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In[ ]:= (*range of values for inclination angle of northern wing*)
RE = 1353 * 103; (*moon radius in m*)
B0 = 5.1369 * 10-9; (*background field magnitude in T*)
B0x = 0;
B0y = 0;
B0z = -B0; (*background magnetic field in system (x,y,z)*)

n0 = 0.11 * 106; (*upstream number density in 1/m^3*)
u0 = 43 * 103; (*magnitude of upstream flow velocity in m/s*)
u0xcor = u0;
u0ycor = 0;
u0zcor = 0; (*upstream flow velocity in system (xcor,ycor,zcor)*)
E0norm = u0 * B0; (*may be needed for normalization purposes*)
mp = 1.6726231 * 10-27; (*proton mass*)
m = 7.5 * mp; (*upstream ion mass*)
mu0 = 4 * 3.14159265359 * 10-7; (*magnetic permeability of vacuum*)

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$$v_A = \frac{B_0}{\sqrt{\mu_0 * n_0 * m}}; (*Alfven velocity*)$$

$$MA = u_0 / v_A (*Alfvenic Mach number*)$$

$$\Theta_{N}[\theta] = \text{ArcTan}\left[\frac{MA \cos[\theta]}{1 - MA \sin[\theta]}\right] * \frac{360}{2 \pi};$$

$$\Theta_{S}[\theta] = \text{ArcTan}\left[\frac{MA \cos[\theta]}{1 + MA \sin[\theta]}\right] * \frac{360}{2 \pi};$$

(\*inclination angle against B0 for NORTHERN Alfven wing\*)

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Out[ ]:= 0.348577
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In[ ]:= (*tilt of flow vector in northern and
southern wing against x axis for fully saturated case*)

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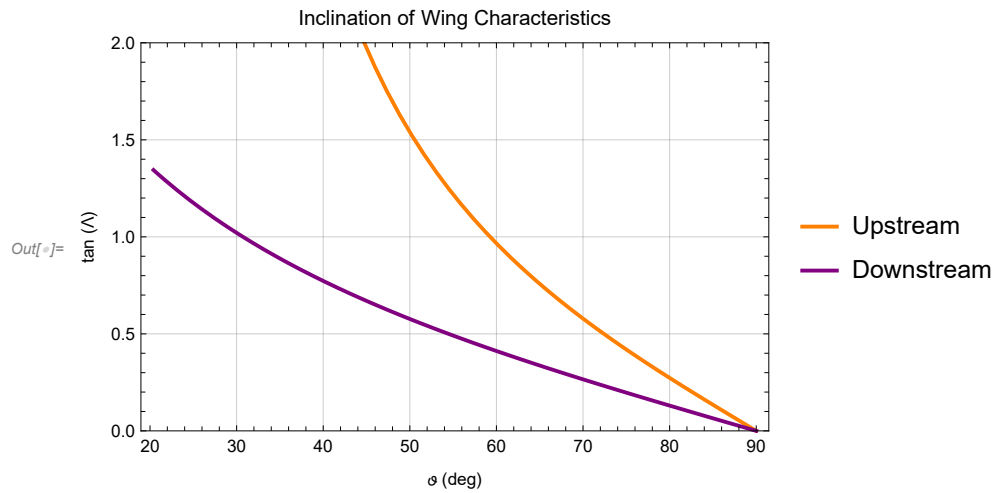
$$\text{flowtiltN}[\theta] = \text{Tan}\left[\left(90 + \Theta_{N}\left[\theta * \frac{2 \pi}{360}\right] - \theta\right) * \frac{2 \pi}{360}\right];$$

$$\text{flowtiltS}[\theta] = \text{Tan}\left[\left(90 - \Theta_{S}\left[\theta * \frac{2 \pi}{360}\right] - \theta\right) * \frac{2 \pi}{360}\right];$$

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In[ ]:= Plot4 = Plot[{flowtiltN[theta], flowtiltS[theta]}, {theta, ArcSin[MA] *  $\frac{360}{2\pi}$ , 90},
  FrameLabel -> {" $\phi$  (deg)", "tan ( $\Delta$ )"}, GridLines -> Automatic,
  Frame -> True, PlotLabel -> "Inclination of Wing Characteristics",
  PlotStyle -> {{Orange, Thick}, {Purple, Thick}},
  PlotLegends -> {"Upstream", "Downstream"}, PlotRange -> {0, 2}]

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In[ ]:= Export["figure4.pdf", Plot4]

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Out[ ]:= figure4.pdf

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