Chapter 3

Optic Flow Based Autopilots: Speed Control and Obstacle Avoidance

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Abstract The explicit control schemes presented here explain how insects may navigate on the sole basis of optic flow (OF) cues without requiring any distance or speed measurements: how they take off and land, follow the terrain, avoid the lateral walls in a corridor, and control their forward speed automatically. The optic flow regulator, a feedback system controlling either the lift, the forward thrust, or the lateral thrust, is described. Three OF regulators account for various insect flight patterns observed over the ground and over still water, under calm and windy conditions, and in straight and tapered corridors. These control schemes were simulated experimentally and/or implemented onboard two types of aerial robots, a micro-helicopter (MH) and a hovercraft (HO), which behaved much like insects when placed in similar environments. These robots were equipped with optoelectronic OF sensors inspired by our electrophysiological findings on houseflies' motion-sensitive visual neurons. The simple, parsimonious control schemes described here require no conventional avionic devices such as rangefinders, groundspeed sensors, or GPS receivers. They are consistent with the neural repertory of flying insects and meet the low avionic payload requirements of autonomous micro-aerial and space vehicles.

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3.1 Introduction

When an insect is flying forward, an image of the ground texture and any lateral obstacles scrolls backward across the ommatidia of its compound eye. This flowing image set up by the animal's own forward motion is called the optic flow (OF). Recent studies have shown that freely flying insects use the OF to avoid collisions [94, 11, 88], follow a corridor [53, 83, 3, 4, 80], cruise, and land [93, 84, 78, 86]. The OF can be described as a vector field where each vector gives the direction and magnitude of the angular velocity at which any point in the environment moves relative to the eye [54]. Several authors have attempted to model the control systems at work in insects during free flight, focusing on specific aspects such as speed control [15, 84], distance or speed servoing [55, 17, 20], course control, and saccadic flight behavior (Egelhaaf et al., Chap. 4 of this volume) [39, 66]. Freely flying flies navigate by making pure translations alternating with fast, saccade-like turns [39, 94, 11, 92]. This idiosyncratic flight behavior was interpreted [94, 60, 11] as an active means of reducing the image flow to its translational components (that depends on the distances to objects [54]).

A biorobotic project was launched in the mid-1980s to investigate how a fly could possibly navigate and avoid collisions based on OF cues. The prototype Robot-Fly ("robot mouche") that we developed [60, 26] was a 50-cm high, fully autonomous *wheeled* robot carrying a compound eye driving 114 OF sensors with an omnidirectional azimuthal field of view (FOV). This reactive robot sensed the *translational OF* while moving straight ahead until detecting an obsta-

cle, which triggered a quick eye and body turn of a suitable amplitude (during which time vision was inhibited). Since the Robot-Fly traveled at a constant speed (50 cm/s), the OF measured during any translation easily gave the object *range*, and the robot was thus able to dodge and slalom to its target lamp through a random array of posts [26]. Despite the success of this early neuromimetic robot, *flying* insects would obviously have to use OF cues differently, since they are not in mechanical contact with the ground and cannot therefore easily estimate their groundspeed. How might flies, or micro-aerial vehicles (MAVs) cope with the severe disturbances (obstacles, wind, etc.) they encounter?

Here we summarize our attempts to model the visuomotor control systems that provide flying insects with a means of close-range autonomous piloting. First, we focus on ground avoidance in the vertical (longitudinal) plane (Sect. 3.3). We then discuss the ability to avoid corridor walls (Sect. 3.4), which has been closely analyzed in honeybees. Independent vertical and horizontal flight control systems (Egelhaaf et al., Chap. 4 of this volume) (see also [59]) were suggested by the performance of flies, which control their movements along the three orthogonal axes independently [94], while keeping their head transiently fixed in space [92, 96] via compensatory head roll and pitch mechanisms [40, 41]. Experimental simulations were performed and our control schemes were tested on two fly-by-sight aerial robots: a micro-helicopter (MH) (Fig. 3.5a) and a miniature hovercraft (HO) (Fig. 3.8a). These aerial robots use neuromorphic OF sensors [7, 25, 8] inspired by the elementary motion detectors (EMDs) previously studied at our laboratory in houseflies [67, 22, 27]. These sensors are briefly described in Sect. 3.2.

3.2 From the Fly EMDs to Electronic Optic Flow Sensors

Conventional cameras produce images at a given frame rate. Each image is scanned line by line at a high frequency. Many authors working in the field of computer vision have presented algorithms for analyzing the OF field based on scanned camera images. Although an OF algorithm has been implemented onboard a highly miniaturized, slow but fully autonomous indoor MAV (Zufferey et al., Chap. 6 of this volume), none of the OF algorithms to be found in the insect brain actually start with a retinal image scanned like a television image. Insects analyze the OF locally, pixel by pixel, via a neural circuit called an "elementary motion detector" (EMD). Further down the neural pathways, well-known collator neurons called "lobula plate tangential cells" (LPTCs) integrate the outputs of large numbers of EMDs and analyze the OF field generated by the animal's locomotion. Some of them transmit electrical signals via the neck to thoracic interneurons directly or indirectly responsible for driving the wing, leg muscles, or head muscles. Other LPTCs send relevant signals to the contralateral eye (see [87, 37, 9, 90]).

To determine the functional principles underlying an EMD, the responses of an LPTC neuron were recorded (H1, Fig. 3.1b) while two neighboring photoreceptors in a single ommatidium were being stimulated using a high-precision instrument (a hybrid between a microscope and a telescope: Fig. 3.1d), where the main objective lens was a single ocular facet (diameter \cong 25 μ m, focal length \cong 50 μ m) (Fig. 3.1a). By illuminating the two photoreceptors successively, a motion occurring in the visual field of the selected ommatidium was "simulated". The H1 neuron responded with a vigorous spike discharge to this "apparent motion," provided the motion was mimicked in the preferred direction (compare top and bottom traces in Fig. 3.1c) [67]. By applying various sequences of light steps and/or pulses to selected receptor pairs, an EMD block diagram was obtained and the dynamics and nonlinearity of each block were characterized [22, 27, 23].

In the mid-1980s, we designed a neuromorphic OF sensor inspired by the results of these electrophysiological studies [7, 25]. By definition, the OF is an angular speed ω corresponding to the inverse of the time Δt taken by a contrasting feature to travel between the visual axes of two adjacent photodiodes separated by an angle $\Delta \varphi$ (Fig. 3.2a). Our OF sensor's scheme is not a "correlator" [38, 65] but rather a "feature-matching scheme" [91], where a given feature (here, a passing edge) is extracted and tracked in time. Each photodiode signal is first band-pass filtered

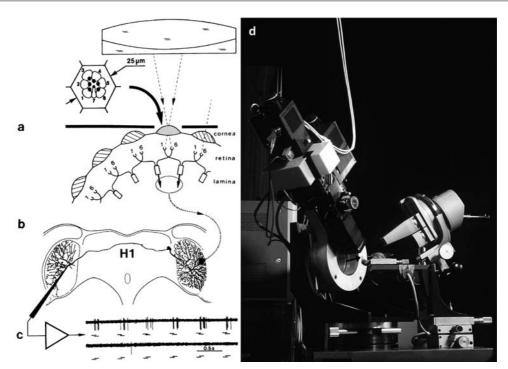


Fig. 3.1 (a)–(c) Experimental scheme used to analyze the principles underlying elementary motion detectors (EMDs) in flies, using single neuron recording and single photoreceptor stimulation. (d) The triple-beam incident light "microscope-telescope" used to deliver a sequence of 1 μm light spots to two neigh-

boring photoreceptors (the fly's head is indicated by the *arrow*). The microelectrode (c) recorded the electrical response (nerve impulses) of the motion-sensitive neuron H1 to this "apparent motion" (from [27])

(Fig. 3.2b), mimicking the analog signals emitted by the large monopolar neurons present in the fly lamina [97]. The next processing step consists in performing hysteresis thresholding and generating a unit pulse. In the EMD version built in 1989 for the Robot-Fly (Fig. 3.2d), the unit pulse from one channel sampled a long-lived decaying exponential function generated by the other channel, via a simple nonlinear circuit called a minimum detector (Fig. 3.2b), giving a monotonically increasing output V_{EMD} with the angular velocity $\omega = \Delta \varphi/\Delta t$ (Fig. 3.2b) [8]. The thresholding makes the EMD respond whatever the texture and contrast encountered, contrary to what occurs with "correlator" EMDs [65, 19] (see also Egelhaaf et al., Chap. 4 of this volume).

A very similar EMD principle was developed, independently, a decade later by Koch's group at CAL-TECH and termed the "facilitate and sample" velocity sensor [47]. These authors patented an aVLSI chip based on this principle [77], another variant of which

was recently presented (Moeckel & Liu, Chap. 8 of this volume).

Our OF sensor actually comprises two parallel EMDs, each of which responds to either positive or negative contrast transitions, as in the fly EMD (cf. Figs. 15 and 16 in [27]). The circuit responds equally efficiently to natural scenes [61]. Our current OF sensors are still based on our original "travel time" principle [7, 25, 8] but for the sake of miniaturization, the signals are processed using a mixed analog + digital approach [75] and the time Δt is converted into ω via a lookup table (Fig. 3.2c). Although they are much larger than any aVLSI (or fly's) EMDs, our current OF sensors (Fig. 3.2e,f) are small and light enough to be mounted on MAVs. Several OF sensors of this type can also be integrated into a miniature FPGA [1, 2]. A different kind of OF sensor was recently designed and mounted on a model aircraft [6, 35]. Optical mouse sensors have also been used as OF sensors (Dahmen et al., Chap. 9 of this volume) [36].

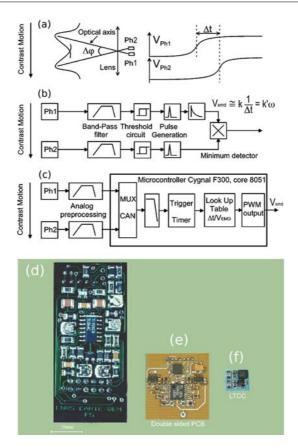


Fig. 3.2 Principle (**a, b**) of the elementary motion detector (EMD) inspired by our electrophysiological findings (cf. Fig. 3.1) (after [7, 25, 8]) and (**d**) completely analog version (mass 5 g) made in 1989 with *small mounted device* (SMD) technology

(from [8]). (**c**, **e**) Hybrid version (mass 0.8 g) based on a microcontroller (from [75]). (**f**) Similar hybrid version (size 7 mm × 7 mm, mass 0.2 g) built using low-temperature co-fired ceramics technology (LTCC) (from [64])

3.3 An Explicit Control Scheme for Ground Avoidance

3.3.1 Avoiding the Ground by Sensing the Ventral Optic Flow

The ventral OF perceived in the vertical plane by airborne creatures (including aircraft pilots) is the *angular velocity* ω generated by a point on the underlying flight track [33, 95]. Based on the definition of the angular velocity (Fig. 3.3A), the ventral OF ω is the ratio between groundspeed V_x and groundheight h as follows:

$$\omega = V_x/h[\text{rad.s}^{-1}] \tag{3.1}$$

Flies and bees are able to measure the *translational* OF, ω , irrespective of the spatial texture and contrast encountered [15, 83, 3], and some of their visual neurons respond monotonically to ω [45]. Neurons facing downward can therefore act as ventral OF sensors and directly assess the groundspeed-to-groundheight ratio V_x/h (Fig. 3.3).

Before Gibson introduced the OF concept [32], Kennedy established that an insect sees and reacts to the OF presented to its *ventral* viewfield [49] (see also [13]). This flowing visual contact with the ground is now known to be essential for insects to be able to orient upwind and migrate toward an attractive source of odor [49, 48] or pheromones [51]. Based on field experiments on locusts, Kennedy developed the "optomotor theory" of flight, according to which locusts have a "preferred retinal velocity" with respect to the

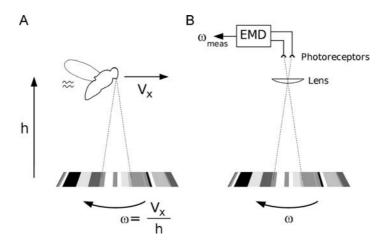


Fig. 3.3 (A) The ventral OF is an angular speed ω [rad s⁻¹] corresponding to the groundspeed-to-groundheight ratio. (B) The OF sensor comprises a microlens and two photoreceptors separated by a small angle $\Delta \varphi$ (see Fig. 3.2a) driving an EMD. The

ground below [49, 50]. In response to wind, for example, they adjust their groundspeed (or groundheight) who restore the velocity of the ground feature images. Kennedy's theory has been repeatedly confirmed during the last 30 years. Flies and bees seem to maintain a

constant OF with respect to the ground while cruising

or landing [13, 62, 84, 86, 4].

The problem is *how* they achieve this remarkable feat, since maintaining a given OF is a kind of chicken-and-egg problem, as illustrated by Eq. (3.1): an insect may hold its perceived OF, ω , constant by controlling V_x (if it knows h) or by controlling h (if it knows V_x). Moreover, it could maintain an OF of say 1 rad/s (i.e., 57°/s) by flying at a speed of 1 m/s at a height of 1 m or by flying at a speed of 2 m/s at a height of 2 m: there exists an infinitely large number of possible combinations of groundspeeds and groundheights generating the same "preferred OF". Kennedy's "theory" therefore lacked an explicit control scheme elucidating:

- 1. the flight variables really involved
- 2. the sensors really required
- 3. the dynamics of the various system components
- 4. the causal and dynamic links between the sensor(s) and the variable(s) to be controlled
- 5. the points of application and the effects of the various disturbances that insects may experience
- the variables insects control to compensate for these disturbances

latter delivers a signal $\omega_{meas} \cong \cong \Delta \varphi / \Delta t = V_x / h$, which serves as a feedback signal in the *OF regulator* (Fig. 3.4A). The one-dimensional random texture is a magnified sample of that shown in Figs. 3.5b and 3.6A (from [28])

Our first attempt to develop a control scheme [55] was not very successful, as we were cornered by the chicken-and-egg OF problem mentioned above and by the assumption prevailing in those days that insect navigation involves *estimating distance* [53, 60, 83, 26, 82]. In the experimental simulations described in 1994, for example [55], we assumed that flying insects (and robots) are able to measure their groundspeed V_x (by whatever means), so that by measuring ω they would then be able to assess the distance h from the ground (Eq. (3.1)) and react accordingly to avoid it. Although this stratagem – which is in line with the Robot-Fly's working principles (Sect. 3.1) – may be acceptable for aerial vehicles able to gauge their own groundspeed [5, 31], it does not tell us *how* insects function.

In 1999, we established (via experimental simulations) how a seeing helicopter (or an insect) might manage to follow a terrain and land on the sole basis of OF cues without measuring its groundspeed or groundheight (see Figs. 4 and 5 in [57]). The landing maneuvers were performed under the permanent feedback control of an OF-sensing eye, and the driving force responsible for the loss of altitude was the decrease in the horizontal flight speed which occurred when the rotorcraft (or the insect) was about to land, either voluntarily or because of an unfavorable headwind. The landing trajectory obtained in these simulations [57] resembled the final approach of bees landing on a flat surface [84]. The 840-g rotorcraft we constructed was able to jump over 1-m-high obstacles (see Fig. 8 in [58]).

3.3.2 The "Optic Flow Regulator"

More recently we developed a genuine "OF-based autopilot" called OCTAVE (which stands for optical altitude control system for aerial vehicles) that enables a micro-helicopter to perform challenging tasks such as takeoff, terrain following, reacting suitably to wind, and landing [70-74]. The idea was to integrate an OF sensor into a feedback loop driving the robot's lift so as to compensate for any deviations of the OF sensor's output from a given set point. This is what we call the OF regulator for ground avoidance. The term "regulator" is used here as in control theory to denote a feedback control system striving to maintain a variable constantly equal to a given set point. The variable regulated is often a temperature, a speed, or a distance, but here it is the variable ω [rad/s] corresponding to the V_x : h ratio, which can be sensed directly by an OF sensor. The OF sensor produces a signal ω_{meas} (Fig. 3.3b) that is compared with an OF set point ω_{Set} (Fig. 3.4A). The error signal $\varepsilon = \omega_{meas} - \omega_{Set}$ drives a controller adjusting the lift L, and hence the groundheight h, so as to minimize ε . All the operator does is to set the pitch angle Θ and hence the forward thrust, and hence the airspeed (see Fig. 3.4A): the *OF regulator* does the rest, keeping the OF, i.e., the V_x :h ratio, constant. In the steady state (i.e., at $t = \infty$), $\omega_{meas} \cong \omega_{Set}$ and the groundheight h becomes proportional to the groundspeed V_r :

$$h = K \cdot V_x \text{ (with } K = 1/\omega_{\text{Set}} = constant)$$
 (3.2)

3.3.3 Micro-Helicopter (MH) with a Downward-Looking Optic Flow Sensing Eye

To test the robustness of the OF regulator, we implemented it on a micro-helicopter (MH) equipped with a ventral OF sensor [70]. The robot (Fig. 3.5a) is tethered to the arm of a flight mill driven in terms of its elevation and azimuth by the MH's lift and forward thrust, respectively (Fig. 3.5b). Any increase in the rotor speed causes the MH to lift and rise, and the slightest (operator mediated) forward tilting induces the MH to gain speed. The flight mill is equipped with ground-truth azimuthal and elevation sensors with which the position and speed of the MH can be measured with great accuracy in real time. The MH is equipped with a minimalistic two-pixel ventral eye driving a single EMD (Fig. 3.2e). The latter senses the OF produced by the underlying arena covered by a richly textured, randomly distributed pattern in terms of both the spatial frequency and the contrast **m** (0.04<**m**<0.3).

Whenever the operator commands the robot to pitch forward or backward, the eye is actively counterrotated by a micro-servo that keeps the gaze oriented vertically downward. This process mimics the pitch stabilization mechanism at work in the eye in flying flies, ensuring suitable alignment of the eyes in space [41]. It eliminates the adverse effects of the *rotational* OF and the resulting need for a "derotation" procedure (Srinivasan et al., Chap. 2 of this volume; Zufferey et al., Chap. 6 this volume).

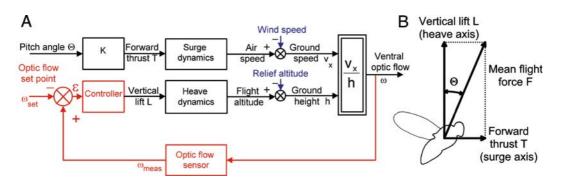
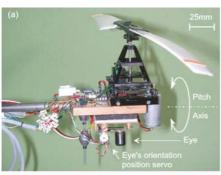


Fig. 3.4 (A) The OCTAVE autopilot consists of a feedback control system, called the *optic flow regulator* (bottom part), that controls the vertical lift, and hence the groundheight, so as to maintain the ventral OF, ω , constant and equal to the set point ω_{set} whatever the groundspeed. (B) Like flies [13] and bees [21],

our micro-helicopter (MH) gains speed by pitching its mean flight force vector forward at an angle Θ with respect to the vertical. Controlling F (via the rotor rpm) amounts to mainly controlling L because Θ remains small ($\Theta_{max} < 10^{\circ}$ for $V_x max = 3$ m/s) (from [28])



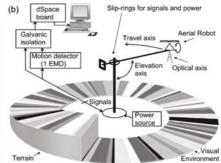


Fig. 3.5 (a) 100-g micro-helicopter (MH) equipped with a ventral OF sensor (Fig. 3.2e) and an OF regulator (Fig. 3.4a). The 5-g rotor is driven by a DC motor via a reducer. (b) The MH, which is mounted on the light pantographic arm of a flight mill, can be remotely pitched forward at an angle Θ while keeping

its roll attitude. The MH circles counterclockwise over a large arena (outside diameter: 4.5 m). In the experiments presented here (Fig. 3.6), the arena was flat, without any rising slopes (from [71])

3.3.4 Insects' Versus the Seeing Helicopter's Behavioral Patterns

OCTAVE's OF regulator scheme (Fig. 3.4A) results in the behavioral patterns shown in Fig. 3.6, which give the MH flight variables monitored during a 70-m flight over a flat terrain [28]. In Fig. 3.6A (left), the operator simply pitched the MH forward rampwise by an angle $\Delta\Theta = +10^{\circ}$ (between arrowheads 1 and 2). The ensuing increase in groundspeed V_x (see B) automatically made the MH take off, since the feedback loop consistently increased h proportionally to V_x to comply with Eq. (3.2). Once it had reached 3 m/s, the MH flew level at a groundheight h of approximately 1 m – the value imposed by the OF set point ω_{Set} (Fig. 3.6C). After covering 42 m, the MH was pitched backward rampwise by an opposite angle $\Delta\Theta = -10^{\circ}$ (between arrowheads 3 and 4 in Fig. 3.6A), and the ensuing deceleration (see 3.6B) automatically triggered a gradual descent until landing occurred. As the landing gear kept the robot's eye 0.3 m above ground (dotted horizontal line), touchdown occurred shortly before the groundspeed V_x had reached zero, and the MH ended its journey with a short ground run.

The MH flight pattern shows that an airborne vehicle can take off, navigate, and even land on flat terrain without having to measure the groundheight or groundspeed, provided it is equipped with an OF sensor facing the ground and an *OF regulator*.

The *OF regulator* concept and the robot's performances – which are extremely reproducible [73] – were found to account for a series of puzzling,

seemingly unconnected flying abilities observed by many authors during the last 70 years in various species (fruitflies, honeybees, moths, mosquitoes, dung beetles, migrating locusts, butterflies, and birds), as discussed in [28]. The most striking parallels with insect behavior focus on the following:

- Automatic terrain following: A gradual increase in relief constitutes a "disturbance" that impinges on the system at a particular point (see Fig. 3.4A). The closed feedback loop overcomes this perturbation by increasing the flight altitude, resulting in a constant groundheight over the rising terrain [70]. This may account for the terrain and canopy following abilities of migrating insects, as described in [28].
- Suitable reactions to headwind: Windspeed is a disturbance that impinges on the system at a different point (see Fig. 3.4A). The feedback loop overcomes this perturbation by forcing the robot to descend by headwind (and even to land smoothly by strong headwind; see Fig. 13 in [73]). A similar reaction was observed in locusts, honeybees, and dung beetles, as reported in [28].
- *Flight over a no-contrast zone*: Here the OF sensor fails to respond, which irremediably causes the robot to crash, just as honeybees crash into mirror-smooth water, as observed in 1963 [42].
- *Landing on a flat surface*: During the *final approach*, which starts when the MH has regained its completely upright position (arrowhead 4 in Fig. 3.6A), the *OF regulator* forces the MH to

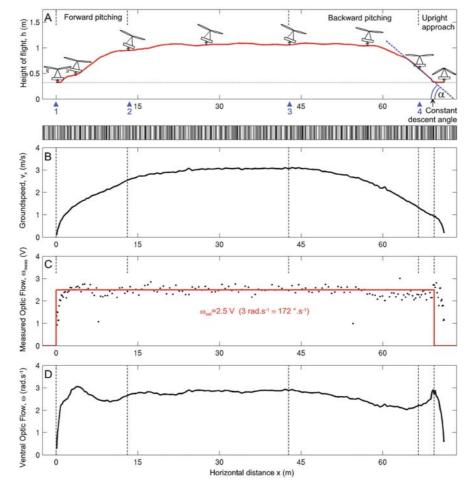


Fig. 3.6 Actual flight path (**A**) and flight parameters monitored during a 70-m flight (consisting of about six laps over the test arena: Fig. 3.5b) performed by the micro-helicopter (MH) equipped with the *OF regulator* (Fig. 3.4A). The complete flight

path (A) over the randomly textured pattern includes takeoff, level flight, and landing. (B) Groundspeed V_x . (C) Output ω_{meas} of the OF sensor. (D) Actual OF output ω (calculated from V_x/h) generated by the OF regulator (from [28])

land smoothly at a *constant descent angle*, α [rad] (Fig. 3.6A), which depends on only two parameters Eq. (3.3):

$$\alpha = -\arctan\left(\frac{1}{\omega_{set}\tau}\right) \tag{3.3}$$

where ω_{Set} [rad s⁻¹] is the OF set point and τ [s] is the surge time constant of the MH ($\tau_{MH} = 2.15$ s) [28].

Honeybees also land with a constant slope on flat surfaces [86]. The way bees may achieve this feat has been explained quite differently, however. Landing bees would follow two rules: "(1) adjusting the speed of forward flight to hold constant the angular velocity of the image of the surface as seen by the eye and (2) making the speed of descent proportional to

the forward speed" [86]. In contrast, our *OF regulator* automatically generates smooth landing with a constant slope by acting neither upon the *forward speed* nor upon the *descent speed* (details in [28]).

In short, the MH's outstanding visuomotor performance and its close resemblance to insects' behavioral patterns show *how* insects and MAVs may take off, follow terrain, and even land smoothly without having to measure the groundheight, groundspeed, or descent speed, provided they are equipped with OF sensors facing the ground and an *OF regulator* that servoes the OF to a reference value.

This model differs markedly from another one where the OF controls the *groundspeed* V_x rather than the *groundheight* h [63, 84, 86, 3]. It can be seen in

Fig. 3.4A that if the ventral OF controlled V_x instead of h, this would generate strikingly different flight patterns, as follows:

- (i) Instead of following a slanting terrain, as migrating butterflies and our MH do, insects would gradually decelerate until touching the rising ground at a negligible speed, thus inopportunely interrupting their journey.
- (ii) Instead of descending in a headwind and rising in a tailwind, as honeybees [68], locusts [50], and our MH do, insects would compensate for an unfavorable headwind by increasing their airspeed without changing their groundheight.

These two models can be reconciled, however, if we add the hypothesis that another *OF regulator* based on OF sensors oriented toward the *lateral* parts of the eyes could be used to control the groundspeed. In the next section, we give the example of flight in a corridor to show *how* the *lateral OF* might control the insect's behavior by determining both the groundspeed and the clearance from the walls.

3.4 An Explicit Control Scheme for Speed Control and Lateral Obstacle Avoidance

Based on behavioral experiments on honeybees conducted at Srinivasan's laboratory and at our own laboratory, we designed the LORA III autopilot (LORA III stands for Lateral Optic flow Regulator Autopilot, Mark III), which is able to control both the forward speed V_x of an aerial vehicle and its lateral distances D_R and D_L from two corridor walls [79].

3.4.1 Effects of Lateral OF on Wall Clearance and Forward Speed

Honeybees flying through a narrow corridor tend to fly along the midline [53, 83]. To explain this "centering behavior," the authors hypothesized that bees might balance the speeds of the retinal images (i.e., the lateral OFs) of the two walls. This was confirmed by experiments where one wall was set in motion: bees flying

in the same direction as the moving wall tended to fly closer to it, whereas bees flying in the opposite direction tended to fly farther away from it. These experiments showed compellingly that the visuomotor control mechanism at work in flying bees depends on the laterally perceived *translational OF* [83].

We recently reported that honeybees trained to fly along a corridor toward a nectar source did not systematically center on the corridor midline, but could hug one wall (Fig. 3.7b,c) [80]. Bees kept on hugging one wall even when part of the opposite wall was missing (Fig. 3.7d). Interestingly, forward speed V_x and distances from the walls, D_R and D_L , were on average such that the speed-to-distance ratio (i.e., the lateral OF, ω) was maintained virtually constant (at about 230°/s in our 95-cm-wide corridor) [80]. These novel findings prompted us to examine how bees may adopt either "centering behavior" (Fig. 3.7a) or "wallfollowing" behavior (Fig. 3.7b-d) on the basis of OF sensing, a prowess that seems to require jointly controlling the forward speed and the clearance from the walls.

Experiments on tethered [34, 10] and free-flying flies [14–16] and bees [84, 3] and tethered locusts [81] have long shown that motion detected in the *lateral* part of these insects' compound eyes affects the forward thrust, and hence the forward speed. When flying through a *tapered corridor* lined with regular black and white vertical stripes, honeybees slowed down as they approached the narrowest section and speeded up when the corridor widened beyond this point [84]. Bees therefore tended to adjust their speed proportionally to the local corridor width. The authors concluded that "the speed of the flight is controlled by regulating the image velocity" [84].

The honeybee's "centering behavior" was successfully mimicked by numerous terrestrial robots [12, 18, 76, 85, 44, 43]. The speed of one of the early robots was also controlled on the basis of the *sum of the lateral OFs* perceived on both sides [76].

3.4.2 New Robotic Demonstrator Based on a Hovercraft (HO)

With a view to explaining honeybees' behavior in a corridor, a miniature hovercraft (HO) was used [79]. With an aerial vehicle of this kind endowed with

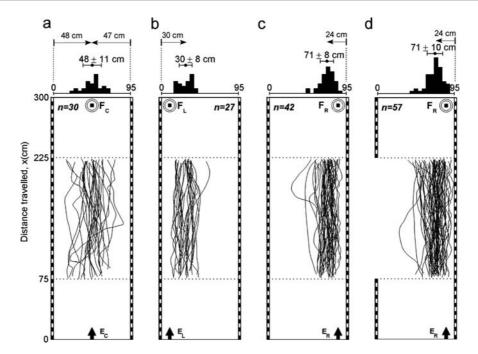


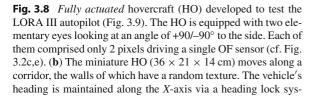
Fig. 3.7 Honeybees' centering and wall-following behavior. Bees were trained to enter an apparatus where sugar solution was provided at the end of a *wide* (width 0.95 m) 3-m-long corridor formed by two 0.25-m-high walls. A digital camera placed above the insect netting covering the corridor filmed the trajectory of individual flying bees over the central part of the corridor. The walls were lined with regularly spaced vertical *white-and-gray*

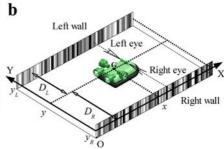
stripes (period 0.1 m; contrast m = 0.27). The bee's entrance E and the feeder F were placed either on the corridor midline (a: E_c and F_c) or on one side (b: E_L and F_L ; c and d: E_R and F_R). In (d), part of the left wall was removed during the trials. The mean ordinate of the trajectories was distributed as shown above (n = 1 number of trajectories recorded under each of the experimental conditions) (from [80])

natural roll and pitch stabilization abilities, planar flight control systems can be developed conveniently. Like flies [94, 92] and sandwasps [96], our modified HO can produce forward and sideward slips independently, because the two lateral thrusters *LT1* and *LT2*

that we have added (Fig. 3.8) make it *fully actuated*. It travels at a constant altitude (~2 mm) and senses the environment with two laterally oriented eyes, which measure the right and left OFs. Each eye contains only two photoreceptors, each driving an EMD of the







tem compensating for any yaw disturbances by activating the two rear thrusters differentially. The two lateral eyes are therefore oriented perpendicular to the corridor axis. The OF experienced by each eye is proportional to the groundspeed V_x and inversely proportional to the distance $(D_R \text{ or } D_L)$ from the wall the eye is looking at (Eqs. (3.4) and (3.5)) (from [79])

type shown in Fig. 3.2e. The HO's heading is maintained along the *X*-axis of the corridor (Fig. 3.8b) by a heading lock system (based on a micro-magnetic compass enhanced by a rate-gyro), which compensates for any yaw disturbances by controlling the two rear thrusters *differentially*. This system mimicks the honeybee's heading lock system, which is based on a polarized light compass [30, 69] and gives the insect an impressively straight course even in the presence of wind [68].

The 820-g HO travels at a groundspeed vector \tilde{V} over a flat surface along a corridor, the walls of which are wallpapered with vertical stripes with a random pattern of spatial frequency and contrast, mimicking a richly textured visual environment (Fig. 3.8b). Since any yaw rotations are compensated for, each eye experiences a purely translational OF, ω_R or ω_L , which can be simply defined as follows:

$$\omega_R = V_X / D_R \tag{3.4}$$

$$\omega_L = V_X / D_L \tag{3.5}$$

where V_x is the hovercraft's groundspeed; D_R and D_L are the distances from the right and the left walls, respectively.

3.4.3 The LORA III Autopilot: A Dual OF Regulator

The hovercraft is controlled by the bio-inspired autopilot called LORA III (Fig. 3.9) [79]). This control scheme has been called the dual OF regulator, because it consists of two strongly interdependent feedback loops, the side control loop and the forward control *loop*, each of which uses the data collected by the OF sensors looking at both sides. Each loop controls the thrust along its own translational degree of freedom (i.e., in the surge or sway direction) and has its own OF set point. The tight coupling existing between the two loops is illustrated, for instance by the fact that the feedforward gain in the side control loop is proportional to the forward speed V_x , which is determined in turn by the *forward control* loop (see Fig. 3.9). The hovercraft reacts to any changes in the lateral OFs by selectively adjusting the two orthogonal components V_x and V_y of its groundspeed vector \vec{V} according to the principles described in the two following subsections:

3.4.3.1 Side Control System

The principle underlying the side control system (Fig. 3.9, bottom loop) is based on findings made on the honeybee's centering and wall-following behaviors described in Sect. 3.4.1. The side control system is a lateral OF regulator akin to the ventral OF regulator for ground avoidance (OCTAVE: see Sect. 3.3). It takes the OF generated by the two corridor walls into account and strives to keep the larger of the two OFs measured (left or right) constantly equal to a sideways OF set point $\omega_{SetSide}$. The hovercraft then reacts to any OF deviations from $\omega_{SetSide}$ by controlling the *lateral* thrust, and hence the side slip speed V_{ν} , and hence the distance from the left or right wall. A maximum criterion selects the larger of the two OFs measured, ω_{Rmeas} or ω_{Lmeas} , and the sign of their difference determines which wall will be followed. The larger OF is compared with the sideways OF set point $\omega_{SetSide}$. The error signal ε_{Side} transmitted to the side controller (Fig. 3.9) is therefore calculated as follows:

$$\varepsilon_{Side} = \operatorname{sign}(\omega_{Lmeas} - \omega_{Rmeas})$$

$$\times (\omega_{SetSide} - \max(\omega_{Lmeas}, \omega_{Rmeas}))$$
(3.6)

In the steady state, the OF selected will become equal to $\omega_{SetSide}$ due to the natural integrator present in this loop. The side control loop is nonlinear since the OF is an inverse function of the variable D_L or D_R controlled. The transfer function of the "side dynamics" (Fig. 3.9) relating the hovercraft's side speed V_y to the control signal (u_{LTI} - u_{LT2}) delivered by the side controller approximates a first-order low-pass filter with an identified time constant of 0.5 s [79]. A proportional-derivative (PD) controller was introduced into the side control system to increase the damping, thus improving the stability and enhancing the response dynamics. Details of the tuning procedures are given in [79].

3.4.3.2 Forward Control System

The forward control system (Fig. 3.9, upper loop) is also based on the OF perceived on *both* sides, as reported in Sect. 3.4.1 for a robot [76] and for the honeybee [84]. This control system strives to hold the *sum* of the left and right OFs measured ($\omega_{Lmeas} + \omega_{Rmeas}$) constant and equal to a *forward OF set point* ω_{SetFwd} by controlling the forward thrust, and hence

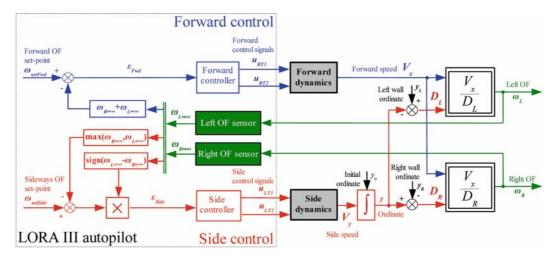


Fig. 3.9 The LORA III autopilot is a *dual OF regulator* consisting of two *interdependent* visual feedback loops: the *forward control loop* and the *side control loop*, each of which controls one degree of freedom (x or y). The forward controller adjusts the forward thrust, and hence the forward speed V_x , so as to minimize the error signal ε_{Fwd} . The side controller adjusts the lateral thrust, and hence the side speed, and hence the ordinate y, so

as to minimize the error signal ε_{Side} . The sign of the difference between the left and right OFs measured determines which wall will be followed. The robot's initial ordinate y_0 and the right and left wall ordinates (y_R, y_L) are regarded by LORA III as disturbances impinging on the system at particular points (black arrows) (from [79])

the forward speed V_x . This control scheme automatically generates a "safe" forward speed, i.e., one commensurate with the local corridor width. More specifically, the feedback signal corresponding to the sum of the two OFs measured is compared with the *forward OF set point* ω_{SetFwd} (Fig. 3.9). In the steady state, the sum of the OF becomes equal to ω_{SetFwd} .

The transfer function of the "forward dynamics" relating the hovercraft's forward speed V_x to the forward control signal $u_{RT1} + u_{RT2}$ (Fig. 3.9) is also given by a first-order low-pass filter with a time constant of 0.5 s [79]. A proportional-integral (PI) controller was introduced here into the feedback loop to improve the closed-loop dynamics and give a zero steady-state error. Further details of the tuning procedures used are given in [79].

3.4.4 "Operating Point" of the Dual OF Regulator

The scheme in Fig. 3.9 requires neither the speed nor the distance from the walls to be measured. The question therefore arises as to whether the system will converge and give a single steady-state "operating point" in terms of a specific forward speed and a specific dis-

tance from the walls. In the following discussion, the LORA III autopilot is assumed to have reached the steady state $(t = \infty)$, that is, (i) the *sum* of the two lateral OFs measured $(\omega_{Rmeas} + \omega_{Lmeas})$ has reached the *forward OF set point* ω_{SetFwd} and (ii) the *larger* of the two lateral OFs measured (here it is taken to be the right one, ω_{Rmeas}) has reached the *sideways OF set point* $\omega_{SetSide}$ (see above). We can therefore write as follows:

$$\begin{cases} \omega_{SetFwd} = \omega_{R\infty} + \omega_{L\infty} \\ \omega_{SetSide} = \omega_{R\infty} \end{cases} \Leftrightarrow \begin{cases} \omega_{SetFwd} = \frac{V_{x\infty}}{D_{R\infty}} + \frac{V_{x\infty}}{D_{L\infty}} \\ \omega_{SetSide} = \frac{V_{x\infty}}{D_{R\infty}} \end{cases}$$

$$(3.7, 3.8)$$

The forward speed $V_{x\infty}$ and the ordinate $y_{\infty} = D_R$ that the hovercraft will reach in the steady state can be said to define an "operating point" $(V_{x\infty}, y_{\infty})$ that can be calculated straightforwardly from Eqs. (3.7) and (3.8) as follows:

$$\begin{cases} V_{x\infty} = \frac{\omega_{\text{SetSide}} \cdot (\omega_{\text{SetFwd}} - \omega_{\text{SetSide}})}{\omega_{\text{SetFwd}}} \cdot D \\ y_{\infty} = D_{R\infty} = \frac{\omega_{\text{SetFwd}} - \omega_{\text{SetSide}}}{\omega_{\text{SetFwd}}} \cdot D, y_{\infty} \in \left[0, \frac{D}{2}\right] \\ D = D_{R\infty} + D_{L\infty}, \end{cases}$$

$$(3.9, 3.10, 3.11)$$

where *D* is the corridor width.

Equations (3.9) and (3.10) have five important consequences:

- Two single parameters, ω_{SetFwd} and $\omega_{SetSide}$, suffice to define the speed and lateral position $(V_{x\infty}, y_{\infty})$ that the hovercraft (or the bee) will adopt in a corridor of a given width D: the flying agent needs to measure neither its speed nor its clearance from the walls nor the corridor width.
- Both the speed $V_{x\infty}$ and the ordinate y_{∞} adopted in the steady state can be seen to be *proportional* to the corridor width D (Eqs. (3.9) and (3.10)). If the corridor is twice as large, the agent will automatically fly twice as fast. This mechanism therefore accounts for the view that "the bee controls its flight speed by monitoring and regulating the angular velocity of the environment's image as represented on the eye, that is, if the tunnel width is doubled, the bee flies twice as fast" [84]. In cases where the agent follows one wall (as in Figs. 3.7b–d), doubling the corridor width will not only double the forward speed but also the clearance from the wall followed.
- If the sideways OF set point is *larger* than half the value of the *forward OF set point* (i.e., ω_{SetSide} > ω_{SetFwd}/2), the agent will reach a final ordinate y_∞ (Eq. (3.10)) that is smaller than half the value of the corridor width. This means that the flying agent will hug one wall, thus generating wall-following behavior.
- At all sideways OF set point values that are smaller than half the forward OF set point (i.e., ω_{SetSide} < ω_{SetFwd}/2), the agent will not reach the final ordinate y_∞ predicted by Eq. (3.10) without causing a change in sign in the error signal ε_{Side} (see Eq. (3.6)). The agent will therefore be forced to maintain the OF value at ω_{SetFwd}/2, i.e., it will consistently navigate along the midline of the corridor, thus showing centering behavior. Oscillations in the trajectory are likely to occur, however, because of the repeatedly changing sign of the error signal ε_{Side} (Eq. (3.6)).

3.4.5 Simulation Results: Flight Paths Along Straight or Tapered Corridors

The implementation of the complete LORA III visuomotor autopilot on the robotic hovercraft is under-

way, and only computer-simulated experiments will be presented here [79]. The computer simulations include the dynamic hovercraft model on both the surge and sway axes, the controllers with their PD or PI compensation, actuator saturations, full optical transfer function of the lens/photoreceptor system of each eye, detailed interactions between each photoreceptor and the random patterns lining the two walls, and complete processing within each OF sensor, according to the EMD principle described in Fig. 3.2c. These simulations were carried out on a standard PC equipped with the MatlabTM/Simulink toolbox at a sampling frequency of 1 kHz. This sampling frequency is the same as the sampling rate of the microcontroller installed onboard the LORA robot (Fig. 3.8a).

3.4.5.1 Wall-Following Behavior Along a Straight Corridor

Figures 3.10, 3.11, and 3.12 show some examples of the computer-simulated behavior of the hovercraft equipped with the LORA III autopilot (Fig. 3.9) [79]. The first visual environment simulated was a straight corridor (3 m long, 1 m wide). The right and left walls were lined with a random pattern of various gray vertical stripes covering a 1-decade contrast range (from 4% to 38%) and a 1.1-decade spatial frequency range (from 0.068 c/° to 0.87 c/° reading from the corridor midline). From the three simulated trajectories shown in Fig. 3.10a, the hovercraft can be seen to navigate without ever colliding with the walls. It reached the same steady speed regardless of its initial ordinate y_0 at the entrance to the corridor (Fig. 3.10b). All three trajectories correspond to the case where the *sideways OF* set point was larger than half the forward OF set point $(\omega_{SetSide} = 230^{\circ}/\text{s} > 150^{\circ}/\text{s} = \omega_{SetFwd}/2)$. Under these conditions, the HO adopted wall-following behavior, as predicted in Sect. 3.4.4, and followed the right or left wall, depending on the initial ordinate y_0 . The HO consistently generated a steady-state clearance of 0.25 m from either wall (left wall: squares and crosses; right wall: full dots) and a "safe" forward speed of $V_{x\infty} = 1$ m/s. Steady-state clearance and speed define a similar "operating point" to that calculated from Eqs. (3.9) and (3.10): taking $\omega_{SetFwd} = 300^{\circ}/\text{s}$, $\omega_{SetSide} = 230^{\circ}/\text{s}$, and a corridor width D = 1 m give an operating point $V_{x\infty} = 0.94$ m/s and $y_{\infty} = 0.23$ m.

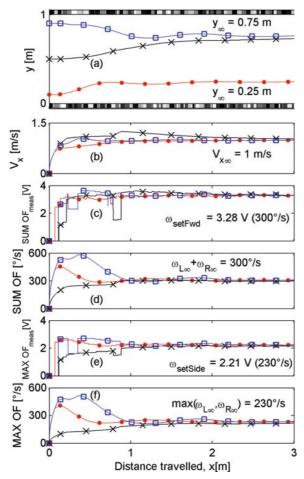


Fig. 3.10 (a) Wall-following behavior of the hovercraft (HO) (time marks on the flight paths are at 0.3-s intervals). The HO moves to the right at a forward OF set point $\omega_{SetFwd} = 300^{\circ}/s$ (3.28 V) and a sideways OF set point $\omega_{SetSide} = 230^{\circ}/s$ (2.21 V), starting at various ordinates y_0 (*squares*: y_0 =0.90 m, *crosses*: y_0 =0.50 m, *full dots*: y_0 = 0.10 m). (b) Irrespective of its initial ordinate, the HO ends up following one wall with a clearance

of 0.25 m, at a forward speed $V_{x\infty} = 1$ m/s. (**c**, **d**) The sum of the lateral *OFs measured* and the sum of the *actual OFs* both eventually equal the forward OF set point $\omega_{SetFwd} = 300^{\circ}$ /s. (**e**, **f**) The larger value of the OFs *measured* and that of the *actual* OFs both eventually equal the sideways OF set point $\omega_{SetFwd} = 230^{\circ}$ /s (from [79])

Whether the hovercraft followed the right or the left wall depended on the sign of the error signal ε_{Side} (Eq. (3.6)). The hovercraft's initial ordinate y_0 was treated like a *disturbance* (see Fig. 3.9), which was rejected by the *dual OF regulator*. It can be seen from Fig. 3.10d,f that both the *sum* and the *larger value* of the lateral OFs eventually reacts the OF set points of 300°/s and 230°/s, respectively.

3.4.5.2 "Centering Behavior": A Particular Case of "Wall-Following Behavior"

Figure 3.11 illustrates the opposite case, where the OFs generated on either side (150°/s: Fig. 3.11d) never

reached the *sideways OF set point*, $\omega_{SetSide}$ (which was set at 90°/s or 110°/s or 130°/s). This is because these required values of $\omega_{SetSide}$ were all *smaller* than half the value of the *forward OF set point* (i.e., $\omega_{SetSide} < \omega_{SetFwd}/2 = 150^\circ$ /s). These low values of $\omega_{SetSide}$ relative to ω_{SetFwd} forced the hovercraft to *center* between the two walls, as predicted at the end of Sect. 3.4.4. In addition, the robot's ordinate can be seen to oscillate about the midline, due to the ever-changing sign of the error signal ε_{Side} (Eq. (3.6)) – as also predicted at the end of Sect. 3.4.4. The LORA III dual OF regulator may therefore even account for the striking oscillatory flight pattern observed in honeybees when they fly centered along a corridor (see Fig. 2a in [53]).

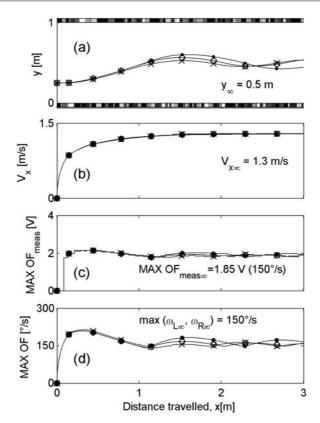


Fig. 3.11 "Centering behavior" as a particular case of "wall-following behavior". (**a, b**) Simulated trajectories of the hovercraft (HO) moving to the right along a straight corridor at a forward OF set point $\omega_{SetFwd} = 300^{\circ}/\text{s}$, with various sideways OF set points (*crosses*: $\omega_{SetSide} = 130^{\circ}/\text{s}$, *open dots*: $\omega_{SetSide} = 110^{\circ}/\text{s}$, *full dots*: $\omega_{SetSide} = 90^{\circ}/\text{s}$). Initial condition $y_0 = 0.25$ m, time marks as in Fig. 3.10. The HO can be seen to consis-

tently end up *centering* between the two walls at a forward speed of 1.3 m/s. (**c**, **d**) The larger of the two OFs *measured* and the larger *actual* OF both eventually equal 150°/s ($=\omega_{SetFwd}/2$). In attempting to reach either of the three sideways OF set points, the LORA III autopilot triggers changes in the sign of the error signal ε_{Side} (Eq. (3.6)), causing oscillations about the midline (see (**a**)) (from [79])

The error signal ε_{Side} is consistently minimum along the midline, but the OF cannot become smaller than 150°/s (i.e., $\omega_{SetFwd}/2$). The hovercraft can be seen to have reached the steady-state forward speed of $V_{x\infty}$ = 1.3 m/s (Fig. 3.11b) in all three cases because all three values of $\omega_{SetSide}$ were below 150°/s. In all three cases, the steady-state operating point of the hovercraft was similar to that predicted in Sect. 3.4.4: at $\omega_{SetFwd} = 300^{\circ}$ /s and $\omega_{SetSide} < 150^{\circ}$ /s, a 1-m-wide corridor gives $V_{x\infty} = 1.31$ m/s and $y_{\infty} = D_{R\infty} = D_{L\infty}$ = 0.5 m (Eqs. (3.9) and (3.10)). The lateral OFs measured on either side reached 150°/s (= $\omega_{SetFwd}/2$) in all three cases (Fig. 3.11c), and their sum was therefore equal to 300°/s ($=\omega_{SetFwd}$). Similar centering behavior occurred at all values of $\omega_{SetSide}$ such that $\omega_{SetSide}$ $\leq \omega_{SetFwd}/2$. Centering behavior can therefore be said to be a particular case of wall-following behavior.

3.4.5.3 Flight Pattern Along a Tapered Corridor

In the last set of computer-simulated experiments presented here, the environment was a 6-m-long *tapered* corridor with a 1.24-m-wide entrance and a 0.5-m-wide constriction located midway (Fig. 3.12). The right and left walls were lined with a random pattern of gray vertical stripes covering a 1-decade contrast range (from 4% to 38%) and a 1.5-decade spatial frequency range (from 0.034 c/° to 1.08 c/° reading from the corridor midline).

As shown in Fig. 3.12a and b, whatever the position of its initial ordinate y_{θ} , the HO automatically slowed down on approaching the narrowest section of the corridor and accelerated when the corridor widened beyond this point. In Fig. 3.12a, the HO adopted wallfollowing behavior because $\omega_{SetSide} > \omega_{SetFwd}/2 =$

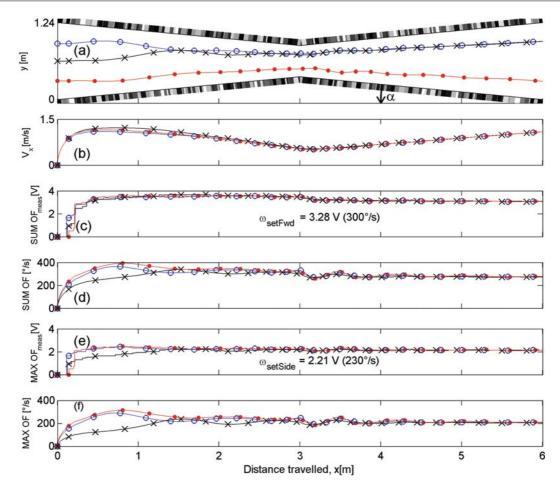


Fig. 3.12 Automatic navigation along a tapered corridor, requiring no data on the corridor width or the tapering angle α (marks on trajectories as in Fig. 3.10). Again, the hovercraft's behavior is entirely determined by its two OF set points: $\omega_{SetFwd} = 300^{\circ}$ /s and $\omega_{SetSide} = 230^{\circ}$ /s. (a) Simulated trajectories of the HO moving to the right along the corridor (tapering angle $\alpha = 7^{\circ}$) with three initial ordinates y_0 (open dots: $y_0 = 0.90$ m, crosses: $y_0 = 0.60$ m, full dots: $y_0 = 0.30$ m). (b) The for-

ward speed decreases and increases linearly with the local corridor width and the distance x traveled. (\mathbf{c} , \mathbf{d}) The sum of the two lateral OFs measured (and that of the actual OFs computed with Eq. (3.4) and (3.5)) is maintained constant and is equal to $300^{\circ}/s$ (= ω_{SetFwd}). (\mathbf{e} , \mathbf{f}) The side control system effectively keeps whichever lateral OF is larger at a constant value of approximately $230^{\circ}/s$ (= $\omega_{SetSide}$) (from [79])

150°/s (see Eq. (3.10)). It can be seen from Fig. 3.12c that the forward control system succeeded in keeping the sum of the two lateral OFs measured constant and equal to the *forward OF set point* $\omega_{SetFwd} = 300^{\circ}$ /s. Likewise, the side control system succeeded in keeping the *larger* of the two lateral OFs measured virtually constant and equal to the *sideways OF set point* $\omega_{SetSide} = 230^{\circ}$ /s (Fig. 3.12e). Not only the initial ordinate y_{θ} but also the (gradually changing) ordinates y_{R} and y_{L} are regarded by the LORA III autopilot as *output perturbations*, which are rejected by the *dual OF regulator* (see the points of application of these disturbances in Fig. 3.9).

The ensuing forward speed profile along the tapered corridor is particularly instructive (Fig. 3.12b): the HO's forward speed V_x tends at all times to be proportional to the distance traveled x, as observed with the flight path of bees flying freely along a tapered corridor [84]. This plot of $V_x = dx/dt$ versus x actually defines a *phase plane*, in which the *linear* change in speed observed with the distance traveled means that the speed $V_x(t)$ is bound to vary as an exponential function of time [79]:

$$V_{x}(t) = V_{x}(t_{0}) \cdot e^{sign(\alpha) \cdot (t - t_{0})/\tau(\alpha)}$$
(3.12)

where the time constant $\tau(\alpha)$ is a monotonic function of the tapering angle α , as follows:

$$\tau(\alpha) = \frac{\omega_{SetFwd}}{2 \cdot tan \, |\alpha| \cdot \omega_{SetSide} \cdot (\omega_{SetFwd} - \omega_{SetSide})}$$
(3.13)

Thus, without having any knowledge of the corridor width D or the tapering angle α , the HO (or the bee) is therefore bound to slow down exponentially with a time constant $\tau(\alpha)$ when entering the narrowing section ($\alpha < 0$) of the corridor and to speed up exponentially with the same time constant $\tau(\alpha)$ after leaving the constriction ($\alpha > 0$), in accord with the observed honeybee's flight path in a tapered tunnel [84].

The behavioral effects of two other types of disturbances affecting the LORA III autopilot were also studied [79]. With a large gap in a wall (as in the bee experiments shown in Fig. 3.7d) the HO was not flummoxed and kept on following the remaining wall [79]. When a wall pattern was moved at a constant speed V_P in the direction of travel (as in the original bee experiments [53]), the robot moved closer to the moving wall, and vice versa [79]. The reason is that the robot's relative speed with respect to the moving wall became V_x – V_P instead of V_x in Eq. (3.8), causing a predictable shift in the robot's operating point, $V_{x\infty}$, y_{∞} as computed from Eqs. (3.9–3.10).

3.5 Conclusion

3.5.1 Is There a Pilot Onboard an Insect?

In this chapter, we have recounted our attempts to specify the types of operations that insects may perform to guide their flight on the sole basis of optic flow (OF) cues. The OCTAVE principle differs markedly from another current OF-based navigation strategy, where OF sensing needs to be completed by groundspeed sensing (based on a GPS, for example) to estimate the groundheight (see Eq. (3.1)) so as to follow terrain and land safely [5, 31]. The OCTAVE and LORA III autopilots harness the power of the *translational OF* more parsimoniously because they do not need to *measure or estimate* any distances or groundspeeds and therefore do not require any sensors other than OF sensors. The purpose of these autopilots is not to *reg*-

ulate any distances or groundspeeds. The only variable they need to regulate (i.e., maintain constant) is the OF – a variable that represents a speed-to-distance ratio and that can be directly measured by a dedicated sensor called an OF sensor. OCTAVE and LORA III autopilots include three interdependent OF regulators in all, which control the lift, lateral thrust, and forward thrust, on which the groundheight, lateral positioning, and groundspeed, respectively, depend. The block diagrams (Figs. 3.4 and 3.9) show which variables need to be measured, which ones are controlled, and which ones are regulated, as well as the point of application of the various disturbances. They also give the causal and dynamic relationships between these variables. These three feedback control loops may enable an agent having no mechanical contact with the ground to automatically attain a given groundspeed, a given groundheight and a given clearance from the walls in a simple environment such as a corridor, without any need for speed or range sensors giving explicit speed or distance data. In a tapered corridor, for example, the hovercraft (HO) automatically tunes both its clearance from the walls and its groundspeed to the local corridor width (Fig. 3.12), although it is completely "unaware" of the exact corridor width, the forward speed, and the clearance from the walls. The behavior depends wholly on three constants which are the three OF set points $\omega_{SetVentr}$, $\omega_{SetSide}$, and ω_{SetFwd} . Experimental simulations and physical demonstrators showed that difficult operations such as automatic takeoff, ground avoidance, terrain following, centering, wallfollowing, suitable reaction to headwind, groundspeed control, and landing can be successfully performed on the basis of these three *OF regulators*.

These simple control schemes actually account for many surprising findings published during the last 70 years on insects' visually guided performances (details in [28]), including honeybees' habit of landing at a constant slope [86] and their flight pattern along a tapered corridor [84]. Our novel finding that bees do not center systematically in a corridor but tend to follow a wall, even when the opposite wall has been removed (Fig. 3.7b–d), cannot be accounted for by the *optic flow balance hypothesis* [53, 83] and is convincingly accounted for by the LORA III model, where "centering behavior" [53, 83] turned out to be a particular case of "wall-following behavior" (Sect. 3.4.5.2). LORA III (Fig. 3.9) would give the bee a safe speed and safe clearance from the walls, whereas OCTAVE

(Fig. 3.4) would give the bee safe clearance from the ground – a clearance commensurate with its forward speed, whatever the speed. These *explicit control* schemes can therefore be viewed as working hypotheses. Onboard insects, the three OF set points may depend on either innate, internal, or external parameters. Recent findings have shown that the forward speed of bees flying along a corridor depends not only on the lateral OF but also partly on the ventral OF [4], which suggests that the ventral and lateral OFs should not perhaps be handled as separately as with OCTAVE and LORA III. Recent experimental simulations have shown how an agent might fly through a tunnel by relying on the OF generated by all four sides: the lateral walls, ground, and roof [61]. Indoor experiments on the autonomously flying microplane MC2 showed that when the plane makes a banked turn to avoid a wall, the ventral OF may originate partly from the *lateral* wall and the lateral OF partly from the ground (Zufferey et al., Chap. 6 of this volume). This particularity may not concern insects, however, since they compensate for the banking and pitching of their thorax [40, 41] by actively stabilizing their "visual platforms" [92, 96]. A high-speed, one-axis oculomotor compensatory mechanism of this kind was recently implemented onboard a 100-g aerial robot [52].

The electronic implementation of an OF regulator is not very demanding (nor is its neural implementation), since it requires only a few linear operations (such as adding, subtracting, and applying various filters) and nonlinear operations (such as minimum and maximum detections). OF sensors are the crux of OF regulators. Our neuromorphic OF sensors deliver an output that grows monotonically with the OF ω , regardless of the spatial frequency and contrast encountered (see Sect. 3.2), much like honeybees' velocity-tuned (VT) neurons [45]. Most importantly, the OF regulator scheme greatly reduces the dynamic range constraints imposed on OF sensors, since it tunes the animal's behavior all the time so that the OF will deviate little from the OF set point [28]. In Fig. 3.6C, for example, the MH holds ω_{meas} virtually constant throughout its journey, despite the large groundspeed variations that occur during takeoff and landing, for instance (see Fig. 3.6B). This and other examples (see Figs. 3.10c,e, 3.11c, and 3.12c,e) show that the 1-decade dynamic range (from 40°/s to 400°/s [75]) of our OF sensors is up to the tasks.

3.5.2 Potential Aeronautics and Aerospace Applications

The control schemes presented here rely on the OF to carry out reputedly difficult tasks such as taking off, terrain following, landing, avoiding lateral walls, and groundspeed control. These simple control schemes are restricted so far to cases where the OF is sensed either ventrally or laterally, perpendicular to the heading direction. The field of view (FOV) of the eyes and the provocatively small number of pixels (2 pixels per eye) and EMDs (one EMD per eye) obviously need to be increased when dealing with navigation in more sparsely textured environments. Increasing the number of motion sensors was found to improve the goal-directed navigation performances of an airship in noisy environments [46], and we recently described how an additional forward-looking EMD might enable our MH to climb steeper rises by providing the autopilot OCTAVE with an anticipatory feedforward signal [74]. It will also be necessary to enlarge the FOV and control the heading direction (about the yaw axis) to enable the HO to successfully negotiate more challenging corridors including L-junctions and T-junctions. The more frontally oriented visual modules required for this purpose could be based on measuring the OF divergence [56–58], a procedure that flies seem to use when they land and trigger body saccades [93, 78, 89].

In the field of aeronautics, these systems could serve to improve navigation aids and automatic maneuvers. Steady measurement of the *ventral* OF could prevent deadly crashes by warning pilots that the current altitude is "too low for the current groundspeed" - without requiring any altitude and speed measurements [29]. An OCTAVE OF regulator implemented onboard an aircraft would enable it to gradually take off "under automatic visual control", to veto any attempt to descend to a groundheight not commensurate with the current groundspeed, and to make it land safely [72, 29]. These systems could also potentially be harnessed to guiding MAVs indoors or through complex terrains such as mountains and urban canyons. Since these control systems are parsimonious and do not rely on GPS or bulky and power-hungry emissive sensors such as FLIRs, RADARs, or LADARs, they meet the strict constraints imposed on the bird and insect scale in terms of their size, mass, and consumption.

For the same reasons, these autopilots could potentially be adapted to micro-space vehicles (MSVs) performing rendezvous and docking missions in space or exploration missions on other celestial bodies. A Martian lander equipped with a more elaborate OCTAVE autopilot could perform smooth automatic landing (see Sect. 3.3 and [29, 72]). A flying reconnaissance rover equipped with more elaborate OCTAVE and LORA III autopilots could take off autonomously and explore an area, skimming the ground and hugging the walls of a canyon, and adapting its groundspeed and clearance from the walls automatically to the width of the canyon. The orbiter (or base station) would simply have to send the rover a set of three *low-bandwidth* signals: the values of the three OF set points [24].

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