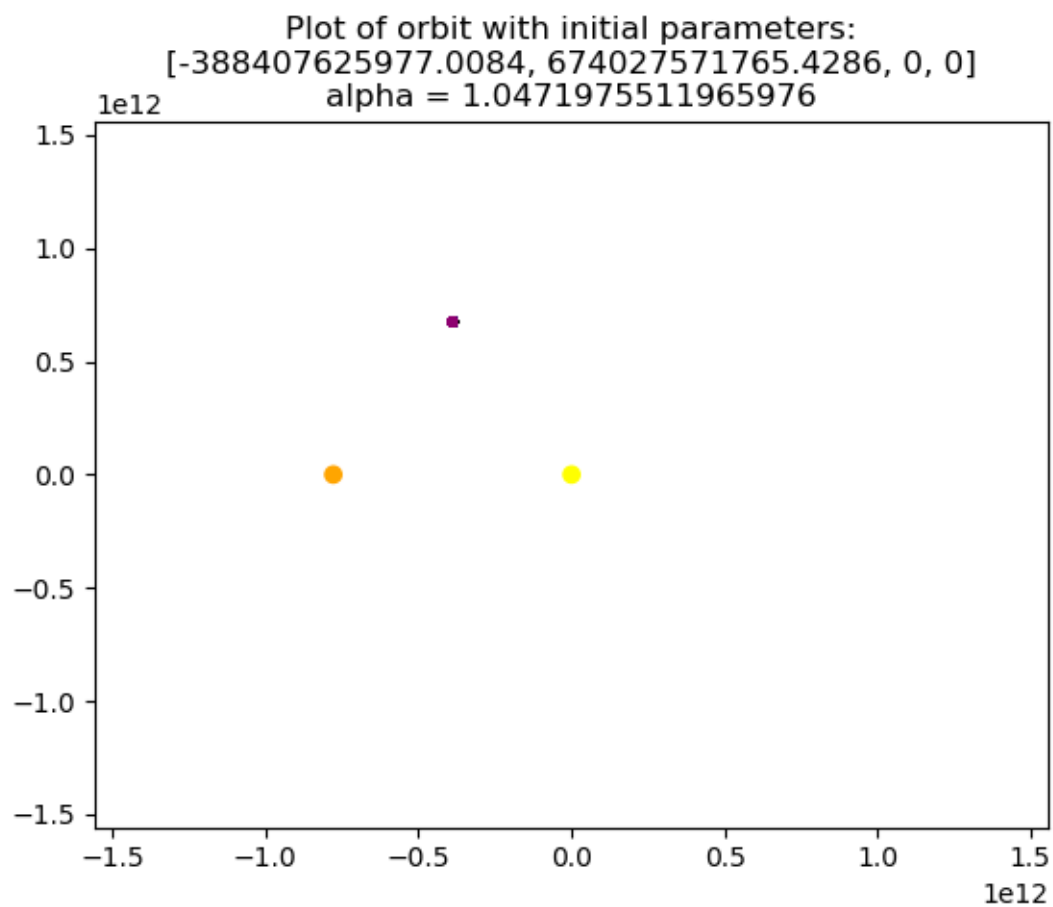


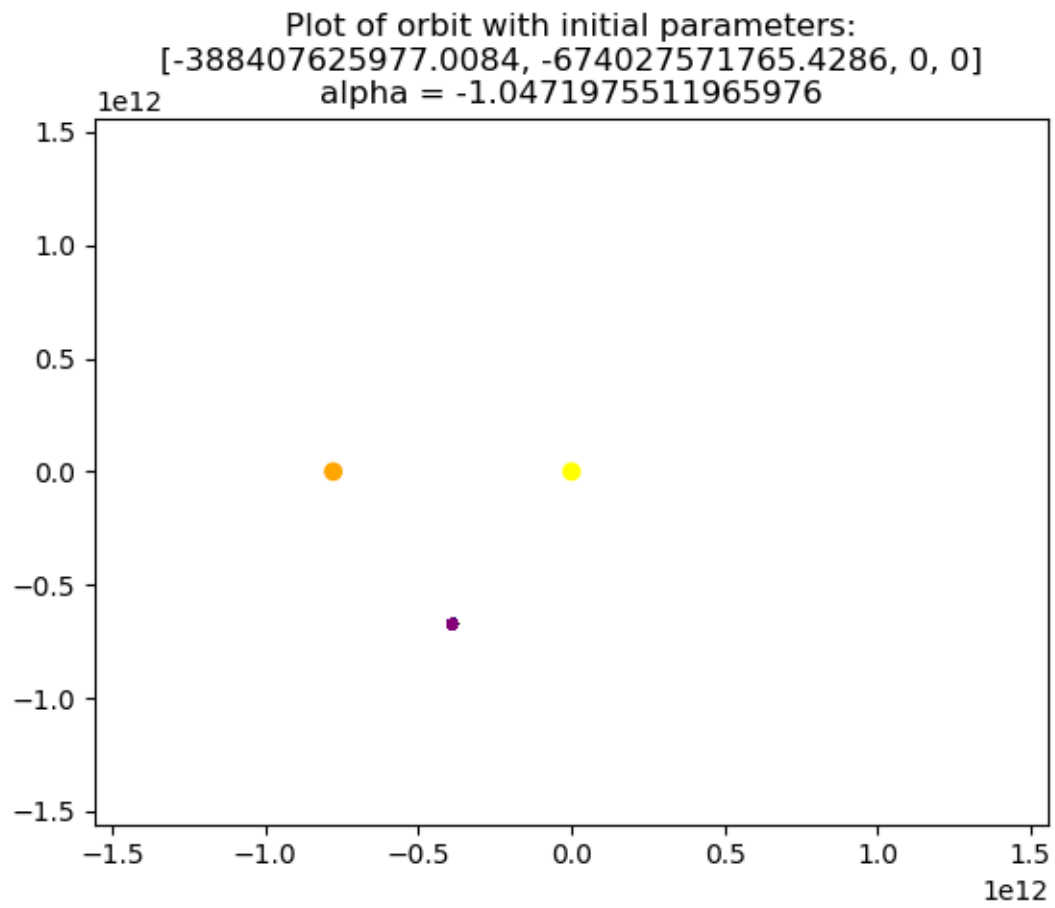
## Homework 2

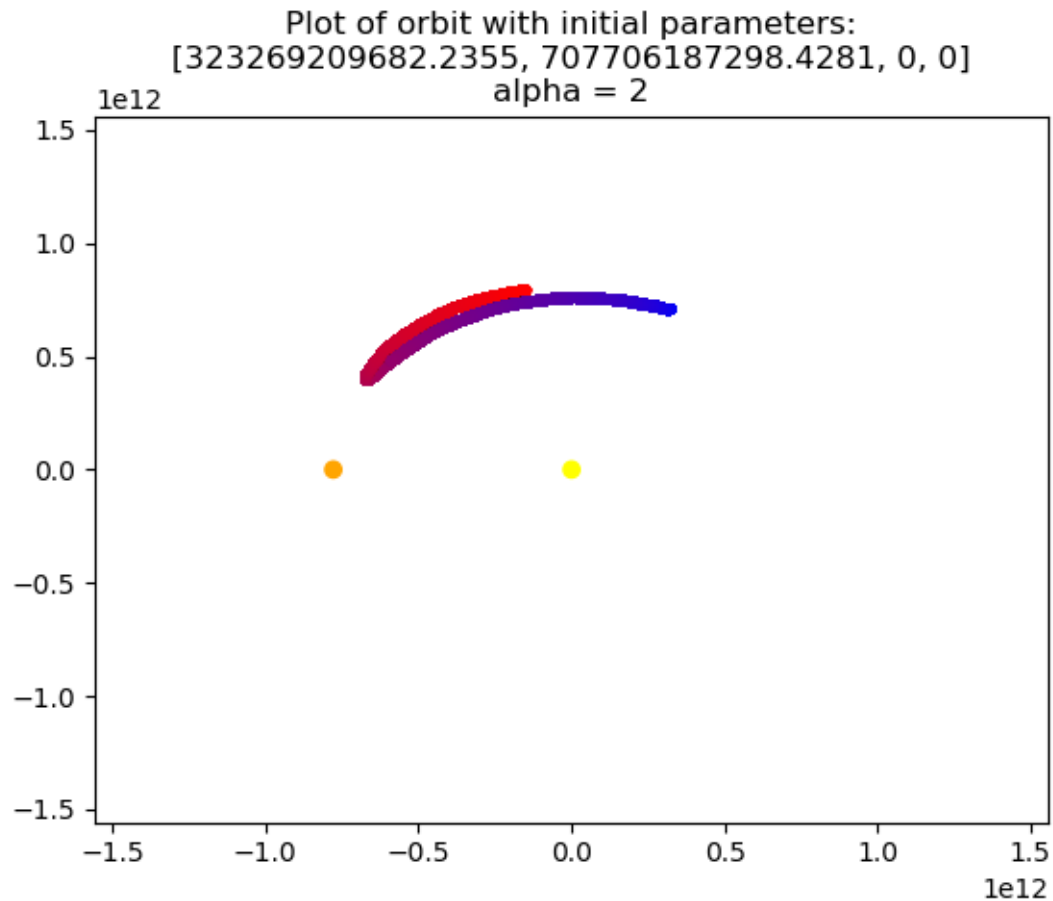
Philip Carr

For all orbit plots, color changes from blue to red as time advances.

1. Orbit plots:

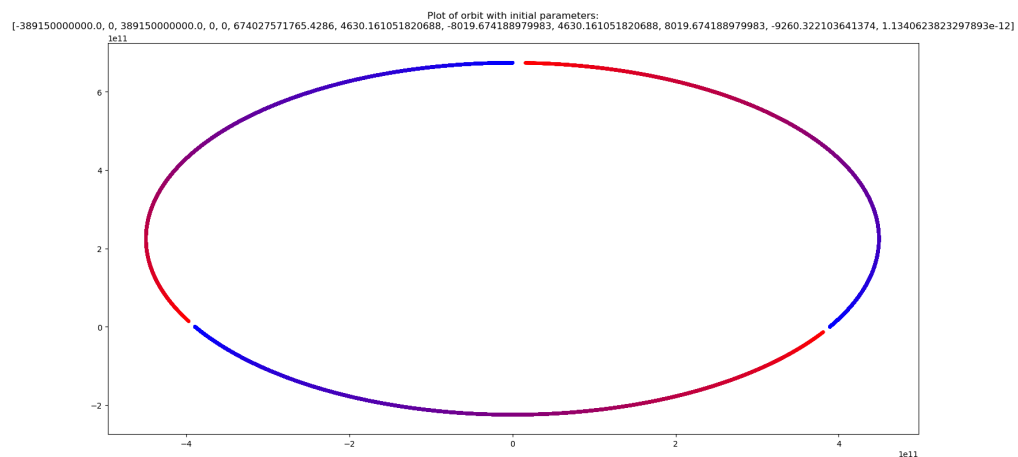






As seen in the above three plots, the Lagrange points specified by  $\alpha = \pi/3$  and  $\alpha = -\pi/3$  are stable.

## 2. Orbit of stable equilateral triangle Lagrange solution:



Equilateral Triangle Orbit Velocity Equation derivation:

Each mass  $M_1 = M_2 = M_3 = M$  orbits the center of mass of the three-body system distance  $r$  away from  $m$ : Let the masses  $M_1$ ,  $M_2$ , and  $M_3$  be located in the shape of an equilateral triangle such that  $M_2$  and  $M_3$

are vertically colinear (along the  $y$ -direction) and  $M_1$  is to the right of  $M_2$  and  $M_3$  (in the  $+x$ -direction). Here, the forces in the  $y$ -direction of  $M_2$  and  $M_3$  acting on  $M_1$  are cancelled out, since  $M_2 = M_3$ ,  $M_2$  and  $M_3$  are the same magnitude of distance away from  $M_1$ , and  $M_2$  and  $M_3$  are equally and oppositely as far away in the  $y$ -direction from  $M_1$ 's  $y$ -position. Therefore, the only nonzero force on  $M_1$  is in the  $x$ -direction here. Therefore,

$$F_x = M_1 a = -\frac{GM_2 M_1 r}{d^3} - \frac{GM_3 M_1 r}{d^3} = -\frac{2GM_1^2 r}{d^3}$$

( $a$  here is centripetal acceleration, since  $M_1$  is in a circular orbit.)

$$\Rightarrow \frac{M_1 v^2}{r_c^2} = \frac{2GM_1^2 r}{d^3}$$

$$\Rightarrow v^2 = \frac{2GM r r_c}{d^3}$$

$$\Rightarrow v = \sqrt{\frac{2GM r r_c}{d^3}}$$

( $r$  is the distance from  $M_1$  to  $M_2$  and equal to the distance from  $M_1$  to  $M_3$  in the  $x$  direction, which is  $r = d \cos(\frac{\pi}{6}) = \frac{\sqrt{3}d}{2}$ .  $r_c$  is the distance from  $M_1$  to the center of mass of the system about which  $M_1$  orbits, which is at the barycentric center of mass of the equilateral triangle, in which each mass in the triangle is away from this center of mass by two-thirds the distance from the mass' location to the opposite side of the triangle by the perpendicular line segment that bisects the opposite side by from the mass' location. Therefore,  $M_1$  is colinear with the center of mass of the system along the  $x$ -direction,  $r_c = \frac{2}{3}r$ .)

$$\Rightarrow v = \sqrt{\frac{2GM r (\frac{2}{3}r)}{d^3}} = \sqrt{\frac{\frac{4}{3}GM r^2}{d^3}} = \sqrt{\frac{\frac{4}{3}GM (\frac{3}{4}d^2)}{d^3}} = \sqrt{\frac{GM}{d}}. \square$$

3. Figure-8 orbit:

