Lion Group

TASKCARD

A/C TYPE	Effectivity	DESCRIPTION							WORK ORDER NO.	
		B737-800 - DAILY CHECK.								
A/C REG.	A/C MSN.	ACCESS							TASKCARD NO.	
									B789-05-999-02-01-BTK-IDN	
A/C TSN.	A/C CSN.	192G	413AL	414AR	423AL	424AR			THRESHOLD	INTERVAL
:									0	0
OPERATOR	PLACE	ZONE							TASK	REVISION
		211 632	212 713	414 734	424 744	531	532	631	DET	22
START DATE	FINISH DATE	NOTE							ATA	SKILL
		☐ ETOPS ☐ RVSM ☐ RNP10 ☐ RII ☐ CDCCL							05-20	

REFERENCE			
Doc No.	**Doc Description**	**Doc No.**	**Doc Description**
AMM 12-11-00-680-801	Fuel System Sumping	AMM 12-13-21-200-801	IDG Oil Level Check
AMM 12-13-21-600-801	IDG Servicing (Oil Fill)	AMM 12-15-21-210-801	Crew Oxygen Cylinder Dispatch Pressure Check
AMM 12-13-31-200-801	APU Oil Level Inspection	AMM 12-15-51-610-802	Add Nitrogen or Air to the Tire
AMM 24-22-00-860-811	Supply Electrical Power	AMM 24-22-00-860-814	Remove External Power
AMM 24-34-00-710-801	The Operational Test of the Standby Power System	AMM 28-10-00-280-021	Collect the Fuel Sample
FIM 31-62 TASK 801	BITE Procedure	AMM 32-00-01-480-801	Landing Gear Downlock Pins Installation
AMM 32-45-00-700-801	Wheels Fast Check (Wheel Installed on the Airplane)	AMM 32-45-00-700-803	Tires Inspection
EI B737NG-EI-34-226	pitot probe covers		

TOOLS REQUIRED		
PART NUMBER	**DESCRIPTION**	**QUANTITY**
1001089-1	COVER - PROBE, PITOT STATIC	1
14-6806-6011	TYRE PRESSURE GAUGE	1
F70199-1	TIRE INFLATION TOOL	1
F80201-1	DRAIN TL	1
H3310	HEADPHONE - 600 OHM, WITH 1/4 INCH MONO RCAAUDIO PLUG / GROUND HEAD SET	1
SPL-1880	EQUIPMENT - DOWNLOCK, NLG AND MLG (FLYAWAY KIT)	1
STD-1055	CONTAINER-OIL RESISTANT, 5 U.S-GAL (19 L)	1
06-5020-3600	DISPENSER - OIL SERVICING - 1 QUART OIL CONTAINER WITH COUPLINGS FOR ENGINE, APU, IDG AND CSD (OPTIONAL: 06-5022-6600)	1

MATERIAL REQUIRED		
PART NUMBER	**DESCRIPTION**	**QUANTITY**
ALCOHOL-70	ISOPROPYL ALCOHOL, REAGENT GRADE	1
BMS3-32TYII	MINERAL HYDRAULIC FLUID & ADDITIVES - DAMPER LANDING GEAR	1
CP2442	OIL	1
G00018	NITROGEN BOTTLE	1
LD4	FLUID - HYDRAULIC, EROSION ARRESTING, FIRE RESISTANT	1

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MATERIAL REQUIRED

PART NUMBER	DESCRIPTION	QUANTITY
MAJUN	CLOTH - CLEAN, DRY, LINT-FREE, WHITE, COTTON	1
MJOILII	OIL	1
NASM20995	OXIDIZED INCONNEL LOCKWIRE DIA 0.032 INCH	1
SKYDROL PE-5	FLUID - HYDRAULIC, EROSION ARRESTING, FIRE RESISTANT	1

ACCOMPLISHMENT

NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
4001	**INTERVAL NOTE:** Perform at interval 24 HR and not to exceed to 48 HR (Elapsed Clock Hours – ECH) **APPLICABLE FOR:** MLI, ILF, DEA and GEF - ALL		
4002	**B.0. JOB SET-UP:** B.1. Ensure that aircraft arrival and parking areas are cleared of debris and obstructions.	A/P	
4003	AMM TASK 32-00-01-480-801 B.2. Install nose and main landing gear ground downlock pins and install Wheel Chocks. Use downlock pin - Downlock, NLG and MLG, Part No.: C32026-15 with Alt. Part No.: C32026-1 or Part No.: C32026-6 or Equivalent WARNING: Before you go near the landing gears to install the CHOCK - WHEEL(S) , make sure that: · The aircraft is fully stopped · The beacon lights are off · All the engines are stopped	A/P	
4004	AMM TASK 30-31-00-730-801 B.3.a. Perform Pitot Probe, AOA Sensor, and TAT Probe Heater - System Test Note: On Procedure step 9, use (a) method “spray the air data sensors with water to check heat”. RESULT / FOUND: _____ (write: NORMAL or found any discrepancy example: “F/O PITOT U/S” then transfer to AFML for rectification) B.3.b. Perform normal parking maintenance procedures as follows: • The Total Air Temperature Probe cover,(on Cold condition)	A/P	

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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	• Pole - Pitot Static Probe Cover (on Cold condition) • The (AOA) sensor angle of attack vane cover or AOA vane protective cover		
4005	AMM TASK 24-22-00-860-811 B.4. Supply external electrical power as necessary	A/P	
4006	**EXTERNAL WALKAROUND CHECK** **C.0. FORWARD FUSELAGE** C.1. Make a general visual inspection as far as visible from the Ground of Forward Fuselage, with particular attention to any missing or loose fastener/rivets. Any related finding outside the AMM or SRM limit needs to be corrected before flight, including: - Passenger / Crew doors, cargo compartment door, - Service panel doors, avionic compartment access doors, - Pitot probes, static ports and AOA sensors: in Service (Protective covers removed), - Avionics equipment ventilation: Air inlet and air outlet valves, - Batteries ventilation outlet (Venturi): No obstruction, - Waste water drain mast, - Antennas: No damage, - Wing and engine scan lights: Cleanliness, secuirity & fluid leaks - Radome: No damage, latches engaged and locked, - Crew oxygen cylinder overpressure indicator: Green disc in place. - Wing-to-fuselage fairings and belly fairing - Anti-collision beacon lights - Ram air inlet, No obstruction	A/P	

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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	- Check exterior lights and lens for condition, security, or Checked burned out bulbs - Check exterior required markings and placards for proper installation and legibility - FWD Cargo - Inspect (General Visual) the cargo compartment floor, ceiling, sidewall, bulkhead, and blowout (pressure relief) panels/liners for holes/tears, condition, and security. - Visually examine the cargo restraint nets, lock & Light at the forward door. Examine the items that follow: (a) Make sure the straps are not worn or frayed. (b) Make sure that each strap attaches to an anchor point.		
4007	**D.0. NOSE LANDING GEAR AND DOORS** D.1. Do a general visual inspection of the nose landing gear and doors from ground as far as visible, including: - Doors and wheel well, - Gear assy structure: Damage, evidence of leakage, and proper height - Shock absorber: Signs of leakage and normal extension, - Shock absorber sliding tube, Wheel steering assemblies: Scoring, damage & cleanliness, NOTE : Please pay attention for the cleanliness of the Shock Absorber. Clean the tube from any dust, dirt, or other unwanted material that could potentially erode the dynamic seal and led to the hydraulic leakage.(if necessary using lint free cloth moist with mineral hydraulic fluid clean tube and dry with a clean cloth) - Clean exposed surfaces of the nose landing gear shock strut. (MP 32-060-00)	A/P	
4008	**E.0. CENTER FUSELAGE**	A/P	

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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	E.1. Do a general visual inspection of the forward center fuselage from ground as far as visible, including: - Emergency exits, - Service panel doors, - Wing-to-fuselage fairings and belly fairing, - Anti-collision beacon lights: Cleanliness, - Drain mast, - Ram air inlet flap, - Pack air intakes and outlets: No obstruction, - Antennas: No damage.		
4009	E.2. Check exterior lights and lens for condition, security, or burned out bulbs	A/P	
4010	E.3. Check all external required placards. Replace or install any damaged or missing required placards.	A/P	
4011	**F.0. WINGS** F.1. Do a general visual inspection of the RH and the LH wing from ground as far as visible, including: - Leading edge slats, for condition and security - Leading edge access panels, for condition and security - Surge tank air intake: No obstruction Surge tank overpressure - Navigation and strobe lights, - Static dischargers for condition and security - Control surfaces, flaps and flap track fairings: Damage, evidence of fluid leakage, - Lower wing surface: Damage, evidence of fuel leakage, - Landing light: No damage, cleanliness.	A/P	

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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
4012	AMM TASK **28-10-00-280-021** F.2. Collect the Fuel Sample Do these steps to collect a fuel sample: NOTE: Each fuel sample must be collected separately for each fuel tank and kept in separate sample containers. (a) Use the fuel sump drain to get a fuel sample: (b) Fill the container for the fuel sampling equipment, COM-1535 or equivalent, with approximately one quart (one liter) of fuel. (c) make sure the fuel sample has some visible water (free water) and some fuel (d) Pour the fuel/water from the container into the fuel sample bottle, STD-6365 or equivalent. NOTE: Do not add any additives, such as food coloring, to identify the presence of water. (e) Continue to collect fuel samples for the remaining tanks. Clean the fuel sampling equipment, COM-1535, again before you collect a new sample. Make sure to collect a fuel sample from each tank.	A/P	
4013	**G.0. POWERPLANT, PYLON & AFT FUSELAGE** G.1. Do a general visual inspection of the power plant and pylon from ground as far as visible, including (cowl doors closed): - Pylon with fairings and fillets - Engine air inlet - Fan blades and spinner For Fan Blade Platform please make sure platform are all seated and not in loose condition	A/P	

Engine Position	Engine #1	Engine #2

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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY

Platform Finding (please state if there is finding or no, and state the platform position)		

See video on this link

bit.ly/4gVxDNQ

For more information please download Engineering Information below

SCAN ME

- Fan cowl doors and thrust reverser cowl doors:

- Cowl doors closed and correctly latched,

- Blocker doors in stowed (closed) position,

- Access doors and pressure relief doors in place, closed and latched,

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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	- Air inlets/outlets: No obstruction, - Fan exhaust: Acoustic lining, exit vanes and struts, thrust reverser pivoting doors, - Turbine exhaust: Last stage LPT blades, nozzle and plug (for damage and metal deposit), - Drain mast and pylon drains: No obstruction, evidence of leakage. - AFT Cargo - Inspect (General Visual) the cargo compartment floor, Net, ceiling, sidewall, bulkhead, and blowout (pressure relief) panels/liners for holes/tears, condition, and security. - Visually examine the aft cargo restraint nets, lock & Light at the door. Examine the items that follow: (a) Make sure the straps are not worn or frayed. (b) Make sure that each strap attaches to an anchor point.		
4014	G.2. Check left and right engine oil quantity and service as required. A reinspection is required to check the oil caps for proper closing. After servicing, physically verify to ensure oil service cap is securely fastened. Service engine oil level to full. Ensure engine oil tank filler cap is properly secured after servicing. Verify all Servicing access doors are closed and securely fastened. And in the Aircraft Flight and Maintenance Log Book. LEFT ENGINE (Uplift:_____Qts) RIGHT ENGINE (Uplift:_____Qts)	A/P	
4015	AMM TASK 12-13-21-200-802, AMM TASK 12-13-21-200-801 & AMM TASK 12-13-21-600-801 b) Detailed Inspection of IDG, delta P Indicator, oil level and service as required (MP Task 24-020-01 & 24-020-02 and 24-030-01 & 24-030-02) IDG #1 Differential Pressure Indicator in : _____ Position If the DPI is in the up position, do actions refer to AMM Check the DPI resets decal (if installed) on the scavenge/charge filter cover for the number of DPI resets that has been done, and make logbook entry.	A/P	

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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	IDG #2 Differential Pressure Indicator in : _____ Position If the DPI is in the up position, do actions refer to AMM Check the DPI resets decal (if installed) on the scavenge/charge filter cover for the number of DPI resets that has been done, and make logbook entry.		
4016	AMM 05-41-03-210-801 & AMM 12-13-31-860-001 G3. Aft of the pressure bulkhead. a) Do a General Visual inspection of the area aft of the pressure bulkhead b) Check the APU oil level on the oil sight glass service as necessary record uplift in the Aircraft Flight and Maintenance Log Book	A/P	
4017	**H.0. MAIN LANDING GEAR / WHEEL-WELL** AMM TASK 32-00-10-100-802 H.1. Clean exposed surfaces of the left and right main landing gear shock strut. (MP TASKS 32-010-01 & 32-010-02)	A/P	
4018	AMM TASK 32-45-00-700-803 & 32-45-00-700-801 H.2. Inspect (General Visual) main landing gear tires and wheels for condition and wear. from ground as far as visible, (MP Task 32-360-00) including: - Check tires for wear, In Service - Check wheels, In Service Include: Rim damage, sheared/missing tie bolts, and hub caps for security	A/P	
4019	AMM TASK 32-41-41-700-801 H.3. Visually check the left and right brake wear pins for minimum extension. (MP Task 32-270-01 and 32-270-02) Inspect and record the brake wear pin on each brake, in service	A/P	
4020	H.4. Measure the remaining indicator pin length for all brake assemblies and make record for the measurement :	A/P	

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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	BRAKE WEAR PINS RECORD (in mm) #1 | #2 | #3 | #4 | | | |		
4021	H.5. - Cross check the control cabin HYD fluid QTY indication with indicators in the HYD system reservoirs and service as required. (AMM 12-12-00-610-801, HYD fluid spec BMS 3-11, Type IV). NOTE: If uplift is required record the uplift in the Aircraft Flight and Maintenance Log Book - Check hydraulic fluid quantity service as necessary - Check hydraulic components for condition, security and signs of fluid leakage.	A/P	
4022	**I.0. TIRES,** AMM TASK 12-15-51-780-801 (MP Task 32-350-00) I.1. Check NLG and MLG tires for correct pressure, and service as necessary. Record tire pressures in table below, and in the Aircraft Flight & Maintenance Log Book **TIRE PRESSURE RECORD**	A/P	

NOSE TIRES PRESSURE			
LEFT		RIGHT	
Before Top-Up	After Top-Up	Before Top-Up	After Top-Up

MAIN TIRES PRESSURE			
LH #1		LH #2	
Before Top-Up	After Top-Up	Before Top-Up	After Top-Up

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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	<table><tr><td></td><td></td><td></td><td></td></tr><tr><td colspan="2">RH #1</td><td colspan="2">RH #2</td></tr><tr><td>Before Top-Up</td><td>After Top-Up</td><td>Before Top-Up</td><td>After Top-Up</td></tr><tr><td></td><td></td><td></td><td></td></tr></table>		
4023	I.3. Check torsion link cotter pin and nut for condition	A/P	
4024	**J.0. FLIGHT COMPARTMENT** AMM TASK 12-15-21-210-801 J.1. Do a “Crew Oxygen Cylinder Dispatch Pressure Check" Make sure the pressure shown on the pressure gage is above the minimum pressure necessary for dispatch as per AMM. Record the C.O.B. Pressure on table below: <table><tr><td>C.O.B. PRESSURE (PSI)</td></tr><tr><td></td></tr></table>	A/P	
4025	AMM TASK 12-13-31-200-801 J.2. Do an APU Oil Level Inspection by using APU BITE Procedure.	A/P	
4026	J.3. Record APU hours, cycles and CT5ATP (EGT) from APU DMM via CDU <table><tr><th>APU Hours</th><th>APU Cycles</th><th>APU EGT (CT5ATP)</th></tr><tr><td></td><td></td><td></td></tr></table> Note: "If the CT5ATP reaches 1172^{O}F (633^{O}C) or higher, ETOPS operation is not allowed as stated in CMP Part D, Item 1a and/or 1b.	A/P	
4027	J.4. Check all cockpit lights and Spare bulb (completed it as necessary)	A/P	
4028	J.5.1. Check flight compartment for condition and cleanliness J.5.2. Check Flight Deck Window from damage description, as follow: AMM TASK 56-11-00-200-803 1. Moisture seal degradation	A/P	

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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	2. Moisture ingress into laminate and UV exposure 3. Interlayer degradation (discoloration, bubbling and cracking) 4. Delamination (separation interlayer from glass/plastic) 5. Bus bar degradation (discoloration and cracking) 6. Deterioration at the junction of the bus bar and conductive heating film 7. Electrical arcing at the junction of the bus bar and conductive heating film 8. Glass ply fracture, scratch and localized overheating condition J.5.3. Captain & First Officer Seat Track & Lock Mechanisme for wear, condition & security		
4029	J.6. Check IDG disconnect switch for condition and security (guarded and wired).	A/P	
4030	J.7. Check emergency, safety equipment and Medical / Doctor's Kit onboard for condition, security and validity Description / QTY / POS# / Available [YES/NO] / Valid Until Halon Firext — 1 ea — Cockpit PBE — 1 ea — Cockpit Medical Kit — 1 Box — Cockpit	A/P	
4031	J.8. Check emergency exit windows and F/O emergency exit handles for condition and security	A/P	
4032	J.9. Operational test all exterior lights: a. Wing illumination lights (AMM 33-41-00-710). b. Fixed landing lights (AMM 33-42-01-960). c. Retractable landing lights (AMM 33-42-02-960). d. Winglet position lights (AMM 33-43-10-710).	A/P	

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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	e. Winglet forward position lights (AMM 33-43-11-960). f. Winglet aft position lights (AMM 33-43-12-960). g. Runway turnoff lights (AMM 33-45-02-960). h. Logo lights (AMM 33-49-00-960). i. Slide/Over-wing lights (AMM 33-51-04-960).		
4033	J.10. Check all circuit breakers, switches, guards and controls for normal position prior to leave the cockpit.	A/P	
4034	AMM 24-34-00-710 J.11. Operational check of the standby power control unit (SPCU). MPD Task 24-080-00 for ALL MLI MPD Task 24-100-00 for ALL ILF, DEA, GEF	A/P	
4053	AMM 73-21-00-740 J12. Interrogate the FMC CDU for left & Right engine faults	A/P	
4035	FIM 31-62 TASK 801 J13. Do the BITE procedure for maintenance messages in CURRENT STATUS on the CDS: on DEU-1 CDS BITE on DEU-2 CDS BITE, Do the BITE procedure for maintenance messages in INFLIGHT FAULTS STATUS on the CDS: on DEU-1 CDS BITE on DEU-2 CDS BITE, RESULT ☐ NO FAULT MASSAGE ☐ FAULT MASSAGE APPEARED If there is fault recorded, find fault isolation task for the applicable maintenance message. write down fault message and rectification process on AFML page	A/P	
4036	**K.0. PASSENGER CABIN.**	A/P	

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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	K.1. Visually check that all emergency exit hatches are secured and that handles are properly stowed.		
4037	K.2.1. Check forward entry doors radius link marked screw have no migration or the straight line mark on screw still inline to the washer. If found the stright line mark have migration or the straight line is not inline, please replace the radius link PIN as per AMM 52-11-21. Result: ..... Note: 1. If found screw migrated, report to following WG hotline: - Hotline WG 6 (0812-8502-9832 / WG#6_btk@batamaerotechnic.com). 2. Please make sure all the STEP on AMM 52-11-21 is followed. Use PIN 141A6076-3, Screw BACS12ER08K6, Washer 141A6077-2, Loctite 7649, and Loctite 242. 3. Please make sure adhesive primer Loctite 7649 and compound Loctite 242 is applied correctly as per AMM 52-11-21. If the loctite is not present, please inform engineering/WG to be applied in Hub Station which material available. 4. When install new pin or found no marking (or marking not clear), put new marking on the screw using PN: 44250 (HIGH TEMP 44 MARKER) or other equivalent permanent marker. Let marker cure for 45 sec to 1 minute before operate the door. K.2.2. Check passenger and emergency doors for condition and security	AIRPL	
4038	K.3. Check aircraft emergency equipment for condition, proper pressure, safeties and security, including the following:	AIRPL	
4039	a) DOOR	A/P	

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ACCOMPLISHMENT

NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY

Description	QTY	POS#	Available [YES/NO]	Valid Until
Escape Slide	1 ea	Main Entry		
Escape Slide	1 ea	Service Door FWD Service Door Aft R/H Service Door Aft L/H		
Escape Slide	1 ea	MID Door L/H MID Door R/H		

4040 b) PASSENGER CABIN — A/P

Description	QTY	POS#	Available [Yes/No]	Valid Until
Halon Firext	1 ea 2 ea	FWD AFT		
Oxygen Bottle 5LB	1 ea	FWD		
Oxygen Bottle 5LB	1 ea	FWD		
Oxygen Bottle 5LB	1 ea	AFT		
Oxygen Bottle 5LB	1 ea	AFT		
Oxygen Bottle 5LB	1 ea	AFT		
PBE	1 ea 1 ea 2 ea	FWD FWD AFT		
H20 Firext	1 ea 1 ea	FWD AFT		
First Aid Kit	1 ea 1 ea 1 ea	FWD AFT L/H AFT R/H		

4041 c) LAVATORY — A/P

Description	QTY	POS	Available [YES/NO]	Valid Until
Fire Extinguiser	1 ea 1 ea 1 ea	FWD AFT R/H AFT L/H		

4042 d) Check portable oxygen bottles correct pressure (Min. 1000 psi), validity and security of installation. (refer to. GEN-EI-35-083). — A/P

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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	NOTE: 1. If the pressure of Portable Oxygen Bottle out of the allowed range (1000 psi to 1850 psi) it must be remove from aircraft and send to main store for refill. 2. If the Hard Time (HT) of the Portable Oxygen Bottle is reached, but the pressure still full, it must remain to remove from aircraft. 3. If the Origin departure from base station (CGK) the pressure Portable Oxygen Bottle can be dispatch with 1500 PSI (the needle close to red band position) 4. If the fleet on the out station, and the pressure approching minimum allowed it must be replaced at nearest station which have replacement portable oxygen bottle facility.		
4043	(e) Check all emergency exits for condition, security and proper operation	A/P	
4044	(f) Check evacuation slides (passenger entrance and galley service doors) for security, proper installation, air bottle pressure, proper instruction placards installed and validity.	A/P	
4045	(g) Check lavatories for general condition and operation. Lavatory water heater for proper operation. And waste bin door for proper operation	A/P	
4046	(h) Check lavatory temperature indicators for condition. Replace lavatory fire extinguisher bottles, if temperature indicators change color (white to black)	A/P	
4047	(i) Check First Aid Kit (FAK) for sealing, damage and validity (If installed).	A/P	
4048	**L.0. FINAL CHECK** L.1. Review AFML, CML, NSRDIL, DBC.	A/P	
4049	L.2. Review deffered maintenance items, log and correct as necessary.	A/P	
4050	L.3. Check the following aircraft document for completeness and validity/update : a) Certificate of Airworthiness (C of A). Valid until b) Certificate of Registration (C of R). Valid until c) Radio Permit & AASL	A/P	

BARCODE:

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Lion Group

TASKCARD

WORK ORDER NO.	A/C REG.	A/C MSN.	A/C Effectivity	OPERATOR	TASK CARD NO.
					B789-05-999-02-01-BTK-IDN

ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	Valid until | d) Weight and Balance. Date issued | e) Swing Compass Date issued | f) Insurance Certificate Valid until | g) Operation Specifications (OPSPEC). Last Revision Date | h) Required Navigation Performance (RNP). Date issued | i) Reduced Vertical Separate Minimal (RVSM). Date issued | j) Noise Certificate Date issued | k) Minimum Equipment List Date issued | l) Dent Buckle & Chart Check any due date of Temp. repair repetitive inspection or Permanent repair if required | YES/NO Report to MCC if there is/are document(s) has expired (not updated)		
4051	L.4. Make necessary entries for work accomplished and sign off airworthiness release in the aircraft maintenance log.	A/P	
52	**M.0. JOB CLOSE-UP** AMM TASK 24-22-00-860-814 M.1. Removed external power source 115 V (+5 V), 400 Hz (+20 Hz) if no longer required.	A/P	

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Lion Group

TASKCARD

WORK ORDER NO.	A/C REG.	A/C MSN.	A/C Effectivity	OPERATOR	TASK CARD NO.
					B789-05-999-02-01-BTK-IDN

ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
4052	---------- END OF TASK ----------		

START TIME(UTC)	FINISH TIME(UTC)	TOTAL MAN HOUR		DEFECT FOUND Y N M.D.R.R. No:
		EST.	ACTUAL	
		2.50		

TASK CARD RELEASE

DATE (UTC) : ..... TIME (UTC) : ..... SIGNATURE : ..... AUTHORIZATION NO. : .....

BARCODE:

B789-05-999-02-01-BTK-IDN