzmann framework. Let us assume a rectangular domain of length  $L_x$  and height  $L_y$ . Initial conditions consist of

$$b(x, t = 0) = [0, b_{v0}, 0].$$



**FIG. 4.** Four-rolls mill at  $u_0 = 0.3$ : results of a convergence analysis by adopting different lattice Boltzmann models. Again, models based on second-order equilibrium ( $M_3$  and  $M_5$ ) exhibit the poorest accuracy.



**FIG. 5**. Four-rolls mill: results of a convergence are reported by adopting  $M_1$  at  $u_0$  = 0.001 (black squares), 0.1 (red circles), and 0.3 (blue triangles). As  $u_0$  increases, the convergence and accuracy of the solution deteriorates due to the remaining Mach-number errors.

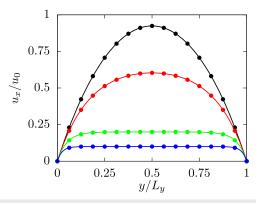
We reproduce the so-called Hartmann flow that it is analogous to the Poiseuille flow, but in the former case, (1) the fluid is assumed to be conductive and (2) a constant uniform vertical magnetic field  $(b_{y0})$  is enforced at the bottom and top walls, where the no-slip condition is enforced too. At the other sides, the magnetic field is assumed to be periodic. Magnetic boundary conditions are assigned according to Ref. 80. The analytical solution of the Hartmann flow reads as follows:

$$u_x(\mathbf{x},t) = \frac{4vu_0}{L_y b_{y0} \tanh(\mathrm{Ha})} \left[ 1 - \frac{\cosh(\mathrm{Ha}y'/L_y)}{\cosh(\mathrm{Ha})} \right], \quad (33)$$

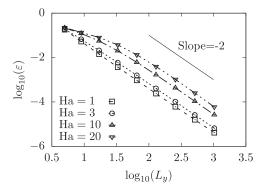
where  $y'=2y-L_y$  and the Hartmann number is defined as Ha =  $b_{y0}L_y/\sqrt{4\rho_0\nu\eta}$ . Moreover, Eq. (33) is prescribed at the inlet section, while an outflow boundary condition is applied at the opposite side of the domain as  $\boldsymbol{n}^{\dagger}\cdot\nabla\boldsymbol{u}=0$ , with  $\boldsymbol{n}^{\dagger}$  being the unit vector normal to the boundary. The presence of the magnetic field gives rise to the Lorentz force  $\boldsymbol{F}=\boldsymbol{j}\times\boldsymbol{b}$ , with  $\boldsymbol{j}$  being the electric current that is computed directly from the populations. 82

In Fig. 6, the results obtained with the present approach are compared to the analytical solution in Eq. (33) for four values of the Hartmann number, i.e., Ha = 1, 3, 10, and 20. From them, it is clear that  $M_1$  is able to correctly recover the physics of the Hartmann flow when the grid mesh is fine enough. The impact of the grid mesh as well as the convergence order of the present method is further studied by evaluating the relative error by Eq. (32) for different values of  $L_y \in [5:1025]$  and using the analytical solution in Eq. (33). In Fig. 7, the relative error is plotted against the number of points required to discretize the vertical dimension. Interestingly, a poor convergence is experienced for low values of  $L_y$ , especially for high values of Ha. This behavior can be explained by the presence of progressively thinner Hartmann layers, the latter requiring a larger number of grid points to be successfully and accurately captured.

In addition, this test was further considered to compare the behavior of the proposed approach with other forcing schemes. Since the Hartmann flow is a purely one-dimensional problem (i.e.,  $u_y = u_z = 0$ ), then only terms proportional to  $u_x$  remain in Eqs. (7) and (9). As a consequence, it is expected that  $M_1$  and  $M_4$  reduce to  $M_5$ , whereas  $M_2$  should recover the behavior of  $M_3$ . Hence, for any



**FIG. 6.** Hartmann flow: normalized axial velocity at Ha = 1 (black), 3 (red), 10 (green), and 20 (blue). Continuous lines and circles, respectively, account for the analytical predictions and the numerical solutions obtained with  $M_1$  using  $L_y$  = 1025.



**FIG. 7.** Hartmann flow simulation using  $M_1$ : convergence analysis for Ha = 1 (squares, dashed line), 3 (circles, dotted line), 10 (triangles, dotted-dashed line), and 20 (inverted triangles, dotted-dotted-dashed line). The continuous line denotes a convergence rate equal to 2.

combination of Ha and  $L_y$ , it has been confirmed that high-order approaches lead exactly to the same results, in terms of relative errors and velocity profiles, as their low-order versions.

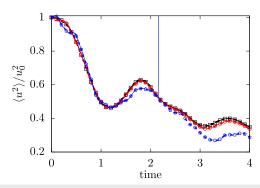
# C. Two-dimensional Orszag-Tang vortex

The next test case consists in the simulation of the Orszag-Tang vortex problem. <sup>81,83</sup> Its simulation represents a hard task since smaller and smaller structures appear in the domain as the Reynolds number grows. This test case is then a perfect candidate to assess (1) the accuracy of the present method to capture the existence of fine flow features as well as (2) its ability to deal with high local gradients that arise during the simulation. <sup>65</sup>

In what follows, a square periodic box of length  $2\pi$  is considered with a Reynolds number of Re = 1600, and the initial conditions are set to

$$u(x, t = 0) = u_0[-\sin(\psi y), \sin(\psi x)],$$
  

$$b(x, t = 0) = b_0[-\sin(\psi y), \sin(2\psi x)].$$
(34)



**FIG. 8.** Orszag-Tang vortex in two dimensions: time history of the kinetic energy normalized by its initial value for different Mach numbers, i.e., Ma = 0.07 (black), 0.14 (red), and 0.28 (blue). Continuous and dashed lines account for simulations carried out by using Hermite polynomials (used for both the equilibrium populations and forcing term) up to the second-  $(M_5)$  and sixth-  $(M_1)$  orders, respectively. The fluid domain consists of 256 points in each direction. The accuracy of these results is assessed by repeating the experiments with a finer grid (512 $^2$  lattice points) and by depicting the corresponding solution with symbols (squares for Ma = 0.07, circles for Ma = 0.14, and pentagons for Ma = 0.28). From them, it is clear that a mesh convergence has been achieved.

In Fig. 8, the mean kinetic energy  $\langle u^2 \rangle$ , normalized by its initial value  $\langle u_0^2 \rangle$  with  $\langle \cdot \rangle$  standing for the average over the whole domain, is reported for different values of the Mach number, i.e.,  $\text{Ma} = u_0/c_s = [0.07, 0.14, 0.28]$  with  $c_s = 1/\sqrt{3}$ . Moreover, it is set  $b_0 = u_0$ . Experiments are carried out by adopting Hermite polynomials up to the second-order and sixth-order for both the equilibrium populations and forcing term, corresponding to the models  $\text{M}_1$  and  $\text{M}_5$ , respectively. Here, the stability properties of the proposed algorithm clearly manifests. In fact, the two strategies are both stable for Ma = 0.07 and 0.14. By further increasing the Mach number,  $\text{M}_5$  becomes unstable, whereas the present approach,  $\text{M}_1$ , leads to the correct behavior.

We also use this scenario to evaluate the accuracy of different LB approaches. In Table III, the peak values of the current  $j_{\rm max}$  and vorticity  $\zeta_{\rm max}$  are compared to reference high-resolution spectral values in Ref. 81. All the models show very good accuracy, with the relative errors lower than 0.22%. To further evaluate their robustness, the Mach number is increased up to Ma = 0.28 for all the involved forcing schemes. From this, it is found that only  $M_5$  manifests stability issues.

**TABLE III.** Orszag-Tang vortex: findings from different LB schemes and reference spectral values from Ref. 81 in terms of peak value of the current,  $j_{\text{max}}$ , and vorticity,  $\zeta_{\text{max}}$ , at two representative time instants. Reproduced with permission from P. J. Dellar, "Lattice kinetic schemes for magnetohydrodynamics," J. Comput. Phys. **179**, 95–126 (2002). Copyright 2002 Elsevier.

	t	81	$M_1$	$M_2$	$M_3$	$M_4$	$M_5$
$j_{ m max}$	0.5 1	18.24 46.59	18.26 46.66	18.26 46.66	18.26 46.66	18.26 46.66	18.26 46.66
					6.756 14.11		

Now that the good numerical properties of our approach have been rigorously demonstrated in the 2D case, its validation will further be carried out considering 3D flows. Due to the non-negligible CPU time required to run these simulations,  $M_1$  will only be compared to the second-order formulations  $M_4$  and  $M_5$ , with the exception of the Rayleigh-Taylor instability test case, where data are already available in the literature.

# D. Three-dimensional Orszag-Tang vortex

This test involves an Orszag-Tang vortex that develops in a three-dimensional cubic periodic box of length  $2\pi$  with the initial conditions defined according to Ref. 84, i.e.,

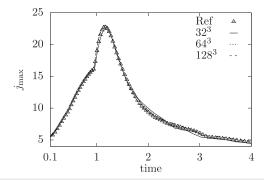
$$u(x, t = 0) = u_0[2 \sin(y), 2 \sin(x), 0],$$
  

$$b(x, t = 0) = 0.8b_0[-2 \sin(2y) + \sin(z),$$
  

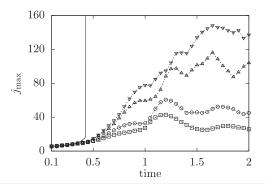
$$\times 2 \sin(x) + \sin(z), \sin(x) + \sin(y)],$$
 (35)

with  $u_0 = b_0 = 0.0203$ . Similar to the two-dimensional case, here the flow is characterized by the presence of intermittent small-scale structures. However, the topology of the reconnection of the magnetic field is considerably more complicated in three dimensions. First, the accuracy of the method is assessed by running a simulation at Re = 100 with three different grid sizes, i.e., N = [32, 64, 128]. In Fig. 9, the time evolution of the peak value of the current magnitude ( $j_{\text{max}}$ ) is depicted for the different grid dimensions. As N increases, the numerical solution gets progressively closer to the reference one obtained from a high-resolution pseudospectral simulation. This is further quantified by defining the vectors  $\mathbf{q}_{\text{ref}}$  and  $\mathbf{q}_{\text{num}}$ , gathering findings from the reference pseudospectral analysis and our results, respectively. Then, we introduce  $\varepsilon_{\infty}$  as the  $L_{\infty}$ -norm of the relative discrepancy, i.e.,

$$\varepsilon_{\infty} = \frac{\max |\mathbf{q}_{\text{ref}} - \mathbf{q}_{\text{num}}|}{\max |\mathbf{q}_{\text{ref}}|},$$



**FIG. 9.** Orszag-Tang vortex in three dimensions: the time evolutions of the peak value of the current magnitude,  $j_{\text{max}}$ , for different grid dimensions, i.e.,  $32^3$  (continuous line),  $64^3$  (dotted line), and  $128^3$  (dashed line), generated by the present method are compared to those from a high-resolution pseudospectral simulation from Ref. 60 (triangles, Ref). Reproduced with permission from A. De Rosis, "Nonorthogonal central-moments-based lattice Boltzmann scheme in three dimensions," Phys. Rev. E **95**, 013310 (2017). Copyright 2017 American Physical Society.



**FIG. 10.** Orszag-Tang vortex in three dimensions: time evolution of the peak value of the magnitude of the current for different Reynolds numbers. Symbols denote results from simulations by CMs at Re = 570 (squares), Re = 1040 (circles), Re = 3040 (triangles), and Re = 5600 (inverted triangles). The BGK run at Re = 570 (continuous line) shows an unstable behavior, while CM-runs always lead to a stable solution. Every run is conducted with N = 128.

which is of order  $10^{-2}$  for the finest grid, thus highlighting the accuracy of the method. Then, in order to evaluate the stability of the algorithm, higher values of the Reynolds number are considered, i.e., Re = [570, 1040, 3040, 5600]. In Fig. 10, the time evolution of  $j_{\text{max}}$  is reported for all these Reynolds numbers. From this, it is clear that the present approach (M<sub>1</sub>) remains stable even for the highest value of the Reynolds number. To further highlight the robustness of M<sub>1</sub>, the case Re = 570 (the one that is expected to be the least prone to the onset of instability) is also simulated using this time the BGK collision operator. 82 One can note that, for the latter collision model, the run blows up at  $t \sim 0.43$ . Conversely, the adoption of CMs (either the sixth-order or the second-order one, i.e., M<sub>1</sub>, M<sub>4</sub>, and M<sub>5</sub>) overcomes this problem, allowing us to simulate a larger time span for the whole desired set of Re. This is consistent with the observations in Ref. 65 regarding two-dimensional magnetohydrodynamic flows simulated by a CM-based D2Q9 model. Nonetheless, by increasing the Mach number, M5 becomes unstable as it was already the case in Sec. III C. Finally, a contour plot of the current magnitude is sketched in Fig. 11 at different time instants for the configuration Re = 570 with N = 128. One can appreciate that the present method is able to capture the existence of fine flow features as well as the presence of strong gradients.

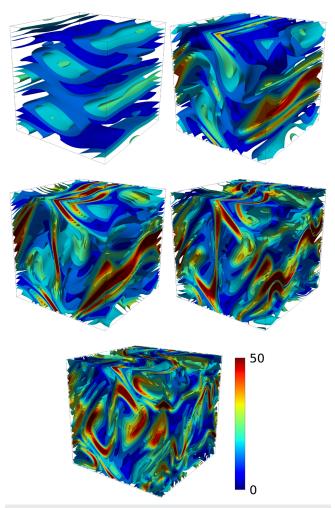
# E. Static droplet

In the following, the behavior of the present model is further investigated in the context of a multiphase flow simulated through the popular Shan-Chen model. Specifically, the latter introduces an interaction force

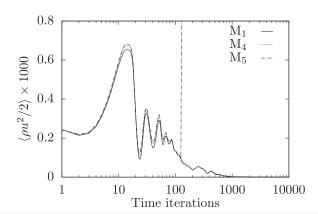
$$F(x,t) = -G\psi(x,t)\sum_{i}w_{i}c_{i}\psi(x+c_{i},t), \qquad (36)$$

which mimics the molecular interactions leading to phase segregation.  $\psi = 1 - e^{-\rho}$  is the so-called pseudopotential, and G is a parameter controlling the strength of the interaction.

Let us consider a periodic domain consisting of 200 lattice points in each direction. A droplet of radius R and density equal to  $\rho_0$  is placed in the center of the domain, while the density is set to  $\rho_0/8$ 



**FIG. 11.** Simulation of the Orszag-Tang vortex in three dimensions using  $M_1$ : time evolution of the (isocontours of the) current magnitude for Re = 570 and N = 128. From top left to bottom, t = 0.1, 0.5, 1.0, 1.5, and 2.0.

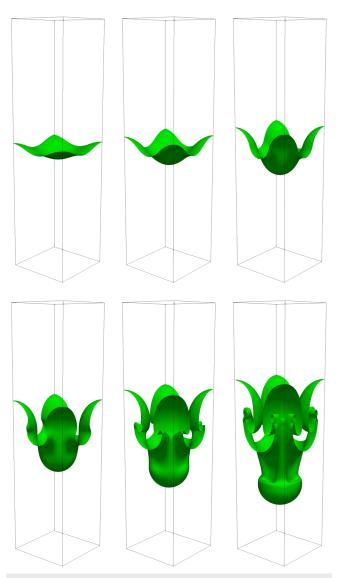


**FIG. 12.** Static droplet: time evolution of the mean kinetic energy by adopting Hermite polynomials up to the second- (solid line) and sixth- (dotted line) orders.

**TABLE IV.** Static droplet: normalized CPU time needed to perform the typical time step by different three-dimensional models.

$\overline{M_1}$	M <sub>4</sub>	M <sub>5</sub>
1	1.2	1.6

elsewhere. The parameter G is set equal to -6. The pseudopotential LBM is known to be prone to numerical instability due to the presence of spurious velocity currents. In Fig. 12, the time evolution of the mean kinetic energy is reported for numerical experiments carried out by using the models  $M_1$ ,  $M_4$ , and  $M_5$ . Here, the beneficial



**FIG. 13.** Rayleigh-Taylor instability: interface evolution at salient time instants, i.e.,  $t/t_0 = 0.5$  (top left), 1.0 (top center), 1.5 (top right), 2.0 (bottom left), 2.5 (bottom center), and 3.0 (bottom right).

effect of using the sixth-order Hermite polynomials for the forcing term is clearly visible. Indeed, the run corresponding to the poorest model (i.e.,  $M_5$ ) blows up after ~120 time iterations. Conversely, the adoption of the sixth-order Hermite polynomials allows us to simulate a considerably larger time span, with the kinetic energy that tends to vanish after initial large-amplitude oscillations. This behavior is slightly impacted by the form of the forcing term, as both  $M_1$  and  $M_4$  lead to stable simulations. A small mismatch between the two forcing schemes is found in the early stage of the simulation, where larger velocity magnitudes are encountered. They emphasize the impact of higher-order Hermite polynomials on the forcing term and suggest a slightly better behavior of  $M_1$ , as compared to  $M_4$ , since the former leads to a lower maximal kinetic energy. This mismatch is supposed to be even greater for droplets moving at a speed  $u \neq 0$ .

In order to compare the computational cost of different approaches, Table IV reports the CPU time needed to perform the typical time step normalized by the fastest one. Each C++ serial program runs on a desktop machine equipped by an Intel(R) Core(TM) i7-6700 CPU-3.40 GHz with 16 MB of L3 cache and 32 GB of RAM. The operating system is CANONICAL UBUNTU 16.10. Interestingly, the proposed approach exhibits an advantage with respect to the remaining three-dimensional methods that we explain as follows: The real bottleneck of CM-LBMs is represented by the reconstruction of postcollision populations by Eq. (20). In this regard, several choices are available. For example, one can be represented by a hand-made implementation of the solution of this linear system by a Cholesky decomposition (the matrix is always positive definite), followed by a back- or forward-substitution. A smarter choice could imply the adoption of hardware-optimized numerical libraries (e.g., PARDISO, 87 SuiteSparse, 88 and Spooles 89). However, here we prefer to operate a priori a symbolic solution of the system (see Matlab script in the supplementary material). By doing so, we can use the fact that many terms in the vector collecting the postcollision CMs  $k_i^*$  are zero, thus involving many zero terms in the resultant expressions of  $f_i^*$  that can be directly neglected in our C++ programs. The good behavior summarized in Table IV stems from the fact that the present model  $M_1$  is the one with the highest number of zero terms in  $k_i^*$ , implying the smallest number of floating-point operations to reconstruct  $f_i^*$ . The possibility of using a parallel implementation (by OpenMP or MPI) does not impact the ratios of mutual performance, which is affected only by the number of floating point operations required by each single lattice point. Consequently, even if the present approach does not show a drastic improvement in terms of accuracy, it leads to the shortest wall-clock time one can expect from CM-LBMs with external forcing.

# F. Rayleigh-Taylor instability

To further investigate the numerical properties of the present approach, the simulation of the Rayleigh-Taylor instability is carried out hereafter. The fundamental mechanism, upon which this kind of instability relies, can be summarized as follows: Let us consider a heavy fluid on top of a lighter one, both being immiscible fluids that are separated by an interface. Assuming that they are subject to a gravitational field, their interface will not be able to stay in its equilibrium state. As a consequence, plumes (spikes) will start forming and flowing downward (upward).

This Rayleigh-Taylor instability mechanism is simulated hereafter by adopting the color-gradient method. 90-92 Recently, Saito et al. 66 have proposed a CM-based color-gradient model by using the third- and second-order terms for the equilibrium and forcing, respectively. Here, the approach is recasted within the present framework, hence adopting the sixth-order Hermite polynomials for both the equilibrium and the forcing term. This problem, as

**TABLE V.** Rayleigh-Taylor instability: time evolution of the position of the spike and the bubble of the interface at salient time instants. The present results are compared with those from (i) a MRT LB study, <sup>93</sup> (ii) a BGK LB model for multiphase flows, <sup>94</sup> (iii) a phase-field MRT LB scheme, <sup>95</sup> and (iv) a solution of the coupled Navier-Stokes-Cahn-Hilliard equations. <sup>96</sup> Reproduced with permission from S. Saito, Y. Abe, and K. Koyama, "Lattice Boltzmann modeling and simulation of liquid jet breakup," Phys. Rev. E **96**, 013317 (2017). Copyright 2017 American Physical Society. Reproduced with permission from He *et al.*, "On the three-dimensional Rayleigh-Taylor instability," Phys. Fluids **11**, 1143–1152 (1999). Copyright 1999 AIP Publishing LLC. Reproduced with permission from N. Wang, H. Liu, and C. Zhang, "Three-dimensional phase-field lattice Boltzmann model for incompressible multiphase flows," J. Comput. Sci. **17**, 340–356 (2016). Copyright 2016 Elsevier. Reproduced with permission from H. G. Lee and J. Kim, "Numerical simulation of the three-dimensional Rayleigh-taylor instability," Comput. Math. Appl. **66**, 1466–1474 (2013). Copyright 2013 Elsevier.

	Spike				Bubble					
$t/t_0$	Present	93	94	95	96	Present	93	94	95	96
0.0	1.897	1.895	1.887	1.888	1.904	2.100	2.095	2.092	2.096	2.101
0.5	1.845	1.864	1.839	1.860	1.869	2.101	2.131	2.113	2.129	2.131
1.0	1.753	1.763	1.744	1.755	1.776	2.202	2.230	2.229	2.228	2.224
1.5	1.591	1.587	1.555	1.569	1.618	2.345	2.377	2.372	2.364	2.372
2.0	1.378	1.357	1.312	1.325	1.396	2.516	2.535	2.545	2.524	2.535
2.5	1.121	1.085	1.022	1.037	1.149	2.682	2.695	2.693	2.672	2.688
3.0	0.791	0.788	0.712	0.740	0.863	2.834	2.847	2.846	2.824	2.856
3.5	0.537	0.481	0.390	0.419	0.572	2.997	2.999	3.009	2.984	3.009
4.0	0.233	0.160	0.060	0.090	0.271	3.184	3.179	3.178	3.164	3.181

well as its solution procedure by the color-gradient method, further shows the universality of the present approach for the simulation of multiphysics flows. The CM-based color-gradient method is briefly outlined in Appendix D.

Let us consider a three-dimensional domain of size  $W \times 4W \times W$ , with W = 64. A fluid of density  $\rho_h$  is placed over a lighter one of density  $\rho_l = 1$ . The initial flow field is perturbed as

$$\rho(\mathbf{x},0) = \rho_h, \text{ if } y > 2W + 0.05W[\cos(2\pi x) + \cos(2\pi z)],$$

$$\rho(\mathbf{x},0) = \rho_l, \text{ otherwise.}$$
(37)

No-slip walls are enforced at the top and bottom sections, while periodic boundaries are assumed at the other sides of the domain. A gravitational body force is considered as

$$\mathbf{F} = -\left[\rho(\mathbf{x}, t) - \frac{\rho_h + \rho_l}{2}\right] \mathbf{g},\tag{38}$$

with g = (0, -g, 0), and g chosen so that  $\sqrt{gW} = 0.04$ . The problem is governed by two dimensionless parameters: the Reynolds number,

$$Re = \frac{W\sqrt{gW}}{v},\tag{39}$$

and the Atwood number,

$$At = \frac{\rho_h - \rho_l}{\rho_h + \rho_l},\tag{40}$$

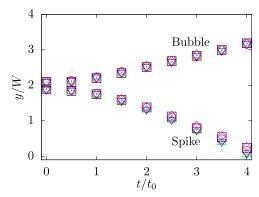


FIG. 14. Rayleigh-Taylor instability: time evolution of the position of the spike and the bubble of the interface at salient time instants. The present results (black squares) are plotted together with those from (i) a MRT LB study<sup>93</sup> (red circles), (ii) a BGK LB model for multiphase flows 94 (green triangles), (iii) a phase-field MRT LB scheme<sup>95</sup> (blue inverted triangles), and (iv) a solution of the coupled Navier-Stokes-Cahn-Hilliard equations <sup>96</sup> (magenta diamonds). Reproduced with permission from S. Saito, Y. Abe, and K. Koyama, "Lattice Boltzmann modeling and simulation of liquid jet breakup," Phys. Rev. E 96, 013317 (2017). Copyright 2017 American Physical Society. Reproduced with permission from He et al., "On the three-dimensional Rayleigh-Taylor instability," Phys. Fluids 11, 1143–1152 (1999). Copyright 1999 AIP Publishing LLC. Reproduced with permission from N. Wang, H. Liu, and C. Zhang, "Three-dimensional phase-field lattice Boltzmann model for incompressible multiphase flows," J. Comput. Sci. 17, 340-356 (2016). Copyright 2016 Elsevier. Reproduced with permission from H. G. Lee and J. Kim, "Numerical simulation of the three-dimensional Rayleigh-Taylor instability," Comput. Math. Appl. 66, 1466-1474 (2013). Copyright 2013 Elsevier.

which are set here to Re = 512 and At = 0.5, respectively. In addition, the characteristic time used for the time evolution of the Rayleigh-Taylor instability is defined as

$$t_0 = \frac{t}{\sqrt{Wg}}. (41)$$

In the following, let us denote the bottom and top points of the interface as spike and bubble, respectively. In Fig. 13, the evolution of the interface between the two fluids is sketched at salient time instants. They show that the edge of the spike rolls up at  $t/t_0 = 3$ . This is consistent with another LB study, <sup>93</sup> where the Rayleigh-Taylor instability has been investigated by means of the multiple-relaxation-time kernel.

More quantitative results, regarding the time evolution of the position of the two reference points, are reported in Table V and depicted in Fig. 14. The present findings are further compared to several models to assess its accuracy: (1) a MRT LB study, <sup>93</sup> (2) a BGK LB model for multiphase flows, <sup>94</sup> (3) a phase-field MRT LB scheme, <sup>95</sup> and (4) a solution of the coupled Navier-Stokes-Cahn-Hilliard equations. <sup>96</sup> From this, it is clear that the present method shows a pretty good agreement with data from the literature. This confirms the good numerical properties of the proposed approach as well as its universality.

### IV. CONCLUSIONS

In this work, a *systematic* way to derive CM-LBMs with external forcing has been proposed for the simulation of multiphysics phenomena. It naturally flows from a previous work based on the D3Q27 velocity discretization, <sup>69</sup> where Galilean invariant CMs were derived from the discrete equilibrium state instead of the continuous Maxwell Boltzmann distribution. Here, by relying on the very same set of 27 Hermite polynomials, Galilean invariant CMs of the forcing term are also obtained but without any assumption on these CMs. By further equilibrating acoustic and high-order CMs, the present method only requires the computation of five precollision CMs. Both points eventually lead to a very concise algorithm.

Interestingly, it is also shown that the present method recovers the behavior of the cascaded LBM. However, different from the latter, the proposed methodology allows the derivation of forcing terms in an *a priori* way, thanks to the tight link existing between the Hermite polynomial expansion, the family of moments considered, and the lattice of discrete velocities. As a consequence, its extension to any kind of collision models and velocity discretizations can be done in a *systematic* and *straightforward* manner by following the instructions proposed in Ref. 37. This is illustrated in Appendix E in the context of the D3Q27-LBM based on RMs.

The simulation of the four-rolls mill, the Hartmann flow, the two- and three-dimensional Orszag-Tang vortex, a static droplet, and the Rayleigh-Taylor instability was conducted to evaluate both the numerical behavior and the universality of the present method. By further comparing the latter with four other forcing schemes on several of these test cases, its good accuracy and robustness properties were properly demonstrated.

All in all, the consistency with the cascaded LBM and the compactness of the algorithm as well as its excellent numerical properties lead the present scheme to be a very good candidate to

perform multiphysics simulations with (or without) the presence of external forces. Regarding future works, the extension to both high-order lattices (that require high-order terms in the Hermite expansion) and other kinds of collision space (raw, Hermite, central Hermite moment, and cumulant) as well as regularized approaches<sup>37</sup> is currently under investigation. The corresponding results shall be presented in the near future.

# SUPPLEMENTARY MATERIAL

See supplementary material for a script that allows the reader to implement the numerical model described in this manuscript and to perform all the involved symbolic manipulations.

# **ACKNOWLEDGMENTS**

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# APPENDIX A: HERMITE POLYNOMIALS

Discrete Hermite polynomials are defined as 67,68

$$\mathcal{H}_{i}^{(n)} = \frac{\left(-c_{s}^{2}\right)^{n}}{\omega(c_{i})} \nabla_{c_{i}}^{n} \omega(c_{i}), \tag{A1}$$

with the corresponding Gaussian weight function

$$w(c_i) = \frac{1}{(2\pi c_s^2)^{D/2}} \exp\left(-\frac{c_i^2}{2c_s^2}\right),$$
 (A2)

where D is the number of physical dimensions and  $c_s$  is the lattice constant. In the 3D case, discrete Hermite tensors can be computed using tensor products of their 1D formulation,

$$\mathcal{H}_{ix...xy...yz...z}^{(nx+ny+nz)} = \frac{(-c_s^2)^{nx+ny+nz}}{w(c_i)} \nabla_{c_i}^{nx+ny+nz} w(c_i)$$

$$= \left[ \frac{(-c_s^2)^{nx}}{w'(c_{ix})} \nabla_{c_{ix}}^{nx} w'(c_{ix}) \right] \left[ \frac{(-c_s^2)^{ny}}{w'(c_{iy})} \nabla_{c_{iy}}^{ny} w'(c_{iy}) \right]$$

$$\times \left[ \frac{(-c_s^2)^{nz}}{w'(c_{iz})} \nabla_{c_{iz}}^{nz} w'(c_{iz}) \right]$$

$$= \mathcal{H}_{ix...x}^{(nx)} \mathcal{H}_{iy...y}^{(ny)} \mathcal{H}_{iz...z}^{(nz)}, \tag{A3}$$

where w' is the one-dimensional version of the Gaussian weight [see Eq. (A2)] and (nx, ny, nz) are the number of occurrences of indices (x, y, z). Since the D3Q27 lattice is a tensor product of three D1Q3 lattices, then only 1D Hermite polynomials of degree up to n = 2 can be used for the construction of their 3D counterparts. This means that the degree of each Hermite polynomial will be at most *two per direction*. Indeed,

$$\mathcal{H}_{i}^{(0)} = 1, \quad \mathcal{H}_{ix}^{(1)} = c_{ix}, \quad \mathcal{H}_{iy}^{(1)} = c_{iy}, \quad \mathcal{H}_{iz}^{(1)} = c_{iz}, \quad \mathcal{H}_{ixx}^{(2)} = c_{ix}^{2} - c_{s}^{2}, \quad \mathcal{H}_{iyy}^{(2)} = c_{iy}^{2} - c_{s}^{2}, \quad \mathcal{H}_{izz}^{(2)} = c_{iz}^{2} - c_{s}^{2}, \quad \mathcal{H}_{ixy}^{(2)} = \mathcal{H}_{ix}^{(1)} \mathcal{H}_{iy}^{(1)},$$

$$\mathcal{H}_{ixz}^{(2)} = \mathcal{H}_{ix}^{(1)} \mathcal{H}_{iz}^{(1)}, \quad \mathcal{H}_{iyz}^{(2)} = \mathcal{H}_{iy}^{(1)} \mathcal{H}_{iz}^{(1)}, \quad \mathcal{H}_{ixxy}^{(3)} = \mathcal{H}_{ixx}^{(2)} \mathcal{H}_{iy}^{(1)}, \quad \mathcal{H}_{iyyz}^{(3)} = \mathcal{H}_{iyy}^{(2)} \mathcal{H}_{iz}^{(1)}, \quad \mathcal{H}_{ixyy}^{(3)} = \mathcal{H}_{ix}^{(1)} \mathcal{H}_{iyy}^{(2)},$$

$$\mathcal{H}_{ixzz}^{(3)} = \mathcal{H}_{ix}^{(1)} \mathcal{H}_{izz}^{(2)}, \quad \mathcal{H}_{iyzz}^{(3)} = \mathcal{H}_{iy}^{(1)} \mathcal{H}_{izz}^{(2)}, \quad \mathcal{H}_{ixyz}^{(3)} = \mathcal{H}_{ix}^{(1)} \mathcal{H}_{iy}^{(1)}, \quad \mathcal{H}_{iz}^{(4)} = \mathcal{H}_{ixx}^{(2)} \mathcal{H}_{iyy}^{(2)}, \quad \mathcal{H}_{ixxyz}^{(4)} = \mathcal{H}_{ixx}^{(2)} \mathcal{H}_{iyz}^{(2)}, \quad \mathcal{H}_{ixyzz}^{(4)} = \mathcal{H}_{ixx}^{(2)} \mathcal{H}_{izz}^{(2)}, \quad \mathcal{H}_{ixyzz}^{(4)} = \mathcal{H}_{ixx}^{(2)} \mathcal{H}_{iyz}^{(2)}, \quad \mathcal{H}_{ixyzz}^{(4)} = \mathcal{H}_{ixx}^{(4)} \mathcal{H}_{iyy}^{(4)} \mathcal{H}_{izz}^{(2)}, \quad \mathcal{H}_{ixxyzz}^{(5)} = \mathcal{H}_{ixx}^{(5)} \mathcal{H}_{iyy}^{(5)} \mathcal{H}_{izz}^{(5)}, \quad \mathcal{H}_{ixxyzzz}^{(5)} = \mathcal{H}_{ixx}^{(5)} \mathcal{H}_{iyy}^{(5)} \mathcal{H}_{izz}^{(5)}, \quad \mathcal{H}_{ixxyzzz}^{(5)} \mathcal{H}_{ixxyzzzz}^{(5)} \mathcal{H}_{ixxyzzzz}^{(5)} \mathcal{H}_{ixxyzzzz}^{(5)} \mathcal{H}_{ixxyzzzz}^{(5)} \mathcal{H}_{ixxyzzzz}^{(5)} \mathcal{H}_{ixxyzzzz}^{(5)} \mathcal{H}_{ixxyzzzz}^{(5)} \mathcal{H}_{ixxyzzzzz}^{(5)} \mathcal{H}_{ixxyzzzzzzzz}$$

# APPENDIX B: TWO-DIMENSIONAL MODEL

Let us consider an Eulerian basis  $\mathbf{x} = [x, y]$ . Here, we derive the two-dimensional LB model by means of the D2Q9 velocity discretization, where lattice directions  $\mathbf{c}_i = [|c_{ix}\rangle, |c_{iy}\rangle]$  are

$$|c_{ix}\rangle = \begin{bmatrix} 0, & 1, -1, & 0, & 0, & 1, -1, & 1, -1 \end{bmatrix}^{\mathsf{T}},$$
  
 $|c_{iy}\rangle = \begin{bmatrix} 0, & 0, & 0, & 1, -1, & 1, -1, -1, & 1 \end{bmatrix}^{\mathsf{T}},$  (B1)

with  $i=0,\ldots,8$ . Let us define the particle distribution functions  $|f_i\rangle=[f_0,\ldots,f_i,\ldots,f_8]^{\mathsf{T}}$  and the velocity vector  $\boldsymbol{u}=[u_x,u_y]$ . In two dimensions, the equilibrium distributions  $|f_i^{eq}\rangle=[f_0^{eq},\ldots,f_i^{eq},\ldots,f_8^{eq}]^{\mathsf{T}}$  can be expanded onto a basis of Hermite polynomials  $\mathcal{H}^{(n)}$  up to the fourth-order, 35,67,68 i.e.,

$$f_{i}^{eq} = w_{i} \rho \left( 1 + \frac{c_{i} \cdot \mathbf{u}}{c_{s}^{2}} + \frac{1}{2c_{s}^{4}} \mathcal{H}_{i}^{(2)} : \mathbf{u}\mathbf{u} + \frac{1}{2c_{s}^{6}} (\mathcal{H}_{ixxy}^{(3)} u_{x}^{2} u_{y} + \mathcal{H}_{ixxyy}^{(3)} u_{x} u_{y}^{2}) + \frac{1}{4c_{s}^{8}} \mathcal{H}_{ixxyy}^{(4)} u_{x}^{2} u_{y}^{2} \right),$$
(B2)

where ":" stands for the full contraction of indices (also known as the Frobenius inner product),  $w_0 = 4/9$ ,  $w_{1...4} = 1/9$ ,  $w_{5...8} = 1/36$ , and  $c_s = 1/\sqrt{3}$  is the lattice sound speed.<sup>4</sup> Again, the development of a CM-based collision operator begins by shifting the lattice directions by the local fluid velocity, i.e.,  $\bar{c}_i = [|\bar{c}_{ix}\rangle, |\bar{c}_{iy}\rangle]$ , where

$$|\overline{c}_{ix}\rangle = |c_{ix} - u_x\rangle, \quad |\overline{c}_{iy}\rangle = |c_{iy} - u_y\rangle.$$
 (B3)

Let us choose the basis  $\overline{T} = [\overline{T}_0, \ldots, \overline{T}_i, \ldots, \overline{T}_8]$  whose components

$$|\overline{T}_{0}\rangle = |1, \dots, 1\rangle, \quad |\overline{T}_{1}\rangle = |\overline{c}_{ix}\rangle, \quad |\overline{T}_{2}\rangle = |\overline{c}_{iy}\rangle,$$

$$|\overline{T}_{3}\rangle = |\overline{c}_{ix}^{2} + \overline{c}_{iy}^{2}\rangle, \quad |\overline{T}_{4}\rangle = |\overline{c}_{ix}^{2} - \overline{c}_{iy}^{2}\rangle, \quad |\overline{T}_{5}\rangle = |\overline{c}_{ix}\overline{c}_{iy}\rangle, \quad |\overline{T}_{6}\rangle = |\overline{c}_{ix}^{2}\overline{c}_{iy}\rangle, \quad |\overline{T}_{7}\rangle = |\overline{c}_{ix}c_{ii}^{2}\rangle, \quad |\overline{T}_{8}\rangle = |\overline{c}_{ix}^{2}\overline{c}_{iy}^{2}\rangle.$$
(B4)

Let us collect precollision, equilibrium, and postcollision CMs as

$$|k_i\rangle = \begin{bmatrix} k_0, \dots, k_i, \dots, k_8 \end{bmatrix}^{\mathsf{T}},$$

$$|k_i^{eq}\rangle = \begin{bmatrix} k_0^{eq}, \dots, k_i^{eq}, \dots, k_8^{eq} \end{bmatrix}^{\mathsf{T}},$$

$$|k_i^{\star}\rangle = \begin{bmatrix} k_0^{\star}, \dots, k_i^{\star}, \dots, k_8^{\star} \end{bmatrix}^{\mathsf{T}},$$
(B5)

respectively. The first two quantities are evaluated by applying the transformation matrix  $\overline{\mathcal{T}}$  to the corresponding distribution, which

$$|k_i\rangle = \overline{\mathcal{T}}^{\mathsf{T}}|f_i\rangle, \quad |k_i^{eq}\rangle = \overline{\mathcal{T}}^{\mathsf{T}}|f_i^{eq}\rangle.$$
 (B6)

By adopting equilibrium populations with the full set of Hermite polynomials [see Eq. (B2)], equilibrium CMs are

$$k_0^{eq} = \rho, \quad k_3^{eq} = 2\rho c_s^2, \quad k_8^{eq} = \rho c_s^4$$
 (B7)

and  $k_{1,2,4,5,6,7}^{eq} = 0$ . In analogy with the 3D case, it should be noted that (1) the equilibrium state is fully Galilean invariant since it does not show any dependence on the fluid velocity and (2) the discrete equilibrium CMs have the same form of the continuous counterpart when the full set of Hermite polynomials is considered.

Again, the postcollision CMs can be written as

$$|k_i^{\star}\rangle = (\mathbf{I} - \mathbf{\Lambda})\overline{\mathcal{T}}^{\mathsf{T}}|f_i\rangle + \mathbf{\Lambda}\overline{\mathcal{T}}^{\mathsf{T}}|f_i^{eq}\rangle + (\mathbf{I} - \frac{\mathbf{\Lambda}}{2})\overline{\mathcal{T}}^{\mathsf{T}}|\mathcal{F}_i\rangle$$
 (B8)

or

$$|k_i^{\star}\rangle = (\mathbf{I} - \mathbf{\Lambda})|k_i\rangle + \mathbf{\Lambda}|k_i^{eq}\rangle + (\mathbf{I} - \frac{\mathbf{\Lambda}}{2})|R_i\rangle,$$
 (B9)

where **I** is the  $(9 \times 9)$  unit tensor and  $\Lambda = \text{diag}[1, 1, 1, 1, \omega, \omega, 1, 1, 1]$ is a  $(9 \times 9)$  relaxation matrix.

Now, let us define the two-dimensional forcing term  $\mathcal{F}_i$ . Specifically, we employ the expression adopted by Huang et al.,

$$\mathcal{F}_{i} = w_{i} \left( \frac{F \cdot c_{i}}{c_{s}} \cdot \mathcal{H}_{i}^{(1)} + \frac{[Fu]}{2c_{s}^{2}} \cdot \mathcal{H}_{i}^{(2)} + \frac{[Fuu]}{6c_{s}^{3}} \cdot \mathcal{H}_{i,[xyy],[xxy]}^{(3)} + \frac{[Fuuu]}{24c_{s}^{4}} \cdot \mathcal{H}_{i,[xxyy]}^{(4)} \right), \tag{B10}$$

where the square bracket in the Hermite coefficient denotes permutations. Note that the popular formula by Guo et al. 72 is recovered if  $\mathcal{H}_i^{(3)}$  and  $\mathcal{H}_i^{(4)}$  are neglected. The CMs of the forcing term can be computed as

$$|R_i\rangle = \overline{\mathcal{T}}^{\mathsf{T}}|\mathcal{F}_i\rangle$$
 (B11)

and read as follows:

$$R_1 = F_x$$
,  $R_2 = F_y$ ,  $R_6 = F_y c_s^2$ ,  $R_7 = F_x c_s^2$ , (B12)

and  $R_{0,3,4,5,8} = 0$ . The resultant expressions of the postcollision

$$k_0^* = \rho, \quad k_1^* = F_x/2, \quad k_2^* = F_y/2, \quad k_3^* = 2\rho c_s^2, \quad k_4^* = (1 - \omega)k_4,$$
  
 $k_5^* = (1 - \omega)k_5, \quad k_6^* = F_y c_s^2/2, \quad k_7^* = F_x c_s^2/2, \quad k_8^* = \rho c_s^4.$ 
(B13)

Again, one can appreciate that the postcollision state in terms of CMs shows simple expressions. Then, the postcollision populations are reconstructed as usual,

$$|f_i^{\star}\rangle = \left(\overline{\mathcal{T}}^{\mathsf{T}}\right)^{-1}|k_i^{\star}\rangle,$$
 (B14)

with  $|f_i^*\rangle = [f_0^*, \dots, f_i^*, \dots, f_8^*]^T$  that are eventually streamed.

# APPENDIX C: LATTICE BOLTZMANN METHOD FOR THE EVOLUTION OF THE MAGNETIC FIELD

The evolution of the magnetic field b is predicted by a second LBM based on the D3Q7 velocity space, which is  $\forall l \in [0, 6]$ ,

$$\mathbf{h}_{l}(\mathbf{x} + \boldsymbol{\xi}_{l}, t + 1) = \mathbf{h}_{l}(\mathbf{x}, t) + \omega_{m} \left[ \mathbf{h}_{l}^{eq}(\mathbf{x}, t) - \mathbf{h}_{l}(\mathbf{x}, t) \right], \tag{C1}$$

with  $\omega_m$  being the relaxation frequency that is related to the magnetic resistivity as  $\eta = c_s^2 (1/\omega_m - 1/2)$ , where  $c_s = 1/2$  for the D3Q7 lattice. For this particular lattice, discrete velocities  $\xi_l = [|\xi_{lx}\rangle, |\xi_{ly}\rangle$ ,  $|\xi_{lz}\rangle$ ] are defined as

$$\begin{aligned} |\xi_{lx}\rangle &= \begin{bmatrix} 0, & 1, -1, & 0, & 0, & 0, & 0 \end{bmatrix}^{\mathsf{T}}, \\ |\xi_{ly}\rangle &= \begin{bmatrix} 0, & 0, & 0, & 1, -1, & 0, & 0 \end{bmatrix}^{\mathsf{T}}, \\ |\xi_{lz}\rangle &= \begin{bmatrix} 0, & 0, & 0, & 0, & 0, & 1, -1 \end{bmatrix}^{\mathsf{T}}. \end{aligned}$$
(C2)

Vector-valued populations  $h_l$  are necessary due to the antisymmetry of the electric tensor.<sup>81</sup> Their equilibrium state is defined as

$$h_{l\alpha}^{eq} = w_l \left[ b_{\alpha} + 4\xi_{l\beta} \left( u_{\beta} b_{\alpha} - b_{\beta} u_{\alpha} \right) \right], \tag{C3}$$

where  $w_0 = 1/4$ ,  $w_{1...6} = 1/8$ , and  $(\alpha, \beta) \in \{x, y, z\}^2$ . Regarding magnetic boundary conditions, we follow the approach in Ref. 80 that is built on the fact that the magnetic field can be computed as

$$\boldsymbol{b} = \sum_{l=0}^{6} \boldsymbol{h}_{l}. \tag{C4}$$

Specifically, let us assume that one wants to enforce a certain magnetic field  $b^{\dagger} = [b_x^{\dagger}, b_y^{\dagger}, b_z^{\dagger}]$  to the bottom section of the considered domain. In this case, the only population to be assigned is  $h_2$  that is evaluated as

$$h_{2x} = b_x^{\dagger} - \sum_{l \neq 2} h_{lx}, \quad h_{2y} = b_y^{\dagger} - \sum_{l \neq 2} h_{ly}, \quad h_{2z} = b_z^{\dagger} - \sum_{l \neq 2} h_{lz}.$$
 (C5)

# APPENDIX D: CM-BASED COLOR-GRADIENT METHOD

Let us consider two immiscible fluids, namely, red and blue. The evolution of populations  $f_i^k$  is

$$f_i^k(\mathbf{x} + \mathbf{c}_i, t + 1) = f_i^k(\mathbf{x}, t) + \Omega_i^k(\mathbf{x}, t),$$
 (D1)

where k = r for the red fluid and k = b for the blue one. Moreover, it is possible to define the total distribution functions as  $f_i = f_i^r + f_i^b$ . The collision process  $\Omega_i^k$  is composed of three substages,

$$\Omega_i^k = \left(\Omega_i^k\right)^{(3)} \left[ \left(\Omega_i^k\right)^{(1)} + \left(\Omega_i^k\right)^{(2)} \right], \tag{D2}$$

where  $\left(\Omega_i^k\right)^{(1)}$ ,  $\left(\Omega_i^k\right)^{(2)}$ , and  $\left(\Omega_i^k\right)^{(3)}$  are the single-phase, perturbation, and recoloring operators, respectively. Within the CM-based framework, <sup>66</sup> the single-phase collision operator can be written as

$$\left(\Omega_{i}^{k}\right)^{(1)} = \left(\overline{\mathcal{T}}^{\mathsf{T}}\right)^{-1} \Lambda \overline{\mathcal{T}}^{\mathsf{T}} \left(\left|f_{i}^{eq}\right\rangle - \left|f_{i}\right\rangle\right) + \left(\mathbf{I} - \frac{\Lambda}{2}\right) \overline{\mathcal{T}}^{\mathsf{T}} \left|\mathcal{F}_{i}\right\rangle, \quad (D3)$$

with  $|\mathcal{F}_i\rangle$ ,  $\overline{\mathcal{T}}^T$ , and  $\Lambda$  that obey Eqs. (9), (11), and (17), respectively. The macroscopic variables are given by

$$\rho_k = \sum_i f_i^k, \quad \rho = \sum_k \rho_k, \quad \rho \mathbf{u} = \sum_i f_i \mathbf{c}_i + \frac{1}{2} \mathbf{F}, \tag{D4}$$

where  $\rho_k$  is the density of the fluid k,  $\rho$  is the total mass density,  $\boldsymbol{u}$  is the total momentum, and  $\boldsymbol{F}$  is a body force. Distributions relax to an enhanced local equilibrium<sup>93,97</sup> that, by adopting the sixth-order Hermite polynomials, reads as follows:

$$f_{i}^{eq}(\rho, \mathbf{u}) = \rho \left\{ \varphi_{i} + w_{i} \left[ \frac{c_{i} \cdot \mathbf{u}}{c_{s}^{2}} + \frac{1}{2c_{s}^{4}} \left[ \mathcal{H}_{ixx}^{(2)} u_{x}^{2} + \mathcal{H}_{iyy}^{(2)} u_{y}^{2} + \mathcal{H}_{izz}^{(2)} u_{z}^{2} + 2 \left( \mathcal{H}_{ixy}^{(2)} u_{x} u_{y} + \mathcal{H}_{ixz}^{(2)} u_{x} u_{z} + \mathcal{H}_{iyz}^{(2)} u_{y} u_{z} \right) \right] \right.$$

$$\left. + \frac{1}{2c_{s}^{6}} \left[ \mathcal{H}_{ixxy}^{(3)} u_{x}^{2} u_{y} + \mathcal{H}_{ixxz}^{(3)} u_{x}^{2} u_{z} + \mathcal{H}_{ixyy}^{(3)} u_{x} u_{y}^{2} + \mathcal{H}_{ixzz}^{(3)} u_{x} u_{z}^{2} + \mathcal{H}_{iyyz}^{(3)} u_{y} u_{z}^{2} + \mathcal{H}_{iyyz}^{(3)} u_{y}^{2} u_{z}^{2} + 2 \mathcal{H}_{ixyyz}^{(3)} u_{x}^{2} u_{z}^{2} + \mathcal{H}_{ixyyz}^{(4)} u_{x}^{2} u_{y}^{2} u_{z}^{2} + \mathcal{H}_{ixyyz}^{(4)} u_{x}^{2} u_{y}^{2} u_{z}^{2} + \mathcal{H}_{ixyyz}^{(4)} u_{x}^{2} u_{y}^{2} u_{z}^{2} + \mathcal{H}_{ixxyz}^{(4)} u_{x}^{2} u_{y}^{2} u_{z}^{2} + \mathcal{H}_{ixxyz}^{(5)} u_{x}^{2} u_{y}^{2} u_{z}^{2} + \mathcal{H}_{ixxyyzz}^{(5)} u_{x}^{2} u_{y}^{2} u_{z}^{2} + \mathcal{H}_{ixxyzzz}^{(5)} u_{x}^{2} u_{y}^{2} u_{z}^{2} u$$

Here, quantities  $\varphi_i$  and  $\Phi_i$  are

$$\varphi_{i} = \begin{cases} \overline{\alpha}, & |c_{i}|^{2} = 0, \\ 2(1 - \overline{\alpha})/19, & |c_{i}|^{2} = 1, \\ (1 - \overline{\alpha})/38, & |c_{i}|^{2} = 2, \\ (1 - \overline{\alpha})/152, & |c_{i}|^{2} = 3. \end{cases}$$
(D6)

and

$$\Phi_{i} = \begin{cases}
-3\overline{v}(\mathbf{u} \cdot \nabla \rho), & |c_{i}|^{2} = 0, \\
+16\overline{v}(\mathbf{G} : c_{i} \otimes c_{i}), & |c_{i}|^{2} = 1, \\
+4\overline{v}(\mathbf{G} : c_{i} \otimes c_{i}), & |c_{i}|^{2} = 2, \\
+1\overline{v}(\mathbf{G} : c_{i} \otimes c_{i}), & |c_{i}|^{2} = 3,
\end{cases}$$
(D7)

where  $\otimes$  is the tensor (outer) product and: stands for the contraction of the index (Frobenius inner product). The tensor **G** is defined as

$$\mathbf{G} = \frac{1}{48} \left[ \mathbf{u} \otimes \nabla \rho + (\mathbf{u} \otimes \nabla \rho)^{\mathsf{T}} \right], \tag{D8}$$

where gradients are computed by a second-order isotropic central scheme.  $^{98}$   $\bar{\nu}$  is the kinematic viscosity interpolated by  $^{92}$ 

$$\frac{1}{v} = \frac{1+\phi}{2} \frac{1}{v_r} + \frac{1-\phi}{2} \frac{1}{v_b}.$$
 (D9)

To distinguish the two components, the order parameter  $\phi$  is introduced, which is

$$\phi = \left(\frac{\rho_r}{\rho_r^0} - \frac{\rho_b}{\rho_b^0}\right) / \left(\frac{\rho_r}{\rho_r^0} + \frac{\rho_b}{\rho_b^0}\right). \tag{D10}$$

The values  $\phi = 1, -1$ , and 0 correspond to a purely red fluid, a purely blue fluid, and the interface, respectively. To obtain a stable interface, the density ratio between the fluids must be taken into account as follows to obtain a stable interface:<sup>99</sup>

$$\frac{\rho_r^0}{\rho_b^0} = \frac{1 - \alpha_b}{1 - \alpha_r},\tag{D11}$$

where the superscript "0" indicates the initial value of the density at the beginning of the simulation. <sup>100</sup> The pressure of the fluid is given as an isothermal equation of state,

$$p = \rho \left(c_s^k\right)^2 = \rho_k \frac{9(1 - \overline{\alpha})}{19},\tag{D12}$$

for the D3Q27 lattice, where  $c_s^k$  is the speed of sound of the fluid k,  $\overline{\alpha}$  is interpolated by

$$\overline{\alpha} = \frac{1+\phi}{2}\alpha_r + \frac{1-\phi}{2}\alpha_b,\tag{D13}$$

with  $\alpha_b = 8/27$ , and  $c_s^b = 1/\sqrt{3}$ .

Following Refs. 91, 98, and 101, the interfacial tension is modeled by the so-called perturbation operator,

$$(\Omega_i)^{(2)} = \frac{A}{2} |\nabla \phi| \left[ w_i \frac{(\boldsymbol{c}_i \cdot \nabla \phi)}{|\nabla \phi|^2} - B_i \right], \tag{D14}$$

where

$$B_{i} = \begin{cases} -10/27, & |c_{i}|^{2} = 0, \\ +2/27, & |c_{i}|^{2} = 1, \\ +1/54, & |c_{i}|^{2} = 2, \\ +1/216, & |c_{i}|^{2} = 3. \end{cases}$$
(D15)

Eventually, the following recoloring operator is applied to promote phase segregation and maintain the interface,

$$(\Omega_i^r)^{(3)} = \frac{\rho_r}{\rho} f_i + \gamma \frac{\rho_r \rho_b}{\rho^2} \cos(\theta_i) f_i^{eq}(\rho, \mathbf{0}), \tag{D16}$$

$$(\Omega_i^b)^{(3)} = \frac{\rho_b}{\rho} f_i - \gamma \frac{\rho_r \rho_b}{\rho^2} \cos(\theta_i) f_i^{eq}(\rho, \mathbf{0}), \tag{D17}$$

where  $\gamma = 0.7^{66,92,93}$  and  $\theta_i$  is the angle between  $\nabla \phi$  and  $c_i$ , which is defined by

$$\cos(\theta_i) = \frac{c_i \cdot \nabla \phi}{|c_i| |\nabla \phi|}.$$
 (D18)

# APPENDIX E: GENERAL MRT LBM

In Sec. II, we derived a CM-based LBM by defining a matrix  $\overline{T}$ , allowing us to move from the velocity space to the CMs one (and vice versa). This matrix can be seen as the product of two contributions,  $^{37,48,102}$ 

$$\overline{T} = ST.$$
 (E1)

The latter depends only on the chosen lattice, while the former accounts for velocity-dependent terms. In other words, the matrix  $\mathcal S$  represents the effect of shifting the lattice direction by the local fluid velocity. Therefore, it plays the role of a *shifting* matrix, allowing us to move from the frame at rest to the comoving one. Interestingly, the classical RMs-based MRT can be immediately derived by the present approach when the shifting matrix is disregarded (i.e.,  $\overline{\mathcal T} = \mathcal T$ ). Consistently, the methodology outlined in Sec. II can be interpreted as a general MRT-LBM, where the CM-based approach degrades into the RMs-based one when  $\mathcal S = I$ .

Let us define precollision, equilibrium, and postcollision RMs as

$$|m_{i}\rangle = [m_{0}, \dots, m_{i}, \dots, m_{26}]^{\mathsf{T}},$$
  
 $|m_{i}^{eq}\rangle = [m_{0}^{eq}, \dots, m_{i}^{eq}, \dots, m_{26}^{eq}]^{\mathsf{T}},$  (E2)  
 $|m_{i}^{\star}\rangle = [m_{0}^{\star}, \dots, m_{i}^{\star}, \dots, m_{26}^{\star}]^{\mathsf{T}},$ 

respectively. Consistent with the CM-based approach, RMs are computed by applying the transformation matrix  $\mathcal T$  to the populations, which is

$$|m_i\rangle = \mathcal{T}^{\mathsf{T}}|f_i\rangle, \quad |m_i^{eq}\rangle = \mathcal{T}^{\mathsf{T}}|f_i^{eq}\rangle.$$
 (E3)

By adopting the full set of Hermite polynomials, equilibrium RMs read as follows:  $^{37}$ 

$$\begin{split} &m_{1}^{eq} = \rho, \\ &m_{1}^{eq} = \rho u_{x}, \\ &m_{2}^{eq} = \rho u_{y}, \\ &m_{3}^{eq} = \rho u_{y}, \\ &m_{4}^{eq} = \rho u_{x} u_{y}, \\ &m_{5}^{eq} = \rho u_{x} u_{z}, \\ &m_{6}^{eq} = \rho u_{y} u_{z}, \\ &m_{7}^{eq} = \rho \left(u_{x}^{2} - u_{y}^{2}\right), \\ &m_{8}^{eq} = \rho \left(u_{x}^{2} - u_{y}^{2}\right), \\ &m_{9}^{eq} = \rho \left(u_{x}^{2} - u_{y}^{2}\right), \\ &m_{10}^{eq} = \rho c_{s}^{2} u_{x} (3u_{y}^{2} + 3u_{z}^{2} + 2), \\ &m_{11}^{eq} = \rho c_{s}^{2} u_{x} (3u_{y}^{2} + 3u_{z}^{2} + 2), \\ &m_{12}^{eq} = \rho c_{s}^{2} u_{z} (3u_{x}^{2} + 3u_{y}^{2} + 2), \\ &m_{13}^{eq} = \rho u_{x} \left(u_{y}^{2} - u_{z}^{2}\right), \\ &m_{14}^{eq} = \rho u_{y} \left(u_{x}^{2} - u_{y}^{2}\right), \\ &m_{15}^{eq} = \rho u_{x} u_{y} u_{z}, \\ &m_{16}^{eq} = \rho u_{x} u_{y} u_{z}, \\ &m_{18}^{eq} = \rho c_{s}^{2} \left[3 \left(u_{x}^{2} u_{y}^{2} + u_{x}^{2} u_{z}^{2} + u_{y}^{2} u_{z}^{2}\right) + 2 \left(u_{x}^{2} + u_{y}^{2} + u_{z}^{2}\right) + 1\right], \\ &m_{18}^{eq} = \rho c_{s}^{2} \left[9 \left(u_{x}^{2} u_{y}^{2} + u_{x}^{2} u_{z}^{2} - u_{y}^{2} u_{z}^{2}\right) + 6u_{x}^{2} + 1\right], \\ &m_{19}^{eq} = \rho c_{s}^{2} \left(u_{y} - u_{z}^{2}\right) \left(2u_{x}^{2} + 1\right), \\ &m_{20}^{eq} = \rho c_{s}^{2} u_{x} u_{z} \left(3u_{y}^{2} + 1\right), \\ &m_{22}^{eq} = \rho c_{s}^{2} u_{x} u_{z} \left(3u_{y}^{2} + 1\right), \\ &m_{23}^{eq} = \rho c_{s}^{4} u_{x} \left(3u_{y}^{2} + 1\right) \left(3u_{z}^{2} + 1\right), \\ &m_{24}^{eq} = \rho c_{s}^{4} u_{z} \left(3u_{x}^{2} + 1\right) \left(3u_{z}^{2} + 1\right), \\ &m_{25}^{eq} = \rho c_{s}^{4} u_{z} \left(3u_{x}^{2} + 1\right) \left(3u_{y}^{2} + 1\right), \\ &m_{26}^{eq} = \rho c_{s}^{4} \left(3u_{x}^{2} + 1\right) \left(3u_{y}^{2} + 1\right), \\ &m_{26}^{eq} = \rho c_{s}^{4} \left(3u_{x}^{2} + 1\right) \left(3u_{y}^{2} + 1\right), \\ &m_{26}^{eq} = \rho c_{s}^{6} \left(3u_{x}^{2} + 1\right) \left(3u_{y}^{2} + 1\right). \\ \end{pmatrix}$$

One can immediately observe that all the equilibrium RMs are nonzero. Moreover, all show a dependence on the fluid velocity. This is explained by the fact that RMs are linear combinations of (nonzero) equilibrium CMs multiplied by powers of the velocity components.<sup>37</sup> In other words, it is not possible to obtain Galilean invariant equilibrium moments when RMs are considered, even if the Hermite polynomial up to the sixth-order are adopted. This has an impact on the numerical stability of the resulting RM-LBM even if its macroscopic behavior remains unchanged.<sup>37</sup>

The postcollision RMs can be written as

$$|m_i^{\star}\rangle = (\mathbf{I} - \mathbf{\Lambda})\mathcal{T}^{\mathsf{T}}|f_i\rangle + \mathbf{\Lambda}\mathcal{T}^{\mathsf{T}}|f_i^{eq}\rangle$$
 (E5)

(E7)

or

$$|m_i^{\star}\rangle = (\mathbf{I} - \mathbf{\Lambda})|m_i\rangle + \mathbf{\Lambda}|m_i^{eq}\rangle$$
 (E6)

and read as follows:

$$m_{1}^{o} = \rho,$$

$$m_{1}^{*} = \rho u_{x},$$

$$m_{2}^{*} = \rho u_{y},$$

$$m_{3}^{*} = \rho u_{z},$$

$$m_{4}^{*} = (1 - \omega) m_{4} + \omega \rho u_{x} u_{y},$$

$$m_{5}^{*} = (1 - \omega) m_{5} + \omega \rho u_{x} u_{z},$$

$$m_{6}^{*} = (1 - \omega) m_{6} + \omega \rho u_{y} u_{z},$$

$$m_{7}^{*} = (1 - \omega) m_{7} + \omega \rho \left(u_{x}^{2} - u_{y}^{2}\right),$$

$$m_{8}^{*} = (1 - \omega) m_{8} + \omega \rho \left(u_{x}^{2} - u_{z}^{2}\right),$$

$$m_{9}^{*} = \rho \left(u_{x}^{2} + u_{y}^{2} + u_{z}^{2} + 1\right),$$

$$m_{10}^{*} = \rho c_{s}^{2} u_{x} \left(3 u_{x}^{2} + 3 u_{z}^{2} + 2\right),$$

$$m_{11}^{*} = \rho c_{s}^{2} u_{y} \left(3 u_{x}^{2} + 3 u_{y}^{2} + 2\right),$$

$$m_{12}^{*} = \rho c_{s}^{2} u_{z} \left(3 u_{x}^{2} + 3 u_{y}^{2} + 2\right),$$

$$m_{13}^{*} = \rho u_{x} \left(u_{y}^{2} - u_{z}^{2}\right),$$

$$m_{14}^{*} = \rho u_{y} \left(u_{x}^{2} - u_{y}^{2}\right),$$

$$m_{15}^{*} = \rho u_{x} \left(u_{y}^{2} - u_{z}^{2}\right),$$

$$m_{16}^{*} = \rho u_{x} u_{y} u_{z},$$

$$m_{17}^{*} = \rho c_{s}^{2} \left[3 \left(u_{x}^{2} u_{y}^{2} + u_{x}^{2} u_{z}^{2} + u_{y}^{2} u_{z}^{2}\right) + 2\left(u_{x}^{2} + u_{y}^{2} + u_{z}^{2}\right) + 1\right],$$

$$m_{18}^{*} = \rho c_{s}^{2} \left[9 \left(u_{x}^{2} u_{y}^{2} + u_{x}^{2} u_{z}^{2} - u_{y}^{2} u_{z}^{2}\right) + 6 u_{x}^{2} + 1\right],$$

$$m_{19}^{*} = \rho c_{s}^{2} \left(u_{y}^{2} - u_{z}^{2}\right) \left(2 u_{x}^{2} + 1\right),$$

$$m_{20}^{*} = \rho c_{s}^{2} u_{x} u_{z} \left(3 u_{y}^{2} + 1\right),$$

$$m_{21}^{*} = \rho c_{s}^{2} u_{x} u_{z} \left(3 u_{y}^{2} + 1\right),$$

$$m_{22}^{*} = \rho c_{s}^{2} u_{x} u_{y} \left(3 u_{x}^{2} + 1\right),$$

$$m_{23}^{*} = \rho c_{s}^{4} u_{x} \left(3 u_{y}^{2} + 1\right) \left(3 u_{z}^{2} + 1\right),$$

$$m_{24}^{*} = \rho c_{s}^{4} u_{y} \left(3 u_{x}^{2} + 1\right) \left(3 u_{y}^{2} + 1\right),$$

$$m_{25}^{*} = \rho c_{s}^{4} \left(3 u_{x}^{2} + 1\right) \left(3 u_{y}^{2} + 1\right),$$

$$m_{26}^{*} = \rho c_{s}^{6} \left(3 u_{x}^{2} + 1\right) \left(3 u_{y}^{2} + 1\right),$$

$$m_{26}^{*} = \rho c_{s}^{6} \left(3 u_{x}^{2} + 1\right) \left(3 u_{y}^{2} + 1\right),$$

$$m_{26}^{*} = \rho c_{s}^{6} \left(3 u_{x}^{2} + 1\right) \left(3 u_{y}^{2} + 1\right),$$

where

$$m_{4} = \sum_{i} f_{i} c_{ix} c_{iy}, \quad m_{5} = \sum_{i} f_{i} c_{ix} c_{iz}, \quad m_{6} = \sum_{i} f_{i} c_{iy} c_{iz},$$

$$m_{7} = \sum_{i} f_{i} (c_{ix}^{2} - c_{iy}^{2}), \quad m_{8} = \sum_{i} f_{i} (c_{ix}^{2} - c_{iz}^{2}).$$
(E8)

Again, all postcollision RMs are nonzero and also show a dependence on the velocity (except for the first conserved moment,  $m_0$ ). Eventually, the postcollision populations are reconstructed as

$$|f_i^{\star}\rangle = (\mathcal{T}^{\mathsf{T}})^{-1}|m_i^{\star}\rangle$$
 (E9)

and streamed [see Eq. (4)].

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