

Delta IV Heavy

The **Delta IV Heavy** (**Delta 9250H**) is an expendable heavy-lift launch vehicle, the largest type of the Delta IV family. It is the world's third highest-capacity launch vehicle in operation, behind NASA's Space Launch System and SpaceX's Falcon Heavy and closely followed by CASC's Long March 5. [4][5] It is manufactured by United Launch Alliance (ULA) and was first launched in 2004. [6] ULA will retire the Delta IV Heavy in 2024. [7] As of June 2023, one flight remains.

The Delta IV Heavy consists of a central Common Booster Core (CBC), with two additional CBCs as liquid rocket boosters instead of the GEM-60 solid rocket motors used by the Delta IV Medium+ versions. At lift off, all three cores operate at full thrust, and 44 seconds later the center core throttles down to 55% to conserve fuel until booster separation. The two side boosters burn out at 242 seconds after launch and are separated as the core booster throttles back up to full thrust. The core burns out 86 seconds later, and the second stage completes the ascent to orbit. [3]

The launch vehicle uses three RS-68 engines, one in the central core and one in each booster. On the last seconds of countdown, the hydrogen fuel flows through the engines and upwards along the booster body, and after the ignition that hydrogen inflames, making the characteristic fireball and charred look of the booster. [9]

History

The Delta IV line of rockets was developed by McDonnell Douglas. The program was later transferred to United Launch Alliance. The Delta IV Heavy is the most powerful member of the line, which also includes the smaller Delta IV Medium. The Delta IV Heavy can lift 28,370 kg (62,550 lb) to low Earth orbit and 13,810 kg (30,450 lb) to geostationary transfer orbit (GTO). It is an all liquid-fueled launch vehicle, consisting of an upper stage, one main booster and two strap-on boosters.

The first launch of the Delta IV Heavy in 2004 carried a boilerplate payload and was a partial failure. Cavitation in the liquid-oxygen propellant lines caused shutdown of both

Delta IV Heavy



Delta IV Heavy launches from Vandenberg Air Force Base

Vandenberg Air Force Base				
Function	Orbital heavy-lift launch vehicle			
Manufacturer	United Launch Alliance			
Country of origin	United States			
Cost per launch	US\$350 million ^[1] NRO: US\$440 million			
Size				
Height	72 m (236 ft)			
Diameter	5 m (16 ft)			
Width	15 m (49 ft)			

733,000 kg

(1,616,000 lb)

Mass



Delta IV Heavy for Exploration Flight Test-1

boosters eight seconds early, and the core engine nine seconds early: this resulted in a lower staging velocity for which the second stage was unable to compensate. payload was left in a lower than intended orbit.[11] Its first operational payload the **DSP-23** was satellite. successfully launched in 2007; it was then used to launch a further five visual and electronic reconnaissance satellites for the National

Reconnaissance

Office (NRO) through 2013.

In December 2014, the Delta IV Heavy was used to launch an un-crewed test flight of the <u>Orion spacecraft</u>, designated <u>Exploration Flight Test 1</u> (EFT-1). After several delays, the mission was successfully launched at 12:05 <u>UTC</u> on 5 December 2014. [12]

On 12 August 2018, the Delta IV Heavy with an additional Star 48BV third stage was used to launch the Parker Solar Probe into an elliptical heliocentric orbit. [13] In May 2023, the final Delta IV Heavy core and boosters finished construction, officially ending Delta IV production and making way for the Vulcan launch vehicle. [14]

Capabilities

Capacity of the Delta IV Heavy:

- Low Earth orbit (LEO), 200 km \times 28.7°: 28,790 kg $(63,470 \text{ lb})^{[15]}$
- Low Earth orbit (ISS), 407 km × 51.6°: 25,980 kg (57,280 lb) [16]
- Geosynchronous transfer orbit (GTO): 14,220 kg (31,350 lb) [15]

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Stages	2+					
Capacity						
Payload to <u>LEO</u>						
Mass	28,790 kg					
	(63,470 lb)					
Payload to GTO						
Mass	14,220 kg					
	(31,350 lb)					
Associated rockets						
Family	Delta					
Comparable	Long March 5					
	Falcon Heavy · Proton-M					
l aunch	history					
Status	Production ended					
Launch sites	Cape Canaveral,					
Ludiioii oitoo	SLC-37B					
	Vandenberg, SLC-6					
Total launches	15					
Success(es)	14					
Partial failure(s)	1					
First flight	21 December 2004					
_	(USA-181)					
Last flight	22 June 2023					
	(<u>NROL-68</u>)					
Type of	EFT-1 · NRO					
passengers/cargo	classified payloads					
	(Kennen · Orion) · Parker Solar Probe					
D I.						
	s (CBC)					
No. boosters	2					
Height	40.8 m (134 ft)					
Diameter	5.1 m (17 ft)					
Empty mass	26,000 kg (57,000 lb)					
Gross mass	226,400 kg					
Gross mass	(499,100 lb)					
Propellant mass	200,400 kg					
	(441,800 lb) ^[2]					

Geosynchronous orbit (GEO): 6,750 kg (14,880 lb) [15]

Lunar transfer orbit (LTO): 10,000 kg (22,000 lb)

■ Mars transfer orbit: 8,000 kg (18,000 lb) [17]

Maximum thrust 3,140 kN (710,000 lbf) Total thrust 6,280 kN (1,410,000 lbf) Specific impulse Sea level: 360 s (3.5 km/s) Vacuum: 412 s (4.04 km/s) Burn time 242 seconds [3] Propellant LH₂ / LOX First stage (CBC) Height 40.8 m (134 ft) Diameter 5.1 m (17 ft) Gross mass 226,400 kg (499,100 lb) Propellant mass 200,400 kg (441,800 lb) Powered by 1 RS-68A Maximum thrust 3,140 kN (710,000 lbf) Specific impulse Sea level: 360 s (3.5 km/s) Vacuum: 412 s (4.04 km/s) Burn time 328 seconds Propellant LH₂ / LOX Second stage (DCSS) Height 13.7 m (45 ft) Diameter 5.1 m (17 ft) Gross mass 30,700 kg (67,700 lb) Propellant mass 27,220 kg (60,010 lb) Propellant mass 27,220 kg (60,010 lb) Ad2 s (4.53 km/s) 462 s (4.53 km/s) Burn time 1125 seconds Propellant LH₂ / LOX	Powered by	1 RS-68A				
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Specific impulse	Total thrust	,				
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	Specific impulse	462 s (4.53 km/s)				
Propellant LH ₂ / LOX	Burn time	1125 seconds				
	Propellant	LH ₂ / LOX				

The Delta IV Heavy's total mass at launch is approximately 733,000 kg (1,616,000 lb) and produce around 952,000 kg (2,099,000 lb) of thrust to power the rocket skyward at liftoff. [18]

Launch history

[hide]

					Outcome	
Flight	Date	Payload ^[19]	Mass	Launch site	[19]	
1	21 December 2004	DemoSat, Sparkie / 3CS-1 and Ralphie / 3CS-2	~13,228 pounds (6000 Cape Canaveral, SLC-37B		Partial failure ^[a]	
2	11 November 2007	DSP-23 Defense Support Program	11,574 pounds (5,250 kilograms)	Cape Canaveral, SLC-37B	Success	
3	18 January 2009	Orion 6 / Mentor 4 (USA- 202 / NROL-26)	Classified	Cape Canaveral, SLC-37B	Success	
4	21 November 2010	Orion 7 / Mentor 5 (<u>USA-</u> 223 / <u>NROL-32</u>)	Classified	Cape Canaveral, SLC-37B	Success	
5	20 January 2011	KH-11 Kennen 15 (USA- 224 / NROL-49)	<37,479 pounds (17,000 kilograms)	Vandenberg, SLC-	Success	
6	29 June 2012	Orion 8 / Mentor 6 (USA- 237 / NROL-15)	Classified	Cape Canaveral, SLC-37B	Success	
7	26 August 2013	KH-11 Kennen 16 (<u>USA-</u> 245 / <u>NROL-65</u>)	<37,479 pounds (17,000 kilograms)	Vandenberg, SLC-	Success	
8	5 December 2014	Orion Exploration Flight Test-1 (EFT-1)	46,297 pounds (21,000 kilograms) [20][b]	Cape Canaveral, SLC-37B	Success	
9	11 June 2016	Orion 9 / Mentor 7 (USA- 268 / NROL-37)	Classified	Cape Canaveral, SLC-37B	Success	
10	12 August 2018	Parker Solar Probe ^[c]	1,510 pounds (685 kilograms)	Cape Canaveral, SLC-37B	Success	
11	19 January 2019	NROL-71	Classified	Vandenberg, SLC-	Success	
12	11 December 2020	Orion 10 / Mentor 8 (USA- 268/ NROL-44) ^{[21][22]}	Classified	Cape Canaveral, SLC-37B	Success	
13	26 April 2021	KH-11 Kennen 17 (NROL- 82)	Classified	Vandenberg, SLC-	Success	
14	24 September 2022	KH-11 Kennen 18 (NROL- 91)	Classified	Vandenberg, SLC-	Success	
15	22 June 2023	Orion 11 / Mentor 9 (NROL- 68)[23]	Classified	Cape Canaveral, SLC-37B	Success	

Future launches

Missions thirteen through sixteen were announced by the <u>National Reconnaissance Office</u>. For the final five missions (12-16) including modifications, ULA was awarded US\$2.2 billion, or US\$440 million per launch. This can be compared with the Falcon Heavy launch price of \$90M to \$150M. As of June 2023, only one remains before ULA retires the Delta IV Heavy.

Flight	Date	Launch site	Payload [19]	Mass	Orbit	Customer	Launch outcome
16	1 March 2024 ^[26]	CCSFS, SLC-37B	NROL-70	Classified	GEO	US NRO	Planned
	Reconnaissance satellite, final flight of the Delta rocket family.						

- a. CBCs underperformed, lower orbit than planned
- b. The officially reported mass of 21,000 kg includes the Launch Abort System (LAS) which did not reach orbit, but excludes the residual mass of the upper stage, which *did* reach orbit, likely offsetting the mass of the LAS.
- c. Star 48BV upper stage

Comparable vehicles

Current:

- Long March 5 (geostationary transfer orbit)
- Long March 5B (low Earth orbit)
- Long March 7A (geostationary transfer orbit)
- Falcon Heavy
- Proton-M

In development:

- Angara A5
- Ariane 6
- New Glenn
- Vulcan Centaur

Retired or cancelled:

- Ariane 5 (retired)
- Atlas V Heavy (proposed, never developed)
- Saturn IB (retired)
- Titan III (retired)
- <u>Titan IV</u> (retired)

See also

- Heavy-lift launch vehicle
- Comparison of orbital launch systems
- Comparison of orbital rocket engines
- Comparison of space station cargo vehicles



National Launch System, (1991–1993) study

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External links

 Delta IV Booster Integration Another Step Toward First Orion Flight (http://www.nasa.gov/content/d elta-iv-booster-integration-another-step-toward-first-orion-flight/#.VHxsvDHF9Zs)

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