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## Unit = 7

### Properties of the

### Chapter - 5

Chapter - 1<sup>st</sup>

- \* Deforming force: Deforming force is a force which changes shape and size of a body.
- \* Restoring force: The force which helps to regain its original shape and size after removal of deforming force is called restoring force.

\* Elasticity: The property of material of body to regain its original shape and size after removal of deforming force is called Elasticity.

\* Perfectly elastic body: A body which regains 100% of its original shape and size after removal of deformation. It is called perfectly elastic body. There is no permanent change in its shape and size.

body which is perfectly elastic but Quirky and phosphorus Bronze are nearly perfectly elastic.

\* Perfectly plastic body: A body which does not regain its original shape and size at all after removal of deforming force. is called perfectly plastic body.

No body is perfectly plastic body but Paraffin wax are examples of nearly plastic body.

\* Elastic limit: Elastic limit is the largest value of deforming force upto which the body will regain its shape and size after removal of deforming force.

Elastic limit is the property of body whereas Elastic limit is the property of material of body.

## Hydraulic

\* Elastic Fatigue: - If it is the property of an elastic body due to which its elastic behaviour becomes less elastic under the action of repeated deforming force.

\* Stress: The internal restoring force acting per unit area of a deformed body is called stress.

$$\text{Stress} = \frac{\text{Restoring force}}{\text{Area}}$$

S.I. unit of stress is  $\text{N/m}^2$

Dimensional formula  $[\text{M}^1 \text{L}^{-1} \text{T}^{-2}]$

Ques: For perfectly elastic body restoring force will be equal to Deforming force.

\* Types of Stress: There are two types of stress

i) Normal Stress: When a deforming force acts normally ( $\perp$ ) over an area of body then ratio of normal Restoring force per unit area is known as normal stress.

Normal force changes size of body (length, Area, Volume) but do not change shape of body. There are 3-types of normal stress.

a) Tensile Stress: If there is an increase in length of a body then normal stretching force acting per unit area is called tensile stress.

b) Compressive Stress: If there is decrease in length of a body then normal compressive force acting per unit area is called comp. stress.

(c) Hydrolic Stress: The hydrolic force (Hydost) acting per unit area is called Hydrolic Stress. It deals with fluids like gases and liquids.

\* Strain: When a deforming force is applied on a body, a change in shape, size, vol<sup>m</sup> etc. of the body take place.

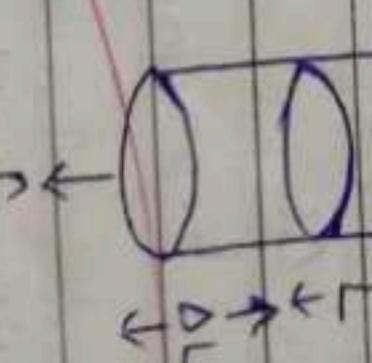
The ratio of change in configuration is called strain. S.I. unit of strain is no unit.

Dimensional formula

doesn't exist.

\* Types of Strain: There are three types of strain.

i) Longitudinal Strain:



The ratio of change in length to original length is

called longitudinal strain.

$$\text{Longitudinal strain} = \frac{\Delta L}{L}$$

It will be unitless or dimensionless.

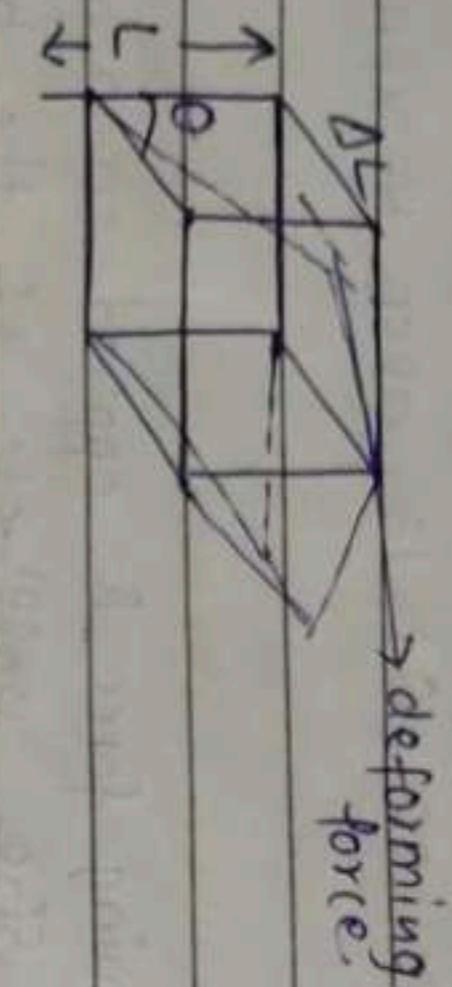
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i) Volumetric Strain: The ratio of change in vol<sup>m</sup> to original vol<sup>m</sup> is called volumetric strain.

$$\text{Volumetric strain} = \frac{\Delta V}{V}$$

g+ will be unitless, dimensionless.

iii) Shearing (Tangential) strain:



When a tangential force is applied, it changes shape of body. If lower plane is fixed than their is relative displacement of upper plane w.r.t lower plane takes place.

The angle by which shape of body changes is called Shearing strain.

$$\text{Shearing strain} = \theta$$

$$\theta = \frac{\Delta L}{L}$$

\* Modulus of elasticity: From Hooke's law

$$\text{Stress} = E \cdot \text{Strain}$$

$$E = \frac{\text{Stress}}{\text{Strain}}$$

$$\text{Unit of } E = \text{Nm}^{-2}$$

$$\text{Dimensional formula} = [M^1 L^1 T^{-2}] [L^2] = [M^1 L^{-1} T^{-2}]$$

i) Young's Modulus of elasticity: There are three types of modulus of elasticity.

g+ is ratio of relative displacement ( $\Delta L$ ) of one plane to the lar dist. (L) from fixed plane.

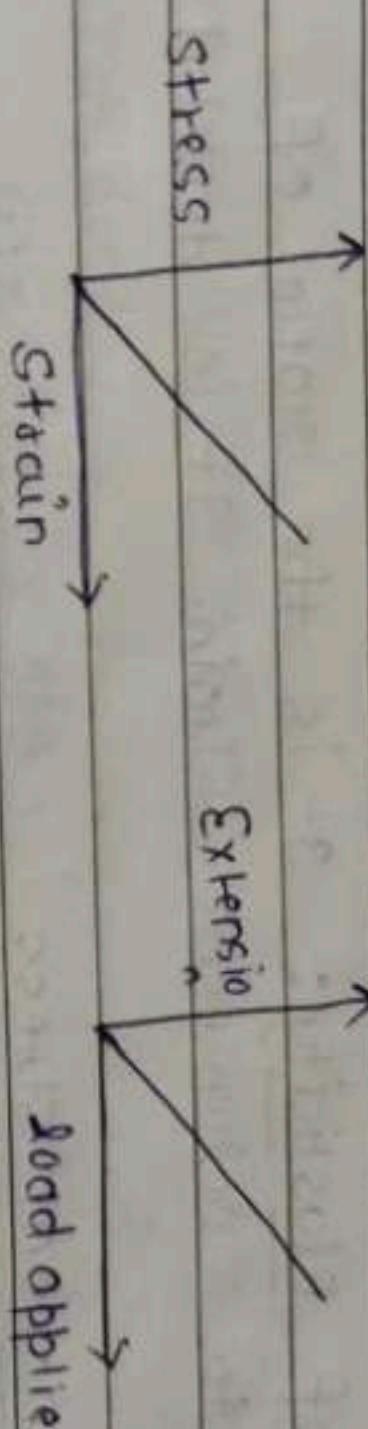
\* Hook's Law in elasticity: Within elastic limit, stress will be directly prop. to strain produced in the body i.e.

stress  $\propto$  strain.

OR

Within elastic limit, extension in body will be directly prop. to load applied.

extension  $\propto$  load applied



\* Modulus of elasticity:

From Hooke's law

$$\text{Stress} = E \cdot \text{Strain}$$

$$E = \frac{\text{Stress}}{\text{Strain}}$$

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$$Y = \frac{F/A}{\Delta L/L}$$

Unit of  $Y = \text{Nm}^{-2}$

D.F. of  $Y = [M^1 L^{-1} T^{-2}]$

2) Bulk Modulus of elasticity: It is the ratio of normal stress to volumetric strain. It is denoted by 'K' or 'B'

$K = \frac{\text{normal stress}}{\text{volumetric strain}}$

$$\text{Volumetric strain} = \frac{\Delta V}{V}$$

$$K = \frac{F/A}{\Delta V/V}$$

$$\text{Unit of } K = \text{Nm}^{-2}$$

$$\text{D.F.} = [M^1 L^{-1} T^{-2}]$$

3) Modulus of Rigidity: It is the ratio of tangential stress to shearing strain. It is denoted by 'G'.

$G = \text{tangential stress}$

Shearing strain

$$\text{Elastic P.E.} = \frac{1}{2} \times \text{stress} \times \text{strain} \times \text{volm}$$

$$\text{as } \frac{F}{A} = \text{stress}, \quad \frac{\Delta L}{L} = \text{strain}$$

$$\Delta L = \text{volm}$$

$$\text{as Young's Modulus of elasticity} = \frac{\text{stress}}{\text{strain}} = Y \times \text{strain} =$$

$$\text{as, } G = \frac{\Delta L/L}{\Delta S/S}$$

$$S_0, \quad G = \frac{F/A}{\Delta L/L}$$

$$\text{Unit of } G = \text{Nm}^{-2}$$

$$\text{D.F.} = [M^1 L^{-1} T^{-2}]$$

\* Elastic P.E. of a stretched wire:

Consider a wire of length 'L' having area of cross-section 'A'. Let a deforming force applied on wire increases from 0 to F.

Then average deforming force is  $\frac{0+F}{2} = \frac{F}{2}$

$$\omega = \text{Force} \times \text{displ.}$$

$$\omega = \frac{F}{2} \times \Delta L$$

as work done is stored in the form of elastic p.e

$$\text{then, } \text{E.P.E.} = \omega = \frac{1}{2} \times F \times \Delta L$$

$$\text{Now, Elastic P.E.} = \frac{1}{2} \times F \times \Delta L \times \Delta L$$

$$= \frac{1}{2} \times A \times \frac{\Delta L}{L} \times \Delta L$$

$$\text{as, } \frac{F}{A} = \text{stress}, \quad \frac{\Delta L}{L} = \text{strain}$$

$$\text{Elastic P.E.} = \frac{1}{2} \times \text{stress} \times \text{strain} \times \text{volm}$$

$$\text{as Young's Modulus of elasticity} = \frac{\text{stress}}{\text{strain}} = Y \times \text{strain} =$$

$$\text{or Strain} = \frac{\text{stress}}{Y}$$

$$\text{Elastic P.E.} = \frac{1}{2} \times Y \times \text{strain} \times \text{strain} \times \text{volm}$$

$$\frac{1}{2} \times Y \times (\text{strain})^2 \times \text{volm}$$

$$\text{Elastic P.E.} = \frac{1}{2} \times \text{stress} \times \text{stress} \times \text{volm}$$

It consists of a master cylinder ('M'). A large piston 'P' and two small pistons  $P_1$  and  $P_2$ . The master cylinder is filled with brake oil. The piston P is connected to break pedal. Small piston  $P_1$  and  $P_2$  are connected to break shoe  $S_1$  and  $S_2$ .

When breaks are applied, we press pedal P. Due to this piston P is pushed inwards. So, a press is created acc to pascal law equal press is shared by piston  $P_1$  and  $P_2$ . Piston  $P_1$  and  $P_2$  will push break shoe  $S_1$  and  $S_2$ . As a result breaks are applied.

\* Atmospheric pressure: The pressure exerted by atmosphere is called atmospheric pressure. The value of atmospheric pressure is  $1.01 \times 10^5 \text{ Nm}^{-2}$  or  $1.01 \times 10^5 \text{ Pascal}$  or  $1.01 \times 10^6 \text{ dyn.cm}^{-2}$  or  $76 \text{ cm}$  of mercury atom. It is measured by Barometer.

\* Buoyant force or Buoyancy: The upward force exerted by fluid on a body when body is immersed in the fluid is called buoyant force. The process is called Buoyancy.

$$\text{Buoyant force } (F_B) = mg$$

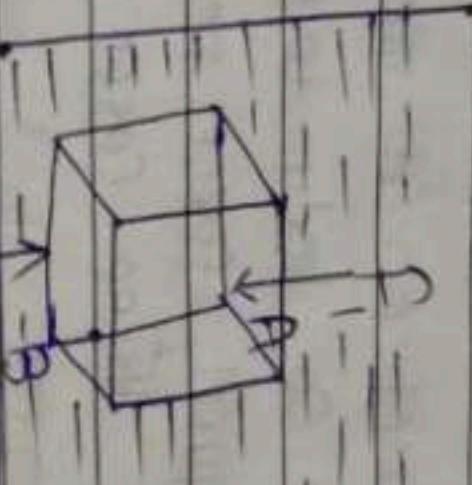
$$F_B = \gamma \cdot V \cdot g$$

Where,  $\gamma$  = density of fluid

$$V = V_{\text{displaced}}$$

\* Archimede's Principle: When a body is immersed in a fluid, loss in weight of body take place. This loss in weight will be equal to weight of fluid displaced by body.

### Proof:



Consider a liquid of density 'γ' is placed in a container. A vessel of cubic shape is immersed in liquid.

$$\text{Let } F_2 \text{ be the force exerted by upper layer of liquid}$$

$$\text{As, we know } P = h \gamma g$$

$$P_1 = \frac{F_1}{A}$$

Where 'A' is area of upper face

$$F_1 = P_1 A$$

$$\text{Let } F_1 \text{ be the force exerted by lower liquid on face}$$

$$P_2 = (\alpha + h) \gamma g$$

$$P_2 = \frac{F_2}{A}$$

$$F_2 = P_2 A$$

$$= (\alpha + h) \gamma g A - 2)$$

$$\text{Net force on the vessel} = F_2 - F_1$$

$$\text{Net force} = (\alpha + h) \gamma g \cdot A - \alpha \gamma g A$$

$$\text{Net force} = \gamma g A$$

$$= V \gamma g [h \cdot A = V \cdot \gamma]$$

Net force =  $mg$  [ $V \cdot \gamma = \text{mass}$ ]  
Where 'mg' is weight of liquid displaced by vessel which is equal to buoyant force.

Total force on cylinder

$$mg + F_1 = F_2$$

$$\text{as, } M = f A h$$

$$F_1 = f A h g = F_2$$

$$P_1 A + f A h g = P_2 A$$

$$A(P_1 + f h g) = P_2 A$$

$$P_1 + f h g = P_2$$

$$P_2 - P_1 = f h g$$

so, with increase in depth, press exerted by liq increases

Special case :-  $gf h = 0$

$$P_2 - P_1 = 0$$

$$P_2 = P_1$$

$$P_2 - P_1 = 0$$

\* Pascal's Law: If effect of gravity is neglected then press at all points inside liq. will be same.

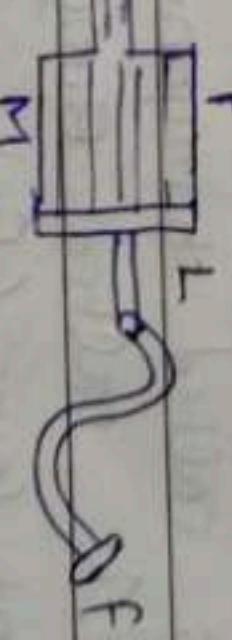
OR

It states that the pressure exerted at 1 point on a liq. will be equally transmitted to all other points of the liq., i.e.  $P_1 = P_2$

$$A >> a \text{ then } F_2 >> F_1$$

2) Hydraulic Brakes: These are used in automobiles vehicles for applying brakes. It's working is based on pascal law.

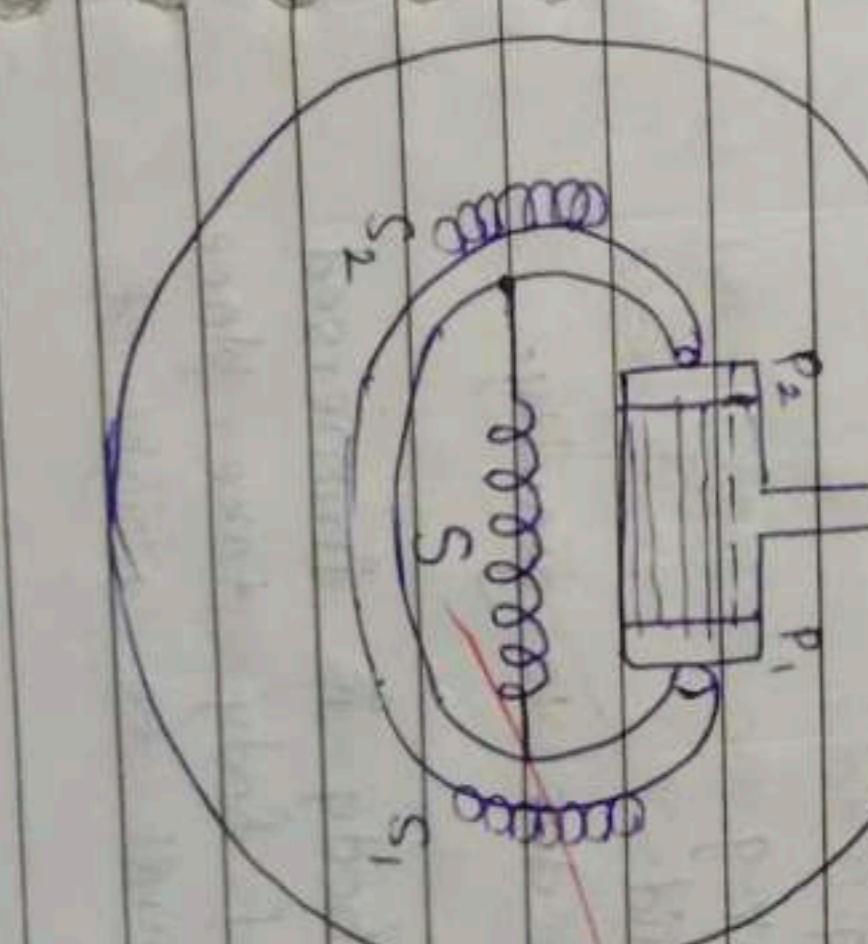
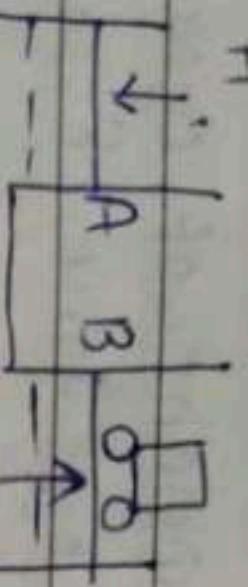
To other wheels



Applications of Pascal's Law:

Hydraulic lift: It is used to lift heavy loads.

Working is based on pascal's law.



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**: Hydrostatics:**

\* **Hydrostatics:** Hydrostatic is the branch of physics which deals with study of properties of fluids at rest.

\* **Fluid:** Fluids are the substances which begin to flow when external force is applied on them. Liquids and gases are examples of fluids.

**Properties:**

- They have definite vol<sup>m</sup>.
- They do not have definite shape.
- Fluids exert force on container wall.

\* **Thrust:** The force exerted by fluid on container wall is called thrust. Its S.I. unit is 'N'.

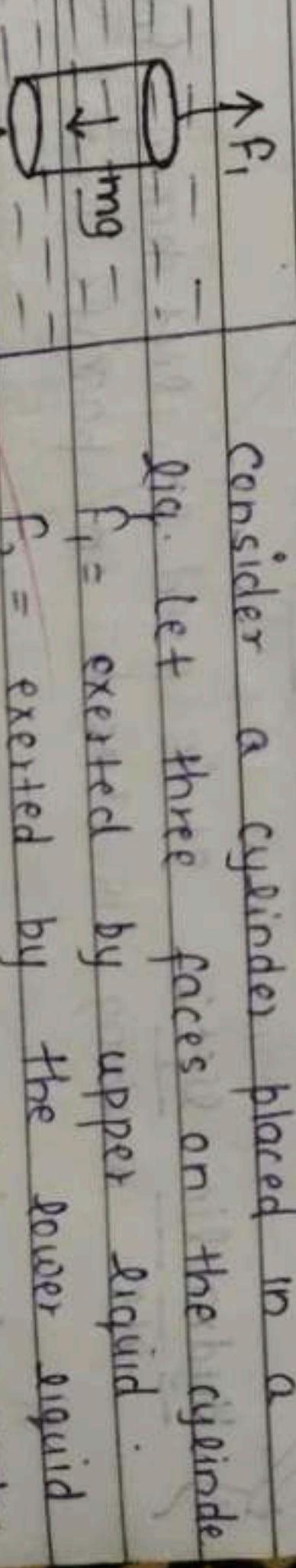
$$D.P. = [M^1 L^{-1} T^{-2}]$$

\* **Pressure of fluid:** Thrust exerted by fluid per unit area is known as hydrostatic press.

$$\rho = \frac{F}{A}$$

$$D.P. = \frac{F}{A}$$

\* **Variation of press. with depth:**



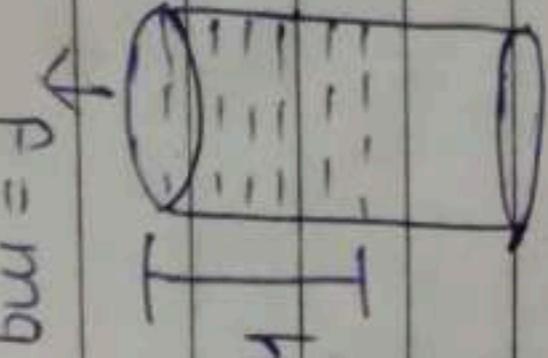
Consider a cylinder placed in a liquid. Let three faces on the cylinder be:-

- $P_1$  = exerted by upper liquid.
- $P_2$  = exerted by the lower liquid.
- $P_3$  = mg = be the weight of cylinder.

\* **Expression for press. exerted by liquid (fluids) :-**

Consider a cylindrical container having

a liq. of density 'g' filled up to height 'h'. 'A' be the area of container.



$$P = mg$$

$F$  be the force exerted by liquid.

$$F = mg \rightarrow i)$$

$$as \ density 'g' = \frac{\text{Mass}}{\text{Volm}}$$

$$I = \frac{M}{V}$$

$$f = \frac{M}{A \times h}$$

$$M = I \cdot A \cdot h \rightarrow 2)$$

Let 'P' be the press. exerted by liq.

$$P = \frac{F}{A}$$

using eqn 1) and ii)

$$P = \frac{mg}{A}$$

$$mg = \frac{I \cdot A \cdot h \cdot g}{A}$$

$$P = I \cdot h \cdot g$$

$$resp. P_1 = \frac{P_1}{A}, P_2 = \frac{P_2}{A}$$

$P_1$  and  $P_2$  be the press. exerted by force  $f_1$  and  $f_2$ .

$$\text{Elastic P.E.} = \frac{1}{2} \times (\text{Stress})^2 \times \text{Vol}^m$$

as, Elastic P.E. density =  $\frac{\text{Elastic P.E.}}{\text{Vol}^m}$

$\Rightarrow$  Elastic P.E. density =  $\frac{1}{2} \times \text{Stress} \times \text{Strain}$

$$\text{Or elastic P.E. density} = \frac{1}{2} \times Y \times (\text{Strain})^2$$

$$\text{Elastic P.E. density} = \frac{1}{2} \times \frac{(\text{Stress})^2}{Y}$$

\* Poisson's Ratio:  $\gamma$  is the ratio of lateral strain

to longitudinal strain.

$$\text{Poisson's Ratio} = \frac{\text{Lateral strain}}{\text{Longit. strain}}$$

$$\text{as, Lateral strain} = -\frac{\Delta R}{R} \quad (\text{change in radius})$$

$$\text{for solids: } Y = \frac{\text{Thermal stress}}{\text{Thermal strain}}$$

$$\text{longitudinal strain} = \frac{\Delta L}{L}$$

$$\text{Poisson's Ratio} = -\frac{\Delta R}{R} = -\frac{\Delta R \times L}{R \times \Delta L} = -\frac{L}{R} \frac{\Delta R}{\Delta L}$$

$$Y = \frac{\text{Thermal stress}}{\alpha \cdot \Delta T}$$

$$\text{Thermal stress} = Y \cdot \alpha \cdot \Delta T$$

$\gamma$  is denoted by ' $\sigma$ '  
 $\gamma$  is unitless, dimensionless

in terms of diameter

$$\sigma = -\frac{\Delta D}{D} = \frac{\Delta L}{L}$$

$$\sigma = -\frac{L}{D} \times \frac{\Delta D}{\Delta L}$$

$$4) \quad \frac{\alpha}{Y} = \frac{1}{B} + \frac{3}{n}$$

\* Thermal stress: When a fixed bar has ends is heated, its length increases.

Let 'L' be the length of rod, ' $\Delta L$ ' be the extension in length and  $\Delta T$  be the temp. provided to the rod.

$$\text{Then, } \Delta L \propto L \quad \text{or} \quad \Delta L \propto \Delta T$$

$$\text{Comparing eqn 1 and 2)} \\ \Delta L \propto L \Delta T$$

$$\Delta L = \alpha L \Delta T$$

where ' $\alpha$ ' is called proportionality constant.

$$\frac{\Delta L}{L} = \alpha \cdot \Delta T$$

where,  $\frac{\Delta L}{L}$  is called thermal stress.

$$Y = \frac{\text{Thermal stress}}{\text{Thermal strain}}$$

$$Y = \frac{\text{Thermal stress}}{\alpha \cdot \Delta T}$$

$$\text{Thermal stress} = Y \cdot \alpha \cdot \Delta T$$

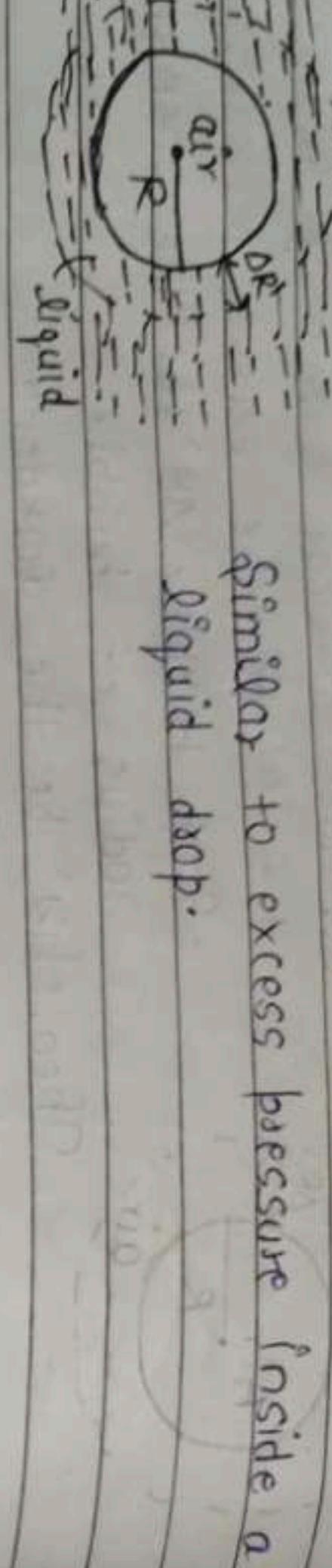
\* Relation b/w  $Y, B, \gamma, \sigma$

$$1) \quad Y = 2B(1 - 2\sigma)$$

$$2) \quad Y = 2\gamma(1 - \sigma)$$

$$3) \quad \sigma = \frac{3B - 2\gamma}{2\gamma + 6B}$$

\* Excess press. inside an Air bubble in water:



Similar to excess pressure inside a liquid drop.

Then force is  
 $F \propto A \cdot \frac{dP}{dr}$  — i)  
 $F \propto A \cdot \frac{dv}{dx} — ii)$   
 where  $\frac{dv}{dx}$  is called velocity gradient

### CHAPTER = 3 Hydrodynamics

$$F = \eta \cdot A \cdot \frac{dv}{dx}$$

Unit of coff. of viscosity ( $\eta$ ) is 'Poise'  
 Dimensional formula =  $\frac{M^1 L^{-2} T^{-1}}{A \cdot L^2}$

$$\eta = [M^1 L^{-1} T^{-1}]$$

\* It is the branch of physics which deals with study of fluids during their motion.

\* Viscosity :-

- → — —  $v + dv$
- → — — ↑ Velocity
- → — —  $v$  increases

The property by which fluid layers oppose their relative motion is called viscosity.

OR

The property by which internal friction b/w diff. layer of fluids takes place which opposes their relative motion is called viscosity.

Let us consider a fluid flowing in layers. The velocity of fluid in diff. layers will increase as we go from bottom to top.  
 let 'dv' be the small change in velocity  
 'dr' be the small gap b/w layers  
 'A' be the area of layers  
 'F' be the viscous force.

\* Factors affecting viscosity:  
 1) If temp. is increased, the viscosity of liquid decrease  
 2) If press. is increased, viscosity of liquid decreases except water. Viscosity of water increases with increase in press. While viscosity of gases do not change with press.

\* Stake's law: If a body is dropped in a container filled with fluid then fluid will apply an opposing force (viscous force on the body) Acc. to this law, viscous force is given by

$$f = 6\pi\eta r v$$

where, 'r' is radius of body

$$\eta = \text{coff. of viscosity}$$

$v_t$  = terminal velocity.

consider a liquid drop or radius 'R'. Let 'P' be the excess pressure inside liquid drop. ' $\Delta R$ ' be the change in radius of drop.

Then, ' $\omega$ ' be the workdone

$$\omega = \text{Force} \times \text{displ.}$$

$$\omega = F \times \Delta R$$

$$\text{as, Press.} = \frac{\text{Force}}{\text{Area}}$$

$$P = \frac{F}{4\pi R^2}$$

$$F = P \cdot 4\pi R^2 - \downarrow$$

$$\omega = P \cdot 4\pi R^2 \cdot \Delta R$$

$$\text{Surface energy} = \omega = \text{Surface tension} \times \frac{\text{Increase in Surface area}}{\text{Surface area}}$$

$$\text{Increase in Surface area} = \frac{\text{Final Surface Area} - \text{Initial Surface Area}}{\text{Initial Surface Area}}$$

$$\text{Increase in Surface area} = 4\pi(R + \Delta R)^2 - 4\pi R^2$$

$$= 4\pi(R^2 + \Delta R^2 + 2R \cdot \Delta R) - 4\pi R^2$$

as,  $\Delta R$  is small so  $\Delta R^2$  will be neglected

Thus,

$$\text{Increase in Surface area} = 4\pi R^2 + 8\pi R \cdot \Delta R - 4\pi R^2$$

Because there are two free surfaces  
Increase in Surface area =  $2[4\pi(R + \Delta R)^2 - 4\pi R^2]$

$$= 2[4\pi(R^2 + \Delta R^2 + 2R \cdot \Delta R) - 4\pi R^2]$$

$$\text{Surface energy} = \omega = S \times 8\pi R \cdot \Delta R$$

$$\text{Equating } ① \text{ and } ② \quad P \cdot 4\pi R^2 \cdot \Delta R = S \times 8\pi R^2 \cdot \Delta R$$

$$\text{Increase in Surface area} = 2[4\pi R^2 + 8\pi R \cdot \Delta R - 4\pi R^2]$$

$$= 16\pi R \cdot \Delta R$$

$$\text{Surface energy} = \omega = S \cdot 16\pi R \cdot \Delta R - \downarrow$$

$$\text{equating } ① \text{ and } ② \quad 4$$

$$P \cdot 4\pi R^2 \cdot \Delta R = S \times 16\pi R \cdot \Delta R$$

$$P = \frac{4S}{R}$$

where 'S' is surface tension and 'R' is radius of soap bubble

Where 'S' is surface tension and 'R' is radius of soap bubble

### \* Excess press. inside a soap bubble:

Consider a soap bubble of Radius 'R'  
Let 'P' be the excess pressure inside  
Soap bubble. ' $\Delta R$ ' be the change in  
radius of bubble.

$$\omega = \text{Force} \times \text{displ.}$$

$$\omega = F \cdot \Delta R$$

$$\text{as, Pressure} = \frac{\text{Force}}{\text{Area}}$$

$$P = \frac{F}{4\pi R^2}$$

$$P = P \cdot 4\pi R^2$$

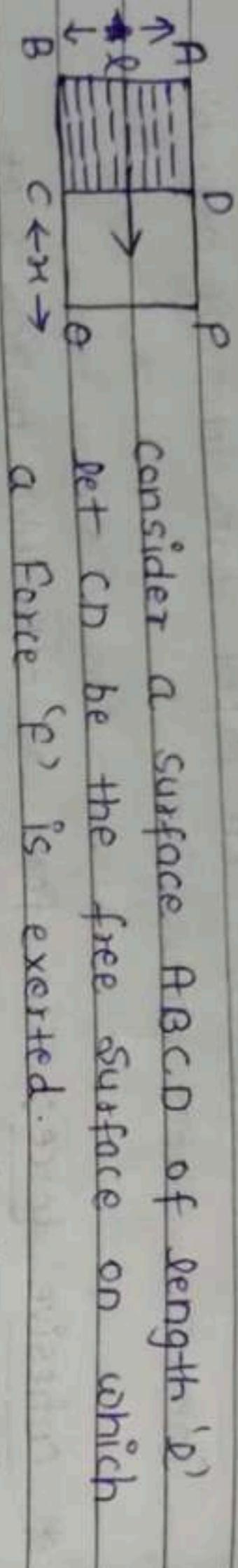
$$\omega = P \cdot 4\pi R^2 \cdot \Delta R$$

$$\text{Increase in Surface area} = 2[4\pi R^2 + 8\pi R \cdot \Delta R - 4\pi R^2]$$

$$= 16\pi R \cdot \Delta R$$

$$\text{Surface energy} = \omega = S \cdot 16\pi R \cdot \Delta R - \downarrow$$

\* Surface Energy: Workdone against surface tension to increase surface area is called surface energy.



Consider a surface ABCD of length 'l' let CD be the free surface on which a force 'F' is exerted.

'x' be the displacement of the surface.

Now, workdone( $\omega$ ) = Force  $\times$  displacement

$$= Fx \quad (i)$$

Let 'S' be the surface tension

$$S = \frac{F}{2l} \left[ \text{Total length of free Surface} = l + x \right]$$

$$S = \frac{F}{2l}$$

$$F = S \cdot 2l$$

Put this value in eq (i)

$$\omega = S \cdot 2l \cdot x$$

So, Surface energy =  $\omega = S \cdot 2l \cdot x$

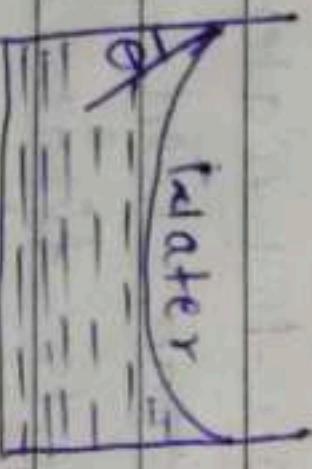
as,  $2l \cdot x$  = Surface area

Surface energy = Surface tension  $\times$  increase in Surface area

### \* Angle of Contact:

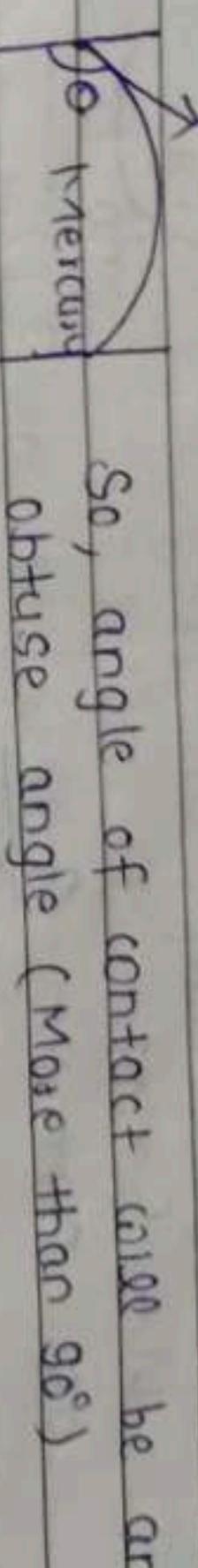
The angle b/w solid surface and a tangent of liquid surface is known as angle of contact.

- i) If Adhesive force  $>$  cohesive force then a concave meniscus will be formed.



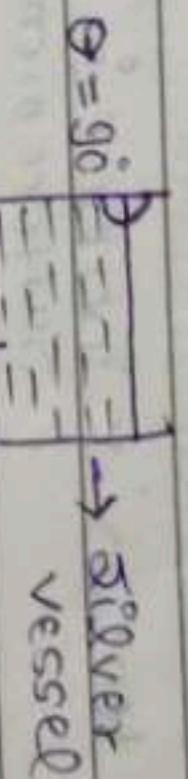
So, angle of contact will be an acute angle less than  $90^\circ$

→ If Adhesive force  $<$  cohesive force then a convex meniscus will be formed.



So, angle of contact will be an obtuse angle (More than  $90^\circ$ )

→ If adhesive force = cohesive force then a plane meniscus will be formed.



Liquid so, angle of contact will be  $90^\circ$

### \* Factors affecting angle of contact:

- i) Nature of solid and liquid in contact.
- ii) Temp. of liquid.
- iii) Medium above free surface of liquid

Surface energy = Surface tension  $\times$  increase in Surface area

- \* Excess pressure: Excess pressure is the pressure difference b/w two sides of surface. Due to excess press. the surface will either lower meniscus or upper meniscus. Due to excess press. surface area of surface increases.

\* Excess pressure due to a liquid drop:



\* Laws of floatation: When a body is immersed in a fluid or upward buoyant force act on it and weight of body acts downward. So, body will float or not, it will depend on buoyant force and weight of body. There will be three cases:-

Case I :- The buoyant force is greater than weight of body i.e.

$$f_B > \omega$$

$$m_g > M$$

$m > M$   $M$  is mass of body where ' $m$ ' is mass of liquid displaced. If density of body is less than density of fluid then the body will float on the surface of fluid.

Case II :- If buoyant force is less than weight

$$f_B < \omega$$

$$m_g < M$$

$$m < M$$

Thus  $f < \sigma$  If density of body greater than density of fluid then body will sink into the fluid.

Case III :- If buoyant force is equal to weight of body

$$\text{i.e } f_B = \omega$$

$$m_g = M$$

$$\text{then, } f = \sigma$$

If density of a body is equal to density of fluid then the body will remain in water inside the fluid.

\* Adhesive force: The force of attraction b/w molecules of two diff. surfaces are called adhesive forces.

Ex: When we write on paper, the force of att. b/w ink and paper molecules.

\* Cohesive force: The force of att. b/w molecules of same substance are called cohesive force. Ex: Solids have definite shape and size due to their strong force of cohesive.

\* Surface Tension: The property of fluid due to which it tends to minimize its surface area is called Surface tension.

It arises due to cohesive force. Mathematically surface tension is the ratio of force per unit length of the surface. It is denoted by 'S'.

$S = \frac{F}{L}$  where,  $F = \text{Force}$   
 $L = \text{Length of surface}$ .

SI unit is  $\text{Nm}^{-1}$

$$\text{D.P} = \frac{[\text{M}^1 \text{L}^{-2}]}{[\text{L}^1]} = [\text{M}^1 \text{L}^0 \text{T}^{-2}]$$

\* Applications of surface tension:

1) Rain drops are generally spherical in shape.

2) Oil drop spread on the surface of cold water but remain as drop on hot water because surface tension of oil is less than cold water but greater than hot water.

3) Small insects float on water surface.

\* Reynold number: It is a pure no. which determines type of flow. It is denoted by  $N_R$ .

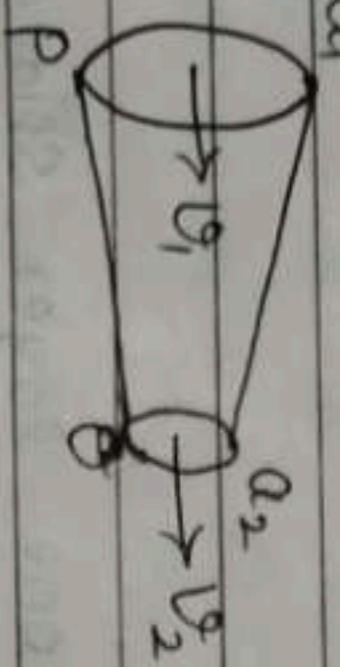
$$N_R = \frac{V_c \cdot f \cdot D}{\eta}$$

If  $N_R < 2000$  then flow is streamline laminar

If  $N_R > 3000$  then flow is turbulent flow.

If  $2000 < N_R < 3000$ , then flow is unstable. Sometimes flow becomes laminar while sometimes it becomes turbulent.

\* Eqn of continuity:



Velocity of fluid coming out from a hole is inversely proportional to area of cross-section of hole such that,

$$V \propto \frac{1}{A}$$

where 'V' is velocity of fluid

\* Energy of fluid: There are three types of energy in a flowing fluid — kinetic energy, potential energy and pressure energy.

1) K.E → The energy which comes due to motion of fluid is known as K.E. It is given by  $\frac{1}{2}mv^2$

K.E per unit mass is given by

$$\frac{K.E}{m} = \frac{\frac{1}{2}mv^2}{m} = \frac{1}{2}fv^2$$

$av = \text{constant}$

This is called eqn of continuity.

\* Proof: Consider a pipe of variable cross-section. Area of cross-section of end P is greater than that of end O

Then mass of fluid = mass of fluid at end O at end P

$$m_1 = m_2$$

Let 'f' be the density of fluid.  $V_1$  and  $V_2$  be the vols of fluid at end P and O resp.

$$fv_1 = fv_2$$

$$v_1 = v_2$$

$$\frac{v_1}{t} = \frac{v_2}{t}$$

$$a_1 \cdot d_1 = a_2 \cdot d_2$$

$$a_1 v_1 = a_2 v_2 \quad [\because d = v]$$

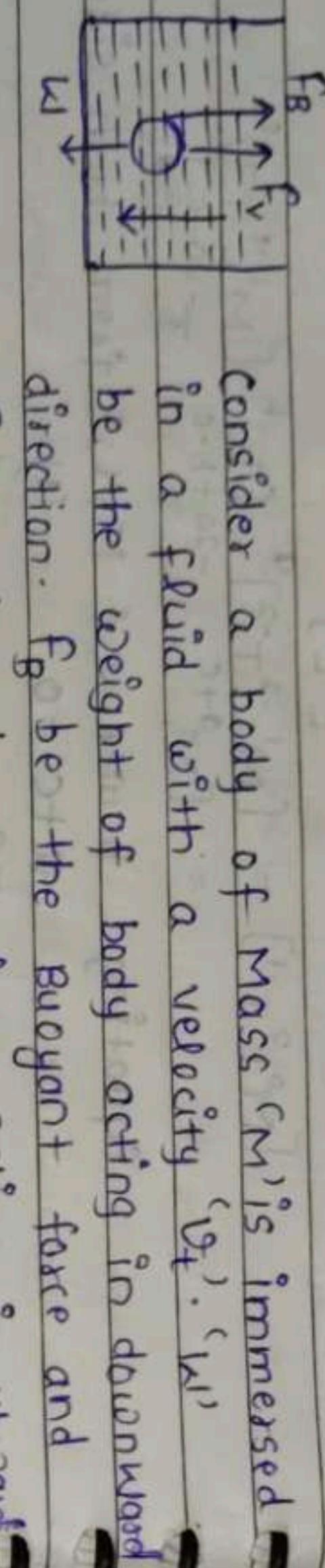
$$\Rightarrow av = \text{constant}$$

$$as, V = a \cdot d$$

Where 'd' is distance travelled by fluid

\* Terminal velocity:  $v_t$  is the maximum constant velocity acquired by body while falling freely in a fluid.  $g_t$  is denoted by  $v_t$ .

$$v_t = \frac{2(\sigma - f) \rho g}{\eta C_D}$$



Consider a body of mass 'M' is immersed in a fluid with a velocity ' $v_t$ '. ' $\omega$ ' be the weight of body acting in downward direction.  $F_B$  be the buoyant force and  $F_v$  be the viscous force acting in upward.

$$\omega = Mg$$

$F_B = \text{Weight of liquid displaced}$

$$F_B = mg$$

$$F_v = 6\pi\eta D v_t$$

Net force acting on body is  $\omega - [F_B + F_v]$

$$Ma = \omega - (F_B + F_v)$$

As velocity is constant then  $a = 0$

$$0 = \omega - (F_B + F_v)$$

$$\omega = F_B + F_v$$

$$\omega = mg + 6\pi\eta D v_t$$

Let  $\sigma$  and  $f$  be the density of body and fluid.

Then,

$$Mg - mg = 6\pi\eta D v_t$$

$$(M-m)g = 6\pi\eta D v_t$$

$$(\sigma - f)g = 6\pi\eta D v_t$$

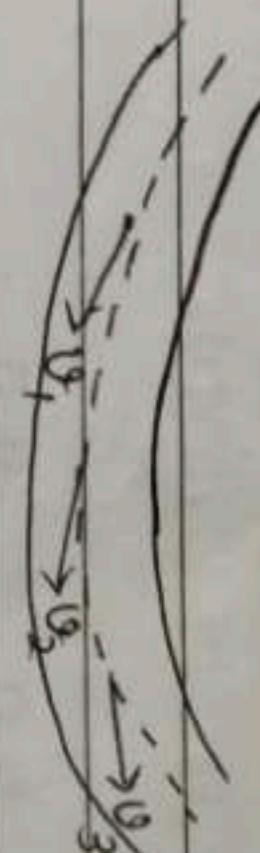
$$v_t = \frac{(\sigma - f)v g}{6\pi\eta D}$$

$$\text{as, } V = \frac{4}{3}\pi r^3$$

$$v_t = \frac{(\sigma - f) \cdot \frac{4}{3}\pi r^3 \cdot g}{6\pi\eta D}$$

#### \* Types of flow:

1) Streamline flow: The flow in which every particle of the liquid follows exactly the path and velocity of its preceding particle is called streamline flow.



2) Laminar flow: In this flow one layer slides over the other layer.

3) Turbulent flow: When velocity of liquid increases beyond a limit and becomes very high, the flow of fluid becomes irregular this irregular flow is called turbulent flow.

4) Critical Velocity: The critical velocity is that velocity of flow after which flow type is changed from streamline to turbulent.  $g_t$  is denoted by  $v_c$ .

$$v_c = \frac{N_r \cdot \eta}{f \cdot D}$$

where,  $N_r \rightarrow$  Reynold number.

$\eta \rightarrow$  coeff. of viscosity  
1  $\rightarrow$  density of fluid  
D  $\rightarrow$  Diameter of tube.

Proof:

$$F = K u^a \eta^b v^c$$

$$F = K u^a \eta^b v^c - \text{---(i)}$$

$$[M^a L^b T^c] = [L]^a [M^b L^c]^b [L T^{-1}]^c$$

$$[M^a L^b T^c] = [M^a L^{a-b+c} T^{-b+c}]$$

Equating power of  $M, L, T$

$$b = 1$$

$$-b + c = -2$$

$$b + c = 2$$

$$a - b + c = 1$$

$$a + c = 2$$

$$a - x + y = 1$$

Put these value in eqn (i)

$$F = K u^a \eta^b v^c$$

$$F = K u^a \eta^b v^c$$

where  $K = 6\pi$  [experimentally]

$$b = 1$$

$$a = 1$$

$$c = 1$$

$$a + c = 2$$

$$a - x + y = 1$$

\* Expression: Rate of flow of  $\propto (P)^\alpha u^b \eta^c$  liquid

$$V = K (P)^\alpha u^b \eta^c - *$$

$$[M^0 L^3 T^{-1}] = [M^b L^{-1} T^{-2}]^a [L]^b [M^b L^{-1} T^{-1}]^c$$

$$[M^0 L^3 T^{-1}] = [M^b L^{-2} T^{-2}]^a [L]^b [M^b L^{-1} T^{-1}]^c$$

Equating power of  $M, L, T$  from both side

$$a + c = 0 - \text{---(1)}$$

$$-2a + b - c = 3 - \text{---(2)}$$

$$-2a - c = -1 - \text{---(3)}$$

$$Add eqn ① and ② Put this value in eqn (i)$$

$$a + c = 0 Put this value in eqn (i)$$

$$-2a - c = -1 Put this value in eqn (i)$$

$$+ a = +1 Put this value in eqn (i)$$

$$a = +1 Put this value in eqn (i)$$

$$-2x_1 + b + (-1) = 3 Put this value in eqn (i)$$

$$b - 1 = 3 Put this value in eqn (i)$$

$$b = 4 Put this value in eqn (i)$$

$$a + c = 0 Put this value in eqn (i)$$

$$V = K (P)^\alpha u^4 \eta^1$$

$$= K \left( \frac{P}{\eta} \right) u^4 \eta^1$$

$$V = \frac{K P u^4}{\eta^2}$$

$$L' = L(1 + \alpha \cdot \Delta T)$$

$\alpha$

$$\alpha = \frac{\Delta L}{L \cdot \Delta T}$$

$$\Delta V = \gamma \cdot V \cdot \Delta T$$

Where ' $\gamma$ ' is called coeff. of cubic expansion

$$\gamma = \frac{\Delta V}{V \cdot \Delta T}$$

\* unit of  $\gamma$  is  $K^{-1}$  or  $^{\circ}C^{-1}$

$(V')$  be the final increase in vol after expansion

Then,

$$V' - V = V \cdot \gamma \cdot \Delta T$$

$$V' = V + \gamma V \cdot \Delta T$$

$$V' = V(1 + \gamma \cdot \Delta T)$$

From eqn (i) and (ii)

$$\Delta S = \beta S \cdot \Delta T$$

Where ' $\beta$ ' is coeff. of superficial expansion

$S'$  be the final increase in surface area

Then change in surface area ( $\Delta S$ ) =  $S' - S$

$$S' - S = \beta \cdot S \cdot \Delta T$$

\* States of Matter: The conversion of one state of matter into other is called change of state of matter.

There are three states of matter— Solid, Liquid and Gas.

The change of state from solid to liquid is called melting

The change of state from liquid to gas is called boiling

The change of state from solid to gas is called sublimation.

The change of state from solid to liquid is called fusion.

The change of state from gas to liquid is condensation.

- 3) cubical expansion: If increase in vol take place then the thermal expansion is called cubical expansion.

Let  $V$  be the vol of solid

$\Delta V$  be the small increase in vol at  $\Delta T$  temp.

$$\Delta V \propto V - 1$$

from i) and ii)

$$\frac{\Delta V}{V} = \frac{V'}{V} - 1$$

$$\Delta V = V \cdot \gamma \cdot \Delta T$$

2) Fahrenheit Scale: In this scale the lowest point is taken as  $32^{\circ}\text{F}$  and highest point is  $212^{\circ}\text{F}$ . The gap between  $32^{\circ}\text{F}$  and  $212^{\circ}\text{F}$  is divided into 180 divisions.

$212^{\circ}\text{F}$

$80^{\circ}\text{R}$

180 divisions

$32^{\circ}\text{F}$

$$\frac{C-0}{100} = \frac{F-32}{180} = \frac{K-273}{100} = \frac{R-0}{80}$$

\* conversion formula:

3) Kelvin Scale: In this scale the lowest point is taken as  $0^{\circ}\text{K}$  and highest point is  $373\text{K}$ . The gap is divided in 100 equal parts. This scale is also extended to zero Kelvin. The temp. of zero Kelvin is known as Absolute zero. Which is equal to  $-273^{\circ}\text{C}$ .

Thermal Expansion of Solids:

When solids are heated they expand. This expansion is called thermal expansion. There are 3 types of thermal expansion.

i) Linear Expansion - If increase in length take place then the expansion is called thermal expansion.

Let ' $L'$ ' be the length of rod.  $\Delta L$  be the small change in length of rod after increasing temp. by  $\Delta T$ .

$$\text{Then, } \Delta L \propto L \quad (1)$$

$$\Delta L \propto \Delta T \quad (2)$$

From eqn (1) and (2)

$$\Delta L \propto L \cdot \Delta T$$

$$\Delta L = \alpha \cdot L \cdot \Delta T$$

Where, ' $\alpha$ ' is called the coefficient of thermal expansion.

4) Reaumer Scale: In this scale the lowest point is taken at  $0^{\circ}\text{R}$  and highest point is  $80^{\circ}\text{R}$ . The gap is divided in equal 80 parts.

— ok

Let ' $L'$ ' be the final increase in length. Then change in length ( $\Delta L$ ) =  $L' - L$

$$L' - L = \alpha \cdot L \cdot \Delta T$$

$$L' = L + \alpha \cdot L \cdot \Delta T$$

\* Application of Bernoulli's thm:

Acc. to Bernoulli's thm

$$\frac{1}{2} \rho v^2 + \rho gh + \frac{P}{\rho} = \text{constant}$$

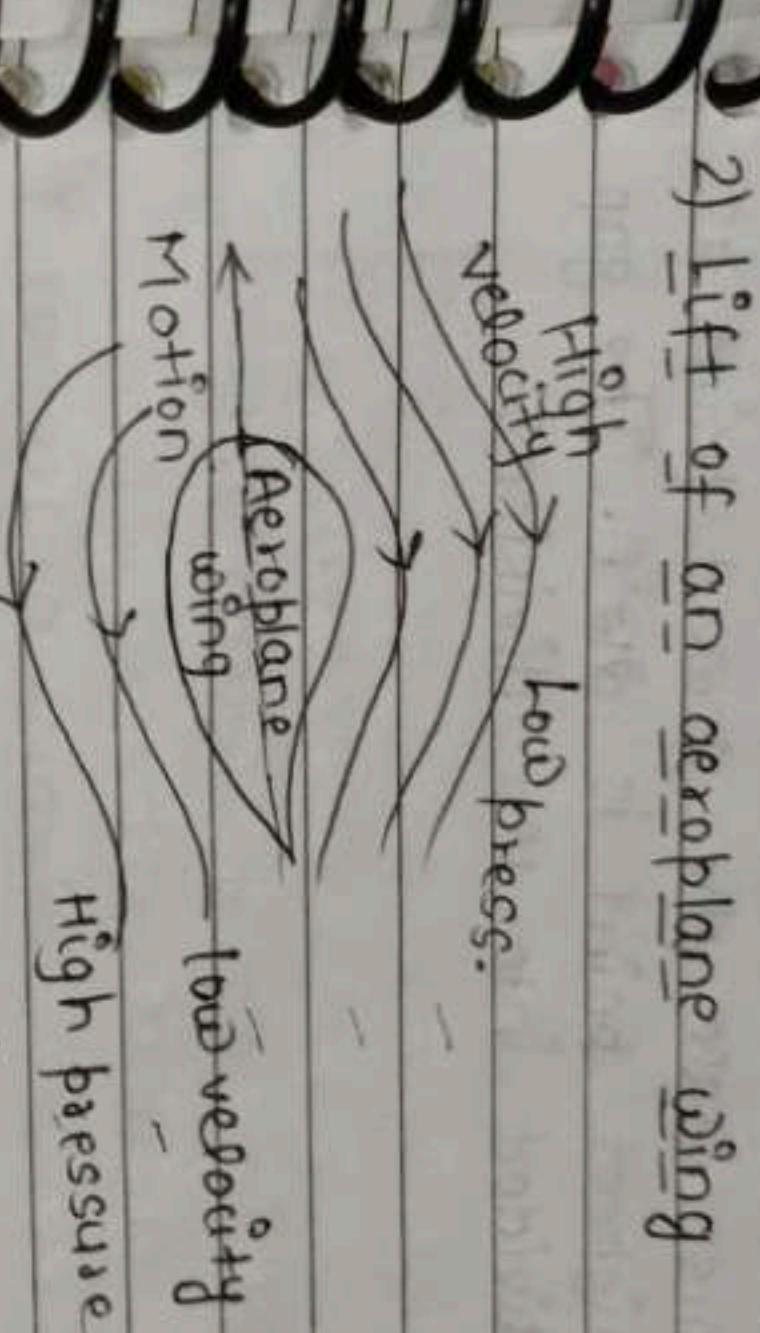
of two points lying at same horizontal dist. than, P.C becomes zero

$$\frac{1}{2} \rho v^2 + P = \text{constant}$$

Case I = If in same area velocity of fluid is low then press. created by fluid in that area will be high.

Case II = If in some areas velocity of fluid is high then, press. created by fluid in that area will be low.

1) Blowing off the roofs during storms



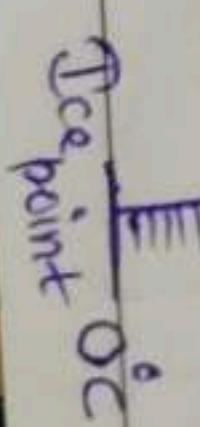
The upper surface of aeroplane wing is more curved than its lower surface. Due to this the wind speed is higher on upper surface as compare to lower surface. So, a high press. is created at lower surface while a low press. is created at upper surface. This pressure difference helps the aeroplane to lift up.

## CHAPTER = 4 THERMAL PROPERTIES

\* Temperature scale: Temp. scale is a way to represent temperature.

i) Celsius scale: This scale was designed by scientist Celsius. In Celsius scale, the lowest point is taken as zero degree Celsius. This is known as ice point or melting point of ice. The highest point is 100°C.

During storms, the wind blows with very high speed over the top side of the roof. This high speed will create a low pressure over the roof. Inside the hut the wind speed is very less as compare to outside. So, a high press. is created below the roof. Since roofs are blown off.



Press. Energy: The energy possessed by a fluid due to its press. is known as pressure energy.

Let  $\propto'$  be the disp. of fluid after applying some pressure on it.

Then, workdone = Pressure energy

$$W = \text{Force} \times \text{disp.}$$

$$W = F \times \propto'$$

$$\omega = \text{Press. Energy} = F \cdot \propto' - (i)$$

$$\text{as, Press.} = \frac{F}{A} = P$$

$$\text{But this value in eqn } A$$

$$W = \text{press. energy} = P \cdot A \cdot \propto'$$

$$\text{Pressure energy} = P \cdot V$$

Pressure energy per unit volm

$$\frac{P \cdot V}{V} = P$$

\* M.M.I @ Bernoulli's Theorem: This thrm deals with law of conservation of energy for flow of an ideal and

incompressible fluid such that for a streamline flow of an ideal fluid, the total energy remains constant at every point.

$$\text{Thus, } \frac{1}{2}mv^2 + mgh + Pv = \text{constant}$$

In terms of density  
 $\frac{1}{2}\rho v^2 + \rho gh + \rho V = \text{constant}$

\* Expression (Proof):

$$\frac{1}{2}\rho v^2 + \rho gh + \rho V = \text{constant}$$

$$\frac{B_{A_2}}{A_2} v_2$$

~~$$\frac{1}{2}\rho v^2 + \rho gh + \rho V = \text{constant}$$~~

Dividing by volm (V) :-

$$\frac{1}{2}m v^2 + mgh + Pv = \text{constant}$$

$$\frac{1}{2}\rho v^2 + \rho gh + \rho V = \text{constant}$$

~~$$\frac{1}{2}\rho v^2 + \rho gh + \rho V = \text{constant}$$~~

$$\frac{1}{2}m v^2 + mgh + Pv = \text{constant}$$

~~$$\frac{1}{2}\rho v^2 + \rho gh + \rho V = \text{constant}$$~~

$$\left[ \frac{1}{2}v^2 + gh + \frac{P}{\rho} = \text{constant} \right]$$

consider a bernoulli tube.

$$\text{As, } h_2 > h_1$$

$$\text{then, P.E at B} > \text{P.E at end A}$$

$$\text{change in P.E} = m_2gh_2 - m_1gh_1 - (1)$$

$$g > v_2 \text{ then,}$$

$$\text{K.E at end B} > \text{K.E at end A}$$

$$\text{change in K.E} = \frac{1}{2}m_2v_2^2 - \frac{1}{2}m_1v_1^2 - (2)$$

$$P_1 > P_2 \text{ then liquid flow from A end to B end.}$$

$$\text{Press. at end B} > \text{Press. Energy at end B}$$

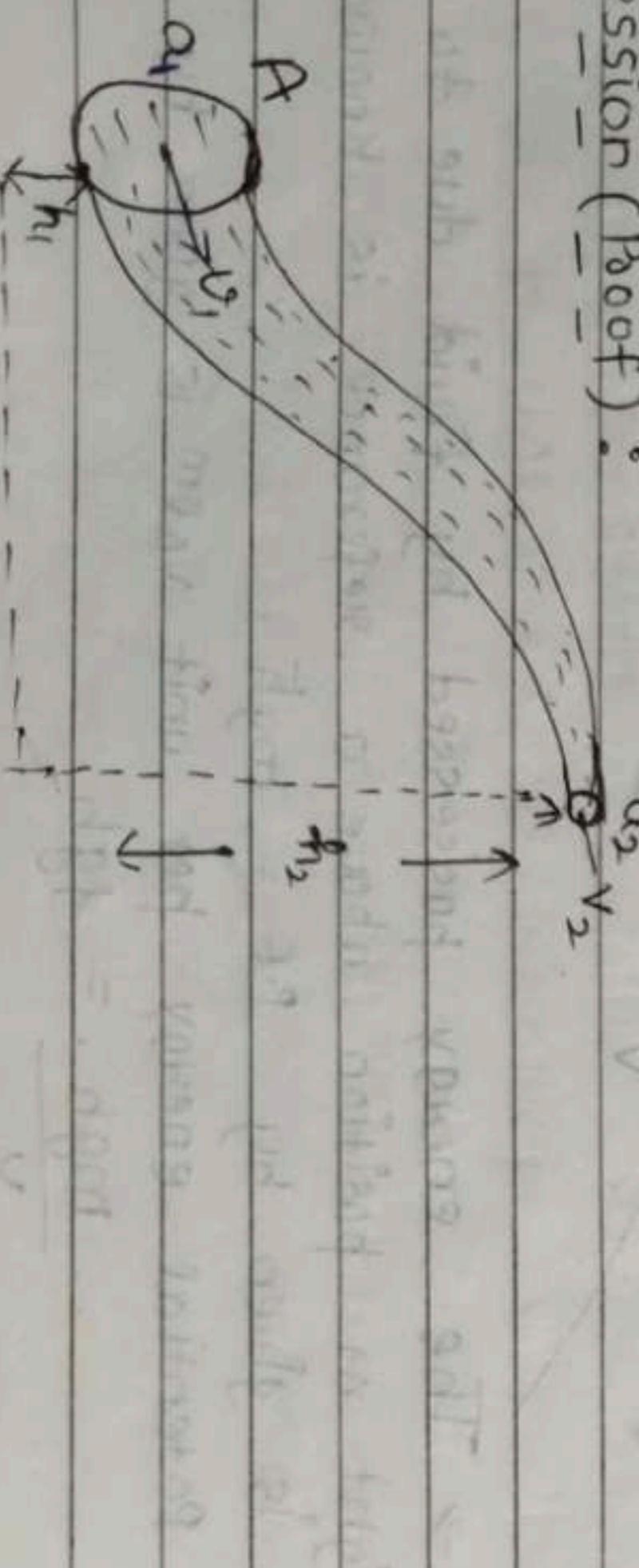
$$P_1V_1 > P_2V_2$$

$$\text{change in press. energy}$$

$$\text{For total energy}$$

$$\text{Adding eqn (1), (2) and (3)}$$

$$\frac{1}{2}m_2v_2^2 - \frac{1}{2}m_1v_1^2 + m_2gh_2 - m_1gh_1 + P_1V_1 - P_2V_2 = \text{const.}$$



\* Melting point: The temp. at which solid converts into liquid is called melting point. For ex - Melting point of ice is zero°C.

\* Boiling point: The temp. at which liq. converts into gas is called boiling point. For ex - The boiling point of water is 100°C.

\* Latent heat: The heat required to change state of matter of a unit mass body is called latent heat.

The heat required to change state of matter of a body depend on mass of body.

$$\Theta = m \cdot L$$

Where,  $\Theta$  = heat energy

$$m = \text{mass body}$$

$L$  = latent heat



$$\text{S.O.G} \text{ unit of latent heat } (L) = \frac{\text{J}}{\text{kg}} = \text{J kg}^{-1}$$

$$\text{D.F.} = \frac{M^1 L^2 T^{-2}}{M} = [M^0 L^2 T^{-2}]$$

\* Specific heat: The heat required to raise the temp.

of a unit mass body by 1°C. is called specific heat. The heat required to raise the temp. of a body depends on mass of body and temp. difference.

$$\Theta = M \cdot S \cdot \Delta T$$

Where  $\Theta$  = Heat energy

$S$  = Specific Heat

$\Delta T$  = temp. diff.

$$S = \Theta$$

$$m \cdot \Delta T$$

S.T unit of specific heat ( $\sigma$ ) =  $J$

$$\text{Kg} \cdot \text{K}$$

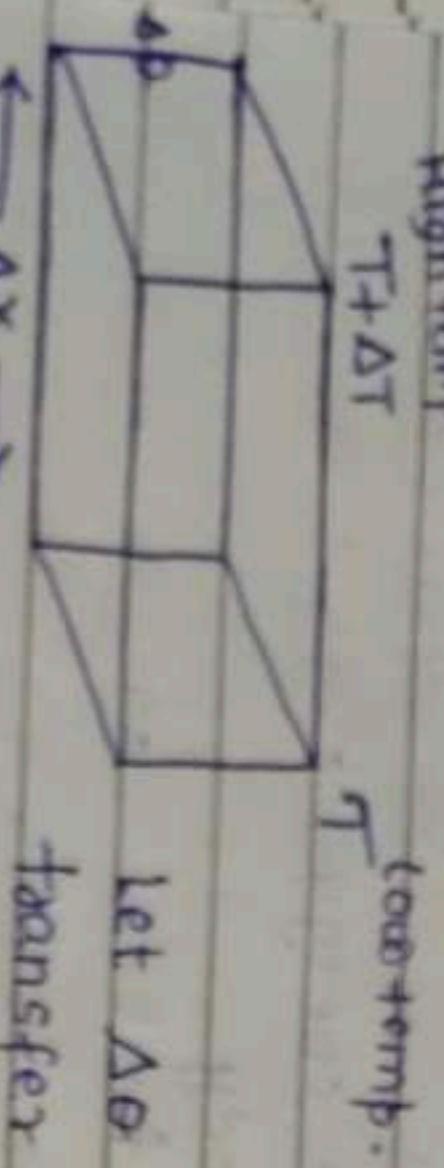
$$\text{D.F. formula} = [M^0 L^2 T^{-2} K^{-1}]$$

$$= \text{J kg}^{-1} \text{K}^{-1}$$

3) Radiation: The phenomena of transfer of heat directly from source to receiver without any movement of particles and without heating medium b/w them.

It is the fastest mode of heat transfer. Heat from Sun comes to us through radiation.

\* Thermal conductivity: It is the ability of solid to conduct heat through it.



Let  $\Delta Q$  be the amount of heat transfer from one face to another force in time  $\Delta t$ .  
Rate of flow of heat =  $\frac{\Delta Q}{\Delta t}$

Then,

$$\frac{\Delta Q}{\Delta t} \propto A \quad \text{(i)}$$

$$\frac{\Delta Q}{\Delta t} \propto T \quad \text{(ii)}$$

$$\frac{\Delta Q}{\Delta t} \propto \frac{1}{\Delta x} \quad \text{(iii)}$$

From (i), (ii) and (iii)

$$\frac{\Delta Q}{\Delta t} \propto A \cdot T \cdot \frac{1}{\Delta x}$$

$$\frac{\Delta Q}{\Delta t} = k \cdot A \cdot T \cdot \frac{1}{\Delta x}$$

Where 'k' is called coeff. of thermal conductivity.

$$k = \frac{\Delta Q}{\Delta t} \cdot \frac{\Delta x}{A \cdot T}$$

Where  $A$  = area of face

$$\Delta x = \text{dist. b/w two ends}$$

$T$  = temp.

$$\text{J.g unit of } k = \text{J.m}^{-1}\text{K}^{-1} = \text{W.m}^{-1}\text{K}^{-1}$$

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$$= 1 \text{ J.m}^{-1}\text{K}^{-1}$$

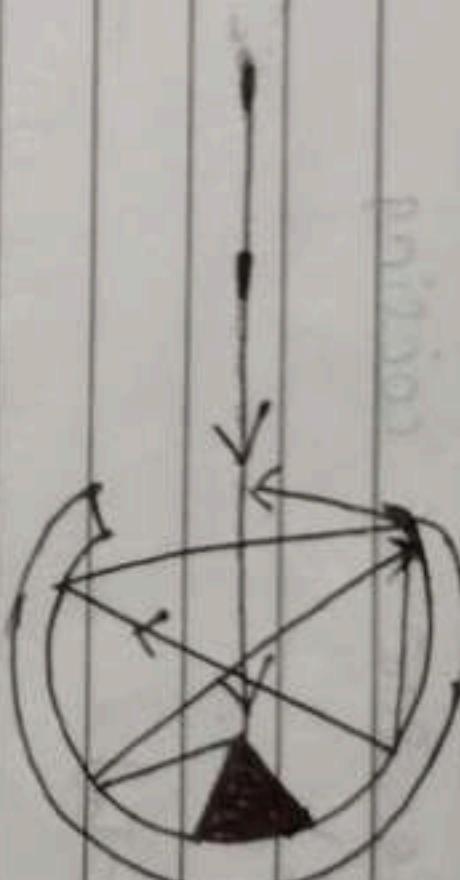
$$\text{D.P. of } K = M^1 L^2 T^{-2} \times T^{-1} \times L^{-1} \times K^{-1} \\ = K = [M^1 L^1 T^{-3} K^{-1}]$$

High temp.

$T^{\text{low temp.}}$

$T^{\text{low temp.}}$

\* Perfectly black body: A perfectly black body is that body which will absorb completely absorb radiation of all wavelength incident on it. The radiation emitted by a black body are known as Blackbody radiation or full radiations of total radiations. A perfectly black body cannot be designed in real practice.



\* Kirchhoff's law: Acc. to this law the ratio of emissive power to absorptive power for a particular wavelength of a given body at a given temp. is always constant such that  $\frac{e_h}{a_h} = \text{constant} = E_h$

$$\frac{\Delta Q}{\Delta t} = k \cdot A \cdot T \cdot \frac{1}{\Delta x}$$

\* Stefan's law: Acc. to this law the amount of heat energy radiated per second by unit area of a perfectly black body is directly prop. to 4<sup>th</sup> power of Absolute temp. of body. It is also called Stefan's bolzmann's law.

Mathematical expression:

$$E \propto T^4$$

$$F = \sigma T^4$$

$$\text{Where, } \sigma = \text{Stefen's constant} = 5.67 \times 10^{-8} \text{ Wm}^{-2}\text{K}^{-4}$$

\* Newton's law of cooling: Acc. to this law the rate of cooling of a body is directly prop. to temp. diff. of body and surrounding.

OR

The rate of loss of heat from a body is directly prop. to temperature diff. of body and surrounding.

Let ( $T$ ) be the temp. of body

$T_0$  = temp. of surrounding

$\frac{-dT}{dt}$  = rate of fall temp.

Then acc. to Newton's law of cooling

$$\frac{-dT}{dt} \propto (T - T_0)$$

$$\frac{-dT}{dt} = K(T - T_0)$$

Where  $K$  is proportionally constant.

~~Rock~~

~~Body~~

~~Surrounding~~

$$\frac{dT}{dt} = -K(T - T_0)$$

$$T = T_0 e^{-Kt}$$