Lion Group

TASKCARD

A/C TYPE	Effectivity	DESCRIPTION							WORK ORDER NO.	
		737NG - DAILY CHECK.								
A/C REG.	A/C MSN.	ACCESS							TASKCARD NO.	
									B789-05-INT-02-01-IDN	
A/C TSN.	A/C CSN.	192G	413AL	414AR	423AL	424AR			THRESHOLD	INTERVAL
:										
OPERATOR	PLACE	ZONE							TASK	REVISION
		211 632	212 713	414 734	424 744	531	532	631	DET	39
START DATE	FINISH DATE	NOTE							ATA	SKILL
		☐ ETOPS ☐ RVSM ☐ RNP10 ☐ RII ☐ CDCCL							05	A/P

REFERENCE			
Doc No.	**Doc Description**	**Doc No.**	**Doc Description**
AMM 12-13-21-200-801	IDG Oil Level Check	AMM 12-13-21-600-801	IDG Servicing (Oil Fill)
AMM 12-15-21-210-801	Crew Oxygen Cylinder Dispatch Pressure Check	AMM 12-15-51-610-802	Add Nitrogen or Air to the Tire
AMM 32-00-01-480-801	Landing Gear Downlock Pins Installation	EA B737NG-EI-34-226	PITOT PROBE COVER
AMM 24-34-00-710-801	The Operational Test of the Standby Power System	SRM 51-10-06	TAP TEST INSPECTION
AMM 30-41-41	Window/Pitot Heat Test	AMM 32-42-00-720-801	ANTISKID/AUTOBRAKE CONTROL UNIT OPERATIONAL TEST
AMM 34-43-00-860-145-002	WEATHER RADAR OPERATIONAL TEST	AMM 34-45-00-710-801	TCAS - OPERATIONAL TEST
AMM 73-21-00-740-803-F00	EEC BITE TEST - RECENT FAULTS	EA B737NG-EA-11-001R11	Installation Tire Pacard Servicing
EI GEN-E-35-083R1	LIMITATION PRESSURE OF PORTABLE OXYGEN BOTTLE MUST BE REMOVE FROM AIRCRAFT	FIM 31-62 TASK 801	BITE Procedure
FIM 32-09 TASK 801	PSEU BITE TEST	AMM 24-22-00-860-811	Supply Electrical Power
AMM 23-51-00-710-801	Flight Interphone System - Operational Test	AMM 23-12-00-710-801	VHF Communication System Operational Test
AMM 23-11-00-710-801	HF Comunication System Operational Test	AMM 12-13-31-200-801	APU Oil Level Inspection
AMM 12-11-00-680-801	Fuel System Sumping	AMM 27-41-71-700-801	Stabilizer Trim Motor Test

TOOLS REQUIRED		
PART NUMBER	**DESCRIPTION**	**QUANTITY**
06-5022-6800	FLUID SERVICING UNIT, CAPACITY 2 GALLON, MAX. PRESS. 200 PSI FOR MINERAL BASE	1
1001089-1	COVER - PROBE, PITOT STATIC	1
14-6806-6011	TYRE PRESSURE GAUGE	1
3001331-1	TAP TEST TOOL	1
C32026-6	EQUIPMENT - DOWNLOCK, NLG AND MLG (FLYAWAY KIT)	1
F70199-1	TIRE INFLATION TOOL	1
F80201-1	DRAIN TL	1
H3310	HEADPHONE - 600 OHM, WITH 1/4 INCH MONO RCAAUDIO PLUG / GROUND HEAD SET	1
SPL-1880	EQUIPMENT - DOWNLOCK, NLG AND MLG (FLYAWAY KIT)	1
STD-1056	CONTAINER - SOLVENT RESISTANT, 5 GALLON (19 LITERS)	1

BARCODE:

B789-05-INT-02-01-IDN

Lion Group

TASKCARD

WORK ORDER NO.	A/C REG.	A/C MSN.	A/C Effectivity	OPERATOR	TASK CARD NO.
					B789-05-INT-02-01-IDN

MATERIAL REQUIRED

PART NUMBER	DESCRIPTION	QUANTITY
141A6076-3	PIN - GUIDE ARM, FORWARD ENTRY DOOR	1
141A6077-2	WASHER	1
44250	HIGH PURITY 44 MARKER, COLOR BASED ON PART NUMBER	1
AEROSHELL GREASE 33	AIRCRAFT GENERAL PURPOSE GREASE, PASTEBMS3-33 TYPE 1	1
AEROSHELL-FLUID-41	FLUID - LANDING GEAR SHOCK STRUT	1
ALCOHOL-70	ISOPROPYL ALCOHOL, REAGENT GRADE	1
BACS12ER08K6	SCREW	1
CP2442	OIL	1
G00018	NITROGEN BOTTLE	1
LOCTITE 242	THREADLOCKING ADHESIVE - MEDIUM STRENGHT	1
LOCTITE 7649	SOLVENT-BASED PRIMER	1
MAJUN	CLOTH - CLEAN, DRY, LINT-FREE, WHITE, COTTON	1
MJOILII	OIL	1
NASM20995	OXIDIZED INCONNEL LOCKWIRE DIA 0.032 INCH	1
SKYDROL PE-5	FLUID - HYDRAULIC, EROSION ARRESTING, FIRE RESISTANT	1

ACCOMPLISHMENT

NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
1	**APPLICABLE FOR: MLI, ETJ and GEF - ALL** The 36 Hour Check (elapsed clock hours – ECH) or known as Daily Check is more comprehensive than the Preflight Check. The Daily Check shall be performed once per calendar day (Accomplishment not to exceed 36 calendar hours).	A/P	

START TIME(UTC)	FINISH TIME(UTC)	TOTAL MAN HOUR		DEFECT FOUND Y N M.D.R.R. No:
		EST.	ACTUAL	
		2.75		

TASK CARD RELEASE

DATE (UTC) : ..... TIME (UTC) : ..... SIGNATURE : ..... AUTHORIZATION NO. : .....

BARCODE:

B789-05-INT-02-01-IDN

TASKCARD

A/C TYPE	Effectivity	DESCRIPTION							WORK ORDER NO.	
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A/C REG.	A/C MSN.	ACCESS							TASKCARD NO.	
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OPERATOR	PLACE	ZONE							TASK	REVISION
		211 632	212 713	414 734	424 744	531	532	631	DET	39
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		☐ ETOPS ☐ RVSM ☐ RNP10 ☐ RII ☐ CDCCL							05	A/P

REFERENCE			
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SRM 51-10-06	TAP TEST INSPECTION	AMM 27-41-71-700-801	Stabilizer Trim Motor Test
AMM 30-41-41	Window/Pitot Heat Test	AMM 32-42-00-720-801	ANTISKID/AUTOBRAKE CONTROL UNIT OPERATIONAL TEST
AMM 34-45-00-710-801	TCAS - OPERATIONAL TEST	AMM 34-43-00-860-145-002	WEATHER RADAR OPERATIONAL TEST
AMM 73-21-00-740-803-F00	EEC BITE TEST - RECENT FAULTS	EA B737NG-EA-11-001R11	Installation Tire Pacard Servicing
EI GEN-E-35-083R1	LIMITATION PRESSURE OF PORTABLE OXYGEN BOTTLE MUST BE REMOVE FROM AIRCRAFT	FIM 31-62 TASK 801	BITE Procedure
FIM 32-09 TASK 801	PSEU BITE TEST	AMM 24-22-00-860-811	Supply Electrical Power
AMM 23-51-00-710-801	Flight Interphone System - Operational Test	AMM 23-12-00-710-801	VHF Communication System Operational Test
AMM 23-11-00-710-801	HF Comunication System Operational Test	AMM 12-13-31-200-801	APU Oil Level Inspection
AMM 12-11-00-680-801	Fuel System Sumping	AMM 24-34-00-710-801	The Operational Test of the Standby Power System

TOOLS REQUIRED		
PART NUMBER	**DESCRIPTION**	**QUANTITY**
06-5022-6800	FLUID SERVICING UNIT, CAPACITY 2 GALLON, MAX. PRESS. 200 PSI FOR MINERAL BASE	1
1001089-1	COVER - PROBE, PITOT STATIC	1
14-6806-6011	TYRE PRESSURE GAUGE	1
3001331-1	TAP TEST TOOL	1
C32026-6	EQUIPMENT - DOWNLOCK, NLG AND MLG (FLYAWAY KIT)	1
F70199-1	TIRE INFLATION TOOL	1
F80201-1	DRAIN TL	1
H3310	HEADPHONE - 600 OHM, WITH 1/4 INCH MONO RCAAUDIO PLUG / GROUND HEAD SET	1
SPL-1880	EQUIPMENT - DOWNLOCK, NLG AND MLG (FLYAWAY KIT)	1
STD-1056	CONTAINER - SOLVENT RESISTANT, 5 GALLON (19 LITERS)	1

BARCODE:

B789-05-INT-02-01-IDN

TASKCARD

WORK ORDER NO.	A/C REG.	A/C MSN.	A/C Effectivity	OPERATOR	TASK CARD NO.
					B789-05-INT-02-01-IDN

MATERIAL REQUIRED		
PART NUMBER	**DESCRIPTION**	**QUANTITY**
141A6076-3	PIN - GUIDE ARM, FORWARD ENTRY DOOR	1
141A6077-2	WASHER	1
44250	HIGH TEMP 44 MARKER / MEDIUM BLACK (DYKEM®)	1
AEROSHELL GREASE 33	GREASE CLAY SAE AMS 3052, BMS3-33 TYPE I (03GBC1)	1
ALCOHOL-70	ISOPROPYL ALCOHOL, REAGENT GRADE	1
BACS12ER08K6	SCREW	1
BMS3-32TYII	FLUID - LANDING GEAR SHOCK STRUT	1
CP2442	OIL	1
G00018	NITROGEN BOTTLE	1
LOCTITE 242	THREADLOCKING ADHESIVE - MEDIUM STRENGHT	1
LOCTITE 7649	SOLVENT-BASED PRIMER	1
MAJUN	CLOTH - CLEAN, DRY, LINT-FREE, WHITE, COTTON	1
MJOILII	OIL	1
NASM20995	OXIDIZED INCONNEL LOCKWIRE DIA 0.032 INCH	1
SKYDROL PE-5	FLUID - HYDRAULIC, EROSION ARRESTING, FIRE RESISTANT	1

ACCOMPLISHMENT

NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
1	**INTERVAL NOTE :** The 36 Hour Check (elapsed clock hours – ECH) or known as Daily Check is more comprehensive than the Preflight Check. The Daily Check shall be performed once per calendar day (Accomplishment not to exceed 36 calendar hours) **APPLICABLE FOR: MLI, ETJ and GEF - ALL** **General** A. Accomplishment of the daily check must be performed by the approved authorization holder or by person who work under his direct supervision. The Daily Check inspection must be released by a person who appointed and approved by Quality Department or a holder of higher degree/ authorization/license B. This task contains with the aircraft security inspections. If object(s) or material (powdery, gaseous or liquid) from un-identified origin is/are found, keep off the object(s) or material and immediately report to MCC In-Charge before taking any necessary action C. If Aircraft in maintenance visit (C-Check) and aircraft in storage condition is not necessary to do the Daily Check, but Storage (Prolong Inspection) program must be performed. If Aircraft Back to a Serviceable Condition after storage, Daily Check must be performed and attached to After Storage Tasks		

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TASKCARD

WORK ORDER NO.	A/C REG.	A/C MSN.	A/C Effectivity	OPERATOR	TASK CARD NO.
					B789-05-INT-02-01-IDN

ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	(B789-10-INT-01-21-IDN, B789-10-INT-01-21-GEF-IDN, B789-10-INT-01-21-ETJ-IDN).		
2	**JOB SET-UP :** (1) Ensure that aircraft arrival and parking areas are cleared of debris and obstructions.	A/P	
3	(2) Supply electrical power source 115 V (+5 V), 400 Hz (+20 Hz). Do this task: Supply Electrical Power AMM TASK 24-22-00-860-811. (if required). Source From APU [.....] or GPU [.....]	A/P	
4	(3) Install nose and main landing gear ground lock pins. Do this task : Landing Gear Downlock Pins Installation. AMM TASK 32-00-01-480-801	A/P	
5	**PROCEDURE.** **A. Aircraft Maintenance Logs**. (1) Review AFML, CML, NSRDI, DBC, DBCS	A/P	
6	(2) Review deffered maintenance items, log and correct as necessary	A/P	
7	**B. Fuselage Inspections** (1) Check fuselage access / service panels, cabin windows, hatches, navigation / communication antennas and radome from ground level for obvious damage and security. Verify that cargo door latches are fully engaged. Masts and drains for fluid leakage	A/P	
8	(2) FWD Cargo Compartment & AFT Cargo Compartment Do general visual inspection of the fwd and aft cargo compartment: - Inspect (General Visual) the forward & aft cargo compartment floor, ceiling, sidewall, bulkhead, and blowout (pressure relief) panels/liners for holes/tears, condition, and security. - Inspect (General Visual) Forward & Aft Cargo compartment Cargo restraint nets, cargo lock and attachment points including light for condition.	A/P	
9	(3) Check (Crew) oxygen blowout disk (green) for security.	A/P	
10	(4) Perform walk-around visual check of airframe, wings and tail for condition, security and evidence of fluid leaks	A/P	
11	(5) Check landing gear, gear doors, wheel-wells and installations for general condition and security including the following:	A/P	

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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
12	a) Check nose and main landing gear struts for fluid leaks and proper height.	A/P	
13	b) Check tires and wheels for condition and hub caps for security.	A/P	
14	c) Check hydraulic reservoir fluid quantity and accumulator pressures. Serviceas required. And Check hydraulic components for condition, security and signs of fluid leakage	A/P	
15	(6) Check pitot tubes and static ports for damage or obstructions	A/P	
16	(7) Check static dischargers for condition and security	A/P	
17	(8) Check exterior lights and lens for condition, security, or burned out bulbs	A/P	
18	(9) Check all external required placards. Replace or install any damaged ormissing required placards.	A/P	
19	(10) 1. Extend LH & RH trailing edge flaps to the 40-unit position. To extend them, do this task: Extend the Trailing Edge Flaps,Ref. AMM TASK No. 27-51-00-860-803. 2. Deactivated Trailing Edge Flap Sys Ref. AMM TASK No. 27-51-00-040-801. 3. Do Detailed Visual Inspection at Upper and Lower Skin of WingTE Inboard Main Flap Wedge for sign of deterioration, delamination. Do DVI and continue to perform Tap test Inspection refer to SRM CHAP 51-10-06 at upper and lower skin of Wing Flap area (Inb TE Main Flap/Outboard TE Flap, Flap Fairing Support) for an identified sign of deterioration, delamination or disbond. For the Initial temporary repair, Tap Test inspection is a must to perform to measure for the Extend of the Damage Area. RESULT : Good Condition/ initial Temporary repair/Found Damage …… P/N & S/N Part Damage …… Damage Measurement …… *Provide the Detail Photo	A/P	

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WORK ORDER NO.	A/C REG.	A/C MSN.	A/C Effectivity	OPERATOR	TASK CARD NO.
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ACCOMPLISHMENT			
NO.	**INSTRUCTION**	**PERFORMED BY**	**INSPECTED BY**
	Note: For Disbond damage detected, please marking the disbond area at the TE Flap/Outboard Flap and Flap Fairing support. There are specific SRM Allowable Damage Limit for evaluation of damage and temporary repair for every Part/Area. **SRM 57-53-01 WING TRAILING EDGE FLAP SKIN Allowable Damage 1-5** **SRM 57-53-02 WING TRAILING EDGE INBOARD MAIN FLAP STRUCTURE** 4.After performing the inspection report the result to Structure Engineering with the following report method: - Scan the QR Code below and fill in the form SHM- FLAP INSPECTION		
20	**C. Flight compartment Inspections.** (1) Do the following tasks: Do a “Crew Oxygen Cylinder Dispatch Pressure Check" (AMM 12-15-21-210-801). Make sure the pressure shown on the pressure gage is above the minimum pressure necessary for dispatch. See the table below for minimum dispatch pressure.	A/P	

C.O.B PRESSURE (PSI)

REQUIRED PRESSURE (PSI) FOR 114/115 CUBIC FOOT CYLINDER				
BOTTLE TEMPERATURE		NUMBER OF CREW USING OXYGEN		
Deg C	Deg F	2	3	4

BARCODE: B789-05-INT-02-01-IDN

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					B789-05-INT-02-01-IDN

ACCOMPLISHMENT

NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	50 122 530 735 945 45 113 520 725 930 40 104 510 715 915 35 95 505 700 900 30 86 495 690 885 25 77 485 680 870 20 68 480 670 860 15 59 470 655 840 10 50 460 645 830 5 41 455 635 815 0 32 445 620 800 -5 23 440 610 785 -10 14 430 600 770		
21	(2) Do an APU Oil Level Inspection by using APU BITE Procedure (AMM 12-13-31-200).	A/P	
22	(3) Record APU hours and cycles from APU DMM via CDU. **APU Hours** | **APU Cycles** |	A/P	
23	(4) Check all circuit breakers, switches, guards with safety wire and controls for normal position prior to leave the cockpit.	A/P	
24	(5) Check all cockpit lights	A/P	
25	(6) Check flight compartment and windows for condition and cleanliness Result: ..... Note: If window glass/acrylic found damage (Scratch, Chips, Delamination,Moisture Ingression, Bubbles, Haze, etc.), report to following WG hotline: - Hotline WG 1 (0812-8502-9831 / WG#1_lion@batamaerotechnic.com) - Hotline WG 2 (0812-8502-9821 / WG#2_lion@batamaerotechnic.com)	A/P	

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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	- Hotline WG 3 (0812-8502-9830 / WG#3_lion@batamaerotechnic.com) - Hotline WG 4 (0812-8502-9823 / WG#4_lion@batamaerotechnic.com)		
26	(7) Check IDG disconnect switch for condition and security (guarded and wired).	A/P	
27	(8) Check emergency, safety equipment and Medical / Doctor's Kit onboard for condition, security and validity	A/P	
28	(9) Check emergency exit windows and F/O emergency exit handles for condition and security	A/P	
29	(10) Operational test all exterior lights: a. Wing illumination lights (AMM 33-41-00-710). b. Fixed landing lights (AMM 33-42-01-960). c. Retractable landing lights (AMM 33-42-02-960). d. Winglet position lights (AMM 33-43-10-710). e. Winglet forward position lights (AMM 33-43-11-960). f. Winglet aft position lights (AMM 33-43-12-960). g. Runway turnoff lights (AMM 33-45-02-960). h. Logo lights (AMM 33-49-00-960). i. Slide/Over-wing lights (AMM 33-51-04-960).	A/P	
30	**(11) These tasks below require IRS alignment:** **1. Weather Radar (WXR) System - Operational Test** **2. TCAS - Operational Test** **TASK 34-43-00-710-803-002** **1. Weather Radar (WXR) System - Operational Test** SUBTASK 34-43-00-860-145-002 (1) Make sure the air data inertial reference unit (ADIRU) is aligned to the NAV mode. To align it, do this task: Air Data Inertial Reference System Alignment from the ISDU, TASK 34-21-00-820-802 or Air Data Inertial Reference System - Alignment from the FMC CDU, TASK 34-21-00-820-801. (a) Make sure the ATT flags on the captains and first officers Primary Flight Display (PFD) are not in view. SUBTASK 34-43-00-860-146-002 (2) Make sure that this circuit breaker is closed: **F/O Electrical System Panel, P6-1** **Row: D, Col: 13, Number: C00120, Name: WEATHER RADAR RT**	A/P	

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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	SUBTASK 34-43-00-860-147-002 (3) Set the switches on the captain's and the first officer's EFIS control panels as follows: (a) Set the left and right ADF/VOR control to OFF. (b) Set the mode selector to VOR. SUBTASK 34-43-00-710-004-002 (4) Select TEST on the WXR control panel. (a) Set the SOURCE switch on the instrument switching module to the ALL ON 1 (ALL ON 2) position. **WARNING: DO NOT OPERATE THE WEATHER RADAR WITHIN 3.3 FEET (1 METER) OF REFUELING OR DE-FUELING OPERATIONS, A FUEL SPILL OR OPEN FUEL CELLS. OPERATION OF THE WEATHER RADAR WITHIN THE 3.3 FEET (1 METER) LIMIT CAN CAUSE A FIRE OR EXPLOSION. A FIRE OR EXPLOSION CAN CAUSE INJURY OR DEATH TO PERSONNEL AND DAMAGE TO EQUIPMENT.** **WARNING: DO NOT OPERATE THE WEATHER RADAR WHEN PERSONNEL ARE IN THE AREA USUALLY ENCLOSED BY THE AIRCRAFT NOSE RADOME OR IN A HANGAR. THESE CONDITIONS CAN CAUSE INJURY TO PERSONNEL.** b) Push the WXR switch on the captain's (first officer's) EFIS control panel to the ON position. NOTE: The hangar restriction DOES NOT apply to the weather radar test mode. NOTE: The fuel spill or open fuel cell warning DOES apply to the weather radar test mode. 1) After 15 seconds: a) Make sure WXR TEST display on the captain's and first officer's display unit (DU) is as shown on Figure 501. 2) Set the SOURCE switch on the instrument switching module to the AUTO position.		
31	**TASK 34-45-00-710-801** **2. TCAS - Operational Test** **A. Prepare for the Operational Test** SUBTASK 34-45-00-860-001 (1) Do this task: Supply Electrical Power, TASK 24-22-00-860-811. SUBTASK 34-45-00-860-002 (2) Make sure these systems are serviceable: (a) Air Data System (TASK 34-21-00-710-801). (b) Air Traffic Control System (TASK 34-53-00-710-801). (c) Common Display System (TASK 31-62-00-710-801). (d) Flight Interphone System (TASK 23-51-00-710-801). (e) Low Range Radio Altimeter System (TASK 34-33-00-710-801). SUBTASK 34-45-00-860-003 (3) Do this task: Air Data Inertial Reference System - Alignment from the ISDU,TASK 34-21-00-820-802 or Air Data Inertial Reference System – Alignment from the FMC CDU, TASK 34-21-00-820-801.	AP	

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	SUBTASK 34-45-00-860-034 (4) Wait until the align light on the MSU goes off. SUBTASK 34-45-00-860-004 (5) Put the display mode switch on the EFIS control panel to the MAP position. SUBTASK 34-45-00-860-035 (6) Set the range knob on the EFIS control panel to 10. SUBTASK 34-45-00-860-005 (7) Put the mode select switch on the TCAS/ATC control panel to the TA/RA position. SUBTASK 34-45-00-860-006 (8) Push the TFC switch on the EFIS control panel. (a) Make sure the inboard display shows TFC. SUBTASK 34-45-00-860-007 (9) Put the transponder select switch on the TCAS/ATC control panel (referred to as the control panel for the rest of this section) to the 1 position. (a) Make sure the FAIL lamp on the control panel is off. SUBTASK 34-45-00-860-008 (10) Put the mode select switch on the control panel to the STBY position. (a) Make sure the inboard displays show the TCAS OFF indication. SUBTASK 34-45-00-860-009 (11) Put the mode select switch on the control panel to the TA position. (a) Make sure the inboard displays show the TA ONLY indication. SUBTASK 34-45-00-860-010 (12) Put the mode select switch on the control panel to the TA/RA position. (a) Make sure TA ONLY continues to show on the inboard displays. **B. Procedure** SUBTASK 34-45-00-740-001 (1) Do the operational test as follows: (a) Turn and release the mode select switch on the control panel to the TEST position and hold it for one second. (b) Make sure these results occur: 1) The inboard displays show a test pattern as described below (Figure 501): a) TCAS TEST shows on the left side of the inboard displays b) The word TRAFFIC shows on the right side of the inboard displays c) An R/A (red square) shows at 3 o'clock and flying level (no arrow). d) A Traffic Advisory (yellow circle) shows at 9 o'clock and climbing (up arrow) e) Proximity Traffic (solid white diamond) shows at 1 o'clock and descending (down arrow). f) Non-Threat Traffic (open white diamond) shows at 11 o'clock and flying level (no arrow). 2) Make sure the outboard displays show the DO NOT CLIMB and DO NOT DESCEND resolution advisory.		

BARCODE:

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TASKCARD

WORK ORDER NO.	A/C REG.	A/C MSN.	A/C Effectivity	OPERATOR	TASK CARD NO.
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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	3) A TCAS SYSTEM TEST OK synthesized voice announcement comes on at the end of the test if the test passes. SUBTASK 34-45-00-740-002 (2) Do the operational test again with the transponder select switch on the control panel set to the 2 position. SUBTASK 34-45-00-840-005 (3) Make sure you return the ATC/TCAS mode switch on the ATC/TCAS panel to STBY position. SUBTASK 34-45-00-860-011 (4)If electrical power is not necessary, do this task: Remove Electrical Power, TASK 24-22-00-860-812. **NOTE: PLEASE PUT IRS MODE SELECTOR TO OFF.**		
32	FIM 31-62 TASK 801 12. Do the BITE procedure for maintenance messages in CURRENT STATUS on the CDS: on DEU-1 CDS BITE ☐Checked on DEU-2 CDS BITE, ☐ Checked Do the BITE procedure for maintenance messages in INFLIGHT FAULTS STATUS on the CDS: on DEU-1 CDS BITE ☐ Checked on DEU-2 CDS BITE, ☐Checked RESULT ☐ NO FAULT MESSAGE ☐ FAULT MESSAGE APPEARED If there is fault recorded, find fault isolation task for the applicable maintenance message. write down fault message and rectification process on AFML page	A/P	

TASKCARD

WORK ORDER NO.	A/C REG.	A/C MSN.	A/C Effectivity	OPERATOR	TASK CARD NO.
					B789-05-INT-02-01-IDN

ACCOMPLISHMENT

NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
33	FIM 32-09 TASK 801 Do PSEU Bite Test refer to FIM 32-09 Task 801 (Latest Revision),to ensure no faults exist. If there is a fault recorded, find a fault isolation task for the applicable maintenance message. write down the fault message and rectification process on the AFML page. Fault Message Recorded: …….....	A/P	
34	**TASK 27-41-71-700-801** **13. Stabilizer Trim Motor Test** Perform Stabilizer Trim Motor Test. This Stabilizer Trim Motor Test reference include:	A/P	
35	(14) Check all circuit breakers, switches, guards and controls for normal position prior to leave the cockpit.	A/P	
36	(15) Do general visual inspection Captain & First Officer seat forcondition and security.	A/P	
37	**GRAY WATER / DRAIN SYSTEM CLEANING TO REMOVE BLOCKAGESFROM CLOGGED DRAIN LINES** Get access to Forward L/H and R/H Sill Drain Float Valve Assembly(See IPC Figure 1 ITEM 38-31-51-33A) Check air flows from the forward drain port. Parameter : OK / Blockages , Result :..... Please see pictures in the end of task card.	A/P	
38	**D. Passenger Cabin and Galley Inspection.** (1) Visually check that all emergency exit hatches are secured and that handlesare properly stowed.	A/P	

Reference	Title
22-11-81-710-801	Stabilizer Trim Motor Autopilot Test (P/B 501)
24-22-00-860-811	Supply Electrical Power (P/B 201)
29-11-00-860-801	Hydraulic System A or B Pressurization (P/B 201)
29-11-00-860-805	Hydraulic System A or B Power Removal (P/B 201)

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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
39	(2) Check passenger and emergency doors for condition and security. Please make sure forward entry doors radius link marked screw have no migration or the straight line mark on screw still inline to the washer. If found the stright line mark have migrated or the straight line is not inline, please replace the radius link pin assy as per AMM 52-11-21. Result:..... Note: 1. If found screw migrated, report to following WG hotline: - Hotline WG 1 (0812-8502-9831 / WG#1_lion@batamaerotechnic.com) - Hotline WG 2 (0812-8502-9821 / WG#2_lion@batamaerotechnic.com) - Hotline WG 3 (0812-8502-9830 / WG#3_lion@batamaerotechnic.com) - Hotline WG 4 (0812-8502-9823 / WG#4_lion@batamaerotechnic.com) 2. Please make sure all the STEP on AMM 52-11-21 is followed. Use PIN 141A6076-3, Screw BACS12ER08K6, Washer 141A6077-2 Loctite 7649, and Loctite 242. 3. Please make sure adhesive primer Loctite 7649 and compound Loctite 242 is applied correctly as per AMM 52-11-21. If the loctite is not present, please inform engineering/WG to be applied in Hub Station which material available. 4. When install new pin or found no marking (or marking not clear), put new straight-line marking on the screw using PN: 44250 (HIGH TEMP 44 MARKER) or other equivalent permanent marker.Let marker cure for 45 sec to 1 minute before operate the door.	A/P	
40	(3) Check aircraft emergency equipment for condition, proper pressure, safetiesand security, including the following:	A/P	

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NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
41	a) Flight Compartment	A/P	

Description	QTY	POS#	Available [YES/NO]	Valid Until
Halon Firext	1 ea	Cockpit		
PBE	1 ea	Cockpit		
Medical Kit	1 Box	Forward		

42	b) Door	A/P	

Description	QTY	POS#	Available [Yes/No]	Valid Until
Escape Slide	1 ea	Main Entry		
Escape Slide	1 ea	Service DoorFWD Service DoorAft R/H Service Door Aft L/H		
Escape Slide	1 ea	MID Door L/H MID Door R/H		

43	c) Passenger Cabin	A/P	

Description	QTY	POS#	Available [Yes/No]	Valid Until
Halon Firext	2 ea 3 ea	FWD AFT		
Oxygen Bottle 5LB	1 ea	FWD		
Oxygen Bottle 5LB	1 ea	FWD		
Oxygen Bottle 5LB	1 ea	MID R/H		
Oxygen Bottle 5LB	1 ea	AFT		
Oxygen Bottle 5LB	1 ea	AFT		

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NO.	INSTRUCTION					PERFORMED BY	INSPECTED BY
	PBE	1 ea 1 ea 4 ea	FWD FW D MID				
	H20 Firext	1 ea 1 ea	FWD AFT				
	First Aid Kit	1 ea 1 ea 1 ea	FWD AFT R/HAFT R/H				
44	d) Lavatory					A/P	
	Description	QTY	POS	Available [Yes/No]	Valid Until		
	Fire Extinguiser	1 ea 1 ea 1 ea	FWD AFT R/H AFT L/H				
45	e) Check portable oxygen bottles correct pressure (Min. 1000 psi), validity and security of installation. (refer to. GEN-EI-35-083R2). NOTE: 1. If the pressure of Portable Oxygen Bottle out of the allowed range (1000 psi to 1850 psi) it must be remove from aircraft and send to main store for refill. 2. If the Hard Time (HT) of the Portable Oxygen Bottle is reached,but the pressure still full, it must remain to remove from aircraft. 3.If the Orignin departure from base station (CGK) the pressure Portable Oxygen Bottle can be dispatch with 1500 PSI (the needle close to red band position) 4. If the fleet on the out station, and the pressure approching minimum allowed it must be replaced at nearest station which have replacement portable oxygen bottle facility.					A/P	
46	(4) Check all emergency exits for condition, security and proper operation					A/P	
47	(5) Check evacuation slides (passenger entrance and galley service doors) for security, proper installation, air bottle pressure, proper instruction placards installed and validity.					A/P	
48	(6) Check lavatories for general condition and operation. Lavatory water heaterfor proper operation. And waste bin door for proper operation					A/P	
49	(7) Check lavatory temperature indicators for condition. Replace lavatory fire extinguisher bottles, if temperature indicators change color (white to black)					A/P	

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ACCOMPLISHMENT			
NO.	**INSTRUCTION**	**PERFORMED BY**	**INSPECTED BY**
50	(8) LAVATORY DOOR HINGE PIN INSPECTION INSTRUCTION 1. Get access to all lavatory (FWD and AFT). 2. Visual Inspect the lavatory door hinge pin. As show in figure 5 and figure 6. 3. If there is finding, please put it on Cabin Maintenance Log.	A/P	
51	(9) Check First Aid Kit (FAK) for sealing, damage and validity (If installed).	A/P	
52	(10) Operational check of the Emergency Light (Ref. AMM 33-51-00-710). (a) Emergency Interior Light: • Dome light, • Aisle light, • Emergency aisle light (EAL), • Overdoor exit sign, • Overwing hatch handle lights, • Exit locator sign, • Exit indicator sign, • Floor proximity light (If apply), • Flood sign (If Apply) (b) Emergency Exterior Light: • Slide light • Wing Lights	A/P	
53	**E. Power Plant Inspection** (1) Check engine cowling, inlet cowls, nose domes and visible fan blades for obvious damage. Check access panels and chip detector/pressure relief door for condition and security. Check for open latches and obvious signs of fuel and hydraulic leaks. For Fan Blade Platform please make sure platform are all seated and not in loose condition	A/P	

Engine Position	Engine #1	Engine #2
Platform Finding (please state if there is finding or no, and state the platform position)		

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ACCOMPLISHMENT			
NO.	**INSTRUCTION**	**PERFORMED BY**	**INSPECTED BY**
	See video on this link bit.ly/4gVxDNQ For more information please download Engineering Information below SCAN ME		
54	(2) Check left and right engine oil quantity and service as required. a) Oil should be added not less than 5 minutes and no greater than 30 minutes after engine shutdown while the oil in the tank is still warm. This will prevent the over-servicing of the engine.	A/P	
55	b) Detailed Inspection of IDG, delta P indicator, oil level and service as require(MP Task 24-020-01 & 24-020-02 and 24-030-01 & 24-030-02) 1) - Inspect (Detail) the left & right IDG delta P indicator (AMM 12-13-21-200-802) - Inspect left & right IDG oil level, do this task: IDG Oil Level Check, (AMM TASK 12-13-21-200-801) 2) - To service IDG oil, do this task: IDG Servicing (Oil Fill), (AMM TASK 12-13-21-600-801)	A/P	
56	**F. Wheel-Well Inspections.** (a) Clean exposed surfaces of the nose, left and right main landing gear shockstrut (MP Task 32-010-01, 32-010-02 and 32-060-00)	A/P	

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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
57	(b) Inspect (General Visual) nose and main landing gear tires and wheels for condition and wear. (MP Task 32-360-00)	A/P	
58	(c) Visually check the left and right brake wear pins for minimum extension.(MP Task 32-270-01 and 32-270-02) **Inspect and record the brake wear pin on each brake.** NOTE **:** make sure to apply parking brake ON/SET prior to perform this task. **WARNING :** MAKE SURE THE DOWNLOCK PINS ARE INSTALLED ON ALL LANDING GEAR WITHOUT THE DOWNLOCK PINS, THE LANDING GEAR COULD RETRACT AND CAUSE INJURIES TO PERSONS AND DAMAGE TO EQUIPMENT. **CAUTION :** DO NOT TOW OR MOVE THE AIRPLANE DURING THE INSPECTION.	A/P	
59	(d) Measure the remaining indicator pin length for all brake assemblies and make record for the measurement : BRAKE WEAR PINS RECORD (in mm) #1 | #2 | #3 | #4 | | | |	A/P	

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ACCOMPLISHMENT

NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
60	(e) Check nose and main landing gear tires for proper inflation (MP 32-350-00). Service asrequired and record the pressure. (B737NG-EA-11-001R11) NOTE : a. Effective B737-800 1. Tire MLG Pressure Limit (MIN – MAX) : PR 28 : 200-210 Psig. 2. Tire MLG Pressure Limit (MIN – MAX) : PR 30 : 200-210 Psig. 3. Tire NLG Pressure Limit (MIN – MAX) : 200-210 Psig. b. Effective B737-900 ER 1. Tire MLG Pressure Limit (MIN – MAX) : 30 PR : 220-225 Psig. 2. Tire MLG Pressure Limit (MIN – MAX) : 28 PR : 220-223 Psig.	A/P	

NOSE TIRES PRESSURE			
LEFT		RIGHT	
Before Top-Up	After Top-Up	Before Top-Up	After Top-Up

MAIN TIRES PRESSURE			
LH #1		LH #2	
Before Top-Up	After Top-Up	Before Top-Up	After Top-Up
RH #1		RH #2	
Before Top-Up	After Top-Up	Before Top-Up	After Top-Up

61	(1)Procedures. a) To servicing the tires, do this task: Add Nitrogen or Air to the Tire,AMM TASK 12-15-51-610-802	A/P	
62	b) Check torsion link cotter pin and nut for condition	A/P	

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NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
63	**G. Wings and Body Fairing Inspections.** (1) Drain fuel from sump drain to prevent water contamination (AMM 12-11-00-680)	A/P	
64	**H. Window/Pitot Heat Module (P5-9) Installation Test** **SUBTASK 30-41-41-860-002** **Please make sure pitot cover is removed before testing pitot heater.** (1) Remove the safety tags and close these circuit breakers:	A/P	

CAPT	Electrical	SystemPanel	P18-3 Row Col Number Name
C	1	C00523	HEATERS CAPT PITOT
C	2	C00238	HEATERS TEMP PROBE
C	3	C01072	HEATERS ALPHA VANE LEFT
C	4	C00236	HEATERS ELEV PITOT LEFT

CAPT Electrical System Panel, P18-3
MLI 001, 002, 005-096, 101-131, 134-141, 143-171, ETJ 210, GEF 950-953

Row	Col	Number	Name
D	1	C00226	WINDOW HEAT CONTROL RIGHT FRONT AC
D	2	C00225	WINDOW HEAT CONTROL LEFT SIDE AC

MLI ALL, ETJ ALL, GEF ALL

Row	Col	Number	Name
D	3	C01071	HEATERS ALPHA VANE RIGHT
D	4	C00237	HEATERS ELEV PITOT RIGHT
D	5	C00525	HEATERS F/O PITOT
D	6	C00524	HEATERS AUX PITOT

MLI 001, 002, 005-096, 101-131, 134-141, 143-171, ETJ 201, GEF 950-953

Row	Col	Number	Name
E	1	C00224	WINDOW HEAT CONTROL LEFT FRONT AC
E	2	C00227	WINDOW HEAT CONTROL RIGHT SIDE AC

F/O Electrical System Panel, P6-3
MLI ALL, ETJ ALL, GEF ALL

Row	Col	Number	Name
F	16	C00570	PROBE INDICATION F/O
F	18	C00569	PROBE INDICATION CAPT

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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	SUBTASK 30-41-41-760-001 (2) On the captain's main instrument panel, P1, set the switch for the master dim and test lights to TEST. (a) Make sure all the lights on the window/pitot heat module [1] come on. SUBTASK 30-41-41-860-009 (3) On the captain's main instrument panel, P1, set the switch for the master dim and test lights to BRT. SUBTASK 30-41-41-710-001 (4) Put the PITOT & PROBE HEAT switches to the ON position 2-5 minute. NOTE: It can take up to 15 seconds for the light to come on. (a) Make sure the pitot heat indications on the module go off. **WARNING:** DO NOT TOUCH THE PROBES WHEN YOU CHECK FOR HEAT. THE PROBE TEMPERATURES CAN REACH 800 DEGREES F (427 DEGREES C) IN STILL AIR. (b) Make sure the pitot probes get hot. NOTE: You can check for heat by spraying the probe with water and watching for steam. **MLI 001, 002, 006, 008-038, ETJ 210, GEF 950-953, ETJ 201 PRE SB 737-30A1063** SUBTASK 30-41-41-860-003 (5) Put the PITOT & PROBE HEAT switches to the OFF position.☐ **MLI 005, 007, 039-099, 101-131, 134-141, 143-184, 301-999; MLI 001, 002, 006, 008-038, ETJ 201 POST SB 737-30A1063** SUBTASK 30-41-41-860-006 (6) Put the PITOT & PROBE HEAT switches to the AUTO position. **MLI ALL, GEF ALL, ETJ ALL** SUBTASK 30-41-41-860-007 (7) Make sure the pitot heat indications on the module come on. SUBTASK 30-41-41-860-004 (8) Put the WINDOW HEAT switches to the ON position. NOTE: Do not power test when all ON lights are illuminated SUBTASK 30-41-41-710-002 (9) Select the OVERHEAT position on the POWER AND OVERHEAT TEST switch, S7. (a) Make sure the OVERHEAT lights come on. SUBTASK 30-41-41-860-005 (10) Set the WINDOW HEAT and POWER AND OVERHEAT TEST switches to the off position.		

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NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
65	Note: Performing the inspection using a personal flashlight to enhance visibility during the inspection, particularly in low-light conditions (11) Install all cover pitot probe, static port. Pitot and static port covers must be installed during **Aircraft RON/AOG**. Check the condition of these covers before installation.	A/P	
66	**TASK 32-42-00-720-801** **I. Antiskid/Autobrake Control Unit Operational Test** **A. Prepare for the Test** **WARNING: OBEY THE PROCEDURE FOR THE INSTALLATION OF THE DOWNLOCK PINS. IF YOU MOVE THE CONTROL LEVER FOR THE LANDING GEAR TO THE UP POSITION, THE LANDING GEAR CAN RETRACT. THIS CAN CAUSE INJURIES TO PERSONNEL, AND DAMAGE TO EQUIPMENT.** (1) If the downlock pins are not installed in the nose and main landing gear,do this task: Landing Gear Downlock Pins Installation, TASK 32-00-01-480-801. (2) Make sure that the tires have chocks installed around them (CHOCK INSTALLATION, PAGEBLOCK 10-11-05/201). (3) Release the parking brake. **WARNING: KEEP PERSONNEL AND EQUIPMENT AWAY FROM THE FLIGHT CONTROL SURFACES, THE THRUST REVERSERS, AND THE LANDING GEAR. THESE COMPONENTS CAN MOVE SUDDENLY WHEN YOU SUPPLY HYDRAULIC POWER. THIS CAN CAUSE INJURIES TO PERSONNEL ANDDAMAGE TO EQUIPMENT.** (4) For the A and B hydraulic systems, do this task: Hydraulic System A orB Pressurization, TASK 29-11-00-860-801. (5) Make sure that the two thrust levers are in the IDLE position. (6) Make sure that the landing gear control lever is in the DN position. (7) Make sure that the spoilers are stowed. (8) Make sure that the Right and Left ADIRSs are in the NAV mode. (a) Align the ADIRS. To do this, do this task: Air Data Inertial Reference System - Alignment from the ISDU, TASK 34-21-00-820-802 or Air Data Inertial Reference System Alignment from the FMC CDU, TASK 34-21-00-820-801.	A/P	

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NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	(9) Make sure that the BRAKE TEST selecto switch on the Antiskid/Autobrake Control Unit is in the NORM position. **B. Antiskid/Autobrake Control Unit Operational Test** (1) Do the steps that follow for Antiskid/Autobrake Control Unit: (a) Push and then release the RESET switch on the Antiskid/Autobrake Control Unit. (b) Make sure that the display on the front panel of the Antiskid/Autobrake Control Unit shows MEM CLR and then clears. (c) Set the AUTOBRAKE selector switch on the P2-2 panel to "OFF" then to position "1". NOTE: If the AUTOBRAKE DISARM light is on when you perform steps (d), (e) and (f), the test will not successfully complete and one or more faults will be displayed on the front panel of the Antiskid/Autobrake Control Unit. (d) Push and hold the ENABLE/VERIFY switch on the Antiskid/Autobrake Control Unit. (e) While you hold the ENABLE/VERIFY switch, push and hold the VERIFY switch on the Antiskid/Autobrake Control Unit. (f) Release the ENABLE/VERIFY switch and the VERIFY switch at the same time. NOTE: The Antiskid/Autobrake Control Unit display will flash WAIT during the test. (g) Make sure that you see the indications that follow: 1) Make sure that the antiskid control unit display shows TEST END after 10 - 20 seconds. 2) Make sure that the ANTISKID INOP light on the P2-2 panel flashes ON/OFF twice. 3) Make sure that the AUTOBRAKE DISARM light on the P2-2 panel is ON, then OFF, and then remains ON at the end of the test. (h) Set the autobrake selector switch to OFF. (2) For the A and B hydraulic systems, do this task: Hydraulic System A or B Power Removal, TASK 29-11-00-860-805. (3) Do this task: Air Data Inertial Reference System - Deactivation, TASK 34-21-00-040-801. (4) Set the parking brake.		
67	**J. HF Communication System- Operational Test** **Perform HF Communication System Operational Test** This HF Communication System Operational Test include: Reference: 23-11-00-710-801 — Title: HF Communication System Operational Test	A/P	
68	**K. VHF Communication System- System Test Perform VHF Communication System Test** This VHF Communication System Test include: Reference: 23-12-00-730-801 — Title: VHF Communication System - System Test	A/P	
69	**L. Flight Interphone System - Operational Test Perform Flight Interphone System Operational Test**	A/P	

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ACCOMPLISHMENT

NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	This Flight Interphone System - Operational Test **Reference** / **Title** 24-22-00-860-811 — Supply Electrical Power (P/B 201) 24-22-00-860-812 — Remove Electrical Power (P/B 201) 35-12-85-910-801 — Crew Oxigen Mask Stowage (P/B 201)		
70	**M. Additional Task** 1) Operational check of the standby power control unit (SPCU). (MP Task 24-080-00).	A/P	
71	2) Interrogate the FMC CDU for left engine faults. AMM 73-21-00-740 (MP TASK 73-020-01) Result..... Note: “If found a fault message, please info to WG and do thetroubleshooting reff to FIM”	A/P	
72	3) Interrogate the FMC CDU for right engine faults. AMM 73-21-00-740 (MP TASK 73-020-02) Result..... Note: “If found a fault message, please info to WG and do thetroubleshooting reff to FIM”	A/P	
73	Perform “EAU Bite Test Reff. FIM 78-31 TASK 801. Result..... Note: “If found a fault message, please info to WG and do thetroubleshooting reff to FIM”	A/P	
74	**N. Final Check Inspections.** (1) Make necessary entries for work accomplished and sign off airworthiness release in the aircraft maintenance log.	A/P	
75	**JOB CLOSE-UP INSPECTION** Removed external power source 115 V (+5 V), 400 Hz (+20 Hz) if no longer required. Do this task: Remove External Power AMM TASK 24-22-00-860-814.	A/P	
76	..... END OF TASK		

START TIME(UTC)	FINISH TIME(UTC)	TOTAL MAN HOUR		DEFECT FOUND ☐ Y ☐ N M.D.R.R. No:
		EST.	ACTUAL	
		2.75		

TASK CARD RELEASE

DATE (UTC) : ..... TIME (UTC) : ..... SIGNATURE : ..... AUTHORIZATION NO. : .....

BARCODE:

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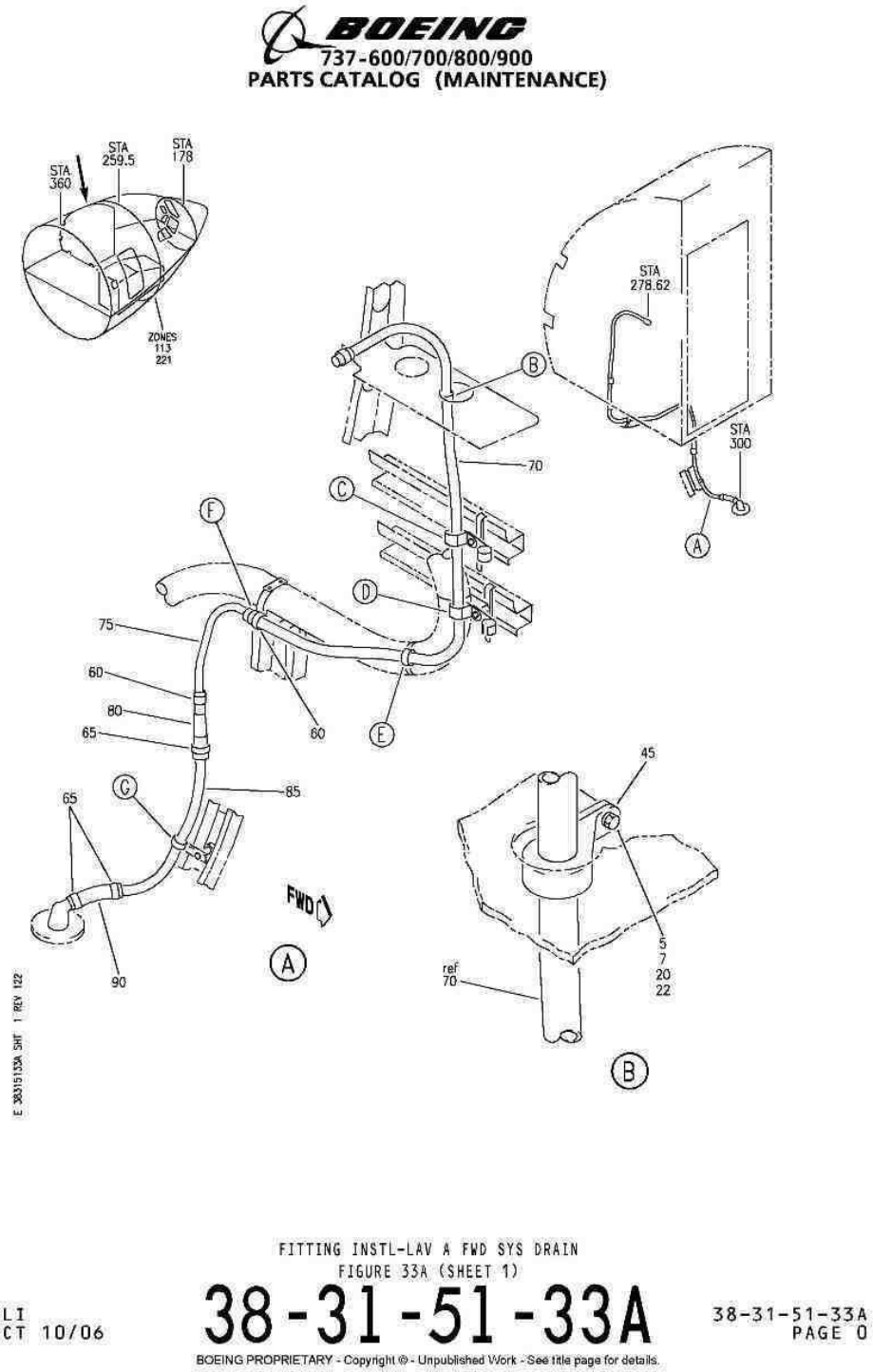
APPENDIX

GRAPHIC
FIGURE 1 IPC ITEM 38-31-51-33A.JPG



GRAPHIC
FIGURE 2. SILL DRAIN AT THE FORWARD ENTRY DOOR.JPG

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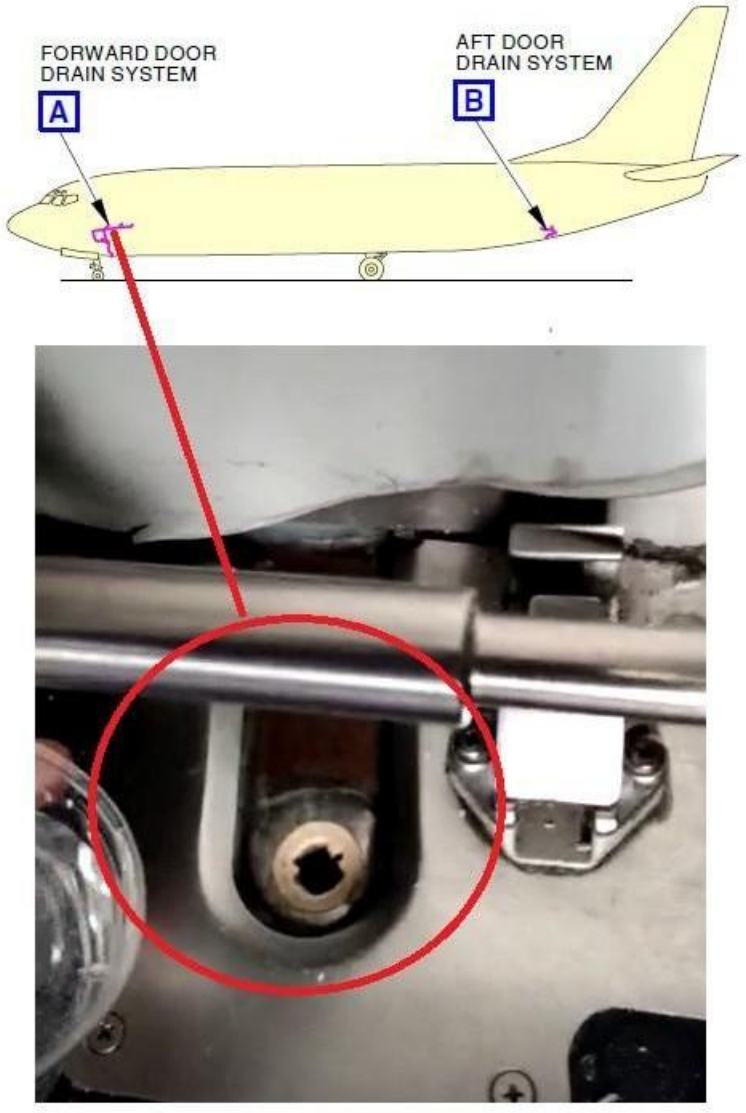
APPENDIX



GRAPHIC
FIGURE 3. WATER DRAIN SYSTEM.JPG

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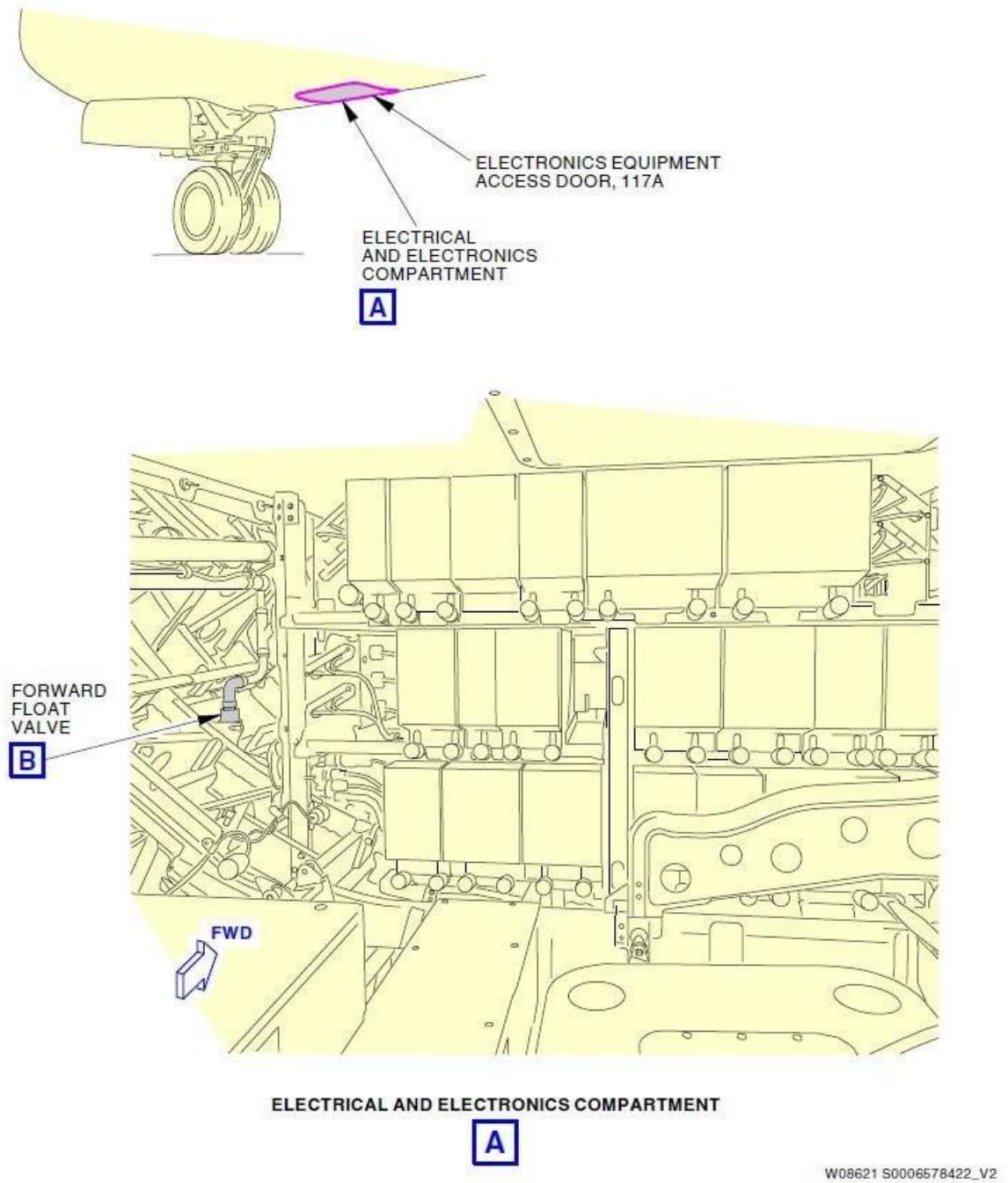
APPENDIX



GRAPHIC
FIGURE 4.JPG

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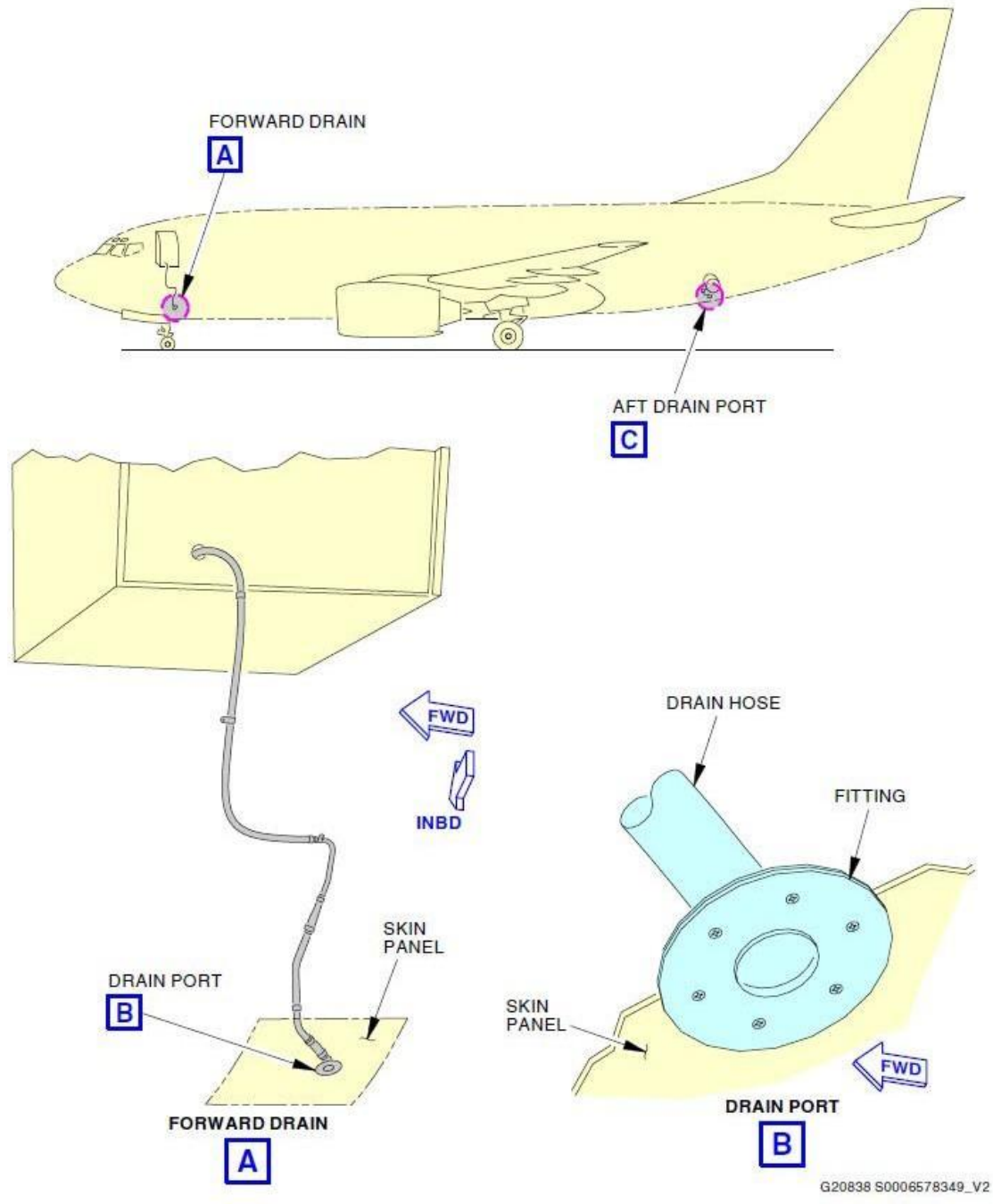
APPENDIX



Water Drain System

GRAPHIC
FIGURE 5.JPEG

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TASKCARD

WORK ORDER NO.	A/C REG.	A/C MSN.	A/C Effectivity	OPERATOR	TASK CARD NO.
					B789-05-INT-02-01-IDN

APPENDIX

FIGURE 2

GRAPHIC
FIGURE 6.JPEG

BARCODE:

B789-05-INT-02-01-IDN

TASKCARD

WORK ORDER NO.	A/C REG.	A/C MSN.	A/C Effectivity	OPERATOR	TASK CARD NO.
					B789-05-INT-02-01-IDN

APPENDIX

FIGURE 1

BARCODE:

Task Card P/N Pre-Draw Print

Task Card: B789-05-INT-02-01-IDN 737NG - DAILY CHECK.

P/N	P/N Description	P/N Category	QTY	Spare	Reserve
06-5022-6800	FLUID SERVICING UNIT, CAPACITY 2 GALLON, MAX. P	TOOLS	1	☐	☑
1001089-1	COVER - PROBE, PITOT STATIC	TOOLS	1	☐	☑
14-6806-6011	TYRE PRESSURE GAUGE	TOOLS	1	☐	☑
141A6076-3	PIN - GUIDE ARM, FORWARD ENTRY DOOR	ACCON	1	☑	☐
141A6077-2	WASHER	ACCON	1	☑	☐
3001331-1	TAP TEST TOOL	TOOLS	1	☐	☑
44250	HIGH TEMP 44 MARKER / MEDIUM BLACK (DYKEM®)	ACCON	1	☑	☐
AEROSHELL GREASE 33	GREASE CLAY SAE AMS 3052, BMS3-33 TYPE I (03GBC1)	ACCON	1	☑	☐
ALCOHOL-70	ISOPROPYL ALCOHOL, REAGENT GRADE	ACCON	1	☑	☐
BACS12ER08K6	SCREW	ACCON	1	☑	☐
BMS3-32TYII	FLUID - LANDING GEAR SHOCK STRUT	ACCON	1	☑	☐
C32026-6	EQUIPMENT - DOWNLOCK, NLG AND MLG (FLYAWAY KIT)	TOOLS	1	☐	☑
CP2442	OIL	ACCON	1	☑	☐
F70199-1	TIRE INFLATION TOOL	TOOLS	1	☐	☑
F80201-1	DRAIN TL	TOOLS	1	☐	☑
G00018	NITROGEN BOTTLE	ACCON	1	☑	☐
H3310	HEADPHONE - 600 OHM, WITH 1/4 INCH MONO RCAAU	TOOLS	1	☐	☑
LOCTITE 242	THREADLOCKING ADHESIVE - MEDIUM STRENGHT	ACCON	1	☑	☐
LOCTITE 7649	SOLVENT-BASED PRIMER	ACCON	1	☑	☐
MAJUN	CLOTH - CLEAN, DRY, LINT-FREE, WHITE, COTTON	ACCON	1	☑	☐
MJOILII	OIL	ACCON	1	☑	☐
NASM20995	OXIDIZED INCONNEL LOCKWIRE DIA 0.032 INCH	ACCON	1	☑	☐
SKYDROL PE-5	FLUID - HYDRAULIC, EROSION ARRESTING, FIRE RESISTANT	ACCON	1	☑	☑
SPL-1880	EQUIPMENT - DOWNLOCK, NLG AND MLG (FLYAWAY KIT)	TOOLS	1	☐	☑
STD-1056	CONTAINER - SOLVENT RESISTANT, 5 GALLON (19 LITERS)	COMM	1	☐	☑