

# Simulation of solid fuel engine

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## 1 Calculations regarding engine thrust

### 1.1 Thrust

We calculate engine thrust by usage of following equation, which as input takes  $P_{ch}$ ,  $A_t$  and  $P_e$ .

$$F = A_t P_0 \sqrt{\frac{2k^2}{k-1} \left( \frac{2}{k+1} \right)^{\frac{k+1}{k-1}} \left[ 1 - \left( \frac{P_e}{P_0} \right)^{\frac{k-1}{k}} \right]} + A_e (P_e - P_a) \quad (1)$$

### 1.2 Exit pressure

Unfortunately, we don't have  $P_e$  so we need to calculate it numerically from following equation for expansion ratio.

$$E = \frac{A_t}{A_e} = \left( \frac{k+1}{2} \right)^{\frac{1}{k-1}} \left( \frac{P_e}{P_0} \right)^{\frac{1}{k}} \sqrt{\left( \frac{k+1}{k-1} \right) \left[ 1 - \left( \frac{P_e}{P_0} \right)^{\frac{k-1}{k}} \right]} \quad (2)$$

Where E is:

$$E = \frac{A_t}{A_e} = \frac{1}{\text{Expansion ratio}}$$

Numerical calculations will be performed by taking the minimum of  $\Delta E$  in order to obtain the corresponding  $P_e$  for which, upon substitution into the formula, we will achieve a value closest to the E calculated from the nozzle parameters.

$$0 \approx \Delta E = \left| E - \left( \frac{k+1}{2} \right)^{\frac{1}{k-1}} \left( \frac{P_e}{P_0} \right)^{\frac{1}{k}} \sqrt{\left( \frac{k+1}{k-1} \right) \left[ 1 - \left( \frac{P_e}{P_0} \right)^{\frac{k-1}{k}} \right]} \right| \quad (3)$$

## 2 Calculations regarding engine chamber parameters

Calculations related to the combustion chamber focus on a key equation for solid fuel rocket engines, namely the equation for pressure in the chamber.

$$P_{ch}(A_b) = K_n^{\frac{1}{1-n}} (c^* \rho_p a)^{\frac{1}{1-n}}, \quad K_n = \frac{A_b}{A_t} \quad (4)$$

Therefore, we treat  $P_{ch}$  as a function of  $A_b$ . To conduct simulations, we need the function  $A_b(x)$ , where x is the burn distance. Since this simulator considers only the BATES geometry, below is the formula for precisely this geometry.

$$A_b(x) = 2\pi N((R+x)(L-2x) + \left( \frac{D}{2} \right)^2 - (R+x)^2) \quad (5)$$

Where  $R = d_p/2$  and D is external grain diameter, L is length of one segment and N is number of segments.

When it comes to computation of  $x$ , we solve it numerically. That means using rectangle method of solving integrals. In this case it's a simple one.

$$x(t) = \int_0^{t_c} r dt = \sum r \Delta t \quad (6)$$

Where regression rate  $r$  is a function of  $P_{ch}$ , therefore for each  $t_c + \Delta t$  we will calculate new value of regression, which we will use for corresponding addition to burn distance.

$$r(P_{ch}) = a(P_{ch})^n$$

As a closing I want to also mention that we use exactly same method to calculate total impuls, that is simple numerical integration.