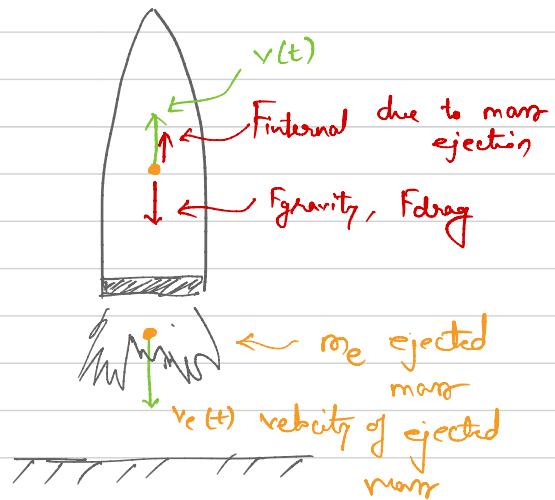


Rocket Equation

- Consider a rocket with initial mass $m(0)$ (kg)

- Rocket is ejecting mass m_e at a prescribed rate \dot{m}_e (kg/s)

at a relative velocity (relative to the rocket) v_e (m/s)



- Let $v = v(t)$, $m = m(t)$ are velocity (upward) (m/s) and mass of rocket (kg) at time $t=0$.

- Assume

- $\dot{m}_e = \dot{m}_e(t)$ is known \rightarrow Notation \dot{f} means $\frac{df}{dt}$

- $v(0) = 0$ is given

- $m(0) = m_0$ is known

- $g = 9.81$ (m/s²) gravity acceleration

- C_d drag coefficient (m/kg) is also known

- $v_e \leq 0$ relative velocity (m/s) of mass ejecting (negative as relative velocity is downward)

- Objective

Find velocity $v = v(t)$ of rocket at any time t , $0 \leq t \leq t_F$

Method

1. Conservation of mass as rocket is changing its mass
2. Conservation of linear momentum or velocity of rocket
(and momentum) is changing in time due to
mass ejection

(A.1) Conservation of mass

$$\begin{aligned} \text{Rate of change of mass of} & \quad - \text{Rate of mass out of rocket} \\ \text{rocket} & = \\ & + \text{Rate of mass into rocket} \end{aligned}$$

$$\Rightarrow \frac{dm}{dt} = - \frac{d m_e}{dt} + 0$$

↑
negative
or mass
goes out of
rocket

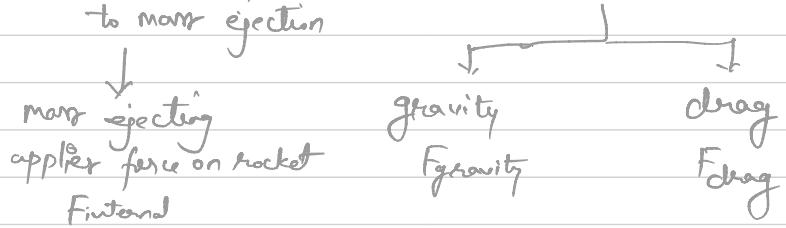
In our problem, we are not adding
any mass to rocket

Using notation $\dot{f} = \frac{df}{dt}$, we have

$$(1) \quad \dot{m}(t) = - \dot{m}_e(t)$$

(B.) Conservation of linear momentum

$$\text{Rate of change of linear momentum} = \text{Internal force due to mass ejection} + \text{External force}$$



- Linear momentum $\phi = \phi(t) = m(t)v(t)$

$$\Rightarrow \frac{d\phi}{dt} = \frac{d}{dt}(mv) = v \frac{dm}{dt} + m \frac{dv}{dt}$$

product rule

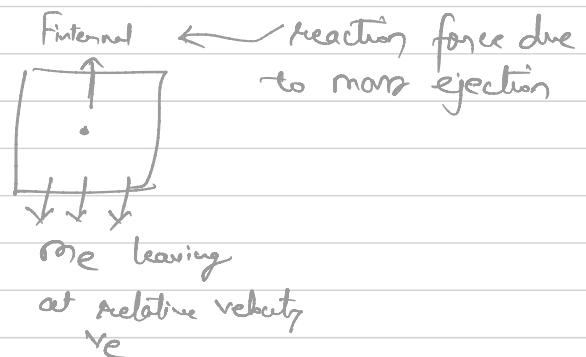
- $F_{\text{gravity}} = -mg$ (Our convention is upward +ve downward -ve)

- $F_{\text{drag}} = -C_d |v| v$ (drag always opposite to velocity)

- $F_{\text{internal}} = -m_e v_e$

Note: v_e is negative as it is downward relative to rocket

\Downarrow
 $F_{\text{internal}} \geq 0$ for $v_e \leq 0$



Thus from conservation of linear momentum law

$$m \frac{dv}{dt} + v \frac{dm}{dt} = -m_e v_e - mg - C_d |v| v$$

$$\therefore \frac{dm}{dt} = -\dot{m}_e \quad (\text{from conservation of mass})$$

we have

$$(2) \quad m \frac{dv}{dt} = \dot{m}_e (v - v_e) - mg - \frac{C_d}{m} \rho v^2$$

or if at time t , $m(t) > 0$, then

$$(3) \quad \frac{dv}{dt} = \frac{\dot{m}_e}{m} (v - v_e) - g - \frac{C_d}{m} \rho v^2$$

Let $m_0(\text{kg})$, $v_0(\text{m/s})$, $v_e(\text{m/s})$, $g(\text{m/s}^2)$, $C_d(\text{kg/m})$
and $\dot{m}_e(t)$ function is given.

Then we can compute $m(t)$, $v(t)$ at time t , $0 < t \leq t_F$, using

$$① \quad m(0) = m_0, \quad v(0) = v_0$$

$$② \quad \dot{m}(t) = -\dot{m}_e(t)$$

$$③ \quad \frac{dv}{dt} = \frac{\dot{m}_e}{m} (v - v_e) - g - \frac{C_d}{m} \rho v^2$$