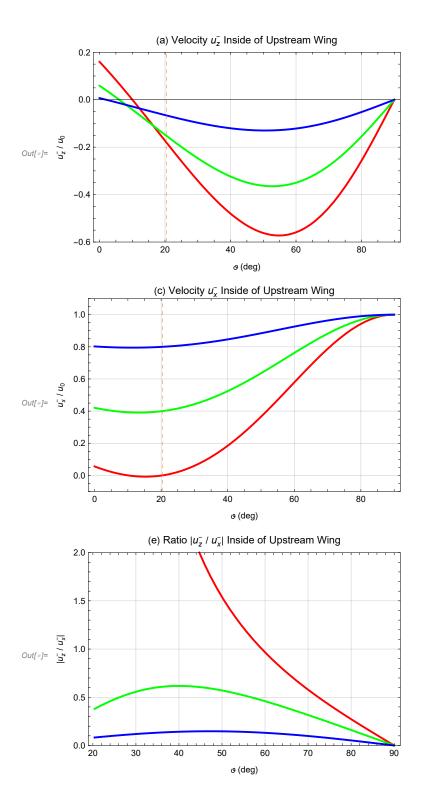
```
ln[-]:= (*05/16/21*)
      (*velocity u z (constant) inside the norhern Alfven wing generated by a box-
      like ionospheric obstacle*)
     B0 = 5.1369 * 10<sup>-9</sup>; (*upstream magnetic field in Tesla*)
     n0 = 0.11 * 10^6; (*upstream number density in 1/m^3*)
     mp = 1.6726231 * 10^{-27}; (*proton mass*)
     m = 7.5 * mp; (*upstream ion mass*)
     mu0 = 4 * 3.14159265359 * 10^{-7}; (*magnetic permeability of vacuum*)
     vA = \frac{B0}{\sqrt{mu0 * n0 * m}}; (*Alfven velocity*)
     u\theta = 43 * 10^3; (*magnitude of upstream flow velocity in m/s*)
     MA = u0 / vA(*Alfvenic Mach number, Europa*)
     uz[theta_, a_] = \frac{B0 * Cos[theta]}{\sqrt{mu0 * n0 * m} (1 + MA^2 - 2 MA Sin[theta])} *
         (-\sqrt{(1 + MA^2 - 2 MA Sin[theta] - MA^2 a^2 (1 - Sin[theta] * Sin[theta]))} +
            MA * (a - 2) * Sin[theta] + MA^2 (1 - a) + 1);
     ux[theta_, a_] = u0 + \frac{B0}{\sqrt{mu0 * n0 * m}} (1 + MA^2 - 2 MA Sin[theta]) * ((Sin[theta] - MA) *
              \sqrt{(1 + MA^2 - 2 MA Sin[theta] - MA^2 a^2 (1 - Sin[theta] * Sin[theta]))} +
             MA * a * Cos[theta] * Cos[theta] - Sin[theta] * (1 + MA^2 - 2 MA Sin[theta]));
Out[*]= 0.348577
```

```
In[*]:= Plot1 = Plot
           \left\{ uz \left[ theta * \frac{2\pi}{360}, 0.0 \right] / (u0), uz \left[ theta * \frac{2\pi}{360}, 0.4 \right] / (u0), uz \left[ theta * \frac{2\pi}{360}, 0.8 \right] / (u0) \right\}
           {theta, 0, 90}, FrameLabel \rightarrow {"0 (deg)", "u_z^- / u_0^-}, GridLines \rightarrow Automatic,
           Frame → True, PlotLabel → "(a) Velocity u, Inside of Upstream Wing",
           PlotStyle → {{Red, Thick}, {Green, Thick}, {Blue, Thick}},
           (*PlotLegends \rightarrow {"\lambda^- = 1.0","\lambda^- = 0.6","\lambda^- = 0.2"},*)
           PlotRange \rightarrow {-0.6, 0.2}, Epilog \rightarrow {Directive[Dashed, Orange],
              InfiniteLine \left[ \left\{ \left\{ ArcSin[MA] * \frac{360}{2}, -0.6 \right\}, \left\{ ArcSin[MA] * \frac{360}{2}, 0.2 \right\} \right] \right] \right]
       Plot2 = Plot \left[ \left\{ ux \left[ theta * \frac{2\pi}{360}, 0.0 \right] \right/ (u0), ux \left[ theta * \frac{2\pi}{360}, 0.4 \right] \right/ (u0), 
             ux[theta * \frac{2\pi}{360}, 0.8] / (u0)}, {theta, 0, 90},
           FrameLabel \rightarrow {"0 (deg)", "u_x^- / u_0"}, GridLines \rightarrow Automatic, Frame \rightarrow True,
           PlotLabel \rightarrow "(c) Velocity u_{\nu}^{-} Inside of Upstream Wing",
           PlotStyle → {{Red, Thick}, {Green, Thick}, {Blue, Thick}},
           (*PlotLegends →{"\lambda^- = 1.0","\lambda^- = 0.6","\lambda^- = 0.2"}, *)
           PlotRange \rightarrow {-0.1, 1.1}, Epilog \rightarrow {Directive[Dashed, Orange],
              InfiniteLine \left[\left\{\left\{ArcSin\left[MA\right] * \frac{360}{2\pi}, -0.1\right\}, \left\{ArcSin\left[MA\right] * \frac{360}{2\pi}, 1.1\right\}\right\}\right]\right\}
       Plot3 = Plot \left[ \left\{ Abs \left[ uz \left[ theta * \frac{2\pi}{360}, 0.0 \right] \middle] \right\} \right]  ux \left[ theta * \frac{2\pi}{360}, 0.0 \right] \right]
            Abs \left[ uz \left[ theta * \frac{2\pi}{360}, 0.4 \right] / ux \left[ theta * \frac{2\pi}{360}, 0.4 \right] \right]
            Abs \left[ uz \left[ theta * \frac{2\pi}{360}, 0.8 \right] / ux \left[ theta * \frac{2\pi}{360}, 0.8 \right] \right] \right\}, \left\{ theta, ArcSin [MA] * \frac{360}{2\pi}, 90 \right\},
           FrameLabel \rightarrow {"0 (deg)", "|u_{\bar{x}}| / u_{\bar{x}} / |u_{\bar{x}}|"}, GridLines \rightarrow Automatic, Frame \rightarrow True,
           PlotLabel \rightarrow "(e) Ratio |u_{\tau}^{-}/u_{\nu}^{-}| Inside of Upstream Wing",
           PlotStyle → {{Red, Thick}, {Green, Thick}, {Blue, Thick}}, (*,
           PlotLegends \rightarrow {"\lambda^+ = 1.0","\lambda^+ = 0.6","\lambda^+ = 0.2"},*)PlotRange \rightarrow {0, 2}
```



```
In[@]:= Export["figure3a_new.pdf", Plot1]
       Export["figure3c_new.pdf", Plot2]
       Export["figure3e new.pdf", Plot3]
Out[*]= figure3a_new.pdf
Out[*]= figure3c_new.pdf
Out[*]= figure3e_new.pdf
ln[*] = uxroots[theta_] = u0 + \frac{1}{\sqrt{mu0 * n0 * m}} (1 + MA^2 - 2 MA Sin[theta])
                                                                                     ------ * ((Sin[theta] - MA) *
                  \sqrt{(1 + MA^2 - 2 MA Sin[theta] - MA^2 a^2 (1 - Sin[theta] * Sin[theta]))} +
                MA * a * Cos[theta] * Cos[theta] - Sin[theta] * ((1 + MA^2 - 2 MA Sin[theta])));
ln[-]:= a = 0.1
Out[*]= 0.1
In[@]:= Solve[uxroots[theta] == 0, theta]
       Solve[uz[theta, 0.1] == 0, theta]
       ... Solve: Inverse functions are being used by Solve, so some solutions may not be found; use Reduce for complete solution
Out[*]= \{ \{ \text{theta} \rightarrow 0.270824 - 0.317337 i \} \}, \{ \text{theta} \rightarrow 0.270824 + 0.317337 i \} \}
         \{\mbox{theta}\rightarrow\mbox{2.87077}-\mbox{0.317337}\ \mbox{$\dot{\mathbb{1}}$}\} , \{\mbox{theta}\rightarrow\mbox{2.87077}+\mbox{0.317337}\ \mbox{$\dot{\mathbb{1}}$}\}\}
       ... Solve: Inverse functions are being used by Solve, so some solutions may not be found; use Reduce for complete solution
             information.
Out[\circ] = \{ \{ \text{theta} \rightarrow -1.5708 \}, \{ \text{theta} \rightarrow 0.15751 \}, \{ \text{theta} \rightarrow 1.5708 \} \}
```