

# TASKCARD

A/C TYPE	Effectivity	DESCRIPTION	WORK ORDER NO.	
		A320 - DAILY CHECK		
A/C REG.	A/C MSN.	ACCESS	TASKCARD NO.	
			A320-052000-99-2-SJV-IDN	
A/C TSN.	A/C CSN.		THRESHOLD	INTERVAL
:				
OPERATOR	PLACE	ZONE	TASK	REVISION
			DET	10
START DATE	FINISH DATE	NOTE	ATA	SKILL
		<input type="checkbox"/> ETOPS <input type="checkbox"/> RVSM <input type="checkbox"/> RNP10 <input type="checkbox"/> RII <input type="checkbox"/> CDCCL		

REFERENCE			
Doc No.	Doc Description	Doc No.	Doc Description
AMM 21-26-00-710-001-A	Operational Check of the Avionics Equipment Ventilation-System via MCDU	AMM 52-10-00-210-004-A	Check Pressure of Emergency Cylinder/ Accumulator of the Passenger/Crew Doors
AMM 12-13-79-610-011-A	ENGINE OIL SERVICING Check oil level and replenish	AMM 23-73-63-710-058-A	Functional Test NO SMOKING/NO MOBILE DEVICE/FASTEN SEAT BELT SIGN
AMM 33-51-00-710-008-A	Operational Test of The Emergency Lighting in The Cabin with The "EMER EXIT LT" switch	AMM 10-11-00-555-013-A	AIRCRAFT PROTECTIVE COVERS/ DEVICES Installation of the Aircraft Protection Equipment

TOOLS REQUIRED		
PART NUMBER	DESCRIPTION	QUANTITY
460005833	SLEEVE-GROUND LOCK	2
98A10001005000	COVER – PROBE, PITOT (AIRBUS)	3
98A10001013000	COVER-TOTAL TEMP SENSOR	2
98A10001500000	COVER - SLIP-ON, AOA SENSOR	3
98A28101000000	DRAIN TL	1
98D10007512000	COVER-EXHAUST DUCT,APU	1
98D10007513000	COVER-OILCOOLER OUTLET,APU	1
98D10103500001	COVER-STATIC PROBE	3
D23080000	PIN-GROUND LOCK,NLG	1
IAE1N20008	COVER, PROTECTIVE	2
IAE1N20400	COVER, PROTECTIVE	2

MATERIAL REQUIRED			
PART NUMBER	DESCRIPTION	QUANTITY	
SKYDROL PE-5	FLUID - HYDRAULIC, EROSION ARRESTING, FIRE RESISTANT	1	

ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
2	A320 - DAILY CHECK	A/P	

<b>BARCODE:</b> <div style="text-align: center;">    A320-052000-99-2-SJV-IDN </div>
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START TIME(UTC)	FINISH TIME(UTC)	TOTAL MAN HOUR		DEFECT FOUND M.D.R.R. No:	Y	N
		EST.	ACTUAL			
		2.00				

TASK CARD RELEASE

DATE (UTC) : ..... TIME (UTC) : ..... SIGNATURE : ..... AUTHORIZATION NO. : .....

BARCODE:




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A/C TSN.	A/C CSN.		THRESHOLD	INTERVAL
:				
OPERATOR	PLACE	ZONE	TASK	REVISION
			DET	10
START DATE	FINISH DATE	NOTE	ATA	SKILL
		<input type="checkbox"/> ETOPS <input type="checkbox"/> RVSM <input type="checkbox"/> RNP10 <input type="checkbox"/> RII <input type="checkbox"/> CDCCL		

REFERENCE			
Doc No.	Doc Description	Doc No.	Doc Description
AMM 12-12-29-03	HYDRAULIC POWER - SERVICING	AMM 12-15-38-613-001	POTABLE WATER SERVICING — Fill the Potable Water Tank System
AMM 12-32-28-281-001-A	Drain Water Content	AMM 21-26-00-710-001-A	Operational Check of the Avionics Equipment Ventilation-System via MCDU
AMM 32-12-00-410-001	MAIN GEAR DOORS — Ground Doors Closing	AMM 32-12-00-010-001	MAIN GEAR DOORS — Ground Doors Opening
AMM 32-42-27-210-003-A	Functional Check of Brake Heat Pack for wear by use of the Wear indicator (Parking Brake Applied)	AMM 32-22-00-010-001	NOSE GEAR DOORS — Ground Doors Opening
AMM 32-22-00-410-001	NOSE GEAR DOORS — Ground Doors Closing	AMM 49-90-00-600-004-A	OIL SERVICING Check Oil Level and Replenish
AMM 52-10-00-210-004-A	Check Pressure of Emergency Cylinder/ Accumulator of the Passenger/Crew Doors	AMM 71-00-00-710-045-A	DRY MOTORING CHECK
AMM 12-13-79-610-011-A	ENGINE OIL SERVICING Check oil level and replenish	AMM 23-73-63-710-058-A	Functional Test NO SMOKING/NO MOBILE DEVICE/FASTEN SEAT BELT SIGN
AMM 33-51-00-710-008-A	Operational Test of The Emergency Lighting in The Cabin with The "EMER EXIT LT" switch	AMM 10-11-00-555-013-A	AIRCRAFT PROTECTIVE COVERS/ DEVICES Installation of the Aircraft Protection Equipment

TOOLS REQUIRED		
PART NUMBER	DESCRIPTION	QUANTITY
460005833	SLEEVE-GROUND LOCK	2
98A10001005000	COVER – PROBE, PITOT (AIRBUS)	3
98A10001013000	COVER-TOTAL TEMP SENSOR	2
98A10001500000	COVER - SLIP-ON, AOA SENSOR	3
98A28101000000	DRAIN TL	1
98D10007512000	COVER-EXHAUST DUCT,APU	1
98D10007513000	COVER-OILCOOLER OUTLET,APU	1
98D10103500001	COVER-STATIC PROBE	3
D23080000	PIN-GROUND LOCK,NLG	1
IAE1N20008	COVER, PROTECTIVE	2
IAE1N20400	COVER, PROTECTIVE	2
14-6806-6011	TYRE PRESSURE GAUGE	1
98A28104000000	WATER DRAIN PURGER	1
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MATERIAL REQUIRED		
PART NUMBER	DESCRIPTION	QUANTITY
SKYDROL PE-5	FLUID - HYDRAULIC, EROSION ARRESTING, FIRE RESISTANT	1
MOBIL JET II	OIL	1

ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
1	<p><b>INTERVAL NOTE:</b> The Daily Check shall be performed once per calendar day/24 hours (accomplishment not to exceed 36 calendar hours).</p> <p><b>APPLICABILITY NOTE:</b></p> <p>FSN 001-999 effective for ALL</p>		
2	<p><b>** ON A/C FSN 001-999</b></p> <p><b>1.0. JOB SET-UP</b> AMM TASK 10-11-00-555-015-A</p> <p>1.1. Install the wheel chocks. <input type="checkbox"/> Installed</p> <p>WARNING: Before you go near the landing gears to install the CHOCK - WHEEL(S) , make sure that:</p> <ul style="list-style-type: none"> <li>· The aircraft is fully stopped</li> <li>· The beacon lights are off</li> <li>· All the engines are stopped</li> </ul> <p>AMM TASK 32-00-00-481-001-A</p> <p>1.2. Install the nose landing gear Pin-Ground lock <input type="checkbox"/> Installed (D23080000 or Equivalent) and the main landing gear ground lock safety devices (460005833 or Equivalent) (as necessary). If aircraft towing is required, place the towing lever in the "towing position" and install pin.</p> <p>AMM TASK 24-41-00-861-002-A, 24-41-51-420-050-A &amp; 24-41-00-861-002-A01</p> <p>1.3. Review the aircraft technical log and start corrective actions as necessary. Do a safety check of the flight compartment for correct position of the controls.</p> <p>AMM TASK 24-42-00-861-001</p> <p>1.4. Energize the ground service network from the external power if necessary. <input type="checkbox"/> Installed</p>	A/P	

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3	<p><b>** ON A/C FSN 001-999</b></p> <p>1.5 If ‘STS’ has appeared on the upper ECAM DU, check for messages in the maintenance section of the STATUS page on the lower ECAM DU. Start corrective actions as required.</p> <p>AMM Task 27-96-00-740-001-A 1.5.1 Operational Check of EFCS by BITE (ground Scanning).</p>	A/P	
4	<p><b>** ON A/C FSN 001-999</b></p> <p>1.6 Post Flight Report. Print PFR, analyse it and take corrective actions if necessary and sent it to technical record with AFML for archiving.</p>	A/P	
5	<p><b>** ON A/C FSN 001-999</b></p> <p>AMM TASKS 10-11-00-555-013-A &amp; 10-11-00-555-069-B 1.7. Installation of the Aircraft Protection Equipment</p> <p>1.7.1. Protection Total Air Temperature Sensor           <input type="checkbox"/> Installed (98A10001013000 or Equivalent)</p> <p>1.7.2. Protection of the 3 pitot probes                         <input type="checkbox"/> Installed (98A10001005000 or Equivalent)</p> <p>1.7.3. Protection of the 3 angle-of-attack sensors          <input type="checkbox"/> Installed (98A10001500000 or Equivalent)</p> <p>1.7.4. Protection of the 6 static probes                      <input type="checkbox"/> Installed (98D10103500001 or Equivalent)</p>	A/P	
6	<p><b>** ON A/C FSN 001-999</b></p> <p>AMM TASK 10-11-00-555-070-B 1.8. Installation of the Protection Devices on the Engines</p> <p>1.8.1. Protection of the engine air intakes                   <input type="checkbox"/> Installed (IAE1N20400 or Equivalent)</p> <p>1.8.2. Protection of the engine exhaust nozzles             <input type="checkbox"/> Installed (IAE1N20008 or Equivalent)</p>	A/P	
7	<p><b>** ON A/C FSN 001-999</b></p>		

  
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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	AMM TASK 10-11-00-555-013-A 1.9. Installation of the Protection Devices on the APU Area  1.9.1. Protection of the APU exhaust duct <input type="checkbox"/> Installed (98D10007512000 or Equivalent)  1.9.2. Protection of the outlet duct of the APU oil cooler <input type="checkbox"/> Installed (98D10007513000 or Equivalent)	A/P	
8	<b>** ON A/C FSN 001-999</b>  AMM TASK 12-13-79-610-011-A 1.10. Check Oil Level and Replenish  <p style="text-align: center;"><b>WARNING:DO NOT LET ENGINE OIL STAY ON YOUR SKIN FOR A LONG TIME. FLUSH THE OIL FROM YOUR SKIN WITH WATER. THE OIL IS POISONOUS AND CAN GO THROUGH YOUR SKIN AND INTO YOUR BODY.</b></p> <p style="text-align: center;"><b>WARNING:BE CAREFUL WHEN YOU DO WORK ON THE ENGINE PARTS AFTER THE ENGINE SHUTDOWN. USE APPLICABLE THERMAL GLOVES. THE ENGINE PARTS CAN STAY HOT FOR ONE HOUR AFTER SHUTDOWN AND CAN BURN YOU.</b></p> <p style="text-align: center;"><b>WARNING:DO NOT REMOVE THE FILLER CAP OF THE OIL TANK IMMEDIATELY AFTER ENGINE OPERATION. LET THE OIL PRESSURE DECREASE FOR A MINIMUM OF 5 MINUTES AFTER THE ENGINE SHUTDOWN. IF YOU OPEN THE FILLER CAP WHEN THERE IS PRESSURE IN THE TANK, THE OIL CAN BURN YOU DANGEROUSLY.</b></p> <p style="text-align: center;"><b>CAUTION:IF THE ENGINE WAS STOPPED FOR MORE THAN 1 HOUR, DO THE SPECIFIC MAINTENANCE ACTION TO MAKE SURE THAT THE OIL LEVEL IS CORRECT. THIS WILL PREVENT OVER-SERVICING IF OIL IS ADDED.</b></p> <p style="text-align: center;"><b>CAUTION:ADDING TOO MUCH OIL TO THE TANK CAN RESULT IN AMBER CROSS QUANTITY INDICATION WARNINGS AND SERVICE DISRUPTION.</b></p> 1.10.1. Check the engine oil level. (1) Where conditions permit, the oil tank should be checked and oil added, if necessary within a period of 5 to 60 minutes	A/P	



BT-PCF-001 / Issued dated July 2019

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NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	<p>after engine shutdown.</p> <p>(2) If the engine was stopped for between 1 and 10 hours, perform an idle run Ref. AMM TASK 71-00-00-710-012 of at least 3 minutes duration prior to servicing to ensure the oil level indicated on the sight glass is correct within a period of 5 to 60 minutes after engine shut down.</p> <p>(3) If the engine was stopped for more than 10 hours, dry crank the engine Ref. AMM TASK 71-00-00-710-045 followed by a start and idle run of a least 3 minutes duration before servicing to ensure the oil level indicated on the sight glass is correct within a period of 5 to 60 minutes after engine shut down.</p> <p>(4) Servicing is not required prior to departure if the engine was serviced on arrival or the oil tank sight glass shows a minimum of 23 quarts or the cockpit indication shows a minimum of 19 quarts and pre-departure walkaround checks confirm no oil pools/drips from drain mast.</p> <p>(5) If extended cranking has been undertaken e.g. following core washing, maintenance activity, etc, prior to servicing, run the engine at idle Ref. AMM TASK 71-00-00-710-012 for a minimum of 3 minutes.</p> <p>1.10.2 Open the oil filler access door 437BL(447BL), in the left fan cowl 437AL(447AL) Ref. AMM TASK 71-13-14-010-801.</p> <p>1.10.3 Remove the filler cap from the engine oil tank.</p> <p>(1) Lift the handle to open the cap.</p> <p>(2) Turn the handle to align the lock with the slots.</p> <p>(3) Remove the filler cap from the filler neck.</p> <p>1.10.4 Fill the engine oil tank - Gravity Filling</p> <p>(1) Fill the engine oil tank with lubricant (engine oil).</p> <p>(2) Make sure that the oil level sight glass shows FULL.</p> <p><b>NOTE: Do not fill the oil tank past the sight glass FULL level. Filling to tank overfill will result in excess oil, leading to amber cross indication warnings and service disruption.</b></p> <p>1.10.5 Install the oil tank filler cap</p> <p>(1) Examine the packing on the filler cap. Replace the packing if it is damaged Ref. AMM TASK 70-23-13-911-010.</p> <p>(2) Apply engine oil to the packing.</p> <p>(3) Make sure that the filler cap handle is in the open position.</p> <p>(4) Align the lock of the cap with the slots in the filler neck.</p>	A/P	

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	<p>(5) Install the cap and turn the handle to the closed position.</p> <p>(6) Lower the handle to lock the cap.</p> <p>1.10.6 Close Access</p> <p>(1) Make sure that the work area is clean and clear of tools and other items.</p> <p><b>WARNING: MAKE SURE THAT ALL LATCHES ARE IN THE DISENGAGED POSITION BEFORE YOU CLOSE THE OIL FILLER DOOR. IF THE LATCHES ARE NOT CORRECTLY DISENGAGED BEFORE YOU CLOSE THE DOOR, IT IS POSSIBLE TO ENGAGE THE LATCHES INCORRECTLY. THIS CAN CAUSE THE DOOR TO BREAK OFF AND CAUSE INJURY.</b></p> <p>(2) Close the oil filler access door Ref. AMM TASK 71-13-14-410-801.</p>	A/P	
9	<p><b>** ON A/C FSN 001-999</b></p> <p>AMM TASK 12-13-49-612-001-A &amp; 49-90-00-600-004-A</p> <p>1.12. Do a check of the APU oil level (on the sight glass). Add approved oil to the oil reservoir as necessary. Record uplift in the aircraft technical log.</p> <p>Record the data below if there are APU oil added :</p> <p>A/C Reg. : _____</p> <p>APU S/N : _____</p> <p>APU Hour : _____</p> <p>APU Cycle : _____</p> <p>Oil Added : _____ (Qrt)</p> <p><b>NOTE:</b> Please take attention to fill the data above because it will be useful for Analysis APU Oil consumption.</p> <p><b>WARNING: PUT THE OIL INTO THE GRAVITY-FILL PORT SLOWLY. THIS WILL PREVENT A BLOW-BACK OF HOT OIL WHICH CAN BURN YOU</b></p> <p><b>CAUTION: DO NOT FILL THE GEARBOX ABOVE THE FULL MARK ON THE SIGHT GLASS. DURING APU OPERATION THE OIL EXPANDS WHICH COULD CREATE AN OVER-FILLED CONDITION RESULTING IN LEAKAGE AND POSSIBLE OIL SMELL IN THE CABIN.</b></p>	A/P	

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10	<b>** ON A/C FSN 001-999</b>  AMM TASKS 12-12-29-611-001-A 1.13. Do a fluid level check of the Green, Blue and Yellow hydraulic reservoirs (On ECAM LOWER DISPLAY, HYD page).  Add Phosphate Ester Base hydraulic fluid (Mat. No. 02-003) to the reservoirs as necessary (Ref. AMM 12-12-29-03). Record uplift(s) in the aircraft technical log.	A/P	
11	<b>2.0 EXTERNAL WALK AROUND</b>	A/P	
12	<b>** ON A/C FSN 001-999</b> AMM TASKS 05-21-00-200-001, 05-22-00-200-001, 05-28-00-200-001  <b>2.1 FORWARD FUSELAGE</b>  Do a general visual inspection of the forward fuselage from ground as far as visible, including: <ul style="list-style-type: none"> <li>- Passenger/Crew doors, cargo compartment door <input type="checkbox"/> Checked</li> <li>- Service panel doors, avionic compartment access doors <input type="checkbox"/> Checked</li> <li>- Pitot probes, static ports and AOA sensors: <input type="checkbox"/> Checked</li> <li>- No obstruction (Protective covers removed), <input type="checkbox"/> Checked</li> <li>- Avionics equipment ventilation: Air inlet and air outlet valves <input type="checkbox"/> Checked</li> <li>- Batteries ventilation outlet (Venturi): No obstruction, <input type="checkbox"/> Checked</li> <li>- Waste water drain mast, <input type="checkbox"/> Checked</li> <li>- Antennas: No damage, <input type="checkbox"/> Checked</li> <li>- Wing and engine scan lights: Cleanliness, <input type="checkbox"/> Checked</li> <li>- Radome: No damage, latches engaged and locked, <input type="checkbox"/> Checked</li> <li>- Crew oxygen cylinder overpressure indicator Green disc in place: <input type="checkbox"/> Checked</li> <li>- FWD Cargo - Visual Check of Cargo Compartment Decompression, Lining, Floor Panel and Pressure Compensating Valve (as far as visible) <input type="checkbox"/> Checked</li> <li>- FWD Cargo- Visual Check of Divider Nets, Door Net, <input type="checkbox"/> Checked</li> </ul>	A/P	

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NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	Net Attachment Point and Light		
13	<p><b>** ON A/C FSN 001-999</b></p> <p>AMM TASK 05-27-00-200-001</p> <p><b>2.2 NOSE LANDING GEAR AND DOORS</b></p> <p>Do a general visual inspection of the nose landing gear and doors from ground as far as visible, including:</p> <ul style="list-style-type: none"> <li>- Doors and wheel well, <input type="checkbox"/> Checked</li> <li>- Gear assy structure: Damage, evidence of leakage, <input type="checkbox"/> Checked</li> <li>- Ground sensing mechanism in proper condition, Bolt, Nut, Cotter pin and Safety locking in good condition</li> </ul> <p>Result : .....</p> <ul style="list-style-type: none"> <li>- Shock absorber: Signs of leakage and normal extension, <input type="checkbox"/> Checked</li> <li>- Shock absorber sliding tube: Scoring, damage, cleanliness, <input type="checkbox"/> Checked</li> </ul> <p><b>NOTE:</b> Please pay attention to the cleanliness of the Shock Absorber chromium surface. Clean the sliding tube from any dust, dirt, or other unwanted material that could potentially erode the dynamic seal and led to hydraulic leakage. (Clean using a lint-free cloth moist with mineral hydraulic fluid (Mat. No. 02-001), clean the tube and dry with a clean cloth)</p> <ul style="list-style-type: none"> <li>- NLG gear actuating-cylinder and Down Lock Actuator : <input type="checkbox"/> Checked Signs of leakage,normal extension, damage/corrosion, cleanliness</li> </ul> <p><b>NOTE:</b> Ref AMM 32-22-00-010/410-001 to open and close the NLG Door for inspection. Put a small quantity of Phosphate Ester Hydraulic Fluid-General Power - - (Material Ref. 02ABA1) on a clean Textile-Lint free Cotton - (Material Ref. 14SBA1) and apply a thin layer phosphate Ester Hydraulic Fluid-General Power - - (Material Ref. 02ABA1) only to the chromium surfaces of:</p> <ul style="list-style-type: none"> <li>• The NLG gear actuating-cylinder</li> <li>• The NLG downlock actuator.</li> </ul> <ul style="list-style-type: none"> <li>- Taxi and take-off lights, runway turnoff lights: Cleanliness, <input type="checkbox"/> Checked</li> <li>- Wheels: Rim damage, sheared/missing tie bolts, <input type="checkbox"/> Checked</li> <li>- Tires: Wear, damage and signs of under inflation. <input type="checkbox"/> Checked</li> </ul>		

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14	<b>** ON A/C FSN 001-999</b>  AMM TASK 05-21-00-200-001, 05-22-00-200-001, 05-28-00-200-001 <b>2.3 CENTER FUSELAGE</b> Do a general visual inspection of the forward center fuselage from ground as far as visible, including: <ul style="list-style-type: none"> <li>- Emergency exits, <input type="checkbox"/> Checked</li> <li>- Service panel doors, <input type="checkbox"/> Checked</li> <li>- Wing-to-fuselage fairings and belly fairing, <input type="checkbox"/> Checked</li> <li>- Anti-collision beacon lights: Cleanliness, <input type="checkbox"/> Checked</li> <li>- Drain mast, <input type="checkbox"/> Checked</li> <li>- Ram air inlet flap, <input type="checkbox"/> Checked</li> <li>- Pack air intakes and outlets: No obstruction, <input type="checkbox"/> Checked</li> <li>- Ram air turbine doors: Correctly closed, <input type="checkbox"/> Checked</li> <li>- Antennas: No damage. <input type="checkbox"/> Checked</li> </ul>	A/P	
15	<b>** ON A/C FSN 001-999</b>  AMM TASK 05-25-00-200-001-A <b>2.4 RIGHT/LEFT WING LEADING EDGE</b>  Do a general visual inspection of the RH and the LH wing leading edge from ground as far as visible, including: <ul style="list-style-type: none"> <li>- Leading edge slats, <input type="checkbox"/> Checked</li> <li>- Leading edge access panels (Pressure relief panels in place), <input type="checkbox"/> Checked</li> <li>- Surge tank NACA air intake: No obstruction Surge tank overpressure protector (White cross is visible). <input type="checkbox"/> Checked</li> </ul>	A/P	

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16	<p><b>** ON A/C FSN 001-999</b></p> <p>AMM TASK 05-24-00-200-001</p> <p><b>2.5 RIGHT/LEFT POWER PLANT AND PYLON</b></p> <p>Do a general visual inspection of the power plant and pylon from ground as far as visible, including (cowl doors closed):</p> <ul style="list-style-type: none"> <li>- Pylon with fairings and fillets: No damage, <input type="checkbox"/> Checked</li> <li>- Engine air inlet: Lip skin, riveting, acoustic panels, and visible probes and sensors (T12), <input type="checkbox"/> Checked</li> <li>- Fan blades and spinner: No damage, check free rotation by hand, <input type="checkbox"/> Checked</li> <li>- Fan cowl doors and thrust reverser cowl doors: <input type="checkbox"/> Checked</li> <li>- Cowl doors closed and correctly latched, <input type="checkbox"/> Checked</li> <li>- Blocker doors in stowed (closed) position, <input type="checkbox"/> Checked</li> <li>- Access doors and pressure relief doors in place, closed and latched, <input type="checkbox"/> Checked</li> <li>- Air inlets/outlets: No obstruction, <input type="checkbox"/> Checked</li> <li>- Fan exhaust: Acoustic lining, exit vanes and struts, thrust reverser pivoting doors, <input type="checkbox"/> Checked</li> <li>- Turbine exhaust: Last stage LPT blades, nozzle and plug (for damage and metal deposit), <input type="checkbox"/> Checked</li> <li>- Drain mast and pylon drains: No obstruction, evidence of leakage. <input type="checkbox"/> Checked</li> </ul>	A/P	

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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
17	<b>** ON A/C FSN 001-999</b>  AMM TASK 05-25-00-200-001 <b>2.6 RIGHT/LEFT WING TIP AND TRAILING EDGE</b>  Do a general visual inspection of the RH and the LH wing tip and trailing edge from ground as far as visible, including:  - Navigation and strobe lights, <input type="checkbox"/> Checked  - Static dischargers, <input type="checkbox"/> Checked  - Control surfaces, flaps and flap track fairings: Damage, evidence of fluid leakage, <input type="checkbox"/> Checked  - Lower wing surface: Damage, evidence of fuel leakage, <input type="checkbox"/> Checked  - Wing tank overpressure protector: Burst disk in place (White cross visible), <input type="checkbox"/> Checked  - Landing light: No damage, cleanliness. <input type="checkbox"/> Checked	A/P	
18	<b>** ON A/C FSN 001-999</b>  AMM 05-27-00-200-001 <b>2.7 RIGHT/LEFT MAIN LANDING GEAR AND DOORS</b>  Do a general visual inspection of the main landing gears and doors from ground as far as visible, including:  - Doors and wheel well, <input type="checkbox"/> Checked  - Gear assy structure: Damage, evidence of leakage, <input type="checkbox"/> Checked  - Shock absorber: Signs of leakage and normal extension, <input type="checkbox"/> Checked  - Shock absorber sliding tube: Scoring/damage, cleanliness,(if necessary using lint - free cloth moist with mineral hydraulic fluid (Mat. No. 02-001), clean tube and dry with a clean cloth), <input type="checkbox"/> Checked  - Proximity detectors: Security, cleanliness, <input type="checkbox"/> Checked  - Brake units: Evidence of leakage or overheating, <input type="checkbox"/> Checked	A/P	

**BARCODE:**

**A320-052000-99-2-SJV-IDN**

# TASKCARD

WORK ORDER NO.	A/C REG.	A/C MSN.	A/C Effectivity	OPERATOR	TASK CARD NO.
					A320-052000-99-2-SJV-IDN

ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	<ul style="list-style-type: none"> <li>- Wheels: Rim damage, sheared/missing tie bolts, <input type="checkbox"/> Checked</li> <li>- Tires: Wear, damage and signs of under inflation. <input type="checkbox"/> Checked</li> </ul>	A/P	
19	<p><b>** ON A/C FSN 001-999</b></p> <p>AMM TASK 32-42-27-210-003</p> <p><b>2.8 MAIN LANDING GEAR BRAKES (MP TASK 324227-01-1)</b></p> <p>Check Heat Pack Wear Indicator (Parking Brake Applied). <input type="checkbox"/> Checked</p> <p>(1) On parking brake panel 110VU, set parking brake control-switch 73GG to the ON position.</p> <p>(2) For an in-service brake Measure the wear indicators when the brake is cold (brake temperature less than 60 deg.C (140 deg.F)).</p> <p>Brake #1: _____mm      Brake #2: _____mm</p> <p>Brake #3: _____mm      Brake #4: _____mm</p> <p>NOTE : Replace if Brake PIN Flush (1 mm/30 cycle).</p> <p>NOTE: Do this test only if the remaining length of the wear indicators was not measured on the aircraft.</p> <p>CAUTION: MAKE SURE THAT NO HYDRAULIC FLUID FALLS ON THE HEAT PACK. THROUGHOUT ALL THE TESTS, MAKE SURE THAT THERE IS NO EXTERNAL LEAKAGE THROUGH THE WALLS.</p> <p>CAUTION: DO NOT PRESSURIZE THE CARBON BRAKE THROUGH THE TWO CIRCUITS AT THE SAME TIME.</p> <p>(a) Apply a pressure of 145 bar (2100 psi) to one of the two circuit of the carbon brake.</p> <p>(b) Measure the length of the wear indicators.  <u>1</u> If the end of one of the two wear indicators is flush with the machined surface of the piston housing (heat pack fully worn), replace the brake (Ref. AMM TASK 32-42-27-000-001) (Ref. AMM TASK 32-42-27-400-001) .</p>	A/P	

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ACCOMPLISHMENT			
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	<p>NOTE: The brake units have two wear indicators.</p> <p>But if a brake does not have the two wear indicators:</p> <ul style="list-style-type: none"> <li>– The brake continues to be serviceable if there is one wear indicator which is in the limits.</li> <li>– If there is no serviceable wear indicator, do the deactivation of the brake (Ref. AMM TASK 32-42-00-040-003) .</li> </ul> <p>(3) For a new brake:</p> <ul style="list-style-type: none"> <li>– With the parking brake on, measure distance D between the end of the wear indicators and the machined surface of the piston housing.</li> <li>– Make sure that distance D is between 49.6 mm (1.953 in.) and 50.4 mm (1.984 in.).</li> <li>– If distance D is out of the limits, replace the brake (Ref. AMM TASK 32-42-27-000-001) (Ref. AMM TASK 32-42-27-400-001).</li> </ul> <p>(4) Do a check of the components that follow to <input type="checkbox"/> Checked</p> <p>make sure that there are no leaks :</p> <ul style="list-style-type: none"> <li>- The piston housing</li> <li>- The piston</li> <li>- The hydraulic lines.</li> </ul>	A/P	
20	<p><b>** ON A/C FSN 001-999</b></p> <p>AMM TASK 32-41-00-210-003</p> <p><b>2.9 WHEELS</b> (MP TASK 324100-01-1)</p> <p>Do a check of the tires for correct pressure. <input checked="" type="checkbox"/> Checked</p> <p>(sub task 32-41-00-210-058-A)</p> <p>To do this, refer to the pressure table. and compare the tire pressure values with those that you find when you use:</p> <ul style="list-style-type: none"> <li>· A GAGE (14-6806-6011 or Equivalent) - PRESSURE, TIRE , or</li> <li>· The wheel-mounted tire-pressure indicators, if installed, or</li> <li>· The Tire Pressure Indicating System (TPIS), if installed.</li> </ul> <p><b>CAUTION:</b> TIRE PRESSURE MUST BE CHECKED AND CORRECTED</p>	A/P	

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**A320-052000-99-2-SJV-IDN**

# TASKCARD

WORK ORDER NO.	A/C REG.	A/C MSN.	A/C Effectivity	OPERATOR	TASK CARD NO.
					A320-052000-99-2-SJV-IDN

ACCOMPLISHMENT																																														
NO.	INSTRUCTION			PERFORMED BY	INSPECTED BY																																									
	<p>WHEN THE TIRES ARE COLD. SERVICE WITH NITROGEN ONLY. IF THE TIRE PRESSURE IS OUT OF LIMIT PERFORM NECESSARY ACTION I.A.W. AMM.</p> <p><b>NOTE: Note tire pressure before inflation.</b></p> <p>NLG: LH _____psi RH _____psi</p> <p>MLG: LH/IB _____psi LH/OB _____psi</p> <p>MLG: RH/IB _____psi RH/OB _____psi</p> <table border="1"> <thead> <tr> <th rowspan="2"></th><th rowspan="2">A/C WEIGHT</th><th rowspan="2">TIRE DIMENSIONS</th><th colspan="2">INFLATION PRESSURE</th></tr> <tr> <th>UNLOADED / BARS (PSI)</th><th>LOADED BAR (PSI)</th></tr> </thead> <tbody> <tr> <td>Main Wheel</td><td>73.9 t</td><td>46 X 17 R20</td><td>13.3 (193)</td><td>13.8 (200)</td></tr> <tr> <td>Nose Wheel</td><td>73.9 t</td><td>30 X 8.8 R15</td><td>11.8 (171)</td><td>12.3 (178)</td></tr> </tbody> </table> <p>NOTE: Please Check the Tire Dimension</p> <table border="1"> <thead> <tr> <th>TIRE PRESSURE</th><th>TIRE PRESSURE</th><th>TIRE CONDITION / ACTION NECESSARY</th></tr> </thead> <tbody> <tr> <td><b>NLG 30X8.8-15 30X8.8R15</b></td><td><b>MLG 46X16-20 46X17R20</b></td><td></td></tr> <tr> <td>More than 12.9 bars (187 psi)</td><td>More than 14.5 bars (210 psi)</td><td>Over-inflation. Make sure that the measurement is correct with another pressure gage. If it is correct, deflate the tire to the maximum normal pressure.</td></tr> <tr> <td>From 12.3 to 12.9 bars (178-187 psi)</td><td>From 13.8 to 14.5 bars (200-210 psi)</td><td>Normal pressure range. Do not adjust the tire pressure.</td></tr> <tr> <td>From 11.7 to less than 12.3 bars (170-178 psi)</td><td>From 13.1 to less than 13.8 bars (190-200 psi)</td><td>Inflate the tire to the maximum normal pressure.</td></tr> <tr> <td>From 11.1 to less than 11.7 bars (161-170 psi)</td><td>From 12.4 to less than 13.1 bars (180-190 psi)</td><td>Inflate the tire to the maximum normal pressure. You must measure the tire pressure again the next day. If the tire is under-inflated again, you must replace the wheel.</td></tr> <tr> <td>From 9.9 to less than 11.1 bars (144-161 psi)</td><td>From 11 to less than 12.4 bars (160-180 psi)</td><td>You must replace the wheel.</td></tr> <tr> <td>From 0 to less than 9.9 bars ( 0-144 psi)</td><td>From 0 to less than 11 bars ( 0-160 psi)</td><td>You must replace the wheel and the adjacent wheel.</td></tr> </tbody> </table>				A/C WEIGHT	TIRE DIMENSIONS	INFLATION PRESSURE		UNLOADED / BARS (PSI)	LOADED BAR (PSI)	Main Wheel	73.9 t	46 X 17 R20	13.3 (193)	13.8 (200)	Nose Wheel	73.9 t	30 X 8.8 R15	11.8 (171)	12.3 (178)	TIRE PRESSURE	TIRE PRESSURE	TIRE CONDITION / ACTION NECESSARY	<b>NLG 30X8.8-15 30X8.8R15</b>	<b>MLG 46X16-20 46X17R20</b>		More than 12.9 bars (187 psi)	More than 14.5 bars (210 psi)	Over-inflation. Make sure that the measurement is correct with another pressure gage. If it is correct, deflate the tire to the maximum normal pressure.	From 12.3 to 12.9 bars (178-187 psi)	From 13.8 to 14.5 bars (200-210 psi)	Normal pressure range. Do not adjust the tire pressure.	From 11.7 to less than 12.3 bars (170-178 psi)	From 13.1 to less than 13.8 bars (190-200 psi)	Inflate the tire to the maximum normal pressure.	From 11.1 to less than 11.7 bars (161-170 psi)	From 12.4 to less than 13.1 bars (180-190 psi)	Inflate the tire to the maximum normal pressure. You must measure the tire pressure again the next day. If the tire is under-inflated again, you must replace the wheel.	From 9.9 to less than 11.1 bars (144-161 psi)	From 11 to less than 12.4 bars (160-180 psi)	You must replace the wheel.	From 0 to less than 9.9 bars ( 0-144 psi)	From 0 to less than 11 bars ( 0-160 psi)	You must replace the wheel and the adjacent wheel.	A/P	
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21	<p><b>** ON A/C FSN 001-999</b></p> <p>AMM TASK 05-21-00-200-001-1, 05-22-00-200-001-1, 05-28-00-200-001</p> <p><b>2.10 AFT FUSELAGE</b></p> <p>Do general visual inspection of the aft fuselage from ground as far as visible, including:</p> <p>- Passenger/Crew, cargo compartment doors, <input type="checkbox"/> Checked</p>			A/P																																										

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**A320-052000-99-2-SJV-IDN**

# TASKCARD

WORK ORDER NO.	A/C REG.	A/C MSN.	A/C Effectivity	OPERATOR	TASK CARD NO.
					A320-052000-99-2-SJV-IDN

ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	<ul style="list-style-type: none"> <li>- Service panel doors, <input type="checkbox"/> Checked</li> <li>- Antennas and beacon, <input type="checkbox"/> Checked</li> <li>- Waste water drain mast, <input type="checkbox"/> Checked</li> <li>- Outflow valve, <input type="checkbox"/> Checked</li> <li>- Fuselage tail damage due to ground contact. <input type="checkbox"/> Checked</li> <li>- AFT Cargo - Visual Check of Cargo Compartment Decompression, Lining, Floor Panel and Pressure Compensating Valve (as far as visible) <input type="checkbox"/> Checked</li> <li>- AFT Cargo - VisualCheck of Divider Nets, <input type="checkbox"/> Checked</li> <li>- Door Nets, Net Attachment Point and Light</li> </ul>	A/P	

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# TASKCARD

22	<p><b>** ON A/C FSN 001-999</b></p> <p>AMM TASK 05-23-00-200-001</p> <p><b>2.11 STABILIZERS AND CONE/REAR FUSELAGE</b></p> <p>Do a general visual inspection of the rear fuselage with cone and stabilizers from ground as far as visible, including:</p> <ul style="list-style-type: none"> <li>- THS and elevators: Damage, evidence of fluid leakage, <input type="checkbox"/> Checked</li> <li>- Fin and rudder: Damage, evidence of fluid leakage, <input type="checkbox"/> Checked</li> <li>- Static dischargers, <input type="checkbox"/> Checked</li> <li>- APU Area: <input type="checkbox"/> Checked</li> <li>- Air intake: Flap closed or open (If APU is Off or On), <input type="checkbox"/> Checked</li> <li>- Doors: Closed and secured, <input type="checkbox"/> Checked</li> <li>- Drain mast, drains and vents, <input type="checkbox"/> Checked</li> <li>- Exhaust, <input type="checkbox"/> Checked</li> <li>- Fire extinguisher overpressure discharge indicator: <input type="checkbox"/> Checked</li> <li style="padding-left: 20px;">Red disc in place,</li> <li>- Tail cone strobe light. <input type="checkbox"/> Checked</li> </ul>	A/P	
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23	<b>3.0 WASTE/WATER SERVICING</b>		
24	<b>** ON A/C FSN 001-999</b>  AMM TASK 12-15-38-613-001-A 3.1.1. Service the potable water system as required (15).  AMM TASK 12-15-38-613-002-A 3.1.2. Service the toilet waste tanks as required (12). In freezing conditions, obey the cold weather maintenance.	A/P	
25	<b>4.0 FLIGHT COMPARTMENT</b>		
26	<b>** ON A/C FSN 001-999</b>  4.1 OVER WING SLIDE AND CREW OXYGEN SYSTEM  On the lower ECAM DU, on the DOOR/OXY page:  1. Make sure that the over wing is indicated in SLIDE. If the SLIDE not shown, do a check of the target switch at overwing and release if target switch release out.  2. And makesure that the oxygen pressure is indicated in Green. If the Amber hal frame is shown, do a check of the oxygenessure for the flight crew and service the system as required.  COB : _____psi  NOTE : Please Refer to FCOM for Limitation Oxygen.	A/P	
27	<b>** ON A/C FSN 001-999</b>  4.2 EXTERNAL LIGHTS Do an operational check of AMM TASK 33-41-00-710-001-A  1. Navigation Lights, Landing Lights, <input type="checkbox"/> Tested  AMM TASK 33-42-00-710-001-A	A/P	

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	2. Landing Light <input type="checkbox"/> Tested AMM TASK 33-43-00-710-001-A 3. Runway Turnoff Lights, <input type="checkbox"/> Tested AMM TASK 33-46-00-710-001-A 4. Taxi and Takeoff Lights, <input type="checkbox"/> Tested AMM TASK 33-47-00-710-002- 5. Logo Lights, <input type="checkbox"/> Tested AMM TASK 33-48-00-710-001-A 6. Anti-collision/Beacon <input type="checkbox"/> Tested AMM TASK 33-49-00-710-001-A 7. Lights, Wing and Engine Scan Lights. <input type="checkbox"/> Tested	A/P	
28	<b>** ON A/C FSN 001-999</b> AMM TASK 33-13-00-710-001-A , 33-51-00-710-008-A , & 23-73-63-710-058-A 4.3 Do a check of the: - flight compartment lighting, including stby compass light <input type="checkbox"/> Checked - missing spare bulbs/fuses, replace as necessary, <input type="checkbox"/> Checked - emergency equipment for presence and correct stowage, flight compartment windshields and the side windows for damage and cleanliness <input type="checkbox"/> Checked - flight compartment for general condition and cleanliness. <input type="checkbox"/> Checked - operational test of the Emergency Lighting in the cabin with the "EMER EXIT LT" switch <input type="checkbox"/> Checked - functional test NO SMOKING / NO MOBILE DEVICE / FASTEN SEAT BELT SIGN <input type="checkbox"/> Checked - Do general visual inspection Captain & FO seat condition <input type="checkbox"/> Checked	A/P	
29	<b>** ON A/C FSN 001-999</b>		

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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	AMM TASK 31-33-00-710-007-A & 237100-710-001-A  <b>4.4 DIGITAL FLIGHT DATA RECORDER &amp; COCKPIT VOICE RECORDER</b> (EASA SIB No.: 2009-28R1) (MP TASK 237135-99-1 & 313300-99-1) (CASR 91 Appendix E)  4.4.1 Operational test of the DFDR <input type="checkbox"/> Tested  4.4.2 Operational test of The CVR. <input type="checkbox"/> Tested	A/P	
30	<b>** ON A/C FSN 001-999</b>  AMM TASK 71-00-00-710-045-A  4.5. Perform Dry-Motoring Check <input type="checkbox"/> Tested  NOTE: with exception to not perform: Subtask 71-00-00-010-089-A, Subtask 71-00-00-040-067-A, Subtask 71-00-00-010-091-A, Subtask 71-00-00-410-090-A, Subtask 71-00-00-440-066-A, Subtask 71-00-00-410-092-A,		
31	<b>**ON A/C FSN 001-999</b>  TASK 21-26-00-710-001-A  4.6. Operational Check of the Avionics Equipment Ventilation <input type="checkbox"/> Tested System via MCDU (MP TASK: 212600-01-1)	A/P	
32	<b>ON A/C FSN ALL</b>  On RMP1 (2 or 3), push VHF3 push button, make sure ACARS/DATA indication is shown in the ACTIVE window. if no, push the transfer pushbutton switch (double arrow) between the two window	A/P	

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33	<b>5.0 PASSENGER COMPARTMENT</b>		
34	<p><b>** ON A/C FSN 001-999</b></p> <p>5.1. Check passenger compartment for general condition and cleanliness. <input type="checkbox"/> Checked</p> <p>5.2. Check of the galleys for general condition, cleanliness and evidence of water leakage. <input type="checkbox"/> Checked</p> <p>5.3. Do a check of the lavatories for general condition, cleanliness and evidence of leakage. <input type="checkbox"/> Checked</p> <p>5.4. Check the closing flap of the waste bins for correct operation <input type="checkbox"/>Checked</p>	A/P	

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	<p>5.5. Do a check of the emergency equipment for presence and correct stowage. <input type="checkbox"/> Checked</p> <p>5.6. Examine the hermetic seal of the first aid Kits for condition. If seal is broken, take action as required. <input type="checkbox"/> Checked</p> <p>5.6. Do a general visual check of the left and right wing upper surface and control surfaces, through cabin windows. <input type="checkbox"/> Checked</p>	A/P	
35	<p><b>** ON A/C FSN 001-999</b></p> <p><b>6.0. EMERGENCY EQUIPMENT</b></p> <p>6.1. Check aircraft emergency equipment for condition, proper pressure, safeties, security and validity date including the following:</p> <p>a) DOOR - Escape Slide / Raft <input type="checkbox"/> Checked</p> <p>b) PASSENGER CABIN</p> <p>- Halon / Water Fire extinguisher <input type="checkbox"/> Checked</p> <p>- PBE <input type="checkbox"/> Checked</p> <p>- Oxygen Bottle <input type="checkbox"/> Checked</p> <p>- First Aid Kit <input type="checkbox"/> Checked</p>	A/P	

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	c) LAVATORY - Fire extinguisher <input type="checkbox"/> Checked	A/P	
36	<b>** ON A/C FSN 001-999</b> AMM TASK 52-10-00-210-004-A <b>7.0. PASSENGER/CREW DOOR</b> (MP TASK 521000-01-1) 7.1. Check emergency cylinder/accumulator pressure.  FWD : R/H _____ PSI L/H _____ PSI AFT : R/H _____ PSI L/H _____ PSI NOTE: IF PRESSURE IS BELOW LIMITATION PLEASE SERVICING DOOR DAMPER	A/P	
37	<b>8.0. WINGS</b>		
38	<b>** ON A/C FSN 001-999</b> AMM 12-32-28-281-001 <b>8.1. TANKS</b> (MP TASK 281100-01-2)  Drain water from wing tanks and center tank at water drain valves.  NOTE: If possible, do the water drain procedure before a refuel. Alternatively, you must wait for one hour after a refuel has been completed before you do the water drain procedure.  To get access to the high water drain valves from the ground, attach the: - ADAPTING PIPE - WATER DRAINING FROM FUEL (98A28101000000) or - PIPE ADAPTING-WATER DRAINING (98D28104000000) to the bottle and funnel of the PURGER - WATER DRAIN (98A28104000000 or Equivalent) .  After drainage, make sure that the drain valves are correctly closed and not leaking.	A/P	
39	<b>9.0. FINAL ITEMS</b>		
40	<b>** ON A/C FSN 001-999</b>  9.1. Review AFML, CML, DMI, NSRDIL, DBC, Deferred Item (from. Phase check or C Check)	A/P	
41	Revision item 9.2, paragraph d and e <b>** ON A/C FSN 001-999</b>	A/P	

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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	<p>9.2. Check the following aircraft document for completeness and validity:</p> <p>a) Certificate of Airworthiness (C of A).</p> <div>Valid until <input type="text"/></div> <p>b) Certificate of Registration (C of R).</p> <div>Valid until <input type="text"/></div> <p>c) Radio Permit &amp; AASL</p> <div>Valid until <input type="text"/></div> <p>d) Weight and Balance.</p> <div>Valid until <input type="text"/></div> <p>e) Swing Compass</p> <div>Valid until <input type="text"/></div> <p>f) Insurance Certificate</p> <div>Valid until <input type="text"/></div> <p>g) Operation Specifications (OPSPEC) and ACL.</p> <div>Last Revision Date <input type="text"/></div> <p>h) Required Navigation Performance (RNP).</p> <div>Date issued <input type="text"/></div> <p>i) Reduced Vertical Separate Minimal (RVSM).</p> <div>Date issued <input type="text"/></div> <p>j) Noise Certificate</p> <div>Date issued <input type="text"/></div> <p>k) Minimum Equipment List</p> <div>Date issued <input type="text"/></div> <p>Report to MCC if there is/are document(s) has expired (not updated)</p>	A/P	
42	<b>** ON A/C FSN 001-999</b>	A/P	

**BARCODE:**

**A320-052000-99-2-SJV-IDN**

# TASKCARD

WORK ORDER NO.	A/C REG.	A/C MSN.	A/C Effectivity	OPERATOR	TASK CARD NO.
					A320-052000-99-2-SJV-IDN

ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	9.3. - Check corrective actions in the aircraft technical log and record each corrected or carried forward item. Record the DAILY check in the aircraft technical log. <input type="checkbox"/> Checked  - De-energize the aircraft electrical network. <input type="checkbox"/> Removed  - Make sure that the cockpit sliding windows are closed. <input type="checkbox"/> Checked  - Make sure that all passenger/crew doors, cargo compartment doors and service panel doors are correct closed and secured. Install the protective covers on the aircraft as necessary. Make sure that an authorized person releases the aircraft. <input type="checkbox"/> Checked	A/P	
43	THE END OF THE TASK		

START TIME(UTC)	FINISH TIME(UTC)	TOTAL MAN HOUR		DEFECT FOUND M.D.R.R. No:	Y	N
		EST.	ACTUAL			
		2.25				

## TASK CARD RELEASE

DATE (UTC) : ..... TIME (UTC) : ..... SIGNATURE : ..... AUTHORIZATION NO. : .....

BARCODE:



A320-052000-99-2-SJV-IDN

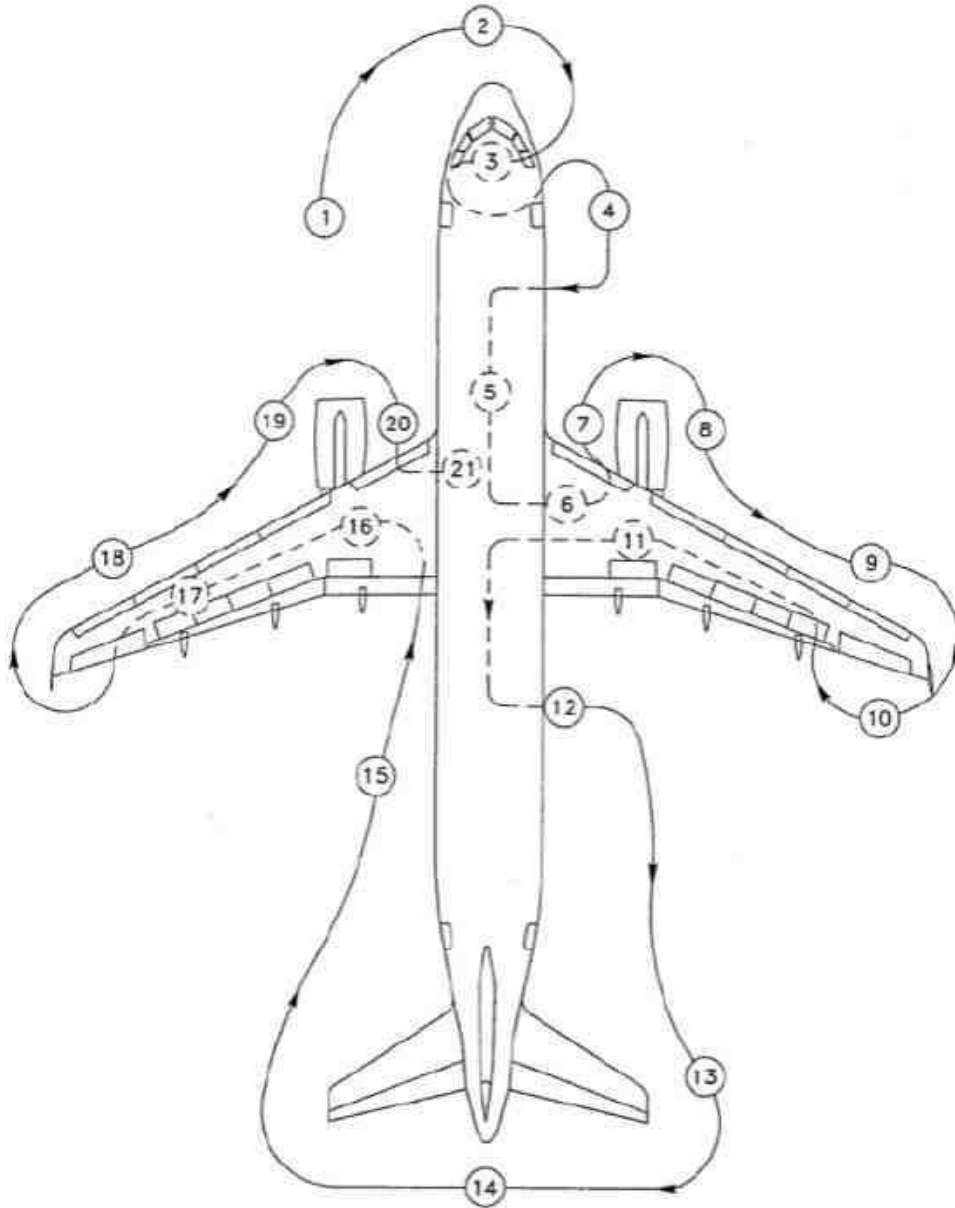
# TASKCARD

WORK ORDER NO.	A/C REG.	A/C MSN.	A/C Effectivity	OPERATOR	TASK CARD NO.
					A320-052000-99-2-SJV-IDN

## APPENDIX

GRAPHIC  
PICTURE.JPG

### EXTERNAL WALK AROUND DIAGRAM



BARCODE:



A320-052000-99-2-SJV-IDN