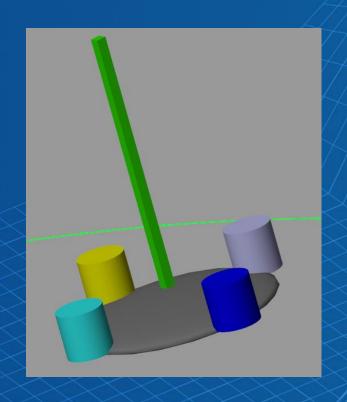
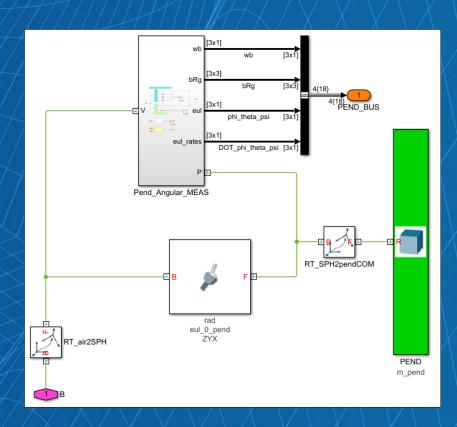


A quadcopter balancing pendulum:

- How to model and control it

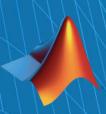




$$\frac{d}{dt} \frac{\partial L}{\partial \dot{q}_k} - \frac{\partial L}{\partial q_k} = Q_k$$

$$Q_k = \sum_{i=1}^{Nf_{nc}} \left(\overrightarrow{F}_i \cdot \frac{\partial \overrightarrow{v}_i}{\partial \dot{q}_k} \right) + \sum_{j=1}^{N\tau_{nc}} \left(\overrightarrow{\tau}_j \cdot \frac{\partial \overrightarrow{\omega}_j}{\partial \dot{q}_k} \right)$$

Brad Horton
Engineer
MathWorks





How do you nurture the **CONFIDENCE**

of students?

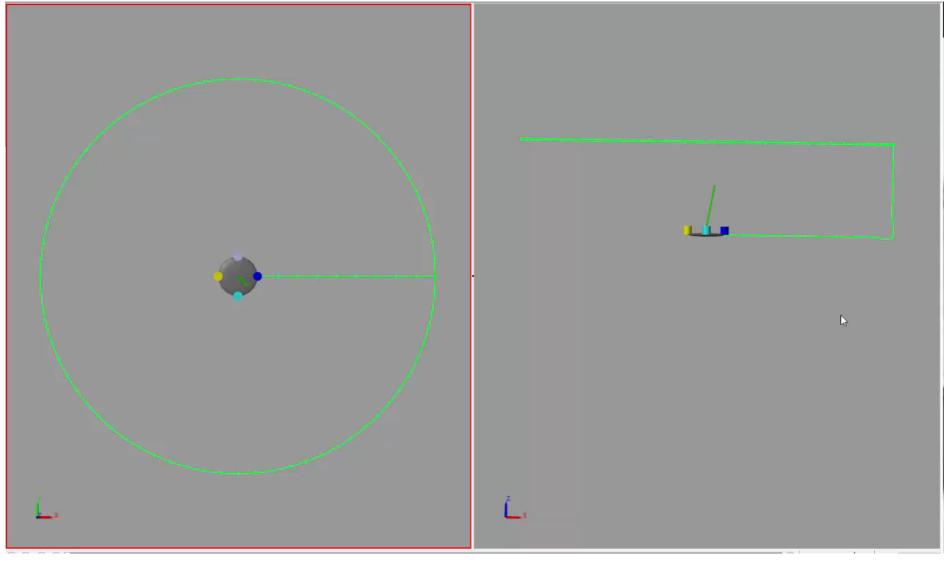
$$\begin{bmatrix} u \\ v \\ z \end{bmatrix}_L = \begin{bmatrix} \cos(\phi) & \sin(\phi) & 0 \\ -\sin(\phi) & \cos(\phi) & 0 \\ 0 & 0 & 1 \end{bmatrix} \cdot \begin{bmatrix} u \\ v \\ z \end{bmatrix}_G$$

 $\omega \times r$

$${}^{B}F = m.({}^{B}\dot{v}_{C} + {}^{B}_{G}\omega_{B} \times {}^{B}_{G}v_{C})$$

$${}^{B}M = {}^{B}I.{}^{B}_{-}\dot{\omega}_{B} + {}^{B}_{G}\omega_{B} \times ({}^{B}I.{}^{B}_{G}\omega_{B})$$

$$\frac{d}{dt}\frac{\partial L}{\partial \dot{q}_k} - \frac{\partial L}{\partial q_k} = Q_k$$





Todays agenda:

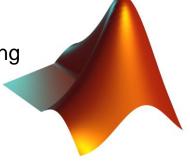
Phase 1

- One of the challenges in Learning Rigid Body Dynamics.
- Computational Thinking Is this the answer ?

Phase 2

Applying Computational Thinking

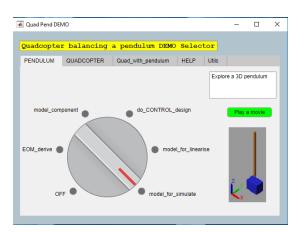
4 Case Studies



Phase 3

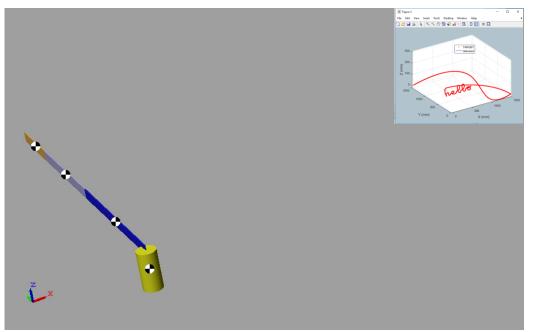
Questions AND Answers

How do you get ALL of the examples that you saw today?

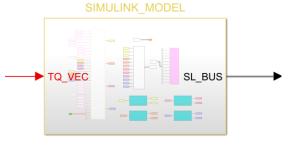




How do you make a rigid body machine move the way you want it to?



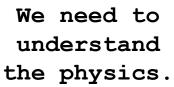
Mathematical model





$$M(q).\ddot{q} + C(\dot{q},q).\dot{q} + K(q).q + g(q) = Q$$

$$\ddot{q} = [M(q)]^{-1}.[Q - C(\dot{q}, q).\dot{q} - K(q).q - g(q)]$$









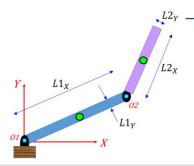
We need to apply Lagrange's equation

Laborious part

$$\frac{d}{dt}\frac{\partial L}{\partial \dot{q}_k} - \frac{\partial L}{\partial q_k} = Q_k$$

$$Q_k =$$

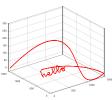
$$Q_k = \sum_{i=1}^{Nf_{nc}} \left(\overrightarrow{F}_i \cdot \frac{\partial \overrightarrow{v}_i}{\partial \dot{q}_k} \right) + \sum_{j=1}^{N\tau_{nc}} \left(\overrightarrow{\tau}_j \cdot \frac{\partial \overrightarrow{\omega}_j}{\partial \dot{q}_k} \right)$$



Q2 s

Laborious?

bh_tmp_EOM_file_WILL_BE_DELETED.txt × + q = TH1 s3 LHS of EOM is: I1G s*TH1 s DD I2G s*TH1 s DD I2G s*TH2 s DD (L1X s^2 TH1 s DD*m1 s)/4 L1X s^2*TH1 s_DD*m2_s (L2X $s^2*TH1 s DD*m2 s)/4$ (L2X $s^2*TH2 s DD*m2 s)/4$ $(L1X_s*g_s*m1_s*cos(TH1_s))/2$ L1X s*g s*m2 s*cos(TH1 s)(L2X s*g s*m2 s*cos(TH1 s + TH2 s))/2 16 L1X s*L2X s*TH1 s DD*m2 s*cos(TH2 s) 17 (L1X s*L2X s*TH2 s DD*m2 s*cos(TH2 s))/218 $-(L1X s*L2X s*TH2 s D^2*m2 s*sin(TH2 s))/2$ -L1X s*L2X s*TH1 s D*TH2 s D*m2 s*sin(TH2 s) 20 ### RHS of EOM is: Q1 s ### q = TH2 s### LHS of EOM is: 28 I2G s*TH1 s DD I2G s*TH2 s DD (L2X $s^2*TH1 s DD*m2 s)/4$ (L2X $s^2*TH2 s DD*m2 s)/4$ 32 (L2X s*g s*m2 s*cos(TH1 s + TH2 s))/2(L1X s*L2X s*TH1 s DD*m2 s*cos(TH2 s))/234 (L1X_s*L2X_s*TH1_s_D^2*m2_s*sin(TH2_s))/2 35 ### RHS of EOM is:



2-dof

Approx 20 lines

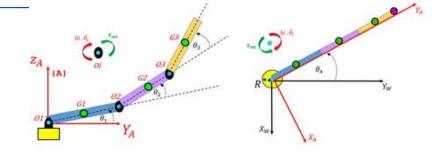
 $\ddot{ heta}_1$ $\ddot{ heta}_2$

4-dof

Approx 200 lines

 $egin{bmatrix} \ddot{ heta}_1 \ \ddot{ heta}_2 \ \ddot{ heta}_3 \ \ddot{ heta} \end{array}$

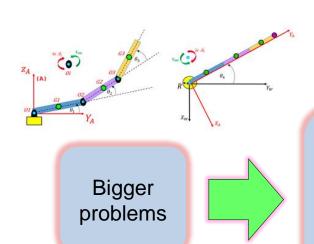
$$\frac{d}{dt}\frac{\partial L}{\partial \dot{q}_k} - \frac{\partial L}{\partial q_k} = Q_k$$



```
### LHS of EOM is:
                 (L1Y_s^2*TH1_s_DD*m1_s)/3
                L1Y s^2*TH1 s DD*m2 s
                L1Y s^2*TH1 s DD*m3 s
                 (L2Y s^2*TH1 s DD*m2 s)/3
                 (L2Y s^2*TH2 s DD*m2 s)/3
                 (L3Y s^2*TH1 s DD*m3 s)/3
                 (L3Y s^2*TH2 s DD*m3 s)/3
                 (L3Y s^2*TH3 s DD*m3 s)/3
                 (L1Z s^2*TH1 s DD*m1 s)/12
                 (L2Z s^2*TH2 s DD*m2 s)/12
                 (L3Z s^2*TH1 s DD*m3 s)/12
                 (L3Z s^2*TH2 s DD*m3 s)/12
                 (L3Z s^2*TH3 s DD*m3 s)/12
                 (L3Y s^2*TH4 s D^2*m3 s*sin(2*TH1 s + 2*TH2 s + 2*TH3 s))/6
                 -(1.37 \text{ s}^2 + \text{TH4 s} \text{ D}^2 + \text{m3 s}^2 + \text{sin}(2 + \text{TH1 s} + 2 + \text{TH2 s} + 2 + \text{TH3 s}))/24
                  -(L1Y s*L3Y s*TH3 s D*TH4 s D*m3 s*sin(2*TH1 s + TH2 s + TH3 s))/2
                 -(L1Y s*L3Y s*TH2 s D*TH4 s D*m3 s*sin(TH2 s + TH3 s))/2
                 -(L1Y_s*L3Y_s*TH3_s_D*TH4_s_D*m3_s*sin(TH2_s + TH3_s))/2
                  -L2Y s*L3Y s*TH1 s D*TH4_s_D*m3_s*sin(2*TH1_s + 2*TH2_s + TH3_s)
                  -L2Y_s*L3Y_s*TH2_s_D*TH4_s_D*m3_s*sin(2*TH1_s + 2*TH2_s + TH3_s)
                  -(L2Y s*L3Y s*TH3 s D*TH4 s D*m3 s*sin(2*TH1 s + 2*TH2 s + TH3 s))/2
                 -(L1Y s*L2Y s*TH2 s D*TH4 s D*m2 s*sin(TH2 s))/2
                  -L1Y s*L2Y s*TH2 s D*TH4 s D*m3 s*sin(TH2 s)
                 -(L2Y s*L3Y s*TH3_s_D*TH4_s_D*m3_s*sin(TH3_s))/2
                 -L1Y s*L2Y s*TH1 s D*TH4 s D*m2 s*sin(2*TH1 s + TH2 s)
                 -2*L1Y_s*L2Y_s*TH1_s_D*TH4_s_D*m3_s*sin(2*TH1_s + TH2_s)
                 -(L1Y_s*L2Y_s*TH2_s_D*TH4_s_D*m2_s*sin(2*TH1_s + TH2_s))/2
                 -L1Y s*L2Y s*TH2 s D*TH4 s D*m3 s*sin(2*TH1 s + TH2 s)
     ### RHS of EOM is:
209
210
```



Encouraging Deeper Learning engagements in your classroom:



The understanding of the problem physics:

- 3D motion
- Inertia matrix
- · Passive Rotations
- Vector sum of angular velocities

Problem Solving and practice

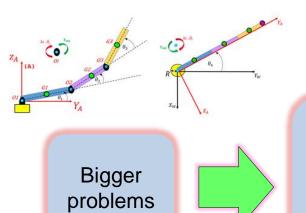
Hand written implementation

BRAIN Conceptual Difficulty



HAND Computational Difficulty







The understanding of the problem physics:

- 3D motion
- Inertia matrix
- Passive Rotations
- Vector sum of angular velocities



Problem Solving and practice

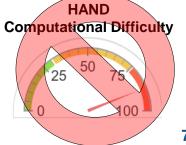
Computational Thinking:

- Brain
- **Technology**



BRAIN Conceptual Difficulty







Enabling Computational Thinking using MATLAB

Problem Solving and practice

Computational Thinking:

- Brain
- Technology

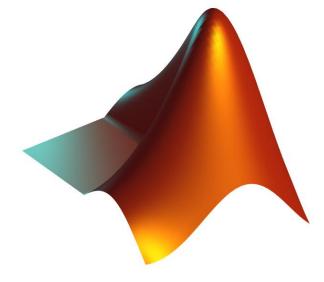


Decomposition

Algorithms +
Automation

Simulation



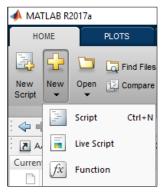


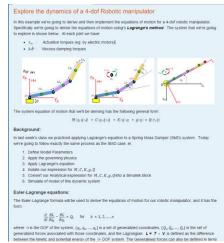


Decomposition

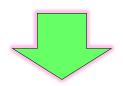


Live Script





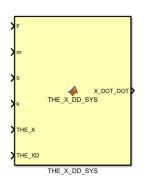
Algorithms + Automation



Symbolic Computing

- >> diff()
- >> matlabFunctionBlock()

our_EOM(t) =
$$m\frac{\partial^2}{\partial t^2} x(t) + k x(t) = F - b \frac{\partial}{\partial t} x(t)$$

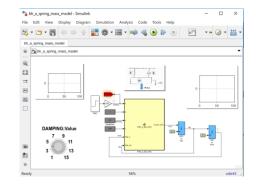


Simulation

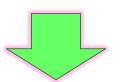


Numeric via Block Diagram

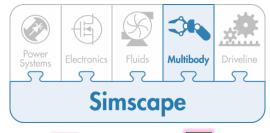
SIMULINK



Simulation (again)



Compare against GOLDEN Reference

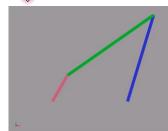






Validate

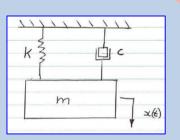
Hand
Derivation
vs
Simscape



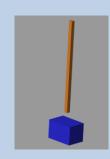
Visualization



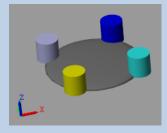
1-dof (Hand derivation workflow)



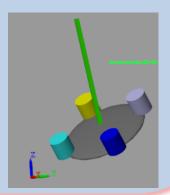
3D inverted
Pendulum
(Simscape Multibody)



Quadcopter (LQI control design)



Quadcopter balancing an inverted pendulum



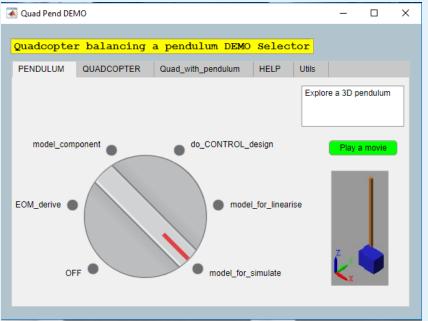
Today's case studies:

- A partnership that scales
 - Same workflow for small and BIG problems
- Divide and Conquer
 - Capture rationale and implementation
- The Modelling choices
 - Symbolic, Numeric, Block Diagram, Simscape
- The opportunities to explore and Discover
 - Visualization
 - Simulation

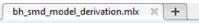


Demo these concepts



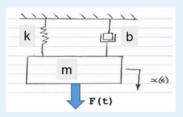


Task: Spring Mass Damper



Explore the dynamics of a 1-dof Spring Mass Damper

In this example we're going to derive and then implement the equations of motion for 1-dof Spring Mass Damper system. Specifically we're going to derive the equations of motion using's Lagrange's method. The system that we're going to explore is shown below.





Background:

From our year 1 class in physics and mechanics, we derived using Newton's 2nd law, the equation of motion for the dynamics of a Spring Mass damper system. Recall that it had the following form:

$$m.\ddot{x} + b.\dot{x} + k.x = F(t)$$

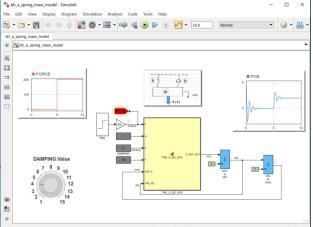
Today we'll use the Lagrangian approach to derive the same equations of motion for spring mass damper. We're going to break this problem down into the following 6 ste

- 1. Define Model Parameters
- 2. Apply the governing physics
- 3. Apply Lagrange's equation
- 4. Isolate our expression for $\ddot{x}(t)$
- 5. Convert our Analytical expression for \ddot{x} into a Simulink block
- 6. Simulate of model of this dynamic system

Euler-Lagrange equations:

Recall our earlier class where we derived and summarised the fundamental Lagrangi equations that allow us to derive system equations of motion:

$$\frac{d}{dt}\frac{\partial L}{\partial \dot{q}_k} - \frac{\partial L}{\partial q_k} = Q_k \qquad \text{where} \qquad Q_k = \sum_{i=1}^{Nf_{nc}} \left(\overrightarrow{F_i} \cdot \frac{\overrightarrow{\partial v_i}}{\partial \dot{q}_k} \right) + \sum_{j=1}^{N\tau_{nc}} \left(\overrightarrow{\tau_j} \cdot \frac{\overrightarrow{\partial \omega_j}}{\partial \dot{q}_k} \right)$$



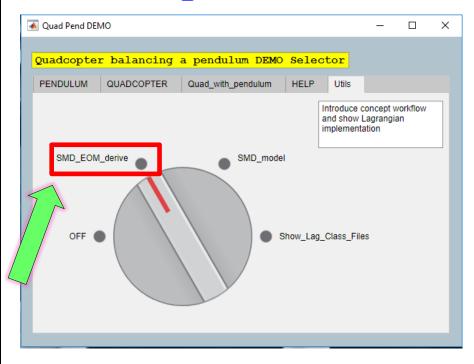




Live Script:



bh_smd_model_derivation.mlx

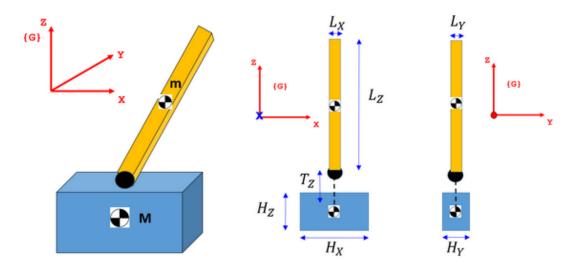


Task: 3-D Inverted Pendulum



Our 3D pendulum looks like this:

bh_derive_LAG_eom_for_3D_pend.mlx × +



- The pendulum is attached to the base cart via a spherical joint.
- The sperical joint permits the pendulum to yaw, pitch and roll.
- The base cart may translate in the inertial X,Y, and Z directions.
- . We will be applying forces to the Base cart only
- · We will represent the base cart as a point mass.

Defining the model parameters:

syms m_pend M_cart g syms Tv Z Lx Ly Lz

Defining the pendulum INERTIA:

So let's define the INERTIA matrix for the pendulum about its body fixed center of mass frame:



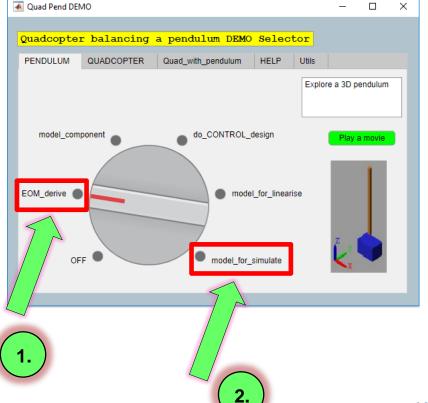


Live Script:



bh_derive_LAG_eom_for_3D_pend.mlx

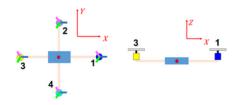
Try it:



Task: Quadcopter

bh_explore_LAG_eom_for_quad_include_spin_props.mlx × +

Task: Lagrangian approach for deriving Eoms for Quadcopter



In this task we're going to look at how the Lagrangian Dynamics approach can be used to derive the equations of motion of a Rigid Body. Steps that we'll take will include:

- What is a PASSIVE rotation matrix ?
- How do i construct a Direction Cosine Matrix (DCM) from a given rotation sequence?
- What is the relationship between BODy rates and EULER rates?
- What is the KE and PE of just the airframe?
- What is the KE and PE of each rotor+Propeller assembly?
- Apply Lagrange's equation to derive the system EoMs

Euler-Lagrange equations:

The Euler-Lagrange formula will be used to derive the equations of motion for a Rigid Body, and it has the form:

$$\frac{d}{dt}\frac{\partial L}{\partial \dot{q}_k} - \frac{\partial L}{\partial q_k} = Q_k \quad \text{for} \quad k = 1, 2, \dots, n$$

where *n* is the DOF of the system, $\{q_1, q_2, \dots, q_n\}$ is a set of generalized coordinates, $\{Q_1, Q_2, \dots, Q_n\}$ is the set of generalized forces associated with those coordinates, and the Lagrangian: L = T - V is defined as the difference between the kinetic and potential energy of the n- DOF system. The Generalised forces can also be defined in terns of the non conservative forces and torques acting on the multibody system. The formula for the generalised forces acting on the system is:

$$Q_{k} = \sum_{i=1}^{Nf_{nc}} \left(\overrightarrow{F_{i}} \cdot \frac{\overrightarrow{\partial v_{i}}}{\overrightarrow{\partial q_{k}}} \right) + \sum_{j=1}^{N\tau_{nc}} \left(\overrightarrow{\tau_{j}} \cdot \frac{\overrightarrow{\partial \omega_{j}}}{\overrightarrow{\partial q_{k}}} \right)$$

where:

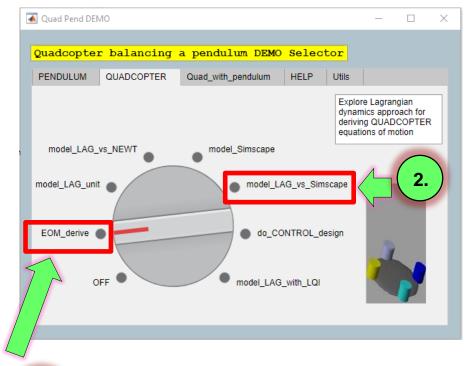
• Q_k : is the generalised force associated with the k^{th} generalised co-ordinate q_k





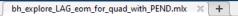
Live Script:

bh_explore_LAG_eom_for_quad_include_spin_props.mlx

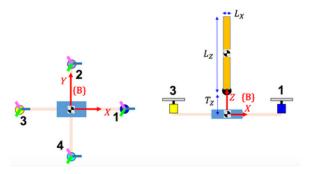




Task: Quadcopter balancing a pendulum



Task: Lagrangian approach for deriving Eoms for a Quadcopter balancing a pendulum



In this task we're going to look at how the Lagrangian Dynamics approach can be used to derive the equations of motion of a quadcopter balancing an inverted pendulum. Steps that we'll take will include:

- What is a PASSIVE rotation matrix ?
- How do i construct a Direction Cosine Matrix (DCM) from a given rotation sequence?
- What is the relationship between BODy rates and EULER rates ?
- What is the KE and PE of just the airframe?
- What is the KE and PE of each rotor+Propeller assembly ?
- · Apply Lagrange's equation to derive the system EoMs

Euler-Lagrange equations:

The Euler-Lagrange formula will be used to derive the equations of motion for a Rigid Body, and it has the form:

$$\frac{d}{dt}\frac{\partial L}{\partial \dot{q}_k} - \frac{\partial L}{\partial q_k} = Q_k \quad \text{for} \quad k = 1, 2, \dots, n$$

where *n* is the DOF of the system, $\{q_1, q_2, \dots, q_n\}$ is a set of generalized coordinates, $\{Q_1, Q_2, \dots, Q_n\}$ is the set of generalized forces associated with those coordinates, and the Lagrangian: L = T - V, is defined as the difference between the kinetic and potential energy of the *n*- DOF system. The Generalised forces can also be defined in terns of the non conservative forces and torques acting on the multibody system. The formula for the generalised forces acting on the system is:

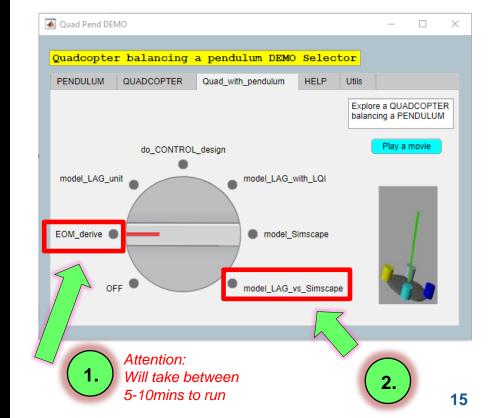




Live Script:



bh_explore_LAG_eom_for_quad_with_PEND.mlx





Wrap up



Enabling Computational Thinking using MATLAB

Problem Solving and practice

Computational Thinking:

- Brain
- Technology



Decomposition

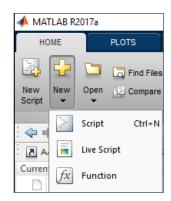
Algorithms +
Automation

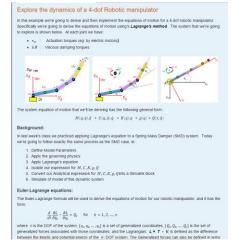
Simulation

Decomposition



Live Script

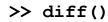




Algorithms + Automation

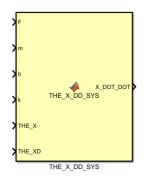


Symbolic Computing



>> matlabFunctionBlock()

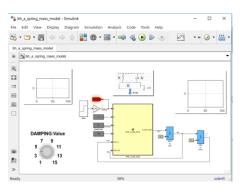
our_EOM(t) = $m\frac{\partial^2}{\partial t^2} x(t) + k x(t) = F - b \frac{\partial}{\partial t} x(t)$

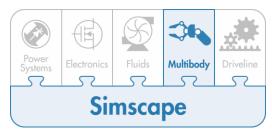


Simulation



Numeric via Block Diagram



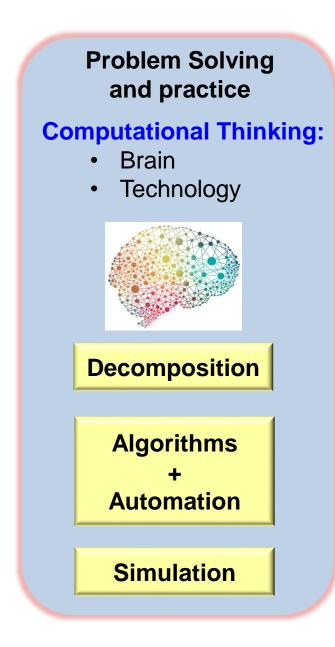


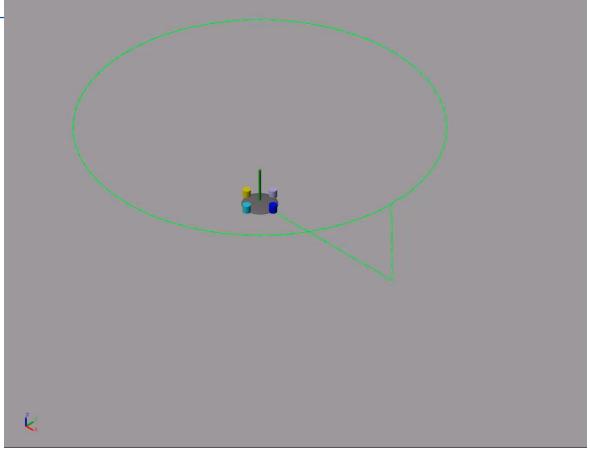
Q/A:

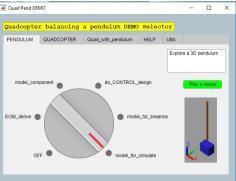
Are there some questions please ?

 Download the examples that you saw today ... and more that you didn't!

R2018a



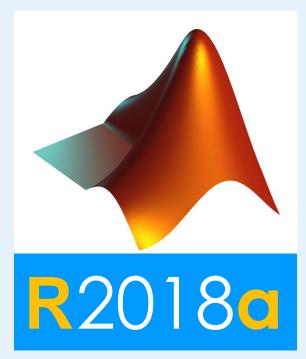


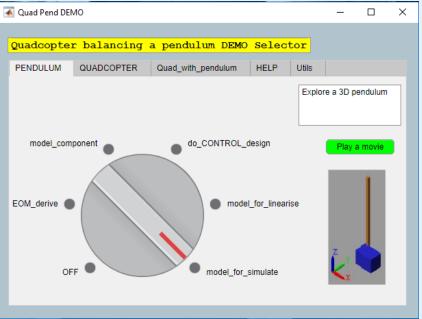


>> bh_startup_quad_and_pendulum



DEMO requirements







What you need to run the demo:

- R2018a (or NEWER)
 - MATLAB
 - Symbolic Maths Toolbox
 - Control System toolbox
 - Simulink
 - Simscape
 - Simscape Multibody
 - Simulink Control Design



These products will probably be on your TAH campus license.

If they're not ... then contact the **MathWorks**

OR

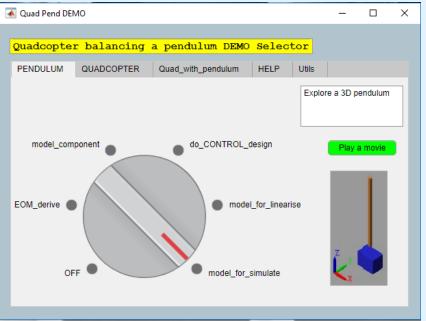
help yourself here:

https://www.mathworks.com/programs/trials/trial_request.html

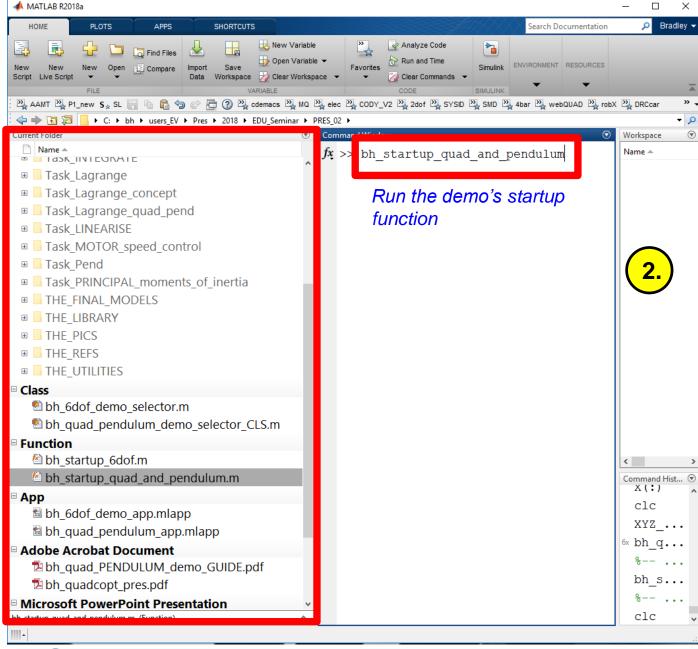


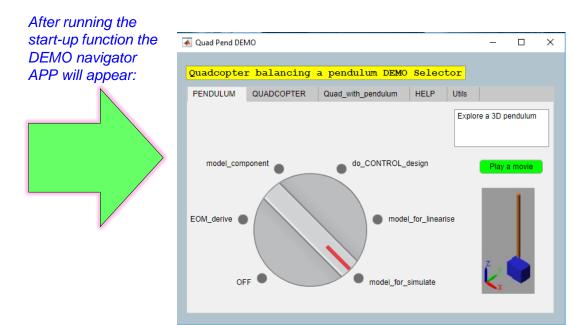
DEMO setup and launch













In MATLAB, navigate to the DEMO folder