

Aerodynamic Module-8 March 2017

Q1 - what r basic factors acc to isa
Ans. Press,density,temp

Q2. Stander temp ,preand density
A.273c,1.023N/M2,1.225kgper m3
B.273c,1.023×105 N/m2,1.225kg per m3

Q3 if fin is very large ie rudder too then what will be condition
A) laterally stable but dirvectioly unstable
B)ds but later unstable
C)strong ds
D strong ls

Q4 speed of sound
340m/s

Q5 condition for longitudinal stability
Ans position of tail plane frm cg

6) rolling is abt which axis
Lateral axis

:7) related to gliding ,when ac is having powerful engine
A plotting curve
B Performance characteristic
C Lift curve

8) coeffi of viscocity
A Dec with increase in altitude
B. as in a,but constant in tropopause
C. As in a ,increase in tropopose
D. As in b ,increase in stratosphere

9) aspect ratio
Span/mean cord

11 when ac is banking then what other factor will it depend on except radius
A. Speed
B.lift
C inertia

10) when a/ c is climbing its speed
A.inc with inc in altitude
B.dec with inc in alt.
C remain const

13) thin airfoil is used to

- A) inc lift
- B) incre speed
- C) inc drag

12) when some pressure is acted on any surface ,then what actally acting

- A press
- B force
- C inertia

15) aspect ratio is _____ when induced drag is halved.

Ans double both r inversly propor

16) when ac is banking at high speed then ac will roll

- A) inward
- B outward
- C doesnt affect

14) if camber of an airfoil is changed thn wht happns

Ans all of the above

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1) In which layer of atmosphere at below its height in sudden change atmosphere takes place

- a) Troposphere
- b) stratosphere
- c) Ionosphere

2) What is density of air?

- a) Weight of air per unit its volume
- b) Mass of air per unit its volume
- c) Weight of air with respect to its viscosity

3) Which laws of mechanics is/are applicable to air

- a) 1st law
- b) 1st & 3rd law
- c) 1st, 2nd & 3rd law

4) Aileron provides — Movement

- a) Pitching
- b) Rolling
- c) Yawing

5) — is the distance travelled by q/c in straight level flight with given weight of fuel.

- a) Endurance range
- b) Safe range

6) Induced drag is inversely proportional to

- a) square of fwd velocity.
- b) Lift produced by q/c .
- c) weight of an q/c .

- 7) The longitudinal stability is depend largely on
- Aspect ratio of Tailplane
 - Center of pressure of Tailplane
 - Center of gravity distance from Tailplane

- 8) The point on upper surface of wing where laminar flow of air turns into Turbulent air
- Transition point
 - Stagnation point
 - Aerodynamic center point

- 9) The center of pressure is the point
- At which low pressure on upper surface of wing gives resultant effect.
 - At which on chordline the resultant lift force acts.
 - At which high pressure below the wing gives resultant effect.

- 10) The longitudinal unstable a/c can restore its stability by
- Adjusting pitching moment about fuselage.
 - Increasing aspect ratio of tailplane
 - Adjusting c_p on tailplane.

- 11) The boundary layer in the flow of relatively slow flow rate
- Ahead of leading edge of wing
 - Close to upper surface of wing
 - Close to the lower surface of wing.

- 12) The induced drag which is produced by lifting component of C_L is
- a) Decreases by increasing weight of C_L
 - b) Increases by increasing square of the density of airflow
 - c) Decreases by increasing square of velocity of C_L