

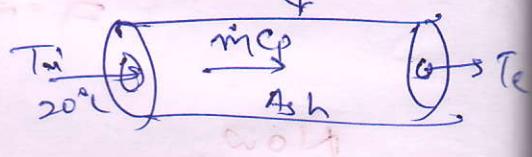
rate of decay depends on the magnitude of $\left(\frac{hA_s}{mc_p}\right)$

$$\left(\frac{hA_s}{mc_p}\right) = NTU \text{ (number of transfer unit)}$$

is measure of effectiveness of heat transfer system (Non-dimensional no.)

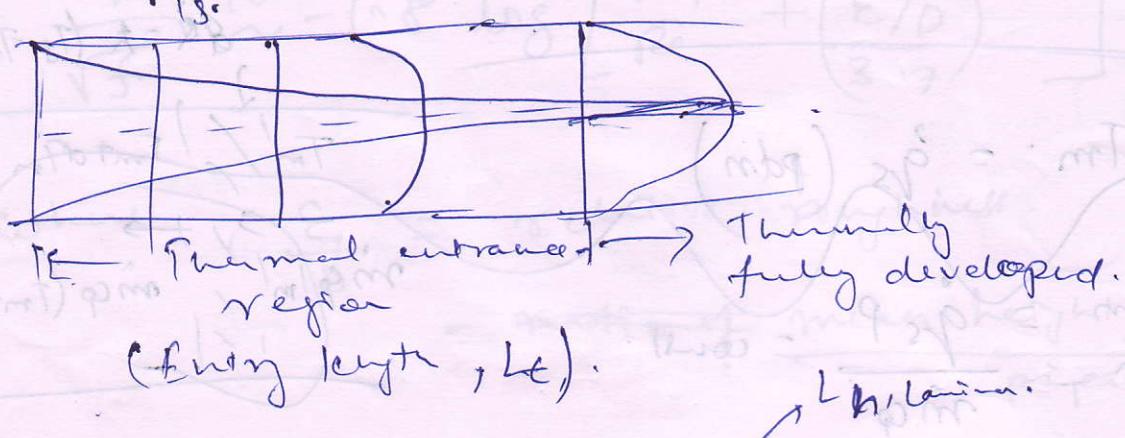
Typically

NTU	$T_e ({}^\circ C)$
0.01	20.8
0.5	23.9 51.5
1	27.0 60.6
5	99.5



So, NTU → give indication of effectiveness
 → Once beyond 5 or 6 may
 or length of tube will be not
 add any value.

Boundary flow

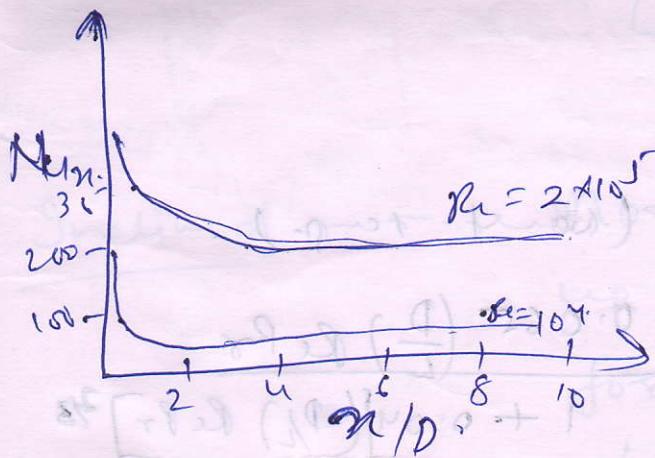


$$L_{t, \text{Laminar}} \approx (0.05 \text{ Re}) \text{ Pr} \cdot D$$

$$\text{For } \text{Re} = 20, \text{ Pr} = 1, L_t = D.$$

$$\text{Re} = 2300 \text{ (Laminar cond.)}, L_t = 115D.$$

$$L_{t, \text{turbulent}} \approx 10D.$$



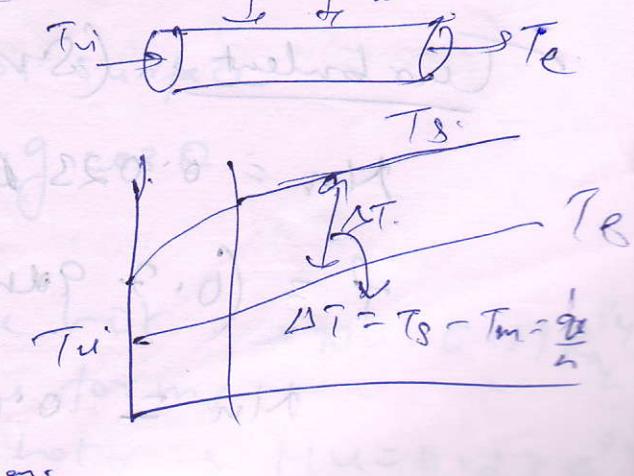
Const. Surface heat flux:

$$\dot{q}_s A_s = m c_p (T_e - T_i)$$

$$\therefore T_e = T_i + \frac{\dot{q}_s A_s}{m c_p} \rightarrow \text{linear}$$

Surf. temp.

$$\dot{q}_{STF} = h_A (T_s - T_m) \rightarrow T_s = T_m + \frac{\dot{q}_s}{h_A}$$

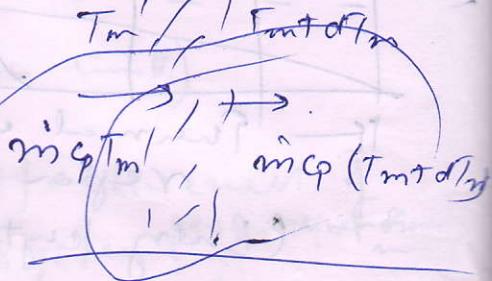


$$T_S - T_m = \Delta T = \text{const.}$$

$$\delta Q = k(T_S - T_m) dA$$

$$\text{mcp} \cdot T_m = q_{rs} (\text{pdn})$$

$$\Rightarrow \frac{dT_m}{dr} = \frac{q_{rs} P}{\text{mcp}} = \text{const.}$$



Reynold No. - (fully developed). ~ Laminar

Circular tube, const. flux: (q_s)

$$\lambda u = \frac{\kappa D}{P} = 4.36,$$

const. surf. Temp (T_S)

$$\lambda u = 3.66.$$

Entrance region ' (const. temp.)

$$\lambda u = 3.66 + \frac{0.065 \left(\frac{D}{L} \right) Re Pr}{1 + 0.04 \left[\left(\frac{D}{L} \right) Re Pr \right]^{2/3}}$$

Turbulent (Smooth)

$$\lambda u = 0.023 Re^{0.8} Pr^{1/3}$$

$0.7 \leq Pr \leq 160$
 $Re > 10,000$

$$f = (6.7 + 9 \ln Re - 1.64)^{-2} \quad (3000 < Re < 5 \times 10^6)$$

$$\lambda u = 0.125 f Re^{0.8} Pr^{1/3}$$

Rough surf.

$$\frac{1}{\sqrt{f}} \approx -1.8 \log \left[\frac{6.9}{Re} + \left(\frac{\epsilon/D}{3.7} \right)^{1.11} \right] \quad (\text{for rough})$$

where $\frac{\epsilon}{D}$ = relative roughness

= ~~ratio of mean height of roughness~~ $\frac{\text{mean height of roughness}}{\text{pipe diameter}}$

Free convection $\rightarrow Nu = f(Ra_L, \beta)$

Vertical $\rightarrow Nu = \left[0.825 + \frac{0.382 Ra_L^{1/8}}{\left[1 + \left(\frac{0.492}{Pr} \right)^{9/16} \right]^{8/27}} \right]^2$

Inclined \rightarrow More on hot surface facing up than vertical \rightarrow Plumes form \rightarrow reduces boundary layer thickness and resistance to heat transfer.

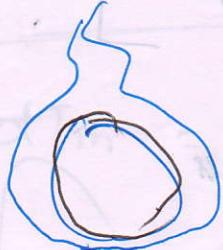
\rightarrow lower hot surface

$$\cancel{g'} = g \cos \theta$$

Horizontal \rightarrow Top surface hot $\rightarrow Nu = 0.59 Ra_L^{4/5}$ \rightarrow plumes form.

\rightarrow lower surface hot $\rightarrow Nu = 0.27 Ra_L^{4/5}$ \rightarrow low Nu

Horizontal cylinders & spheres



- BLT increases from bottom to top
- Local Nu at bottom is higher than that at top.

Natural Convection Inside Enclosure

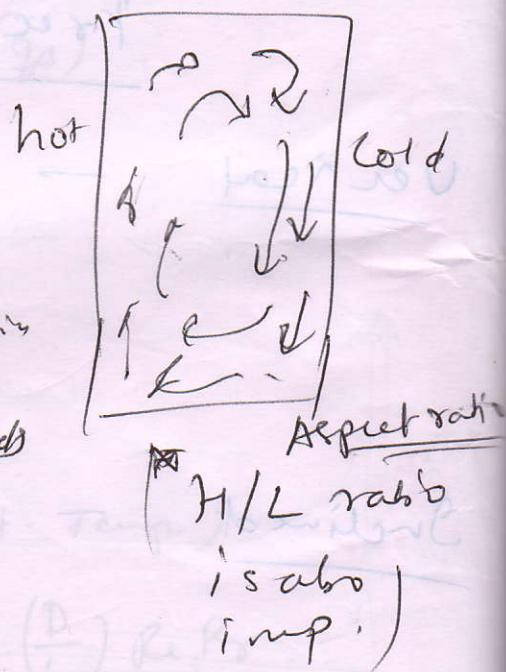
If top is hot

↳ pure conduction,
↳ $\text{Nu} \approx 1$

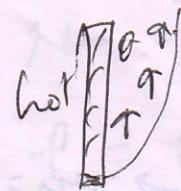
If bottom is hot

↳ If $\text{Ra}_L \ll 1708$, pure conduction

↳ $\text{Ra}_L \geq 1708 \rightarrow$ natural convection



Combined Natural & Forced Convection ($\frac{\text{Gr}}{\text{Re}^2}$)



↑↑↑

Assisting flow



↓↓↓

(Opposing flow)



Transverse flow

$$(Nu)_{\text{combined}} = (Nu_{\text{forced}}^n \pm Nu_{\text{natural}}^n)^{1/n}$$

assisting / transverse $\rightarrow +ve$
 opposing $\rightarrow -ve$.

$$n \rightarrow 3-4.$$

Tutorial - 9

Q. Heat loss through Double-pane window

$$K = 0.02416 \text{ W/m.K}$$

$$Pr = 0.7344, \text{ width} = 2 \text{ m}$$

$$\gamma = 1.4 \times 10^{-5} \frac{\text{m}^2}{\text{s}}$$

$$\beta = \frac{1}{T_{\text{avg}}} = \frac{1}{280}$$

$$L_c = L = 0.02 \text{ m}$$

$$Ra_2 = \frac{g \beta (T_1 - T_2) L_c^3}{\gamma^3} Pr. \quad * \text{Heat loss through radiation}$$

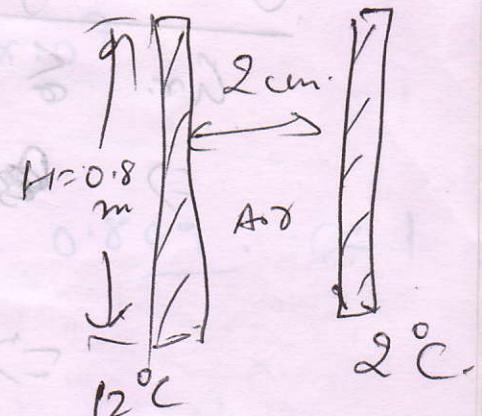
$$= \frac{9.81 \times [1/280][12-2](0.02)^3}{1.4 \times 10^{-5}} (0.7344)$$

$$= 1.05 \times 10^7$$

$$\text{Aspect Ratio}, \frac{H}{L} = \frac{0.8}{0.02} = 40.$$

$$\therefore Nu = 0.42 Ra_2^{1/4} Pr^{0.012} \left(\frac{H}{L}\right)^{-0.3}$$

$$= 1.4.$$



$$A_s = H \times W = 0.8 \times 2 = 1.6 \text{ m}^2$$

$$\text{Q} = h A_s (T_1 - T_2)$$

$$= \left(\frac{1 \leq \text{Nu}}{2} \right) A_s (T_1 - T_2)$$

$$= \left(\frac{0.02416 \times 1.4}{0.02} \right) \times 1.6 \times (12 - 2)$$

$$= \underline{27.1 \text{ W}}$$

Condⁿ for Nu = 1

$$\text{Gr.} \leq 1708$$

$$\Rightarrow L_c^3 \leq \frac{1708}{\left(g \beta (T_1 - T_2) \rho \right)} \quad Q = 1.3 \times 10^{-6}$$

$$\Rightarrow L_c < 0.011 \text{ m (11 cm)}$$

Radiative heat loss

$$= \epsilon \sigma A_s (T_h^4 - T_c^4)$$

$$= 1 \times 5.67 \times 10^{-8} \times (2 \times 0.8) \times (285^4 - 235^4)$$

$$= \underline{79.68 \text{ W}}$$

* Glass acts as insulator for IR radiation.

Q. 0.2×0.2 vertical plate.

$K_{air} = 0.02588 \frac{W}{m \cdot K}$

$\nu = 1.608 \times 10^{-5} \frac{m^2}{s}$

$\Pr = 0.7282$

$\beta = \frac{1}{T_f} = \frac{1}{\frac{(313 + 293)}{2}} = 0.0033 \frac{K^{-1}}{m}$

$$Re = \frac{VL}{\nu} = \frac{0.4 \times 0.2}{1.608 \times 10^{-5}} = 4925$$

$$Gr_L = \frac{g \beta (T_s - T_b) L^3}{\nu^2} = 2 \times 10^7$$

$$\therefore \frac{Gr_L}{Re^2} = \frac{2 \times 10^7}{(4925)^2} = 0.809 \approx 1$$

(Both natural & forced convection are significant)

$$Ra = \frac{h_r \cdot \Pr}{\nu}$$

$$Nu_{natural} = 0.59 Ra_L^{0.4} = 0.59 (2 \times 10^7 \times 0.7282)^{0.4} = 36.4 L$$

$$Nu_{forced} = 0.664 Re^{0.5} Pr^{0.3} = 42.44$$

\therefore Assisting flow,

$$Nu_{combined} = (Nu_{nat} + Nu_{for})^{1/3}$$

Opposing flow $= 29.8 W$

$$Nu_{comb} = (Nu_{nat}^3 - Nu_{for}^3)^{1/3} = 29.8 W$$

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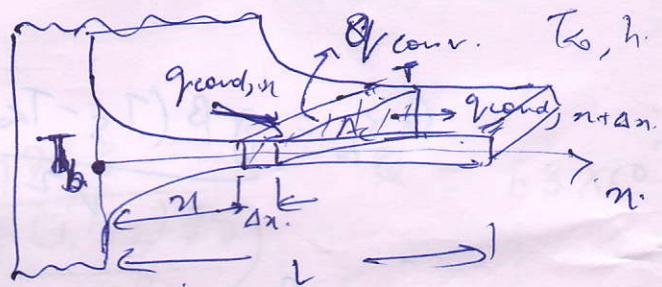
Finned Surface

$$q_{\text{conv}} = hA(T_s - T_o) \quad \text{fixed}$$

to ↑ q_{conv} → ↑ L or ↑ A_s ,
 ↓ Fin.

Too many fins. → Restricts convective flow

fin equation



under steady state condition, energy balance.

$$\left(\begin{array}{l} \text{Rate of heat} \\ \text{conduction into} \\ \text{the element } n \end{array} \right) = \left(\begin{array}{l} \text{Rate of} \\ \text{heat cond.} \\ \text{from } n+1 \end{array} \right) + \left(\begin{array}{l} \text{Rate of} \\ \text{heat conv.} \\ \text{from element } n \end{array} \right)$$

$$\Rightarrow q_{\text{cond},n} = q_{\text{cond},n+1} + q_{\text{conv}}.$$

$$\text{where, } q_{\text{conv.}} = h(\rho \Delta n)(T - T_b).$$

$$\therefore \frac{q_{\text{cond},n} - q_{\text{cond},n+1}}{\Delta n} + hp(T - T_b) = 0.$$

as $\Delta n \rightarrow 0$,

$$\frac{dq_{\text{cond}}}{dn} + hp(T - T_b) = 0$$

Now, from Fourier's law, $\frac{d\theta}{dn} + h_p \theta = -KA_c \frac{dT}{dn}$ (A_c = cross section area).

$$\Rightarrow \frac{d}{dn} \left(-KA_c \frac{dT}{dn} \right) + h_p \frac{d\theta}{dn} (T - T_b) = 0.$$

$$\Rightarrow \frac{d^2 T}{dn^2} - \frac{h_p}{KA_c} (T - T_b) = 0.$$

Dimensionalizing T ,

$$(T - T_b) = 0 \Rightarrow \frac{d^2 \theta}{dn^2} = \frac{d^2 T}{dn^2}.$$

and let, $m = \sqrt{\frac{h_p}{KA_c}}$.

$$\Rightarrow \frac{d^2 \theta}{dn^2} - m^2 \theta = 0 \rightarrow \text{linear, homogeneous, } 2^{\text{nd}} \text{ order eqn with const. coeff'}$$

Sol¹ $\rightarrow \theta(n) = C_1 e^{mn} + C_2 e^{-mn}$

B.C \rightarrow let T_b = temp. of plate attached to fin.

i) at fin base $\therefore \theta(0) = T_b - T_{\infty}$.

ii) Fin tip.

a) Infinite long fin ($T_L = T_{\infty}$)

$$\therefore \theta(L) = T(L) - T_{\infty} = 0 \text{ as } L \rightarrow \infty.$$

But as $k \rightarrow \infty \rightarrow e^{mn} \rightarrow \infty$.

Hence, C_1 must be 0. Not a prospective soln.

only soln is,

$$\Theta(n) = C_2 e^{-mn}$$

\therefore at $n=b \rightarrow \Theta(0) = T_b - T_{\infty}$.

$$T_b - T_{\infty} = C_2 e^{-mb}$$

$$\Rightarrow C_2 = T_b - T_{\infty}$$

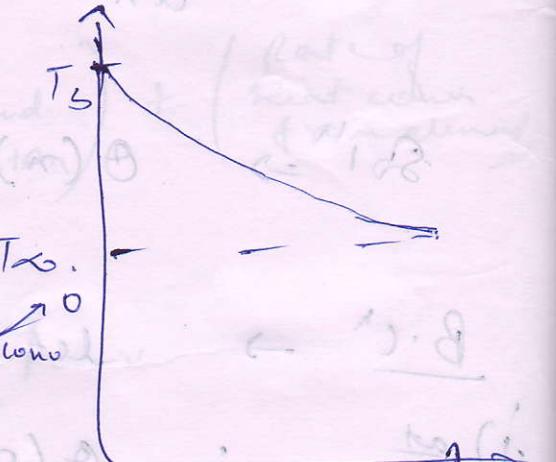
$$\therefore \Theta(n) = (T_b - T_{\infty}) e^{-mn}$$

$$\Rightarrow \frac{T_n - T_{\infty}}{T_b - T_{\infty}} = e^{-mn} = e^{-n\sqrt{hp/KA_c}}$$

Total heat loss from fin

Also, also, at base ($n=0$)

$$q_{\text{flow fin}} = -KA_c \frac{d(T)}{dn}|_{n=0} + q_{\text{flow}}$$



$$\left. \frac{dT}{dn} \right|_{n=0}$$

$$= -(T_b - T_{\infty}) \sqrt{\frac{hp}{KA_c}} \cdot e^{-m(0)} \approx 1.$$

as φ

$$= -\sqrt{\frac{hp}{KA_c}} (T_b - T_{\infty}) \cdot x - KA_c (0)$$

$$= \sqrt{hpKA_c} (T_b - T_{\infty}) x$$

b) Negligible heat loss from fin tip / ~~flame~~
 (Adiabatic fin tip, $Q_{\text{fin tip}} = 0$).

↳ fin tip area is negligible compared to total fin area.

↳ Fin tip at higher temp. but negligible heat loss.

$$\rightarrow \frac{\partial \Theta}{\partial n} \Big|_{n=L} = 0.$$

$$\Rightarrow \text{at } \Theta_b \text{ base} \rightarrow \text{same} \rightarrow \Theta_b = C_1 + C_2.$$

$$\text{at tip} \rightarrow \frac{\partial \Theta}{\partial n} \Big|_{n=L} = mC_1 e^{mL} - mC_2 e^{-mL} = 0.$$

$$\Rightarrow C_1 = \frac{\Theta_b}{1 + e^{2mL}}, \quad C_2 = \frac{\Theta_b}{1 + e^{-2mL}}$$

$$\Rightarrow \Theta(x) = \frac{\Theta_b}{1 + e^{2mL}} e^{mx} + \frac{\Theta_b}{1 + e^{-2mL}} e^{-mx}$$

Can be rearranged as, $(\cosh mx = \frac{e^x + e^{-x}}{2})$

$$\frac{T_b - T_\infty}{T_b - T_\infty} = \frac{\cosh m(L-x)}{\cosh mL}$$

Total heat transfer rate

$$Q_{\text{radi. tip}} = -KA_c \frac{dT}{dx} \Big|_{n=L} = hPKA_c (T_b - T_\infty) \tanh mL$$

c) Specified Temp. ($T_{fin, tip} = T_L$). (d)

↳ a generalised case of $T_{fin} = T_\infty$

$$\rightarrow \underline{B \cdot C'} \text{ at tip} \rightarrow \theta(L) = (T_L - T_\infty)$$

we get $\sinh mL = 0$

$$\frac{T(n) - T_\infty}{T_b - T_\infty} = \frac{[(T_L - T_\infty)/(T_b - T_\infty)] \sinh(mn)}{\sinh(mL)}$$

$$\text{and } q_{\text{spec. temp.}} = \frac{\sqrt{h \rho k A_c} (T_b - T_\infty) \cosh(mL) [(T_L - T_\infty)/(T_b - T_\infty)]}{\sinh(mL)}$$

conv. from fin tip.

↳ many real case scenarios

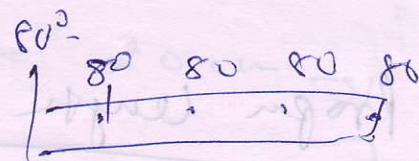
↳ short fins

$$q_F \rightarrow -k A_c \left[\frac{dT}{dr} \right]_{n=L} = h A_c [T_L - T_\infty]$$

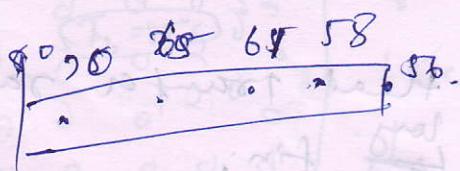
and

$$= \frac{+ \left(\frac{h}{mK} \right) \sinh m(L-n)}{mL + \left(\frac{h}{mK} \right) \sinh(mL)} \cdot$$

$$= \frac{(T_b - T_\infty) \sinh mL + \left(\frac{h}{mK} \right) \cosh mL}{\cosh mL + \left(\frac{h}{mK} \right) \sinh mL}$$

Fin efficiency

$$\eta_{fin} = \frac{q_{fin}}{q_{fin,max}}$$



= Actual heat transfer rate from the fin

Ideal heat transfer rate from the fin.
if entire fin were at base temp.

e.g. long fin $\rightarrow \eta_{long-fin} = \frac{q_{fin}}{q_{fin,max}} = \frac{\sqrt{h_p K A_c} (T_b - T_a)}{h A_{fin} (T_b - T_a)} = \frac{1}{L} \sqrt{\frac{K A_c}{h_p}} = \frac{1}{m^2}$

adiabatic fin $\rightarrow \eta_{adiabatic-fin} = \frac{q_{fin}}{q_{fin,max}} = \frac{\sqrt{h_p K A_c} (T_b - T_a) + h u m L}{h A_{fin} (T_b - T_a)} = \frac{\tanh m L}{m L}$

$$\epsilon_{fin} = \frac{q_{fin}}{q_{no-fin}} = \frac{q_{fin}}{h A_b (T_b - T_a)}$$

= Heat transfer rate from the fin
of base Area A_b

Heat transfer rate from the
surface of area A_b (w/o fin)

long fin w/ uniform cross-section area ($A_3 = A_1$)

$$\epsilon_{long-fin} = \frac{q_{fin}}{q_{no-fin}} = \frac{\sqrt{h_p K A_c} (T_b - T_a)}{h A_b (T_b - T_a)} = \sqrt{\frac{K / p}{h A}}$$

$\Rightarrow K/T, P/T$ (thin plate, slender pin), ϵ/T for low h applications

more justified to
use if h is low

Also - $\epsilon - \eta$ relation

$$\epsilon_{fin} = \frac{q_{fin}}{q_{no-fin}} = \frac{q_{fin}}{h A_b (T_b - T_a)} = \frac{\eta_{fin} \cdot h A_{fin} (T_b - T_a)}{h A_b (T_b - T_a)}$$

$$\boxed{\epsilon_{fin} = \frac{\eta_{fin}}{A_{fin} \cdot \eta_{fin}}}$$

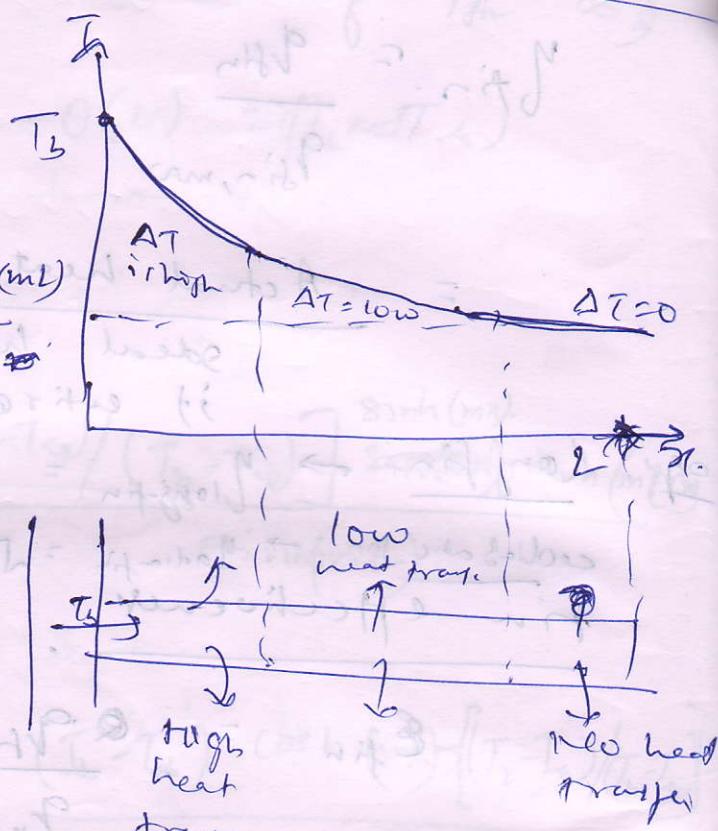
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Proper length of fin - Approx. for infinite heat flux

Heat transfer ratio for long fin;

$$\frac{q_{fin}}{q_{long\ fin}} = \frac{\sqrt{h_p F A_c} (T_b - T_s) \tanh(mL)}{\sqrt{h_p F A_c} (T_b - T_s) +}$$

$$= \tanh(mL)$$



mL	$\tanh(mL) \left(= \frac{q_{fin}}{q_{long\ fin}} \right)$
0.1	0.1
0.5	0.462
1	0.762
2	0.999
5	1

fin approaches No heat loss zone at $mL = 5$

or,
$$LT = \frac{5}{m} \rightarrow$$
 length considered to be infinitely long fin beyond which no heat transfer will take place.

rotator p-3 - OA

$$h_p F A_c (T_b - T_s) \sinh(mL) + h_p m L = -2 \rho \nu \frac{m^2}{23}$$

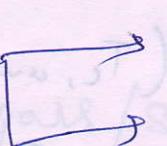
$$(T_b - T_s) \sinh(mL) + h_p m L = -2 \rho \nu \frac{m^2}{23}$$

Heat exch.

Types of heat exch.

① Parallel flow

② Counter flow

③ Cross flow  un mixed
 mixed.

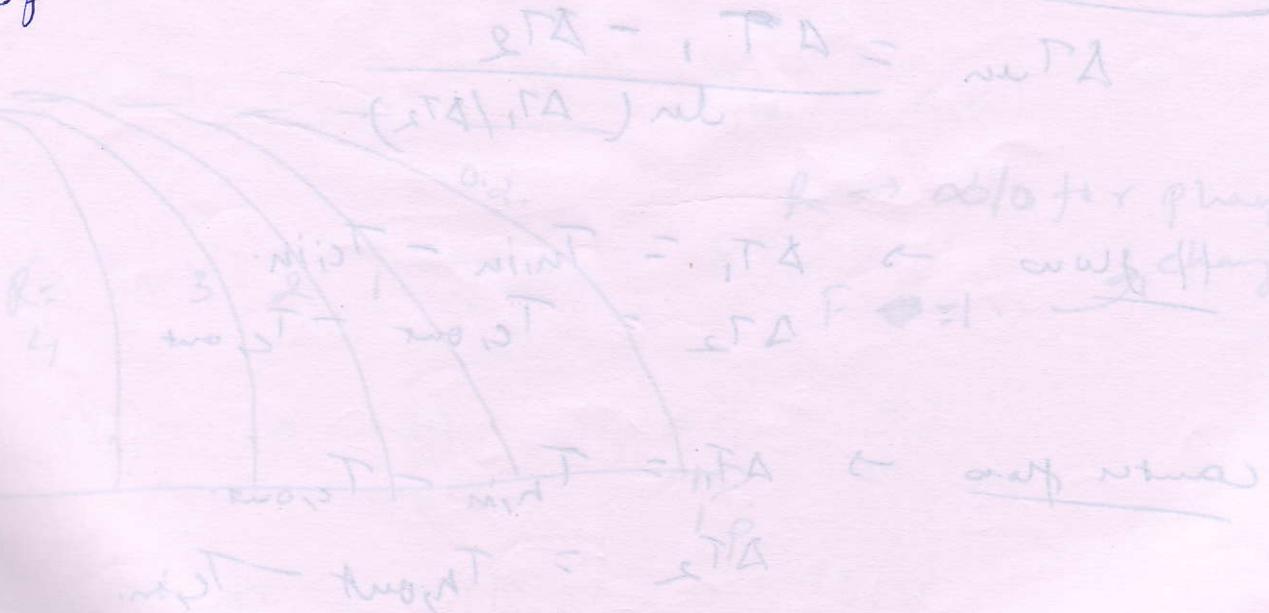
④ Shell - and - tube type.  One shell - 1 2 tube pass
 2 shell. 4 4 tube pass etc.

⑤ Plate & frame type.

⑥ Regenerative type (high specific heat - heat storage ceramic wool)

⑦ Condenser / Boiler \rightarrow Phase change Heat exch. type.

⑧ Space radiator



Analysis of heat exchangers.

LMTD -

$$\Delta T_{\text{LMTD}} \Rightarrow$$

Heat capacity rate.

$$q = m_c C_p (T_{c,\text{out}} - T_{c,\text{in}})$$

$$= C_c (T_{c,\text{out}} - T_{c,\text{in}})$$

$$= m_h C_p (T_{h,\text{out}} - T_{h,\text{in}})$$

$$= C_h (T_{h,\text{in}} - T_{h,\text{out}})$$

C_h & C_c are called heat capacity rates

~~Phase Change Heat Exchangers~~

$$\Delta T \rightarrow 0 \Rightarrow C \rightarrow \infty$$

→ limited by C_{\min} → heat transfer,
 ~~reservoir~~

LMTD -

$$\Delta T_{\text{LMTD}} = \frac{\Delta T_1 - \Delta T_2}{\ln(\Delta T_1 / \Delta T_2)}$$

$$\text{for } \text{fw} \rightarrow \Delta T_1 = T_{h,\text{in}} - T_{c,\text{in}}$$

$$\Delta T_2 = T_{c,\text{out}} - T_{h,\text{out}}$$

$$\text{for } \text{cw} \rightarrow \Delta T_1 = T_{h,\text{in}} - T_{c,\text{out}}$$

$$\Delta T_2 = T_{h,\text{out}} - T_{c,\text{in}}$$

when

In counter flow, when $C_c = C_m$

$$\Delta T_{in} = \frac{0}{0} \Rightarrow \Delta T_{in} = \Delta T_1 - \Delta T_2.$$

For plain change heat exchanger

ΔT_{in} is same for II or counter flow

Cross flow, multi pass / shell-tube type heat exchanger

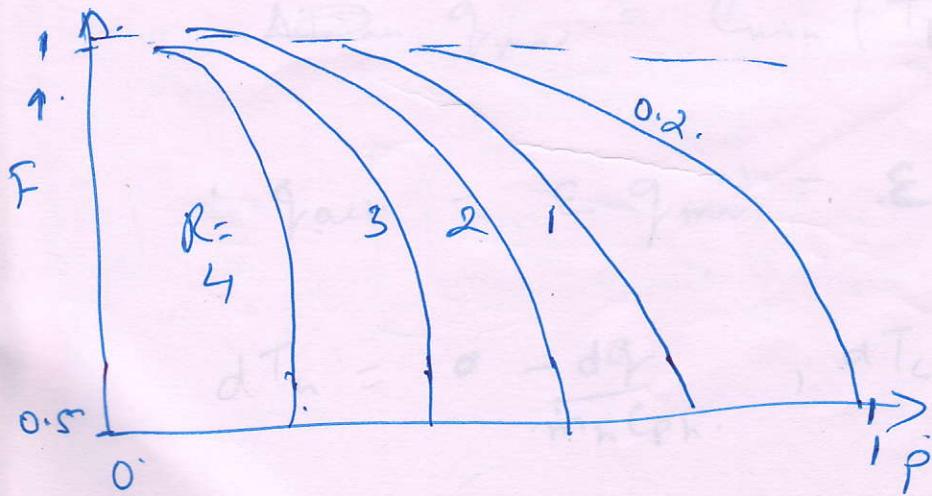
$$\Delta T_{in} = F \Delta T_{in, CF}$$

where, F = correction factor

= correlated with respect to R & R .

where, $R = \frac{t_2 - t_1}{T_1 - t_1}$, $R = \frac{T_1 - T_2}{t_2 - t_1} = \frac{(\dot{m}_p)_{\text{tube}}}{(\dot{m}_p)_{\text{shell}}}$

1 → inlet, T → Shell (outside)
 2 → outlet, t = tube (inside).



$f \rightarrow \infty$ for plain
 $F \neq 1$. \rightarrow change.

$$\Delta T_b - \Delta T_c = -d(T_b - T_c) = -d\left(\frac{1}{\dot{m}_c C_p} + \frac{1}{\dot{m}_b C_p}\right)$$

Performance Analysis

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Compactness of heat Exchangers

$$\text{Area density, } \beta = \frac{\text{Surface area of heat exchange}}{\text{Volume of heat exchanger}}$$

$$\beta > 700 \frac{\text{m}^2}{\text{m}^3} \rightarrow \text{Heat Exchangers compact HE}$$

$$\text{Car radiation} \approx 1000 \text{ m}^2/\text{m}^2$$

$$\text{Human lung, } \beta = 20,000 \text{ m}^2/\text{m}^3.$$

Effectiveness - NTU method

$$\text{Effectiveness, } \epsilon = \frac{q_{\text{actual}}}{q_{\text{max.}}}$$

$$q_{\text{actual}} = C_e (T_{c,\text{out}} - T_{c,\text{in}})$$

$$= C_m (T_{n,\text{in}} - T_{n,\text{out}})$$

~~$$q_{\text{max}} = C_m (T_{n,\text{in}} - T_{c,\text{in}})$$~~

$$q_{\text{actual}} = \epsilon q_{\text{max}} = \epsilon (C_m (T_{n,\text{in}} - T_{c,\text{in}}))$$

$$dT_n = - \frac{dq}{m_n C_{p,n}}, \quad dT_c = \frac{dq}{m_c C_{p,c}}$$

$$dT_n - dT_c = -d(T_n - T_c) = -dq \left(\frac{1}{m_n C_{p,c}} + \frac{1}{m_c C_{p,c}} \right)$$

$$\text{also, } dq = V(T_h - T_c) dA_s$$

$$\therefore \frac{d(T_h - T_c)}{(T_h - T_c)} = -V dA_s \left(\frac{1}{m_h C_p h} + \frac{1}{m_c C_p c} \right)$$

integrating.

$$\Rightarrow \ln \left(\frac{T_{h,out} - T_{c,out}}{T_{h,in} - T_{c,in}} \right) = V A_s \left(\frac{1}{m_h C_p h} + \frac{1}{m_c C_p c} \right) = \frac{V A_s}{C_c} \left(\frac{C_c}{C_h} + 1 \right)$$

Also,

$$C_h (T_{h,in} - T_{h,out}) = C_c (T_{c,out} - T_{c,in})$$

$$\Rightarrow T_{h,out} = T_{h,in} - \frac{C_c}{C_h} (T_{c,out} - T_{c,in}) \checkmark$$

$$\Rightarrow \ln \left(\frac{T_{h,in} - T_{c,in} + T_{c,in} - T_{c,out} - \frac{C_c}{C_h} (T_{c,out} - T_{c,in})}{T_{h,in} - T_{c,in}} \right)$$

$$(m_h T = \frac{V A_s}{C_c} \left(1 + \frac{C_c}{C_h} \right))$$

$$\Rightarrow \ln \left[1 = \left(1 + \frac{C_c}{C_h} \right) \left(\frac{T_{c,out} - T_{c,in}}{T_{h,in} - T_{c,in}} \right) \right] = \frac{V A_s}{C_c} \left(1 + \frac{C_c}{C_h} \right)$$

$$\frac{T_b}{T_b - nT_b} = \frac{T_b}{T_b} \quad \frac{T_b - nT_b}{T_b} = nT_b$$

$$\left(\frac{1}{n+1} \right) T_b - = (nT_b - T_b) b = T_b - nT_b$$

$$\text{Ans}, \quad \varepsilon = \frac{q}{q_{\max}} = \frac{C_c(T_{c,\text{out}} - T_{c,\text{in}})}{C_{\min}(T_{h,\text{in}} - T_{c,\text{in}})}$$

$$\Rightarrow \frac{T_{c,\text{out}} - T_{c,\text{in}}}{T_{h,\text{in}} - T_{c,\text{in}}} = \varepsilon \frac{C_{\min}}{C_c}$$

$$\Rightarrow \ln \left[1 - \left(1 + \frac{C_c}{C_h} \right) \left(\varepsilon \frac{C_{\min}}{C_c} \right) \right] = -\frac{U A_s}{C_c} \left(1 + \frac{C_c}{C_h} \right)$$

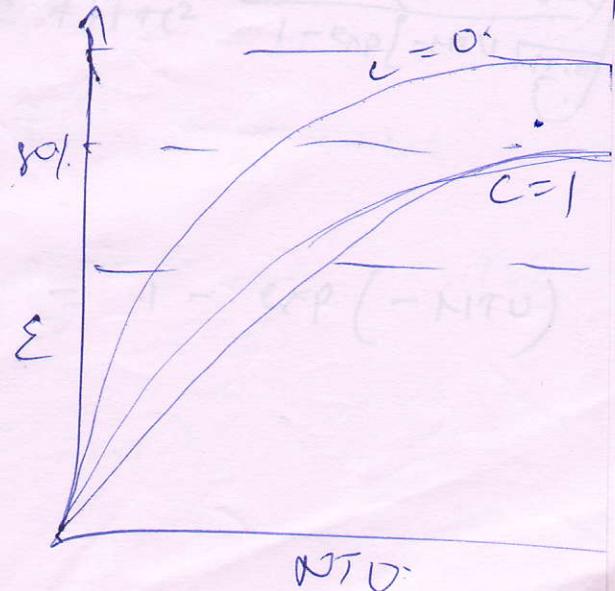
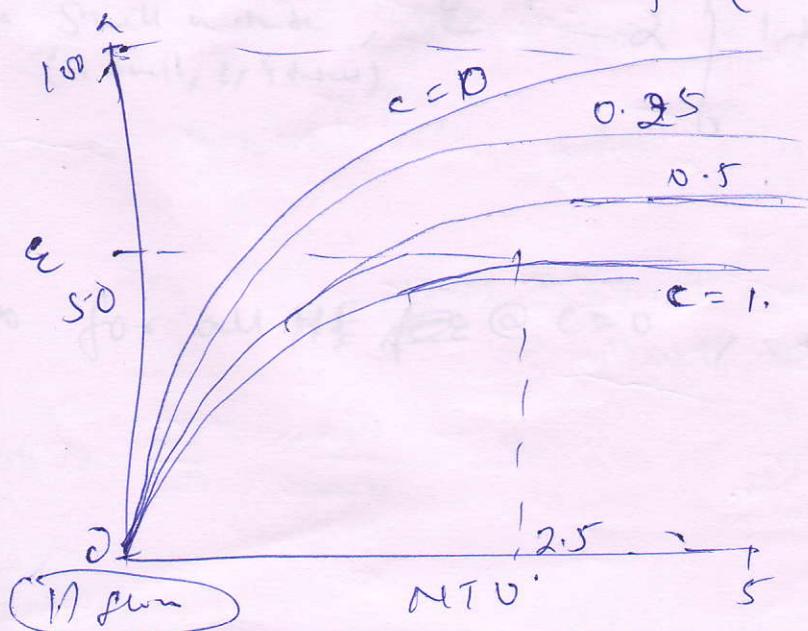
$$\Rightarrow \boxed{\begin{aligned} \varepsilon_{\text{parallel flow}} &= 1 - \exp \left[-\frac{U A_s}{C_c} \left(1 + \frac{C_c}{C_h} \right) \right] \\ &= \frac{\left(1 + \frac{C_c}{C_h} \right) \left(\frac{C_{\min}}{C_c} \right)}{1 + \frac{C_{\min}}{C_{\max}}} \\ &= 1 - \exp \left[-\frac{U A_s}{C_{\min}} \left(1 + \frac{C_{\min}}{C_{\max}} \right) \right] \end{aligned}}$$

$$NTU = \frac{U A_s}{(\min C_p)_{\min}} = \frac{U A_s}{C_{\min}}$$

$$c = C_{\min}/C_{\max}$$

$$\Rightarrow \boxed{\varepsilon_{\text{parallel}} = \frac{1 - \exp[-NTU(1+c)]}{1+c}}$$

$$\varepsilon = f(NTU, c)$$



5/11/10

NTU

$$\frac{V_{AS}}{C_{min}} = \frac{V_{AS}}{(mC_p)_{min}}$$

for a specified V & C_{min} NTU represents measure of the heat transfer susceptibility (α)

$$\therefore \epsilon = f(NTU, \frac{C_{min}}{C_{out}})$$

$$= f(NTU, c)$$

* In flow, $\epsilon = \frac{1 - \exp(-NTU(1+c))}{1+c}$

* counter flow, $\epsilon = \frac{1 - \exp[-NTU(1+c)]}{1 - c \exp[-NTU(1+c)]} \quad (\text{for } c < 1)$

$$= \frac{NTU}{1 + NTU} \quad (\text{For } c = 1).$$

* Shell & tube, $\epsilon = 2 \int_{1+c}^{1+c+\sqrt{1+c^2}} \frac{1 + \exp(-NTU\sqrt{1+c})}{1 - \exp(-NTU\sqrt{1+c})} \, dt$

* for all HE ~~$\epsilon @ c=0$~~ $\epsilon = 1 - \exp(-NTU)$

- ϵ change in ϵ is negligible beyond $NTU \approx 3$.
- for a given NTU, $\epsilon_{\text{outflow}}$ is high & ϵ_{inflow} is lowest.
- For a given heat exchanger type, ϵ is independent (or constant) of C for $NTU < 0.3$.
- once C & NTU have been evaluated, the ϵ value can be determined from Chilton's formula / correlation for given type of heat exchanger.

Tutorial 13

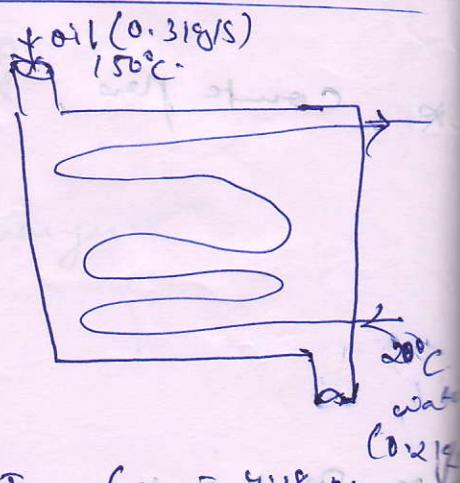
Q. ~~Not~~

1 shell, 8-tube passes heat exchanger. The tubes are thin walled, made of copper.

Tube - ID = 1.4 cm, length of each pass $= 5 \text{ m}$.

$$U = 310 \text{ W/m}^2 \cdot \text{K}, C_{D,1} = 2.13 \frac{\text{kg}}{\text{m}^2 \cdot \text{K}}, C_{P,1,0} = 4.18 \frac{\text{J}}{\text{kg} \cdot \text{K}}$$

Determine i) q , ii) $T_{w,\text{out}}$, iii) $T_{D,1,\text{out}}$.



$$\text{Soln: } C_n = m_n C_{P,n} = 0.639 \text{ kW/K}$$

$$C_c = 0.836 \text{ kW/K}$$

$$\therefore C_{\min} = C_h \Rightarrow C = \frac{C_{\max}}{C_{\min}} = \frac{C_n}{C_c} = 0.76$$

Selection of HE

- Head & pump rate
- cost
- Pumping power } effort creating turbulence
to heat for enhancing h'
- Size & weight
- Type.
- Materials - more on HTF than interface metals.

Refr AC & psychrometrics

6/11/18

Moisture & temp. control - AC.

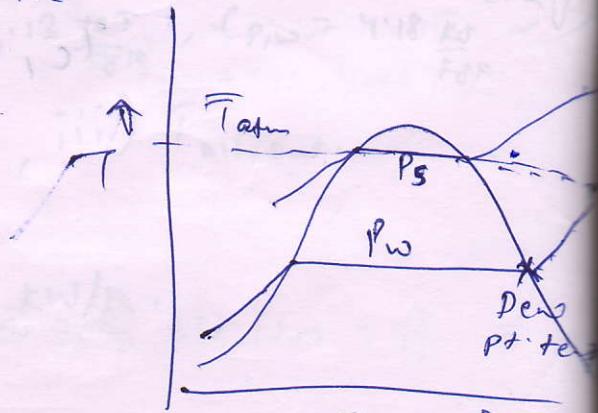
$$P = P_a + P_w$$

Mole fraction of water vapour,

$$\chi_w = \frac{P_w}{P} = P_w \quad (P_a = 1 \text{ atm}).$$

Relative humidity (RH, ϕ) - ratio of partial pr. of wat-vap. (P_w), to the sat. pr. (P_s) at the same temp. of mixture.

$$\therefore RH(\phi) = \frac{P_w}{P_s}$$



$$P_w V = m_w R_{H_2O} T = n_w \bar{R} T$$

$$\therefore P_s V = m_s R_{H_2O} T = n_s \bar{R} T.$$

$$\Rightarrow \phi = \frac{P_w}{P_s} = \frac{m_w}{m_s} = \frac{n_w}{n_s} = \frac{\chi_w}{\chi_s}$$

Specific humidity or Humidity Ratio (w)

mass of wet. vap. (or moisture) per unit mass of dry air in a mix. of air + wet. vap.

\therefore if, a = mass of dry air,
 m = mass of wet. vap.

$$w = \frac{m}{a}$$

$$w_{\max} = \frac{m_s}{a} \rightarrow \text{saturated cond.}$$

Assuming dry air and wet. vap. as ideal gases,

$$P_w V = m R_a T \Rightarrow m = \frac{P_w V}{R_a T}$$

$$P_a V = G R_a T \Rightarrow G = \frac{P_a V}{R_a T}$$

$$\therefore w = \frac{m}{G} = \frac{\left(\frac{P_w V}{R_a T} \right)}{\left(\frac{P_a V}{R_a T} \right)} = \frac{P_w}{P_a} \times \frac{R_a}{R_a}$$

$$= \left(\frac{P_w}{P - P_w} \right) \times \left(\frac{R_a}{R_a M_a} \right)$$

$$= \left(\frac{P_w}{P - P_w} \right) \times \left(\frac{M_a}{R_a} \right) = \left(\frac{P_w}{P - P_w} \right) \times \frac{18}{28.96}$$

$$\boxed{W = 0.622 \left(\frac{P_w}{P - P_w} \right)}$$

at' Saturated condition $\rightarrow W_s = 0.622 \times \left(\frac{P_s}{P - P_s} \right)$

Degree of Saturation (μ) \rightarrow ratio of actual specific humidity q to saturated specific humidity both at same temp. T .

$$\therefore \mu = \frac{w}{w_s} = \frac{0.622 \times \frac{P_w}{P - P_w}}{0.622 \times \frac{P_s}{P - P_s}} = \left(\frac{P_w}{P_s} \right) \left(\frac{P - P_w}{P - P_s} \right)$$

Now, if $\phi = 0 \rightarrow 0\% \text{ RH.} \rightarrow P_w = 0$

$$\Rightarrow \mu = 0.$$

if $\phi = 100\% \rightarrow 100\% \text{ RH.} \rightarrow P_w = P_s$.

$$\Rightarrow \mu = 1$$

μ varies between 0 & 1.

Dew point temp. (DBT_0) (T_{dp}) \rightarrow The temp. at which wat. vap. starts condensing (at const. pressure) is called DBT.

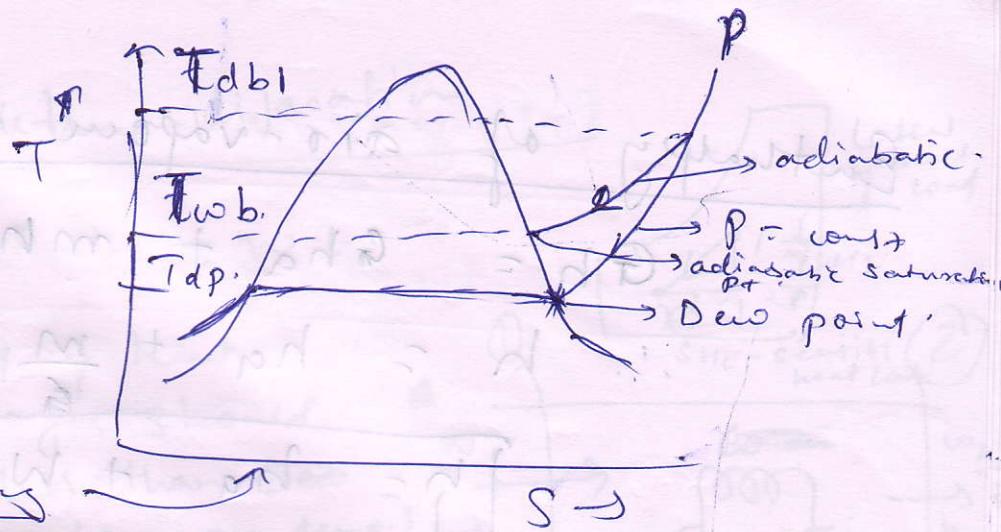
Dry bulb temp. (DBT) \rightarrow temp. recorded by thermometer at dry bulb.

Wet bulb temp. \rightarrow temp. recorded when water vap. is added to air with evaporation of water adiabatic condition.

Psychrometer

Adiabatic Saturation Temp.

\rightarrow The temp.



Adiabatic Cooling

Energy balance.

$$G h_{a1} + m_1 h_{g1} + \Delta m h_{f2} = G h_{a2} + m_2 h_{g2}$$

div. by G $\Rightarrow G h_{a1} + m_1 h_{g1} + (m_2 - m_1) h_{f2} = G h_{a2} + m_2 h_{g2}$

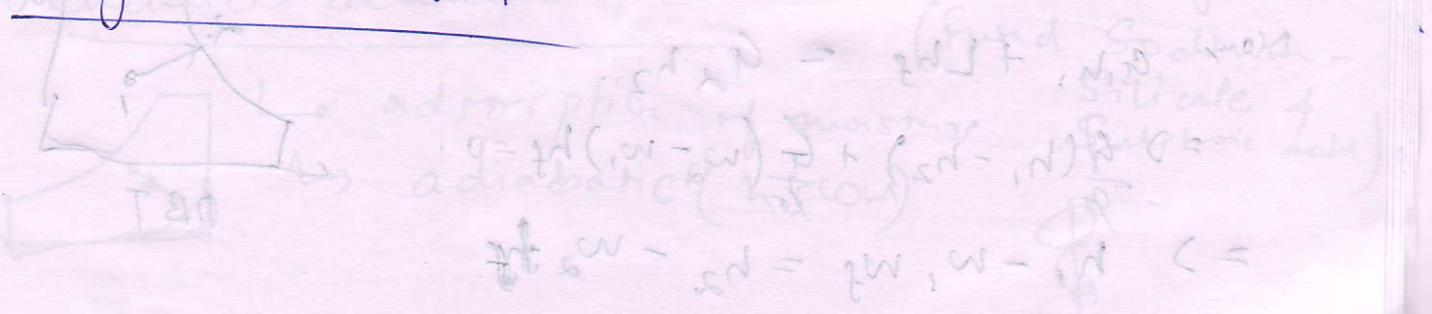
$$\Rightarrow h_{a1} + w_1 h_{g1} + (w_2 - w_1) h_{f2} = h_{a2} + w_2 h_{g2}$$

$$\Rightarrow w_1 = \frac{(h_{a2} - h_{a1}) + (w_2 h_{g2} - w_2 h_{f2})}{(1 - w_1)}$$

$$\boxed{w_1 = \frac{c_{p,a}(T_2 - T_1) + w_2 h_{fg2}}{w_1 h_{g1} + h_{f2}}}.$$

Where, $w_2 = \frac{m_2}{a} = \frac{m_s}{\bar{v}} = \frac{0.622 P_s}{P - P_s}$

Psychrometric Chart -



Enthalpy of air-vapour mixture.

$$G_h = G_ha + m h_w,$$

$$\therefore h = h_a + \frac{m h_w}{G}$$

$$[h = h_a + W h_w]$$

→ Enthalpy of mixture per kg of dry air.

Standard Psychrometric Process

① Sensible heating or cooling ($w = \text{const.}$)

$$G_1 h_1 + Q_{1-2} = G_2 h_2.$$

↳ heat addition / removal only

$$Q_{1-2} = G(h_2 - h_1).$$

↳ no change in moisture content

$$= G(h_{a2} + W h_{w2} - h_a - W h_{w1})$$

$$= G[C_p a(T_2 - T_1) - W C_p w(T_2 - T_1)] = G[(C_p a + W C_p w)(T_2 - T_1)].$$

② Adiabatic evaporative cooling. ($n = \text{const.}$)

↳ addition of moisture.

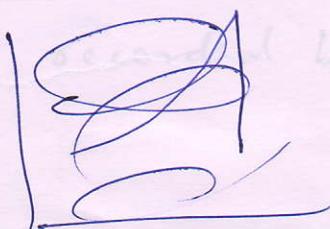
↳ enthalpy of air used for evaporation.

$$G_1 = G_2 = G$$

\downarrow amount of water added.

$$\Rightarrow G_1 w_1 + L = G_2 w_2.$$

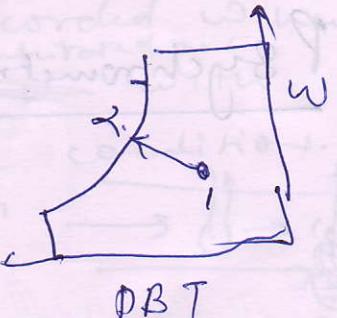
$$\Rightarrow L = G_2 w_2 - G_1 w_1 = G(w_2 - w_1)$$



$$\text{Now, } G_1 h_1 + L h_f = G_2 h_2.$$

$$\Rightarrow \frac{G}{G}(h_1 - h_2) + \frac{G}{G}(w_2 - w_1) h_f = 0.$$

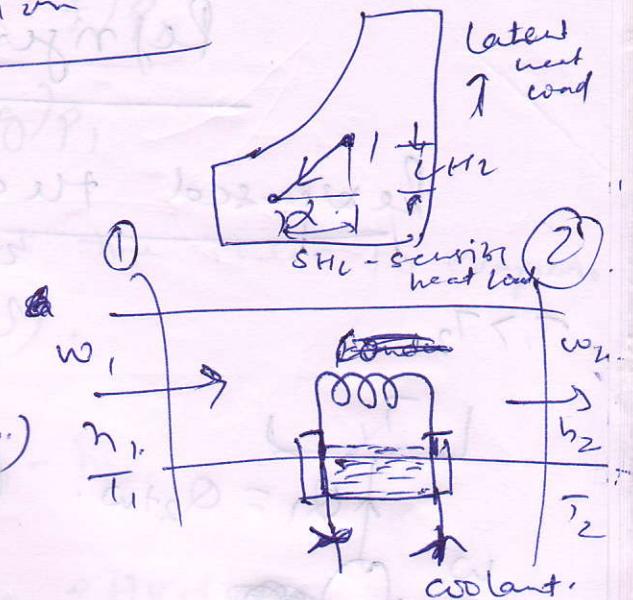
$$\Rightarrow h_1 - w_1 w_f = h_2 - w_2 f_f$$



③ Cooling & dehumidification

→ DBT ↓, w ↓

↳ decap. of heat
exchanger should
be less than DPT
(few pt. temp.)



$$m_1 = m_2 + L \rightarrow \text{moisture removed}$$

$$\therefore L = G(w_1 - w_2)$$

energy balance,

$$G_1 h_1 = G_2 h_2 + Q_{1-2} + L h_{f2}$$

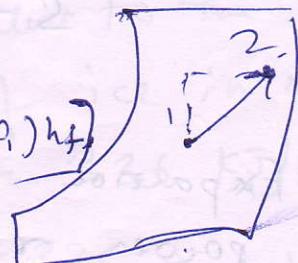
$$\therefore G_1 h_1 = G_2 h_2 + Q_{1-2} + G(w_1 - w_2) h_{f2} \quad \dots$$

$$\Rightarrow Q_{1-2} = G \left[(h_1 - h_2) - (w_1 - w_2) h_{f2} \right]$$

④ Heating & humidification.

~~G~~

$$Q_{1-2} = G \left[(h_2 - h_1) - (w_2 - w_1) h_f \right]$$



⑤ Chemical dehumidification → Sri saget

(Fixed Sodium -

↳ adsorption of moisture

Silicate &
Sulphonic acid)

↳ adiabatic (n = const)



8/11/18

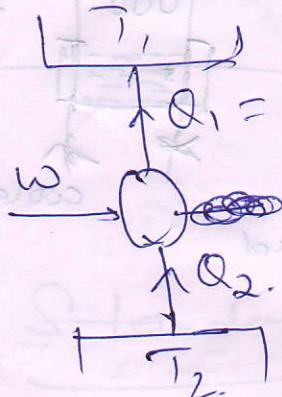
Refrigeration

Reversed heat engine cycle

→ Heat pump

↔ Refrigeration cycle

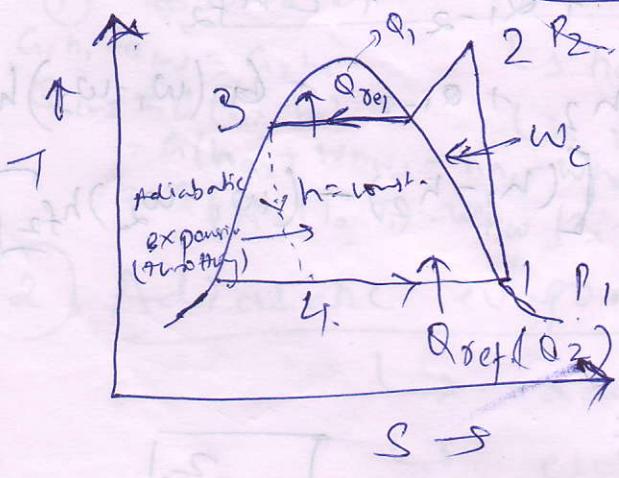
T_1, T_2



$$Q_1 = Q_2 + w \cdot \text{ (COP)}_{H.P.} = \frac{Q_1}{w} > 1.$$

$$(\text{COP})_{\text{Ref.}} = \frac{Q_2}{w}$$

Vapour compression ref. cycle.



Adiabatic expansion / turbine

$$TdS = dh - vdp \quad (\text{flow process})$$

$$\therefore \int dS = - \int_{P_1}^{P_2} \frac{vdp}{T} \quad (dh=0) \quad (M)$$

→ Expansion engine or turbine is not used as power recovery is small & doesn't justify the cost.

Energy analysis

$$\text{Comp.} \rightarrow h_2 = h_1 + w_c \Rightarrow w_c = (h_2 - h_1)$$

$$\text{condenser} \rightarrow Q_1 = (h_2 - h_3)$$

$$\text{Expansion} \rightarrow h_3 = h_4$$

$$(h_f)_{P_2} = (h_f)_{P_1} + x_f(h_{fg})$$

$$\therefore \eta_y = \frac{(h_f)_{P_2} - (h_f)_{P_1}}{(h_{fg})_{P_1}}$$

quality of ref. at the inlet to evaporator
(flash gas fraction).

Evaporator, $\text{Q}_2 = h_i - h_y$.

\uparrow
refrigerating effect.

$$COP = \frac{\text{Q}_2}{w_c} = \frac{h_i - h_y}{h_2 - h_1}$$

if, 'w' is the mass flow of ref. in kg/s.
the heat removal rate. $= w(h_i - h_y) \frac{\text{KJ}}{\text{s}}$
 $= w(h_i - h_y) \times 3600 \frac{\text{KJ}}{\text{h}}$.

One tonne of ref. \rightarrow defined as the ratio of heat removal from surrounding equivalent to the heat required for ~~melting~~^{fusing} 1 tonne of ice in one day.
(Latent heat of fusion of ice = 336 KJ/kg)

$$\therefore 1 \text{ tonne } \overset{\text{ice}}{\cancel{\text{ref}}} \equiv \frac{1000 \times 336}{24} = 14000 \frac{\text{KJ}}{\text{h}}$$

Capacity of ref. plant

$$= \frac{w(h_i - h_y) \times 3600}{14000} \text{ tonnes.}$$