TASKCARD

A/C TYPE	Effectivity	DESCRIPTION	WORK ORDER NO.	
		A320 - DAILY CHECK		
A/C REG.	A/C MSN.	ACCESS	TASKCARD NO.	
			A320-052000-99-2-SJV-IDN	
A/C TSN.	A/C CSN.		THRESHOLD	INTERVAL
:				
OPERATOR	PLACE	ZONE	TASK	REVISION
			DET	09
START DATE	FINISH DATE	NOTE	ATA	SKILL
		☐ ETOPS ☐ RVSM ☐ RNP10 ☐ RII ☐ CDCCL		

REFERENCE

Doc No.	Doc Description	Doc No.	Doc Description
AMM 12-12-29-03	HYDRAULIC POWER - SERVICING	AMM 12-15-38-613-001	POTABLE WATER SERVICING — Fill the Potable Water Tank System
AMM 12-32-28-281-001-A	Drain Water Content	AMM 21-26-00-710-001-A	Operational Check of the Avionics Equipment Ventilation-System via MCDU
AMM 32-12-00-410-001	MAIN GEAR DOORS — Ground Doors Closing	AMM 32-12-00-010-001	MAIN GEAR DOORS — Ground Doors Opening
AMM 32-42-27-210-003-A	Functional Check of Brake Heat Pack for wear by use of the Wear indicator (Parking Brake Applied)	AMM 32-22-00-010-001	NOSE GEAR DOORS — Ground Doors Opening
AMM 32-22-00-410-001	NOSE GEAR DOORS — Ground Doors Closing	AMM 49-90-00-600-004-A	OIL SERVICING Check Oil Level and Replenish
AMM 52-10-00-210-004-A	Check Pressure of Emergency Cylinder/ Accumulator of the Passenger/Crew Doors	AMM 71-00-00-710-045-A	DRY MOTORING CHECK
AMM 12-13-79-610-011-A	ENGINE OIL SERVICING Check oil level and replenish	AMM 23-73-63-710-058-A	Functional Test NO SMOKING/NO MOBILE DEVICE/FASTEN SEAT BELT SIGN
AMM 33-51-00-710-008-A	Operational Test of The Emergency Lighting in The Cabin with The "EMER EXIT LT" switch	AMM 10-11-00-555-013-A	AIRCRAFT PROTECTIVE COVERS/ DEVICES Installation of the Aircraft Protection Equipment

TOOLS REQUIRED

PART NUMBER	DESCRIPTION	QUANTITY
460005833	SLEEVE-GROUND LOCK	2
98A10001005000	COVER – PROBE, PITOT (AIRBUS)	3
98A10001013000	COVER-TOTAL TEMP SENSOR	2
98A10001500000	COVER - SLIP-ON, AOA SENSOR	3
98A28101000000	DRAIN TL	1
98D10007512000	COVER-EXHAUST DUCT,APU	1
98D10007513000	COVER-OILCOOLER OUTLET,APU	1
98D10103500001	COVER-STATIC PROBE	3
D23080000	PIN-GROUND LOCK,NLG	1
IAE1N20008	COVER, PROTECTIVE	2
IAE1N20400	COVER, PROTECTIVE	2
14-6806-6011	TYRE PRESSURE GAUGE	1
98A28104000000	WATER DRAIN PURGER	1

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MATERIAL REQUIRED		
PART NUMBER	**DESCRIPTION**	**QUANTITY**
SKYDROL PE-5	FLUID - HYDRAULIC, EROSION ARRESTING, FIRE RESISTANT	1
MOBIL JET II	OIL	1

ACCOMPLISHMENT

NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
1	**INTERVAL NOTE:** The Daily Check shall be performed once per calendar day/24 hours (accomplishment not to exceed 36 calendar hours). **APPLICABILITY NOTE:** FSN 001-999 effective for ALL		
2	***** ON A/C FSN 001-999*** **1.0. JOB SET-UP** AMM TASK 10-11-00-555-015-A 1.1. Install the wheel chocks. ☐ Installed WARNING: Before you go near the landing gears to install the CHOCK - WHEEL(S) , make sure that: · The aircraft is fully stopped · The beacon lights are off · All the engines are stopped AMM TASK 32-00-00-481-001-A 1.2. Install the nose landing gear Pin-Ground lock ☐ Installed (D23080000 or Equivalent) and the main landing gear ground lock safety devices (460005833 or Equivalent) (as necessary). If aircraft towing is required, place the towing lever in the "towing position" and install pin. AMM TASK 24-41-00-861-002-A, 24-41-51-420-050-A &24-41-00-861-002-A01 1.3. Review the aircraft technical log and start corrective actions as necessary. Do a safety check of the flight compartment for correct position of the controls. AMM TASK 24-42-00-861-001 1.4. Energize the ground service network from the external power if necessary. ☐ Installed	A/P	

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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
3	**** ON A/C FSN 001-999*** 1.5 If 'STS' has appeared on the upper ECAM DU, check for messages in the maintenance section of the STATUS page on the lower ECAM DU. Start corrective actions as required. AMM Task 27-96-00-740-001-A 1.5.1 Operational Check of EFCS by BITE (ground Scanning).	A/P	
4	**** ON A/C FSN 001-999*** 1.6 Post Flight Report. Print PFR, analyse it and take corrective actions if necessary and sent it to technical record with AFML for archiving.	A/P	
5	**** ON A/C FSN 001-999*** AMM TASKS 10-11-00-555-013-A & 10-11-00-555-069-B 1.7. Installation of the Aircraft Protection Equipment 1.7.1. Protection Total Air Temperature Sensor ☐ Installed (98A10001013000 or Equivalent) 1.7.2. Protection of the 3 pitot probes ☐ Installed (98A10001005000 or Equivalent) 1.7.3. Protection of the 3 angle-of-attack sensors ☐ Installed (98A10001500000 or Equivalent) 1.7.4. Protection of the 6 static probes ☐ Installed (98D10103500001 or Equivalent)	A/P	
6	**** ON A/C FSN 001-999*** AMM TASK 10-11-00-555-070-B 1.8. Installation of the Protection Devices on the Engines 1.8.1. Protection of the engine air intakes ☐ Installed (IAE1N20400 or Equivalent) 1.8.2. Protection of the engine exhaust nozzles ☐ Installed (IAE1N20008 or Equivalent)	A/P	
7	**** ON A/C FSN 001-999***	A/P	

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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	AMM TASK 10-11-00-555-013-A 1.9. Installation of the Protection Devices on the APU Area 1.9.1. Protection of the APU exhaust duct ☐ Installed (98D10007512000 or Equivalent) 1.9.2. Protection of the outlet duct of the APU oil cooler ☐ Installed (98D10007513000 or Equivalent)		
8	***** ON A/C FSN 001-999*** AMM TASK 12-13-79-610-011-A 1.10. Check Oil Level and Replenish **WARNING: DO NOT LET ENGINE OIL STAY ON YOUR SKIN FOR A LONG TIME. FLUSH THE OIL FROM YOUR SKIN WITH WATER. THE OIL IS POISONOUS AND CAN GO THROUGH YOUR SKIN AND INTO YOUR BODY.** **WARNING: BE CAREFUL WHEN YOU DO WORK ON THE ENGINE PARTS AFTER THE ENGINE SHUTDOWN. USE APPLICABLE THERMAL GLOVES. THE ENGINE PARTS CAN STAY HOT FOR ONE HOUR AFTER SHUTDOWN AND CAN BURN YOU.** **WARNING: DO NOT REMOVE THE FILLER CAP OF THE OIL TANK IMMEDIATELY AFTER ENGINE OPERATION. LET THE OIL PRESSURE DECREASE FOR A MINIMUM OF 5 MINUTES AFTER THE ENGINE SHUTDOWN. IF YOU OPEN THE FILLER CAP WHEN THERE IS PRESSURE IN THE TANK, THE OIL CAN BURN YOU DANGEROUSLY.** **CAUTION: IF THE ENGINE WAS STOPPED FOR MORE THAN 1 HOUR, DO THE SPECIFIC MAINTENANCE ACTION TO MAKE SURE THAT THE OIL LEVEL IS CORRECT. THIS WILL PREVENT OVER-SERVICING IF OIL IS ADDED.** **CAUTION: ADDING TOO MUCH OIL TO THE TANK CAN RESULT IN AMBER CROSS QUANTITY INDICATION WARNINGS AND SERVICE DISRUPTION.** 1.10.1. Check the engine oil level. (1) Where conditions permit, the oil tank should be checked and oil added, if necessary within a period of 5 to 60 minutes	A/P	

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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	after engine shutdown. (2) If the engine was stopped for between 1 and 10 hours, perform an idle run Ref. AMM TASK 71-00-00-710-012 of at least 3 minutes duration prior to servicing to ensure the oil level indicated on the sight glass is correct within a period of 5 to 60 minutes after engine shut down. (3) If the engine was stopped for more than 10 hours, dry crank the engine Ref. AMM TASK 71-00-00-710-045 followed by a start and idle run of a least 3 minutes duration before servicing to ensure the oil level indicated on the sight glass is correct within a period of 5 to 60 minutes after engine shut down. (4) Servicing is not required prior to departure if the engine was serviced on arrival or the oil tank sight glass shows a minimum of 23 quarts or the cockpit indication shows a minimum of 19 quarts and pre-departure walkaround checks confirm no oil pools/drips from drain mast. (5) If extended cranking has been undertaken e.g. following core washing, maintenance activity, etc, prior to servicing, run the engine at idle Ref. AMM TASK 71-00-00-710-012 for a minimum of 3 minutes. 1.10.2 Open the oil filler access door 437BL(447BL), in the left fan cowl 437AL(447AL) Ref. AMM TASK 71-13-14-010-801. 1.10.3 Remove the filler cap from the engine oil tank. (1)Lift the handle to open the cap. (2)Turn the handle to align the lock with the slots. (3)Remove the filler cap from the filler neck. 1.10.4 Fill the engine oil tank - Gravity Filling (1)Fill the engine oil tank with lubricant (engine oil). (2)Make sure that the oil level sight glass shows FULL. **NOTE:Do not fill the oil tank past the sight glass FULL level. Filling to tank overfill will result in excess oil, leading to amber cross indication warnings and service disruption.** 1.10.5 Install the oil tank filler cap (1)Examine the packing on the filler cap. Replace the packing if it is damaged Ref. AMM TASK 70-23-13-911-010. (2)Apply engine oil to the packing. (3)Make sure that the filler cap handle is in the open position. (4)Align the lock of the cap with the slots in the filler neck.		

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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	(5)Install the cap and turn the handle to the closed position. (6)Lower the handle to lock the cap. 1.10.6 Close Access (1)Make sure that the work area is clean and clear of tools and other items. **WARNING:MAKE SURE THAT ALL LATCHES ARE IN THE DISENGAGED POSITION BEFORE YOU CLOSE THE OIL FILLER DOOR. IF THE LATCHES ARE NOT CORRECTLY DISENGAGED BEFORE YOU CLOSE THE DOOR, IT IS POSSIBLE TO ENGAGE THE LATCHES INCORRECTLY. THIS CAN CAUSE THE DOOR TO BREAK OFF AND CAUSE INJURY.** (2)Close the oil filler access door Ref. AMM TASK 71-13-14-410-801.		
9	***** ON A/C FSN 001-999*** AMM TASK 12-13-49-612-001-A & 49-90-00-600-004-A 1.12. Do a check of the APU oil level (on the sight glass). Add approved oil to the oil reservoir as necessary. Record uplift in the aircraft technical log. Record the data below if there are APU oil added : A/C Reg. : _____ APU SN : _____ APU Hour : _____ APU Cycle : _____ Oil Added : _____ (Qrt) **NOTE:** Please take attention to fill the data above because it will be useful for Analysis APU Oil consumption. **WARNING: PUT THE OIL INTO THE GRAVITY-FILL PORT SLOWLY. THIS WILL PREVENT A BLOW-BACK OF HOT OIL WHICH CAN BURN YOU** **CAUTION:DO NOT FILL THE GEARBOX ABOVE THE FULL MARK ON THE SIGHT GLASS. DURING APU OPERATION THE OIL EXPANDS WHICH COULD CREATE AN OVER-FILLED CONDITION RESULTING IN LEAKAGE AND POSSIBLE OIL SMELL IN THE CABIN.**	A/P	

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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
10	**** ON A/C FSN 001-999*** AMM TASKS 12-12-29-611-001-A 1.13. Do a fluid level check of the Green, Blue and Yellow hydraulic reservoirs (On ECAM LOWER DISPLAY, HYD page). Add Phosphate Ester Base hydraulic fluid (Mat. No. 02-003) to the reservoirs as necessary (Ref. AMM 12-12-29-03). Record uplift(s) in the aircraft technical log.	A/P	
11	**2.0 EXTERNAL WALK AROUND**	A/P	
12	**** ON A/C FSN 001-999*** AMM TASKS 05-21-00-200-001, 05-22-00-200-001, 05-28-00-200-001 **2.1 FORWARD FUSELAGE** Do a general visual inspection of the forward fuselage from ground as far as visible, including: - Passenger/Crew doors, cargo compartment door, ☐ Checked - Service panel doors, avionic compartment access doors, Checked - Pitot probes, static ports and AOA sensors: Checked No obstruction (Protective covers removed), - Avionics equipment ventilation: Air inlet and Checked air outlet valves, - Batteries ventilation outlet (Venturi): No obstruction, Checked - Waste water drain mast, Checked - Antennas: No damage, Checked - Wing and engine scan lights: Cleanliness, Checked - Radome: No damage, latches engaged and locked, Checked - Crew oxygen cylinder overpressure indicator: Checked Green disc in place. - FWD Cargo - VIsual Check of Cargo Compartment Checked Decompression, Lining, FLoor Panel and Pressure Compensating Valve (as far as visible) - FWD Cargo- VIsual Check of Divider Nets, Door Net, Checked	A/P	

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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	Net Attachment Point and Light		
13	***** ON A/C FSN 001-999*** AMM TASK 05-27-00-200-001 **2.2 NOSE LANDING GEAR AND DOORS** Do a general visual inspection of the nose landing gear and doors from ground as far as visible, including: - Doors and wheel well, ☐ Checked - Gear assy structure: Damage, evidence of leakage, ☐ Checked - Shock absorber: Signs of leakage and normal extension, ☐ Checked - Shock absorber sliding tube: Scoring, damage, ☐ Checked cleanliness, **NOTE:** Please pay attention to the cleanliness of the Shock Absorber chromium surface. Clean the sliding tube from any dust, dirt, or other unwanted material that could potentially erode the dynamic seal and led to hydraulic leakage. (Clean using a lint-free cloth moist with mineral hydraulic fluid (Mat. No. 02-001), clean the tube and dry with a clean cloth) - NLG gear actuating-cylinder and Down Lock Actuator: Signs of leakage,normal extension, damage/corrosion, cleanliness ☐ Checked **NOTE**: Ref AMM 32-22-00-010/410-001 to open and close the NLG Door for inspection. Put a small quantity of Phosphate Ester Hydraulic Fluid-General Power - - (Material Ref. 02ABA1) on a clean Textile-Lint free Cotton - (Material Ref. 14SBA1) and apply a thin layer phosphate Ester Hydraulic Fluid-General Power - - (Material Ref. 02ABA1) only to the chromium surfaces of: • The NLG gear actuating-cylinder • The NLG downlock actuator.	A/P	

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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	- Taxi and take-off lights, runway turnoff lights: Cleanliness, ☐ Checked - Wheels: Rim damage, sheared/missing tie bolts, ☐ Checked - Tires: Wear, damage and signs of under inflation. ☐ Checked		
14	***** ON A/C FSN 001-999*** AMM TASK 05-21-00-200-001, 05-22-00-200-001, 05-28-00-200-001 **2.3 CENTER FUSELAGE** Do a general visual inspection of the forward center fuselage from ground as far as visible, including: - Emergency exits, ☐ Checked - Service panel doors, ☐ Checked - Wing-to-fuselage fairings and belly fairing, ☐ Checked - Anti-collision beacon lights: Cleanliness, ☐ Checked - Drain mast, ☐ Checked - Ram air inlet flap, ☐ Checked - Pack air intakes and outlets: No obstruction, ☐ Checked - Ram air turbine doors: Correctly closed, ☐ Checked - Antennas: No damage. ☐ Checked	A/P	
15	***** ON A/C FSN 001-999*** AMM TASK 05-25-00-200-001-A **2.4 RIGHT/LEFT WING LEADING EDGE** Do a general visual inspection of the RH and the LH wing leading edge from ground as far as visible, including: - Leading edge slats, ☐ Checked	A/P	

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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	- Leading edge access panels ☐ Checked (Pressure relief panels in place), - Surge tank NACA air intake: No obstruction Surge ☐ Checked tank overpressure protector (White cross is visible).		
16	***** ON A/C FSN 001-999*** AMM TASK 05-24-00-200-001 **2.5 RIGHT/LEFT POWER PLANT AND PYLON** Do a general visual inspection of the power plant and pylon from ground as far as visible, including (cowl doors closed): - Pylon with fairings and fillets: No damage, ☐ Checked - Engine air inlet: Lip skin, riveting, acoustic panels, and ☐ Checked visible probes and sensors (T12), - Fan blades and spinner: No damage, ☐ Checked check free rotation by hand, - Fan cowl doors and thrust reverser cowl doors: ☐ Checked - Cowl doors closed and correctly latched, ☐ Checked - Blocker doors in stowed (closed) position, - Access doors and pressure relief doors in place, ☐ Checked closed and latched, - Air inlets/outlets: No obstruction, ☐ Checked - Fan exhaust: Acoustic lining, exit vanes and ☐ Checked struts, thrust reverser pivoting doors, - Turbine exhaust: Last stage LPT blades, ☐ Checked nozzle and plug (for damage and metal deposit), - Drain mast and pylon drains: No obstruction, ☐ Checked evidence of leakage.	A/P	
17		A/P	

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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	***** ON A/C FSN 001-999*** AMM TASK 05-25-00-200-001 **2.6 RIGHT/LEFT WING TIP AND TRAILING EDGE** Do a general visual inspection of the RH and the LH wing tip and trailing edge from ground as far as visible, including: - Navigation and strobe lights, ☐ Checked - Static dischargers, ☐ Checked - Control surfaces, flaps and flap track fairings: Damage, evidence of fluid leakage, ☐ Checked - Lower wing surface: Damage, evidence of fuel leakage, ☐ Checked - Wing tank overpressure protector: Burst disk in place (White cross visible), ☐ Checked - Landing light: No damage, cleanliness. ☐ Checked		
18	***** ON A/C FSN 001-999*** AMM 05-27-00-200-001 **2.7 RIGHT/LEFT MAIN LANDING GEAR AND DOORS** Do a general visual inspection of the main landing gears and doors from ground as far as visible, including: - Doors and wheel well, ☐ Checked - Gear assy structure: Damage, evidence of leakage, ☐ Checked - Shock absorber: Signs of leakage and normal extension, ☐ Checked - Shock absorber sliding tube: Scoring/damage, cleanliness,(if necessary using lint - free cloth moist with mineral hydraulic fluid (Mat. No. 02-001), clean tube and dry with a clean cloth), ☐ Checked - Proximity detectors: Security, cleanliness, ☐ Checked - Brake units: Evidence of leakage or overheating, ☐ Checked	A/P	

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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	- Wheels: Rim damage, sheared/missing tie bolts, ☐ Checked - Tires: Wear, damage and signs of under inflation. ☐ Checked		
19	**** ON A/C FSN 001-999*** AMM TASK 32-42-27-210-003 **2.8 MAIN LANDING GEAR BRAKES** (MP TASK 324227-01-1) Check Heat Pack Wear Indicator (Parking Brake Applied). ☐ Checked (1) On parking brake panel 110VU, set parking brake control-switch 73GG to the ON position. (2) For an in-service brake Measure the wear indicators when the brake is cold (brake temperature less than 60 deg.C (140 deg.F)). Brake #1: _____ mm Brake #2: _____mm Brake #3: _____mm Brake #4: _____mm NOTE : Replace if Brake PIN Flush (1 mm/30 cycle). NOTE: Do this test only if the remaining length of the wear indicators was not measured on the aircraft. CAUTION: MAKE SURE THAT NO HYDRAULIC FLUID FALLS ON THE HEAT PACK. THROUGHOUT ALL THE TESTS, MAKE SURE THAT THERE IS NO EXTERNAL LEAKAGE THROUGH THE WALLS. CAUTION: DO NOT PRESSURIZE THE CARBON BRAKE THROUGH THE TWO CIRCUITS AT THE SAME TIME. (a) Apply a pressure of 145 bar (2100 psi) to one of the two circuit of the carbon brake. (b) Measure the length of the wear indicators. 1 If the end of one of the two wear indicators is flush with the machined surface of the piston housing (heat pack fully worn), replace the brake (Ref. AMM TASK 32-42-27-000-001) (Ref. AMM TASK 32-42-27-400-001) .	A/P	

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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	NOTE: The brake units have two wear indicators. But if a brake does not have the two wear indicators: − The brake continues to be serviceable if there is one wear indicator which is in the limits. − If there is no serviceable wear indicator, do the deactivation of the brake (Ref. AMM TASK 32-42-00-040-003) . (3) For a new brake: − With the parking brake on, measure distance D between the end of the wear indicators and the machined surface of the piston housing. − Make sure that distance D is between 49.6 mm (1.953 in.) and 50.4 mm (1.984 in.). − If distance D is out of the limits, replace the brake (Ref. AMM TASK 32-42-27-000-001) (Ref. AMM TASK 32-42-27-400-001). (4) Do a check of the components that follow to make sure that there are no leaks: - The piston housing - The piston - The hydraulic lines. ☐ Checked		
20	***** ON A/C FSN 001-999*** AMM TASK 32-41-00-210-003 **2.9 WHEELS** (MP TASK 324100-01-1) Do a check of the tires for correct pressure. Checked (sub task 32-41-00-210-058-A) To do this, refer to the pressure table. and compare the tire pressure values with those that you find when you use: · A GAGE (14-6806-6011 or Equivalent) - PRESSURE, TIRE , or · The wheel-mounted tire-pressure indicators, if installed, or · The Tire Pressure Indicating System (TPIS), if installed. **CAUTION:** TIRE PRESSURE MUST BE CHECKED AND CORRECTED	A/P	

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ACCOMPLISHMENT

NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	WHEN THE TIRES ARE COLD. SERVICE WITH NITROGEN ONLY. IF THE TIRE PRESSURE IS OUT OF LIMIT PERFORM NECESSARY ACTION I.A.W. AMM. **NOTE: Note tire pressure before inflation.** NLG: LH _____ psi RH_____psi MLG: LH/IB _____ psi LH/OB _____ psi MLG: RH/IB _____ psi RH/OB_____ psi		

	A/C WEIGHT	TIRE DIMENSIONS	INFLATION PRESSURE	
			UNLOADED / BARS (PSI)	LOADED BARS (PSI)
Main Wheel	73.9 t	46 X 17 R20	13.3 (193)	13.8 (200)
Nose Wheel	73.9 t	30 X 8.8 R15	11.8 (171)	12.3 (178)
NOTE: Please Check the Tire Dimension				

TIRE PRESSURE	TIRE PRESSURE	
NLG 30X8.8-15 30X8.8R15	**MLG 46X16-20 46X17R20**	**TIRE CONDITION / ACTION NECESSARY**
More than 12.9 bars (187 psi)	More than 14.5 bars (210 psi)	Over-inflation. Make sure that the measurement is correct with another pressure gage. If it is correct, deflate the tire to the maximum normal pressure.
From 12.3 to 12.9 bars (178-187 psi)	From 13.8 to 14.5 bars (200-210 psi)	Normal pressure range. Do not adjust the tire pressure.
From 11.7 to less than 12.3 bars (170-178 psi)	From 13.1 to less than 13.8 bars (190-200 psi)	Inflate the tire to the maximum normal pressure.
From 11.1 to less than 11.7 bars (161-170 psi)	From 12.4 to less than 13.1 bars (180-190 psi)	Inflate the tire to the maximum normal pressure. You must measure the tire pressure again the next day. If the tire is under-inflated again, you must replace the wheel.
From 9.9 to less than 11.1 bars (144-161 psi)	From 11 to less than 12.4 bars (160-180 psi)	You must replace the wheel.
From 0 to less than 9.9 bars (0-144 psi)	From 0 to less than 11 bars (0-160 psi)	You must replace the wheel and the adjacent wheel.

21	***** ON A/C FSN 001-999*** AMM TASK 05-21-00-200-001-1, 05-22-00-200-001-1, 05-28-00-200-001 **2.10 AFT FUSELAGE** Do general visual inspection of the aft fuselage from ground as far as visible, including: - Passenger/Crew, cargo compartment doors, Checked	A/P	

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Lion Group

TASKCARD

WORK ORDER NO.	A/C REG.	A/C MSN.	A/C Effectivity	OPERATOR	TASK CARD NO.
					A320-052000-99-2-SJV-IDN

ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	- Service panel doors, Checked - Antennas and beacon, Checked - Waste water drain mast, Checked - Outflow valve, Checked - Fuselage tail damage due to ground contact. Checked - AFT Cargo - Visual Check of Cargo Compartment Checked Decompression, Lining, Floor Panel and Pressure Compensating Valve (as far as visible) - AFT Cargo - VisualCheck of Devider Nets, Checked Door Nets, Net Attachment Point and LIght		
22	***** ON A/C FSN 001-999*** AMM TASK 05-23-00-200-001 **2.11 STABILIZERS AND CONE/REAR FUSELAGE** Do a general visual inspection of the rear fuselage with cone and stabilizers from ground as far as visible, including: - THS and elevators: Damage, evidence of fluid leakage, Checked - Fin and rudder: Damage, evidence of fluid leakage, Checked - Static dischargers, Checked - APU Area: Checked - Air intake: Flap closed or open (If APU is Off or On), Checked - Doors: Closed and secured, Checked - Drain mast, drains and vents, Checked - Exhaust, Checked - Fire extinguisher overpressure discharge indicator: Checked Red disc in place,	A/P	

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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	- Tail cone strobe light. Checked		
23	**3.0 WASTE/WATER SERVICING**		
24	***** ON A/C FSN 001-999*** AMM TASK 12-15-38-613-001-A 3.1.1. Service the potable water system as required (15). AMM TASK 12-15-38-613-002-A 3.1.2. Service the toilet waste tanks as required (12). In freezing conditions, obey the cold weather maintenance.	A/P	
25	**4.0 FLIGHT COMPARTMENT**		
26	***** ON A/C FSN 001-999*** 4.1 OVER WING SLIDE AND CREW OXYGEN SYSTEM On the lower ECAM DU, on the DOOR/OXY page: 1. Make sure that the over wing is indicated in SLIDE. If the SLIDE not shown, do a check of the target switch at overwing and release if target switch release out. 2. And makesure that the oxygen pressure is indicated in Green. If the Amber hal frame is shown, do a check of the oxygenessure for the flight crew and service the system as required. COB : _____ psi NOTE : Please Refer to FCOM for Limitation Oxygen.	A/P	
27	***** ON A/C FSN 001-999*** 4.2 EXTERNAL LIGHTS Do an operational check of AMM TASK 33-41-00-710-001-A 1. Navigation Lights, Landing Lights, ☐ Tested AMM TASK 33-42-00-710-001-A	A/P	

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TASKCARD

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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	2. Landing Light ☐ Tested AMM TASK 33-43-00-710-001-A 3. Runway Turnoff Lights, ☐ Tested AMM TASK 33-46-00-710-001-A 4. Taxi and Takeoff Lights, ☐ Tested AMM TASK 33-47-00-710-002- 5. Logo Lights, ☐ Tested AMM TASK 33-48-00-710-001-A 6. Anti-collision/Beacon ☐ Tested AMM TASK 33-49-00-710-001-A 7. Lights, Wing and Engine Scan Lights. ☐ Tested		
28	***** ON A/C FSN 001-999*** AMM TASK 33-13-00-710-001-A , 33-51-00-710-008-A , & 23-73-63-710-058-A 4.3 Do a check of the: - flight compartment lighting, including stby compass light Checked - missing spare bulbs/fuses, replace as necessary, Checked - emergency equipment for presence and correct stowage, Checked - flight compartment windshields and the side windows for damage and cleanliness, Checked - flight compartment for general condition and cleanliness. ☐ Checked - operational test of the Emergency Lighting in the cabin with the "EMER EXIT LT" switch ☐ Checked - functional test NO SMOKING / NO MOBILE DEVICE / - FASTEN SEAT BELT SIGN ☐ Checked - Do general visual inspection Captain & FO seat condition. ☐ Checked	A/P	
29	***** ON A/C FSN 001-999***	A/P	

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Lion Group

TASKCARD

WORK ORDER NO.	A/C REG.	A/C MSN.	A/C Effectivity	OPERATOR	TASK CARD NO.
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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	AMM TASK 31-33-00-710-007-A & 237100-710-001-A 4.4 DIGITAL FLIGHT DATA RECORDER & COCKPIT VOICE RECORDER (EASA SIB No.: 2009-28R1) (MP TASK 237135-99-1 & 313300-99-1) (CASR 91 Appendix E) 4.4.1 Operational test of the DFDR Tested 4.4.2 Operational test of The CVR. ☐ Tested		
30	***** ON A/C FSN 001-999*** AMM TASK 71-00-00-710-045-A 4.5. Perform Dry-Motoring Check ☐ Tested NOTE: with exception to not perform: Subtask 71-00-00-010-089-A,Subtask 71-00-00-040-067-A, Subtask 71-00-00-010-091-A, Subtask 71-00-00-410-090-A, Subtask 71-00-00-440-066-A, Subtask 71-00-00-410-092-A,		
31	*****ON A/C FSN 001-999*** TASK 21-26-00-710-001-A 4.6. Operational Check of the Avionics Equipment Ventilation ☐ Tested System via MCDU (MP TASK: 212600-01-1)	A/P	
32	***ON A/C FSN ALL*** On RMP1 (2 or 3), push VHF3 push button, make sure ACARS/DATA indication is shown in the ACTIVE window. if no, push the transfer pushbutton switch (double arrow) between thetwowindow	A/P	

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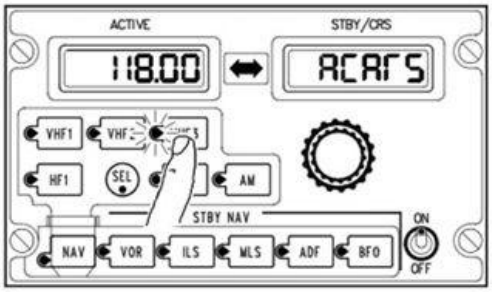
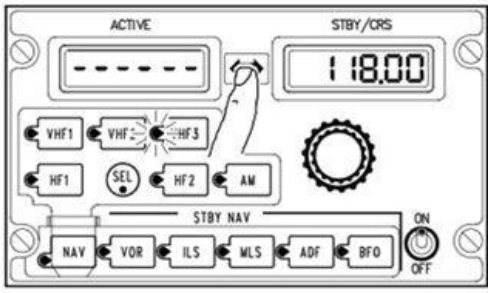
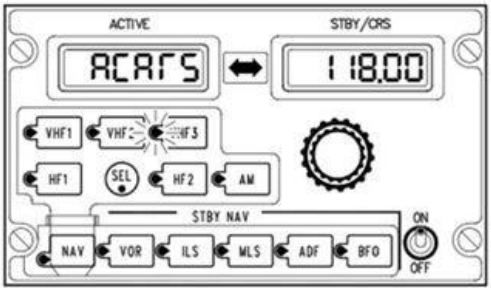
					A320-052000-99-2-SJV-IDN

ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	  		
33	**5.0 PASSENGER COMPARTMENT**		
34	***** ON A/C FSN 001-999*** 5.1. Check passenger compartment for general condition and cleanliness. ☐ Checked 5.2. Check of the galleys for general condition, cleanliness and evidence of water leakage. ☐ Checked 5.3. Do a check of the lavatories for general condition, cleanliness and evidence of leakage. ☐ Checked 5.4. Check the closing flap of the waste bins for correct operation. ☐ Checked	A/P	

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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	5.5. Do a check of the emergency equipment for presence and correct stowage. ☐ Checked 5.6. Examine the hermetic seal of the first aid Kits for condition. If seal is broken, take action as required. ☐ Checked 5.6. Do a general visual check of the left and right wing upper surface and control surfaces, through cabin windows. ☐ Checked		
35	***** ON A/C FSN 001-999*** **6.0. EMERGENCY EQUIPMENT** 6.1. Check aircraft emergency equipment for condition,proper pressure, safeties, security and validity date including the following: a) DOOR - Escape Slide / Raft ☐ Checked b) PASSENGER CABIN - Halon / Water Fire extinguisher ☐ Checked - PBE ☐ Checked - Oxygen Bottle ☐ Checked - First Aid Kit ☐ Checked	A/P	

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TASKCARD

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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	c) LAVATORY - Fire extinguisher ☐ Checked		
36	***** ON A/C FSN 001-999*** AMM TASK 52–10–00–210–004-A **7.0. PASSENGER/CREW DOOR** (MP TASK 521000-01-1) 7.1. Check emergency cylinder/accumulator pressure. FWD : R/H _____PSI L/H_____PSI AFT : R/H_____PSI L/H_____PSI NOTE: IF PRESSURE IS BELOW LIMITATION PLEASE SERVICING DOOR DAMPER	A/P	
37	**8.0. WINGS**		
38	***** ON A/C FSN 001-999*** AMM 12-32-28-281-001 8.1. TANKS (MP TASK 281100-01-2) Drain water from wing tanks and center tank at water drain valves. NOTE: If possible, do the water drain procedure before a refuel. Alternatively, you must wait for one hour after a refuel has been completed before you do the water drain procedure. To get access to the high water drain valves from the ground, attach the: - ADAPTING PIPE - WATER DRAINING FROM FUEL (98A28101000000) or - PIPE ADAPTING-WATER DRAINING (98D28104000000) to the bottle and funnel of the PURGER - WATER DRAIN (98A28104000000 or Equivalent) . After drainage, make sure that the drain valves are correctly closed and not leaking.	A/P	
39	**9.0. FINAL ITEMS**		
40	***** ON A/C FSN 001-999*** 9.1. Review AFML, CML, DMI, NSRDIL, DBC, Deferred Item (from. Phase check or C Check)	A/P	
41	Revision item 9.2, paragraph d and e ***** ON A/C FSN 001-999***	A/P	

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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	9.2. Check the following aircraft document for completeness and validity: a) Certificate of Airworthiness (C of A). Valid until b) Certificate of Registration (C of R). Valid until c) Radio Permit & AASL Valid until d) Weight and Balance. Valid until e) Swing Compass Valid until f) Insurance Certificate Valid until g) Operation Specifications (OPSPEC) and ACL. Last Revision Date h) Required Navigation Performance (RNP). Date issued i) Reduced Vertical Separate Minimal (RVSM). Date issued j) Noise Certificate Date issued k) Minimum Equipment List Date issued Report to MCC if there is/are document(s) has expired (not updated)		
42	***** ON A/C FSN 001-999***	A/P	

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ACCOMPLISHMENT			
NO.	INSTRUCTION	PERFORMED BY	INSPECTED BY
	9.3. - Check corrective actions in the aircraft technical log and record each corrected or carried forward item. Record the DAILY check in the aircraft technical log. — Checked - De-energize the aircraft electrical network. Make sure that the cockpit sliding windows are closed. — Removed - Make sure that all passenger/crew doors, cargo compartment doors and service panel doors are correct closed and secured. Install the protective covers on the aircraft as necessary. Make sure that an authorized person releases the aircraft. — Checked		
43	**THE END OF THE TASK**		

START TIME(UTC)	FINISH TIME(UTC)	TOTAL MAN HOUR		DEFECT FOUND Y N M.D.R.R. No:
		EST.	ACTUAL	
		2.25		

TASK CARD RELEASE

DATE (UTC) : ..... TIME (UTC) : ..... SIGNATURE : ..... AUTHORIZATION NO. : .....

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TASKCARD

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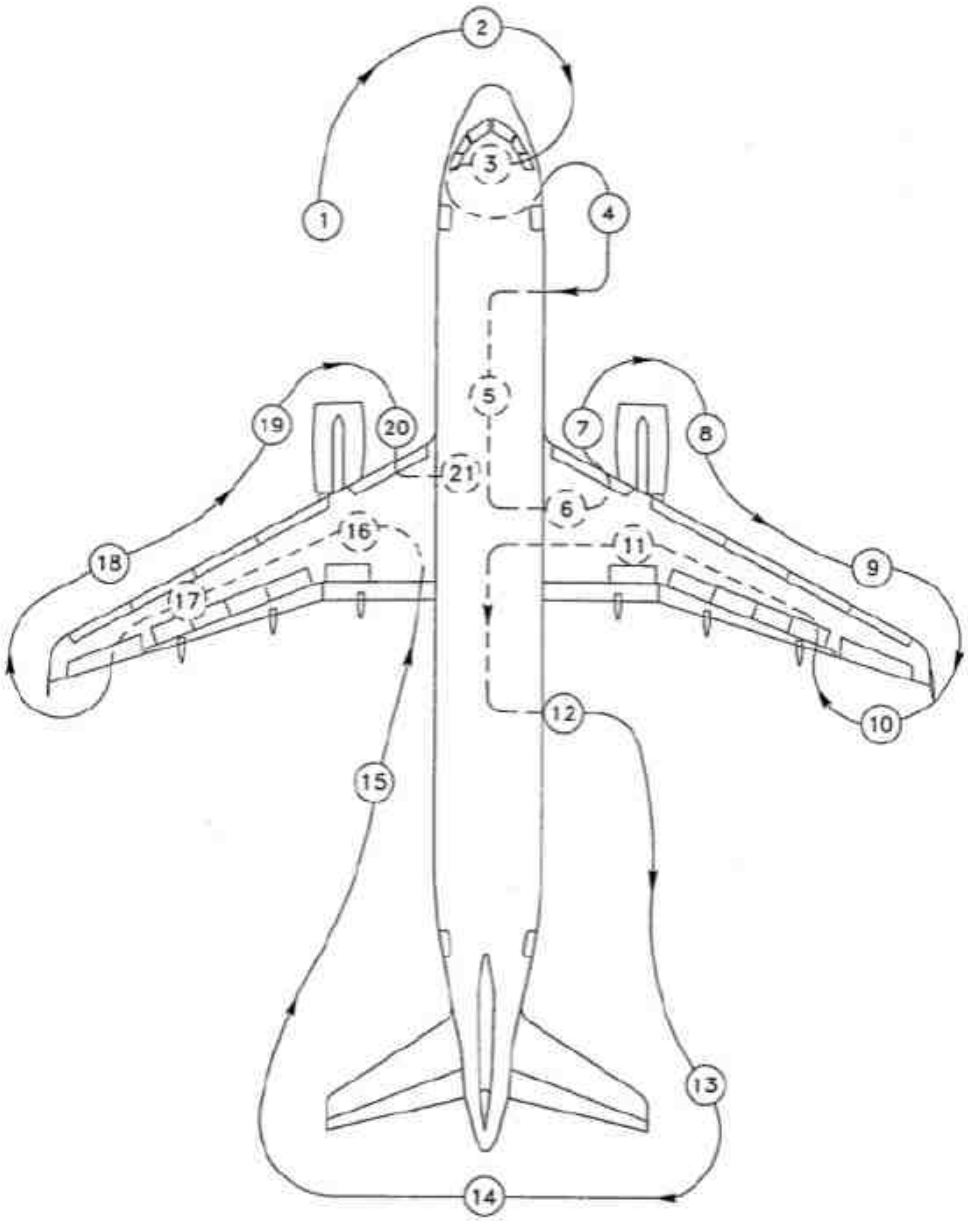
APPENDIX

GRAPHIC
PICTURE.JPG

EXTERNAL WALK AROUND DIAGRAM



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