

Reverse Engineering the Apollo 11 Guidance  
Computer (AGC)  
Source Code for Lunar Module (Luminary099)

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# Chapter 1

## Luminary099

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# DPS AND APS ENGINE PARAMETERS

SETLOC P40S  
BANK  
COUNT\* \$\$/P40

# \*\*\* THE ORDER OF THE FOLLOWING SIX CONSTANTS MUST NOT BE CHANGED \*\*\*

FDPS	2DEC	4.3670 B-7	# 9817.5 LBS FORCE IN NEWTONS
MDOTDPS	2DEC	0.1480 B-3	# 32.62 LBS/SEC IN KGS/CS
DTDECAY	2DEC	-38	
FAPS	2DEC	1.5569 B-7	# 3500 LBS FORCE IN NEWTONS
MDOTAPS	2DEC	0.05135 B-3	# 11.32 LBS/SEC IN KGS/CS
ATDECAY	2DEC	-10	

# \*\*\*\*\*

FRCS4	2DEC	0.17792 B-7	# 400 LBS FORCE IN NEWTONS
FRCS2	2DEC	0.08896 B-7	# 200 LBS FORCE IN NEWTONS

SETLOC P40S1  
BANK  
COUNT\* \$\$/P40

# \*\*\* APS IMPULSE DATA FOR P42 \*\*\*\*\*

K1VAL	2DEC	124.55 B-23	# 2800 LB-SEC
K2VAL	2DEC	31.138 B-24	# 700 LB-SEC
K3VAL	2DEC	1.5569 B-10	# FAPS (3500 LBS THRUST)

# \*\*\*\*\*

S40.136	2DEC	.4671 B-9	# .4671 M NEWTONS (DPS)
S40.136_	2DEC	.4671 B+1	# S40.136 SHIFTED LEFT 10.

SETLOC ASENT1  
BANK  
COUNT\* \$\$/P70

(1/DV)A	2DEC	15.20 B-7	# 2 SECONDS WORTH OF INITIAL ASCENT
---------	------	-----------	-------------------------------------

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```

# STAGE ACCELERATION -- INVERTED (M/0
# 1) PREDICATED ON A LIFTOFF MASS OF
# 4869.9 KG (SNA-8-D-027 7/11/68)
# 2) PREDICATED ON A CONTRIBUTION TO
# ICLE ACCELERATION FROM RCS THRU
# EQUIV. TO 1 JET ON CONTINUOUSLY

K(1/DV)      2DEC      436.70 B-9      # DPS ENGINE THRUST IN NEWTONS / 100

(AT)A        2DEC      3.2883 E-4 B9    # INITIAL ASC. STG. ACCELERATION ** M
# ASSUMPTIONS SAME AS FOR (1/DV)A.

(TBUP)A      2DEC      91902 B-17      # ESTIMATED BURN-UP TIME OF THE ASCEN
# ASSUMPTIONS SAME AS FOR (1/DV)A WI
# ADDITIONAL ASSUMPTION THAT NET MASS
# RATE = 5.299 KG/SEC = 5.135 (APS) -
# .164 (1 RCS JET).

                SETLOC  ASENT
                BANK
                COUNT*  $$/ASENT
AT/RCS      2DEC      .0000785 B+10    # 4 JETS IN A DRY LEM

                SETLOC  SERVICES
                BANK
                COUNT*  $$/SERV

# *** THE ORDER OF THE FOLLOWING TWO CONSTANTS MUST NOT BE CHANGED *****

APSVEX      DEC      -3030 E-2 B-5      # 9942 FT/SEC IN M/CS.
DPSVEX      DEC*     -2.95588868 E+1 B-05*  # VE (DPS) +2.95588868E+ 3

# *****

                SETLOC  F2DPS*31
                BANK
                COUNT*  $$/F2DPS

TRIMACCL    2DEC*     +3.50132708 E-5 B+08*  # A (T) +3.50132708E- 1

```

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## # THROTTLING AND THRUST DETECTION PARAMETERS

SETLOC P40S  
 BANK  
 COUNT\* \$\$/P40

THRESH1 DEC 24  
 THRESH3 DEC 12  
 HIRTHROT = BIT13

SETLOC FFTAG5  
 BANK  
 COUNT\* \$\$/P40

THRESH2 DEC 308

SETLOC FTHROT  
 BANK  
 COUNT\* \$\$/THROT

FMAXODD	DEC	+3841	# FSAT	+4.81454413 E+4
FMAXPOS	DEC	+3467	# FMAX	+4.34546769 E+4
THROTLAG	DEC	+20	# TAU (TH)	+1.99999999 E-1
SCALEFAC	2DEC*	+7.97959872 E+2 B-16*	# BITPERF	+7.97959872 E-2

SETLOC F2DPS\*32  
 BANK  
 COUNT\* \$\$/F2DPS

DPSTHRSH DEC 36 # (THRESH1 + THRESH3 FOR P63)

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## # LM HARDWARE-RELATED PARAMETERS

	SETLOC	RADARUPT	
	BANK		
	COUNT*	\$\$/RRUPT	
LVELBIAS	DEC	-12288	# LANDING RADAR BIAS FOR 153.6 KC.
RDOTBIAS	2DEC	17000	# BIAS COUNT FOR RR RANGE RATE.
	SETLOC	LRS22	
	BANK		
	COUNT*	\$\$/LRS22	
RDOTCONV	2DEC	-.0019135344 B7	# CONVERTS RR RDOT READING TO M/CS AT
RANGCONV	2DEC	2.859024 B-3	# CONVERTS RR RANGE READING TO M. AT
	SETLOC	SERVICES	
	BANK		
	COUNT*	\$\$/SERV	
HBEAMANT	2DEC	-.4687018041	# RANGE BEAM IN LR ANTENNA COORDINATE
	2DEC	0	
	2DEC	-.1741224271	
HSCAL	2DEC	-.3288792	# SCALES 1.079 FT/BIT TO 2(22)M.
# ***** THE SEQUENCE OF THE FOLLOWING CONSTANTS MUST BE PRESERVED *****			
VZSCAL	2DEC	+.5410829105	# SCALES .8668 FT/SEC/BIT TO 2(18) M/CS.
VYSCAL	2DEC	+.7565672446	# SCALES 1.212 FT/SEC/BIT TO 2(18) M/CS.
VXSCAL	2DEC	-.4020043770	# SCALES -.644 FT/SEC/BIT TO 2(18) M/CS.
# *****			
KPIP	DEC	.0512	# SCALES DELV TO UNITS OF 2(5) M/CS.
KPIP1	2DEC	.0128	# SCALES DELV TO UNITS OF 2(7) M/CS.
KPIP2	2DEC	.0064	# SCALES DELV TO UNITS OF 2(8) M/CS.

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ALTCONV	2DEC	1.399078846 B-4	# CONVERTS M*2(-24) TO BIT UNITS *2(-28).
ARCONV1	2DEC	656.167979 B-10	# CONV. ALTRATE COMP. TO BIT UNITS<
SETLOC R10			
BANK			
COUNT* \$\$/R10			
ARCONV	OCT	24402	# 656.1679798B-10 CONV ALTRATE TO BIT UNIT
ARTOA	DEC	.1066098 B-1	# .25/2.345 B-1 4X/SEC CYCLE RATE.
ARTOA2	DEC	.0021322 B8	# (.5)/(2.345)(100)
VELCONV	OCT	22316	# 588.914 B-10 CONV VEL. TO BIT UNITS.
KPIP1(5)	DEC	.0512	# SCALES DELV TO M/CS*2(-5).
MAXVBITS	OCT	00547	# MAX. DISPLAYED VELOCITY 199.9989 FT/SEC.
SETLOC DAPS3			
BANK			
COUNT* \$\$/DAPAO			
TORKJET1	DEC	.03757	# 550 / .2 SCALED AT (+16) 64 / 180

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# PARAMETERS RELATING TO MASS, INERTIA, AND VEHICLE DIMENSIONS

	SETLOC	FRANDRES	
	BANK		
	COUNT*	\$\$/START	
FULLAPS	DEC	5050 B-16	# NOMINAL FULL ASCENT MASS -- 2(16) K
	SETLOC	LOADDAP1	
	BANK		
	COUNT*	\$\$/R03	
MINLMD	DEC	-2850 B-16	# MIN. DESCENT STAGE MASS -- 2(16) K
MINMINLM	DEC	-2200 B-16	# MIN ASCENT STAGE MASS -- 2(16) KG.
MINCSM	=	BIT11	# MIN CSM MASS (OK FOR 1/ACCS) = 9050
	SETLOC	DAPS3	
	BANK		
	COUNT*	\$\$/DAPAD	
LOASCENT	DEC	2200 B-16	# MIN ASCENT LEM MASS -- 2(16) KG.
HIDESCNT	DEC	15300 B-16	# MAX DESCENT LEM MASS -- 2(16) KG.
LODESCNT	DEC	1750 B-16	# MIN DESCENT STAGE (ALONE) -- 2(16)

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# PHYSICAL CONSTANTS ( TIME - INVARIANT )

SETLOC IMU2  
BANK  
COUNT\* \$\$/P07

OMEG/MS 2DEC .24339048

SETLOC R30LOC  
BANK  
COUNT\* \$\$/R30

# \*\*\* THE ORDER OF THE FOLLOWING TWO CONSTANTS MUST BE PRESERVED \*\*\*\*\*

1/RTMUM 2DEC\* .45162595 E-4 B14\*  
1/RTMUE 2DEC\* .50087529 E-5 B17\*

# \*\*\*\*\*

SETLOC P40S1  
BANK  
COUNT\* \$\$/S40.9

EARTHMU 2DEC\* -3.986032 E10 B-36\* # M(3)/CS(2)

SETLOC ASENT1  
BANK  
COUNT\* \$\$/P12

MUM(-37) 2DEC\* 4.9027780 E8 B-37\*  
MOONRATE 2DEC\* .26616994890062991 E-7 B+19\* # RAD/CS.

SETLOC SERVICES  
BANK  
COUNT\* \$\$/SERV

# \*\*\* THE ORDER OF THE FOLLOWING TWO CONSTANTS MUST BE PRESERVED \*\*\*\*\*

-MUDT 2DEC\* -7.9720645 E+12 B-44\*  
-MUDT1 2DEC\* -9.8055560 E+10 B-44\*

# \*\*\*\*\*

-MUDTMUN 2DEC\* -9.8055560 E+10 B-38\*  
RESQ 2DEC\* 40.6809913 E12 B-58\*

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20J 2DEC 3.24692010 E-2  
 2J 2DEC 3.24692010 E-3

SETLOC P50S1  
 BANK  
 COUNT\* \$\$/LOSAM

RSUBEM 2DEC 384402000 B-29  
 RSUBM 2DEC 1738090 B-29  
 RSUBE 2DEC 6378166 B-29  
 ROE 2DEC .00257125

SETLOC CONICS1  
 BANK  
 COUNT\* \$\$/LT-LG

ERAD 2DEC 6373338 B-29 # PAD RADIUS  
 504RM 2DEC 1738090 B-29 # METERS B-29 (EQUATORIAL MOON RADIUS)

SETLOC CONICS1  
 BANK  
 COUNT\* \$\$/CONIC

# \*\*\* THE ORDER OF THE FOLLOWING CONSTANTS MUST BE PRESERVED \*\*\*\*\*

MUTABLE 2DEC\* 3.986032 E10 B-36\* # MUE  
 2DEC\* .25087606 E-10 B+34\* # 1/MUE  
 2DEC\* 1.99650495 E5 B-18\* # SQRT(MUE)  
 2DEC\* .50087529 E-5 B+17\* # 1/SQRT(MUE)  
 2DEC\* 4.902778 E8 B-30\* # MUM  
 2DEC\* .203966 E-8 B+28\* # 1/MUM  
 2DEC\* 2.21422176 E4 B-15\* # SQRT(MUM)  
 2DEC\* .45162595 E-4 B+14\* # 1/SQRT(MUM)

# \*\*\*\*\*



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SETLOC INTINIT  
BANK  
COUNT\* \$\$/INTIN

OMEGMOON 2DEC\* 2.66169947 E-8 B+23\*

SETLOC ORBITAL2  
BANK  
COUNT\* \$\$/ORBIT

# \*\*\* THE ORDER OF THE FOLLOWING CONSTANTS MUST NOT BE CHANGED \*\*\*\*\*

2DEC\* 1.32715445 E16 B-54\*  
MUM 2DEC\* 4.9027780 E8 B-30\*  
MUEARTH 2DEC\* 3.986032 E10 B-36\*  
2DEC 0  
J4REQ/J3 2DEC\* .4991607391 E7 B-26\*  
2DEC -176236.02 B-25  
2J3RE/J2 2DEC\* -.1355426363 E5 B-27\*  
2DEC\* .3067493316 E18 B-60\*  
J2REQSQ 2DEC\* 1.75501139 E21 B-72\*  
3J22R2MU 2DEC\* 9.20479048 E16 B-58\*

# \*\*\*\*\*

SETLOC TOF-FF1  
BANK  
COUNT\* \$\$/TFF

1/RTMU 2DEC\* .5005750271 E-5 B17\* # MODIFIED EARTH MU

SETLOC SBAND  
BANK  
COUNT\* \$\$/R05

REMDIST 2DEC 384402000 B-29 # MEAN DISTANCE BETWEEN EARTH AND MOON.

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# PHYSICAL CONSTANTS (TIME - VARIANT)

SETLOC	STARTAB			
BANK				
COUNT*	\$\$/STARS			
2DEC	+.8342971408 B-1	# STAR 37	X	
2DEC	-.2392481515 B-1	# STAR 37	Y	
2DEC	-.4966976975 B-1	# STAR 37	Z	
2DEC	+.8139832631 B-1	# STAR 36	X	
2DEC	-.5557243189 B-1	# STAR 36	Y	
2DEC	+.1691204557 B-1	# STAR 36	Z	
2DEC	+.4541086270 B-1	# STAR 35	X	
2DEC	-.5392368197 B-1	# STAR 35	Y	
2DEC	+.7092312789 B-1	# STAR 35	Z	
2DEC	+.3201817378 B-1	# STAR 34	X	
2DEC	-.4436021946 B-1	# STAR 34	Y	
2DEC	-.8370786986 B-1	# STAR 34	Z	
2DEC	+.5520184464 B-1	# STAR 33	X	
2DEC	-.7933187400 B-1	# STAR 33	Y	
2DEC	-.2567508745 B-1	# STAR 33	Z	
2DEC	+.4537196908 B-1	# STAR 32	X	
2DEC	-.8779508801 B-1	# STAR 32	Y	
2DEC	+.1527766153 B-1	# STAR 32	Z	
2DEC	+.2069525789 B-1	# STAR 31	X	
2DEC	-.8719885748 B-1	# STAR 31	Y	
2DEC	-.4436288486 B-1	# STAR 31	Z	
2DEC	+.1217293692 B-1	# STAR 30	X	
2DEC	-.7702732847 B-1	# STAR 30	Y	

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2DEC	+.6259880410 B-1	# STAR 30	Z
2DEC	-.1124304773 B-1	# STAR 29	X
2DEC	-.9694934200 B-1	# STAR 29	Y
2DEC	+.2178116072 B-1	# STAR 29	Z
2DEC	-.1146237858 B-1	# STAR 28	X
2DEC	-.3399692557 B-1	# STAR 28	Y
2DEC	-.9334250333 B-1	# STAR 28	Z
2DEC	-.3516499609 B-1	# STAR 27	X
2DEC	-.8240752703 B-1	# STAR 27	Y
2DEC	-.4441196390 B-1	# STAR 27	Z
2DEC	-.5326876930 B-1	# STAR 26	X
2DEC	-.7160644554 B-1	# STAR 26	Y
2DEC	+.4511047742 B-1	# STAR 26	Z
2DEC	-.7861763936 B-1	# STAR 25	X
2DEC	-.5217996305 B-1	# STAR 25	Y
2DEC	+.3311371675 B-1	# STAR 25	Z
2DEC	-.6898393233 B-1	# STAR 24	X
2DEC	-.4182330640 B-1	# STAR 24	Y
2DEC	-.5909338474 B-1	# STAR 24	Z
2DEC	-.5812035376 B-1	# STAR 23	X
2DEC	-.2909171294 B-1	# STAR 23	Y
2DEC	+.7599800468 B-1	# STAR 23	Z
2DEC	-.9170097662 B-1	# STAR 22	X
2DEC	-.3502146628 B-1	# STAR 22	Y
2DEC	-.1908999176 B-1	# STAR 22	Z

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2DEC	-.4523440203	B-1	# STAR 21	X
2DEC	-.0493710140	B-1	# STAR 21	Y
2DEC	-.8904759346	B-1	# STAR 21	Z
2DEC	-.9525211695	B-1	# STAR 20	X
2DEC	-.0593434796	B-1	# STAR 20	Y
2DEC	-.2986331746	B-1	# STAR 20	Z
2DEC	-.9656605484	B-1	# STAR 19	X
2DEC	+.0525933156	B-1	# STAR 19	Y
2DEC	+.2544280809	B-1	# STAR 19	Z
2DEC	-.8608205219	B-1	# STAR 18	X
2DEC	+.4636213989	B-1	# STAR 18	Y
2DEC	+.2098647835	B-1	# STAR 18	Z
2DEC	-.7742591356	B-1	# STAR 17	X
2DEC	+.6152504197	B-1	# STAR 17	Y
2DEC	-.1482892839	B-1	# STAR 17	Z
2DEC	-.4657947941	B-1	# STAR 16	X
2DEC	+.4774785033	B-1	# STAR 16	Y
2DEC	+.7450164351	B-1	# STAR 16	Z
2DEC	-.3612508532	B-1	# STAR 15	X
2DEC	+.5747270840	B-1	# STAR 15	Y
2DEC	-.7342932655	B-1	# STAR 15	Z
2DEC	-.4118589524	B-1	# STAR 14	X
2DEC	+.9065485360	B-1	# STAR 14	Y
2DEC	+.0924226975	B-1	# STAR 14	Z
2DEC	-.1820751783	B-1	# STAR 13	X

21  $\langle \text{Page LM0050 21} \rangle \equiv$ 

(8 724)

2DEC	+.9404899869 B-1	# STAR 13	Y
2DEC	-.2869271926 B-1	# STAR 13	Z
2DEC	-.0614937230 B-1	# STAR 12	X
2DEC	+.6031563286 B-1	# STAR 12	Y
2DEC	-.7952489957 B-1	# STAR 12	Z
2DEC	+.1371725575 B-1	# STAR 11	X
2DEC	+.6813721061 B-1	# STAR 11	Y
2DEC	+.7189685267 B-1	# STAR 11	Z
2DEC	+.2011399589 B-1	# STAR 10	X
2DEC	+.9690337941 B-1	# STAR 10	Y
2DEC	-.1432348512 B-1	# STAR 10	Z
2DEC	+.3507315038 B-1	# STAR 9	X
2DEC	+.8926333307 B-1	# STAR 9	Y
2DEC	+.2831839492 B-1	# STAR 9	Z
2DEC	+.4105636020 B-1	# STAR 8	X
2DEC	+.4988110001 B-1	# STAR 8	Y
2DEC	+.7632988371 B-1	# STAR 8	Z
2DEC	+.7032235469 B-1	# STAR 7	X
2DEC	+.7075846047 B-1	# STAR 7	Y
2DEC	+.0692868685 B-1	# STAR 7	Z
2DEC	+.5450107404 B-1	# STAR 6	X
2DEC	+.5314955466 B-1	# STAR 6	Y
2DEC	-.6484410356 B-1	# STAR 6	Z
2DEC	+.0130968840 B-1	# STAR 5	X
2DEC	+.0078062795 B-1	# STAR 5	Y

22 (Page LM0051 22)≡

(8 724)

2DEC	+.9998837600 B-1	# STAR 5	Z
2DEC	+.4917678276 B-1	# STAR 4	X
2DEC	+.2204887125 B-1	# STAR 4	Y
2DEC	-.8423473935 B-1	# STAR 4	Z
2DEC	+.4775639450 B-1	# STAR 3	X
2DEC	+.1166004340 B-1	# STAR 3	Y
2DEC	+.8708254803 B-1	# STAR 3	Z
2DEC	+.9342640400 B-1	# STAR 2	X
2DEC	+.1735073142 B-1	# STAR 2	Y
2DEC	-.3115219339 B-1	# STAR 2	Z
2DEC	+.8748658918 B-1	# STAR 1	X
2DEC	+.0260879174 B-1	# STAR 1	Y
2DEC	+.4836621670 B-1	# STAR 1	Z

CATLOG DEC 6970

# \*\*\*\*\*

SETLOC EPHEM1  
 BANK  
 COUNT\* \$\$/EPHEM

KONMAT	2DEC	1.0 B-1	# *****
	2DEC	0	# *
	2DEC	0	# *
	2DEC	0	# *
	2DEC	.91745 B-1	# K1 COS(OBL) *
	2DEC	-.03571 B-1	# K2 SIN(OBL)SIN(IM) *
	2DEC	0	# *
	2DEC	.39784 B-1	# K3 SIN(OBL) *

23a (Page LM0052 23a)≡

(8 724)

	2DEC	.082354 B-1	# K4 COS(OBL)SIN(IM) *	
CSTODAY	2DEC	8640000 B-33	#	* NOTE: *
RCB-13	OCT	00002	#	* TABLES CONTAIN *
	OCT	00000	#	* CONSTANTS FOR *
RATESP	2DEC	.03660098 B+4	# LOMR	* 1969 - 1970 *
	2DEC	.00273779 B+4	# LOSR	
	2DEC	-.00014719 B+4	# LONR	
	2DEC	.815282336	# LOMO	
	2DEC	.274674910	# LOSO	
	2DEC	.986209499	# LONO	
VAL67	2DEC*	.01726666666 B+1*	# AMOD	
	2DEC	.530784445	# AARG	
	2DEC	.036291712 B+1	# 1/27	
	2DEC	.003505277 B+1	# BMOD	
	2DEC	.585365625	# BARG	
	2DEC	.03125 B+1	# 1/32	
	2DEC	.005325277 B+1	# CMOD	
	2DEC	-.01106341036	# CARG	
	2DEC	.002737925 B+1	# 1/365	

# \*\*\*\*\*

SETLOC PLANTIN2  
 BANK  
 COUNT\* \$\$/LUROT

COSI	2DEC	.99964173 B-1	# COS (5521.5 SEC.) B-1
SINI	2DEC	.02676579 B-1	# SIN (5521.5 SEC.) B-1
NODDOT	2DEC	-.457335121 E-2	# REV/CSEC B+28 = -1.07047011 E-8 RAD/SEC
FDOT	2DEC	.570863327	# REV/CSEC B+27 = 2.67240410 E-6 RAD/SEC

23b (Page LM0053 23b)≡

(8 724)

BDOT	2DEC	-3.07500686 E-8	# REV/CSEC B+28 = -7.19757301 E-14 RAD/SEC
NODIO	2DEC	.986209434	# REVS B-D = 6.19653663041 RAD
FSUBO	2DEC	.829090536	# REVS B-D = 5.20932947829 RAD
BSUBO	2DEC	.0651201393	# REVS B-D = 0.40916190299 RAD
WEARTH	2DEC	.973561595	# REV/CSEC B+23 = 7.29211494 E-5 RAD/SEC

## 1.2 input output channel bit descriptions

$$\begin{aligned}
 24 \quad \langle \textit{input output channel bit descriptions 24} \rangle \equiv & \quad (7) \\
 & \langle \textit{Page LM0054 25} \rangle \\
 & \langle \textit{Page LM0055 26} \rangle \\
 & \langle \textit{Page LM0056 27} \rangle \\
 & \langle \textit{Page LM0057 28} \rangle \\
 & \langle \textit{Page LM0058 29} \rangle \\
 & \langle \textit{Page LM0059 30a} \rangle \\
 & \langle \textit{Page LM0060 30b} \rangle
 \end{aligned}$$



# \*\*\* CHANNEL DESCRIPTION WORDS ARE ALLOCATED IN ERASABLE ASSIGNMENTS \*\*\*

# CHANNEL 1 IDENTICAL TO COMPUTER REGISTER L (0001)

# CHANNEL 2 IDENTICAL TO COMPUTER REGISTER Q (0002)

# CHANNEL 3 HISCALAR: INPUT CHANNEL; MOST SIGNIFICANT 14 BITS FROM 33 STAGE BINARY COUNTER.  
# FACTOR IS B23 IN CSEC, SO MAX VALUE ABOUT 23.3 HOURS AND LEAST SIGNIFICANT BIT

# CHANNEL 4 LOSCALAR: INPUT CHANNEL; NEXT MOST SIGNIFICANT 14 BITS FROM THE 33 STAGE BINARY  
# ASSOCIATED WITH CHANNEL 3. SCALE FACTOR IS B9 IN CSEC. SO MAX VAL IS 5.12 SEC A  
# SIGNIFICANT BIT IS 1/3200 SEC. SCALE FACTOR OF D.P. WORD WITH CHANNEL 3 IS B23

# CHANNEL 5 PYJETS: OUTPUT CHANNEL; PITCH RCS JET CONTROL. (REACTION CONTROL SYSTEM) USES

# CHANNEL 6 ROLLJETS: OUTPUT CHANNEL; ROLL RCS JET CONTROL. (REACTION CONTROL SYSTEM) USES

# CHANNEL 7 SUPERBNK: OUTPUT CHANNEL; NOT RESET BY RESTART; FIXED EXTENSION BITS USED TO SE  
# APPROPRIATE FIXED MEMORY BANK IF FBANK IS 30 OCTAL OR MORE. USES BITS 5-7.

# CHANNEL 10 OUTO: OUTPUT CHANNEL; REGISTER USED TO TRANSMIT LATCHING-RELAY DRIVING INFORMAT  
# THE DISPLAY SYSTEM. BITS 15-12 ARE SET TO THE ROW NUMBER (1-14 OCTAL) OF THE R  
# CHANGED AND BITS 11-1 CONTAIN THE REQUIRED SETTINGS FOR THE RELAYS IN THE ROW.

# CHANNEL 11 DSALMOUT: OUTPUT CHANNEL; REGISTER WHOSE BITS ARE USED FOR ENGINE ON-OFF CONTR  
# DRIVE INDIVIDUAL INDICATORS OF THE DISPLAY SYSTEM. BITS 1-7 ARE A RELAYS.

#  
# BIT 1 ISS WARNING  
# BIT 2 LIGHT COMPUTER ACTIVITY LAMP  
# BIT 3 LIGHT UPLINK ACTIVITY LAMP  
# BIT 4 LIGHT TEMP CAUTION LAMP  
# BIT 5 LIGHT KEYBOARD RELEASE LAMP  
# BIT 6 FLASH VERB AND NOUN LAMPS  
# BIT 7 LIGHT OPERATOR ERROR LAMP

26 (Page LM0055 26)≡

(24 739)

#	BIT 8	SPARE
#	BIT 9	TEST CONNECTOR OUTBIT
#	BIT 10	CAUTION RESET
#	BIT 11	SPARE
#	BIT 12	SPARE
#	BIT 13	ENGINE ON
#	BIT 14	ENGINE OFF
#	BIT 15	SPARE
# CHANNEL 12	CHAN12: OUTPUT CHANNEL; BITS USED TO DRIVE NAVIGATION AND SPACECRAFT	
#		
#	BIT 1	ZERO RR CDU; CDU'S GIVE RRADAR INFORMATION FOR LM
#	BIT 2	ENABLE CDU RADAR ERROR COUNTERS
#	BIT 3	NOT USED
#	BIT 4	COARSE ALIGN ENABLE OF IMU
#	BIT 5	ZERO IMU CDU'S
#	BIT 6	ENABLE IMU ERROR COUNTER, CDU ERROR COUNTER.
#	BIT 7	SPARE
#	BIT 8	DISPLAY INERTIAL DATA
#	BIT 9	-PITCH GIMBAL TRIM (BELL MOTION) DESCENT ENGINE
#	BIT 10	+PITCH GIMBAL TRIM (BELL MOTION) DESCENT ENGINE
#	BIT 11	-ROLL GIMBAL TRIM (BELL MOTION) DESCENT ENGINE
#	BIT 12	+ROLL GIMBAL TRIM (BELL MOTION) DESCENT ENGINE
#	BIT 13	LR POSITION 2 COMMAND
#	BIT 14	ENABLE RENDEZVOUS RADAR LOCK-ON; AUTO ANGLE TRACK'G
#	BIT 15	ISS TURN ON DELAY COMPLETE

27 (Page LM0056 27)≡

(24 739)

```

# CHANNEL 13  CHAN13: OUTPUT CHANNEL.
#
#          BIT 1          RADAR C          PROPER SETTING OF THE A,B,C MATRIX
#          BIT 2          RADAR B          SELECTS CERTAIN RADAR
#          BIT 3          RADAR A          PARAMETERS TO BE READ.
#          BIT 4          RADAR ACTIVITY
#          BIT 5          NOT USED (CONNECTS AN ALTERNATE INPUT TO UPLINK)
#          BIT 6          BLOCK INPUTS TO UPLINK CELL
#          BIT 7          DOWNLINK TELEMETRY WORD ORDER CODE BIT
#          BIT 8          RHC COUNTER ENABLE (READ HAND CONTROLLER ANGLES)
#          BIT 9          START RHC READ INTO COUNTERS IS BIT 8 SET
#          BIT 10         TEST ALARMS, TEST DSKY LIGHTS
#          BIT 11         ENABLE STANDBY
#          BIT 12         RESET TRAP 31-A          ALWAYS APPEAR TO BE SET TO 0
#          BIT 13         RESET TRAP 31-B          ALWAYS APPEAR TO BE SET TO 0
#          BIT 14         RESET TRAP 32           ALWAYS APPEAR TO BE SET TO 0
#          BIT 15         ENABLE T6 RUPT

# CHANNEL 14  CHAN14: OUTPUT CHANNEL; USED TO CONTROL COMPUTER COUNTER CELLS (CDU, GYRO, SPAC
#
#          BIT 1          OUTLINK ACTIVITY (NOT USED)
#          BIT 2          ALTITUDE RATE OR ALTITUDE SELECTOR
#          BIT 3          ALTITUDE METER ACTIVITY
#          BIT 4          THRUST DRIVE ACTIVITY FOR DESCENT ENGINE
#          BIT 5          SPARE
#          BIT 6          GYRO ENABLE POWER FOR PULSES
#          BIT 7          GYRO SELECT B          PAIR OF BITS IDENTIFIES AXIS OF
#          BIT 8          GYRO SELECT A          GYRO SYSTEM TO BE TORQUED.
#          BIT 9          GYRO TORQUING COMMAND IN NEGATIVE DIRECTION.

```

28 (Page LM0057 28)≡

(24 739)

#	BIT 10	GYRO ACTIVITY
#	BIT 11	DRIVE CDU S
#	BIT 12	DRIVE CDU T
#	BIT 13	DRIVE CDU Z
#	BIT 14	DRIVE CDU Y
#	BIT 15	DRIVE CDU X

# CHANNEL 15 MNKEYIN: INPUT CHANNEL; KEY CODE INPUT FROM KEYBOARD OF DSKY, SENSED  
 # PROGRAM INTERRUPT #5 IS RECEIVED. USED BITS 5-1

# CHANNEL 16 NAVKEYIN: INPUT CHANNEL; OPTICS MARK INFORMATION AND NAVIGATION PANEL  
 # CONTROL (LM) SENSED BY PROGRAM THEN PROGRAM INTERRUPT #6 IS RECEIVED

#	BIT 1	NOT ASSIGNED
#	BIT 2	NOT ASSIGNED
#	BIT 3	OPTICS X-AXIS MARK SIGNAL FOR ALIGN OPTICAL TSCOPE
#	BIT 4	OPTICS Y-AXIS MARK SIGNAL FOR AOT
#	BIT 5	OPTICS MARK REJECT SIGNAL
#	BIT 6	DESCENT+ ; CREW DESIRED SLOWING RATE OF DESCENT
#	BIT 7	DESCENT- ; CREW DESIRED SPEEDING UP RATE OF D'CENT

# NOTE: ALL BITS IN CHANNELS 30-33 ARE INVERTED AS SENSED BY THE PROGRAM, SO THAT A V  
 # THAT THE INDICATED SIGNAL IS PRESENT.

# CHANNEL 30 INPUT CHANNEL

#	BIT 1	ABORT WITH DESCENT STAGE
#	BIT 2	UNUSED
#	BIT 3	ENGINE ARMED SIGNAL
#	BIT 4	ABORT WITH ASCENT ENGINE STAGE
#	BIT 5	AUTO THROTTLE; COMPUTER CONTROL OF DESCENT ENGINE

29 (Page LM0058 29)≡

(24 739)

```

# BIT 6 DISPLAY INERTIAL DATA
# BIT 7 RR CDU FAIL
# BIT 8 SPARE
# BIT 9 IMU OPERATE WITH NO MALFUNCTION
# BIT 10 LM COMPUTER (NOT AGS) HAS CONTROL OF LM.
# BIT 11 IMU CAGE COMMAND TO DRIVE IMU GIMBAL ANGLES TO 0.
# BIT 12 IMU CDU FAIL (MALFUNCTION OF IMU CDU,S)
# BIT 13 IMU FAIL (MALFUNCTION OF IMU STABILIZATION LOOPS)
# BIT 14 ISS TURN ON REQUESTED
# BIT 15 TEMPERATURE OF STABLE MEMBER WITHIN DESIGN LIMITS

```

```

# CHANNEL 31 INPUT CHANNEL; BITS ASSOCIATED WITH THE ATTITUDE CONTROLLER, TRANSLATIONAL CONT
# AND SPACECRAFT ATTITUDE CONTROL; USED BY RCS DAP.
#

```

```

# BIT 1 ROTATION (BY RHC) COMMANDED IN POSITIVE PITCH DIRECTION; MUST BE
# ALSO POSITIVE ELEVATION CHANGE FOR LANDING POINT DESIGNATOR
# BIT 2 AS BIT 1 EXCEPT NEGATIVE PITCH AND ELEVATION.
# BIT 3 ROTATION (BY RHC) COMMANDED IN POSITIVE YAW DIRECTION; MUST BE
# AS BIT 3 EXCEPT NEGATIVE YAW
# BIT 5 ROTATION (BY RHC) COMMANDED IN POSITIVE ROLL DIRECTION; MUST BE
# ALSO POSITIVE AZIMUTH CHANGE FOR LANDING POINT DESIGNATOR.
# BIT 6 AS BIT 5 EXCEPT NEGATIVE ROLL AND AZIMUTH
# BIT 7 TRANSLATION IN +X DIRECTION COMMANDED BY THC
# BIT 8 TRANSLATION IN -X DIRECTION COMMANDED BY THC
# BIT 9 TRANSLATION IN +Y DIRECTION COMMANDED BY THC
# BIT 10 TRANSLATION IN -Y DIRECTION COMMANDED BY THC
# BIT 11 TRANSLATION IN +Z DIRECTION COMMANDED BY THC
# BIT 12 TRANSLATION IN -Z DIRECTION COMMANDED BY THC

```

30a  $\langle$ Page LM0059 30a $\rangle \equiv$ 

(24 739)

# BIT 13 ATTITUDE HOLD MODE ON SCS MODE CONTROL SWITCH  
 # BIT 14 AUTO STABILIZATION OF ATTITUDE ON SCS MODE SWITCH  
 # BIT 15 ATTITUDE CONTROL OUT OF DETENT (RHC NOT IN NEUTRAL)

# CHANNEL 32 INPUT CHANNEL.

#  
 # BIT 1 THRUSTERS 2 & 4 DISABLED BY CREW  
 # BIT 2 THRUSTERS 5 & 8 DISABLED BY CREW  
 # BIT 3 THRUSTERS 1 & 3 DISABLED BY CREW  
 # BIT 4 THRUSTERS 6 & 7 DISABLED BY CREW  
 # BIT 5 THRUSTERS 14 & 16 DISABLED BY CREW  
 # BIT 6 THRUSTERS 13 & 15 DISABLED BY CREW  
 # BIT 7 THRUSTERS 9 & 12 DISABLED BY CREW  
 # BIT 8 THRUSTERS 10 & 11 DISABLED BY CREW  
 # BIT 9 DESCENT ENGINE DISABLED BY CREW  
 # BIT 10 APPARENT DESCENT ENGINE GIMBAL FAILURE  
 # BIT 14 INDICATES PROCEED KEY IS DEPRESSED

# CHANNEL 33 CHAN33: INPUT CHANNEL; FOR HARDWARE STATUS AND COMMAND INFORMATION.  
 # FLOP BITS RESET BY A CHANNEL "WRITE" COMMAND THAT ARE RESET BY A RES

#  
 # BIT 1 SPARE  
 # BIT 2 RR AUTO-POWER ON  
 # BIT 3 RR RANGE LOW SCALE  
 # BIT 4 RR DATA GOOD  
 # BIT 5 LR RANGE DATA GOOD  
 # BIT 6 LR POS1  
 # BIT 7 LR POS2

30b  $\langle$ Page LM0060 30b $\rangle \equiv$ 

(24 739)

# BIT 8 LR VEL DATA GOOD  
 # BIT 9 LR RANGE LOW SCALE  
 # BIT 10 BLOCK UPLINK INPUT  
 # BIT 11 UPLINK TOO FAST  
 # BIT 12 DOWNLINK TOO FAST  
 # BIT 13 PIPA FAIL  
 # BIT 14 WARNING OF REPEATED ALARMS: RESTART, COUNTER FAIL, VO  
 # BIT 15 LGC OSCILLATOR STOPPED

# CHANNEL 34 DNT M1: OUTPUT CHANNEL; DOWNLINK 1: FIRST OF TWO WORDS SERIALIZATION

# CHANNEL 35 DNT M2: OUTPUT CHANNEL; DOWNLINK 2: SECOND OF TWO WORDS SERIALIZATION

### 1.3 flagword assignments

$$\begin{aligned}
 31 \quad \langle \textit{flagword assignments } 31 \rangle \equiv & \quad (7) \\
 & \langle \textit{Page LM0061 } 32 \rangle \\
 & \langle \textit{Page LM0062 } 33 \rangle \\
 & \langle \textit{Page LM0063 } 35 \rangle \\
 & \langle \textit{Page LM0064 } 37 \rangle \\
 & \langle \textit{Page LM0065 } 39 \rangle \\
 & \langle \textit{Page LM0066 } 40 \rangle \\
 & \langle \textit{Page LM0067 } 41 \rangle \\
 & \langle \textit{Page LM0068 } 42 \rangle \\
 & \langle \textit{Page LM0069 } 43 \rangle \\
 & \langle \textit{Page LM0070 } 44 \rangle \\
 & \langle \textit{Page LM0071 } 45 \rangle \\
 & \langle \textit{Page LM0072 } 46 \rangle \\
 & \langle \textit{Page LM0073 } 48 \rangle \\
 & \langle \textit{Page LM0074 } 49 \rangle \\
 & \langle \textit{Page LM0075 } 50 \rangle \\
 & \langle \textit{Page LM0076 } 51 \rangle \\
 & \langle \textit{Page LM0077 } 52 \rangle \\
 & \langle \textit{Page LM0078 } 53 \rangle \\
 & \langle \textit{Page LM0079 } 54 \rangle \\
 & \langle \textit{Page LM0080 } 55 \rangle \\
 & \langle \textit{Page LM0081 } 56 \rangle \\
 & \langle \textit{Page LM0082 } 57 \rangle \\
 & \langle \textit{Page LM0083 } 59 \rangle \\
 & \langle \textit{Page LM0084 } 60 \rangle \\
 & \langle \textit{Page LM0085 } 61 \rangle \\
 & \langle \textit{Page LM0086 } 62 \rangle \\
 & \langle \textit{Page LM0087 } 63 \rangle \\
 & \langle \textit{Page LM0088 } 64\text{a} \rangle \\
 & \langle \textit{Page LM0089 } 64\text{b} \rangle
 \end{aligned}$$

32 (Page LM0061 32)≡

(31 733)

# FLAGWORDS 0-11

ARE DOWNLINKED AND CAN BE SET AND CLEARED BY UP-FLAG AND DOWN-FLAG.  
 INTERPRETER. THESE WERE PREVIOUSLY LISTED UNDER "INTERPRETING".  
 THE ERASABLE LOG SECTION. FLAGWORDS 12 & 13 WERE PREVIOUSLY LISTED  
 ARE STILL DOWNLINKED UNDER THOSE NAMES.

## # ALPHABETICAL LIST OF FLAGWORDS

#

# FLAGWORD	DEC. NUMBER	BIT AND FLAG	BIT NAME
# ACCOKFLG	207	BIT 3 FLAG 13	ACCSOKAY
# ACC4-2FL	199	BIT 11 FLAG 13	ACC4OR2X
# ACMODFLG	032	BIT 13 FLAG 2	ACMODBIT
# ALTSCALE	186	BIT 9 FLAG 12	ALTSCBIT
# ANTENFLG	183	BIT 12 FLAG 12	ANTENBIT
# AORBSFLG	205	BIT 5 FLAG 13	AORBSYST
# AORBTRFLG	200	BIT 10 FLAG 13	AORBTRAN
# APSESW	130	BIT 5 FLAG 8	APSESBIT
# APSFLAG	152	BIT 13 FLAG 10	APSFLBIT
# ASTNFLAG	108	BIT 12 FLAG 7	ASTNBIT
# ATTFLAG	104	BIT 1 FLAG 6	ATTFLBIT
# AUTOMODE	193	BIT 2 FLAG 12	AUTOMBIT
# AUTR1FLG	209	BIT 1 FLAG 13	AUTRATE1
# AUTR2FLG	208	BIT 2 FLAG 13	AUTRATE2
# AUXFLAG	103	BIT 2 FLAG 6	AUXFLBIT
# AVEGFLAG	115	BIT 5 FLAG 7	AVEGFBIT
# AVEMIDSW	149	BIT 1 FLAG 9	AVEMDBIT
# AVFLAG	040	BIT 5 FLAG 2	AVFLBIT
# CALCMAN2	043	BIT 2 FLAG 2	CALC2BIT
# CALCMAN3	042	BIT 3 FLAG 2	CALC3BIT
# CDESFLAG	180	BIT 15 FLAG 12	CDESBIT
# CMOONFLG	123	BIT 12 FLAG 8	CMOONBIT
# COGAFLAG	131	BIT 4 FLAG 8	COGAFBIT
# CSMDKFLG	197	BIT 13 FLAG 13	CSMDOCKD
# CULTFLAG	053	BIT 7 FLAG 3	CULTBIT
# DAPBOOLS		FLGWRD13	
# DBSELFLG	206	BIT 4 FLAG 13	DBSELECT
# DESIGFLG	185	BIT 10 FLAG 12	DESIGBIT
# DIDFLAG	016	BIT 14 FLAG	DIDFLBIT
# DIMOFLAG	059	BIT 1 FLAG 3	DIMOBIT
# DMENFLG	081	BIT 9 FLAG 5	DMENFBIT
# DRIFTDFL	202	BIT 8 FLAG 13	DRIFTBIT
# DRIFTFLG	030	BIT 15 FLAG 2	DRFTBIT
# DSKYFLAG	075	BIT 15 FLAG 5	DSKYFBIT



33 (Page LM0062 33)≡

(31 733)

# D6OR9FLG	058	BIT 2 FLAG 3	D6OR9BIT
# ENGONFLG	083	BIT 7 FLAG 5	ENGONBIT
# ERADFLAG	017	BIT 13 FLAG 1	ERADFBIT
# ETPIFLAG	038	BIT 7 FLAG 2	ETPIBIT
# FINALFLG	039	BIT 6 FLAG 2	FINALBIT
# FLAGWRD0	(000-014)	(STATE +0)	
# FLAGWRD1	(015-029)	(STATE +1)	
# FLAGWRD2	(030-044)	(STATE +2)	
# FLAGWRD3	(045-059)	(STATE +3)	
# FLAGWRD4	(060-074)	(STATE +4)	
# FLAGWRD5	(075-089)	(STATE +5)	
# FLAGWRD6	(090-104)	(STATE +6)	
# FLAGWRD7	(105-119)	(STATE +7)	
# FLAGWRD8	(120-134)	(STATE +8D)	
# FLAGWRD9	(135-149)	(STATE +9D)	
# FLAP	142	BIT 8 FLAG 9	FLAPBIT
# FLGWRD10	(150-164)	(STATE +10D)	
# FLGWRD11	(165-179)	(STATE +11D)	
# FLGWRD12	(180-194)	(STATE +12D)	
# FLGWRD13	(195-209)	(STATE +13D)	
# FLPC	138	BIT 12 FLAG 9	FLPCBIT
# FLPI	139	BIT 11 FLAG 9	FLPIBIT
# FLRCS	149	BIT 10 FLAG 9	FLRCSBIT
# FLUNDISP	125	BIT 10 FLAG 8	FLUNDBIT
# FLVR	136	BIT 14 FLAG 9	FLVRBIT
# FREEFLAG	012	BIT 3 FLAG 0	FREEFBIT
# FSPASFLG	005	BIT 10 FLAG 0	FSPASBIT
# GLOKFAIL	046	BIT 14 FLAG 3	GLOKFBIT
# GMBDRVSW	095	BIT 10 FLAG 6	GMBDRBIT
# GUESSW	028	BIT 2 FLAG 1	GUESSBIT
# HFLSHFLG	179	BIT 1 FLAG 11	HFLSHBIT
# IDLEFLAG	113	BIT 7 FLAG 7	IDLEFBIT
# IGNFLAG	107	BIT 13 FLAG 7	IGNFLBIT
# IMPULSW	036	BIT 9 FLAG 2	IMPULBIT
# IMUSE	007	BIT 8 FLAG 0	IMUSEBIT
# INFINFLG	128	BIT 7 FLAG 8	INFINBIT
# INITALGN	133	BIT 2 FLAG 8	INITABIT
# INTFLAG	151	BIT 14 FLAG 10	INTFLBIT
# INTYPFLG	056	BIT 4 FLAG 3	INTYPBIT
# ITSWICH	105	BIT 15 FLAG 7	ITSWBIT
# JSWITCH	001	BIT 14 FLAG 0	JSWCHBIT
# LETABORT	141	BIT 9 FLAG 9	LETABBIT
# LMOONFLG	124	BIT 11 FLAG 8	LMOONBIT
# LOKONSW	010	BIT 5 FLAG 0	LOKONBIT
# LOSCMFLG	033	BIT 12 FLAG 2	LOSCMBIT

EQUIVALENT FLAG NAME: D

# LRALTFLG	190	BIT 5 FLAG 12	LRALTBIT
# LRBYPASS	165	BIT 15 FLAG 11	LRBYBIT
# LRINH	172	BIT 8 FLAG 11	LRINHBIT
# LRPOSFLG	189	BIT 6 FLAG 12	LRPOSBIT
# LRVELFLG	187	BIT 8 FLAG 12	LRVELBIT

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(31 733)

# LUNAFLAG	048	BIT 12 FLAG 3	LUNABIT
# MANUFLAG	106	BIT 14 FLAG 7	MANUFBIT
# MGLVFLAG	088	BIT 2 FLAG 5	MGLVFBIT
# MIDAVFLG	148	BIT 2 FLAG 9	MIDAVBIT
# MIDFLAG	002	BIT 13 FLAG 0	MIDFLBIT
# MID1FLAG	147	BIT 3 FLAG 9	MID1BIT
# MKOVFLAG	072	BIT 3 FLAG 4	MKOVBIT
# MOONFLAG	003	BIT 12 FLAG 0	MOONBIT
# MRKIDFLG	060	BIT 15 FLAG 4	MRKIDBIT
# MRKNVFLG	066	BIT 9 FLAG 4	MRKNVBIT
# MRUPTFLG	070	BIT 5 FLAG 4	MRUPTBIT
# MUNFLAG	097	BIT 8 FLAG 6	MUNFLBIT
# MWAITFLG	064	BIT 11 FLAG 4	MWAITBIT
# NEEDLFLG	011	BIT 4 FLAG 0	NEEDLBIT
# NEWIFLG	122	BIT 13 FLAG 8	NEWIBIT
# NJETSFLG	015	BIT 15 FLAG	NJETSBIT
# NODOFLAG	044	BIT 1 FLAG 2	NODOBIT
# NOLRREAD	170	BIT 10 FLAG 11	NOLRRBIT
# NORMSW	110	BIT 10 FLAG 7	NORMSBIT
# NORRMON	086	BIT 4 FLAG 5	NORRMBIT
# NOR29FLG	049	BIT 11 FLAG 3	NR29FBIT
# NOTHROTL	078	BIT 12 FLAG 5	NOTHRBIT
# NOUPFLAG	024	BIT 6 FLAG 1	NOUPFBIT
# NRMNVFLG	067	BIT 8 FLAG 4	NRMNVBIT
# NRMIDFLG	062	BIT 13 FLAG 4	NRMIDBIT
# NRUPTFLG	071	BIT 4 FLAG 4	NRUPTBIT
# NTARGFLG	102	BIT 3 FLAG 6	NTARGBIT
# NWAITFLG	065	BIT 10 FLAG 4	NWAITBIT
# OLDESEFLG	014	BIT 1 FLAG 0	OLDESBIT
# OPTNSW	038	BIT 7 FLAG 2	OPTNBIT
# ORBWFLAG	054	BIT 6 FLAG 3	ORBWFBIT
# ORDERSW	129	BIT 6 FLAG 8	ORDERBIT
# OURRCFLG	198	BIT 12 FLAG 13	OURRCBIT
# PDSPFLAG	063	BIT 12 FLAG 4	PDSPFBIT
# PFRATFLG	041	BIT 4 FLAG 2	PFRATBIT
# PINBRFLG	069	BIT 6 FLAG 4	PINBRBIT
# PRECIFLG	052	BIT 8 FLAG 3	PRECIBIT
# PRIODFLG	061	BIT 14 FLAG 1	PRIODBIT
# PRONVFLG	068	BIT 7 FLAG 4	PRONVBIT
# PSTHIGAT	169	BIT 11 FLAG 11	PSTHIBIT
# PULSEFLG	195	BIT 15 FLAG 13	PULSES
# P21FLAG	004	BIT 11 FLAG 0	P21FLBIT
# P25FLAG	006	BIT 9 FLAG 0	P25FLBIT
# P39/79SW	126	BIT 9 FLAG 8	P39SWBIT
# QUITFLAG	145	BIT 5 FLAG 9	QUITBIT

EQUIVALENT FLAG NAME: E

# RADMODES		FLGWRD12	
# RASFLAG		FLGWRD10	
# RCDUFAIL	188	BIT 7 FLAG 12	RCDUFBIT
# RCDUOFLG	182	BIT 13 FLAG 12	RCDUOBIT
# READLR	174	BIT 6 FLAG 11	READLBIT

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(31 733)

# READRFLG	051	BIT 9 FLAG 3	READRBIT
# READVEL	175	BIT 5 FLAG 11	READVBIT
# REDFLAG	099	BIT 6 FLAG 6	REDFLBIT
# REFSMFLG	047	BIT 13 FLAG 3	REFSMBIT
# REINTFLG	158	BIT 7 FLAG 10	REINTBIT
# REMODFLG	181	BIT 14 FLAG 12	REMODBIT
# RENDWFLG	089	BIT 1 FLAG 5	RENDWBIT
# REPOS MON	184	BIT 11 FLAG 12	REPOSBIT
# RHCSCFLG	203	BIT 7 FLAG 13	RHCSCALE
# RNDVZFLG	008	BIT 7 FLAG 0	RNDVZBIT
# RNGEDATA	176	BIT 4 FLAG 11	RNGEDBIT
# RNGSCFLG	080	BIT 10 FLAG 5	RNGSCBIT
# RODFLAG	018	BIT 12 FLAG 1	RODFLBIT
# ROTFLAG	144	BIT 6 FLAG 9	ROTFLBIT
# RPQFLAG	120	BIT 15 FLAG 8	RPQFLBIT
# RRDATAFL	191	BIT 4 FLAG 12	RRDATABT
# RRNBSW	009	BIT 6 FLAG 0	RRNBBIT
# RRRSFLAG	192	BIT 3 FLAG 12	RRRSBIT
# RVSW	111	BIT 9 FLAG 7	RVSWBIT
# R04FLAG	051	BIT 9 FLAG 3	R04FLBIT
# R10FLAG	013	BIT 2 FLAG 0	R10FLBIT
# R61FLAG	020	BIT 10 FLAG 1	R61FLBIT
# R77FLAG	079	BIT 11 FLAG 5	R77FLBIT
# SCALBAD	177	BIT 3 FLAG 11	SCABBIT
# SLOPESW	027	BIT 3 FLAG 1	SLOPEBIT
# SNUFFER	077	BIT 13 FLAG 5	SNUFFBIT
# SOLNSW	087	BIT 3 FLAG 5	SOLNSBIT
# SRCHOPTN	031	BIT 14 FLAG 2	SRCHOBIT
# STATEFLG	055	BIT 5 FLAG 3	STATEBIT
# STEERSW	034	BIT 11 FLAG 2	STEERBIT
# SURFFLAG	127	BIT 8 FLAG 8	SURFFBIT
# SWANDISP	109	BIT 11 FLAG 7	SWANDBIT
# S32.1F1	090	BIT 15 FLAG 6	S32BIT1
# S32.1F2	091	BIT 14 FLAG 6	S32BIT2
# S32.1F3A	092	BIT 13 FLAG 6	S32BIT3A
# S32.1F3B	093	BIT 12 FLAG 6	S32BIT3B
# TFFSW	119	BIT 1 FLAG 7	TFFSWBIT
# TRACKFLG	025	BIT 5 FLAG 1	TRACKBIT
# TURNONFL	194	BIT 1 FLAG 12	TURNONBT
# ULLAGFLG	204	BIT 6 FLAG 13	ULLAGER
# UPDATFLG	023	BIT 7 FLAG 1	UPDATBIT
# UPLOCKFL	116	BIT 4 FLAG 7	UPLOCBIT
# USEQRFLG	196	BIT 14 FLAG 13	USEQRJTS
# VEHUPFLG	022	BIT 8 FLAG 1	VEHUPBIT
# VELDATA	173	BIT 7 FLAG 11	VELDABIT

EQUIVALENT FLAG NAME FC

EQUIVALENT FLAG NAME:

# VERIFLAG	117	BIT 3 FLAG 7	VERIFBIT
# VFLAG	050	BIT 10 FLAG 3	VFLAGBIT
# VFLSHFLG	178	BIT 2 FLAG 11	VFLSHBIT
# VINTFLAG	057	BIT 3 FLAG 3	VINTFBIT
# VXINH	168	BIT 12 FLAG 11	VXINHBIT

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(31 733)

# V37FLAG	114	BIT 6 FLAG 7	V37FLBIT
# V67FLAG	112	BIT 8 FLAG 7	V67FLBIT
# V82EMFLG	118	BIT 2 FLAG 7	V82EMBIT
# XDELVFLG	037	BIT 8 FLAG 2	XDELVBIT
# XDSPFLAG	074	BIT 1 FLAG 4	XDSPBIT
# XORFLG	171	BIT 9 FLAG 11	XORFLBIT
# XOVINFLG	201	BIT 9 FLAG 13	XOVINHIB
# 3AXISFLG	084	BIT 6 FLAG 5	3AXISBIT
# 360SW	134	BIT 1 FLAG 8	360SWBIT

## # ASSIGNMENT AND DESCRIPTION OF FLAGWORDS

FLAGWRDO	=	STATE +0	# (000-014)	
			# (SET)	(RESET)
# BIT 15 FLAG 0 (S)				
	=	000D		
	=	BIT15		
# BIT 14 FLAG 0 (S)				
JSWITCH	=	001D	# INTEGRATION OF W	INTEGRATION OF
JSWCHBIT	=	BIT14	# MATRIX	VECTOR
# BIT 13 FLAG 0 (S)				
MIDFLAG	=	002D	# INTEGRATION WITH	INTEGRATION WITH
			# SECONDARY BODY AND	SOLAR PERTURBAT
MIDFLBIT	=	BIT13	# SOLAR PERTURBATIONS	
# BIT 12 FLAG 0 (L)				
MOONFLAG	=	003D	# MOON IS SPHERE OF	EARTH IS SPHERE
MOONBIT	=	BIT12	# INFLUENCE	INFLUENCE
# BIT 11 FLAG 0				
P21FLAG	=	004D	# USE BASE VECTORS	1ST PASS -- CAL
P21FLBIT	=	BIT11	# ALREADY CALCULATED	ULATE BASE VECT
# BIT 10 FLAG 0				
FSPASFLG	=	005D	# FIRST PASS THROUGH	NOT FIRST PASS
FSPASBIT	=	BIT10	# REPOSITION ROUTINE	REPOSITION ROUT

40 (Page LM0066 40)≡

(31 733)

# BIT 9 FLAG 0 (S)

P25FLAG = 006D

# P25 OPERATING

P25 M

P25FLBIT = BIT9

# BIT 8 FLAG 0 (S)

IMUSE = 007D

# IMU IN USE

IMU M

IMUSEBIT = BIT8

# BIT 7 FLAG 0 (S)

RNDVZFLG = 008D

# P20 RUNNING (RADAR

P20 M

RNDVZBIT = BIT7

# IN USE)

# BIT 6 FLAG 0 (S)

RRNBSW = 009D

# RADAR TARGET IN

RADAR

RRNBBIT = BIT6

# NB COORDINATES

SM CO

# BIT 5 FLAG 0 (S)

LOKONSW = 010D

# RADAR LOCK-ON

RADAR

LOKONBIT = BIT5

# DESIRED

DESI

# BIT 4 FLAG 0 (S)

NEEDLFLG = 011D

# TOTAL ATTITUDE

A/P R

NEEDLBIT = BIT4

# ERROR DISPLAYED

ERROR

# BIT 3 FLAG 0

FREEFLAG = 012D

# (USED BY P51-53 TEMP IN MANY DIFFER

FREEFBIT = BIT3

# ROUTINES &amp; BY LUNAR + SOLAR EPHEMER

# BIT 2 FLAG 0

R10FLAG = 013D

# R10 OUTPUTS DATA TO

BESID

R10FLBIT = BIT2

# ALTITUDE &amp; ALTITUDE

SET,

# RATE METERS ONLY

TO F

#

VELO

# BIT 1 FLAG 0 (L)

OLDESFLG = 014D

# R29 GYRO CMD LOOP

R29 C

OLDESBIT = BIT1

# REQUESTED

NOT R

FLAGWRD1 = STATE +1

# (015-029)



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				#	(31 733) (SET)	(RESET)
# BIT 15 FLAG 1 (S)						
NJETSFLG	=	015D		#	TWO JET RCS BURN	FOUR JET RCS BU
NJETSBIT	=	BIT15				
# BIT 14 FLAG 1 (L)						
DIDFLAG	=	016D		#	INERTIAL DATA IS	PERFORM DATA DI
DIDFLBIT	=	BIT14		#	AVAILABLE	INITIALIZATION
# BIT 13 FLAG 1 (S)						
ERADFLAG	=	017D		#	COMPUTE REARTH	USE CONSTANT RE
ERADFBIT	=	BIT13		#	FISCHER ELLIPSOID	PAD RADIUS
# BIT 12 FLAG 1						
RODFLAG	=	018D		#	IF IN P66, NORMAL	IF IN P66, RE-I
RODFLBIT	=	BIT12		#	OPERATION CONTINUES.	IALIZATION IS P
				#	RESTART CLEARS FLAG	FORMED AND FLAG
# BIT 11 FLAG 1						
	=	019D				
	=	BIT11				
# BIT 10 FLAG 1 (L)						
R61FLAG	=	020D		#	RUN R61 LEM	RUN R65 LEM
R61FLBIT	=	BIT10				
# BIT 9 FLAG 1						
	=	021D				
	=	BIT9				
# BIT 8 FLAG 1 (S)						
VEHUPFLG	=	022D		#	CSM STATE-VECTOR	LEM STATE VECTO
VEHUPBIT	=	BIT8		#	BEING UPDATED	BEING UPDATED
# BIT 7 FLAG 1 (S)						
UPDATFLG	=	023D		#	UPDATING BY MARKS	UPDATING BY MAR
UPDATBIT	=	BIT7		#	ALLOWED	NOT ALLOWED
# BIT 6 FLAG 1 (S)						
NOUPFLAG	=	024D		#	NEITHER CSM	EITHER STATE
				#	NOR LM STATE VECTOR	VECTOR MAY BE
NOUPFBIT	=	BIT6		#	MAY BE UPDATED	UPDATED

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(31 733)

# BIT 5 FLAG 1 (S)

TRACKFLG = 025D

TRACKBIT = BIT5

# TRACKING ALLOWED

TRACK

# BIT 4 FLAG 1

= 026D

= BIT4

# BIT 3 FLAG 1 (S)

SLOPESW = 027D

SLOPEBIT = BIT3

# ITERATE WITH BIAS  
# METHOD IN ITERATOR

ITERA

FALS

ITERA

# BIT 2 FLAG 1 (S)

GUESSW = 028D

GUESSBIT = BIT2

# NO STARTING VALUE  
# FOR ITERATION

STAR

ITERA

# BIT 1 FLAG 1

= 029D

= BIT1

# OH 2009-05-15 Scan does not have th

FLAGWRD2 = STATE +2

# (030-044)

# (SET)

(RES

# BIT 15 FLAG 2 (S)

DRIFTFLG = 030D

DRFTBIT = BIT15

# T3RUPT CALLS GYRO  
# COMPENSATION

T3RU

COMPI

# BIT 14 FLAG 2 (S)

SRCHOPTN = 031D

SRCHOBIT = BIT14

# RADAR IN AUTOMATIC  
# SEARCH OPTION (R24)

RADAR

MATIO

# BIT 13 FLAG 2 (S)

ACMODFLG = 032D

ACMODBIT = BIT13

# MANUAL ACQUISITION  
# BY RENDEZVOUS RADAR

AUTO

BY R

# BIT 12 FLAG 2 (S)

LOSCMFLG = 033D

LOSCMBIT = BIT12

# LINE OF SIGHT BEING  
# COMPUTED (R21)

LINE

BEIN

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(31 733)

# BIT 11 FLAG 2 (S)

STEERSW = 034D  
STEERBIT = BIT11

# SUFFICIENT THRUST INSUFFICIENT TH  
# IS PRESENT IS PRESENT

# BIT 10 FLAG 2 (S)

= 035D  
= BIT10

# OH 2009-05-15 These two line don't appear in

# BIT 9 FLAG 2 (S)

IMPULSW = 036D  
IMPULBIT = BIT9

# MINIMUM IMPULSE STEERING BURN (C  
# BURN (CUTOFF TIME CUTOFF TIME YET  
# SPECIFIED) AVAILABLE)

# BIT 8 FLAG 2 (S)

XDELVFLG = 037D  
XDELVBIT = BIT8

# EXTERNAL DELTAV VG LAMBERT (AIMPO  
# COMPUTATION VG COMPUTATION

# BIT 7 FLAG 2 (S)

ETPIFLAG = 038D  
ETPIBIT = BIT7

# ELEVATION ANGLE TPI TIME SUPPLI  
# SUPPLIED FOR FOR P34,74 TO C  
# P34,74 ELEVATION

# BIT 7 FLAG 2 (L)

OPTNSW = ETPIFLAG  
OPTNBIT = BIT7

# SOI PHASE OF P38/78 SOR PHASE OF P3

# BIT 6 FLAG 2 (S)

FINALFLG = 039D  
FINALBIT = BIT6

# LAST PASS THROUGH INTERIM PASS TH  
# RENDEZVOUS PROGRAM RENDEZVOUS PROG  
# COMPUTATIONS COMPUTATIONS

# BIT 5 FLAG 2 (S)

AVFLAG = 040D  
AVFLBIT = BIT5

# LEM IS ACTIVE CSM IS ACTIVE  
# VEHICLE VEHICLE

# BIT 4 FLAG 2 (S)

PFRATFLG = 041D  
PFRATBIT = BIT4

# PREFERRED ATTITUDE PREFERRED ATTIT  
# COMPUTED NOT COMPUTED

# BIT 3 FLAG 2 (S)

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CALCMAN3 = 042D  
 CALC3BIT = BIT3

(31 733)  
 # NO FINAL ROLL FINAL  
 # NECES

# BIT 2 FLAG 2 (S)  
 CALCMAN2 = 043D  
 CALC2BIT = BIT2

# PERFORM MANEUVER BYPAS  
 # STARTING PROCEDURE PROCE

# BIT 1 FLAG 2 (S)  
 NODOFLAG = 044D  
 NODOBIT = BIT1

# V37 NOT PERMITTED V37 P

FLAGWRD3 = STATE +3

# (045-059)

# (SET) (RES)

# BIT 15 FLAG 3  
 = 045D  
 = BIT15

#  
 # OH 2009-05-15 This line is not in s

# BIT 14 FLAG 3 (S)  
 GLOKFAIL = 046D  
 GLOKFBIT = BIT14

# GIMBAL LOCK HAS NOT  
 # OCCURRED

# BIT 13 FLAG 3 \*\*\* PROTECTED FROM FRESH START \*\*\*  
 REFSMFLG = 047D  
 REFSMBIT = BIT13

# REFSMMAT GOOD REFS

# BIT 12 FLAG 3 (S)  
 LUNAFLAG = 048D  
 LUNABIT = BIT12

# LUNAR LAT-LONG EART

# BIT 11 FLAG 3 (L)  
 NOR29FLG = 049D  
 NR29FBIT = BIT11

# R29 NOT ALLOWED R29 A  
 # IGNAT

# BIT 10 FLAG 3 (S)  
 VFLAG = 050D  
 VFLAGBIT = BIT10

# LESS THAN TWO STARS TWO S  
 # IN FIELD OF VIEW OF V

# BIT 9 FLAG 3 (S)  
 R04FLAG = 051D

# ALARM 521 ALAR  
 # SUPPRESSED

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*(Page LM0071 45)*≡

(31 733)

R04FLBIT = BIT9

# BIT 9 FLAG 3 (L)

READRFLG = R04FLAG

READRBIT = BIT9

# READING RR DATA NOT READING RR  
# PURSUANT TO R29 PURSUANT TO R29

# BIT 8 FLAG 3 (S)

PRECIFLG = 052D

PRECIBIT = BIT8

# NORMAL INTEGRATION ENGAGES 4-TIME  
# IN P00 (P00) LOGIC IN  
# GRATION

# BIT 7 FLAG 3 (S)

CULTFLAG = 053D

CULTBIT = BIT7

# STAR OCCULTED STAR NOT OCCULTED

# BIT 6 FLAG 3 (S)

ORBWFLAG = 054D

ORBWFBIT = BIT6

# W MATRIX VALID FOR W MATRIX INVALID  
# ORBITAL NAVIGATION ORBITAL NAVIGATION

# BIT 5 FLAG 3 (S)

STATEFLG = 055D

STATEBIT = BIT5

# PERMANENT STATE PERMANENT STATE  
# VECTOR UPDATED VECTOR NOT UPDATED

# BIT 4 FLAG 3 (S)

INTYPFLG = 056D

INTYPBIT = BIT4

# CONIC INTEGRATION ENCKE INTEGRATION

# BIT 3 FLAG 3 (S)

VINTFLAG = 057D

VINTFBIT = BIT3

# CSM STATE VECTOR LEM STATE VECTOR  
# BEING INTEGRATED BEING INTEGRATED

# BIT 2 FLAG 3 (S)

D6OR9FLG = 058D

D6OR9BIT = BIT2

# DIMENSION OF W IS 9 DIMENSION OF W  
# FOR INTEGRATION FOR INTEGRATION

# BIT 1 FLAG 3 (S)

DIM0FLAG = 059D

DIM0BIT = BIT1

# W MATRIX IS TO BE W MATRIX IS NOT  
# USED USED

FLAGWRD4 = STATE +4

# (060-074)

46 (Page LM0072 46)≡

Page LM0072 46)≡			#	(31 733) (SET)	(RES)
# BIT 15 FLAG 4 (S)					
MRKIDFLG	=	060D	#	MARK DISPLAY IN	NO MA
MRKIDBIT	=	BIT15	#	ENDIDLE	ENDID
# BIT 14 FLAG 4 (S)					
PRIODFLG	=	061D	#	PRIORITY DISPLAY IN	NO PR
PRIODBIT	=	BIT14	#	ENDIDLE	IN ED
# BIT 13 FLAG 4 (S)					
NRMIDFLG	=	062D	#	NORMAL DISPLAY IN	NO NO
NRMIDBIT	=	BIT13	#	ENDIDLE	IN ED
# BIT 12 FLAG 4 (S)					
PDSPFLAG	=	063D	#	P20 SETS SO AS TO	LEAVE
			#	TURN A NORMAL DIS-	
PDSPFBIT	=	BIT12	#	PLAY INTO A PRIORITY	
			#	DISPLAY IN R60	
# BIT 11 FLAG 4 (S)					
MWAITFLG	=	064D	#	HIGHER PRIORITY	NO H
			#	DISPLAY OPERATING	DISPI
MWAITBIT	=	BIT11	#	WHEN MARK	WHEN
			#	DISPLAY INITIATED	INITI
# BIT 10 FLAG 4 (S)					
NWAITFLG	=	065D	#	HIGHER PRIORITY	NO H
			#	DISPLAY OPERATING	DISPI
NWAITBIT	=	BIT10	#	WHEN NORMAL	WHEN
			#	DISPLAY INITIATED	INITI
# BIT 9 FLAG 4 (S)					
MRKNVFLG	=	066D	#	ASTRONAUT USING	ASTRO
			#	KEYBOARD WHEN MARK	KEYBO
MRKNVBIT	=	BIT9	#	DISPLAY INITIATED	DISPI
# BIT 8 FLAG 4 (S)					
NRMNVFLG	=	067D	#	ASTRONAUT USING	ASTRO
			#	KEYBOARD WHEN	KEYBO
NRMNVBIT	=	BIT8	#	NORMAL DISPLAY	NORMA
			#	INITIATED	INITI
# BIT 7 FLAG 4 (S)					
PRONVFLG	=	068D	#	ASTRONAUT USING	ASTRO

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PRONVBIT = BIT7

(31 733)  
# KEYBOARD WHEN  
# PRIORITY DISPLAY  
# INITIATED

# BIT 6 FLAG 4 (S)  
PINBRFLG = 069D

# ASTRONAUT HAS  
# INTERFERED WITH  
# EXISTING DISPLAY

PINBRBIT = BIT6

# BIT 5 FLAG 4 (S)  
MRUPTFLG = 070D

# MARK DISPLAY  
# INTERRUPTED BY  
# PRIORITY DISPLAY

MRUPTBIT = BIT5

# BIT 4 FLAG 4 (S)  
NRUPTFLG = 071D

# NORMAL DISPLAY  
# INTERRUPTED BY  
# PRIORITY OR MARK  
# DISPLAY

NRUPTBIT = BIT4

# BIT 3 FLAG 4 (S)  
MKOVFLAG = 072D  
MKOVBIT = BIT3

# MARK DISPLAY OVER  
# NORMAL

# BIT 2 FLAG 4  
= 073D  
= BIT2

# OH 2009-05-15 Not in scan.

# BIT 1 FLAG 4 (S)  
XDSPFLAG = 074D  
XDSPBIT = BIT1

# MARK DISPLAY NOT  
# TO BE INTERRUPTED

FLAGWRD5 = STATE +5

# (075-089)

# (SET) (RES)

# BIT 15 FLAG 5 (S)  
DSKYFLAG = 075D  
DSKYFBIT = BIT15

# DISPLAYS SENT TO  
# DSKY

# BIT 14 FLAG 5  
= 076D  
= BIT14

NO D



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(31 733)

# BIT 13 FLAG 5 (S,L)

SNUFFER = 077D

# U,V JETS DISABLED

U,V JETS ENABLED

# DURING DPS

DURING DPS

SNUFFBIT = BIT13

# BURNS (V65)

BURNS (V75)

# BIT 12 FLAG 5 (S)

NOTHROTL = 078D

# INHIBIT FULL

PERMIT FULL THR

NOTHRBIT = BIT12

# THROTTLE

# BIT 11 FLAG 5 (S,L)

R77FLAG = 079D

# R77 IS ON,

R77 IS NOT ON.

# SUPPRESS ALL RADAR

# ALARMS AND TRACKER

R77FLBIT = BIT11

# FAILS

# BIT 10 FLAG 5 (S)

RNGSCFLG = 080D

# SCALE CHANGE HAS

NO SCALE CHANGE

# OCCURRED DURING

OCCURRED DURING

RNGSCBIT = BIT10

# RR READING

RR READING

# BIT 9 FLAG 5 (S)

DMENFLG = 081D

# DIMENSION OF W IS 9

DIMENSION OF W

DMENFBIT = BIT9

# FOR INCORPORATION

FOR INCORPORATI

# BIT 8 FLAG 5 (S)

= 082D

= BIT8

# BIT 7 FLAG 5 (S)

ENGONFLG = 083D

# ENGINE TURNED ON

ENGINE TURNED C

ENGONBIT = BIT7

#

# BIT 6 FLAG 5 (S)

3AXISFLG = 084D

# MANEUVER SPECIFIED

MANEUVER SPECIF

# BY THREE AXES

BY ONE AXIS; R6

3AXISBIT = BIT6

#

CALLS VECPOINT.

# BIT 5 FLAG 5

= 085D

= BIT5

# OH 2009-05-15 Not in scan

# BIT 4 FLAG 5 (S)

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NORRMON = 086D  
 NORRMBIT = BIT4

(31 733)  
 # BYPASS RR GIMBAL PERFO  
 # MONITOR RR G

# BIT 3 FLAG 5 (S)  
 SOLNSW = 087D  
 SOLNSBIT = BIT3

# LAMBERT DOES NOT LAMBE  
 # CONVERGE, OR TIME-RAD TIME-  
 # NEARLY CIRCULAR CIRC

# BIT 2 FLAG 5 (S)  
 MGLVFLAG = 088D  
 MGLVFBIT = BIT2

# LOCAL VERTICAL MIDD  
 # COORDINATES COMPU  
 # COMPUTED

# BIT 1 FLAG 5 (S)  
 RENWFLG = 089D  
 RENWBIT = BIT1

# W MATRIX VALID W MA  
 # FOR RENDEZVOUS FOR R  
 # NAVIGATION NAVIO

FLAGWRD6 = STATE +6

# (090-104)  
 # (SET) (RESI

# BIT 15 FLAG 6 (S)  
 S32.1F1 = 090D  
 S32BIT1 = BIT15

# DELTA V AT CSI TIME DVT1  
 # ONE EXCEEDS MAX

# BIT 14 FLAG 6 (S)  
 S32.1F2 = 091D  
 S32BIT2 = BIT14

# FIRST PASS OF REITI  
 # NEWTON ITERATION NEWTO

# BIT 13 FLAG 6 (S)  
 S32.1F3A = 092D  
 S32BIT3A = BIT13

# BIT 13 AND BIT 12 FUNCTION AS AN OR  
 # PAIR (13,12) INDICATING THE POSSIBL  
 # CURRENCE OF 2 NEWTON ITERATIONS FOR  
 # IN THE PROGRAM IN THE FOLLOWING OR  
 # (0,1) (I.E. BIT 13 RESET, BIT 12 S  
 # = FIRST NEWTON ITERATION BEING  
 # (0,0)= FIRST PASS OF SECOND NEWTON  
 # (1,1)= 50 FT/SEC STAGE OF SECOND NE  
 # (1,0)= REMAINDER OF SECOND NEWTON 1

# BIT 12 FLAG 6 (S)  
 S32.1F3B = 093D  
 S32BIT3B = BIT12

# BIT 11 FLAG 6 (S)  
 = 094D  
 = BIT11

#  
 #

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51 (Page LM0076 51)≡ (31 733)

# BIT 10 FLAG 6 (S)				
GMBDRVSW	=	095D	#	TRIMGIMB OVER
GMBDRBIT	=	BIT10	#	TRIMGIMB NOT OV
# BIT 9 FLAG 6				
	=	096D	#	
	=	BIT9	#	
# BIT 8 FLAG 6 (S)				
MUNFLAG	=	097D	#	SERVICER CALLS
MUNFLBIT	=	BIT8	#	MUNRVG
				SERVICER CALLS
				CALCRVG
# BIT 7 FLAG 6 (L)				
	=	098D	#	
	=	BIT7	#	
# BIT 6 FLAG 6 (L)				
REDFLAG	=	099D	#	LANDING SITE
			#	REDESIGNATION
REDFLBIT	=	BIT6	#	PERMITTED
				LANDING SITE
				REDESIGNATION M
				PERMITTED
# BIT 5 FLAG 6				
	=	100D	#	
	=	BIT5	#	OH 2009-05-15 Not in scan
# BIT 4 FLAG 6				
	=	101D	#	
	=	BIT4	#	OH 2009-05-15 Not in scan
# BIT 3 FLAG 6 (S)				
NTARGFLG	=	102D	#	ASTRONAUT DID
			#	OVERWRITE DELTA
NTARGBIT	=	BIT3	#	VELOCITY AT TPI
			#	OR TPM (P34,35)
				ASTRONAUT DID M
				OVERWRITE DELTA
				VELOCITY
# BIT 2 FLAG 6				
AUXFLAG	=	103D	#	PROVIDING IDLEFLAG
AUXFLBIT	=	BIT2	#	IS NOT SET, SERV-
			#	ICER WILL EXERCISE
			#	DVMON ON ITS NEXT
			#	PASS.
			#	SERVICER WILL S
				DVMON ON ITS NE
				PASS EVEN IF TH
				IDLEFLAG IS NOT
				IT WILL THEN SE
				AUXFLAG.
# BIT 1 FLAG 6 (L)				
ATTFLAG	=	104D	#	LEM ATTITUDE EXISTS
				NO LEM ATTITUDE

			#	IN MOON-FIXED	AVAIL	
52	$\langle \text{Page LM0077 52} \rangle \equiv$			(31 733)		
	ATTFLBIT	=	BIT1	#	COORDINATES	FIXED
	FLAGWRD7	=	STATE +7	#	(105-119)	
			#	(SET)	(RES)	
	# BIT 15 FLAG 7 (S)					
	ITSWICH	=	105D	#	R34;TPI TIME TO BE	TPI R
	ITSWBIT	=	BIT15	#	COMPUTED	COMPU
	# BIT 14 FLAG 7 (S)					
	MANUFLAG	=	106D	#	ATTITUDE MANEUVER	NO A
				#	GOING DURING RR	DURIN
	MANUFBIT	=	BIT14	#	SEARCH	
	# BIT 13 FLAG 7 (S)					
	IGNFLAG	=	107D	#	TIG HAS ARRIVED	TIG R
	IGNFLBIT	=	BIT13	#		
	# BIT 12 FLAG 7 (S)					
	ASTNFLAG	=	108D	#	ASTRONAUT HAS	ASTRO
	ASTNBIT	=	BIT12	#	OKAYED IGNITION	OKAYE
	# BIT 11 FLAG 7 (L)					
	SWANDISP	=	109D	#	LANDING ANALOG	LAND
	SWANDBIT	=	BIT11	#	DISPLAYS ENABLED	DISPI
	# BIT 10 FLAG 7 (S)					
	NORMSW	=	110D	#	UNIT NORMAL INPUT	LAMBE
	NORMSBIT	=	BIT10	#	TO LAMBERT	OWN U
	# BIT 9 FLAG 7 (S)					
	RVSW	=	111D	#	DO NOT COMPUTE	COMPU
				#	FINAL STATE VECTOR	VECTO
	RVSWBIT	=	BIT9	#	IN TIME-DELTA	
	# BIT 8 FLAG 7 (S)					
	V67FLAG	=	112D	#	ASTRONAUT OVERWRITE	ASTRO
				#	W-MATRIX INITIAL	OVERW
	V67FLBIT	=	BIT8	#	VALUES	INIT

53 (Page LM0078 53)≡

(31 733)

# BIT 7 FLAG 7 (S)

IDLEFLAG = 113D

IDLEFBIT = BIT7

# NO DV MONITOR

# CONNECT DV MONI

# BIT 6 FLAG 7 (S)

V37FLAG = 114D

V37FLBIT = BIT6

# AVERAGEG (SERVICER)

# RUNNING AVERAGEG (SERVI

# BIT 5 FLAG 7 (S)

AVEGFLAG = 115D

AVEGFBIT = BIT5

# AVERAGEG (SERVICER)

# DESIRED AVERAGEG (SERVI

# BIT 4 FLAG 7 (S)

UPLOCKFL = 116D

UPLOCBIT = BIT4

# K-KBAR-K FAIL

# NO K-KBAR-K FAI

# BIT 3 FLAG 7 (S)

VERIFLAG = 117D

VERIFBIT = BIT3

# CHANGED WHEN V33E OCCURS AT END OF P27

#

# BIT 2 FLAG 7 (L,C)

V82EMFLG = 118D

V82EMBIT = BIT2

# MOON VICINITY

# EARTH VICINITY

# BIT 1 FLAG 7 (S)

TFFSW = 119D

TFFSWBIT = BIT1

# CALCULATE TPERIGEE

# CALCULATE TFF

FLAGWRD8 = STATE +8D

# (120-134)

# (SET)

(RESET)

# BIT 15 FLAG 8 (S)

RPQFLAG = 120D

RPQFLBIT = BIT15

# RPQ NOT COMPUTED

# (RPQ = VECTOR BE-

# TWEEN SECONDARY BODY

# AND PRIMARY BODY)

RPQ COMPUTED

# BIT 14 FLAG 8

= 121D

= BIT14

#

#

54 (*Page LM0079* 54)≡

(31 733)

# BIT 13 FLAG 8 (S)

NEWIFLG = 122D  
 NEWIBIT = BIT13

# FIRST PASS THROUGH  
 # INTEGRATION OF ID

# BIT 12 FLAG 8 \*\*\* PROTECTED FROM FRESH START \*\*\*

CMOONFLG = 123D  
 CMOONBIT = BIT12

# PERMANENT CSM STATE  
 # IN LUNAR SPHERE PERMA  
 IN EA

# BIT 11 FLAG 8 \*\*\* PROTECTED FROM FRESH START \*\*\*

LMOONFLG = 124D  
 LMOONBIT = BIT11

# PERMANENT LM STATE  
 # IN LUNAR SPHERE PERMA  
 IN EA

# BIT 10 FLAG 8 (L)

FLUNDISP = 125D  
 FLUNDBIT = BIT10

# CURRENT GUIDANCE  
 # DISPLAYS INHIBITED CURR  
 DISPI

# BIT 9 FLAG 8 (L)

P39/79SW = 126D  
 P39SWBIT = BIT9

# P39/79 OPERATING P38/7  
 #

# BIT 8 FLAG 8 \*\*\* PROTECTED FROM FRESH START \*\*\*

SURFFLAG = 127D  
 SURFFBIT = BIT8

# LM ON LUNAR SURFACE LM NO  
 # SURFA

# BIT 7 FLAG 8 (S)

INFINFLG = 128D  
 INFINBIT = BIT7

# NO CONIC SOLUTION  
 # (CLOSURE THROUGH  
 # INFINITY REQUIRED) CONIC  
 EXIS

# BIT 6 FLAG 8 (S)

ORDERSW = 129D  
 ORDERBIT = BIT6

# ITERATOR USES 2ND  
 # ORDER MINIMUM MODE ITERA  
 ORDER

# BIT 5 FLAG 8 (S)

APSESW = 130D  
 APSESBIT = BIT5

# RDESIRED OUTSIDE  
 # PERICENTER-APOCENTER  
 # RANGE IN TIME-RADIUS RDES  
 PERIO  
 RANG

# BIT 4 FLAG 8 (S)

COGAFLAG = 131D

# NO CONIC SOLUTION --  
 # TOO CLOSE TO RECTI- CONIC  
 EXIS

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```
55  (Page LM0080 55)≡
    COGAFBIT      =      BIT4      #      (31 733)
                                LINEAR (COGA OVERFLWS)  OVERFLOW)

    # BIT 3 FLAG 8 (S)
    =      132D      #
    =      BIT3      # OH 2009-05-15 Line not in scan

    # BIT 2 FLAG 8 (L)
    INITIALGN      =      133D      #      INITIAL PASS THRU      SECOND PASS THRU
    INITABIT        =      BIT2      #      P57                      (CHECK RESET-MI

    # BIT 1 FLAG 8 (S)
    360SW           =      134D      #      TRANSFER ANGLE NEAR      TRANSFER ANGLE
    360SWBIT         =      BIT1      #      360 DEGREES              NEAR 360 DEGREE

    FLAGWRD9        =      STATE +9D  # (135-149)
                                #      (SET)                      (RESET)

    # BIT 15 FLAG 9
    =      135D      #
    =      BIT15     #

    # BIT 14 FLAG 9 (L)
    FLVR            =      136D      #      VERTICAL RISE      NON-VERTICAL RI
    FLVRBIT         =      BIT14     #      (ASCENT GUIDANCE)

    # BIT 13 FLAG 9
    =      137D      #
    =      BIT13     # OH 2009-05-15 Line not in scan

    # BIT 12 FLAG 9 (L)
    FLPC            =      138D      #      NO POSITION CONTROL      POSITION CONTR
    FLPCBIT         =      BIT12     #      (ASCENT GUIDANCE)

    # BIT 11 FLAG 9 (L)
    FLPI            =      139D      #      PRE-IGNITION PHASE      REGULAR GUIDANC
    FLPIBIT         =      BIT11     #      (ASCENT GUIDANCE)

    # BIT 10 FLAG 9 (L)
    FLRCS           =      140D      #      RCS INJECTION MODE      MAIN ENGINE MOD
    FLRCSBIT        =      BIT10     #      (ASCENT GUIDANCE)

    # BIT 9 FLAG 9 (L)
```

56 (Page LM0081 56)≡

LETABORT = 141D  
 LETABBIT = BIT9

(31 733)  
 # ABORT PROGRAMS ABORT  
 # ARE ENABLED ARE M

# BIT 8 FLAG 9 (L)  
 FLAP = 142D

# APS CONTINUED ABORT APS A  
 # AFTER DPS STAGING CONT  
 # (ASCENT GUIDANCE)

FLAPBIT = BIT8

# BIT 7 FLAG 9 (L)  
 = 143D  
 = BIT7

# OH 2009-05-15 Line not in scan

# BIT 6 FLAG 9 (L)  
 ROTFLAG = 144D  
 ROTFLBIT = BIT6

# P70 AND P71 WILL P70 A  
 # FORCE VEHICLE FORC  
 # ROTATION IN THE ROTAT  
 # PREFERRED DIRECTION PREFE

# BIT 5 FLAG 9 (S)  
 QUITFLAG = 145D  
 QUITBIT = BIT5

# DISCONTINUE INTEGR. CONT  
 #

# BIT 4 FLAG 9  
 = 146D  
 = BIT4

#  
 #

# BIT 3 FLAG 9 (L)  
 MID1FLAG = 147D  
 MID1FBIT = BIT3

# INTEGRAT TO TDEC INTE  
 # THEN-

# BIT 2 FLAG 9 (L)  
 MIDAVFLG = 148D  
 MIDAVBIT = BIT2

# INTEGRATION ENTERED INTE  
 # FROM ONE OF MIDTOAV NOT P  
 # PORTALS MIDTO

# BIT 1 FLAG 9 (S)  
 AVEMIDSW = 149D  
 AVEMDBIT = BIT1

# AVETOMID CALLING NO AV  
 # FOR W.MATRIX INTEGR ALLOW  
 # DON'T WRITE OVER RN, PIPT  
 # VN,PIPTIME

RASFLAG EQUALS FLGWRD10

# WAS ONLY AN INSTALL-ERASTALL FLAG



```

57  (Page LM0082 57)≡
    FLGWRD10      =      STATE +10D      # (150-164)      (31 733)

                                                    #      (SET)      (RESET)

# BIT 15 FLAG 10 (S)
      =      150D      #
      =      BIT15      # OH 2009-05-15 Line not in scan

# BIT 14 FLAG 10 (L,C)
INTFLAG      =      151D      #      INTEGRATION IN      INTEGRATION NOT
INTFLBIT      =      BIT14      #      PROGRESS      PROGRESS

# BIT 13 FLAG 10 (S,L)
APSFLAG      =      152D      #      ASCENT STAGE      DESCENT STAGE
APSFLBIT      =      BIT13      #      *** PROTECTED FROM FRESH START ***

# BIT 12 FLAG 10
      =      153D      #
      =      BIT12      # OH 2009-05-15 Line not in scan

# BIT 11 FLAG 10
      =      154D      #
      =      BIT11      # OH 2009-05-15 Line not in scan

# BIT 10 FLAG 10
      =      155D      #
      =      BIT10      # OH 2009-05-15 Line not in scan

# BIT 9 FLAG 10
      =      156D      #
      =      BIT9      # OH 2009-05-15 Line not in scan

# BIT 8 FLAG 10
      =      157D      #
      =      BIT8      # OH 2009-05-15 Line not in scan

# BIT 7 FLAG 10 (L,C)
REINTFLG      =      158D      #      INTEGRATION ROUTINE      INTEGRATION ROU
REINTBIT      =      BIT7      #      TO BE RESTARTED      NOT TO BE RESTA

# BIT 6 FLAG 10
      =      159D      #
      =      BIT6      # OH 2009-05-15 Line not in scan

# BIT 5 FLAG 10

```

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= 160D  
= BIT5

#  
# OH 2009-05-15 Line not in scan

59

*(Page LM0083 59)*≡

(31 733)

# BIT 4 FLAG 10

= 161D  
 = BIT4

#  
 # OH 2009-05-15 Line not in scan

# BIT 3 FLAG 10

= 162D  
 = BIT3

#  
 # OH 2009-05-15 Line not in scan

# BIT 2 FLAG 10

= 163D  
 = BIT2

#  
 # OH 2009-05-15 Line not in scan

# BIT 1 FLAG 10

= 164D  
 = BIT1

#  
 # OH 2009-05-15 Line not in scan

FLGWRD11 = STATE +11D

# (165-179)

# (SET) (RESET)

# BIT 15 FLAG 11 (L)(R12)

LRBYPASS = 165D  
 LRBYBIT = BIT15

# BYPASS ALL LANDING DO NOT BYPASS L  
 # RADAR UPDATES UPDATES

# BIT 14 FLAG 11

= 166D  
 = BIT14

#  
 #

# BIT 13 FLAG 11

= 167D  
 = BIT13

#  
 #

# BIT 12 FLAG 11 (L)(R12)

VXINH = 168D  
 VXINHBIT = BIT12

# IF Z VELOCITY DATA UPDATE X AXIS  
 # UNREASONABLE, VELOCITY  
 # BYPASS X VELOCITY  
 # UPDATE ON NEXT PASS

# BIT 11 FLAG 11 (L)(R12)

PSTHIGAT = 169D  
 PSTHIBIT = BIT11

# PAST HIGATE PREHIGATE  
 #

# BIT 10 FLAG 11 (L)(R12)

60  $\langle$ Page LM0084 60 $\rangle \equiv$  (31 733)

NOLRREAD	=	170D	#	LANDING RADAR	LR NO
			#	REPOSITIONING;	
NOLRRBIT	=	BIT10	#	BYPASS UPDATE	
# BIT 9 FLAG 11 (L)(R12)					
XORFLG	=	171D	#	BELOW LIMIT	ABOVE
			#	INHIBIT X AXIS	NOT 1
XORFLBIT	=	BIT9	#	OVERRIDE	
# BIT 8 FLAG 11					
LRINH	=	172D	#	LANDING RADAR UP-	LR UP
LRINHBIT	=	BIT8	#	DATES PERMITTED	BY AS
			#	BY ASTRONAUT	
# BIT 7 FLAG 11 (L)(R12)					
VELDATA	=	173D	#	LR VELOCITY	LR VE
VELDABIT	=	BIT7	#	MEASUREMENT MADE	NOT M
# BIT 6 FLAG 11 (L)(R12)					
READLR	=	174D	#	OK TO READ LR	DO NO
READLBIT	=	BIT6	#	RANGE DATA	DATA
# BIT 5 FLAG 11 (L)(R12)					
READVEL	=	175D	#	OK TO READ LR	DO NO
READVBIT	=	BIT5	#	VELOCITY DATA	VELO
# BIT 4 FLAG 11 (L)(R12)					
RNGEDATA	=	176D	#	LR ALTITUDE	LR AI
RNGEDBIT	=	BIT4	#	MEASUREMENT MADE	NOT M
# BIT 3 FLAG 11					
SCALBAD	=	177D	#	LR LOW SCALE DISP-	LS SO
SCABBIT	=	BIT3	#	CREATE NOT PRESENT	APPEA
			#	WHEN IT SHOULD	
# BIT 2 FLAG 11 (L)(R12)					
VFLSHFLG	=	178D	#	LR VELOCITY FAIL	LR VE
			#	LAMP SHOULD BE	SHOU
VFLSHBIT	=	BIT2	#	FLASHING	
# BIT 1 FLAG 11 (L)(R12)					

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61 (Page LM0085 61)≡ (31 733)

HFLSHFLG	=	179D	#	LR ALTITUDE FAIL	LR ALTITUDE FAI
HFLSHBIT	=	BIT1	#	LAMP SHOULD BE	LAMP SHOULD NOT
			#	FLASHING	FLASHING
RADMODES	EQUALS	FLGWRD12	#	RADAR FLAG WORD	
FLGWRD12	=	STATE +12D	#	(180-194)	WAS RADMODES
			#	(SET)	(RESET)
# BIT 15 FLAG 12					
CDESFLAG	=	180D	#	CONTINUOUS DESIG-	LGC CHECKS FOR
CDESBIT	=	BIT15	#	NATE, LGC COMMANDS	ON WHEN ANTENNA
			#	RR REGARDLESS OF	BEING DESIGNATE
			#	LOCK-ON	
# BIT 14 FLAG 12					
REMODFLG	=	181D	#	CHANGE IN ANTENNA	NO REMODE REQUE
REMODBIT	=	BIT14	#	MODE BEEN REQUESTED	OR OCCURRING
			#	I.E., REMODE	
# BIT 13 FLAG 12					
RCDUOFLG	=	182D	#	RR CDU'S BEING	RR CDU'S NOT BE
RCDUOBIT	=	BIT13	#	ZEROED	ZEROED
# BIT 12 FLAG 12					
ANTENFLG	=	183D	#	RR ANTENNA MODE IS	RR ANTENNA IN M
ANTENBIT	=	BIT12	#	MODE 2	
# BIT 11 FLAG 12					
REPOSMON	=	184D	#	REPOSITION MONITOR.	NO REPOSITION T
REPOSBIT	=	BIT11	#	RR REPOSITION IS	PLACE
			#	TAKING PLACE	
# BIT 10 FLAG 12					
DESIGFLG	=	185D	#	RR DESIGNATE	RR DESIGNATE NO
DESIGBIT	=	BIT10	#	REQUESTED OR IN	REQUESTED OR IN
			#	PROGRESS	PROGRESS
# BIT 9 FLAG 12					
ALTSCALE	=	186D	#	LR ALTITUDE READING	LR ALTITUDE REA
ALTSCBIT	=	BIT9	#	IS ON HIGH SCALE	IS ON LOW SCALE

62 (Page LM0086 62)≡

(31 733)

# BIT 8 FLAG 12

LRVELFLG = 187D

LRVELBIT = BIT8

# LR VELOCITY DATA NO L  
# FAIL FAIL

# BIT 7 FLAG 12

RCDUFAIL = 188D

RCDUFBIT = BIT7

# RR CDU FAIL HAS RR C  
# NOT OCCURRED

# BIT 6 FLAG 12

LRPOSFLG = 189D

LRPOSBIT = BIT6

# LANDING RADAR LR P  
# POSITION 2

# BIT 5 FLAG 12

LRALTFLG = 190D

LRALTBIT = BIT5

# LR ALTITUDE DATA NO L  
# FAIL. COULD NOT BE FAIL  
# READ SUCCESSFULLY.

# BIT 4 FLAG 12

RRDATAFL = 191D

RRDATABT = BIT4

# RR DATA FAIL. NO R  
# DATA COULD NOT BE  
# READ SUCCESSFULLY

# BIT 3 FLAG 12

RRRSFLAG = 192D

RRRSBIT = BIT3

# RR RANGE READING RR R  
# ON THE HIGH SCALE THE I

# BIT 2 FLAG 12

AUTOMODE = 193D

AUTOMBIT = BIT2

# RR NOT IN AUTO MODE. RR I  
# AUTO MODE DISCRETE  
# IS NOT PRESENT

# BIT 1 FLAG 12

TURNONFL = 194D

TURNONBT = BIT1

# RR TURN-ON SEQUENCE NO R  
# IN PROGRESS. (ZERO SEQU  
# CDU'S, FIX ANTENNA  
# MODE)

DAPBOOLS EQUALS FLGWRD13

# DIGITAL AUTOPILOT FLAGWORD

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63 (Page LM0087 63)≡

FLGWRD13	=	STATE +13D	# (195-209)	(31 733) WAS DAPBOOLS
			#	(SET) (RESET)
# BIT 15 FLAG 13				
PULSEFLG	=	195D	#	MINIMUM IMPUSE NOT IN MINIMUM
PULSES	=	BIT15	#	COMMAND MODE IN IMPULSE COMMAND
			#	"ATT HOLD" (V76) (V77)
# BIT 14 FLAG 13				
USEQRFLG	=	196D	#	GIMBAL UNUSABLE. TRIM GIMBAL MAY
USEQRJTS	=	BIT14	#	USE JETS ONLY. USED.
# BIT 13 FLAG 13				
CSMDKFLG	=	197D	#	CSM DOCKED. USE CSM NOT DOCKED
CSMDOCKD	=	BIT13	#	BACKUP DAP
# BIT 12 FLAG 13				
OURRCFLG	=	198D	#	CURRENT DAP PASS CURRENT DAP PAS
OURRCBIT	=	BIT12	#	IS RATE COMMAND NOT RATE COMMAND
# BIT 11 FLAG 13				
ACC4-2FL	=	199D	#	4 JET X-AXIS TRANS- 2 JET X-AXIS TR
ACC4OR2X	=	BIT11	#	LATION REQUESTED LATION REQUESTED
# BIT 10 FLAG 13				
AORBTFLG	=	200D	#	B SYSTEM FOR X- A SYSTEM FOR X-
AORBTRAN	=	BIT10	#	TRANSLATION TRANSLATION PRE
# BIT 9 FLAG 13				
XOVINFLG	=	201D	#	X-AXIS OVERRIDE X-AXIS OVERRIDE
XOVINHIB	=	BIT9	#	LOCKED OUT
# BIT 8 FLAG 13				
DRIFTDFL	=	202D	#	ASSUME 0 OFFSET USE OFFSET ACCE
DRIFTBIT	=	BIT8	#	DRIFTING FLIGHT ION ESTIMATE
# BIT 7 FLAG 13				
RHCSCFLG	=	203D	#	NORMAL RHC SCALING FINE RHC SCALIM
RHCSCALE	=	BIT7	#	REQUESTED REQUESTED

64a	$\langle$ Page LM0088 64a $\rangle \equiv$	(31 733)		
	# BIT 6 FLAG 13			
	ULLAGFLG = 204D	#	ULLAGE REQUEST BY	NO ID
	ULLAGER = BIT6	#	MISSION PROGRAM	REQU
	# BIT 5 FLAG 13			
	AORBSFLG = 205D	#	P-AXIS COUPLES 7.15	P-AXI
	AORBSYST = BIT5	#	AND 8.16 PREFERRED	AND 3
	# BIT 4 FLAG 13			
	DBSELFLG = 206D	#	MAX DB SELECTED	MIN I
	DBSELECT = BIT4	#	BY CREW (5 DEG)	CREW
	# BIT 3 FLAG 13			
	ACCOKFLG = 207D	#	CONTROL AUTHORITY	RESTA
	ACCSOKAY = BIT3	#	VALUES FROM 1/ACCS	SINCE
		#	USABLE	OUTPU
	# BIT 2 FLAG 13			
	AUTR2FLG = 208D	#	THESE FLAGS ARE USED TOGETHER TO ID	
	AUTRATE2 = BIT2	#	ASTRONAUT-CHOSEN KALCMANU MANEUVER	
	# BIT 1 FLAG 13	#	(0,0)=(BIT2,BIT1)=	0.2 DEG/SEC
	AUTR1FLG = 209D	#	(0,1)=	0.5 DEG/SEC
	AUTRATE1 = BIT1	#	(1,0)=	2.0 DEG/SEC
		#	(1,1)=	10.0 DEG/SEC

64b  $\langle$ Page LM0089 64b $\rangle \equiv$  (31 733)

## 1.4 rcs failure monitor

64c  $\langle$ rcs failure monitor 64c $\rangle \equiv$  (7)

$\langle$ Page LM0190 65 $\rangle$

$\langle$ Page LM0191 67 $\rangle$

$\langle$ Page LM0192 69 $\rangle$



65

(Page LM0190 65)≡

(64c 762)

```

# PROGRAM DESCRIPTION:
#
# AUTHOR: J. S. MILLER
#
# MODIFIED 6 MARCH 1968 BY P. S. WEISSMAN TO SET UP JOB FOR 1/ACCS WHEN THE MASKS ARE CHANGED.
#
# THIS ROUTINE IS ATTACHED TO T4RUPT, AND IS ENTERED EVERY 480 MS. ITS FUNCTION IS TO EXAMINE
# OF CHANNEL 32 TO SEE IF ANY ISOLATION-VALVE CLOSURE BITS HAVE APPEARED OR DISAPPEARED (THE CR
# FAILURES BY LAMPS LIT BY THE GRUMMAN FAILURE-DETECTION CIRCUITRY; THEY MAY RESPOND BY OPERATI
# ISOLATE PAIRS OF JETS FROM THE PROPELLANT TANKS AND SET BITS IN CHANNEL 32). IN THE EVENT TH
# DIFFER FROM 'PVALVEST', THE RECORD OF ACTIONS TAKEN BY THIS ROUTINE, THE APPROPRIATE BITS IN
# 'CH6MASK', USED BY THE DAP JET-SELECTION LOGIC, ARE UPDATED, AS IS 'PVALVEST'. TO SPEED UP &
# ROUTINE, NO MORE THAN ONE CHANGE IS ACCEPTED PER ENTRY. THE HIGHEST-NUMBERED BIT IN CHANNEL
# ACTION IS THE ONE PROCESSED.
#
# THE CODING IN THE FAILURE MONITOR HAS BEEN WRITTEN SO AS TO HAVE ALMOST COMPLETE RESTART PROT
# EXAMPLE, NO ASSUMPTION IS MADE WHEN SETTING A 'CH5MASK' BIT TO 1 THAT THE PREVIOUS STATE IS C
# COURSE SHOULD BE. ONE CASE WHICH MAY BE SEEN TO EVADE PROTECTION IS THE OCCURRENCE OF A REST
# ONE OR BOTH DAP MASK-WORDS BUT BEFORE UPDATING 'PVALVEST', COUPLED WITH A CHANGE IN THE VALVE
# FORMER STATE. THE CONSEQUENCE OF THIS IS THAT THE NEXT ENTRY WOULD NOT SEE THE CHANGE INCOMP
# ORATED BY THE LAST PASS (BECAUSE IT WENT AWAY AT JUST THE RIGHT TIME), BUT THE DAP MASK-WORDS
# THIS COMBINATION OF EVENTS SEEMS QUITE REMOTE, BUT NOT IMPOSSIBLE UNLESS THE CREW OPERATES TH
# SECOND INTERVALS OR LONGER. IN ANY EVENT, A DISAGREEMENT BETWEEN REALITY AND THE DAP MASKS W
# THE MISINTERPRETED SWITCH IS REVERSED AND THEN RESTORED TO ITS CORRECT POSITION (SLOWLY).
#
# CALLING SEQUENCE:
#
#          TCF          RCSMONIT          # (IN INTERRUPT MODE, EVERY 480 MS.)
#
# EXIT: TCF RCSMONEX (ALL PATHS EXIT VIA SUCH AN INSTRUCTION)

RCSMONEX          EQUALS  RESUME

# ERASABLE INITIALIZATION REQUIRED:
#
#          VIA FRESH START:          PVALVEST          =          +0          (ALL JETS ENABLED)
#                                     CH5MASK,CH6MASK =          +0          (ALL JETS OK)
#
# OUTPUT:          CH5MASK & CH6MASK UPDATED (1'S WHERE JETS NOT TO BE USED, IN CHANNEL 5 & 6 FORM
#                  PVALTEST UPDATED (1'S WHEN VALVE CLOSURES HAVE BEEN TRANSLATED INTO CH5MASK & C
#                  JOB TO DO 1/ACCS.
#
# DEBRIS:  A, L, AND Q AND DEBRIS OF NOVAC.
#
# SUBROUTINE CALLED:  NOVAC.

```

EBANK= CH5MASK

BANK 23

SETLOC RCSMONT

BANK

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67 (Page LM0191 67)≡

(64c 762)

```
COUNT*  $$/T4RCS

RCSMONIT  EQUALS  RCSMON

RCSMON    CS      ZERO
          EXTEND
          RXOR     CHAN32      # PICK UP + INVERT INVERTED CHANNEL 32.
          MASK     LOW8        # KEEP JET-FAIL BITS ONLY.
          TS       Q

          CS      PVALVEST    #
          MASK     Q          # FORM PC + PC.
          TS       L          # (P = PREVIOUS ISOLATION VALVE STATE,
          CS       Q          # C = CURRENT VALVE STATE (CH32)).
          MASK     PVALVEST
          ADS      L          # RESULT NZ INDICATES ACTION REQUIRED.

          EXTEND
          BZF      RCSMONEX    # QUIT IF NO ACTION REQUIRED.

          EXTEND
          MP       BIT7        # MOVE BITS 8-1 OF A TO 14-7 OF L.
          XCH      L          # ZERO TO L IN THE PROCESS.

-3         INCR     L
          DOUBLE
          OVSK
          TCF      -3

          INDEX    L
          CA       BIT8 -1    # SAVE THE RELEVANT BIT (8-1).
          TS       Q
          MASK     PVALVEST    # LOOK AT PREVIOUS VALVE STATE BIT.
          CCS      A
          TCF      VOPENED     # THE VALVE HAS JUST BEEN OPENED.

          CS       CH5MASK     # THE VALVE HAS JUST BEEN CLOSED.
          INDEX    L
          MASK     5FAILTAB
          ADS      CH5MASK     # SET INHIBIT BIT FOR CHANNEL 5 JET.

          CS       CH6MASK
          INDEX    L
          MASK     6FAILTAB
          ADS      CH6MASK     # SET INHIBIT BIT FOR CHANNEL 6 JET
```

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CA Q  
ADS PVALVEST  
TCF 1/ACCFIX

# RECORD ACTION TAKEN.

# SET UP 1/ACCJOB AND EXIT.

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69 (Page LM0192 69)≡

(64c 762)

```
VOPENED      INDEX  L
               CS    5FAILTAB
               MASK   CH5MASK
               TS     CH5MASK

               INDEX  L
               CS    6FAILTAB
               MASK   CH6MASK
               TS     CH6MASK

               CS     Q
               MASK   PVALVEST
               TS     PVALVEST

1/ACCFIX      CAF    PRI027
               TC     NOVAC
               EBANK= AOSQ
               2CADR  1/ACCJOB

               TCF    RCSMONEX

5FAILTAB      EQUALS -1
               OCT    00040
               OCT    00020
               OCT    00100
               OCT    00200
               OCT    00010
               OCT    00001
               OCT    00004
               OCT    00002

6FAILTAB      EQUALS -1
               OCT    00010
               OCT    00020
               OCT    00004
               OCT    00200
               OCT    00001
               OCT    00002
               OCT    00040
               OCT    00100

# A VALVE HAS JUST BEEN OPENED.

# REMOVE INHIBIT BIT FOR CHANNEL 5 JET.

# REMOVE INHIBIT BIT FOR CHANNEL 6 JET.

# RECORD ACTION TAKEN.

# SET UP 1/ACCS SO THAT THE SWITCH CURVES
#     FOR TJETLAW CAN BE MODIFIED IF CH5MASK
#     HAS BEEN ALTERED.

# EXIT.

# CH 5 JET BIT CORRESPONDING TO CH 32 BIT:
# 8
# 7
# 6
# 5
# 4
# 3
# 2
# 1

# CH 6 JET BIT CORRESPONDING TO CH 32 BIT:
# 8
# 7
# 6
# 5
# 4
# 3
# 2
# 1
```

## 1.5 ags initialization

$$\begin{aligned}
 70 \quad \langle ags \text{ initialization } 70 \rangle &\equiv & (7) \\
 &\langle Page \text{ LM0206 } 71 \rangle \\
 &\langle Page \text{ LM0207 } 73 \rangle \\
 &\langle Page \text{ LM0208 } 75 \rangle \\
 &\langle Page \text{ LM0209 } 77 \rangle \\
 &\langle Page \text{ LM0210 } 78a \rangle
 \end{aligned}$$

```

# PROGRAM NAME:  AGS INITIALIZATION (R47)
#
# WRITTEN BY:  RHODE/KILROY/FOLLETT
#
# MOD NO.:      0
# DATE:         23 MARCH 1967
# MOD BY:       KILROY
#
# MOD NO.:      1
# DATE:         28 OCTOBER 1967
# MOD BY:       FOLLETT
#
# FUNCT. DESC.: (1) TO PROVIDE THE AGS ABORT ELECTRONICS ASSEMBLY (AEA) WITH THE LEM AND CSM ST
#                (POSITION,VELOCITY,TIME) IN LEM IMU COORDINATES BY MEANS OF THE LGC DIGITAL DOW
#
#                (2) TO ZERO THE ICDU, LGC, AND AEA GIMBAL ANGLE COUNTER SIMULTANEOUSLY IN ORDER
#                COMMON ZERO REFERENCE FOR THE MEASUREMENT OF GIMBAL (EULER) ANGLES WHICH DEFINE
#
#                (3) TO ESTABLISH THE GROUND ELAPSED TIME OF AEA CLOCK ZERO.  (IF AN AEA CLOCK Z
#                REQUESTED DURING THIS PROGRAM
#
# LOG SECTION:  AGS INITIALIZATION
#
# CALLING SEQ:  PROGRAM IS ENTERED WHEN ASTRONAUT KEYS V47E ON DSKY.
#                R47 MAY BE CALLED AT ANY TIME EXCEPT WHEN ANOTHER EXTENDED VERB IS IN PROGRESS
#
# SUBROUTINES
# CALLED:
#
# NORMAL EXIT:  ENDEXT
#
# ALARM/ABORT:  ALARM -- BAD REFSMMAT -- CODE:220
#                OPERATOR ERROR IF V47 SELECTED DURING ANOTHER EXTENDED VERB.
#
# ERASABLES
# USED:         SAMPTIME          (2)      TIME OF :ENTER: KEYSTROKE
#                AGSK              (2)      GROUND ELAPSED TIME OF THE AEA CLOCK :ZERO:
#                AGSBUFF           (140)    CONTAINS AGS INITIALIZATION DATA (SEE :OUTPUT: BELOW)
#                AGSWORD           (1)      PREVIOUS DOWNLIST SAVED HERE
#
# EBANK=  AGSBUFF
#
# BANK      40
# SETLOC    R47

```

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BANK

COUNT\* \$\$/R47

AGSINIT

CAF REFSMBIT  
MASK FLAGWRD3  
CCS A

# CHECK REFSMFLG.



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(70 713)

	TC	REDSPTM	# REFSMMAT IS OK
	TC	ALARM	# REFSMMAT IS BAD
	OCT	220	
	TC	ENDEXT	
NEWAGS	EXTEND		
	DCA	SAMPTIME	# TIME OF THE :ENTER: KEYSTROKE
	DXCH	AGSK	# BECOMES NEW AEA CLOCK :ZERO:
REDSPTM	EXTEND		
	DCA	AGSK	
	DXCH	DSPTMX	
AGSDISPK	CAF	V06N16	
	TC	BANKCALL	# R1 = 00XXX. HRS., R2 = 000XX MIN.,
	CADR	GOMARKF	# R3 = 0XX.XX SEC.
	TC	ENDEXT	# TERMINATE RETURN
	TC	AGSVCALC	# PROCEED RETURN
	CS	BIT6	# IS ENTER VIA A V32
	AD	MPAC	
	EXTEND		
	BZF	NEWAGS	# YES, USE KEYSTROKE TIME FOR NEW AGSK
	EXTEND		
	DCA	DSPTMX	# NO, NEW AGSK LOADED VIA V25
	TC	REDSPTM -1	# LOADED INTO DSPTMX BY KEYING
			# V25E FOLLOWED BY HRS.,MINS.,SECS.
			# DISPLAY THE NEW K.
AGSVCALC	TC	INTPRET	
	SET		
		NODOFLAG	# DON'T ALLOW V37
	SET	EXIT	
		XDSPFLAG	
	CAF	V06N16	
	TC	BANKCALL	
	CADR	EXDSPRET	
	TC	INTPRET	# EXTRAPOLATE LEM AND CSM STATE VECTORS
	RTB		# TO THE PRESENT TIME
		LOADTIME	# LOAD MPAC WITH TIME2,TIME1
	STCALL	TDEC1	# CALCULATE LEM STATE VECTOR
		LEMPREC	
	CALL		# CALL ROUTINE TO CONVERT TO SM COORDS
		SCALEVEC	# PROVIDE PROPER SCALING
	STODL	AGSBUFF	# (LEMPREC AND CSMPREC LEAVE TDEC1 IN T

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                 TAT  
STCALL    TDEC1  
                 CSMPREC  
CALL  
                 SCALEVEC

# TAT = TIME TO WHICH RATT1 A  
# COMPUTED (CSEC SINCE CLOCK  
# CALCULATE CSM STATE VECTOR

75 (Page LM0208 75)≡

(70 713)

```

STODL  AGSBUFF +6
        TAT
DSU     DDV          # CALCULATE AND STORE THE TIME
        AGSK
        TSCALE
STORE   AGSBUFF +12D
EXIT

CAF     LAGSLIST
TS      DNLSTCOD

CAF     20SEC        # DELAY FOR 20 SEC WHILE THE AGS
TC      BANKCALL     # DOWNLIST IS TRANSMITTED
CADR    DELAYJOB

CA      AGSWORD
TS      DNLSTCOD        # RETURN TO THE OLD DOWNLIST
CAF     IMUSEBIT
MASK    FLAGWRDO        # CHECK IMUSE FLAG.
CCS     A
TC      AGSEND          # IMU IS BEING USED -- DO NOT ZERO
CKSTALL CCS IMUCADR     # CHECK FOR IMU USAGE WHICH AVOIDS THE
TCF     +3              # IMUSE BIT: I.E., IMU COMPENSATION.
TCF     +6              # FREE. GO AHEAD WITH THE IMU ZERO.
TCF     +1
+3      CAF TEN         # WAIT .1 SEC AND TRY AGAIN.
        TC BANKCALL
        CADR DELAYJOB
        TCF CKSTALL

+6      TC BANKCALL     # IMU IS NOT IN USE
        CADR IMUZERO    # SET IMU ZERO DISCRETE FOR 320 MSECS.
        TC BANKCALL     # WAIT 3 SEC FOR COUNTERS TO INCREMENT
        CADR IMUSTALL
        TC AGSEND
AGSEND  TC DOWNFLAG     # ALLOW V37
        ADRES NODOFLAG

CAF     V50N16
TC      BANKCALL
CADR    GOMARK3
TCF     ENDEXT
TCF     ENDEXT
TC      ENDEXT

```

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SCALEVEC	VLOAD	MXV
		VATT1
		REFSMMAT
	VXSC	VSL2
		VSCALE

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(70 713)

VAD	VAD	# THIS SECTION ROUNDS THE VECTOR, AND
	AGSRND1	# CORRECTS FOR THE FACT THAT THE AGS
	AGSRND2	# IS A 2'S COMPLEMENT MACHINE WHILE THE
RTB		# LGC IS A 1'S COMPLEMENT MACHINE.
	VECSGNAG	
STOVL	VATT1	
	RATT1	
MXV	VXSC	
	REFSMMAT	
	RSCALE	
VSL8	VAD	# AGAIN THIS SECTION ROUNDS. TWO VECTO
	AGSRND1	# ARE ADDED TO DEFEAT ALSIGNAG IN THE
VAD	RTB	# CASE OF A HIGH-ORDER ZERO COUPLED WIT
	AGSRND2	# A LOW ORDER NEGATIVE PART.
	VECSGNAG	
LXA,1		
	VATT1	
SXA,1	LXA,1	
	MPAC +1	
	VATT1 +2	
SXA,1	LXA,1	
	MPAC +4	
	VATT1 +4	
SXA,1	RVQ	
	MPAC +6	
LAGSLIST	=	ONE
V01N14	VN	0114
V50N00A	VN	5000
V00N25	EQUALS	OCT31
V06N16	VN	0616
V00N34	EQUALS	34DEC
V50N16	VN	5016
TSCALE	2DEC	100 B-10
20SEC	DEC	2000
RSCALE	2DEC	3.280839 B-3
VSCALE	2DEC	3.280839 E2 B-9
AGSRND1	2OCT	0000060000
	2OCT	0000060000
	2OCT	0000060000
AGSRND2	2OCT	0000037777
	2OCT	0000037777
S		

# CSEC TO SEC SCALE FACTOR

# METERS TO FEET SCALE FACTOR

# METERS/CS TO FEET/SEC SCALE FACTOR

78a     $\langle \text{Page LM0210 78a} \rangle \equiv$  (70 713)

20CT    0000037777

SBANK=    LOWSUPER    # FOR SUBSEQUENT LOW 2CADRS.

## 1.6 aotmark routine

78b     $\langle \text{aotmark routine 78b} \rangle \equiv$  (7)

$\langle \text{Page LM0244 79} \rangle$   
 $\langle \text{Page LM0245 80} \rangle$   
 $\langle \text{Page LM0246 81} \rangle$   
 $\langle \text{Page LM0247 82} \rangle$   
 $\langle \text{Page LM0248 83} \rangle$   
 $\langle \text{Page LM0249 84} \rangle$   
 $\langle \text{Page LM0250 86} \rangle$   
 $\langle \text{Page LM0251 87} \rangle$   
 $\langle \text{Page LM0252 88} \rangle$   
 $\langle \text{Page LM0253 90} \rangle$   
 $\langle \text{Page LM0254 92} \rangle$   
 $\langle \text{Page LM0255 94} \rangle$   
 $\langle \text{Page LM0256 95} \rangle$   
 $\langle \text{Page LM0257 96} \rangle$   
 $\langle \text{Page LM0258 97} \rangle$   
 $\langle \text{Page LM0259 98} \rangle$   
 $\langle \text{Page LM0260 99} \rangle$   
 $\langle \text{Page LM0261 100} \rangle$

79 (Page LM0244 79)≡

(78b 716)

	BANK	12	
	SETLOC	AOTMARK1	
	BANK		
	EBANK=	XYMARK	
	COUNT*	\$\$/MARK	
AOTMARK	INHINT		
	CCS	MARKSTAT	# SEE IF AOTMARK BUSY
	TC	+2	# MARK SYSTEM BUSY -- DO ALARM
	TC	EXTVBCHK	
	TC	POOD00	
	OCT	00105	
EXTVBCHK	CAF	SIX	# SEE IF EXT. VERB WORKING
	MASK	EXTVBACT	
	CCS	A	
	TCF	MKABORT	# YES -- ABORT
	CAF	BIT2	# NO -- DISALLOW SOME EXTENDED VERB ACTION
	ADS	EXTVBACT	# BIT2 RESET IN ENDMARK
MKVAC	CCS	VAC1USE	# LOOK FOR A VAC AREAD -- DO ABORT IF
	TCF	MKVACFND	# NONE AVAILABLE
	CCS	VAC2USE	
	TCF	MKVACFND	
	CCS	VAC3USE	
	TCF	MKVACFND	
	CCS	VAC4USE	
	TCF	MKVACFND	
	CCS	VAC5USE	
	TCF	MKVACFND	
	DXCH	BUF2	
	TC	BAILOUT1	# ALL VAC AREAS OCCUPIED -- ABORT.
	OCT	01207	
MKVACFND	AD	TWO	
	TS	MARKSTAT	# STORE VAC ADR IN LOW 9 OF MARKSTAT
	CAF	ZERO	
	INDEX	MARKSTAT	
	TS	0 -1	# ZERO IN VACUSE REG TO SHOW VAC OCCUPIED
	CAF	PRI015	
	TC	FINDVAC	# SET UP JOB FOR GETDAT
	EBANK=	XYMARK	

	2CADR	GETDAT	
	RELINT		
	TCF	SWRETURN	
80	$\langle \text{Page } LM0245 \text{ } 80 \rangle \equiv$		(78b 716)
	MKABORT	DXCH	BUF2
		TC	BAILOUT1
		OCT	01211
			# CONFLICT WITH EXTENDED VERB
	MKRELEAS	CAF	ZERO
		XCH	MARKSTAT
			# SET MARKSTAT TO ZERO
		MASK	LOW9
			# PICK UP VAC AREA AOR
		CCS	A
		INDEX	A
		TS	0
			# SHOW MKVAC AREA AVAILABLE
		CAF	ONE
		TC	IBNKCALL
		CADR	GOODEND
			# GO WAKE UP CALLING JOB



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81 (Page LM0246 81)≡

(78b 716)

KILLAOT	CAF	ZERO	
	TS	EXTVBACT	# TERMINATE AOTMARK -- ALLOW EXT VERB
	TC	GOTPOOH	
GETDAT	CS	MARKSTAT	# SET BIT12 TO DISCOURAGE MARKRUPT
	MASK	BIT12	# BIT12 RESET AT GETMARK
	ADS	MARKSTAT	
	CAF	V01N71	# DISPLAY DETENT AND STAR CODE
	TC	BANKCALL	
	CADR	GOMARKF	
	TCF	KILLAOT	# V34 -- DOES GOTPOOH
	TCF	DODAT	# V33 -- PROCEED -- USE THIS STAR FOR MARKS
ENTERDAT	TCF	GETDAT	# ENTER -- REDISPLAY STAR CODE
	CAF	HIGH9	# PICK DETENT CODE FROM BITS7-9 OF AOTCODE
DODAT	MASK	AOTCODE	# AND SEE IF CODE 1 TO 6
	EXTEND		
	MP	BIT9	
	TS	XYMARK	# STORE DETENT
	EXTEND		
	BZMF	GETDAT	# COAS CALIBRATION CODE - NO GOOD HERE
	AD	NEG7	# SEE IF DETENT 7 FOR COAS
	EXTEND		
	BZF	CODE7	
	TCF	CODE1TO6	
CODE7	CAF	V06N87*	# CODE 7, COAS SIGHTING, GET OPTIC AXIS
	TC	BANKCALL	# AZ AND EL OF SIGHTING DEVICE FROM ASTRO
	CADR	GOMARKF	
	TCF	KILLAOT	# V34 -- DOES GOTPOOH
	TCF	+2	# PROCEED
	TCF	CODE7	# ON ENTER, RECYCLE
	EXTEND		
	DCA	AZ	# PICK UP AZ AND EL IN SP 25 COMP
	INDEX	FIXLOC	
	DXCH	8D	# STORE IN 8D AND 9D OF LOCAL VAC
	CAF	ZERO	# BACKUP SYSTEM TO BE USED
	TCF	COASCODE	# ZERO APPARENT ROTATION
CODE1TO6	INDEX	XYMARK	# INDEX AOT POSITION BY DET CODE

	CA	AOTEL -1	
	INDEX	FIXLOC	
	TS	9D	# STORE ELEVATION IN VAC+9D
	INDEX	XYMARK	# INDEX DET CODE 1,2 OR 3
82	<i>(Page LM0247 82)≡</i>		(78b 716)
	CA	AOTAZ -1	
	INDEX	FIXLOC	
	TS	8D	# STORE AZIMUTH IN VAC +8D
	CA	AOTAZ +1	# COMPENSATION FOR APPARENT ROTATION OF
	EXTEND		# AOT FIELD OF VIEW IN LEFT AND RIGHT
	INDEX	FIXLOC	# DETENTS IS STORED IN VAC +10D IN SP
	MSU	8D	# PRECISION ONE'S COMPLEMENT
COASCODE	INDEX	FIXLOC	
	TS	10D	# ROT ANGLE
	TC	INTPRET	# COMPUTE X AND Y PLANE VECTORS

83 (Page LM0248 83)≡

(78b 716)

```

# THE OPTAXIS SUBROUTINE COMPUTES THE X AND Y MARK PLANE VECs AND
# ROTATES THEM THRU THE APPARENT FIELD OF VIEW ROTATION UNIQUE TO AOT
# OPTAXIS USES OANB TO COMPUTE THE OPTIC AXIS
#
#      INPUT --      AZIMUTH ANGLE IN SINGLE PREC AT CDU SCALE IN 8D OF JOB VAC
#                    ELEVATION ANGLE IN SINGLE PREC AT CDU SCALE IN 9D OF JOB VAC
#                    ROTATION ANGLE IN SINGLE PREC IS COMP SCALED BY PI IN 10D OF VAC
#
#      OUTPUT --     OPTIC AXIS VEC IN NG COORDS IN SCAXIS
#                    X-MARK PLANE 1/4VEC IN NB COORDS AT 18D OF JOB VAC
#                    Y-MARK PLANE 1/4VEC IN NB COORDS AT 12D OF JOB VAC

OPTAXIS      CALL      # GO COMPUTE OA AN X AND Y PLANE VECs
              OANB
              SLOAD    SR1      # LOAD APP ROTATION IN ONES COMP
              10D        # RESCALE BY 2PI
              PUSH     SIN      # 1/2SIN(ROT) 0-1
              PDDL     COS
              PUSH     VXSC     # 1/2COS(ROT) 2-3
              18D
              PDDL     VXSC     # 1/4COS(ROT)UYP 4-9
              0
              24D        # 1/4SIN(ROT)UXP
              BVSU     STADR     # UP 4-9
              STODL    12D      # YPNB=1/4(COS(ROT)UYP-SIN(ROT)UXP)
              VXSC     PDDL     # UP 2-3 UP 0-1 FOR EXCHANGE
              24D        # 1/4COS(ROT)UXP          PUSH 0-5
              VXSC     VAD       # 1/4SIN(ROT)UYP
              18D        # UP 0-5
              STADR
              STOVL     18D      # XPNB=1/4(COS(ROT)UXP+SIN(ROT)UYP)
              LO6ZEROS  # INITIALIZE AVE STAR VEC ACCUMULATOR
              STORE    STARAD +6
              EXIT
              TCF      GETMKS

```

84 (Page LM0249 84)≡

(78b 716)

```

# THE OANB SUBROUTINE COMPUTES THE OPTIC AXIS OF THE SIGHTING INSTRUMENT
# FROM AZIMUTH AND ELEVATION INPUT FROM THE ASTRONAUT.
#
#      INPUT --      AZIMUTH ANGLE IN SINGLE PREC 2'S COMP IN 8D OF JOB VAC
#                      ELEVATION ANGLE IN SINGLE PREC 2'S COMP IN 9D OF VAC
#
#      OUTPUT --     OPTIC AXIS IN NB COORDS. IN SCAXIS
#                      X-PLANE 1/2VEC IN NB COORDS AT 24D OF VAC
#                      Y-PLANE 1/2VEC IN NB COORDS AT 18D OF VAC

      BANK      05
      SETLOC    AOTMARK2
      BANK

      COUNT*    $$/MARK

OANB      SETPD   STQ
           0
           GCTR      # STORE RETURN
      SLOAD    RTB
           9D        # PICK UP SP ELV
           CDULOGIC
      PUSH     COS
      PDDL     SIN      # 1/2COS(ELV)   PD 0-1
      STADR
      STODL    SCAXIS    # OAX=1/2SIN(ELV)
           8D
      RTB
           CDULOGIC
      PUSH     COS
      STORE    20D      # STORE UYP(Y)   20-21
      PDDL     SIN      # 1/2COS(AZ)     PD 2-3
      PUSH     DCOMP    # PUSH 1/2S IN (AZ)   4-5
      STODL    22D      # STORE UYP(Z)   22-23
           L06ZEROS
      STODL    18D      # STORE UYP(X)   18-19
      DMP      SL1
           0
      STODL    SCAXIS +2 # OAY=1/2COS(ELV)SIN(AZ)
      DMP      SL1      # UP      2-3
      STADR    # UP      0-1
      STOVL    SCAXIS +4 # OAZ=1/2COS(ELV)COS(AZ)
           18D          # LOAD UYP VEC
      VXV      UNIT
           SCAXIS      # UXP VEC=UYP X OA

```

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STORE 24D  
GOTO  
GCTR

# STORE UXP

86 (Page LM0250 86)≡

(78b 716)

```

# SURFSTAR COMPUTES A STAR VECTOR IN SM COORDINATES FOR LUNAR
# SURFACE ALIGNMENT AND EXITS TO AVEIT TO AVERAGE STAR VECTORS.
#
#      GIVEN   X-MARK PLANE 1/4 VEC IN NB AT 18D OF LOCAL VAC
#              Y-MARK PLANE 1/4 VEC IN NB AT 12D OF LOCAL VAC
#              CURSOR SP 2COMP AT POSITION 1 OF INDEXED MARKVAC
#              SPIRAL SP 2COMP AT POSITION 3 OF INDEXED MARKVAC
#              CDUY,Z,X AT POSITIONS 0,2,4 OF INDEXED MARKVAC

BANK      15
SETLOC    P50S
BANK
COUNT*   $$/R59

SURFSTAR   VLOAD*
           0,1          # PUT X-MARK CDUS IN CDUSPOT FOR TRG*NBSM
STORE      CDUSPOT
SLOAD*     RTB
           1,1          # PICK UP YROT
           CDULOGIC
STORE      24D          # STORE CURSOR FOR SPIRAL COMP (REVS)
BZE
JUSTZY     YZCHK        # IF YROT ZERO -- SEE IF SROT ZERO
PUSH       COS
PDDL       SIN          # 1/2COS(YROT) 0-1
VXSC       PDDL         # UP 0-1          1/8SIN(YROT)UXP 0-5
           18D
VXSC       VSU          # UP      0-5
           12D          # UYP
UNIT       VXV
           SCAXIS
UNIT       PUSH
SLOAD*     RTB
           3,1          # PICK UP SPIRAL
           CDULOGIC
STORE      26D          # STORE SPIRAL (REVS)
DSU        DAD
           24D
           ABOUTONE
DMP
           DP1/12
STORE      26D          # SEP=(360 + SPIRAL -CURSOR)/12
SIN        VXSC         # UP      0-5
VSL1       PDDL         # 1/2SIN(SEP)(UPP X 0A) 0-5
           26D

```

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		COS	VXSC	
			SCAXIS	
		VSL1	VAD	# UP 0-5
	JUSTOA	UNIT	CALL	
			TRG*NBSM	
		STCALL	24D	# STAR VEC IN SM
			AVEIT	# GO AVERAGE
87	$\langle$ Page LM0251 87 $\rangle \equiv$			(78b 716)
	ABOUTONE	2DEC	.99999999	
	DP1/12	EQUALS	DEG30	# .08333333
		BANK	7	
		SETLOC	AOTMARK1	
		BANK		
		COUNT*	\$\$/MARK	
	YZCHK	SLOAD*	BZE	# YROT ZERO AND IF SROT ZERO FORCE STAR
			3,1	# ALONG OPTIC AXIS
			YSZERO	
		DLOAD	GOTO	
			24D	
			JUSTZY	# SROT NOT ZERO -- CONTINUE NORMALLY
	YSZERO	VLOAD	GOTO	
			SCAXIS	
			JUSTOA	

88 (Page LM0252 88)≡

(78b 716)

# THE GETMKS ROUTINE INITIALIZES THE SIGHTING MARK PROCEDURE

GETMKS	CAF	ZERO	# INITIALIZE MARK ID REGISTER AND MARK CNT
	TS	XYMARK	
	TS	MARKCNTR	
	CAF	LOW9	# ZERO BITS10 TO 15 RETAINING MKVAC ADR
	MASK	MARKSTAT	
	TS	MARKSTAT	
	CAF	MKVB54*	# DISPLAY VB54 INITIALLY
PASTIT	TC	BANKCALL	
	CADR	GOMARK4	
	TCF	KILLAOT	# V34 -- DOES GOTOPOOH
	TCF	MARKCHEX	# VB33 -- PROCEED, GOT MARKS, COMPUTE LOS
	TCF	GETDAT	# ENTER -- RECYCLE TO V01N71
MARKCHEX	CS	MARKSTAT	# SET BIT12 TO DISCOURAGE MARKRUPT
	MASK	BIT12	
	ADS	MARKSTAT	
	MASK	LOW9	
	TS	XYMARK	# JAM MARK VAC ADR IN XYMARK FOR AVESTAR
	CAF	ZERO	
	TS	MKDEX	# SET MKDEX ZERO FOR LOS VEC CNTR
	CA	MARKSTAT	
	MASK	PRI03	# SEE IF LAST MK PART COMPLETE
	TS	L	
	CAF	PRI03	# BITS10 AND 11
	EXTEND		
	RXOR	LCHAN	
	EXTEND		
	BZF	AVESTAR	# LAST PAIR COMPLETE -- TO COMPUTE LOS
CNTCHK	CCS	MARKCNTR	# NO PAIR SHOWING -- SEE IF PAIR IN HOLD
	TCF	+2	# PAIR BURIED -- DECREMENT COUNTER
	TCF	MKALARM	# NO PAIR -- ALARM
	TS	MARKCNTR	# STORE DECREMENTED COUNTER
AVESTAR	CAF	BIT12	# INITIALIZE MKDEX FOR STAR LOS COUNTER
	ADS	MKDEX	# MKDEX WAS INITIALIZED ZERO IN MARKCHEX
	CS	MARKCNTR	
	EXTEND		
	MP	SIX	# GET C(L) = -6 MARKCNTR
	CS	XYMARK	
	AD	L	# ADD -- MARK VAC ADR SET IN MARKCHEX
	INDEX	FIXLOC	
	TS	X1	# JAM -- CDU ADR OF X-MARK IN X1



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CA	FIXLOC	# SET PD POINTER TO ZERO
TS	PUSHLOC	
TC	INTPRET	

90 (Page LM0253 90)≡

(78b 716)

	BON	VLOAD*	
		SURFFLAG	# IF ON SURFACE COMPUTE VEC AT SURFSTAR
		SURFSTAR	
		1,1	# PUT Y-MARK CDUS IN CDUSPOT FOR TRG*NBSM
	STOVL	CDUSPOT	
		12D	# LOAD Y-PLANE VECTOR IN NG
	CALL		
		TRG*NBSM	# CONVERT IT TO STABLE MEMBER
	PUSH	VLOAD*	
		0,1	# PUT X-MARK CDUS IN CDUSPOT FOR TRG*NBSM
	STOVL	CDUSPOT	
		18D	# LOAD X-PLANE VECTOR IN NB
	CALL		
		TRG*NBSM	# CONVERT IT TO STABLE-MEMBER
	VXV	UNIT	# UNIT(XPSM * YPSM)
	STADR		
	STORE	24D	
AVEIT	SLOAD	PDVL	# N(NUMBER OF VECs) IN 0-1
		MKDEX	
		24D	# LOAD CURRENT VECTOR
	VSR3	V/SC	
		0	
	STODL	24D	# VEC/N
		0	
	DSU	DDV	
		DP1/8	# (N-1)/N
	VXSC	VAD	
		STARAD +6	# ADD VEC TO PREVIOUSLY AVERAGED VECTOR
		24D	# (N-1)/N AVESTVEC + VEC/N
	STORE	STARAD +6	# AVERAGE STAR VECTOR
	STORE	STARSAV2	
	EXIT		
	CCS	MARKCNTR	# SEE IF ANOTHER MARK PAIR IN MKVAC
	TCF	AVESTAR -1	# THERE IS -- GO GET IT -- DECREMENT COUNTER
ENDMARKS	CAF	FIVE	# NO MORE MARKS -- TERMINATE AOTMARK
	INHINT		
	TC	WAITLIST	
	EBANK=	XYMARK	
	2CADR	MKRELEAS	
	TC	ENDMARK	
MKALARM	TC	ALARM	# NOT A PAIR TO PROCESS -- DO GETMKS
	OCT	111	

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	TCF	GETMKS
V01N71	VN	171
V06N87*	VN	687

92 (Page LM0254 92)≡

(78b 716)

# MARKRUPT IS ENTERED FROM INTERRUPT LEAD-INS AND PROCESSES CHANNEL 16  
 # CAUSED BY X,Y MARK OR MARK REJECT OR BY THE RATE OF DESCENT SWITCH

MARKRUPT	TS	BANKRUPT	
	CA	CDUY	# STORE CDUS AND TIME NOW -- THEN SEE IF
	TS	ITEMP3	# WE NEED THEM
	CA	CDUZ	
	TS	ITEMP4	
	CA	CDUX	
	TS	ITEMP5	
	EXTEND		
	DCA	TIME2	
	DXCH	ITEMP1	
	XCH	Q	
	TS	QRUPT	
	CAF	OCT34	# SEE IF X OR Y MARK OR MKREJECT
	EXTEND		
	RAND	NAVKEYIN	
	CCS	A	
	TCF	+2	# ITS A LIVE ONE -- SEE IF ITS WANTED
	TCF	SOMEKEY	# ITS SOME OTHER KEY
	CAF	BIT12	# ARE WE ASKING FOR A MARK
	MASK	MARKSTAT	
	CCS	A	
	TC	RESUME	# DON'T WANT MARK OR MKREJECT -- DO NOTHING
	CCS	MARKSTAT	# ARE MARKS BEING ACCEPTED
	TCF	FINDKEY	# THEY ARE -- WHICH ONE IS IT
	TC	ALARM	# MARKS NOT BEING ACCEPTED -- DO ALARM
	OCT	112	
	TC	RESUME	
FINDKEY	CAF	BIT5	# SEE IF MARK REJECT.
	EXTEND		
	RAND	NAVKEYIN	
	CCS	A	
	TCF	MKREJ	# IT'S A MARK REJECT
	CAF	BIT4	# SEE IF Y MARK
	EXTEND		
	RAND	NAVKEYIN	
	CCS	A	

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TCF	YMKRUPT	# IT'S A Y MARK
CAF	BIT3	# SEE IF X MARK
EXTEND		
RAND	NAVKEYIN	

94 (Page LM0255 94)≡

(78b 716)

	CCS	A	
	TCF	XMKRUPT	# IT'S A X MARK
SOMEKEY	CAF	OCT140	# NOT MARK OR MKREJECT -- SEE IF DESCENT BITS
	EXTEND		
	RAND	NAVKEYIN	
	EXTEND		
	BZF	+3	# IF NO BITS
	TC	POSTJUMP	# IF DESCENT BITS
	CADR	DESCBITS	
	TC	ALARM	# NO INBITS IN CHANNEL 16.
	OCT	113	
	TC	RESUME	
XMKRUPT	CAF	ZERO	
	TS	RUPTREG1	# SET X MARK STORE INDEX TO ZERO
	CAF	BIT10	
	TCF	+4	
YMKRUPT	CAF	ONE	
	TS	RUPTREG1	# SET Y MARK STORE INDEX TO ONE
	CAF	BIT11	
	TS	XYMARK	# SET MARK IDENTIFICATION
	TC	MARKTYPE	# SEE IF SURFACE MARK
	TCF	SURFSTOR	# SURFACE MARK -- JUST STORE CDUS
	CAF	BIT14	# GOT A MARK -- SEE IF MARK PARI MADE
	MASK	MARKSTAT	
	EXTEND		
	BZF	VERIFYMK	# NOT A PAIR, NORMAL PROCEDURE
	CS	MARKCNTR	# GO A PAIR, SEE IF ANOTHER CAN BE MADE
	AD	FOUR	# IF SO, INCREMENT POINTER, CLEAR BITS 10,11
	EXTEND		
	BZMF	5MKALARM	# HAVE FIVE MARK PAIRS -- DON'T ALLOW MARK
	INCR	MARKCNTR	# OK FOR ANOTHER PAIR, INCR POINTER
	CS	PRI023	# CLEAR BITS 10,11,14 FOR NEXT PAIR
	MASK	MARKSTAT	
	TS	MARKSTAT	
VERIFYMK	CA	XYMARK	
	MASK	MARKSTAT	
	CCS	A	

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TCF	+2	# THIS MARK NOT DESIRED
TCF	VACSTOR	# MARK DESIRED -- STORE CDUS
TC	ALARM	
OCT	114	
TC	RESUME	# RESUME -- DISPLAY UNCHANGED -- WAIT FOR ACTION

95	$\langle$ Page LM0256 95 $\rangle \equiv$		(78b 716)	
	5MKALARM	TC	ALARM	# ATTEMPTING TO MAKE MORE THAN 5 MK PAIRS
		OCT	107	
		TC	MARKTYPE	# SEE IF SURFACE MARK
		TCF	DSPV6N79	# IT IS
		TC	RESUME	# DON'T CHANGE DISPLAY -- DO NOTHING

96 (Page LM0257 96)≡

(78b 716)

MKREJ	TC	MARKTYPE	# SEE IF SURFACE
	TCF	SURFREJ	# SURFACE -- JUST CHECK MARK COUNTER
	CAF	PRI03	# INFLIGHT -- SEE IF MARKS MADE
	MASK	MARKSTAT	
	CCS	A	
	TCF	REJECT	# MARKS MADE -- REJECT ONE
REJALM	TC	ALARM	# NO MARK TO REJECT -- BAD PROCEDURE -- ALARM
	OCT	115	
	TC	RESUME	# DESIRED ACTION DISPLAYED
REJECT	CS	PRI030	# ZERO BIT14, SHOW REJ., SEE IF MARK SINCE
	MASK	MARKSTAT	# LAST REJECT
	AD	BIT13	
	XCH	MARKSTAT	
	MASK	BIT13	
	CCS	A	
	TCF	REJECT2	# ANOTHER REJECT SET BIT 10+11 TO ZERO
RENEWMK	CS	XYMARK	# MARK MADE SINCE REJECT -- REJECT MARK IN 11
	MASK	MARKSTAT	
	TS	MARKSTAT	
	TCF	REMARK	# GO REQUEST NEW MARK ACTION
REJECT2	CS	PRI03	# ON SECOND REJECT -- DISPLAY VB53 AGAIN
	TCF	RENEWMK	
SURFREJ	CCS	MARKCNTR	# IF MARK DECREMENT COUNTER
	TCF	+2	
	TCF	REJALM	# NO MARKS TO REJECT -- ALARM
	TS	MARKCNTR	
	TC	RESUME	



97 (Page LM0258 97)≡

(78b 716)

# MARKTYPE TESTS TO SEE IF LEM ON LUNAR SURFACE. IF IT IS RETURN TO LOC+1

MARKTYPE	CS	FLAGWRD8	# SURFFLAG ***** TEMPORARY *****
	MASK	BIT8	
	CCS	A	
	INCR	Q	# IF SURFACE MARK RETURN TO LOC +1
	TC	Q	# IF INFLIGHT MARK RETURN TO LOC +2

SURFSTOR	CAF	ZERO	# FOR SURFACE MARK ZERO MARK KIND INDEX
	TS	RUPTREG1	

	CS	MARKSTAT	# SET BITS10,11 TO SHOW SURFACE MARK
	MASK	PRI03	# FOR MARKCHEX
	ADS	MARKSTAT	

VACSTOR	CAF	LOW9	
	MASK	MARKSTAT	# STORE MARK VAC ADR IN RUPTREG2
	TS	RUPTREG2	
	EXTEND		
	DCA	ITEMP1	# PICK UP MARKTIME
	DXCH	TSIGHT	# STORE LAST MARK TIME
	CA	MARKCNTR	# 6 X MARKCNTR FOR STORE INDEX
	EXTEND		
	MP	SIX	
	XCH	L	# GET INDEX FROM LOW ORDER PART
	AD	RUPTREG2	# SET CDU STORE INDEX TO MARKVAC
	ADS	RUPTREG1	# INCREMENT VAC PICKUP BY MARK FOR FLIGHT
	TS	MKDEX	# STORE HERE IN CASE OF SURFACE MARK
	CA	ITEMP3	
	INDEX	RUPTREG1	
	TS	0	# STORE CDUY
	CA	ITEMP4	
	INDEX	RUPTREG1	
	TS	2	# STORE CDUZ
	CA	ITEMP5	
	INDEX	RUPTREG1	
	TS	4	# STORE CDUX
	TC	MARKTYPE	# IF SURFACE MARK -- JUST DO SURFJOB
	TCF	SURFJOB	

	CAF	BIT13	# CLEAR BIT13 TO SHOW MARK MADE
	AD	XYMARK	# SET MARK ID IN MARKSTAT
	COM		
	MASK	MARKSTAT	
	AD	XYMARK	

```
TS      MARKSTAT
MASK    PRI03      # SEE IF X, Y MARK MADE
TS      L
```

98  $\langle$ Page LM0259 98 $\rangle \equiv$ 

(78b 716)

```
CA      PRI03
EXTEND
RXOR    LCHAN
CCS     A
TCF     REMARK      # NOT PAIR YET, DISPLAY MARK ACTION
CS      MARKSTAT    # MARK PAIR COMPLETE -- SET BIT14
MASK    BIT14
ADS     MARKSTAT
TCF     REMARK      # GO DISPLAY V54
```

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99 (Page LM0260 99)≡

(78b 716)

REMARK	CAF	PRI03	# BITS 10 AND 11
	MASK	MARKSTAT	
	EXTEND		
	MP	BIT6	# SHIFT MARK IDS TO BE 0 TO 3 FOR INDEX
	TS	MKDEX	# STORE VERB INDEX
SURFJOB	CAF	PRI015	
	TC	NOVAC	# ENTER JOB TO CHANGE DISPLAY TO
	EBANK=	XYMARK	# REQUEST NEXT ACTION
	2CADR	CHANGEVB	
	TC	RESUME	
CHANGEVB	TC	MARKTYPE	
	TCF	DSPV6N79	# SURFACE -- DISPLAY V 06 N 79
	INDEX	MKDEX	# INFLIGHT -- PICK UP MARK VB INDEX
	CAF	MKVB54	
	TC	PASTIT	# PASTE UP NEXT MK VERB DISPLAY

# THE FOUR MKVBS ARE INDEXED -- THEIR ORDER CANNOT BE CHANGED

MKVB54	VN	5471	# MAKE X OR Y MARK
MKVB53	VN	5371	# MAKE Y MARK
MKVB52	VN	5271	# MAKE X MARK
MKVB54*	VN	5471	# MAKE X OR Y MARK
DP1/8	2DEC	.125	
OCT34	OCT	34	
V06N71	VN	671	
V06N79*	VN	679	

100 &lt;Page LM0261 100&gt;≡

(78b 716)

# ROUTINE TO REQUEST CURSOR AND SPIRAL MEASUREMENTS

COUNT\* \$\$/R59

DSPV6N79

CAF V06N79\*  
TC BANKCALL  
CADR GOMARKF

# CURSOR -- SPIRAL DISPLAY

TCF KILLAOT  
TCF SURFEND  
CAF BIT6  
MASK MPAC  
CCS A  
TCF SURFAGAN  
TCF DSPV6N79# V34 -- DOES GOTOP00H  
# V33 -- PROCEED, END MARKING  
# IF V32(OCT40) IN MPAC DO RECYCLE  
# OTHERWISE IT IS LOAD VB ENTER SO  
# RE-DISPLAY V06N79  
# VB32 -- RECYCLE  
# ENTER

SURFEND

CS BIT14  
MASK MARKSTAT  
AD BIT14  
TS MARKSTAT

# SET BIT14 TO SHOW MARK END

SURFAGAN

CA CURSOR  
INDEX MKDEX  
TS 1  
CA SPIRAL  
INDEX MKDEX  
TS 3# HOLDS VAC AREA POINTER FOR SURF MARKING  
# STORE CURSOR SP 2COMP  
  
# STORE SPIRALCS MARKSTAT  
MASK BIT14  
EXTEND

# IF BIT 14 SET -- END MARKING

BZF MARKCHEX  
CA MARKCNTR  
AD ONE  
COM  
AD FIVE  
EXTEND

# THIS IS RECYCLE -- SEE IF 5 MARKS ALREADY

BZMF 5MKALARM  
INCR MARKCNTR  
TCF GETMKS +3# CAN'T RECYCLE -- TOO MANY MARKS -- ALARM  
# OF FOR RECYCLE -- INCR COUNTER  
# GO DISPLAY MARK VB

1.7 lem geometry

101

$$\langle lem\ geometry\ 101 \rangle \equiv$$
$$\langle page\ LM0320\ 102 \rangle$$
$$\langle page\ LM0321\ 103 \rangle$$
$$\langle page\ LM0322\ 104 \rangle$$
$$\langle page\ LM0323\ 105 \rangle$$
$$\langle page\ LM0324\ 106 \rangle$$
$$\langle page\ LM0325\ 107a \rangle$$

(7)

102  $\langle$ page LM0320 102 $\rangle=$ 

(101 745)

BANK 23  
 SETLOC LEMGEOM  
 BANK

SBANK= LOWSUPER  
 EBANK= XSM

# THESE TWO ROUTINES COMPUTE THE ACTUAL STATE VECTOR FOR LM,CSM BY ADDING  
 # THE CONIC R,V AND THE DEVIATIONS R,V. THE STATE VECTORS ARE CONVERTED TO  
 # METERS B-29 AND METERS/CSEC B-7 AND STORED APPROPRIATELY IN RN,VN OR  
 # R-OTHER,V-OTHER FOR DOWNLINK. THE ROUTINES NAMES ARE SWITCHED IN THE  
 # OTHER VEHICLES COMPUTER.

#

# INPUT

# STATE VECTOR IN TEMPORARY STORAGE AREA  
 # IF STATE VECTOR IS SCALED POS B27 AND VEL B5  
 # SET X2 TO +2  
 # IF STATE VECTOR IS SCALED POS B29 AND VEL B7  
 # SET X2 TO 0

#

# OUTPUT

# R(T) IN RN, V(T) IN VN, T IN PIPTIME

# OR

# R(T) IN R-OTHER, V(T) IN V-OTHER (T IS DEFINED BY T-OTHER)

                  COUNT\* \$\$/GEOM  
 SVDWN2 BOF RVQ # SW=1=AVETOMID DOING W-MATRIX INTEG.  
                   AVEMIDSW  
                   +1  
                   VLOAD VSL\*  
                   TDELTA  
                   0 -7,2  
                   VAD VSL\*  
                   RCV  
                   0,2  
                   STOVL RN  
                   TNUV  
                   VSL\* VAD  
                   0 -4,2  
                   VCV  
                   VSL\*  
                   0,2  
                   STODL VN  
                   TET  
                   STORE PIPTIME

RVQ

103

$\langle$ page LM0321 103 $\rangle \equiv$

SVDWN1

(101 745)

VLOAD

VSL\*

TDELTA

0

-7,2

VAD

VSL\*

RCV

0,2

STOVL

R-OTHER

TNUV

VSL\*

VAD

0

-4,2

VSL\*

VCV

0,2

STORE

V-OTHER

RVQ

104  $\langle$ page LM0322 104 $\rangle=$ 

(101 745)

```

# THE FOLLOWING ROUTINE TAKES A HALF UNIT TARGET VECTOR REFERRED TO NAV BASE COORDINATE
# GIMBAL ORIENTATIONS AT WHICH THE RR MIGHT SIGHT THE TARGET.  THE GIMBAL ANGLES CORRESPONDING
# ARE LEFT IN MODEA AND THOSE WHICH WOULD BE USED AFTER A REMODE IN MODEB.  THIS ROUTINE
# ANGLE LESS THAN 90 DEGS IN ABS VALUE WITH ARBITRARY SHAFT, WITH A CORRESPONDING DEGREE
# SELECTION AND LIMIT CHECKING ARE DONE ELSEWHERE.
#
# THE MODE 1 CONFIGURATION IS CALCULATED FROM THE VECTOR AND THEN MODE 2 IS FOUND USING
#
#       S(2) = 180 + S(1)
#       T(2) = 180 - T(1)
#
# THE VECTOR ARRIVES IN MPAC WHERE TRG*SMNG OR *SMNB* WILL HAVE LEFT IT.

```

```

RRANGLES      STORE  32D
               DLOAD  DCOMP      # SINCE WE WILL FIND THE MODE 1 SHAFT
               34D      # ANGLE LATER, WE CAN FIND THE MODE 1
               SETPD  ASIN      # TRUNNION BY SIMPLY TAKING THE ARCSIN OF
               0        # THE Y COMPONENT, THE ASIN GIVIN AN
               PUSH   BDSU      # ANSWER WHOSE ABS VAL IS LESS THAN 90 DEG.
               LODPHALF
               STODL   4        # MODE 2 TRUNNION TO 4.

               L06ZEROS
               STOVL   34D      # UNIT THE PROJECTION OF THE VECTOR
               32D      # IN THE X-Z PLANE
               UNIT    BOVB      # IF OVERFLOW, TARGET VECTOR IS ALONG Y
               LUNDESCH  # CALL FOR MANEUVER UNLESS ON LUNAR SURF
               STODL   32D      # PROJECTION VECTOR.
               32D
               SR1     STQ
               S2
               STODL   SINTH      # USE ARCTRIG SINCE SHAFT COULD BE ARB.
               36D
               SR1
               STCALL  COSTH
               ARCTRIG

```



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105	$\langle page\ LM0323\ 105 \rangle \equiv$		(101 745)
	PUSH	DAD	# MODE 1 SHAFT TO 2.
		LODPHALF	
	STOVL	6	
		4	
	RTB		# FIND MODE 2 CDU ANGLES.
		2V1STO2S	
	STOVL	MODEB	
		0	
	RTB		# MODE 1 ANGLES TO MODE A.
		2V1STO2S	
	STORE	MODEA	
	EXIT		
	CS	RADMODES	# SWAP MODEA AND MODEB IF RR IN MODE 2.
	MASK	ANTENBIT	
	CCS	A	
	TCF	+4	
	DXCH	MODEA	
	DXCH	MODEB	
	DXCH	MODEA	
	TC	INTPRET	
	GOTO		
		S2	

106  $\langle$ page LM0324 106 $\rangle \equiv$ 

(101 745)

# GIVEN RR TRUNNION AND SHAFT (T,S) IN TANGNB,+1, FIND THE ASSOCIATED  
 # LINE OF SIGHT IN NAV BASE AXES. THE HALF UNIT VECTOR,  $.5(\sin(S)\cos(T)$ ,  
 #  $-\sin(T), \cos(S)\cos(T))$  IS LEFT IN MPAC AND 32D.

SETLOC INFLIGHT  
 BANK

COUNT\* \$\$/GEOM

RRNB

SLOAD RTB  
 TANGNB  
 CDULOGIC

SETPD PUSH # TRUNNION ANGLE TO 0  
 0

SIN DCOMP  
 STODL 34D # Y COMPONENT

COS PUSH # .5 COS(T) TO 0  
 SLOAD RTB

TANGNB +1  
 CDULOGIC

RRNB1

PUSH COS # SHAFT ANGLE TO 2

DMP SL1  
 0

STODL 36D # Z COMPONENT

SIN DMP

SL1

STOVL 32D

32D

RVQ

# THIS ENTRY TO RRNB REQUIRES THE TRUNNION AND SHAFT ANGLES IN MPAC AND MPAC +1 RESP

RRNBMPAC

STODL 20D # SAVE SHAFT CDU IN 21.

MPAC # SET MODE TO DP. (THE PRECEEDING STORE  
 # MAY BE DP, TP OR VECTOR.)

RTB SETPD  
 CDULOGIC  
 0

PUSH SIN # TRUNNION ANGLE TO 0

DCOMP

STODL 34D # Y COMPONENT

COS PUSH # .5COS(T) TO 0

SLOAD RTB # PICK UP CDU'S.

21D  
CDULOGIC  
GOTO  
RRNB1

107a     $\langle \textit{page LM0325 107a} \rangle \equiv$  (101 745)  
      # (This page has nothing on it.)

1.8    r63 routine

107b     $\langle \textit{r63 routine 107b} \rangle \equiv$  (7)  
       $\langle \textit{Page LM0338 108} \rangle$   
       $\langle \textit{Page LM0339 110} \rangle$   
       $\langle \textit{Page LM0340 111a} \rangle$   
       $\langle \textit{Page LM0341 111b} \rangle$

108    *(Page LM0338 108)*≡ (107b 760)

```

# SUBROUTINE NAME:      V89CALL
# MOD NO:              0          DATE:          9 JAN 1968
# MOD BY:              DIGITAL DEVEL GROUP    LOG SECTION:  R63
#
# FUNCTIONAL DESCRIPTION:
#
# CALLED BY VERB 89 ENTER DURING P00.  Prio 10 USED.  CALCULATES AND
# DISPLAYS FINAL FDAI BALL ANGLES TO POINT LM +X OR +Z AXIS AT CSM.
#
# 1. KEY IN V 89 E ONLY IF IN PROG 00.  IF NOT IN P00, OPERATOR ERROR AND
# EXIT R63, OTHERWISE CONTINUE.
#
# 2. IF IN P00, DO IMU STATUS CHECK ROUTINE (R02BOTH).  IF IMU ON AND ITS
# ORIENTATION KNOWN TO LGC, CONTINUE.
#
# 3. FLASH DISPLAY V 04 N 06.  R2 INDICATES WHICH SPACECRAFT AXIS IS TO
# BE POINTED AT CSM.  INITIAL CHOICE IS PREFERRED (+Z) AXIS (R2=1).
# ASTRONAUT CAN CHANGE TO (+X) AXIS (R2 NOT =1) BY V 22 E 2 E.  CONTINUE
# AFTER KEYING IN PROCEED.
#
# 4. BOTH VEHICLE STATE VECTORS UPDATED BY CONIC EQS.
#
# 5. HALF MAGNITUDE UNIT LOS VECTOR (IN STABLE MEMBER COORDINATES) AND
# HALF MAGNITUDE UNIT SPACECRAFT AXIS VECTOR (IN BODY COORDINATES)
# PREPARED FOR VECPOINT.
#
# 6. GIMBAL ANGLES FROM VECPOINT TRANSFORMED INTO FDAI BALL ANGLES BY
# BALLANGS.  FLASH DISPLAY V 06 N 18 AND AWAIT RESPONSE.
#
# 7      RECYCLE -- RETURN TO STEP 4.
#      TERMINATE -- EXIT R63
#      PROCEED -- RESET 3AXISFLAG AND CALL R60LEM FOR ATTITUDE MANEUVER.
#
# CALLING SEQUENCE:      V 89 E.
#
# SUBROUTINES CALLED:    CHECKPOOH, R02BOTH, GOXDSPF, CSMCONIC, LEMCONIC,
#                        VECPOINT, BALLANGS, R60LEM.
#
# NORMAL EXIT MODES:      TC ENDEXT
#
# ALARMS:                1. OPERATOR ERROR IF NOT IN P00.
#                        2. PROGRAM ALARM IF IMU IS OFF.
#                        3. PROGRAM ALARM IF IMU ORIENTATION IS UNKNOWN.
#
# OUTPUT:                NONE

```

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```
#
# ERASABLE INITIALIZATION REQUIRED:  NONE
#
# DEBRIS:      OPTION1, +1, TDEC1, PCINTVSM, SCAXIS, CPHI, CTHETA, CPSI,
```

110 (Page LM0339 110)≡

(107b 760)

```

# 3AXISFLAG.

EBANK= RONE
BANK 32
SETLOC BAWLANGS
BANK

COUNT* $$/R63
V89CALL TC BANKCALL # IMU STATUS CHECK. RETURNS IF ORIENTATION
CADR R02BOTH # KNOWN. ALARMS IF NOT.
CAF THREE # ALLOW ASTRONAUT TO SELECT DESIRED
TS OPTIONX # TRACKING ATTITUDE AXIS.
CAF ONE
TS OPTIONX +1
CAF VB04N12 # V 04 N 12.
TC BANKCALL
CADR GOFLASH
TC ENDEXT # TERMINATE
TC +2 # PROCEED
TC -5 # DATA IN. OPTION1+1 = 1 FOR Z AXIS
V89RECL TC INTERPRET # 2 FOR X AXIS
RTB DAD
LOADTIME # READ PRESENT TIME
DP1MIN
STORE TSTART82 # SAVE TIME FOR LEMCONIC CALL
STCALL TDEC1 # STORE TIME FOR CSMCONIC CALL
CSMCONIC # CSM STATE VECTOR UPDATE
VLOAD # CSMCONIC LEFT R VECTOR IN RATT
RATT
STODL RONE # SAVE FOR LINE OF SIGHT (LOS) COMPUTATION
TSTART82
STCALL TDEC1 # STORE TIME FOR LEMCONIC CALL
LEMCONIC # LEM STATE VECTOR UPDATE
VLOAD VSU # CSM POSITION -- LEM POSITION -- LOS
RONE # LOS VECTOR LEFT IN MPAC
RATT
MXV RTB # (REFSMAT X LOS). TRANSFORMS LOS FROM
REFSMAT # REFERENCE COORD TO STAB MEMB COORD.
NORMUNIT
STORE POINTVSM # STORE LOS FOR VECPOINT CALCULATION
EXIT
CS OPTIONX +1 # 1 FOR Z AXIS. 2 FOR X AXIS.
AD ONE
EXTEND
BZF ALINEZ

```

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	ALINEX	TC	INTPRET	# X AXIS ALIGNMENT
		VLOAD		
			UNITX	# READ (.5, 0, 0)
111a	(Page LM0340 111a)≡			(107b 760)
	V89CALL1	STCALL	SCAXIS	# STORE SELECTED ALIGNMENT AXIS
			VECPOINT	# PUTS DESIRED GIM ANG (OG,IG,MG) IN TMPAC
		STORE	CPHI	# STOR GIMBAL ANGLES FOR BALLANGS CALL
		EXIT		
		TC	BANKCALL	
		CADR	BALLANGS	# PUTS DESIRED BALL ANGLE IN FDAIX,Y,Z
		CAF	VB06N18	# V 06 N 18
		TC	BANKCALL	# NOUN 18 REFERS TO FDAIX,Y,Z
		CADR	GOFLASH	
		TC	ENDEXT	# TERMINATE
		TC	+2	# PROCEED
		TC	V89RECL	# RECYCLE
		TC	DOWNFLAG	# RESET 3 AXIS FLAG
		ADRES	3AXISFLG	# RESET BIT6 FLAG WORD 5
		TC	BANKCALL	# PERFORMS LEM MANEUVER TO ALIGN SELECTED
		CADR	R60LEM	# SPACECRAFT AXIS TO CSM.
		TCF	ENDEXT	# TERMINATE R63
	ALINEZ	TC	INTPRET	# Z AXIS ALIGNMENT
		VLOAD	GOTO	
			UNITZ	# READ (0, 0, .5)
			V89CALL1	
	VB04N12	VN	412	
	VB06N18	VN	0618	
111b	(Page LM0341 111b)≡			(107b 760)
	DP1MIN	2DEC	6000	

## 1.9 attitude maneuver routine

$$\begin{aligned}
 112 \quad \langle \textit{attitude maneuver routine } 112 \rangle \equiv & \quad (7) \\
 & \langle \textit{Page LM0342 } 113 \rangle \\
 & \langle \textit{Page LM0343 } 115 \rangle \\
 & \langle \textit{Page LM0344 } 117 \rangle \\
 & \langle \textit{Page LM0345 } 119 \rangle \\
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 \end{aligned}$$



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# BLOCK 2 LGC ATTITUDE MANEUVER ROUTINE -- KALCMANU

#

# MOD 2                    DATE 5/1/67            BY DON KEENE

#

# PROGRAM DESCRIPTION

#

# KALCMANU IS A ROUTINE WHICH GENERATES COMMANDS FOR THE LM DAP TO CHANGE THE ATTITUDE OF THE S/C DURING FREE FALL. IT IS DESIGNED TO MANEUVER THE SPACECRAFT FROM ITS INITIAL ORIENTATION TO THE ORIENTATION SPECIFIED BY THE PROGRAM WHICH CALLS KALCMANU, AVOIDING GIMBAL LOCK IN THE PROCESS. MOD 2 VERSION, THIS DESIRED ATTITUDE IS SPECIFIED BY A SET OF THREE COMMANDED CDU ANGLES (CPhi, CTheta, CPSI) SINGLE PRECISION ANGLES IN THE THREE CONSECUTIVE LOCATIONS, CPhi, CTheta, CPSI, WHERE

#

#                    CPhi = COMMANDED OUTER GIMBAL ANGLE  
 #                    CTheta = COMMANDED INNER GIMBAL ANGLE  
 #                    CPSI = COMMANDED MIDDLE GIMBAL ANGLE

#

# WHEN POINTING A SPACECRAFT AXIS (I.E., X, Y, Z, THE AOT, THRUST AXIS, ETC.) THE SUBROUTINE VELOC USED TO GENERATE THIS SET OF DESIRED CDU ANGLES (SEE DESCRIPTION IN R60).

#

# WITH THIS INFORMATION KALCMANU DETERMINES THE DIRECTION OF THE SINGLE EQUIVALENT ROTATION (COF) AND MAGNITUDE OF THE ROTATION (AM) TO BRING THE S/C FROM ITS INITIAL ORIENTATION TO ITS FINAL ORIENTATION. THIS DIRECTION REMAINS FIXED BOTH IN INERTIAL COORDINATES AND IN COMMANDED S/C AXES THROUGHOUT

#

# MANEUVER. ONCE COF AND AM HAVE BEEN DETERMINED, KALCMANU THEN EXAMINES THE MANEUVER TO SEE IF IT

#

# THE S/C THROUGH GIMBAL LOCK. IF SO, COF AND AM ARE READJUSTED SO THAT THE S/C WILL JUST SKIM THE LOCK ZONE AND ALIGN THE X-AXIS. IN GENERAL A FINAL YAW ABOUT X WILL BE NECESSARY TO COMPLETE THE MANEUVER. NEEDLESS TO SAY, NEITHER THE INITIAL NOR THE FINAL ORIENTATION CAN BE IN GIMBAL LOCK.

#

# FOR PROPER ATTITUDE CONTROL THE DIGITAL AUTOPILOT MUST BE GIVEN AN ATTITUDE REFERENCE WHICH IS THE DESIRED ORIENTATION. KALCMANU DOES THIS BY GENERATING A REFERENCE OF DESIRED GIMBAL ANGLES (CDUXD, CDUYD, CDUZD) WHICH IS USED EVERY ONE SECOND DURING THE MANEUVER. TO ACHIEVE A SMOOTHER SEQUENCE OF COMMANDS BETWEEN SUCH MANEUVERS THE PROGRAM ALSO GENERATES A SET OF INCREMENTAL CDU ANGLES (DELDCDU) TO BE ADDED TO CDU DESIRED ANGLES BY THE AUTOPILOT. KALCMANU ALSO CALCULATES THE COMPONENT MANEUVER RATES (OMEGAPD, OMEGAQD, OMEGARD) WHICH

#

# BE DETERMINED SIMPLY BY MULTIPLYING COF BY SOME SCALAR (ARATE) CORRESPONDING TO THE DESIRED RATE OF CHANGE OF THE

#

# AUTOMATIC MANEUVERS ARE TIMED WITH THE HELP OF WAITLIST SO THAT AFTER A SPECIFIED INTERVAL THE DESIRED RATES ARE SET TO ZERO AND THE DESIRED CDU ANGLES (CDUYD, CDUZD) ARE SET EQUAL TO THE DESIRED ANGLES (CTheta, CPSI). IF ANY YAW REMAINS DUE TO GIMBAL LOCK AVOIDANCE, THE FINAL YAW MANEUVER IS CALCULATED AND THE DESIRED YAW RATE SET TO SOME FIXED VALUE (ROLLRATE = + OR - 2 DEGREES PER SECOND). IN THIS CASE ONLY AN INCREMENTAL CDUX ANGLE (DELFROLL) IS SUPPLIED TO THE DAP. AT THE END OF THE MANEUVER OR IN THE EVENT THAT THERE WAS NO FINAL YAW, CDUXD IS SET EQUAL TO CPhi AND THE X-AXIS YAW RATE SET TO ZERO. THUS, UPON COMPLETION OF THE MANEUVER THE S/C WILL FINISH UP IN A LIMIT CYCLE WITH THE DESIRED GIMBAL ANGLES.

```
#  
# PROGRAM LOGIC FLOW  
#  
# KALCMANU IS CALLED AS A HIGH PRIORITY JOB WITH ENTRY POINTS AT KALCMAN3 AND VECPOINT3  
# UP THE CURRENT CDU ANGLES TO BE USED AS THE BASIS FOR ALL COMPUTATIONS INVOLVING THE
```

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```

# IT THEN DETERMINES THE DIRECTION COSINE MATRICES RELATING BOTH THE INITIAL AND FINAL S/C ORIE
#
#      *      *
# MEMBER AXES (MIS,MFS). IT ALSO COMPUTES THE MATRIX RELATING FINAL S/C AXES TO INITIAL S/C AX
# ANGLE OF ROTATION (AM) IS THEN EXTRACTED FROM THIS MATRIX, AND TEST ARE MADE TO DETERMINE IF
#
#      A)      AM LESS THAN .25 DEGREES (MINANG)
#      B)      AM GREATER THAN 170 DEGREES (MAXANG)
#
# IF AM IS LESS THAN .25 DEGREES, NO COMPLICATED AUTOMATIC MANEUVERING IS NECESSARY. THEREFORE
# SET CDU DESIRED EQUAL TO THE FINAL CDU DESIRED ANGLES AND TERMINATE THE JOB.
#
# IF AM IS GREATER THAN .25 DEGREES BUT LESS THAN 170 DEGREES THE AXES OF THE SINGLE EQUIVALENT
#
#      -      *
# (COF) IS EXTRACTED FROM THE SKEW SYMMETRIC COMPONENTS OF MFI.
#
#
#      *      *
# IF AM GREATER THAN 170 DEGREES AN ALTERNATE METHOD EMPLOYING THE SYMMETRIC PART OF MFI (MFI SY
#
#      -
# TO DETERMINE COF.
#
#
# THE PROGRAM THEN CHECKS TO SEE IF THE MANEUVER AS COMPUTED WILL BRING THE S/C THROUGH GIMBAL
# SO, A NEW MANEUVER IS CALCULATED WHICH WILL JUST SKIM THE GIMBAL LOCK ZONE AND ALIGN THE S/C
# METHOD ASSURES THAT THE ADDITIONAL MANEUVERING TO AVOID GIMBAL LOCK WILL BE KEPT TO A MINIMUM
# P AXIS YAW WILL BE NECESSARY, A SWITCH IS RESET (STATE SWITCH 31) TO ALLOW FOR THE COMPUTATIO
# YAW.
#
#
# AS STATED PREVIOUSLY, KALCMANU GENERATES A SEQUENCE OF DESIRED GIMBAL ANGLES WHICH ARE UPDATE
#
# SECOND. THIS IS ACCOMPLISHED BY A SMALL ROTATION OF THE DESIRED S/C FRAME ABOUT THE VECTOR C
# DESIRED REFERENCE MATRIX IS THEN,
#
#      *      *      *
#      MIS      =      MIS      DEL
#      N+1      N
#
#      *
# WHERE DEL IS THE MATRIX CORRESPONDING TO THIS SMALL ROTATION. THE NEW CDU ANGLES CAN THEN BE
#
#      *
# FROM MIS.
#
#
# AT THE BEGINNING OF THE MANEUVER THE AUTOPILOT DESIRED RATES (OMEGAPD, OMEGAQD, OMEGARD) AND
# MANEUVER TIMINGS ARE ESTABLISHED. ON THE FIRST PASS AND ON ALL SUBSEQUENT UPDATES THE CDU DE
# ANGLES ARE LOADED WITH THE APPROPRIATE VALUES AND THE INCREMENTAL CDU ANGLES ARE COMPUTED. T
# (TIME1 AND TIME2) ARE THEN CHECKED TO SEE IF THE MANEUVER WILL TERMINATE BEFORE THE NEXT UPDA
# NOT, KALCMANU CALLS FOR ANOTHER UPDATE (RUN AS A JOB WITH PRIORITY TBD) IN ONE SECOND. ANY D
# CALLING SEQUENCE ARE AUTOMATICALLY COMPENSATED IN CALLING FOR THE NEXT UPDATE.
#
#
# IF IT IS FOUND THAT THE MANEUVER IS TO TERMINATE BEFORE THE NEXT UPDATE A ROUTINE IS CALLED (

```

# LIST TASK) TO STOP THE MANEUVER AT THE APPROPRIATE TIME AS EXPLAINED ABOVE.

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# CALLING SEQUENCE

#

# IN ORDER TO PERFORM A KALCMANU SUPERVISED MANEUVER, THE COMMANDED GIMBAL ANGLES MUST BE PRECO  
 # STORED IN LOCATIONS CPHI, CTHETA, CPSI. THE USER'S PROGRAM MUST THEN CLEAR STATE SWITCH NO 3  
 # ATTITUDE MANEUVER ROUTINE TO PERFORM ANY FINAL P-AXIS YAW INCURRED BY AVOIDING GIMBAL LOCK.  
 # THEN INITIATED BY ESTABLISHING THE FOLLOWING EXECUTIVE JOB

#

\*

# CAF PRIO XX

#

--

# INHINT

# TC FINDVAC

# 2CADR KALCMAN3

# RELINT

#

# THE USER'S PROGRAM MAY EITHER CONTINUE OR WAIT FOR THE TERMINATION OF THE MANEUVER. IF THE U  
 # WAIT, HE MAY PUT HIS JOB TO SLEEP WITH THE FOLLOWING INSTRUCTIONS:

#

# L TC BANKCALL

# L+1 CADR ATTSTALL

# L+2 (BAD RETURN)

# L+3 (GOOD RETURN)

#

# UPON COMPLETION OF THE MANEUVER, THE PROGRAM WILL BE AWAKENED AT L+3 IF THE MANEUVER WAS COMPE  
 # SUCCESSFULLY, OR AT L+2 IF THE MANEUVER WAS ABORTED. THIS ABORT WOULD OCCUR IF THE INITIAL C  
 # WAS IN GIMBAL LOCK.

#

# \*\*\* NOTA BENE \*\*\* IF IT IS ASSUMED THAT THE DESIRED MANEUVERING RATE (0.5, 2, 5, 10 DEG/SEC)  
 # KEYBOARD ENTRY PRIOR TO THE EXECUTION OF KALCMANU.

#

# IT IS ALSO ASSUMED THAT THE AUTOPILOT IS IN THE AUTO MODE. IF THE MODE SWITCH IS CHANGED DUR  
 # MANEUVER, KALCMANU WILL TERMINATE VIA GOODEND WITHIN 1 SECOND SO THAT R60 MAY REQUEST A TRIM  
 # SUBROUTINES.

#

# KALCMANU USES A NUMBER OF INTERPRETIVE SUBROUTINES WHICH MAY BE OF GENERAL INTEREST. SINCE T  
 # WERE PROGRAMMED EXCLUSIVELY FOR KALCMANU, THEY ARE NOT, AS YET, GENERALLY AVAILABLE FOR USE B

#

# MXM3

# ----

#

# THIS SUBROUTINE MULTIPLIES TWO 3X3 MATRICES AND LEAVES THE RESULT IN THE FIRST 18 LOCATIONS C  
 # DOWN LIST, I.E.,

# [ M M M ]

# [ 0 1 2 ]

# \* [ ]

# M = [ M M M ] = M1 X M2

```
#           [ 3      4      5 ]
#           [                ]
#           [ M      M      M ]
#           [ 6      7      8 ]
```

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```

#
# INDEX REGISTER X1 MUST BE LOADED WITH THE COMPLEMENT OF THE STARTING ADDRESS FOR M1, AND X2 M
#
# LOADED WITH THE COMPLEMENT OF THE STARTING ADDRESS FOR M2.  THE ROUTINE USES THE FIRST 20 LOC
# DOWN LIST.  THE FIRST ELEMENT OF THE MATRIX APPEARS IN PDO.  PUSH UP FOR M .
#
# TRANSPOS
# -----
#
# THIS ROUTINE TRANSPOSES A 3X3 MATRIX AND LEAVES THE RESULT IN THE PUSH DOWN LIST, I.E.,
#
#      *      * T
#      M      =      M1
#
# INDEX REGISTER X1 MUST CONTAIN THE COMPLEMENT OF THE STARTING ADDRESS FOR M1.  PUSH UP FOR TH
#
# SEQUENT COMPONENTS OF M.  THIS SUBROUTINE ALSO USES THE FIRST 20 LOCATIONS OF THE PUSH DOWN L
#
# CDU TO DCM
# -----
#
# THIS SUBROUTINE CONVERTS THREE CDU ANGLES IN T(MPAC) TO A DIRECTION COSINE MATRIX (SCALED BY
# THE CORRESPONDING S/C ORIENTATIONS TO THE STABLE MEMBER FRAME.  THE FORMULAS FOR THIS CONVERS
#
#      M      =      COSY COSZ
#      0
#
#      M      =      -COSY SINZ COSX + SINY SINX
#      1
#
#      M      =      COSY SINZ SINX + SINY COSX
#      2
#
#      M      =      SINZ
#      3
#
#      M      =      COSZ COSX
#      4
#
#      M      =      -COSZ SINX
#      5
#
#      M      =      -SINY COSZ
#      6
#

```

$$\begin{array}{l} \# \\ \# \end{array} \quad \begin{array}{l} M \\ 7 \end{array} \quad = \quad \text{SINY SINZ COSX} + \text{COSY SINX}$$



```

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#      M      =      -SINY SINZ SINX + COSY COSX
#      8
#
# WHERE      X      =      OUTER GIMBAL ANGLE
#            Y      =      INNER GIMBAL ANGLE
#            Z      =      MIDDLE GIMBAL ANGLE
#
# THE INTERPRETATION OF THIS MATRIX IS AS FOLLOWS:
#
# IF A , A , A REPRESENT THE COMPONENTS OF A VECTOR IN S/C AXES THEN THE COMPONENTS OF THE SAME
#      X      Y      Z
# STABLE MEMBER AXES (B , B , B ) ARE
#      X      Y      Z
#
#      [ B ]      [ A ]
#      [ X ]      [ X ]
#      [   ]      [   ]
#      [ B ]      *      [ A ]
#      [ Y ]      =      M      [ Y ]
#      [   ]      [   ]
#      [ B ]      [ B ]
#      [ Z ]      [ Z ]
#
# THE SUBROUTINE WILL STORE THIS MATRIX IN SEQUENTIAL LOCATIONS OF ERASABLE MEMORY AS SPECIFIED
#
# PROGRAM. TO DO THIS THE CALLING PROGRAM MUST FIRST LOAD X2 WITH THE COMPLEMENT OF THE STARTING ADDRESS
#
# INTERNALLY, THE ROUTINE USES THE FIRST 16 LOCATIONS OF THE PUSH DOWN LIST, ALSO STEP REGISTER
# REGISTER X2.
#
# DCM TO CDU
# -----
#
# THIS ROUTINE EXTRACTS THE CDU ANGLES FROM A DIRECTION COSINE MATRIX (M SCALED BY 2) RELATING
#
# STABLE MEMBER AXES. X1 MUST CONTAIN THE COMPLEMENT OF THE STARTING ADDRESS FOR M. THE SUBROUTINE
# CORRESPONDING GIMBAL ANGLES IN V(MPAC) AS DOUBLE PRECISION 1'S COMPLEMENT ANGLES SCALED BY 2P
# FOR THIS CONVERSION ARE
#
#      Z      =      ARCSIN (M / 3)
#
#      Y      =      ARCSIN (-M / COSZ)
#
#      6
#

```

```
# IF M IS NEGATIVE, Y IS REPLACED BY PI SGN Y - Y.  
#      0
```



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# [ Y X ]  
#

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```

#
#      -
#      U      =      UNIT ROTATION VECTOR RESOLVED INTO S/C AXES.
#      A      =      ROTATION ANGLE
#
#
#      *
# THE INTERPRETATION OF DEL IS AS FOLLOWS:
#
# IF A , A , A REPRESENT THE COMPONENTS OF A VECTOR IN THE ROTATED FRAME, THEN THE COMPONENTS
# X Y Z
# VECTOR IN THE ORIGINAL S/C AXES (B , B , B ) ARE
# X Y Z
#
#
#      [ B ]      [ A ]
#      [ X ]      [ X ]
#      [   ]      [   ]
#      [ B ]      *      [ A ]
#      [ Y ]      = DEL  [ Y ]
#      [   ]      [   ]
#      [ B ]      [ B ]
#      [ Z ]      [ Z ]
#
#
# THE ROUTINE WILL STORE THIS MATRIX (SCALED UNITY) IN SEQUENTIAL LOCATIONS OF ERASABLE MEMORY
#
# THE LOCATION CALLED DEL. IN ORDER TO USE THE ROUTINE, THE CALLING PROGRAM MUST FIRST STORE U
# DOUBLE PRECISION VECTOR) IN THE SET OF ERASABLE LOCATIONS BEGINNING WITH THE ADDRESS CALLED C
# MUST THEN BE LOADED INTO D(MPAC).
#
# INTERNALLY, THE PROGRAM ALSO USES THE FIRST 10 LOCATIONS OF THE PUSH DOWN LIST.
#
# READCDUK
# -----
#
# THIS BASIC LANGUAGE SUBROUTINE LOADS T(MPAC) WITH THE THREE CDU ANGLES.
#
# SIGNMPAC
# -----
#
# THIS IS A BASIC LANGUAGE SUBROUTINE WHICH LIMITS THE MAGNITUDE OF D(MPAC) TO + OR - DPOSMAX C
#
# PROGRAM STORAGE ALLOCATION
#
#      1)      FIXED MEMORY      1059 WORDS
#      2)      ERASABLE MEMORY   98
#      3)      STATE SWITCHES    3

```

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# 4) FLAGS 1

#

# JOB PRIORITIES

#

# 1) KALCMANU TBD

# 2) ONE SECOND UPDATE TBD

#

# SUMMARY OF STATE SWITCHES AND FLAGWORDS USED BY KALCMANU.

#

STATE SWITCH NO.	FLAGWRD 2 BIT NO.	SETTING	MEANING
---------------------	----------------------	---------	---------

#

# \*

31	14	0	MANEUVER WENT THROUGH GIMBAL
		1	MANEUVER DID NOT GO THROUGH C

#

# \*

32	13	0	CONTINUE UPDATE PROCESS
----	----	---	-------------------------

		1	START UPDATE PROCESS
--	--	---	----------------------

#

33	12	0	PERFORM FINAL P AXIS YAW IF P
		1	IGNORE ANY FINAL P-AXIS YAW

#

34	11	0	SIGNAL END OF KALCMANU
----	----	---	------------------------

		1	KALCMANU IN PROCESS. USER
--	--	---	---------------------------

#

# \* INTERNAL TO KALCMANU

#

# SUGGESTIONS FOR PROGRAM INTEGRATION

#

# THE FOLLOWING VARIABLES SHOULD BE ASSIGNED TO UNSWITCH ERASABLE:

#

# CPHI

# CTHETA

# CPSI

# POINTVSM +5

# SCAXIS +5

# DELDCDU

# DELDCDU1

# DELDCDU2

# RATEINDX

#

# THE FOLLOWING SUBROUTINES MAY BE PUT IN A DIFFERENT BANK

#

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$\rangle \equiv$

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#

TRANSPGS

#

SIGNMPAC

#

READCDUK

#

CDUTODCM

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BANK 15  
 SETLOC KALCMON1  
 BANK

EBANK= BCDU

# THE THREE DESIRED CDU ANGLES MUST BE STORED AS SINGLE PRECISION TWO'S COMPLEMENT AND  
 # LOCATIONS, CPHI, CTHETA, CPSI.

KALCMAN3	COUNT* \$\$/KALC		
	TC INTPRET	# PICK UP THE CURRENT CDU ANGLES AND	
	RTB	# COMPUTE THE MATRIX FROM INITIAL S/C	
	READCDUK	# AXES TO FINAL S/C AXES.	
	STORE BCDU	# STORE INITIAL S/C ANGLES	
	SLOAD ABS	# CHECK THE MAGNITUDE OF THE DESIRED	
	CPSI	# MIDDLE GIMBAL ANGLE	
	DSU BPL		
	LOCKANGL	# IF GREATER THAN 70 DEG ABORT MANEUVER	
	TOOBADF		
	AXC,2 TLOAD		
	MIS		
	BCDU		
	CALL	# COMPUTE THE TRANSFORMATION FROM INITIAL	
	CDUTODCM	# S/C AXES TO STABLE MEMBER AXES	
	AXC,2 TLOAD		
	MFS	# PREPARE TO CALCULATE ARRAY MFS	
	CPHI		
	CALL		
	CDUTODCM		
SECAD	AXC,1 CALL	# MIS AND MFS ARRAYS CALCULATED	\$2
	MIS		
	TRANSPOS		
	VLOAD STADR		
	STOVL TMIS +12D		
	STADR		
	STOVL TMIS +6		
	STADR		
	STORE TMIS	# TMIS = TRANSPOSE(MIS) SCALED BY 2	
	AXC,1 AXC,2		
	TMIS		
	MFS		
	CALL		
	MXM3		
	VLOAD STADR		
	STOVL MFI +12D		



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```
STADR
STOVL  MFI +6
STADR
STORE  MFI      # MFI = TMIS MFS (SCALED BY 4)
SETPD  CALL     # TRANSPOSE MFI IN PD LIST
```

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```

                                18D
                                TRNSPSPD
VLOAD      STADR
STOVL      TMFI      +12D
STADR
STOVL      TMFI      +6
STADR
STORE      TMFI      # TMFI = TRANSPOSE (MFI) SCALED BY 4

# CALCULATE COFSKEW AND MFISYM

DLOAD      DSU
            TMFI      +2
            MFI      +2
PDDL      DSU      # CALCULATE COF SCALED BY 2/SIN(AM)
            MFI      +4
            TMFI      +4
PDDL      DSU
            TMFI      +10D
            MFI      +10D
VDEF
STORE      COFSKEW      # EQUALS MFISKEW

# CALCULATE AM AND PROCEED ACCORDING TO ITS MAGNITUDE

DLOAD      DAD
            MFI
            MFI      +16D
DSU      DAD
            DP1/4TH
            MFI      +8D
STORE      CAM      # CAM = (MFIO+MFI4+MFI8-1)/2 HALF SCALE
ARCCOS
STORE      AM      # AM=ARCCOS(CAM)      (AM SCALED BY 2)
DSU      BPL
            MINANG
            CHECKMAX

TLOAD      # MANEUVER LESS THAN .25 DEGREES
            CPHI      # GO DIRECTLY INTO ATTITUDE HOLD
STCALL     CDUXD      # ABOUT COMMANDED ANGLES
            TOOBADI    # STOP RATE AND EXIT

CHECKMAX    DLOAD      DSU
            AM
            MAXANG

```

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BPL	VLOAD	
	ALTCALC	# UNIT
	COFSKEW	# COFSKEW
UNIT		
STORE	COF	# COF IS THE MANEUVER AXIS

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```

      GOTO          # SEE IF MANEUVER GOES THRU GIMBAL LOCK
      LOCSKIRT
ALTCALC  VLOAD      VAD          # IF AM GREATER THAN 170 DEGREES
          MFI
          TMFI
          VSR1
          STOVL      MFISYM
          MFI        +6
          VAD        VSR1
          TMFI       +6
          STOVL      MFISYM +6
          MFI        +12D
          VAD        VSR1
          TMFI       +12D
          STORE      MFISYM +12D  # MFISYM=(MFI+TMFI)/2    SCALED BY 4

      # CALCULATE COF

      DLOAD      SR1
          CAM
      PDDL      DSU          # PDO CAM          $4
          DPHALF
          CAM
      BOVB      PDDL          # PS2 1 - CAM          $2
          SIGNMPAC
          MFISYM +16D
      DSU      DDV
          0
          2
      SQRT      PDDL          # COFZ = SQRT(MFISYM8-CAM)/(1-CAM)
          MFISYM +8D          # $ ROOT 2
      DSU      DDV
          0
          2
      SQRT      PDDL          # COFY = SQRT(MFISYM4-CAM)/(1-CAM) $ROOT2
          MFISYM
      DSU      DDV
          0
          2
      SQRT      VDEF          # COFX = SQRT(MFISYM-CAM)/(1-CAM) $ROOT 2
      UNIT
      STORE      COF

      # DETERMINE LARGEST COF AND ADJUST ACCORDINGLY

```

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COFMAXGO	DLOAD	DSU		
		COF		
		COF	+2	
	BMN	DLOAD		# COFY G COFX

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		COMP12		
		COF		
	DSU	BMN		
		COF	+4	
		METHOD3		# COFZ G COFX OR COFY
	GOTO			
		METHOD1		# COFX G COFY OR COFZ
COMP12	DLOAD	DSU		
		COF	+2	
		COF	+4	
	BMN			
		METHOD3		# COFZ G COFY OR COFX
METHOD2	DLOAD	BPL		# COFY MAX
		COFSKEW	+2	# UY
		U2POS		
	VLOAD	VCOMP		
		COF		
	STORE	COF		
U2POS	DLOAD	BPL		
		MFISYM	+2	# UX UY
		OKU21		
	DLOAD	DCOMP		# SIGN OF UX OPPOSITE garbled
		COF		
	STORE	COF		
OKU21	DLOAD	BPL		
		MFISYM	+10D	# UY UZ
		LOCSKIRT		
	DLOAD	DCOMP		# SIGN OF UZ OPPOSITE TO UY
		COF	+4	
	STORE	COF	+4	
	GOTO			
		LOCSKIRT		
METHOD1	DLOAD	BPL		# COFX MAX
		COFSKEW		# UX
		U1POS		
	VLOAD	VCOMP		
		COF		
	STORE	COF		
U1POS	DLOAD	BPL		
		MFISYM	+2	# UX UY
		OKU12		
	DLOAD	DCOMP		
		COF	+2	# SIGN OF UY OPPOSITE TO UX
	STORE	COF	+2	

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```
OKU12      DLOAD  BPL
            MFISYM  +4    # UX UZ
            LOCSKIRT
            DLOAD  DCOMP    # SIGN OF UZ OPPOSITE TO UY
            COF      +4
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            STORE  COF      +4
            GOTO
            LOCSKIRT
METHOD3     DLOAD  BPL      # COFZ MAX
            COFSKEW +4    # UZ
            U3POS
            VLOAD  VCOMP
            COF
            STORE  COF
U3POS       DLOAD  BPL
            MFISYM  +4    # UX UZ
            OKU31
            DLOAD  DCOMP
            COF      # SIGN OF UX OPPOSITE TO UZ
            STORE  COF
OKU31       DLOAD  BPL
            MFISYM  +10D   # UY UZ
            LOCSKIRT
            DLOAD  DCOMP
            COF      +2    # SIGN OF UY OPPOSITE TO UZ
            STORE  COF      +2
            GOTO
            LOCSKIRT
```

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(112 720)

# MATRIX OPERATIONS

BANK 13  
 SETLOC KALCMON2  
 BANK

EBANK= BCDU

MXM3

SETPD VLOAD\*  
 0  
 0,1  
 VXM\* PDVL\*  
 0,2  
 6,1  
 VXM\* PDVL\*  
 0,2  
 12D,1  
 VXM\* PUSH  
 0,2  
 RVQ

# MXM3 MULTIPLIES 2 3X3 MATRICES  
 # AND LEAVES RESULT IN PD LIST  
 # AND MPAC

# RETURN WITH MIXM2 IN PD LIST

TRANSPOS

SETPD VLOAD\*  
 0  
 0,1  
 PDVL\* PDVL\*  
 6,1  
 12D,1

# TRANSPOS TRANSPOSES A 3X3 MATRIX  
 # AND LEAVES RESULT IN PD LIST  
 # MATRIX ADDRESS IN XR1

TRNSPSPD

PUSH  
 EXIT  
 INDEX FIXLOC  
 DXCH 12  
 INDEX FIXLOC  
 DXCH 16  
 INDEX FIXLOC  
 DXCH 12  
 INDEX FIXLOC  
 DXCH 14  
 INDEX FIXLOC  
 DXCH 4  
 INDEX FIXLOC  
 DXCH 14  
 INDEX FIXLOC  
 DXCH 2

# MATRIX IN PD  
 # ENTER WITH MATRIX AT 0 IN PD LIST



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INDEX	FIXLOC
DXCH	6
INDEX	FIXLOC
DXCH	2

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(112 720)

```

TC      INTPRET
RVQ

BANK    15
SETLOC  KALCMON1
BANK

EBANK=  BCDU

MINANG   2DEC    0.00069375

MAXANG   2DEC    0.47222222

# GIMBAL LOCK CONSTANTS

# D = MGA CORRESPONDING TO GIMBAL LOCK = 60 DEGREES
#      NGL = BUFFER ANGLE (TO AVOID DIVISIONS BY ZERO) = 2 DEGREES

SD        2DEC    .433015      # = SIN(D)                $2

K3S1      2DEC    .86603       # = SIN(D)                $1

K4         2DEC    -.25        # = -COS(D)               $2

K4SQ       2DEC    .125        # = COS(D)COS(D)          $2

SNGLCD     2DEC    .008725     # = SIN(NGL)COS(D)        $2

CNGL       2DEC    .499695     # COS(NGL)                 $2

LOCKANGL   DEC     .388889     # = 70 DEGREES

# INTERPRETIVE SUBROUTINE TO READ THE CDU ANGLES

READCDUK   CA      CDUZ      # LOAD T(MPAC) WITH CDU ANGLES
            TS      MPAC      +2
            EXTEND
            DCA     CDUX      # AND CHANGE MODE TO TRIPLE PRECISION
            TCF     TLOAD     +6

CDUTODCM   AXT,1   SSP
            OCT     3
            S1
            OCT     1      # SET XR1, S1, AND PD FOR LOOP
            STORE   7

```

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139

```

      SETPD
      0
LOOPSIN  SLOAD* RTB
          10D,1
          CDULOGIC
```

140  $\langle$ Page LM0358 140 $\rangle \equiv$ 

```

                                (112 720)
STORE 10D # LOAD PD WITH 0 SIN(PHI)
SIN PDDL # 2 COS(PHI)
        10D # 4 SIN(THETA)
COS PUSH # 6 COS(THETA)
TIX,1 DLOAD # 8 SIN(PSI)
        LOOPSIN # 10 COS(PSI)
        6
DMP SL1
        10D
STORE 0,2 # C0 = COS(THETA)COS(PSI)
DLOAD DMP
        4
        0
PDDL DMP # (PD6 SIN(THETA)SIN(PHI))
        6
        8D
DMP SL1
        2
BDSU SL1
        12D
STORE 2,2 # C1=-COS(THETA)SIN(PSI)COS(PHI)
DLOAD DMP
        2
        4
PDDL DMP # (PD7 COS(PHI)SIN(THETA)) SCALED 4
        6
        8D
DMP SL1
        0
DAD SL1
        14D
STORE 4,2 # C2=COS(THETA)SIN(PSI)SIN(PHI)
DLOAD
        8D
STORE 6,2 # C3=SIN(PSI)
DLOAD
        10D
DMP SL1
        2
STORE 8D,2 # C4=COS(PSI)COS(PHI)
DLOAD DMP
        10D
        0
DCOMP SL1
STORE 10D,2 # C5=-COS(PSI)SIN(PHI)

```

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DLOAD	DMP	
	4	
	10D	
DCOMP	SL1	
STORE	12D,2	# C6=-SIN(THETA)COS(PSI)

142 (Page LM0359 142)≡

(112 720)

```

DLOAD
DMP      SL1      # (PUSH UP 7)
          8D
PDDL     DMP      # (PD7 COS(PHI)SIN(THETA)SIN(PSI)) SCALE 4
          6
          0
DAD      SL1      # (PUSH UP 7)
STADR    # C7=COS(PHI)SIN(THETA)SIN(PSI)
STORE    14D,2    # +COS(THETA)SIN(PHI)
DLOAD
DMP      SL1      # (PUSH UP 6)
          8D
PDDL     DMP      # (PD6 SIN(THETA)SIN(PHI)SIN(PSI)) SCALE 4
          6
          2
DSU      SL1      # (PUSH UP 6)
STADR
STORE    16D,2    # C8=-SIN(THETA)SIN(PHI)SIN(PSI)
RVQ      # +COS(THETA)COS(PHI)

# CALCULATION OF THE MATRIX DEL.....
#
#      *      *      --T      *
#      DEL = (IDMATRIX)COS(A)+UU (1-COS(A))+UX SIN(A)          SCALED 1
#
#      -
#      WHERE U IS A UNIT VECTOR (DP SCALED 2) ALONG THE AXIS OF ROTATION.
#      A IS THE ANGLE OF ROTATION (DP SCALED 2)
#
#      UPON ENTRY, THE STARTING ADDRESS OF U IS COF, AND A IS IN MPAC

DELCOMP   SETPD   PUSH      # MPAC CONTAINS THE ANGLE A
          0
          SIN     PDDL      # PD0 = SIN(A)
          COS     PUSH      # PD2 = COS(A)
          SR2     PDDL      # PD2 = COS(A)
          BDSU    BOVB
          DPHALF
          SIGNMPAC
          PDDL      # PDA = 1-COS(A)

# COMPUTE THE DIAGONAL COMPONENTS OF DEL

          COF
DSQ        DMP
          4

```

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DAD	SL3
	2
BOVB	
	SIGNMPAC

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(112 720)

STODL	KEL		# UX UX(1-COS(A))	+COS(A)	\$1
	COF	+2			
DSQ	DMP				
	4				
DAD	SL3				
	2				
BOVB					
	SIGNMPAC				
STODL	KEL	+8D	# UY UY(1-COS(A))	+COS(A)	\$1
	COF	+4			
DSQ	DMP				
	4				
DAD	SL3				
	2				
BOVB					
	SIGNMPAC				
STORE	KEL	+16D	# UZ UZ(1-COS(A))	+COS(A)	\$1

# COMPUTE THE OFF DIAGONAL TERMS OF DEL

DLOAD	DMP				
	COF				
	COF	+2			
DMP	SL1				
	4				
PDDL	DMP		# D6	UX UY (1-COS A)	\$4
	COF	+4			
	0				
PUSH	DAD		# D8	UZ SIN A	\$4
	6				
SL2	BOVB				
	SIGNMPAC				
STODL	KEL	+6			
BDSU	SL2				
BOVB					
	SIGNMPAC				
STODL	KEL	+2			
	COF				
DMP	DMP				
	COF	+4			
	4				
SL1	PDDL		# D6	UX UZ (1-COS A)	\$4
	COF	+2			
DMP	PUSH		# D8	UY SIN(A)	
	0				



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DAD	SL2		
	6		
BOVB			
	SIGNMPAC		
STODL	KEL	+4	# UX UZ (1-COS(A))+UY SIN(A)

146 *(Page LM0361 146)*≡

(112 720)

```

      BDSU      SL2
      BOVB
      SIGNMPAC
      STODL     KEL      +12D    # UX UZ (1-COS(A))-UY SIN(A)
      COF      +2
      DMP       DMP
      COF      +4
      4
      SL1       PDDL      # D6      UY UZ (1-COS(A))          $ 4
      COF
      DMP       PUSH      # D8      UX SIN(A)
      0
      DAD       SL2
      6
      BOVB
      SIGNMPAC
      STODL     KEL      +14D    # UY UZ(1-COS(A)) +UX SIN(A)
      BDSU      SL2
      BOVB
      SIGNMPAC
      STORE     KEL      +10D    # UY UZ (1-COS(A)) -UX SIN(A)
      RVQ

```

```

# DIRECTION COSINE MATRIX TO CDU ANGLE ROUTINE
# X1 CONTAINS THE COMPLEMENT OF THE STARTING ADDRESS FOR MATRIX (SCALED 2).
# LEAVE CDU ANGLES SCALED 2PI IN V(MPAC).
# COS(MGA) WILL BE LEFT IN S1 (SCALED 1).
#
# THE DIRECTION COSINE MATRIX RELATING S/C AXES TO STABLE MEMBER AXES CAN BE WRITTEN
#
#      C  = COS(THETA) COS(PHI)
#      0
#
#      C  = -COS(THETA) SIN(PHI) COS(PHI) + SIN(THETA) SIN(PHI)
#      1
#
#      C  = COS(THETA) SIN(PHI) SIN(PHI) + SIN(THETA) COS(PHI)
#      2
#
#      C  = SIN(PHI)
#      3
#
#      C  = COS(PHI) COS(PHI)
#      4
#

```

```
#      C  = -COS(PHI) SIN(PHI)
#      5
#
#      C  = -SIN(THETA) COS(PHI)
#      6
#
#      C  = SIN(THETA) SIN(PHI) COS(PHI) + COS (THETA) SIN(PHI)
#      7
#
#      C  = -SIN(THETA) SIN(PHI) SIN(PHI) + COS(THETA)COS(PHI)
#      8
```

148  $\langle$ Page LM0362 148 $\rangle \equiv$ 

(112 720)

```

#
# WHERE PHI = OGA
#       THETA = IGA
#       PSI = MGA

DCMTOCDU      DLOAD*  ARCSIN
                  6,1
                  PUSH  COS          # PD +0          PSI
                  SL1   BOVB
                  SIGNMPAC
                  STORE  S1
                  DLOAD*  DCOMP
                  12D,1
                  DDV    ARCSIN
                  S1
                  PDDL*  BPL          # PD +2          THETA
                  0,1          # MUST CHECK THE SIGN OF COS(THETA)
                  OKTHETA          # TO DETERMINE THE PROPER QUADRANT.
                  DLOAD  DCOMP
                  BPL    DAD
                  SUHALFA
                  DPHALF
                  GOTO
                  CALCPHI
SUHALFA        DSU
                  DPHALF
CALCPHI        PUSH
OKTHETA        DLOAD*  DCOMP
                  10D,1
                  DDV    ARCSIN
                  S1
                  PDDL*  BPL          # PUSH DOWN PHI
                  8D,1
                  OKPHI
                  DLOAD  DCOMP          # PUSH UP PHI
                  BPL    DAD
                  SUHALFAP
                  DPHALF
                  GOTO
                  VECOFANG
SUHALFAP       DSU
                  GOTO
                  DPHALF
                  VECOFANG
OKPHI          DLOAD
VECOFANG       VDEF  RVQ          # PUSH UP PHI

```

149a

⟨Page LM0363 149a⟩≡

(112 720)

# ROUTINES FOR TERMINATING THE AUTOMATIC MANEUVER AND RETURNING TO USER.

TOOBADF

EXIT

TC

OCT

ALARM

00401

TCF

NOGO

# DO NOT ZERO ATTITUDE ERRORS

TC

CADR

BANKCALL

ZATTEROR

# ZERO ATTITUDE ERRORS

NOGO

TC

CADR

BANKCALL

STOPRATE

# STOP RATES

CAF

INHINT

TC

EBANK=

2CADR

TWO

WAITLIST

BCDU

GOODMANU

# ALL RETURNS ARE NOW MADE VIA GOODEND

TCF

ENDOFJOB

TOOBADI

EXIT

TCF

NOGO

1.10 imu performance test 2

149b

⟨imu performance test 2 149b⟩≡

(7)

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⟨Page LM0374 152⟩

⟨Page LM0375 154⟩

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⟨Page LM0380 160⟩

⟨Page LM0381 162⟩

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(149b 735)

```

# NAME --          IMU PERFORMANCE TESTS 2
#
# DATE --          MARCH 20, 1967
#
# BY --            SYSTEM TEST GROUP 864-6900 EXT. 1274
#
# MODNO. --        ZERO
#
# FUNCTIONAL DESCRIPTION
#
# POSITIONING ROUTINES FOR THE IMU PERFORMANCE TESTS AS WELL AS SOME OF
# THE TESTS THEMSELVES.  FOR A DESCRIPTION OF THESE SUBROUTINES AND THE
# OPERATING PROCEDURES (TYPICALLY) SEE STG MEMO 685.  THEORETICAL REF. E-1973

BANK      33
SETLOC    IMU2
BANK
EBANK=    POSITON
COUNT*   $$/P07

REDO      TC      NEWMODEX
          MM      07

GEOIMUTT  TC      IMUZERR
IMUBACK   CA      ZERO
          TS      NDXCTR
          TS      TORQNDX
          TS      TORQNDX +1
          TS      OVFLOWCK
NBPOSPL   CA      DEC17
          TS      ZERONDX
          CA      XNBADR
          TC      ZEROING
          CA      HALF
          TS      XNB
GUESS     TC      INTPRET
LATAZCHK  DLOAD    SL2
          LATITUDE
          STODL    DSPTEM1 +1
          AZIMUTH
          RTB      EXIT
          1ST02S
          XCH      MPAC
          TS      DSPTEM1
          CAF      VN0641

```

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TC	BANKCALL
CADR	GOFLASH
TC	ENDTEST1
TC	+2
TC	-5

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(149b 735)

	TC	INTPRET	
	SLOAD	RTB	
		DSPTM1	
		CDULOGIC	
	STORE	AZIMUTH	
	SLOAD	SR2	
		DSPTM1 +1	
	STORE	LATITUDE	
	COS	DCOMP	
	SL1		
	STODL	WANGI	
		LATITUDE	
	SIN	SL1	
	STODL	WANGO	
		AZIMUTH	
	PUSH	SIN	
	STORE	YNB	+2
	STODL	ZNB	+4
	COS		
	STORE	YNB	+4
	DCOMP		
POSGMBL	STCALL	ZNB	+2
		CALCGA	
	EXIT		
	TC	BANKCALL	
	CADR	IMUCOARS	
	CAF	BIT14	# IF BIT14 SET, GIMBAL LOCK
	MASK	FLAGWRD3	
	EXTEND		
	BZF	+2	
	INCR	NDXCTR	# +1 IF IN GIMBAL LOCK, OTHERWISE 0
	TC	DOWNFLAG	
	ADRES	GLOKFAIL	# RESET GIMBAL LOCK FLAG
	TC	IMUSLLLG	
	CCS	NDXCTR	# IF ONE GO AND DO A PIPA TEST ONLY
	TC	PIPACHK	# ALIGN AND MEASURE VERTICAL PIPA RATE
	TC	FINIMUDD	
	EXTEND		
	DCA	PERFDLAY	
	TC	LONGCALL	# DELAY WHILE SUSPENSION STABILIZES
	EBANK=	POSITON	
	2CADR	GOESTIMS	
	CA	ESTICADR	
	TC	JOBSLEEP	



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GOESTIMS	CA	ESTICADR
	TC	JOBWAKE
	TC	TASKOVER
ESTICADR	CADR	ESTIMS
TORQUE	CA	ZERO

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(149b 735)

	TS	DSPTM2	
	CA	DRIFTI	
	TS	DSPTM2 +1	
	INDEX	POSITON	
	TS	SOUTHDR -1	
	TC	SHOW	
PIPACHK	INDEX	NDXCTR	# PIPA TEST
	TC	+1	
	TC	EARTH* <sup>*</sup>	
	CA	DEC17	# ALLOW PIP COUNTER TO OVERFLOW 17 TIMES
	TS	DATAPL +4	# IN THE ALLOTTED TIME INTERVAL
	CA	DEC58	
	TS	LENGTHOT	
	CA	ONE	
	TS	RESULTCT	
	CA	ZERO	
	INDEX	PIPINDEX	
	TS	PIPA	
	TS	DATAPL	
	TC	CHECKG	
	INHINT		
	CAF	TWO	
	TC	TWIDDLE	
	EBANK=	XSM	
	ADRES	PIPATASK	
	TC	ENDOFJOB	
PIPATASK	EXTEND		
	DIM	LENGTHOT	
	CA	LENGTHOT	
	EXTEND		
	BZMF	STARTPIP	
	CAF	BIT10	
	TC	TWIDDLE	
	EBANK=	XSM	
	ADRES	PIPATASK	
STARTPIP	CAF	PRI020	
	TC	FINDVAC	
	EBANK=	XSM	
	2CADR	PIPJOB	
	TC	TASKOVER	
PIPJOB	INDEX	NDXCTR	

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TC	+1
TC	EARTH*
CA	LENGTHOT

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```

EXTEND
BZMF      +2
TC        ENDOFJOB

CA        FIVE
TS        RESULTCT
TC        CHECKG
CCS       DATAPL  +1
TC        +4
TC        CSHOLE
CS        DATAPL  +4
TS        DATAPL  +4
EXTEND
DCS       DATAPL
DAS       DATAPL  +4

TC        INTPRET
DLOAD    DSU
          DATAPL  +6
          DATAPL  +2
BPL       CALL
          AINGOTN
          OVERFFIX
AINGOTN   PDDL    DDV
          DATAPL  +4
DMPR      RTB
          DEC585      # DEC585 HAS BEEN REDEFINED FOR LEM
          SGNAGREE
STORE     DSPTM2
EXIT
CCS       NDXCTR
TC        COAALIGN    # TAKE PLATFORM OUT OF GIMBAL LOCK
TC        SHOW
VERTDRFT  CA        3990DEC  # ABOUT 1 HOUR VERTICAL DRIFT TEST
TS        LENGTHOT
INDEX     POSITON
CS        SOUTHDR -2
TS        DRIFTT
CCS       PIPINDEX    # OFFSET PLATFORM TO MISS PIP DEAD-ZONES
TCF       PON4        # Z-UP IN POS 4
PON2      CS        BIT5  # X-UP
ADS       ERCOMP  +2
CA        BIT5
ADS       ERCOMP  +4
TCF       PON

```

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PON4	CS	BIT5
	ADS	ERCOMP +2
	CA	BIT5
	ADS	ERCOMP
PON	TC	EARTH*

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(149b 735)

# ALLOW ONLY SOUTH GYRO EARTH RATE COMPENS

	CA	ZERO
	TS	ERVECTOR
	TS	ERVECTOR +1
GUESS1	CAF	POSMAX
	TS	TORQNDX
	TS	TORQNDX +1
	CA	CDUX
	TS	LOSVEC
	TC	ESTIMS
VALMIS	CA	DRIFTO
	TS	DSPTM2 +1
	CA	ZERO
	TS	DSPTM2
	TC	SHOW
ENDTEST1	TC	DOWNFLAG
	ADRES	IMUSE
	CS	ZERO
	TC	NEWMODEA
	TC	ENDEXT

158  $\langle$ Page LM0378 158 $\rangle \equiv$ 

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```

OVERFFIX      DAD      DAD
                DPPOS MAX
                ONEDPP
                RVQ

COAALIGN      EXTEND
                QXCH     ZERONDX
                CA       ZERO
                TS       THETAD
                TS       THETAD +1
                TS       THETAD +2
                TC       BANKCALL
                CADR     IMUCOARS
ALIGNCOA      TC       BANKCALL
                CADR     IMUSTALL
                TC       SOMERR2
                TC       ZERONDX

IMUSLLLG      EXTEND
                QXCH     ZERONDX
                TC       ALIGNCOA

FINIMUDD      EXTEND
                QXCH     ZERONDX
                TC       BANKCALL
                CADR     IMUFINE
                TC       ALIGNCOA

IMUZERR       EXTEND
                QXCH     ZERONDX
                TC       BANKCALL
                CADR     IMUZERO
                TC       ALIGNCOA

CHECKG        EXTEND
                QXCH     QPLACE
                TC       +6
CHECKG1       RELINT
                CA       NEWJOB
                EXTEND
                BZMF     +6
                TC       CHANG1
                INHINT
                INDEX    PIPINDEX
                CS       PIPAX

# COARSE ALIGN SUBROUTINE

# PIP PULSE CATCHING ROUTINE

```

	TS	ZERONDX	
	INHINT		
159	$\langle \text{Page } LM0379 \text{ 159} \rangle \equiv$		(149b 735)
	INDEX	PIPINDEX	
	CA	PIPAX	
	AD	ZERONDX	
	EXTEND		
	BZF	CHECKG1	
	INDEX	PIPINDEX	
	CA	PIPAX	
	INDEX	RESULTCT	
	TS	DATAPL	
	TC	FINETIME	
	INDEX	RESULTCT	
	TS	DATAPL +1	
	INDEX	RESULTCT	
	LXCH	DATAPL +2	
	RELINT		
ENDCHKG	TC	QPLACE	
ZEROING	TS	L	
	TCF	+2	
ZEROING1	TS	ZERONDX	
	CAF	ZERO	
	INDEX	L	
	TS	0	
	INCR	L	
	CCS	ZERONDX	
	TCF	ZEROING1	
	TC	Q	

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ERTHRVSE	DLOAD	PDDL	SCHZEROS	# PD24 = (SIN	-COS	0) (OMEG/MS
			LATITUDE			
	COS	DCOMP				
	PDDL	SIN				
		LATITUDE				
	VDEF	VXSC				
		OMEG/MS				
	STORE	ERVECTOR				
	RTB					
		LOADTIME				
	STOVL	TMARK				
		SCHZEROS				
	STORE	ERCOMP				
	RVQ					
EARTHR	ITA	RTB				
		S2				
		LOADTIME				
	STORE	TEMPTIME				
	DSU	BPL				
		TMARK				
		ERTHR				
	CALL					
ERTHR	SL	OVERFFIX				
		VXSC				
		9D				
		ERVECTOR				
	MXV	VAD				
		XSM				
		ERCOMP				
	STODL	ERCOMP				
		TEMPTIME				
	STORE	TMARK				
	AXT, 1	RTB				
	ECADR	ERCOMP				
		PULSEIMU				
	GOTO					
		S2				
EARTHR*	EXTEND					
	QXCH	QPLACES				
	TC	INTPRET				
	CALL					
		EARTHR				



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EXIT  
TC IMUSLLLG  
TC QPLACES

SHOW EXTEND

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	QXCH	QPLACE	
SHOW1	CA	POSITON	
	TS	DSPTM2 +2	
	CA	VB06N98	
	TC	BANKCALL	
	CADR	GOFLASH	
	TC	ENDTEST1	# V34
	TC	QPLACE	# V33
	TCF	SHOW1	
3990DEC	DEC	3990	
VB06N98	VN	0698	
VN0641	VN	0641	
DEC17	=	ND1	
DEC58	DEC	58	
OGCPL	ECADR	OGC	
1SECX	=	1SEC	
XNBADR	GENADR	XNB	
XSMADR	GENADR	XSM	
	BLOCK	2	
	COUNT*	\$\$/P07	
FINETIME	INHINT		# RETURNS WITH INTERRUPT INHIBITED
	EXTEND		
	READ	LOSCALAR	
	TS	L	
	EXTEND		
	RXOR	LOSCALAR	
	EXTEND		
	BZF	+4	
	EXTEND		
	READ	LOSCALAR	
	TS	L	
+4	CS	POSMAX	
	AD	L	
	EXTEND		
	BZF	FINETIME +1	
	EXTEND		
	READ	HISCALAR	
	TC	Q	

## 1.11 imu performance tests 4

163a  $\langle \text{imu performance tests 4 163a} \rangle \equiv$  (7)

$\langle \text{Page LM0382 163b} \rangle$   
 $\langle \text{Page LM0383 164} \rangle$   
 $\langle \text{Page LM0384 165} \rangle$   
 $\langle \text{Page LM0385 166} \rangle$   
 $\langle \text{Page LM0386 168} \rangle$   
 $\langle \text{Page LM0387 170} \rangle$   
 $\langle \text{Page LM0388 172} \rangle$   
 $\langle \text{Page LM0389 173a} \rangle$

163b  $\langle \text{Page LM0382 163b} \rangle \equiv$  (163a 737)

```
# PROGRAM --      IMU PERFORMANCE TESTS 4
# DATE --         NOV 15, 1966
# BY --           GEORGE SCHMIDT IL7-146 EXT 1126
# MOD NO-ZERO
#
# FUNCTIONAL DESCRIPTION
#
# THIS SECTION CONSISTS OF THE FILTER FOR THE GYRO DRIFT TESTS.  NO COMPASS
# IS DONE IN LEM.  FOR A DESCRIPTION OF THE FILTER SEE E-1973.  THIS
# SECTION IS ENTERED FROM IMU 2.  IT RETURNS THERE AT END OF TEST.
#
# EARTH,OGC ZERO,ERTHRVSE
#
# NORMAL EXIT
#
# LENGTHOT GOES TO ZERO -- RETURN TO IMU PERF TESTS 2 CONTROL
#
# ALARMS
#
# 1600  OVERFLOW IN DRIFT TEST
# 1601  BAD IMU MODING IN ANY ROUTINE THAT USES IMUSTALL
#       OUTPUT
#
# FLASHING DISPLAY OF RESULTS -- CONTROLLED IN IMU PERF TESTS 2
#
# DEBRIS
#
# ALL CENTRALS -- ALL OF EBANK XSM
```

164 (Page LM0383 164)≡

(163a 737)

	BANK	33	
	SETLOC	IMU4	
	BANK		
	COUNT*	\$\$/P07	
	EBANK=	XSM	
ESTIMS	INHINT		
	CAE	1SECXT	
	TC	TWIDDLE	
	EBANK=	XSM	
	ADRES	ALLOOP	
	CAF	ZERO	# ZERO THE PIPAS
	TS	PIPAX	
	TS	PIPAY	
	TS	PIPAZ	
	RELINT		
	CA	77DECML	
	TS	ZERONDX	
	CA	ALXXXZ	
	TC	ZEROING	
	TC	INTPRET	
	SLOAD		
		SCHZEROS	
	STOVL	GCOMP SW -1	
		INTVAL +2	
	STOVL	ALX1S	
		SCHZEROS	
	STORE	DEL VX	
	STORE	GCOMP	
	SLOAD		
		TORQNDX	
	DCOMP	BMN	
		VERTSKIP	
	CALL		
		ERTHRVSE	
VERTSKIP	EXIT		
	TC	SLEEPIE +1	

165     <Page LM0384 165>≡     (163a 737)

```

      ALLOOP      CA      OVFLOWCK
      EXTEND
      BZF      +2
      TC      TASKOVER
      CCS      ALTIM
      CA      A      # SHOULD NEVER HIT THIS LOCATION
      TS      ALTIMS
      CS      A
      TS      ALTIM
      CS      ONE
      AD      GEOCOMPS
      EXTEND
      BZF      +4
      CA      LENGTHOT
      EXTEND
      BZMF      +5
      CAE      1SECXT
      TC      TWIDDLE
      EBANK=    XSM
      ADRES     ALLOOP
      CAF      ZERO
      XCH      PIPAX
      TS      DELVX
      CAF      ZERO
      XCH      PIPAY
      TS      DELVY
      CAF      ZERO
      XCH      PIPAZ
      TS      DELVZ
      SPECSTS   CAF      PRI020
      TC      FINDVAC
      EBANK=    XSM
      2CADR     ALFLT      # START THE JOB

      TC      TASKOVER

```

166 (Page LM0385 166)≡

(163a 737)

ALFLT	CCS	GEOCOMPS	
	TC	+2	
	TC	NORMLOP	
	TC	BANKCALL	
	CADR	1/PIPA	
NORMLOP	TC	INTPRET	
	DLOAD		
		INTVAL	
	STOVL	S1	
		DELVX	
	VXM	VSL1	
		XSM	
	DLOAD	DCOMP	
		MPAC +3	
	STODL	DPIPAY	
		MPAC +5	
	STORE	DPIPZ	
	SETPD	AXT,1	
		0	
		8D	
	SLOAD	DCOMP	
		GEOCOMPS	
	BMN		
		PERFERAS	
ALCGKK	SLOAD	BMN	
		ALTIMS	
		ALFLT3	
ALKCG	AXT,2	LXA,1	# LOADS SLOPES AND TIME CONSTANTS AT RQST
		12D	
		ALX1S	
ALKCG2	DLOAD*	INCR,1	
		ALFDK +144D,1	
	DEC	-2	
	STORE	ALDK +10D,2	
	TIX,2	SXA,1	
		ALKCG2	
		ALX1S	
ALFLT3	AXT,1		
		8D	
DEMLP	DLOAD*	DMP	
		DPIPAY +8D,1	
		PIPASC	
	SLR	BDSU*	

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	9D	
	INTY	+8D,1
STORE	INTY	+8D,1
PDDL	DMP*	
	VELSC	

168 (Page LM0386 168)≡

(163a 737)

		VLAUN	+8D,1
	SL2R		
	DSU	STADR	
	STORE	DELM	+8D,1
	STORE	DELM	+10D,1
	TIX,1	AXT,2	
		DEMLP	
		4	
ALILP	DLOAD*	DMPR*	
		ALK	+4,2
		ALDK	+4,2
	STORE	ALK	+4,2
	TIX,2	AXT,2	
		ALILP	
		8D	
ALKLP	LXC,1	SXA,1	
		CMPX1	
		CMPX1	
	DLOAD*	DMPR*	
		ALK	+1,1
		DELM	+8D,2
	DAD*		
		INTY	+8D,2
	STORE	INTY	+8D,2
	DLOAD*	DAD*	
		ALK	+12D,2
		ALDK	+12D,2
	STORE	ALK	+12D,2
	DMPR*	DAD*	
		DELM	+8D,2
		INTY	+16D,2
	STORE	INTY	+16D,2
	DLOAD*	DMP*	
		ALSK	+1,1
		DELM	+8D,2
	SL1R	DAD*	
		VLAUN	+8D,2
	STORE	VLAUN	+8D,2
	TIX,2	AXT,1	
		ALKLP	
		8D	
LOOSE	DLOAD*	PDDL*	
		ACCWD	+8D,1
		VLAUN	+8D,1



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169

PDDL*	VDEF	
	POSNV	+8D,1
MXV	VSL1	
	TRANSM1	

170 (Page LM0387 170)≡

(163a 737)

```

DLOAD
MPAC
STORE POSNV +8D,1
DLOAD
MPAC +3
STORE VLAUN +8D,1
DLOAD
MPAC +5
STORE ACCWD +8D,1
TIX,1
LOOSE

AXT,2 AXT,1 # EVALUATE SINES AND COSINES
6
2
BOOP DLOAD* DMPR
ANGX +2,1
GEORGEJ

SR2R
PUSH SIN
SL3R XAD,1
X1
STORE 16D,2
DLOAD
COS
STORE 22D,2 # COSINES
TIX,2
BOOP

PERFERAS EXIT
CA EBANK7
TS EBANK
EBANK= ATIGINC
TC ATIGINC # GOTO ERASABLE TO CALCULATE ONLY TO RETN

# CAUTION
#
# THE ERASABLE PROGRAM THAT DOES THE CALCULATIONS MUST BE LOADED
# BEFORE ANY ATTEMPT IS MAKE TO RUN THE IMU PERFORMANCE TEST

EBANK= AZIMUTH
CCS LENGTHOT
TC SLEEPIE
CCS TORQNDX
TCF +2

```

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TC	SETUPER1		
CA	CDUX		
TS	LOSVEC	+1	# FOR TROUBLESHOOTING VD POSNS 2\$4

172 &lt;Page LM0388 172&gt;≡

(163a 737)

SETUPER1	TC	INTPRET	
	DLOAD	PDDL	# ANGLES FROM DRIFT TEST ONLY
		ANGZ	
		ANGY	
	PDDL	VDEF	
		ANGX	
	VCOMP	VXSC	
		GEORGEJ	
	MXV	VSR1	
		XSM	
	STORE	OGC	
	EXIT		
	CA	OGCPL	
	TC	BANKCALL	
	CADR	IMUPULSE	
	TC	IMUSLLLG	
GEOSTR4	CCS	TORQNDX	# ONLY POSITIVE IF IN VERTICAL DRIFT TEST
	TC	VALMIS	
	TC	INTPRET	
	CALL		
		ERTHRVSE	
	EXIT		
	TC	TORQUE	
SLEEPIE	TS	LENGTHOT	# TEST NOT OVER-DECREMENT LENGTHOT
	CCS	TORQNDX	# ARE WE DOING VERTDRIFT
	TC	EARTH*	
	TC	ENDOFJOB	
SOMEERRR	CA	EBANK5	
	TS	EBANK	
	CA	ONE	
	TS	OVFLOWCK	# STOP ALLOOP FROM CALLING ITSELF
	TC	ALARM	
	OCT	1600	
	TC	ENDTEST1	
SOMERR2	CAF	OCT1601	
	TC	VARALARM	
	TC	DOWNFLAG	
	ADRES	IMUSE	
	TC	ENDOFJOB	
OCT1601	OCT	01601	
DEC585	OCT	06200	# 3200 B+14 ORDER IS IMPORTANT

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	SCHZEROS	2DEC	.00000000	
173a	$\langle \text{Page LM0389 173a} \rangle \equiv$			(163a 737)
		2DEC	.00000000	
		OCT	00000	
	ONEDPP	OCT	00000	# ORDER IS IMPORTANT
		OCT	00001	
		OCT	4	
		OCT	2	
		DEC	144	
		DEC	-1	
	SOUPPLY	2DEC	.93505870	# INITIAL GAINS FOR PIP OUTPUTS
		2DEC	.26266423	# INITIAL GAINS/4 FOR ERECTION ANGLES
	77DECML	DEC	77	
	ALXXXZ	GENADR	ALX1S -1	
	PIPASC	2DEC	.13055869	
	VELSC	2DEC	-.52223476	# 512/980.402
	ALSK	2DEC	.17329931	# SSWAY VEL GAIN X 980.402/4096
		2DEC	-.00835370	# SSWAY ACCEL GAIN X 980.402/4096
	GEORGEJ	2DEC	.63661977	
	GEORGEK	2DEC	.59737013	

### 1.12 s band antenna for lm

173b	$\langle s \text{ band antenna for lm 173b} \rangle \equiv$	(7)
	$\langle \text{Page LM0486 174} \rangle$	
	$\langle \text{Page LM0487 176} \rangle$	
	$\langle \text{Page LM0488 178} \rangle$	
	$\langle \text{Page LM0489 179a} \rangle$	

174 *<Page LM0486 174>≡*

(173b 764)

```

# SUBROUTINE NAME: R05 -- S-BAND ANTENNA FOR LM
#
# MOD0 BY T. JAMES
# MOD1 BY P. SHAKIR
#
# FUNCTIONAL DESCRIPTION
#
# THE S-BAND ANTENNA ROUTINE, R05, COMPUTES AND DISPLAYS THE PITCH AND
# YAW ANTENNA GIMBAL ANGLES REQUIRED TO POINT THE LM STEERABLE ANTENNA
# TOWARD THE CENTER OF THE EARTH. THIS ROUTINE IS SELECTED BY THE ASTRO-
# NAUT VIA DSKY ENTRY DURING COASTING FLIGHT OR WHEN THE LM IS ON THE MOON
# SURFACE. THE EARTH OR MOON REFERENCE COORDINATE SYSTEM IS USED DEPENDING
# ON WHETHER THE LM IS ABOUT TO ENTER OR HAS ALREADY ENTERED THE MOON
# SPHERE OF INFLUENCE, RESPECTIVELY.
#
# TO CALL SUBROUTINE, ASTRONAUT KEYS IN V 64 E
#
# SUBROUTINES CALLED ---
#     R02BOTH
#     INTPRET
#     LOADTIME
#     LEMCONIC
#     LUNPOS
#     CDUTRIG
#     *SMNB*
#     BANKCALL
#     B50OFF
#     ENDOFJOB
#     BLANKRET
#
# RETURNS WITH
#     PITCH ANGLE IN PITCHANG          REV. BO
#     YAW ANGLE IN YAWANG              REV. BO
#
# ERASABLES USED
#     PITCHANG
#     YAWANG
#     RLM
#     VAC AREA

```

BANK    41  
SETLOC SBAND  
BANK

EBANK= WHOCARES

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	COUNT*	\$\$/R05
SBANDANT	TC	BANKCALL

176 (Page LM0487 176)≡

(173b 764)

```

      CADR    R02BOTH      # CHECK IF IMU IS ON AND ALIGNED
      TC      INTPRET
      SETPD   RTB
              OD
              LOADTIME    # PICK UP CURRENT TIME
      STCALL  TDEC1        # ADVANCE INTEGRATION TO TIME IN TDEC1
              LEMCONIC    # USING CONIC INTEGRATION
      SLOAD   BHIZ
              X2           # X2 =0 EARTH SPHERE, X2 =2 MOON SPHERE
              CONV4
      VLOAD
              RATT
      STODL   RLM
              TAT
CONV3        CALL
              LUNPOS      # UNIT POSITION VECTOR FROM EARTH TO MOON
      VLOAD   VXSC
              VMOON
              REMDIST     # MEAN DISTANCE FROM EARTH TO MOON
      VSL1    VAD
              RLM
      GOTO    CONV5
CONV4        VLOAD
              RATT        # UE = -UNIT(RATT)          EARTH SPHERE
CONV5        SETPD   UNIT  # UE = -UNIT((REM)(UEM) + RL)    MOON SPHERE
              OD         # SET PL POINTER TO 0
      VCOMP   CALL
              CDUTRIG    # COMPUTE SINES AND COSINES OF CDU ANGLES
      MXV     VSL1      # TRANSFORM REF. COORDINATE SYSTEM TO
              REFMMAT    # STABLE MEMBER B-1 X B-1 X B+1 = B-1
      PUSH    DLOAD    # 8D
              HI6ZEROS
      STORE   PITCHANG
      STOVL   YAWANG   # ZERO OUT ANGLES
      CALL
              *SMNB*
      STODL   RLM      # PRE-MULTIPLY RLM BY (NBSA) MATRIX(B0)
              RLM      +2
      PUSH    DSU
              RLM
      DMP
              10VSQRT2
      STODL   RLM      +2
      DAD     DMP

```



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	RLM	
	10VSQRT2	
STOVL	RLM	# R B-1
	RLM	
UNIT	PDVL	

178 (Page LM0488 178)≡

(173b 764)

```

                                RLM
                                VPROJ  VSL2      # PROJECTION OF R ONTO LM XZ PLANE.
                                HIUNITY
                                BVSU    BOV        # CLEAR OVERFLOW INDICATOR IF ON
                                RLM
                                COVCNV
                                COVCNV  UNIT  BOV        # EXIT ON OVERFLOW
                                SBANDEX
                                PUSH    VXV        # URP VECTOR B-1
                                HIUNITZ
                                VSL1    VCOMP      # UZ X URP = -(URP X UZ)
                                STORE    RLM        # X VEC B-1
                                DOT      PDVL      # SGN(X.UY) UNSCALED
                                HIUNITY
                                RLM
                                ABVAL    SIGN
                                ASIN      # ASIN((SGN(X.UY))ABV(X))      REV BO
                                STOVL    PITCHANG
                                URP
                                DOT      BPL
                                HIUNITZ
                                NOADJUST  # YES, -90 TO +90
                                DLOAD    DSU
                                HIDPHALF
                                PITCHANG
                                STORE    PITCHANG
                                NOADJUST  VLOAD    VXV
                                UR        # Z = (UR X URP)
                                URP
                                VSL1
                                STODL    RLM        # Z VEC B-1
                                PITCHANG
                                SIN      VXSC
                                HIUNITZ
                                PDDL     COS
                                PITCHANG
                                VXSC     VSU
                                HIUNITX  # (UX COS ALPHA) - (UZ SIN ALPHA)
                                DOT      PDVL      # YAW.Z
                                RLM
                                RLM
                                ABVAL    SIGN
                                ASIN
                                STORE    YAWANG
                                SBANDEX  EXIT

```

179a

CA

EXTVBACT

MASK

BIT5

EXTEND

BZF

ENDEXT

CAF

PRI05

# IS BIT5 STILL ON

# NO

TC

PRI0CHNG

CAF

V06N51

TC

BANKCALL

CADR

GOMARKFR

TC

B5OFF

TC

B5OFF

TC

ENDOFJOB

CAF

BIT3

TC

BLANKET

CAF

PRI04

TC

PRI0CHNG

TC

SBANDANT +2

V06N51

VN

10VSQRT2

2DEC

UR

EQUALS

URP

EQUALS

SBANK=

OD

6D

LOWSUPER

# YES, CONTINUE DISPLAYING ANGLES.

# 1/SQRT(2)

# END OF LNYAIDE .001 \*\*\*

1.13 radar leadin routines

179b

radar leadin routines

179b

180

181a

(7)

180 (Page LM0490 180)≡

(179b 761)

BANK 25  
 SETLOC RRLEADIN  
 BANK

EBANK= RSTACK

# RADAR SAMPLING LOOP.

RADSAMP	COUNT*	\$\$/RLEAD	
	CCS	RSAMPDT	# TIMES NORMAL ONCE-PER-SECOND SAMPLING.
	TCF	+2	
	TCF	TASKOVER	# +0 INSERTED MANUALLY TERMINATES TEST.

TC WAITLIST  
 EBANK= RSTACK  
 2CADR RADSAMP

CAF PRI025  
 TC NOVAC  
 EBANK= RSTACK  
 2CADR DORSAMP

CAF	BIT14	# FOR CYCLIC SAMPLING, RTSTDEX =
EXTEND		# RTSTLOC/2 + RTSTBASE
MP	RTSTLOC	
AD	RTSTBASE	# 0 FOR RR, 2 FOR LR.
TS	RTSTDEX	
TCF	TASKOVER	

# DO THE ACTUAL RADAR SAMPLE.

DORSAMP	TC	VARADAR	# SELECTS VARIABLE RADAR CHANNEL.
	TC	BANKCALL	
	CADR	RADSTALL	

INCR	RFAILCNT	# ADVANCE FAIL COUNTER BUT ACCEPT BAD DATA
------	----------	--

DORSAMP2	INHINT	
	CA	FLAGWRD5
	MASK	R77FLBIT
	CCS	A
	TCF	+4

# DON'T UPDATE RSTACK IF IN R77.

DXCH SAMPLSUM  
 INDEX RTSTLOC

```
DXCH      RSTACK

CS        RTSTLOC      # CYCLE RTSTLOC.
AD        RTSTMAX
EXTEND
```

181a     $\langle$ Page LM0491 181a $\rangle \equiv$  (179b 761)

```
BZF      +3
CA       RTSTLOC
AD       TWO          # STORAGE IS DP
TS       RTSTLOC
TCF      ENDOFJOB     # CONTINUOUS SAMPLING AND 2N TRIES -- GONE.
```

# VARIABLE RADAR DATA CALLER FOR ONE MEASUREMENT ONLY.

```
VARADAR   CAF      ONE          # WILL BE SENT TO RADAR ROUTINE IN A BY
          TS       BUF2         # SWCALL
          INDEX    RTSTDEX
          CAF      RDRLOCS
          TCF      SWCALL       # NOT TOUCHING Q.
```

```
RDRLOCS   CADR     RRRANGE      # =0
          CADR     RRRDOT       # =1
          CADR     LRVELX       # =2
          CADR     LRVELY       # =3
          CADR     LRVELZ       # =4
          CADR     LRALT        # =5
```

1.14    p30-p37 routines

181b     $\langle$ p30-p37 routines 181b $\rangle \equiv$  (7)

$\langle$ Page LM0614 182 $\rangle$

$\langle$ Page LM0615 183 $\rangle$

$\langle$ Page LM0616 184 $\rangle$

$\langle$ Page LM0617 186 $\rangle$

182 &lt;Page LM0614 182&gt;≡

(181b 750)

```

# PROGRAM DESCRIPTION P30          DATE 3-6-67
#
# MOD.1 BY RAMA AIYAWAR
#
# FUNCTIONAL DESCRIPTIONS
#     ACCEPT ASTRONAUT INPUTS OF TIG,DELV(LV)
#     CALL IMU STATUS CHECK ROUTINE (R02)
#     DISPLAY TIME TO GO, APOGEE, PERIGEE, DELV(MAG), MGA AT IGN
#     REQUEST BURN PROGRAM
#
# CALLING SEQUENCE VIA JOB FROM V37
#
# EXIT VIA V37 CALL OR TO GOTOP00H (V34E)
#
# SUBROUTINE CALLS --  FLAGUP, PHASCHNG, BANKCALL, ENDOFJOB, GOFLASH, GOFLASHR
#                      GOPERF3R, INTPRET, BLANKET, GOTOP00H, R02BOTH, S30.1,
#                      TIG/N35, MIDGIM, DISPMGA
#
# ERASABLE INITIALIZATION -- STATE VECTOR
#
# OUTPUT --          RINIT, VINIT, +MGA, VTIG, RTIG, DELVSIN, DELVSAB, DELVSLV, HAPO,
#                      HPER, TTOGO
#
# DEBRIS -- A, L, MPAC, PUSHLIST

                BANK    32
                SETLOC  P30S
                BANK
                EBANK=  +MGA
                COUNT*  $$/P30
P30             TC      UPFLAG          # SET UPDATE FLAG
                ADRES   UPDATFLG
                TC      UPFLAG          # SET TRACK FLAG
                ADRES   TRACKFLG

P30N33          CAF     V06N33          # T OF IGN
                TC      VNP00H          # RETURN ON PROCEED, POOH ON TERMINATE

                CAF     V06N81          # DISPLAY DELTA V (LV)
                TC      VNP00H          #      REDISPLAY ON RECYCLE

                TC      DOWNFLAG        # RESET UPDATE FLAG
                ADRES   UPDATFLG
                TC      INTPRET
                CALL

```

		S30.1	
	SET	EXIT	
		UPDATFLG	
PARAM30	CAF	V06N42	# DISPLAY APOGEE,PERIGEE,DELTA V
	TC	VNP00H	

183      *⟨Page LM0615 183⟩*≡      (181b 750)

	TC	INTPRET	
	SETGO		
		XDELVFLG	# FOR P40'S: EXTERNAL DELTA-V GUIDANCE.
		REVN1645	# TRKMKCNT, T60, +MGA DISPLAY
V06N33	VN	0633	
V06N42	VN	0642	

184 (Page LM0616 184)≡ (181b 750)

```

# PROGRAM DESCRIPTION S30.1      DATE 9NOV66
# MOD NO 1                      LOG SECTION P30,P37
# MOD BY RAMA AIYAWAR **
#
# FUNCTIONAL DESCRIPTION
#   BASED ON STORED TARGET PARAMETERS (R OF IGNITION (RTIG), V OF
#   IGNITION (VTIG), TIME OF IGNITION (TIG)), COMPUTE PERIGEE ALTITUDE
#   APOGEE ALTITUDE AND DELTAV REQUIRED (DELVSIN).
#
# CALLING SEQUENCE
#   L          CALL
#   L+1        s30.1
#
# NORMAL EXIT MODE
#   AT L+2 OR CALLING SEQUENCE (GOTO L+2)
#
# SUBROUTINES CALLED
#   LEMPREC
#   PERIAPO
#
# ALARM OR ABORT EXIT MODES
#   NONE
#
# ERASABLE INITIALIZATION REQUIRED
#   TIG          TIME OF IGNITION          DP B28CS
#   DELVSLV      SPECIFIED DELTA-V IN LOCAL VERT.
#               COORDS. OF ACTIVE VEHICLE AT
#               TIME OF IGNITION          VECTOR B+7 METERS/CS
#
# OUTPUT
#   RTIG         POSITION AT TIG            VECTOR B+29 METERS
#   VTIG         VELOCITY AT TIG          VECTOR B+29 METERS/CS
#   PDL 4D       APOGEE ALTITUDE          DP B+29 M, B+27 METERS.
#   HAPO        APOGEE ALTITUDE          DP B+29 METERS
#   PDL 8D       PERIGEE ALTITUDE          DP B+29 M, B+27 METERS.
#   HPER        PERIGEE ALTITUDE          DP B+29 METERS
#   DELVSIN      SPECIFIED DELTA-V IN INTERTIAL
#               COORD. OF ACTIVE VEHICLE AT
#               TIME OF IGNITION          VECTOR B+7 METERS/CS
#   DELVSAB      MAG. OF DELVSIN          VECTOR B+7 METERS/CS
#
# DEBRIS        QTEMP    TEMP.ERASABLE
#               QPRET, MPAC
#               PUSHLIST

```



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```

      SETLOC  P30S1
      BANK
      COUNT*  $$/S30S
S30.1  STQ     DLOAD
      QTEMP
      TIG      # TIME IGNITION SCALED AT 2(+28)CS
      STCALL   TDEC1
      LEMPREC  # ENCKE ROUTINE FOR LEM
      VLOAD    SXA,2
```

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(181b 750)

```

RATT
RTX2
STORE RTIG # RADIUS VECTOR AT IGNITION TIME
UNIT VCOMP
STOVL DELVSIN # ZRF/LV IN DELVSIN SCALED AT 2
VATT # VELOCITY VECTOR AT TIG, SCALED 2(7) M/CS
STORE VTIG
VXV UNIT
RTIG
SETPD SXA,1
0
RTX1
PUSH VXV # YRF/LV PDL 0 SCALED AT 2
DELVSIN
VSL1 PDVL
PDVL PDVL # YRF/LV PDL 6 SCALED AT 2
DELVSIN # ZRF/LV PDL 12D SCALED AT 2
DELVSLV
VXM VSL1
0
STORE DELVSIN # DELTAV IN INERT. COOR. SCALED TO B+7M/CS
ABVAL
STOVL DELVSAB # DELTA V MAG.
RTIG # (FOR PERIAPO)
PDVL VAD # VREQUIRED = VTIG + DELVSIN (FOR PERIAPO)
VTIG
DELVSIN
CALL PERIAPO1
CALL SHIFTR1 # RESCALE IF NEEDED
CALL # LIMIT DISPLAY TO 9999.9 N. MI.
MAXCHK
STODL HPER # PERIGEE ALT 2(29) METERS FOR DISPLAY
4D
CALL SHIFTR1 # RESCALE IF NEEDED
CALL # LIMIT DISPLAY TO 9999.9 N. MI.
MAXCHK
STCALL HAPO # APOGEE ALT 2(29) METERS FOR DISPLAY
QTEMP

```

## 1.15 p32-p35 p72-p75 routines

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 $\langle Page\ LM0649\ 235a \rangle$   
 $\langle Page\ LM0650\ 235b \rangle$

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```

# COELLIPTIC SEQUENCE INITIATION (CSI) PROGRAMS (P32 AND P72)
#
# MOD NO -1      LOG SECTION -- P32-P35, P72-P75
# MOD BY WHITE.P  DATE 1JUNE67
#
# PURPOSE

#      (1)      TO CALCULATE PARAMETERS ASSOCIATED WITH THE TIME FOLLOWING
#                CONCENTRIC FLIGHT PLAN MANEUVERS -- THE CO-ELLIPTIC SEQUENCE
#                INITIATION (CSI) MANEUVER AND THE CONSTANT DELTA ALTITUDE
#                (CDH) MANEUVER.

#      (2)      TO CALCULATE THESE PARAMETERS BASED UPON MANEUVER DATA
#                APPROVED AND KEYED INTO THE DSKY BY THE ASTRONAUT.

#      (3)      TO DISPLAY TO THE ASTRONAUT AND THE GROUND DEPENDENT VARIABLES
#                ASSOCIATED WITH THE CONCENTRIC FLIGHT PLAN MANEUVERS FOR
#                APPROVAL BY THE ASTRONAUT/GROUND.

#      (4)      TO STORE THE CSI TARGET PARAMETERS FOR USE BY THE DESIRED
#                THRUSTING PROGRAM.
#
# ASSUMPTIONS

#      (1)      AT A SELECTED TPI TIME THE LINE OF SIGHT BETWEEN THE ACTIVE
#                AND PASSIVE VEHICLES IS SELECTED TO BE A PRESCRIBED ANGLE (E)
#                FROM THE HORIZONTAL PLANE DEFINED BY THE ACTIVE VEHICLE
#                POSITION.

#      (2)      THE TIME BETWEEN CSI IGNITION AND CDH IGNITION MUST BE
#                COMPUTED TO BE GREATER THAN 10 MINUTES FOR SUCCESSFUL
#                COMPLETION OF THE PROGRAM.

#      (3)      THE TIME BETWEEN CDH IGNITION AND TPI IGNITION MUST BE
#                COMPUTED TO BE GREATER THAN 10 MINUTES FOR SUCCESSFUL
#                COMPLETION OF THE PROGRAM.

#      (4)      CDH DELTA V IS SELECTED TO MINIMIZE THE VARIATION OF THE
#                ALTITUDE DIFFERENCE BETWEEN THE ORBITS.

#      (5)      CSI BURN IS DEFINED SUCH THAT THE IMPULSIVE DELTA V IS IN THE
#                HORIZONTAL PLANE DEFINED BY THE ACTIVE VEHICLE POSITION AT CSI
#                IGNITION.

#      (6)      THE PERICENTER ALTITUDE OF THE ORBIT FOLLOWING CSI AND CDH

```

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# MUST BE GREATER THAN 35,000 FT (LUNAR ORBIT) OR 85 NM (EARTH  
# ORBIT) FOR SUCCESSFUL COMPLETION OF THIS PROGRAM.

# (7) THE CSI AND CDH MANEUVERS ARE ORIGINALLY ASSUMED TO BE  
# PARALLEL TO THE PLANE OF THE CSM ORBIT. HOWEVER, CREW

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(187 751)

```
#      MODIFICATION OF DELTA V (LV) COMPONENTS MAY RESULT IN AN
#      OUT-OF-PLANE CSI MANEUVER

#      (8)      STATE VECTOR UPDATES BY P27 ARE DISALLOWED DURING AUTOMATIC
#               STATE VECTOR UPDATING INITIATED BY P20 (SEE ASSUMPTION 10).

#      (9)      COMPUTED VARIABLES MAY BE STORED FOR LATER VERIFICATION BY
#               THE GROUND.  THESE STORAGE CAPABILITIES ARE NORMALLY LIMITED
#               ONLY TO THE PARAMETERS FOR ONE THRUSTING MANEUVER AT A TIME
#               EXCEPT FOR CONCENTRIC FLIGHT PLAN MANEUVER SEQUENCES.

#      (10)     THE RENDEZVOUS RADAR MAY OR MAY NOT BE USED TO UPDATE THE LM
#               OR CSM STATE VECTORS FOR THIS PROGRAM.  IF RADAR USE IS
#               DESIRED THE RADAR WAS TURNED ON AND LOCKED BY THE CSM BY
#               PREVIOUS SELECTION OF P20.  RADAR SIGHTING MARKS WILL BE MADE
#               AUTOMATICALLY APPROXIMATELY ONCE A MINUTE WHEN ENABLED BY THE
#               TRACK AND UPDATE FLAGS (SEE P20).  THE RENDEZVOUS TRACKING
#               MARK COUNTER IS ZEROED BY THE SELECTION OF P20 AND AFTER EACH
#               THRUSTING MANEUVER.

#      (11)     THE ISS NEED NOT BE ON TO COMPLETE THIS PROGRAM.

#      (12)     THE OPERATION OF THE PROGRAM UTILIZES THE FOLLOWING FLAGS --
#
#               ACTIVE VEHICLE FLAG -- DESIGNATES THE VEHICLE WHICH IS
#               DOING RENDEZVOUS THRUSTING MANEUVERS TO THE PROGRAM WHICH
#               CALCULATES THE MANEUVER PARAMETERS.  SET AT THE START OF
#               EACH RENDEZVOUS PRE-THRUSTING PROGRAM.
#
#               FINAL FLAG -- SELECTS FINAL PROGRAM DISPLAYS AFTER CREW HAS
#               COMPLETED THE FINAL MANEUVER COMPUTATION AND DISPLAY
#               CYCLE.
#
#               EXTERNAL DELTA V STEERING FLAG -- DESIGNATES THE TYPE OF
#               STEERING REQUIRED FOR EXECUTION OF THIS MANEUVER BY THE
#               THRUSTING PROGRAM SELECTED AFTER COMPLETION OF THIS
#               PROGRAM.

#      (13)     IT IS NORMALLY REQUIRED THAT THE ISS BE ON FOR 1 HOUR PRIOR TO
#               A THRUSTING MANEUVER.

#      (14)     THIS PROGRAM IS SELECTED BY THE ASTRONAUT BY DSKY ENTRY
#
#               P32 IF THIS VEHICLE IS ACTIVE VEHICLE.
#
```

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```
# P72 IF THIS VEHICLE IS THE PASSIVE VEHICLE.  
#  
# INPUT  
  
# (1) TCSI TIME OF THE CSI MANEUVER
```

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#	(2)	NN	NUMBER OF APSIDAL CROSSINGS THRU WHICH THE ACTIVE
#			VEHICLE ORBIT CAN BE ADVANCED TO OBTAIN THE CDH
#			MANEUVER POINT.
#	(3)	ELEV	DESIRED LOS ANGLE AT TPI
#	(4)	TTPI	TIME OF THE TPI MANEUVER
#			
#	OUTPUT		
#	(1)	TRKMKCNT	NUMBER OF MARKS
#	(2)	TTGO	TIME TO GO
#	(3)	+MGA	MIDDLE GIMBAL ANGLE
#	(4)	DIFFALT	DELTA ALTITUDE AT CDH
#	(5)	T1TOT2	DELTA TIME FROM CSI TO CDH
#	(6)	T2TOT3	DELTA TIME FROM CDH TO TPI
#	(7)	DELVLVC	DELTA VELOCITY AT CSI -- LOCAL VERTICAL COORDINATES
#	(8)	DELVLVC	DELTA VELOCITY AT CDH -- LOCAL VERTICAL COORDINATES
#			
#	DOWNLINK		
#	(1)	TCSI	TIME OF THE CSI MANEUVER
#	(2)	TCDH	TIME OF THE CDH MANEUVER
#	(3)	TTPI	TIME OF THE TPI MANEUVER
#	(4)	TIG	TIME OF THE CSI MANEUVER
#	(5)	DELVEET1	DELTA VELOCITY AT CSI -- REFERENCE COORDINATES
#	(6)	DELVEET2	DELTA VELOCITY AT CDH -- REFERENCE COORDINATES
#	(7)	DIFFALT	DELTA ALTITUDE AT CDH
#	(8)	NN	NUMBER OF APSIDAL CROSSINGS THRU WHICH THE ACTIVE
#			VEHICLE ORBIT CAN BE ADVANCED TO OBTAIN THE CDH
#			MANEUVER POINT
#	(9)	ELEV	DESIRED LOS ANGLE AT TPI
#			
#	COMMUNICATION TO THRUSTING PROGRAM		
#	(1)	TIG	TIME OF THE CSI MANEUVER
#	(2)	RTIG	POSITION OF ACTIVE VEHICLE AT CSI -- BEFORE ROTATION
#			INTO PLANE OF PASSIVE VEHICLE
#	(3)	VTIG	VELOCITY OF ACTIVE VEHICLE AT CSE -- BEFORE ROTATION
#			INTO PLANE OF PASSIVE VEHICLE
#	(4)	DELVSIN	DELTA VELOCITY AT CSI -- REFERENCE COORDINATES
#	(5)	DELVSAB	MAGNITUDE OF DELTA VELOCITY AT CSI
#	(6)	XDELVFLG	SET TO INDICATE EXTERNAL DELTA V VG COMPUTATION
#			
#	SUBROUTINES USED		
#	AVFLAGA		



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```
# AVFLAGP
# P20FLGON
# VARALARM
# BANKCALL
# GOFLASH
# GOTOPOOH
```

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```

#      VNPOOH
#      GOFLASHR
#      BLANKET
#      ENDOFJOB
#      SELECTMU
#      ADVANCE
#      INTINT
#      PASSIVE
#      CSI/A
#      S32/33.1
#      DISDVLVC
#      VN1645

BANK    35
SETLOC  CSI/CDH
BANK
EBANK=  SUBEXIT
COUNT* $$/P3272
P32      TC      AVFLAGA
          TC      P32STRT
P72      TC      AVFLAGP
P32STRT  EXTEND
          DCA     P30ZERO
          DXCH    CENTANG
          TC      P32/P72A
ALMXITA  SXA,2
          CSIALRM
ALMXIT   LXC,1
          CSIALRM
          SLOAD*  EXIT
          ALARM/TB -1,1
          CA      MPAC
          TC      VARALARM
          CAF     V05N09
          TC      BANKCALL
          CADR    GOFLASH
          TC      GOTOP00H
          TC      -4
P32/P72A TC      P20FLGON
          CAF     P30ZERO
          TS      NN      +1
          TS      TCSI
          TS      TCSI    +1
VN0611   CAF     V06N11    # TCSI
          TC      VNPOOH

```

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TC	INTPRET
DLOAD	DCOMP
	TCSI
BMN	DLOAD
	VN0655

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		TETLEM	
	STCALL	TDEC1	
		PRECSET	
	VLOAD	VSR*	
		RACT3	
		0,2	
	STOVL	RVEC	
		VACT3	
	VSR*	SET	
		0,2	
		RVSW	
	STODL	VVEC	
		DPPOS MAX	
	STCALL	RDESIRED	
		TIMERAD	
	DAD		
		TDEC2	
	STORE	TCSI	
	EXIT		
	TC	VN0611	
VN0655	EXIT		
	CAF	V06N55	# NN, ELEV(RGLOS)
	TC	BANKCALL	
	CADR	GOFLASH	
	TC	GOTOP00H	
	TC	+2	
	TC	-5	
	CAF	V06N37	# TTPI
	TC	VNPO0H	
	TC	INTPRET	
	DLOAD		
		TCSI	
	STCALL	TIG	
		SELECTMU	
P32/P72B	CALL		
		ADVANCE	
	SETPD	VLOAD	
		OD	
		VPASS1	
	PDVL	PDDL	
		RPASS1	
		TCSI	
	PDDL	PDDL	
		TTPI	
		TWOPI	

		PUSH	CALL	
			INTINT	
		CALL		
			PASSIVE	
		CALL		
197	$\langle$ Page LM0623 197 $\rangle \equiv$			(187 751)
	P32/P72C	BON	CSI/A SET FINALFLG P32/P72D UPDATFLG	
	P32/P72D	DLOAD		
	P32/P72E	STORE DSU	T1TOT2 T1TOT2 BPL 60MIN P32/P72E	
		DLOAD		
	P32/P72F	STORE DSU	T2TOT3 T2TOT3 BPL 60MIN P32/P72F	
		EXIT		
		CAF	V06N75	
		TC	VNP00H	
		TC	INTPRET	
		VLOAD	CALL	
			DELVEET1	
			S32/33.1	
		STOVL	DELVEET1	
			RACT2	
		STOVL	RACT1	
			DELVEET2	
		AXT,1	CALL	
		VN	0682	
			DISDVLVC	
		DLOAD		
			TTPI	
		STCALL	TTPIO	
			VN1645	
		GOTO		
			P32/P72B	

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(187 751)

# CONSTANT DELTA HEIGHT (CDH) PROGRAMS (P33 AND P73)

# MOD NO -1

LOC SECTION -- P32-P35, P72-P75

# MOD BY WHITE, P.

DATE: 1 JUNE 67

#

# PURPOSE

#

# (1) TO CALCULATE PARAMETERS ASSOCIATED WITH THE CONSTANT DELTA  
# ALTITUDE MANEUVER (CDH).

#

# (2) TO CALCULATE THESE PARAMETERS BASED UPON MANEUVER DATA  
# APPROVED AND KEYED INTO THE DSKY BY THE ASTRONAUT.

#

# (3) TO DISPLAY TO THE ASTRONAUT AND THE GROUND DEPENDENT VARIABLES  
# ASSOCIATED WITH THE CDH MANEUVER FOR APPROVAL BY THE  
# ASTRONAUT/GROUND.

#

# (4) TO STORE THE CDH TARGET PARAMETERS FOR USE BY THE DESIRED  
# THRUSTING PROGRAM.

#

# ASSUMPTIONS

#

# (1) THIS PROGRAM IS BASED UPON PREVIOUS COMPLETION OF THE  
# CO-ELLIPTIC SEQUENCE INITIATION (CSI) PROGRAM (P32/P72).  
# THEREFORE --

#

# (A) AT A SELECTED TPI TIME (NOW IN STORAGE) THE LINE OF SIGHT  
# BETWEEN THE ACTIVE AND PASSIVE VEHICLES WAS SELECTED TO BE  
# A PRESCRIBED ANGLE (E) (NOW IN STORAGE) FROM THE  
# HORIZONTAL PLANE DEFINED BY THE ACTIVE VEHICLE POSITION.

#

# (B) THE TIME BETWEEN CSI IGNITION AND CDH IGNITION WAS  
# COMPUTED TO BE GREATER THAN 10 MINUTES.

#

# (C) THE TIME BETWEEN CDH IGNITION AND TPI IGNITION WAS  
# COMPUTED TO BE GREATER THAN 10 MINUTES.

#

# (D) THE VARIATION OF THE ALTITUDE DIFFERENCE BETWEEN THE  
# ORBITS WAS MINIMIZED.

#

# (E) CSI BURN WAS DEFINED SUCH THAT THE IMPULSIVE DELTA V WAS  
# IN THE HORIZONTAL PLANE DEFINED BY ACTIVE VEHICLE  
# POSITION AT CSI IGNITION.

#

# (F) THE PERICENTER ALTITUDES OF THE ORBITS FOLLOWING CSI AND  
# CDH WERE COMPUTED TO BE GREATER THAN 35,000 FT FOR LUNAR

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# ORBIT OR 85 NM FOR EARTH ORBIT.  
#  
# (G) THE CSI AND CDH MANEUVERS WERE ASSUMED TO BE PARALLEL TO  
# THE PLANE OF THE PASSIVE VEHICLE ORBIT. HOWEVER, CREW

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(187 751)

# MODIFICATION OF DELTA V (LV) COMPONENTS MAY HAVE RESULTED  
# IN AN OUT-OF-PLANE MANEUVER.

# (2) STATE VECTOR UPDATES BY P27 ARE DISALLOWED DURING AUTOMATIC  
# STATE VECTOR UPDATING INITIATED BY P20 (SEE ASSUMPTION 4).

# (3) COMPUTED VARIABLES MAY BE STORED FOR LATER VERIFICATION BY  
# THE GROUND. THESE STORAGE CAPABILITIES ARE NORMALLY LIMITED  
# ONLY TO THE PARAMETERS FOR ONE THRUSTING MANEUVER AT A TIME  
# EXCEPT FOR CONCENTRIC FLIGHT PLAN MANEUVER SEQUENCES.

# (4) THE RENDEZVOUS RADAR MAY OR MAY NOT BE USED TO UPDATE THE LM.  
# OR CSM STATE VECTORS FOR THIS PROGRAM. IF RADAR USE IS  
# DESIRED THE RADAR WAS TURNED ON AND LOCKED ON THE CSM BY  
# PREVIOUS SELECTION OF P20. RADAR SIGHTING MARKS WILL BE MADE  
# AUTOMATICALLY APPROXIMATELY ONCE A MINUTE WHEN ENABLED BY THE  
# TRACK AND UPDATE FLAGS (SEE P20). THE RENDEZVOUS TRACKING  
# MARK COUNTER IS ZEROED BY THE SELECTION OF P20 AND AFTER EACH  
# THRUSTING MANEUVER.

# (5) THE ISS NEED NOT BE ON TO COMPLETE THIS PROGRAM.

# (6) THE OPERATION OF THE PROGRAM UTILIZES THE FOLLOWING FLAGS --

# ACTIVE VEHICLE FLAG -- DESIGNATES THE VEHICLE WHICH IS  
# DOING RENDEZVOUS THRUSTING MANEUVERS TO THE PROGRAM WHICH  
# CALCULATES THE MANEUVER PARAMETERS. SET AT THE START OF  
# EACH RENDEZVOUS PRE-THRUSTING PROGRAM.

# FINAL FLAG -- SELECTS FINAL PROGRAM DISPLAYS AFTER CREW HAS  
# COMPLETED THE FINAL MANEUVER COMPUTATION AND DISPLAY  
# CYCLE.

# EXTERNAL DELTA V STEERING FLAG -- DESIGNATES THE TYPE OF  
# STEERING REQUIRED FOR EXECUTION OF THIS MANEUVER BY THE  
# THRUSTING PROGRAM SELECTED AFTER COMPLETION OF THIS  
# PROGRAM.

# (7) IT IS NORMALLY REQUIRED THAT THE ISS BE ON FOR 1 HOUR PRIOR TO  
# A THRUSTING MANEUVER.

# (8) THIS PROGRAM IS SELECTED BY THE ASTRONAUT BY DSKY ENTRY.

# P33 IF THIS VEHICLE IS ACTIVE VEHICLE.

#



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```
# P73 IF THIS VEHICLE IS PASSIVE VEHICLE.
#
# INPUT
#
# (1) TTPIO TIME OF THE TPI MANEUVER -- SAVED FROM P32/P72
```

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```

#      (2)      ELEV      DESIRED LOS ANGLE AT TPI -- SAVED FROM P32/P72
#      (3)      TCDH      TIME OF THE CDH MANEUVER
#
# OUTPUT
#
#      (1)      TRKMKCNT      NUMBER OF MARKS
#      (2)      TTOGO        TIME TO GO
#      (3)      +MGA         MIDDLE GIMBAL ANGLE
#      (4)      DIFFALT      DELTA ALTITUDE AT CDH
#      (5)      T2TOT3       DELTA TIME FROM CDH TO COMPUTED TPI
#      (6)      NOMTPI       DELTA TIME FROM NOMINAL TPI TO COMPUTED TPI
#      (7)      DELVLVC      DELTA VELOCITY AT CDH -- LOCAL VERTICAL COORDINATES
#
# DOWNLINK
#
#      (1)      TCDH         TIME OF THE CDH MANEUVER
#      (2)      TTPI        TIME OF THE TPI MANEUVER
#      (3)      TIG         TIME OF THE CDH MANEUVER
#      (4)      DELLVEET2    DELTA VELOCITY AT CDH -- REFERENCE COORDINATES
#      (5)      DIFFALT      DELTA ALTITUDE AT CDH
#      (6)      ELEV        DESIRED LOS ANGLE AT TPI
#
# COMMUNICATION TO THRUSTING PROGRAMS
#
#      (1)      TIG         TIME OF THE CDH MANEUVER
#      (2)      RTIG        POSITION OF ACTIVE VEHICLE AT CDH -- BEFORE ROTATION
#                               INTO PLANE OF PASSIVE VEHICLE.
#      (3)      VTIG        VELOCITY OF ACTIVE VEHICLE AT CDH -- BEFORE ROTATION
#                               INTO PLANE OF PASSIVE VEHICLE.
#      (4)      DELVSIN      DELTA VELOCITY AT CDH -- REFERENCE COORDINATES.
#      (5)      DELVSAB      MAGNITUDE OF DELTA VELOCITY AT CDH.
#      (6)      XDELVFLG     SET TO INDICATE EXTERNAL DELTA V VG COMPUTATION.
#
# SUBROUTINES USED
#
#      AVFLAGA
#      AVFLAGP
#      P20FLGON
#      VNPOOH
#      SELECTMU
#      ADVANCE
#      CDHMVR
#      INTINT3P
#      ACTIVE
#      PASSIVE

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# S33/S34.1  
# ALARM  
# BANKCALL  
# GOFLASH  
# GOTOP00H  
# S32/33.1

204  $\langle \text{Page } LM0627 \text{ } 204 \rangle \equiv$   
 # VN1645

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	COUNT*	\$\$/P3373	
P33	TC	AVFLAGA	
	TC	P33/P73A	
P73	TC	AVFLAGP	
P33/P73A	TC	P20FLGON	
	CAF	V06N13	# TCDH
	TC	VNP00H	
	TC	INTPRET	
	DLOAD		
		TTPIO	
	STODL	TTPI	
		TCDH	
	STCALL	TIG	
		SELECTMU	
P33/P73B	CALL		
		ADVANCE	
	CALL		
		CDHMVR	
	SETPD	VLOAD	
		OD	
		VACT3	
	PDVL	CALL	
		RACT2	
		INTINT3P	
	CALL		
		ACTIVE	
	SETPD	VLOAD	
		OD	
		VPASS2	
	PDVL	CALL	
		RPASS2	
		INTINT3P	
	CALL		
		PASSIVE	
	DLOAD	SET	
		P30ZERO	
		ITSWICH	
	STCALL	NOMTPI	
		S33/34.1	
	BZE	EXIT	
		P33/P73C	
	TC	ALARM	
	OCT	611	

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CAF	V05N09
TC	BANKCALL
CADR	GOFLASH
TC	GOTOP00H
TC	+2

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	TC	P33/P73A
	TC	INTPRET
	DLOAD	
		P30ZERO
	STORE	NOMTPI
P33/P73C	BON	SET
		FINALFLG
		P33/P73D
P33/P73D	DLOAD	UPDATFLG
		DAD
		NOMTPI
		TTPI
	STORE	TTPI
	DSU	
P33/P73E		TCDH
	DSU	BPL
		60MIN
		P33/P73E
	DAD	
		60MIN
	STODL	T1TOT2
		TTPI
	DSU	PUSH
		TTPIO
P33/P73F	ABS	DSU
		60MIN
	BPL	DAD
		P33/P73F
		60MIN
	SIGN	STADR
	STORE	T2TOT3
	EXIT	
	CAF	V06N75
	TC	VNP00H
	TC	INTPRET
	VLOAD	CALL
		DELVEET2
		S32/33.1
	STCALL	DELVEET2
		VN1645
	GOTO	
		P33/P73B

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```
# ***** ADFLAG/P *****
#
# SUBROUTINES USED
#
#         UPFLAG
#         DOWNFLAG

AVFLAGA      EXTEND      # AVFLAG = LEM
               QXCH      SUBEXIT
               TC         UPFLAG
               ADRES      AVFLAG
               TC         SUBEXIT
AVFLAGP      EXTEND      # AVFLAG = CSM
               QXCH      SUBEXIT
               TC         DOWNFLAG
               ADRES      AVFLAG
               TC         SUBEXIT
P20FLGON     EXTEND
               QXCH      SUBEXIT
               TC         UPFLAG
               ADRES      UPDATFLG      # SET UPDATFLG
               TC         UPFLAG
               ADRES      TRACKFLG      # SET TRACKFLG
               TC         SUBEXIT
```

208  $\langle$ Page LM0630 208 $\rangle \equiv$  (187 751)

```

# ***** DISDVLVC *****
#
# SUBROUTINES USED
#
#      S32/33.X
#      VNPOOH

DISDVLVC      STORE  DELVLVC
               STQ    CALL
               NORMEX
               S32/33.X
               VLOAD  MXV
               DELVLVC
               OD
               VSL1   SXA,1
               VERBNOUN
               STORE  DELVLVC
               EXIT
               CA     VERBNOUN
               TC     VNPOOH
               TC     INTPRET
               GOTO
               NORMEX

```



209      $\langle \text{Page } LM0631 \text{ 209} \rangle \equiv$      (187 751)

# \*\*\*\*\* CONSTANTS \*\*\*\*\*

V06N11	VN	0611
V06N13	VN	0613
V06N75	VN	0675
SN359+	2DEC	-.000086601

CS359+	2DEC	+.499999992
--------	------	-------------

P30ZERO	2DEC	0
---------	------	---

60MIN	2DEC	360000
-------	------	--------

ALARM/TB	OCT	00600	# NO	1
	OCT	00601	#	2
	OCT	00602	#	3
	OCT	00603	#	4
	OCT	00604	#	5
	OCT	00605	#	6
	OCT	00606	#	7

210  $\langle$ Page LM0632 210 $\rangle \equiv$ 

(187 751)

# \*\*\*\*\* CSI/A \*\*\*\*\*

#

# SUBROUTINES USED

#

# VECSHIFT

# TIMETHET

# PERIAP0

# SHIFTR1

# INTINT2C

# CDHMVR

# PERIAP01

# INTINT

# ACTIVE

BANK 34

SETLOC CSI/CDH1

BANK

EBANK= SUBEXIT

COUNT\* \$\$/CSI

LOOPMX 2DEC 16

INITST 2DEC .03048 B-7 # INITIAL DELDV = 10 FPS

DVMAX1 2DEC 3.0480 B-7 # MAXIMUM DV1 = 1000 FPS

DVMAX2 2DEC 3.014472 B-7 # 989 FPS

1DPB2 2DEC 1.0 B-2

1DPB28 2DEC 1

PMINE 2DEC 157420 B-29 # 85 NM -- MUST BE 8 WORDS BEFORE PMINM

EPSILN1 2DEC .0003048 B-7 # .1 FPS

NICKELDP 2DEC .021336 B-7 # 7 FPS (CHANGED FROM .05 FPS)

FIFPSDP 2DEC -.152400 B-7 # 50 FPS

PMINM 2DEC 10668 B-29 # 35000 FT -- MUST BE 8 WORDS AFTER PMINE

DELMAX1 2DEC .6096000 B-7 # 200 FPS

ONETHTH 2DEC .0001 B-3

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TMIN	2DEC	60000	# 10 MIN
CSI/A	CLEAR	SET	# INITIALIZE INDICATORS
		S32.1F1	# DVT1 HAS EXCEEDED MAX INDICATOR
		S32.1F2	# FIRST PASS FOR NEWTON ITERATION INDICATOR

212 (Page LM0633 212)≡

(187 751)

	CLEAR	SET		
		S32.1F3A	#	00=1ST 2 PASSES 2ND CYCLE, 01=FIRST CYCLE
		S32.1F3B	#	10=2ND CYCLE, 11=50 FPS STAGE 2ND CYCLE
	DLOAD	P30ZERO		
	STORE	LOOPCT		
	STORE	CSIALRM		
CSI/B	SETPD	VLOAD		
		OD		
		RACT1		
	ABVAL	PUSH	# RA1	B29 PL02D
	NORM	SR1		
		X2	#	B29-N2+ B1 PL04D
	PDVL	ABVAL		
		RPASS3		
	NORM	BDDV	# RA1/RP3	B1 PL02D
		X1		
	XSU,2	SR*	#	B2
		X1		
		1,2		
	DAD	DMP	# (1+(RA1/RP3))RA1	B29+B2=B31 PL00D
		1DPB2		
	NORM	PDDL	#	PL02D
		X1		
		RTMU		
	SR1	DDV	#	B38-B31= B7 PL00D
	SL*	SQRT	#	B7
		0	-7,1	
	PDVL	UNIT	#	PL02D
		RACT1		
	PDVL	VXV		
		UP1		
	UNIT		#	UNIT(URP1 X UVP1 X URA1) = UH1
	DOT	SL1	#	VA1 . UH1 B7
		VACT1		
	BDSU	STADR	#	PL00D
	STODL	DELVCSI		
		INITST	#	10 FPS
	STORE	DELDV		
CSI/B1	DLOAD	DAD	#	IF LOOPCT = 16
		LOOPCT		
		1DPB28		
	STORE	LOOPCT		
	DSU	AXT,2		
		LOOPMX		

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6

BPL

SCNDSOL

CSI/B2

SETPD

OD

214 (Page LM0634 214)≡

(187 751)

	DLOAD	ABS	
		DELVCSI	
	DSU	BMN	
		DVMAX1	
		CSI/B23	
	AXT,2	BON	
		7	
		S32.1F1	
		SCNDSOL	
	BOFF	BON	
		S32.1F3A	
		CSI/B22	# FLAG 3 NEQ 3
		S32.1F3B	
		SCNDSOL	
CSI/B22	SET	DLOAD	
		S32.1F1	
		DVMAX2	
	SIGN		
		DELVCSI	
	STORE	DELVCSI	
CSI/B23	VLOAD	PUSH	
		RACT1	
	UNIT	PDVL	
		UP1	
	VXV	UNIT	# UNIT (URP1 X UVP1 X URA1) = UH1
	VXSC	VSL1	
		DELVCSI	
	STORE	DELVEET1	
	VAD	BOV	
		VACT1	
		CSI/B23D	
CSI/B23D	STCALL	VACT4	
		VECSHIFT	
	STOVL	VVEC	
	SET		
		RVS	
	STOVL	RVEC	
		SN359+	
	STCALL	SNTH	# ALSO CSTD
		TIMETHET	
	SR1	LXA,1	
		RTX1	
	STCALL	HAFPA1	
		PERIAPO	
	CALL		

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	SHIFTR1
STODL	POSTCSI
	CENTANG
BZE	GOTO
	+2

216  $\langle \text{Page LM0635 216} \rangle \equiv$ 

(187 751)

		CIRCL		
	DLOAD	ECC		
	DSU	BMN		
		ONETHTH		
		CIRCL		
	DLOAD	CALL		
		R1		
		SHIFTR1		
	SETPD	NORM		
		2D		
		X1		
	PDVL	DOT	#	PL041
		RACT1		
		VACT4		
	ABS	DDV		
		02D	# (/RDOTV/)/R1	B38-B29= B7
	SL*	DSU		
		0,1		
		NICKELDP		
	BMN	DLOAD		
		CIRCL		
		P		
	SL2	DSU		
		1DPB2		
	STODL	14D		
		RTSR1/MU		
	SR1	DDV	# (1/ROOTMU)/R1	B-16-B29 = B-45 PL021
	PDDL	DMP		
		P		
		R1		
	CALL			
		SHIFTR1		
	SL4	SL1		
	SQRT	DMP	# ((P/MU)**.5)/R1	B14+B-14 = B-31 PL021
	BOFF	SL3		
		CMOONFLG		
		CSI/B3		
CSI/B3	PDVL	DOT		
		RACT1		
		VACT4		
	STORE	RDOTV		
	ABS			
	NORM	DMP	# ((P/MU)**.5)RDOTV/R1	PL021
		X2		



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XSU,1	SL*	#	B-31+B36-B3 = B2
	X2		
	3,1		
STODL	12D		
	P30ZERO		

218  $\langle$ Page LM0636 218 $\rangle \equiv$  (187 751)

	STORE	16D	
	VLOAD	UNIT	
		12D	
	STOVL	SNTH	# ALSO STORES CSTH AND 0
		RACT1	
	PDVL	SIGN	
		VACT4	
		RDOTV	
	VCOMP	CALL	
		VECSHIFT	
	STOVL	VVEC	
	SET		
		RVSW	
	STCALL	RVEC	
		TIMETHET	
	PDDL	BPL	
		RDOTV	
		NTP/2	
	DLOAD	DSU	
		HAFPA1	
	PUSH	GOTO	
		NTP/2	
CIRCL	SETPD	DLOAD	
		OOD	
		P30ZERO	
	PUSH		
NTP/2	DLOAD	DMP	
		NN	
		HAFPA1	
	SL	DSU	
		14D	
	DAD		
		TCSI	
	STORE	TCDH	
	BDSU	AXT, 2	
		TTPI	
		5D	
	BMN	SETPD	
		SCNDSOL	
		OD	
	VLOAD	PDVL	
		VACT4	
		RACT1	
	CALL		
		INTINT2C	

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STOVL	RACT2
	VATT
STOVL	VACT2
	VPASS1
SETPD	PDVL

220  $\langle$ Page LM0637 220 $\rangle \equiv$ 

(187 751)

	OD		
	RPASS1		
CALL			
	INTINT2C		
STOVL	RPASS2		
	VATT		
STCALL	VPASS2		
	CDHMVR		
VLOAD	SETPD		
	RACT2		
	OD		
PDVL	CALL		
	VACT3		
	PERIAP01		
CALL			
	SHIFTR1		
STOVL	POSTCDH		
	VACT3		
SETPD	PDVL		
	OD		
	RACT2		
PDDL	PDDL		
	TCDH		
	TTPI		
PDDL	PUSH		
	TWOPI		
CALL			
	INTINT		
CALL			
	ACTIVE		
DLOAD			
	ELEV		
SETPD	SINE		
	6D		
PDVL	UNIT		
	RACT3		
STORE	00D	# URA3 AT 00D	
PDVL	VXV	# PL14D, PL08D	
	UP1		
UNIT			
PDDL	COSINE	# UNIT(URA3 X UVA3 X URA3) = UH3	B1 PL
	ELEV		
VXSC	STADR	# (COSLOS)(UH3)	B2 PL
STORE	18D	# PLUS	
DLOAD	VXSC	# (SINLOS)(URA3) = U	B2 PL

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VAD	VSL1	#	B1
	18D	#	PL06D
PUSH	DOT	# (U . RA3) = TEMP1	B1 +B29 = B30
	RACT3	#	B29 PL08D
SL1	PUSH		

222 (Page LM0638 222)≡

(187 751)

	DSQ	TLOAD	# TEMP1**2	B58
		MPAC		
	PDVL	DOT	#	PI
		RACT3		
		RACT3		
	TLOAD	DCOMP	# RA3 . RA3	
		MPAC		
	PDVL	DOT	# RP3 . RP3	B58 PI
		RPASS3		
		RPASS3	#	PI
	TAD	TAD	# TEMP1**2 + RA3.RA3 + RP3.RP3 = TEMP2	PI
	BPL	DLOAD		
		K10RK2		
		LOOPCT		
	DSU	AXT,2		
		1DPB28		
		1D		
	BZE			
		ALMXITA		
	DLOAD	SR1		
		DELDV		
	STORE	DELDV		
	BDSU			
		DVPREV		
	STCALL	DELVCSI		
		CSI/B1		
K10RK2	SQRT	PUSH	# TEMP3 = TEMP2**.5	B29 PI
	DCOMP	DSU		
		06D	# -TEMP1-TEMP3 = K2 AT 10D	
	STODL	10D	#	PI
	DSU	STADR	#	PI
	STORE	12D	# -TEMP1+TEMP3 = K1 AT 12D	
	ABS			
	STODL	14D		
		10D		
	ABS	DSU		
		14D		
	BMN	DLOAD		
		K2.		
		12D		
	STORE	10D	# K = K1	
K2.	DLOAD			
		10D		
	VXSC	VSL1		
	VAD	UNIT	# V = RA3 + KU UNIT	B1

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	RACT3		
PDVL	UNIT		
	RPASS3	#	PL06D
PDVL	UNIT		
	VPASS3	#	PL12D

224 (Page LM0639 224)≡

(187 751)

```

VXV    PDVL    # UVP3 X URP3
        06D
        06D
VXV    DOT
        OOD
STADR
STOVL   12D    # (URP3 X V).(URP3 X URP3)=TEMP
DOT     SL1    #
ARCCOS  SIGN
        12D    #
SR1     PUSH   # GAMMA = SIGN(TEMP)ARCOS(UNITV.URP3)
BON     DLOAD
        S32.1F2
        FRSTPAS
        OOD    # NOT THE FIRST PASS OF A CYCLE
DSU     PDDL   # GAMMA-GAMPREV
        GAMPREV
        DELVCSI
DSU     NORM   #
        DVPREV
        X1
BDDV    PDDL   # (GAM-GAMPREV)/(DV-DVPREV)
        02D    # = SLOPE
        DELVCSI
STORE   DVPREV
BOFF    BOFF
        S32.1F3A
        THRDCHK
        S32.1F3B
        THRDCHK
DLOAD   DMP
        02D
        GAMPREV
BPL     DLOAD
        FIFTYFPS
        INITST
SIGN
        DELDV
STORE   DELDV
SET     CLEAR
        S32.1F3A
        S32.1F3B
FRSTPAS DLOAD
        OOD
STODL   GAMPREV

```



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	DELVCSI
STORE	DVPREV
DSU	CLEAR
	DELDV
	S32.1F2

226 (Page LM0640 226)≡

(187 751)

	STCALL	DELVCSI
		CSI/B1
THRDCHK	BON	BON
		S32.1F3A
		NEWTN
		S32.1F3B
		NEWTN
FIFTYFPS	DLOAD	SIGN
		FIFPSDP
		04D
	SIGN	
		GAMPREV
	STORE	DELDV
	DCOMP	DAD
		DELVCSI
	STODL	DELVCSI
		OOD
	SET	SET
		S32.1F3B
		S32.1F3A
	STCALL	GAMPREV
		CSI/B2
NEWTN	DLOAD	NORM
		04D
		X2
	BDDV	XSU,1
		OOD
		X2
	SR*	
		0,1
	STODL	DELDV
		OOD
	STORE	GAMPREV
	DLOAD	ABS
		DELDV
	PUSH	DSU
		EPSILN1
	BMN	DLOAD
		CSI/SOL
	DSU	BMN
		DELMAX1
		CSISTEP
	DLOAD	SIGN
		DELMAX1
		DELDV

#

PI

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CSISTEP	STORE	DELDV
	DLOAD	DSU
		DELVCSI
		DELDV
	STCALL	DELVCSI

228 &lt;Page LM0641 228&gt;≡

(187 751)

		CSI/B1
CSI/SOL	DLOAD	AXT,2
		POSTCSI
		2
	LXA,1	
		RTX1
	DSU*	BMN
		PMINE -2,1
		SCNDSOL
	AXT,2	DLOAD
		3
		POSTCDH
	DSU*	BMN
		PMINE -2,1
		SCNDSOL
	DLOAD	DSU
		TCDH
		TCSI
	STORE	T1TOT2
	AXT,2	DSU
		4
		TMIN
	BMN	AXT,2
		SCNDSOL
		5
	DLOAD	DSU
		TTPI
		TCDH
	STORE	T2TOT3
	DSU	BPL
		TMIN
		P32/P72C
SCNDSOL	BON	BOFF
		S32.1F3A
		ALMXIT
		S32.1F3B
		ALMXIT
	SXA,2	DLOAD
		CSIALRM
		P30ZERO
	CLEAR	SET
		S32.1F1
		S32.1F2
	CLEAR	CLEAR
		S32.1F3A

S32.1F3B  
 STCALL LOOPCT  
 CSI/B

229

$\langle \text{Page } LM0642 \text{ 229} \rangle \equiv$

(187 751)

```
# ***** ADVANCE *****
#
# SUBROUTINES USED
#     PRECSET
#     ROTATE

ADVANCE      STQ      DLOAD
                SUBEXIT
                TIG
                STCALL TDEC1
                PRECSET
                SET     VLOAD
                XDELVFLG
                VPASS3
                STORE  VPASS2
                STOVL  VPASS1
                RPASS3
                STORE  RPASS2
                STORE  RPASS1
                UNIT   VXV
                VPASS1
                UNIT
                STOVL  UP1
                RACT3
                STCALL RTIG
                ROTATE
                STORE  RACT2
                STOVL  RACT1
                VACT3
                STCALL VTIG
                ROTATE
                STORE  VACT2
                STCALL VACT1
                SUBEXIT
```

230a  $\langle \text{Page } LM0643 \text{ 230a} \rangle \equiv$  (187 751)

# \*\*\*\*\* ROTATE \*\*\*\*\*

ROTATE	PUSH	PUSH
	DOT	VXSC
		UP1
		UP1
	VSL2	BVSU
	UNIT	PDVL
	ABVAL	VXSC
	VSL1	RVQ

230b  $\langle \text{Page } LM0644 \text{ 230b} \rangle \equiv$  (187 751)

# \*\*\*\*\* INTINTNA \*\*\*\*\*

INTINT2C	PDDL	PDDL
		TCSI
		TCDH
	PDDL	PUSH
		TWOPI
	GOTO	
		INTINT
INTINT3P	PDDL	PDDL
		TCDH
		TTPI
	PDDL	PUSH
		P30ZERO
	GOTO	
		INTINT

231a  $\langle$ Page LM0645 231a $\rangle \equiv$  (187 751)

```
# ***** S32/33.1 *****
#
# SUBROUTINES USED
#       S32/33.X
```

```
S32/33.1      STQ      AXT,1
                SUBEXIT
                VN      0681
                CALL
                DISDVLVC
                CALL
                S32/33.X
                VLOAD   VXM
                DELVLVC
                OD
                VSL1
                STORE   DELVSIN
                PUSH    ABVAL
                STOVL   DELVSAB
                GOTO
                SUBEXIT
```

231b  $\langle$ Page LM0646 231b $\rangle \equiv$  (187 751)

```
# ***** S32/33.X *****
```

```
S32/33.X      SETPD   VLOAD
                6D
                UP1
                VCOMP  PDVL
                RACT1
                UNIT   VCOMP
                PUSH   VXV
                UP1
                VSL1
                STORE  OD
                RVQ
```

232  $\langle \text{Page LM0647 232} \rangle \equiv$  (187 751)

# \*\*\*\*\* CDHMVR \*\*\*\*\*

#

# SUBROUTINES USED

# VECSHIFT

# TIMETHET

# SHIFTR1

CDHMVR	STQ	VLOAD	
		SUBEXIT	
		RACT2	
	PUSH	UNIT	
	STOVL	UNVEC	# UR SUB A
		RPASS2	
	UNIT	DOT	
		UNVEC	
	PUSH	SL1	
	STODL	CSTH	
	DSQ	PDDL	
		DP1/4TH	
	SR2	DSU	
	SQRT	SL1	
	PDVL	VCOMP	
	VXV		
		RPASS2	
	DOT	PDDL	
		UP1	
	SIGN	STADR	
	STOVL	SNTH	
		RPASS2	
	PDVL	CALL	
		VPASS2	
		VECSHIFT	
	STOVL	VVEC	
	CLEAR		
		RVSU	
	STCALL	RVEC	
		TIMETHET	
	LXA,2	VSL*	
		RTX2	
		O,2	
	STORE	18D	
	DOT	SL1R	
		UNVEC	
	PDVL	ABVAL	# OD = V SUB PV
	SL*	PDVL	



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	0,2	
	RACT2	
ABVAL	PDDL	# 2D = LENGTH OF R SUB A
DSU		

234 (Page LM0648 234)≡

(187 751)

	02D				
STODL	DIFFALT	# DELTA H IN METERS	B+29		
	R1A				
NORM	PDDL	# 2 - R V**/MU		04D	
	X1				
	R1				
CALL					
	SHIFTR1				
SR1R	DDV				
SL*	PUSH				
	0	-5,1			
DSU	PDDL	# A SUB A	B+29	04D	
	DIFFALT				
SR2	DDV	# A SUB P	B+31		
	04D	#	B+2		
PUSH	SQRT	# A SUB P/A SUB A		06D	
DMPR	DMP				
	06D				
	00D				
SL3R	PDDL	# V SUB AV METERS/CS	B+7	08D	
	02D	# R SUB A MAGNITUDE	B+29		
NORM	PDDL				
	X1				
	RTMU				
SR1	DDV	# 2MU	B+38		
SL*	PDDL	# 2 MU/R SUBAA	B+14	10D	
	0	-5,1			
	04D	# ASUBA	B+29		
NORM	PDDL				
	X2				
	RTMU				
SR1	DDV				
SL*	BDSU				
	0	-6,2	# 2U/R - U/A	B+14 (METERS/CS)SQ	
PDDL	DSQ	#		10D	
	08D				
BDSU	SQRT				
PDVL	VXV	# SQRT(MU(2/R SUB A-1/A SUB A)-VSUBA2)		10D	
	UP1				
	UNVEC				
UNIT	VXSC				
	10D				
PDVL	VXSC				
	UNVEC				
	08D				

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VAD VSL1  
STADR  
STORE VACT3  
VSU  
VACT2

235a  $\langle$ Page LM0649 235a $\rangle \equiv$  (187 751)  
STCALL DELVEET2 # DELTA VCDH -- REFERENCE COORDINATES  
SUBEXIT

235b  $\langle$ Page LM0650 235b $\rangle \equiv$  (187 751)  
# \*\*\*\*\* COMPTGO \*\*\*\*\*  
#  
# SUBROUTINES USED  
# CLOKTASK  
# 2PHSCHNG

BANK 35  
SETLOC CSI/CDH  
BANK

EBANK= RTRN  
COUNT\* \$\$/P3575

COMPTGO  
EXTEND  
QXCH RTRN  
CAF ZERO  
TS DISPDEX  
CAF BIT2  
INHINT  
TC WAITLIST  
EBANK= WHICH  
2CADR CLOKTASK  
  
TC 2PHSCHNG  
OCT 40036  
OCT 05024  
OCT 13000  
TC RTRN

## 1.16 lambert aimpoint guidance

$$\begin{aligned}
 236 \quad \langle \textit{lambert aimpoint guidance } 236 \rangle &\equiv & (7) \\
 &\langle \textit{Page LM0651 } 237 \rangle \\
 &\langle \textit{Page LM0652 } 239 \rangle \\
 &\langle \textit{Page LM0653 } 240 \rangle
 \end{aligned}$$

```
# GENERAL LAMBERT AIMPOINT GUIDANCE **
# WRITTEN BY RAMA M AIYAWAR

# PROGRAM P-31 DESCRIPTION **
#
# 1.    TO ACCEPT TARGETING PARAMETERS OBTAINED FROM A SOURCE EXTERNAL
#       TO THE LEM AND COMPUTE THERE FROM THE REQUIRED-VELOCITY AND
#       OTHER INITIAL CONDITIONS REQUIRED BY LM FOR DESIRED MANEUVER.
#       THE TARGETING PARAMETERS ARE TIG (TIME OF IGNITION), TARGET
#       VECTOR (RTARG), AND THE TIME FROM TIG UNTIL THE TARGET IS
#       REACHED (DELT4), DESIRED TIME OF FLIGHT FROM RINIT TO RTARG.

# ASSUMPTIONS **
#
# 1.    THE TARGET PARAMETERS MAY HAVE BEEN LOADED PRIOR TO THE
#       EXECUTION OF THIS PROGRAM.
# 2.    THIS PROGRAM IS APPLICABLE IN EITHER EARTH OR LUNAR ORBIT.
# 3.    THIS PROGRAM IS DESIGNED FOR ONE-MAN OPERATION, AND SHOULD
#       BE SELECTED BY THE ASTRONAUT BY DSKY ENTRY V37 E31.

# SUBROUTINES USED **
#
# MANUPARM, TTG/N35, R02BOTH, MIDGIM, DISPMGA, FLAGDOWN, BANKCALL,
# GOTOPOOH, ENDOFJOB, PHASCHNG, GOFLASHR, GOFLASH.
#
# MANUPARM      CALCULATES APOGEE, PERIGEE ALTITUDES AND DELTAV DESIRED
#               FOR THE MANEUVER.
#
# TTG/N35       CLOCKTASK - UPDATES CLOCK.
#
# MIDGIM        CALCULATES MIDDLE GIMBAL ANGLE FOR DISPLAY.
#
# R02BOTH       IMU - STATUS CHECK ROUTINE.

# DISPLAYS USED IN P-31LM **
#
# V06N33        DISPLAY SOTRED TIG (IN HRS. MINS. SECS.)
# V06N42        DISPLAY APOGEE, PERIGEE, DELTAV.
# V16N35        DISPLAY TIME FROM TIG.
# V06N45        TIME FROM IGNITION AND MIDDLE GIMBAL ANGLE.

# ERASABLE INITIALIZATION REQUIRED **
#
# TIG           TIME OF IGNITION           DP      (B+28) CS.
```

```
#
# DELT4      DESIRED TIME OF FLIGHT      DP      (B+28) CS
#            FROM RINIT TO RTARG.
#
# RTARG      RADIUS VECTOR OF TARGET POSITION VECTOR
#            RADIUS VECTOR SCALED TO (B+29)METERS IF EARTH ORBIT
```

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```

#          RADIUS VECTOR SCALED TO (B+27)METERS IF MOON ORBIT

# OUTPUT **
#
# HAPO          APOGEE ALTITUDE
# HPER          PERIGEE ALTITUDE
# VGDISP        MAG. OF DELTAV FOR DISPLAY, SCALING      B+7 M/CS EARTH
#              MAG. OF DELTAV FOR DISPLAY, SCALING      B+5 M/CS MOON
# MIDGIM        MIDDLE GIMBAL ANGLE
# XDELVFLG      RESETS XDELVFLG FOR LAMBERT VG COMPUTATIONS

# ALARMS OR ABORTS      NONE **

# RESTARTS ARE VIA GROUP 4 **

          SETLOC  GLM
          BANK

          EBANK=  SUBEXIT

          COUNT*  $$/P31
P31        TC      P20FLGON
          CAF      V06N33      # T16
          TC      VNP00H
          TC      INTPRET
          CLEAR    DLOAD
          UPDATFLG
          TIG
          STCALL   TDEC1      # INTEGRATE STATE VECTORS TO TIG
          LEMPREC
          VLOAD    SETPD
          RATT
          OD
          STORE    RTIG
          STOVL    RINIT
          VATT
          STORE    VTIG
          STODL    VINIT
          P30ZERO
          PUSH     PDDL      # E4 AND NUMIT = 0
          DELLT4
          DAD      SXA,1
          TIG
          RTX1
          STORE    TPASS4

```

```

          SXA,2  CALL
                RTX2
                INITVEL
          VLOAD  PUSH

240  <Page LM0653 240>≡                                     (236 742)
                DELVEET3
          STORE  DELVSIN
          ABVAL  CLEAR
                XDELVFLG
          STCALL VGDISP
                GET.LVC
          VLOAD  PDVL
                RTIG
                VIPRIME
          CALL
                PERIAP01
          CALL
                SHIFTR1
          CALL                                     # LIMIT DISPLAY TO 9999.9 N. MI.
                MAXCHK
          STODL  HPER
                4D
          CALL
                SHIFTR1
          CALL                                     # LIMIT DISPLAY TO 9999.9 N. MI.
                MAXCHK
          STORE  HAPO
          EXIT
          CAF    V06N81                                     # DELVLVC
          TC     VNPO0H
          CAF    V06N42                                     # HAPO, HPER, VGDISP
          TC     VNPO0H
          TC     INTPRET
          REVN1645 SET  CALL                                     # TRKMKCNT, TTOGO, +MGA
                FINALFLG
                VN1645
          GOTO
                REVN1645

```

```

# *** END OF LEMP30S .103 ***

```



## 1.17 burn-baby-burn master ignition routine

241  $\langle \text{burn-baby-burn master ignition routine 241} \rangle \equiv$  (7)

$\langle \text{Page LM0731 242} \rangle$   
 $\langle \text{Page LM0732 244} \rangle$   
 $\langle \text{Page LM0733 245} \rangle$   
 $\langle \text{Page LM0734 246} \rangle$   
 $\langle \text{Page LM0735 248} \rangle$   
 $\langle \text{Page LM0736 250} \rangle$   
 $\langle \text{Page LM0737 252} \rangle$   
 $\langle \text{Page LM0738 254} \rangle$   
 $\langle \text{Page LM0739 256} \rangle$   
 $\langle \text{Page LM0740 258} \rangle$   
 $\langle \text{Page LM0741 260} \rangle$   
 $\langle \text{Page LM0742 262} \rangle$   
 $\langle \text{Page LM0743 264} \rangle$   
 $\langle \text{Page LM0744 266} \rangle$   
 $\langle \text{Page LM0745 267} \rangle$   
 $\langle \text{Page LM0746 269} \rangle$   
 $\langle \text{Page LM0747 271} \rangle$   
 $\langle \text{Page LM0748 273} \rangle$   
 $\langle \text{Page LM0749 274} \rangle$   
 $\langle \text{Page LM0750 275} \rangle$   
 $\langle \text{Page LM0751 277} \rangle$

HONI SOIT QUI MAL Y PENS:

"May he be shamed who thinks badly of it"

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(241 722)

# BURN, BABY, BURN -- MASTER IGNITION ROUTINE

BANK 36  
SETLOC P40S  
BANK  
EBANK= WHICH  
COUNT\* \$\$/P40

# THE MASTER IGNITION ROUTINE IS DESIGNED FOR USE BY THE FOLLOWING LEM PROGRAMS: P12  
# IT PERFORMS ALL FUNCTIONS IMMEDIATELY ASSOCIATED WITH APS OR DPS IGNITION: IN PART  
# BETWEEN THE PRE-IGNITION TIME CHECK -- ARE WE WITHIN 45 SECONDS OF TIG? -- AND TIG  
# PROGRAMS THROTTLE UP.

#

# VARIATIONS AMONG PROGRAMS ARE ACCOMODATED BY MEANS OF TABLES CONTAINING CONSTANTS (P12  
# WAITLIST, FOR PINBALL) AND TCF INSTRUCTIONS. USERS PLACE THE ADRES OF THE HEAD OF  
# (OF P61TABLE FOR P61LM, FOR EXAMPLE) IN ERASABLE REGISTER 'WHICH' (E4). THE IGNITION  
# WHICH TO OBTAIN OR EXECUTE THE PROPER TABLE ENTRY. THE IGNITION ROUTINE IS INITIATED  
# THROUGH BANKJUMP IF NECESSARY. THERE IS NO RETURN.

#

# THE MASTER IGNITION ROUTINE WAS CONCEIVED AND EXECUTED, AND (NOTA BENE) IS MAINTAINED

#

# HONI SOIT QUI MAL Y PENSE

#

# \*\*\*\*\*

# TABLES FOR THE IGNITION ROUTINE

# \*\*\*\*\*

#

# NOLI SE TANGERE

P12TABLE	VN	0674	# (0)	
	TCF	ULLGNOT	# (1)	
	TCF	COMFAIL3	# (2)	
	TCF	GOCUTOFF	# (3)	
	TCF	TASKOVER	# (4)	
	TCF	P12SPOT	# (5)	
	DEC	0	# (6)	NO ULLAGE
	EBANK=	WHICH		
	2CADR	SERVEXIT	# (7)	
	TCF	DISPCHNG	# (11)	
	TCF	WAITABIT	# (12)	
	TCF	P12IGN	# (13)	

P40TABLE	VN	0640	# (0)
	TCF	ULLGNOT	# (1)
	TCF	COMFAIL4	# (2)
	TCF	GOPOST	# (3)
	TCF	TASKOVER	# (4)
	TCF	P40SPOT	# (5)

244      $\langle \text{Page LM0732 244} \rangle \equiv$  (241 722)

	DEC	2240	# (6)
	EBANK=	OMEGAQ	
	2CADR	STEERING	# (7)
	TCF	P40SJUNK	# (11)
	TCF	WAITABIT	# (12)
	TCF	P40IGN	# (13)
	TCF	REP40ALM	# (14)
P41TABLE	TCF	P41SPOT	# (5)
	DEC	-1	# (6)
	EBANK=	OMEGAQ	
	2CADR	CALCN85	# (7)
	TCF	COMMON	# (11)
	TCF	TIGTASK	# (12)
P42TABLE	VN	0640	# (0)
	TCF	WANTAPS	# (1)
	TCF	COMFAIL4	# (2)
	TCF	GOPOST	# (3)
	TCF	TASKOVER	# (4)
	TCF	P42SPOT	# (5)
	DEC	2640	# (6)
	EBANK=	OMEGAQ	
	2CADR	STEERING	# (7)
	TCF	P40SJUNK	# (11)
	TCF	WAITABIT	# (12)
	TCF	P42IGN	# (13)
	TCF	P42STAGE	# (14)
P63TABLE	VN	0662	# (0)
	TCF	ULLGNOT	# (1)
	TCF	COMFAIL3	# (2)
	TCF	V99RECYC	# (3)
	TCF	TASKOVER	# (4)
	TCF	P63SPOT	# (5)
	DEC	2240	# (6)
	EBANK=	WHICH	
	2CADR	SERVEXIT	# (7)
	TCF	DISPCHNG	# (11)
	TCF	WAITABIT	# (12)

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	TCF	P63IGN	# (13)
ABRTABLE	VN	0663	# (0)
	TCF	ULLGNOT	# (1)
	TCF	COMFAIL3	# (2)
	TCF	GOCUTOFF	# (3)
	TCF	TASKOVER	# (4)
	NOOP		# (5)
	NOOP		# (6)
	NOOP		# (7)
	NOOP		
	TCF	DISPCHNG	# (11)
	TCF	WAITABIT	# (12)
	TCF	ABRTIGN	# (13)

```
# *****
# GENERAL PURPOSE IGNITION ROUTINES
# *****
```

BURNBABY	TC	PHASCHNG	# GROUP 4 RESTARTS HERE
	OCT	04024	
	CAF	ZERO	# EXTIRPATE JUNK LEFT IN DVTOTAL
	TS	DVTOTAL	
	TS	DVTOTAL +1	
	TC	BANKCALL	# P40AUTO MUST BE BANKCALLED EVEN FROM ITS
	CADR	P40AUTO	# OWN BANK TO SET UP RETURN PROPERLY
B*RNB*B*	EXTEND		
	DCA	TIG	# STORE NOMINAL TIG FOR OBLATENESS COMP.
	DXCH	GOBLTIME	# AND FOR P70 OR P71.
	INHINT		
	TC	IBNKCALL	
	CADR	ENGNOF3	
	RELINT		
	INDEX	WHICH	
	TCF	5	
P42SPOT	=	P40SPOT	# (5)
P12SPOT	=	P40SPOT	# (5)
P63SPOT	=	P41SPOT	# (5) IN P63 CLOKTASK ALREADY GOING
P40SPOT	CS	CNTDNDEX	# (5)

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(241 722)

	TC	BANKCALL	# MUST BE BANKCALLED FOR GENERALIZED
	CADR	STCLOK2	# RETURN
P41SPOT	TC	INTPRET	# (5)
	DLOAD	DSU	
		TIG	
		D29.9SEC	
	STCALL	TDEC1	
		INITCDUW	
	BOFF	CALL	
		MUNFLAG	
		GOMIDAV	
		CSMPREC	
	VLOAD	MXV	
		VATT1	
		REFSMMAT	
	VSR1		
	STOVL	V(CSM)	# CSM VELOCITY -- M/CS*2(7)
		RATT1	
	VSL4	MXV	
		REFSMMAT	
	STCALL	R(CSM)	# CSM POSITION -- M*2(24)
		MUNGRAV	
	STODL	G(CSM)	# CSM GRAVITY VEC. -- M/CS*2(7)
		TAT	
	STORE	TDEC1	# RELOAD TDEC1 FOR MIDTOAV.
GOMIDAV	CALRB		
		MIDTOAV1	
	TCF	CALLT-35	# MADE IT IN TIME.
	EXTEND		# TIG WAS SLIPPED, SO RESET TIG TO 29.9
	DCA	PIPTIME1	# SECONDS AFTER THE TIME TO WHICH WE DID
	DXCH	TIG	# INTEGRATE.
	EXTEND		
	DCA	D29.9SEC	
	DAS	TIG	
CALLT-35	DXCH	MPAC	
	DXCH	SAVET-30	# DELTA-T UNTIL TIG-30
	EXTEND		
	DCS	5SECDP	
	DAS	SAVET-30	# DELTA-T UNTIL TIG-35
	EXTEND		
	DCA	SAVET-30	
	TC	LONGCALL	
	EBANK=	TTGO	

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2CADR TIG-35

TC PHASCHNG

OCT 20254

# 4.25SPOT FOR TIG-35 RESTART.

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(241 722)

```

TC      CHECKMM
DEC      63
TCF      ENDOFJOB      # NOT P63
CS      CNTDNDEX      # P63 CAN START DISPLAYING NOW.
TS      DISPDEX
TC      INTPRET
VLOAD    ABVAL
          VN1
STORE    ABVEL      # INITIALIZE ABVEL FOR P63 DISPLAY
EXIT
TCF      ENDOFJOB

#      *****

TIG-35    CAF      5SEC
          TC      TWIDDLE
          ADRES    TIG-30

          TC      PHASCHNG
          OCT      40154      # 4.15SPOT FOR TIG-30 RESTART

          CS      BLANKDEX      # BLANK DSKY FOR 5 SECONDS
          TS      DISPDEX

          INDEX    WHICH
          CS      6      # CHECK ULLAGE TIME.
          EXTEND
          BZMF     TASKOVER
          CAF      4.9SEC      # SET UP TASK TO RESTORE DISPLAY AT TIG-30
          TC      TWIDDLE
          ADRES    TIG-30.1

          CAF      PRI017      # A NEGATIVE ULLAGE TIME INDICATES P41, IN
          TC      NOVAC      # WHICH CASE WE HAVE TO SET UP A JOB TO
          EBANK=    TTOGO      # BLANK THE DSKY FOR FIVE SECONDS, SINCE
          2CADR     P41BLANK    # CLOKJOB IS NOT RUNNING DURING P41.

          TCF      TASKOVER

P41BLANK  TC      BANKCALL      # BLANK DSKY.
          CADR     CLEANDSP
          TCF      ENDOFJOB

TIG-30.1  CAF      PRI017      # SET UP JOB TO RESTORE DISPLAY AT TIG-30
          TC      NOVAC

```



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EBANK= TTOGO  
2CADR TIG-30A  
  
TCF TASKOVER

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(241 722)

TIG-30A CAF V16N85B  
 TC BANKCALL # RESTORE DISPLAY.  
 CADR REGODSP # REGODSP DOES A TCF ENDOFJOB

# \*\*\*\*\*

TIG-30 CAF S24.9SEC  
 TC TWIDDLE  
 ADRES TIG-5

CS CNTDNDEX # START UP CLOKTASK AGAIN  
 TS DISPDEX

INDEX WHICH # PICK UP APPROPRIATE ULLAGE -- ON TIME  
 CA 6 # Was CAF --- RSB 2009.  
 EXTEND  
 BZMF ULLGNOT # DON'T SET UP ULLAGE IF DT IS NEG OR ZERO  
 TS SAVET-30 # SAVE DELTA-T FOR RESTART  
 TC TWIDDLE  
 ADRES ULLGTASK

CA THREE # RESTART PROTECT ULLGTASK (1.3SPOT)  
 TS L  
 CS THREE  
 DXCH -PHASE1  
 CS TIME1  
 TS TBASE1

INDEX WHICH  
 TCF 1

WANTAPS CS FLGWRD10 # (1) FOR P42 ENSURE APSFLAG IS SET. IF IT  
 MASK APSFLBIT # WASN'T SET, DAP WILL BE INITIALIZED TO  
 ADS FLGWRD10 # ASCENT VALUES BY 1/ACCS IN 2 SECONDS.

ULLGNOT EXTEND # (1)  
 INDEX WHICH  
 DCA 7 # LOAD AVEGEXIT WITH APPROPRIATE 2CADR  
 DXCH AVEGEXIT

CAF TWO # 4.2SPOT RESTARTS IMMEDIATELY AT REDO4.2  
 TS L  
 CS TWO # AND ALSO AT TIG-5 AT THE CORRECT TIME.  
 DXCH -PHASE4

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CS TIME1  
TS TBASE4

# SET TBASE4 FOR TIG-5 RESTART

RED02.17

EXTEND

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```

DCA      NEG0      # CLEAR OUT GROUP 2 SO LAMBERT CAN START
DXCH     -PHASE2    # IF NEEDED.

REDO4.2   CCS      PHASE5      # IF SERVICER GOING?
          TCF      TASKOVER    # YES, DON'T START IT UP AGAIN.

          TC       POSTJUMP
          CADR     PREREAD      # PREREAD END THIS TASK

# *****

ULLGTASK  TC       ONULLAGE     # THIS COMES AT TIG-7.5 OR TIG-3.5
          TC       PHASCHNG
          OCT      1
          TCF      TASKOVER

# *****

TIG-5     EXTEND
          DCA      NEG0      # INSURE THAT GROUP 3 IS INACTIVE.
          DXCH     -PHASE3

          CAF      5SEC
          TC       TWIDDLE
          ADRES    TIG-0

          TC       DOWNFLAG    # RESET IGNFLAG AND ASINFLAG
          ADRES    IGNFLAG      # FOR LIGHT-UP LOGIC.
          TC       DOWNFLAG
          ADRES    ASTNFLAG

          INDEX    WHICH
          TCF      11

P40SJUNK  CCS      PHASE3      # (11) P40 AND P42. S40.13 IN PROGRESS?
          TCF      DISPCHNG    # YES

          CAF      PRI020
          TC       FINDVAC
          EBANK=   TTOGO
          2CADR    S40.13

          TC       PHASCHNG    # 3.5SPOT FOR S40.13
          OCT      00053
          CS       VB99DEX      # (11)
DISPCHNG

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TS DISPDEX

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(241 722)

COMMON	TC	PHASCHNG	# RESTART TIG-0 (4.7SPOT)
	OCT	40074	
	TCF	TASKOVER	

# \*\*\*\*\*

TIG-0	CS	FLAGWRD7	# SET IGNFLAG SINCE TIG HAS ARRIVED
	MASK	IGNFLBIT	
	ADS	FLAGWRD7	

TC	CHECKMM	# IN P63 CASE, THROTTLE-UP IS ZOOMTIME
DEC	63	# AFTER NOMINAL IGNITION, NOT ACTUAL

TCF	IGNYET?
CA	ZOOMTIME
TC	WAITLIST
EBANK=	DVCNTR
2CADR	P63ZOOM

TC	2PHSCHNG
OCT	40033

OCT	05014
OCT	77777

IGNYET?	CAF	ASTNBIT	# CHECK ASTNFLAG: HAS ASTRONAUT RESPONDED
	MASK	FLAGWRD7	# TO OUR ENGINE ENABLE REQUEST?

EXTEND		
INDEX	WHICH	
BZF	12	# BRANCH IF HE HAS NOT RESPONDED YET

IGNITION	CS	FLAGWRD5	# INSURE ENGONFLG IS SET.
----------	----	----------	---------------------------

MASK	ENGONBIT
ADS	FLAGWRD5
CS	PRI030

# TURN ON THE ENGINE.

EXTEND	
RAND	DSALMOUT
AD	BIT13

EXTEND	
WRITE	DSALMOUT

# SET TEVENT FOR DOWNLINK

EXTEND	
DCA	TIME2
DXCH	TEVENT

EXTEND	
DCA	TGO

# UPDATE TIG USING TGO FROM S40.13

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DXCH	TIG
EXTEND	
DCA	TIME2
DAS	TIG

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(241 722)

	CS	FLUNDBIT	# PERMIT GUIDANCE LOOP DISPLAYS
	MASK	FLAGWRD8	
	TS	FLAGWRD8	
	INDEX	WHICH	
	TCF	13	
P63IGN	EXTEND		# (13) INITIATE BURN DISPLAYS
	DCA	DSP2CADR	
	DXCH	AVGEXIT	
	CA	Z	# ASSASSINATE CLOKTASK
	TS	DISPDEX	
	CS	FLAGWRD9	# SET FLAG FOR P70-P71
	MASK	LETABBIT	
	ADS	FLAGWRD9	
	CS	FLAGWRD7	# SET SWANDISP TO ENABLE R10.
	MASK	SWANDBIT	
	ADS	FLAGWRD7	
	CS	PULSES	# MAKE SURE DAP IS NOT IN MINIMUM-IMPULSE
	MASK	DAPBOOLS	# MODE, IN CASE OF SWITCH TO P66
	TS	DAPBOOLS	
	EXTEND		# INITIALIZE TIG FOR P70 AND P71.
	DCA	TIME2	
	DXCH	TIG	
	CAF	ZERO	# INITIALIZE WCHPHASE, AND FLPASSO
	TS	WCHPHASE	
	TS	WCHPHOLD	# ALSO WHCPHOLD
	CA	TWO	
	TS	FLPASSO	
P40IGN	TCF	P42IGN	
	CS	FLAGWRD5	# (13)
	MASK	NOTHRBIT	
	EXTEND		
	BZF	P42IGN	
	CA	ZOOMTIME	
	TC	WAITLIST	
	EBANK=	DVCNTR	
	2CADR	P40ZOOM	



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P63IGN1

TC 2PHSCHNG  
OCT 40033  
OCT 05014  
OCT 77777

# 3.3SPOT FOR ZOOM RESTART.  
# TYPE C RESTARTS HERE IMMEDIATELY

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(241 722)

P12IGN	TCF	P42IGN	
	CAF	EBANK6	
	TS	EBANK	
	EBANK=	AOSQ	
	CA	IGNAOSQ	# INITIALIZE DAP BIAS ACCELERATION
	TS	AOSQ	# ESTIMATES AT P12 IGNITION.
	CA	IGNAOSR	
	TS	AOSR	
	CAF	EBANK7	
	TS	EBANK	
	EBANK=	DVCNTR	
ABRTIGN	CA	Z	# (13) KILL CLOKTASK
	TS	DISPDEX	
	EXTEND		# CONNECT ASCENT GUIDANCE TO SERVICER.
	DCA	ATMAGADR	
	DXCH	AVGEXIT	
	CS	FLAGWRD7	# ENABLE R10.
	MASK	SWANDBIT	
	ADS	FLAGWRD7	
P42IGN	CS	DRIFTBIT	# ENSURE THAT POWERED-FLIGHT SWITCHING
	MASK	DAPBOOLS	# CURVES ARE USED.
	TS	DAPBOOLS	
	CAF	IMPULBIT	# EXAMINE IMPULSE SWITCH
	MASK	FLAGWRD2	
	CCS	A	
	TCF	IMPLBURN	
DVMONCON	TC	DOWNFLAG	
	ADRES	IGNFLAG	# CONNECT DVMON
	TC	DOWNFLAG	
	ADRES	ASTNFLAG	
	TC	DOWNFLAG	
	ADRES	IDLEFLAG	
	TC	PHASCHNG	
	OCT	40054	
	TC	FIXDELAY	# TURN ULLAGE OFF HALF A SECOND AFTER
	DEC	50	# LIGHT UP.

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ULLAGOFF TC NOULLAGE

WAITABIT EXTEND # KILL GROUP 4  
DCA NEG0

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	DXCH	-PHASE4	
	TCF	TASKOVER	
TIGTASK	TC	POSTJUMP	# (12)
	CADR	TIGTASK1	

# \*\*\*\*\*

	BANK	31	
	SETLOC	P40S3	
	BANK		
	COUNT*	\$\$/P40	
TIGTASK1	CAF	PRI016	
	TC	NOVAC	
	EBANK=	TRKMKCNT	
	2CADR	TIGNOW	
	TC	PHASCHNG	
	OCT	6	# KILL GROUP 6
	TCF	TASKOVER	

# \*\*\*\*\*

P63ZOOM	EXTEND		
	DCA	LUNLANAD	
	DXCH	AVEGEXIT	
	TC	IBNKCALL	
	CADR	FLATOUT	
	TCF	P40ZOOMA	
P40ZOOM	CAF	BIT13	
	TS	THRUST	
	CAF	BIT4	
	EXTEND		
	WOR	CHAN14	
P40ZOOMA	TC	PHASCHNG	
	OCT	3	
	TCF	TASKOVER	

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LUNLANAD	EBANK=	DVCNTR
	2CADR	LUNLAND

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(241 722)

```

ZOOM      =      P40ZOOMA
           BANK    36
           SETLOC  P40S
           BANK
           COUNT*  $$/P40

```

# \*\*\*\*\*

```

COMFAIL    TC      UPFLAG      # (15)
           ADRES   IDLEFLAG
           TC      UPFLAG      # SET FLAG TO SUPPRESS CONFLICTING DISPLAY
           ADRES   FLUNDISP
           CAF     FOUR        # RESET DVMON
           TS      DVCNTR
           CCS     PHASE6      # CLOKTASK ACTIVE?
           TCF     +3          # YES
           TC      BANKCALL    # OTHERWISE, START IT UP
           CADR    STCLOK1
           +3      CS      VB97DEX
           TS      DISPDEX
           TC      PHASCHNG    # TURN OFF GROUP 4.
           OCT     00004
           TCF     ENDOFJOB

COMFAIL1    INDEX   WHICH
           TCF      2

COMFAIL3    CA      Z          # (15) KILL CLOKTASK USING Z
           TCF      +2

COMFAIL4    CS      CNTDNDEX
           TS      DISPDEX

           TC      DOWNFLAG    # RECONNECT DV MONITOR
           ADRES   IDLEFLAG
           TC      DOWNFLAG    # PERMIT GUIDANCE LOOP DISPLAYS
           ADRES   FLUNDISP
           TCF     ENDOFJOB

COMFAIL2    TC      PHASCHNG    # KILL ZOOM RESTART PROTECTION
           OCT     00003

           INHINT
           TC      KILLTASK    # KILL ZOOM IN CASE IT'S STILL TO COME
           CADR    ZOOM

```

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TC	IBNKCALL	# COMMAND ENGINE OFF
CADR	ENGINEOF4	
TC	UPFLAG	# SET THE DRIFT BIT FOR THE DAP.
ADRES	DRIFTDFL	

264 (Page LM0743 264)≡

(241 722)

```

TC      INVFLAG      # USE OTHER RCS SYSTEM
ADRES   AORBTFLG
TC      UPFLAG       # TURN ON ULLAGE
ADRES   ULLAGFLG
CAF     BIT1
INHINT
TC      TWIDDLE
ADRES   TIG-5
TCF     ENDOFJOB

# *****
# SUBROUTINES OF THE IGNITION ROUTINE
# *****

INVFLAG  CA      Q
          TC      DEBIT
          COM
          EXTEND
          RXOR    LCHAN
          TCF     COMFLAG

# *****

NOULLAGE  CS      ULLAGER      # MUST BE CALLED IN A TASK OR UNDER INHINT
          MASK    DAPBOOLS
          TS      DAPBOOLS
          TC      Q

# *****

ONULLAGE  CS      DAPBOOLS      # TURN ON ULLAGE.  MUST BE CALLED IN
          MASK    ULLAGER      # A TASK OR WHILE INHINTED.
          ADS     DAPBOOLS
          TC      Q

# *****

STCLOK1  CA      ZERO          # THIS ROUTINE STARTS THE COUNT-DOWN
STCLOK2  TS      DISPDEX       # (CLOKTASK AND CLOKJOB).  SETTING
STCLOK3  TC      MAKECADR      # SETTING DISPDEX POSITIVE KILLS IT.
          TS      TBASE4       # RETURN SAVE (NOT FOR RESTARTS).
          EXTEND
          DCA     TIG
          DXCH    MPAC
          EXTEND

```



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DCS TIME2

266 (Page LM0744 266)≡

(241 722)

	DAS	MPAC	# HAVE TIG -- TIME2, UNDOUBTEDLY A + NUMBER
	TC	TPAGREE	# POSITIVE, SINCE WE PASSED THE
	CAF	1SEC	# 45 SECOND CHECK.
	TS	Q	
	DXCH	MPAC	
	MASK	LOW5	# RESTRICT MAGNITUDE OF NUMBER IN A
	EXTEND		
	DV	Q	
	CA	L	# GET REMAINDER
	AD	TWO	
	INHINT		
	TC	TWIDDLE	
	ADRES	CLOKTASK	
	TC	2PHSCHNG	
	OCT	40036	# 6.3SPOT FOR CLOKTASK
	OCT	05024	
	OCT	13000	
	CA	TBASE4	
	TC	BANKJUMP	
CLOKTASK	CS	TIME1	# SET TBASE6 FOR GROUP 6 RESTART
	TS	TBASE6	
	CCS	DISPDEX	
	TCF	KILLCLOK	
	NOOP		
	CAF	PRI027	
	TC	NOVAC	
	EBANK=	TTOGO	
	2CADR	CLOKJOB	
	TC	FIXDELAY	# WAIT A SECOND BEFORE STARTING OVER
	DEC	100	
	TCF	CLOKTASK	
KILLCLOK	EXTEND		# KILL RESTART
	DCA	NEGO	
	DXCH	-PHASE6	
	TCF	TASKOVER	
CLOKJOB	EXTEND		
	DCS	TIG	
	DXCH	TTOGO	
	EXTEND		

267 (Page LM0745 267)≡

(241 722)

```

DCA      TIME2
DAS      TTOGO
INHINT
CCS      DISPDEX      # IF DISPDEX HAS BEEN SET POSITIVE BY A
TCF      ENDOFJOB      # TASK OR A HIGHER PRIORITY JOB SINCE THE
TCF      ENDOFJOB      # LAST CLOKTASK, AVOID USING IT AS AN
COM
RELINT
INDEX    A            # ***** DISPDEX MUST NEVER B -0 *****
TCF      DISPNOT -1    # (-1 DUE TO EFFECT OF CCS)

VB97DEX  =            OCT35      # NEGATIVE OF THIS IS PROPER FOR DISPDEX

-35      CS          ZERO        # INDICATE VERB 97 PASTE
          TS          NVWORD1
          CA          NVWORD +2    # NVWORD+2 CONTAINS VO6 & APPROPRIATE NOUN
          TC          BANKCALL
          CADR        CLOCPLAY
          TCF         STOPCLOK      # TERMINATE CLOKTASK ON THE WAY TO POOH
          TCF         COMFAIL1
          TCF         COMFAIL2

-25      CAF          VO6N61      # THIS DISPLAY IS CALLED VIA ASTNCLOK
          TC          BANKCALL      # IT IS PRIMARILY USED BY THE CREW IN P63
          CADR        REFLASH      # TO RESET HIS EVENT TIMER TO AGREE WITH
          TCF         STOPCLOK      # TIG.
          TCF         ASTNRETN
          TCF         -6

CNTDNDEX =            LOW4        # OCT17:  NEGATIVE PROPER FOR DISPDEX

-17      INDEX      WHICH          # THIS DISPLAY COMES UP AT ONE SECOND
          # Was CAF --- RSB 2009
          CA          0            # INTERVALS.  IT IS NORMALLY OPERATED
          TC          BANKCALL      # BETWEEN TIG-30 SECONDS AND TIG-5 SECONDS
          CADR        REGODSP      # REGODSP DOES ITS OWN TCF ENDOFJOB

VB99DEX  =            ELEVEN      # OCT13:  NEGATIVE PROPER FOR DISPDEX

V99RECYC EQUALS

-13      CS          BIT9          # INDICATE VERB 99 PASTE
          TS          NVWORD1
          INDEX      WHICH          # THIS IS THE "PLEASE ENABLE ENGINE"

```

```
# Was CAF --- RSB 2004
```

```
CA      0      # DISPLAY; IT IS INITIATED AT TIG-5 SEC.  
TC      BANKCALL  # THE DISPLAY IS A V99NXX, WHERE XX IS  
CADR    CLOCPLAY  # NOUN THAT HAD PREVIOUSLY BEEN DISPLAYED  
TCF     STOPCLOK  # TERMINATE GOTOP00H TURNS OFF ULLAGE.  
TCF     *PROCEED  
TCF     *ENTER
```

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```
269  (Page LM0746 269)≡
      BLANKDEX      =      TWO      # NEGATIVE OF THIS IS PROPER FOR DISPDEX
                                     (241 722)
      -2      TC      BANKCALL      # BLANK DSKY. THE DSKY IS BLANKED FOR
      CADR      CLEANDSP      # 5 SECONDS AT TIG-35 TO INDICATE THAT
      DISPNOT      TCF      ENDOFJOB      # AVERAGE G IS STARTING.
      STOPCLOK      TC      NULLCLOK      # STOP CLOKTASK & TURN OFF ULLAGE ON THE
      TCF      GOTOPOOH      # WAY TO P00 (GOTOPOOH RELINTS)
      NULLCLOK      INHINT
      EXTEND
      QXCH      P40/RET
      TC      NOULLAGE      # TURN OFF ULLAGE ...
      TC      KILLTASK      # DON'T LET IT COME ON, EITHER ...
      CADR      ULLGTASK
      TC      PHASCHNG      #
      OCT      1      # NOT EVEN IF THERE'S A RESTART.
      CA      Z      # KILL CLOKTASK
      TS      DISPDEX
      TC      P40/RET
      ASTNRET      TC      PHASCHNG
      OCT      04024
      CAF      ZERO      # STOP DISPLAYING BUT KEEP RUNNING
      TS      DISPDEX
      CAF      PRI013
      TC      FINDVAC
      EBANK=      STARIND
      2CADR      ASTNRET
      TCF      ENDOFJOB
      *PROCEED      TC      UPFLAG
      ADRES      ASTNFLAG
      TCF      IGNITE
      *ENTER      INHINT
      INDEX      WHICH
      TCF      3
      GOPOST      CAF      PRI012      # (3) MUST BE LOWER PRIORITY THAN CLOKJOB
      TC      FINDVAC
      EBANK=      TTOGO
      2CADR      POSTBURN
```



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```
271  (Page LM0747 271)≡
                                     (241 722)
                                     # SET UP THE DAP FOR COASTING FLIGHT.
                                     INHINT
                                     TC      IBNKCALL
                                     CADR    ALLCOAST
                                     TC      NULLCLOK
                                     TC      PHASCHNG
                                     OCT      00134
                                     # 4.13 RESTART FOR POSTBURN
                                     TCF      ENDOFJOB
GOCUTOFF
CAF      PRI017      # (3)
TC      FINDVAC
EBANK=   TGO
2CADR    CUTOFF
TC      DOWNFLAG
ADRES    FLUNDISP
                                     INHINT
                                     TC      IBNKCALL
                                     CADR    ALLCOAST
                                     TC      NULLCLOK
                                     TC      PHASCHNG
                                     OCT      07024
                                     OCT      17000
                                     EBANK=   TGO
                                     2CADR    CUTOFF
                                     TCF      ENDOFJOB
IGNITE
CS      FLAGWRD7      # (2)
MASK    IGNFLBIT
CCS      A
TCF      IGNITE1
CAF      BIT1
INHINT
TC      TWIDDLE
ADRES    IGNITION
CAF      OCT23      # IMMEDIATE RESTART AT IGNITION
TS      L
COM
DXCH     -PHASE4
IGNITE1
CS      CNTDNDEX      # RESTORE OLD DISPLAY.
TS      DISPDEX
```

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TCF      ENDOFJOB



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273 (Page LM0748 273)≡

(241 722)

```
# *****

P40ALM      TC      ALARM      # PROGRAM SELECTION NOT CONSISTENT WITH
            OCT      1706      # VEHICLE CONFIGURATION

REP40ALM    CAF      V05N09      # (14)
            TC      BANKCALL
            CADR     GOFLASH

            TCF      GOTOP00H      # V34E      TERMINATE
            TCF      +2            # PROCEED      CHECK FOR P42
            TCF      REP40ALM      # V32E      REDISPLAY ALARM

            INDEX   WHICH          # FOR P42, ALLOW CREW TO PROCEED EVEN
            TCF     14            # THOUGH VEHICLE IS UNSTAGED.

# *****

            BANK     31
            SETLOC   P40S2
            BANK

            COUNT*   $$/P40

P40AUTO      TC      MAKECADR      # HELLO THERE.
            TS      TEMPR60      # FOR GENERALIZED RETURN TO OTHER BANKS.
P40A/P      TC      BANKCALL      # SUBROUTINE TO CHECK PGNC'S CONTROL
            CADR     G+N,AUTO      # AND AUTO STABILIZATION MODES
            CCS      A            # +0 INDICATES IN PGNC'S, IN AUTO
            TCF      TURNITON      # + INDICATES NOT IN PGNC'S AND/OR AUTO
            CAF      APSFLBIT      # ARE WE ON THE DESCENT STAGE?
            MASK     FLGWRD10
            CCS      A
            TCF      GOBACK          # RETURN
            CAF      BIT5          # YES, CHECK FOR AUTO-THROTTLE MODE
            EXTEND
            RAND     CHAN30
            EXTEND
            BZF      GOBACK          # IN AUTO-THROTTLE MODE -- RETURN
TURNITON     CAF      P40A/PMD      # DISPLAYS V50N25 R1=203 PLEASE PERFORM
            TC      BANKCALL      # CHECKLIST 203 TURN ON PGNC'S ETC.
            CADR     GOPERF1
            TCF      GOTOP00H      # V34E TERMINATE
            TCF      P40A/P        # RECYCLE
GOBACK       CA      TEMPR60
```

```

                TC      BANKJUMP      # GOODBYE.  COME AGAIN SOON.

P40A/PMD      OCT      00203

```

274  $\langle$ Page LM0749 274 $\rangle \equiv$ 

(241 722)

```

                BANK      36
                SETLOC    P40S
                BANK

                COUNT*    $$/P40

#      *****
#      CONSTANTS FOR THE IGNITION ROUTINE
#      *****

SERVCADR      =      P63TABLE +7

P40ADRES      ADRES      P40TABLE

P41ADRES      ADRES      P41TABLE -5

P42ADRES      ADRES      P42TABLE

                EBANK=    DVCNTR
DSP2CADR      2CADR      P63DISPS -2

                EBANK=    DVCNTR
ATMAGADR      2CADR      ATMAG

?      =      GOTOP00H

D29.9SEC      2DEC      2990

S24.9SEC      DEC      2490

4.9SEC      DEC      490

OCT20      =      BIT5

V06N61      VN      0661

```

275 (Page LM0750 275)≡

(241 722)

```

# KILLTASK
# MOD NO:  NEW PROGRAM
# MOD BY:  COVELLI
#
# FUNCTIONAL DESCRIPTION:
#
#       KILLTASK IS USED TO REMOVE A TASK FROM THE WAITLIST BY SUBSTITUTING A NULL TASK CALLED
#       WHICH MERELY DOES A TC TASKOVER.  IF THE SAME TASK IS SCHEDULED MORE THAN ONCE, ONLY THE
#       FIRST IS REMOVED.  IF THE TASK IS NOT SCHEDULED, KILLTASK TAKES NO ACTION AND RETURNS W
#       LEAVES INTERRUPTS INHIBITED SO CALLER MUST RELINT
#
# CALLING SEQUENCE
#       L      TC      KILLTASK      # IN FIXED-FIXED
#       L+1    CADR    ?????????    # CADR (NOT 2CADR) OF TASK TO BE REMOVED.
#       L+2    (RELINT)      # RETURN
#
# EXIT MODE:  AT L+2 OF CALLING SEQUENCE.
#
# ERASABLE INITIALIZATION:  NONE.
#
# OUTPUT:  2CADR OF NULLTASK IN LST2
#
# DEBRIS:  ITEMP1 - ITEMP4, A, L, Q.

```

```

EBANK=  LST2
BLOCK   3      # KILLTASK MUST BE IN FIXED-FIXED.
SETLOC  FFTAG6
BANK
COUNT* $$/KILL
KILLTASK CA    KILLBB
INHINT
LXCH    A
INDEX   Q
CA      0      # GET CADR.
LXCH    BBANK
TCF     KILLTSK2  # CONTINUE IN SWITCHED FIXED.

```

```

EBANK=  LST2
KILLBB  BBCON  KILLTSK2

BANK    27

SETLOC  P40S1
BANK
COUNT* $$/KILL

```

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KILLTSK2

LXCH

ITEMP2

# SAVE CALLER'S BBANK

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277	⟨Page LM0751 277⟩≡			(241 722)
		INCR	Q	
		EXTEND		
		QXCH	ITEMP1	# RETURN 2ADR IN ITEMP1,ITEMP2
		TS	ITEMP3	# CADR IS IN A
		MASK	LOW10	
		AD	BIT11	
		TS	ITEMP4	# GENADR OF TASK
		CS	LOW10	
		MASK	ITEMP3	
		TS	ITEMP3	# FBANK OF TASK
		ZL		
ADRSCAN		INDEX	L	
		CS	LST2	
		AD	ITEMP4	# COMPARE GENADRS
		EXTEND		
		BZF	TSTFBANK	# IF THEY MATCH, COMPARE FBANKS
LETITLIV		CS	LSTLIM	
		AD	L	
		EXTEND		# ARE WE DONE?
		BZF	DEAD	# YES -- DONE, SO RETURN
		INCR	L	
		INCR	L	
		TCF	ADRSCAN	# CONTINUE LOOP.
DEAD		DXCH	ITEMP1	
		DTCB		
TSTFBANK		CS	LOW10	
		INDEX	L	
		MASK	LST2 +1	# COMPARE FBANKS ONLY.
		EXTEND		
		SU	ITEMP3	
		EXTEND		
		BZF	KILLDEAD	# MATCH -- KILL IT.
		TCF	LETITLIV	# NO MATCH -- CONTINUE.
KILLDEAD		CA	TCTSKOVR	
		INDEX	L	
		TS	LST2	# REMOVE TASK BY INSERTING TASKOVER
		TCF	DEAD	
LSTLIM		EQUALS	BIT5	# DEC 16

## 1.18 the lunar landing

$$\begin{aligned} 278 \quad \langle \textit{the lunar landing} \ 278 \rangle \equiv & \hspace{10em} (7) \\ & \langle \textit{Page LM0785} \ 279 \rangle \\ & \langle \textit{Page LM0786} \ 281 \rangle \\ & \langle \textit{Page LM0787} \ 283 \rangle \\ & \langle \textit{Page LM0788} \ 285 \rangle \\ & \langle \textit{Page LM0789} \ 287\textit{a} \rangle \\ & \langle \textit{Page LM0790} \ 287\textit{b} \rangle \\ & \langle \textit{Page LM0791} \ 288 \rangle \\ & \langle \textit{Page LM0792} \ 289\textit{a} \rangle \end{aligned}$$

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279 (Page LM0785 279)≡ (278 770)

BANK 32  
SETLOC F2DPS\*32  
BANK

EBANK= E2DPS

# \*\*\*\*\*  
# P63: THE LUNAR LANDING, BRAKING PHASE  
# \*\*\*\*\*

COUNT\* \$\$/P63

P63LM

TC PHASCHNG  
OCT 04024

TC BANKCALL # DO IMU STATUS CHECK ROUTINE R02  
CADR R02BOTH

CAF P63ADRES # INITIALIZE WHICH FOR BURNBABY  
TS WHICH

CAF DPSTHRSH # INITIALIZE DVMON  
TS DVTHRUSH  
CAF FOUR  
TS DVCNTR

CS ONE # INITIALIZE WCHPHASE AND FLPASSO  
TS WCHPHASE

CA ZERO  
TS FLPASSO

CS BIT14  
EXTEND  
WAND CHAN12 # REMOVE TRACK-ENABLE DISCRETE.

FLAGORGY

TC INTPRET # DIONYSIAN FLAG WAVING  
CLEAR CLEAR  
NOTHROTL  
REDFLAG

CLEAR SET  
LRBYPASS  
MUNFLAG

CLEAR CLEAR # TERMINATE P25 IF IT IS RUNNING.  
P25FLAG

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RNDVZFLG

# TERMINATE P20 IF IT IS RUNNING.

# \*\*\*\*\*

IGNALG

SETPD

VLOAD

# FIRST SET UP INPUTS FOR RP-TO-R:



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281 (Page LM0786 281)≡

(278 770)

```

                                0
                                RLS
                                # AT OD LANDING SITE IN MOON FIXED FRAME
                                # AT 6D ESTIMATED TIME OF LANDING
PDDL PUSH                        # MPAC NON-ZERO TO INDICATE LUNAR CASE
                                TLAND
STCALL TPIP                      # ALSO SET TPIP FOR FIRST GUIDANCE PASS
                                RP-TO-R
VSL4 MXV
                                REFSMMAT
STCALL LAND
                                GUIDINIT # GUIDINIT INITIALIZES WM AND /LAND/
DLOAD DSU
                                TLAND
                                GUIDDURN
STCALL TDEC1                    # INTEGRATE STATE FORWARD TO THAT TIME
                                LEMPREC
SSP VLOAD
                                NIGNLOOP
                                4OD
                                UNITX
STOVL CG
                                UNITY
STOVL CG +6
                                UNITZ
STODL CG +14
                                99999CON
STOVL DELTAH                    # INITIALIZE DELTAH FOR V16N68 DISPLAY
                                ZEROVECS
STODL UNFC/2                    # INITIALIZE TRIM VELOCITY CORRECTION TERM
                                HI6ZEROS
STORE TTF/8

IGNALOOP DLOAD
                                TAT
STOVL PIPTIME1
                                RATT1
VSL4 MXV
                                REFSMMAT
STCALL R
                                MUNGRAV
STCALL GDT/2
                                ?GUIDSUB # WHICH DELIVERS N PASSES OF GUIDANCE

# DDUMCALC IS PROGRAMMED AS FOLLOWS:
#
# (RIGNZ - RGU )/16 + 16(RGU )KIGNY/B8 + (RGU - RIGNX)KIGNX/B4 + (ABVAL(VGU) - VI
```

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```
#           2           1           0
# DDUM = -----
#           10
#           2 (VGU - 16 VGU KIGNX/B4)
#           2           0
```

283 (Page LM0787 283)≡

(278 770)

# disconnected from their respective variables

# THE NUMERATOR IS SCALED IN METERS AT 2(28). THE DENOMINATOR IS A VELOCITY IN UNITS OF 2(10)

# THE QUOTIENT IS THUS A TIME IN UNITS OF 2(18) CENTISECONDS. THE FINAL SHIFT RESCALES TO UNITS

# THERE IS NO DAMPING FACTOR. THE CONSTANTS KIGNX/B4, KIGNY/B8 AND KIGNV/B4 ARE ALL NEGATIVE

```

DDUMCALC      TS      NIGNLOOP
               TC      INTPRET
               DLOAD    DMPR          # FORM DENOMINATOR FIRST
               VGU
               KIGNX/B4
               SL4R     BDSU
               VGU +4
               PDDL     DSU
               RIGNZ
               RGU +4
               SR4R     PDDL
               RGU +2
               DSQ       DMPR
               KIGNY/B8
               SL4R     PDDL
               RGU
               DSU       DMPR
               RIGNX
               KIGNX/B4
               PDVL     ABVAL
               VGU
               DSU       DMPR
               VIGN
               KIGNV/B4
               DAD       DAD
               DAD       DDV
               SRR
               10D
               PUSH     DAD
               PIPTIME1
               STODL    TDEC1          # STORE NEW GUESS FOR NEXT INTEGRATION
               ABS      DSU
               DDUMCRIT
               BMN      CALL
               DDUMGOOD
               INTSTALL
               SET       SET
               INTYPFLG
               MOONFLAG

```

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DLOAD

PIPTIME1

STOVL

TET

# HOPEFULLY ?GUIDSUB DID NOT

RATT1

# CLOBBER RATT1 AND VATT1

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285 (Page LM0788 285)≡

(278 770)

```

      STOVL  RCV
      VATT1
      STCALL VCV
      INTEGRVS
      GOTO    IGNALoop

DDUMGOOD  SLOAD  SR
           ZOOMTIME
           14D
      BDSU
           TDEC1
      STOVL  TIG
           V
      VXV    UNIT
           R
      DOT    SL1
           LAND
R60INIT    STOVL  OUTFPLN
           UNFC/2
      STORE  R60VSAVE
      EXIT
           # COMPUTE DISTANCE LANDING SITE WILL BE
           # OUT OF LM'S ORBITAL PLANE AT IGNITION:
           # SIGN IS + IF LANDING SITE IS TO THE
           # RIGHT, NORTH; - IF TO THE LEFT, SOUTH.

           # *****

           # INITIALIZATION FOR CALCMANU
           # STORE UNFC/2 TEMPORARILY IN R60SAVE

           # *****

IGNALGRT  TC    PHASCHNG
           OCT   04024
           # PREVENT REPEATING IGNALG

ASTNCLOK  CS    ASTNDEX
           TC    BANKCALL
           CADR  STCLOK2
           TCF   ENDOFJOB
           # RETURN IN NEW JOB AND IN EBANK FIVE

ASTNRET   TC    INTPRET
           SSP   RTB
           QMAJ
           FCADR P63SPOT2
           R51P63
           # GO PICK UP DISPLAY AT END OF R51:
           # "PROCEED" WILL DO A FINE ALIGNMENT
           # "ENTER" WILL RETURN TO P63SPOT2

P63SPOT2  VLOAD  UNIT
           R60VSAVE
           STOVL POINTVSM
           UNITX
           STORE SCAXIS
           EXIT
           CAF   EBANK7
```

TS EBANK

INHINT

TC IBNKCALL

CADR PFLITEDB

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287a (Page LM0789 287a)≡

(278 770)

```
RELINT

TC      BANKCALL
CADR    R60LEM

TC      PHASCHNG      # PREVENT RECALLING R60
OCT     04024

P63SPOT3  CA      BIT6      # IS THE LR ANTENNA IN POSITION 1 YET
EXTEND
RAND     CHAN33
EXTEND
BZF      P63SPOT4      # BRANCH IF ANTENNA ALREADY IN POSITION 1

CAF      CODE500      # ASTRONAUT:  PLEASE CRANK THE
TC        BANKCALL      #
CADR      GOPERF1
TCF      GOTOP00H      # TERMINATE
TCF      P63SPOT3      # PROCEED      SEE IF HE'S LYING

P63SPOT4  TC      BANKCALL      # ENTER      INITIALIZE LANDING RADAR
CADR      SETPOS1

TC        POSTJUMP      # OFF TO SEE THE WIZARD ...
CADR      BURNBABY

# -----

# CONSTANTS FOR P63LM AND IGNALG

P63ADRES      GENADR  P63TABLE

ASTNDEX      =      MD1      # OCT 25:  INDEX FOR CLOKTASK

CODE500      OCT      00500

99999CON      2DEC      30479.7 B-24

GUIDDURN      2DEC      +66440      # GUIDDURN +6.64400314 E+2
DDUMCRIT      2DEC      +8 B-28      # CRITERION FOR IGNALG CONVERGENCE
```

287b (Page LM0790 287b)≡

(278 770)

# -----

288 (Page LM0791 288)≡

(278 770)

```
# *****
# P68: LANDING CONFIRMATION
# *****
```

```
BANK 31
SETLOC F2DPS*31
BANK
```

```
COUNT* $$/P6567
```

```
LANDJUNK TC PHASCHNG
OCT 04024
```

```
INHINT
```

```
TC BANKCALL
CADR ZATTEROR
```

# ZERO ATTITUDE ERROR

```
TC BANKCALL
CADR SETMAXDB
```

# SET 5 DEGREE DEADBAND

```
TC INTERPRET
SET CLEAR
```

# TO INTERPRETIVE AS TIME IS NOT CRITICAL

```
SURFFLAG
LETABORT
```

```
SET VLOAD
APSFLAG
RN
```

```
STODL ALPHAV
PIPTIME
```

```
SET CALL
LUNAFLAG
LAT-LONG
```

```
SETPD VLOAD
O
```

# COMPUTE RLS AND STORE IT AWAY

```
RN
VSL2 PDDL
```

```
PIPTIME
PUSH CALL
```

```
R-TO-RP
STORE RLS
```

```
EXIT
```

```
CAF V06N43*
TC BANKCALL
```

# ASTRONAUT: NOW LOOK WHERE YOU ENDED UP

```
CADR GOFLASH
TCF GOTOP00H
```

# TERMINATE



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289a	$\langle \text{Page LM0792 289a} \rangle \equiv$		(278 770)
	VLOAD		# INITIALIZE GSAV AND (USING REFMF)
		UNITX	# YNBSAV, ZNBSAV AND ATTFLAG FOR P57
	STCALL	GSAV	
		REFMF	
	EXIT		
	TCF	GOTOP00H	# ASTRONAUT: PLEASE SELECT P57
	V06N43*	VN	0643

## 1.19 throttle control routines

$$\begin{aligned} 289b \quad & \langle \textit{throttle control routines } 289b \rangle \equiv \\ & \langle \textit{Page LM0793 } 290 \rangle \\ & \langle \textit{Page LM0794 } 291 \rangle \\ & \langle \textit{Page LM0795 } 293 \rangle \\ & \langle \textit{Page LM0796 } 295 \rangle \\ & \langle \textit{Page LM0797 } 296a \rangle \end{aligned} \tag{7}$$

290 (Page LM0793 290)≡

(289b 771)

BANK 31  
 SETLOC FTHROT  
 BANK  
 EBANK= PIF  
 COUNT\* \$\$/THROT

# \* \* \* \* \*  
 # HERE FC, DESIRED THRUST, AND FP, PRESENT THRUST, UNWEIGHTED, ARE COMPUTED.

THROTTLE CA ABDELV # COMPUTE PRESENT ACCELERATION IN UNITS OF  
 EXTEND # 2(-4) M/CS/CS, SAVING SERVICER TROUBLE  
 MP /AF/CNST  
 +3 EXTEND  
 QXCH RTNHOLD  
 AFDUMP TC MASSMULT  
 DXCH FP # FP = PRESENT THRUST  
 EXTEND  
 DCA /AFC/  
 TC MASSMULT  
 TS FC # FC = THRUST DESIRED BY GUIDANCE  
 DXCH FCODD # FCODD = WHAT IT IS GOING TO GET

# IF IT HAS BEEN LESS THAN 3 SECONDS SINCE THE LAST THROTTLING, AUGMENT FP USING THE

CS TTHROT # THIS CODING ASSUMES A FLATOUT WITHIN  
 AD TIME1 # 80 SECONDS BEFORE FIRST THROTTLE CALL  
 MASK POSMAX  
 COM  
 AD 3SECS  
 EXTEND  
 BZMF WHERETO # BRANCH IF (TIME1-TTHROT +1) > 3 SECONDS  
 EXTEND  
 DCA FWEIGHT  
 DAS FP

# THIS LOGIC DETERMINES THE THROTTLING IN THE REGION 10% - 94%. THE MANUAL THROTTLE  
 # MINIMUM BY ASTRONAUT OR MISSION CONTROL PROGRAMS, PROVIDES THE LOWER BOUND. A STOP  
 # PROVIDES THE UPPER.

WHERE TO CA EBANK5 # INITIALIZE L\*WCR\*T AND H\*GHCR\*T FROM  
 TS EBANK # PAD LOADED ERASABLES IN W-MATRIX

291 (Page LM0794 291)≡

(289b 771)

```

EBANK= LOWCRIT
EXTEND
DCA LOWCRIT
DXCH L*WCR*T
CA EBANK7
TS EBANK
EBANK= PIF
CS ZERO # INITIALIZE PIFPSET
TS PIFPSET
CS H*GHCR*T
AD FCOLD
EXTEND
BZMF LOWFCOLD # BRANCH IF FCOLD < OR = HIGHCRIT
CS L*WCR*T
AD FCODD
EXTEND
BZMF FCOMPSET # BRANCH IF FC < OR = LOWCRIT
CA FP # SEE NOTE 1
TCF FLATOUT1

FCOMPSET CS FMAXODD # SEE NOTE 2
AD FP
TCF FLATOUT2

LOWFCOLD CS H*GHCR*T
AD FCODD
EXTEND
BZMF DOPIF # BRANCH IF FC < OR = HIGHCRIT

CA FMAXPOS # NO: THROTTLE-UP
FLATOUT1 DXCH FCODD
CA FEXTRA
FLATOUT2 TS PIFPSET

# NOTE 1 FC IS SET EQUAL TO FP SO PIF WILL BE ZERO. THIS IS DESIRABLE
# AS THERE IS ACTUALLY NO THROTTLE CHANGE.
#
# NOTE2 HERE, SINCE WE ARE ABOUT TO RETURN TO THE THROTTLEABLE REGION
# (BELOW 55%) THE QUANTITY -(FMAXODD-FP) IS COMPUTED AND PUT
# INTO PIFPSET TO COMPENSATE FOR THE DIFFERENCE BETWEEN THE
# NUMBER OF BITS CORRESPONDING TO FULL THROTTLE (FMAXODD) AND THE
# NUMBER CORRESPONDING TO ACTUAL THRUST (FP). THUS THE TOTAL
# THROTTLE COMMAND PIF = FC - FP - (FMAXODD - FP) = FC - FMAXODD.

DOPIF TC FASTCHNG

```

```
EXTEND
DCA      FCODD
TS        FCOLD
DXCH      PIF
EXTEND
```

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(289b 771)

```
      DCS      FP
      DAS      PIF      # PIF = FC - FP, NEVER EQUALS +0

DOIT      CA      PIF
          AD      PIFPSET      # ADD IN PIFPSET, WITHOUT CHANGING PIF
          TS      PSEUDO55
          TS      THRUST
          CAF      BIT4
          EXTEND
          WOR      CHAN14
          CA      TIME1
          TS      TTHROT
```

```
# SINCE /AF/ IS NOT AN INSTANTANEOUS ACCELERATION, BUT RATHER AN "AVERAGE" OF THE ACCELERATION
# THE PRECEEDING PIPA INTERVAL, AND SINCE FP IS COMPUTED DIRECTLY FROM /AF/, FP IN ORDER TO COR
# ACTUAL THRUST LEVEL AT THE END OF THE INTERVAL MUST BE WEIGHTED BY
```

```
#
#      PIF(PPROCESS + TL)      PIF /PIF/
#      FWEIGHT = ----- + -----
#                  PGUID          2 PGUID FRATE
#
```

```
# WHERE PROCESS IS THE TIME BETWEEN PIPA READING AND THE START OF THROTTLING, PGUID IS THE GUID
# FRATE IS THE THROTTLING RATE (32 UNITS PER CENTISECOND). PGUID IS EITHER 1 OR 2 SECONDS. TH
# FIRST TERM REPRESENTS THE ENGINE'S RESPONSE LAG. HERE FWEIGHT IS COMPUTED FOR USE NEXT PASS.
```

```
      CA      THISTPIP +1      # INITIALIZE FWEIGHT COMP AS IF FOR P66
      TS      BUF

      CS      MODREG      # ARE WE IN FACT IN P66?
      AD      DEC66
      EXTEND
      BZF      FWCOMP      # YES

      CA      PIPTIME +1      # NO: INITIALIZE FOR TWO SECOND PERIOD
      TS      BUF
      CAF      4SECS
      TCF      FWCOMP +1

FWCOMP      CAF      2SECS
+1          TS      Q
          EXTEND
          MP      BIT6
          LXCH      BUF +1
          CS      BUF      # TIME OF LAST PIPA READING.
          AD      TIME1
```

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AD	THROTLAG	# COMPENSATE FOR ENGINE RESPONSE LAG
MASK	LOW8	# MAKE SURE SMALL AND POSITIVE
ZL		
EXTEND		

295      *(Page LM0796 295)*≡      (289b 771)

DV      Q  
 EXTEND  
 MP      PIF  
 DOUBLE  
 DXCH    FWEIGHT  
 CCS      PIF  
 AD      ONE  
 TCF      +2  
 AD      ONE  
 EXTEND  
 MP      PIF  
 EXTEND  
 DV      BUF +1  
 ZL  
 DAS      FWEIGHT

THDUMP      TC      RTNHOLD

# FLATOUT THROTTLES UP THE DESCENT ENGINE, AND IS CALLED AS A BASIC SUBROUTINE.

FLATOUT      CAF      BIT13      # 4096 PULSES  
 WHATOUT      TS      PIFPSET      # USE PIFPSET SO FWEIGHT WILL BE ZERO  
              CS      ZERO  
              TS      FCOLD  
              TS      PIF  
              EXTEND  
              QXCH    RTNHOLD  
              TCF      DOIT

# MASSMULT SCALES ACCELERATION, ARRIVING IN A AND L IN UNITS OF 2(-4) M/CS/CS, TO FORCE IN PULS

MASSMULT      EXTEND  
              QXCH    BUF  
              DXCH    MPAC  
              TC      DMP  
              ADRES    MASS  
              TC      DMP      # LEAVES PROPERLY SCALED FORCE IN MPAC  
              ADRES    SCALEFAC  
              TC      TPAGREE  
              CA      MPAC  
              EXTEND  
              BZF      +3  
              CAF      POSMAX  
              TC      BUF  
              DXCH    MPAC +1

```

                                TC      BUF
296a  <Page LM0797 296a>≡ (289b 771)
      # CONSTANTS --

      FEXTRA      =      BIT13      # FEXT +5.13309020 E+4

      /AF/CNST    DEC      .13107

      # * * * * *

```

## 1.20 lunar landing guidance equations

```

296b  <lunar landing guidance equations 296b>≡ (7)
      <Page LM0798 297>
      <Page LM0799 298>
      <Page LM0800 300>
      <Page LM0801 302>
      <Page LM0802 304>
      <Page LM0803 306>
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      <Page LM0825 344>
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      <Page LM0827 347a>
      <Page LM0828 347b>

```



```
EBANK= E2DPS
COUNT* $$/F2DPS
```

```
# *****
# LUNAR LANDING FLIGHT SEQUENCE TABLES
# *****
```

```
# FLIGHT SEQUENCE TABLES ARE ARRANGED BY FUNCTION.  THEY ARE REFERENCED USING AS AN INDEX THE F
# WCHPHASE = -1 ---> IGNALG
# WCHPHASE = 0 ---> BRAKQUAD
# WCHPHASE = 1 ---> APPRQUAD
# WCHPHASE = 2 ---> VERTICAL
```

```
*****
```

```
# ROUTINES FOR STARTING NEW GUIDANCE PHASES:
```

```
NEWPHASE      TCF      TTFINCR      # IGNALG
               TCF      TTFINCR      # BRAKQUAD
               TCF      STARTP64      # APPRQUAD
               TCF      P65START      # VERTICAL
```

```
# PRE-GUIDANCE COMPUTATIONS:
```

```
PREGUIDE      TCF      CALCRGVG      # IGNALG
               TCF      RGVGCALC      # BRAKQUAD
               TCF      REDESIG       # APPRQUAD
               TCF      RGVGCALC      # VERTICAL
```

```
# GUIDANCE EQUATIONS:
```

```
WHATGUID      TCF      TTF/8CL       # IGNALG
               TCF      TTF/8CL       # BRAKQUAD
               TCF      TTF/8CL       # APPRQUAD
               TCF      VERTGUID      # VERTICAL
```

```
# POST GUIDANCE EQUATION COMPUTATIONS:
```

```
AFTRGUID      TCF      CGCALC        # IGNALG
               TCF      CGCALC        # BRAKQUAD
               TCF      CGCALC        # APPRQUAD
               TCF      STEER?        # VERTICAL
```

298  $\langle$ Page LM0799 298 $\rangle \equiv$ 

(296b 746)

# WINDOW VECTOR COMPUTATIONS:

	TCF	EXGSUB	# IGNALG
WHATEXIT	TCF	EXBRAK	# BRAKQUAD
	TCF	EXNORM	# APPRQUAD

# DISPLAY ROUTINES:

WHATDISP	TCF	P63DISPS	# BRAKQUAD
	TCF	P64DISPS	# APPRQUAD
	TCF	VERTDISP	# VERTICAL

# ALARM ROUTINE FOR TTF COMPUTATION:

	TCF	1406POO	# IGNALG
WHATALM	TCF	1406ALM	# BRAKQUAD
	TCF	1406ALM	# APPRQUAD

# INDICES FOR REFERENCING TARGET PARAMETERS

	OCT	0	# IGNALG
TARGETDEX	OCT	0	# BRAKQUAD
	OCT	34	# APPRQUAD

\*\*\*\*\*  
# ENTRY POINTS: ?GUIDSUB FOR THE IGNITION ALGORITHM, LUNLAND FOR SERVOUT  
\*\*\*\*\*

# IGNITION ALGORITHM ENTRY: DELIVERS N PASSES OF QUADRATIC GUIDANCE

?GUIDSUB	EXIT		
	CAF	TWO	# N = 3
	TS	NGUIDSUB	
	TCF	GUILDRET +2	
GUIDSUB	TS	NGUIDSUB	# ON SUCCEEDING PASSES SKIP TTFINCR
	TCF	CALCRGVG	

# NORMAL ENTRY: CONTROL COMES HERE FROM SERVOUT

LUNLAND	TC	PHASCHNG	
	OCT	00035	# GROUP 5: RETAIN ONLY PIPA TASK
	TC	PHASCHNG	
	OCT	05023	# GROUP 3: PROTECT GUIDANCE WITH PRIO 21
	OCT	21000	# JUST HIGHER THAN SERVICER'S PRIORITY

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(296b 746)

```

*****
# GULDENSTERN:  AUTO-MODES MONITOR (R13)
*****

```

COUNT\* \$\$/R13

```

# HERE IS THE PHILOSOPHY OF GULDENSTERN:      ON EVERY APPEARANCE OR DISAPPEARANCE
# DISCRETE TO SELECT P67 OR P66 RESPECTIVELY:  ON EVERY APPEARANCE OF THE ATTITUDE-I
# UNLESS THE CURRENT PROGRAM IS P67 IN WHICH CASE THERE IS NO CHANGE

```

```

GULDEN      EXTEND      # IS UN-AUTO-THROTTLE DISCRETE PRESENT?
# STERN      # RSB 2009: Not originally a comment.

```

```

      READ CHAN30
      MASK      BIT5
      CCS      A
      TCF      STARTP67      # YES
P67NOW?      TC      CHECKMM      # NO:  ARE WE IN P67 NOW?
      DEC      67
      TCF      STABL?      # NO
STARTP66      TC      FASTCHNG      # YES
      TC      NEWMODEX
DEC66      DEC      66
      EXTEND
      DCA      HDOTDISP      # SET DESIRED ALTITUDE RATE = CURRENT
      DXCH      VDGVERT      #      ALTITUDE RATE.
STRTP66A      TC      INTPRET
      SLOAD      PUSH
      PBIASZ
      SLOAD      PUSH
      PBIASY
      SLOAD      VDEF
      PBIASX
      VXSC      SET
      BIASFACT
      RODFLAG
      STOVL      VBIAS
      TEMX
      VCOMP
      STOVL      OLDPIPAX
      ZEROVECS
      STODL      DELVROD
      RODSCALE
      STODL      RODSCAL1
      PIPTIME
      STORE      LASTTPIP

```

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	EXIT	
	CAF	ZERO
	TS	FCOLD
	TS	FWEIGHT
	TS	FWEIGHT +1
VRTSTART	TS	WCHVERT

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(296b 746)

	CAF	TWO	# WCHPHASE = 2 ---> VERTICAL: P65,P66,P67
	TS	WCHPHOLD	
	TS	WCHPHASE	
	TC	BANKCALL	# TEMPORARY, I HOPE HOPE HOPE
	CADR	STOPRATE	# TEMPORARY, I HOPE HOPE HOPE
	TC	DOWNFLAG	# PERMIT X-AXIS OVERRIDE
	ADRES	XOVINFLG	
	TC	DOWNFLAG	
	ADRES	REDFLAG	
	TCF	VERTGUID	
STARTP67	TC	NEWMODEX	# NO HARM IN "STARTING" P67 OVER AND OVER
	DEC	67	# SO NO NEED FOR A FASTCHNG AND NO NEED
	CAF	ZERO	# TO SEE IF ALREADY IN P67.
	TS	RODCOUNT	
	CAF	TEN	
	TCF	VRTSTART	
STABL?	CAF	BIT13	# IS UN-ATTITUDE-HOLD DISCRETE PRESENT?
	EXTEND		
	RAND	CHAN31	
	CCS	A	
	TCF	GUILDRET	# YES ALL'S WELL
P66NOW?	CS	MODREG	
	AD	DEC66	
	EXTEND		
	BZF	RESTART?	
	CA	RODCOUNT	# NO. HAS THE ROD SWITCH BEEN "CLICKED"?
	EXTEND		
	BZF	GUILDRET	# NO. CONTINUE WITH AUTOMATIC LANDING
	TCF	STARTP66	# YES. SWITCH INTO THE ROD MODE.
RESTART?	CA	FLAGWRD1	# HAS THERE BEEN A RESTART?
	MASK	RODFLBIT	
	EXTEND		
	BZF	STRTP66A	# YES. REINITIALIZE BUT LEAVE VDGVERT AS
			# IS.
	TCF	VERTGUID	# NO: CONTINUE WITH R.O.D.

# \*\*\*\*\*  
# INITIALIZATION FOR THIS PASS  
# \*\*\*\*\*

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303

COUNT\* \$\$/F2DPS

GUILDRET

CAF  
TS

ZERO  
RODCOUNT

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```

+2      EXTEND
        DCA      TPIP
        DXCH     TPIPOLD

        TC       FASTCHNG

        EXTEND
        DCA      PIPTIME1
        DXCH     TPIP

        EXTEND
        DCA      TTF/8
        DXCH     TTF/8TMP

        CCS      FLPASSO
        TCF      TTFINCR

BRSPOT1  INDEX   WCHPHASE
        TCF      NEWPHASE

```

```

# *****
# ROUTINES TO START NEW PHASES
# *****

```

```

P65START  TC      NEWMODEX
          DEC      65
          CS       TWO
          TS       WCHVERT
          TC       DOWNFLAG      # PERMIT X-AXIS OVERRIDE
          ADRES    XOVINFLG
          TCF      TTFINCR

STARTP64  TC      NEWMODEX
          DEC      64
          CA       DELTTFAP      # AUGMENT TTF/8
          ADS      TTF/8TMP
          CA       BIT12         # ENABLE RUPT10
          EXTEND
          WOR      CHAN13
          TC       DOWNFLAG      # INITIALIZE REDESIGNATION FLAG
          ADRES    REDFLAG

```

```

#          (CONTINUE TO TTFINCR)

```



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```
# *****
# INCREMENT TTF/8, UPDATE LAND FOR LUNAR ROTATION, DO OTHER USEFUL THINGS
# *****
#
#           TTFINCR COMPUTATIONS ARE AS FOLLOWS --
```

```

306  (Page LM0803 306)≡ (296b 746)
#      TTF/8 UPDATED FOR TIME SINCE LAST PASS:
#      TTF/8 = TTF/8 + (TPIP - TPIPOLD)/8
#      LANDING SITE VECTOR UPDATED FOR LUNAR ROTATION:
#      ----
#      LAND = /LAND/ UNIT(LAND - LAND(TPIP - TPIPOLD) * WM)
#      SLANT RANGE TO LANDING SITE, FOR DISPLAY:
#      ----
#      RANGEDSP = ABVAL(LAND - R)

TTFINCR      TC      INTPRET
              DLOAD   DSU
              TPIP
              TPIPOLD
              SLR      PUSH          # SHIFT SCALES DELTA TIME TO 2(17) CSECS
              11D
              VXSC     VXV
              LAND
              WM
              BVSU     RTB
              LAND
              NORMUNIT
              VXSC     VSL1
              /LAND/
              STODL    LANDTEMP
              EXIT

              DXCH     MPAC
              DAS      TTF/8TMP      # NOW HAVE INCREMENTED TTF/8 IN TTF/8TMP

              TC      FASTCHNG

              EXTEND
              DCA      TTF/8TMP
              DXCH     TTF/8

              EXTEND
              DCA      LANDTEMP
              DXCH     LAND
              EXTEND
              DCA      LANDTEMP +2
              DXCH     LAND      +2
              EXTEND
              DCA      LANDTEMP +4
              DXCH     LAND      +4

```

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(296b 746)

```
TC      TDISPSET
TC      FASTCHNG      # SINCE REDESIG MAY CHANGE LANDTEMP

BRSPOT2      INDEX    WCHPHASE
              TCF      PREGUIDE
```

```
# *****
# LANDING SITE PERTURBATION EQUATIONS
# *****
```

```
REDESIG      CA      FLAGWRD6      # IS REDFLAG SET?
              MASK    REDFLBIT
              EXTEND
              BZF      RGVGCALC      # NO:  SKIP REDESIGNATION LOGIC

              CA      TREDES      # YES:  HAS TREDES REACHED ZERO?
              EXTEND
              BZF      RGVGCALC      # YES:  SKIP REDESIGNATION LOGIC

              INHINT
              CA      ELINCR1
              TS      ELINCR
              CA      AZINCR1
              TS      AZINCR
              TC      FASTCHNG

              CA      ZERO
              TS      ELINCR1
              TS      AZINCR1
              TS      ELINCR  +1
              TS      AZINCR  +1

              CA      FIXLOC      # SET PD TO 0
              TS      PUSHLOC

              TC      INTERPRET
              VLOAD    VSU
                      LAND
                      R      #
              RTB      PUSH      # PUSH DOWN UNIT (LAND - R)
                      NORMUNIT
              VXV      VSL1
                      YNBPIP      #
              VXSC      PDDL      # PUSH DOWN - ELINCR(YNB * UNIT(LAND - R))
                      ELINCR
```

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AZINCR  
VXSC VSU  
YNBPIP  
VAD PUSH

# RESULTING VECTOR IS 1/2 REAL SIZE

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(296b 746)

```

                                DLOAD  DSU          # MAKE SURE REDESIGNATION IS NOT
                                0          #      TOO CLOSE TO THE HORIZON.
                                DEPRCRIT
    BMN      DLOAD
                                REDES1
                                DEPRCRIT
    REDES1   STORE  0
                                DLOAD  DSU
                                LAND
                                R
    DDV      VXSC
                                0
    VAD      UNIT
                                R
    VXSC      VSL1
                                /LAND/
    STORE    LANDTEMP
    EXIT                                           # LOOKANGL WILL BE COMPUTED AT RGVGCALC

    TC      FASTCHNG

    EXTEND
    DCA      LANDTEMP
    DXCH     LAND
    EXTEND
    DCA      LANDTEMP +2
    DXCH     LAND +2
    EXTEND
    DCA      LANDTEMP +4
    DXCH     LAND +4

    TCF      RGVGCALC

# *****
# COMPUTE STATE IN GUIDANCE COORDINATES
# *****
#
#      RGVGCALC COMPUTATIONS ARE AS FOLLOWS:--
#      VELOCITY RELATIVE TO THE SURFACE:
#
#      -----  -  -  --
#      ANGTERM = V + R * WM
#      STATE IN GUIDANCE COORDINATES:
#
#      *      -  -----
#      RGU = CG (R - LAND)

```

$$\begin{array}{l} \# \\ \# \end{array} \quad \begin{array}{c} \text{---} \quad * \\ \text{VGU} = \text{CG} \left( \bar{\text{V}} - \bar{\text{WM}} * \bar{\text{R}} \right) \end{array}$$

311 (Page LM0806 311)≡

(296b 746)

```

#
#      HORIZONTAL VELOCITY FOR DISPLAY
#
#      VHORIZ = 8 ABVAL (0, VG , VG )
#                  2      1
#      DEPRESSION ANGLE FOR DISPLAY:
#
#      LOOKANGL = ARCSIN(UNIT(R - LAND).XMBPIP)
#
CALCRGVG      TC      INTPRET      # IN IGNALG, COMPUTE V FROM INTEGRATION
VLOAD      MXV      #      OUTPUT AND TRIM CORRECTION TERM
            VATT1      #      COMPUTED LAST PASS AND LEFT IN UNFC/2
            REFSMMAT
VSR1      VAD
            UNFC/2
STORE      V
EXIT

RGVGCALC      TC      INTPRET      # ENTER HERE TO RECOMPUTE RG AND VG
VLOAD      VXV
            R
            WM
VAD      VSR2      # RESCALE TO UNITS OF 2(9) M/CS
            V
STORE      ANGTERM
MXV
            CG      # NO SHIFT SINCE ANGTERM IS DOUBLE SIZED
STORE      VGU
PDDL      VDEF      # FORM (0,VG ,VG ) IN UNITS OF 2(10) M/CS
            ZEROVECS      #      2      1
ABVAL      SL3
STOVL      VHORIZ      # VHORIZ FOR DISPLAY DURING P65.
            R      #
VSU      PUSH      # PUSH DOWN R - LAND
            LAND
MXV      VSL1
            CG
STORE      RGU
ABVAL
STOVL      RANGEDSP
RTB      DOT      # NOW IN MPAC IS SINE(LOOKANGL)/4
            NORMUNIT
            XNBPIP
EXIT

```

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CA      FIXLOC  
TS      PUSHLOC

# RESET PUSH DOWN POINTER



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313 (Page LM0807 313)≡

(296b 746)

```

      CA      MPAC      # COMPUTE LOOKANGLE ITSELF
DOUBLE
      TC      BANKCALL
      CADR     SPARCSIN -1
      AD       1/2DEG
EXTEND
      MP      180DEGS
      TS      LOOKANGL      # LOOKANGL FOR DISPLAY DURING P64

BRSPOT3      INDEX      WCHPHASE
              TCF        WHATGUID

# *****
# TTF/8 COMPUTATION
# *****

TTF/8CL      TC      INTPRETX
              DLOAD*
              JDG2TTF,1
STODL*      TABLTTF +6      # A(3) = 8 JDG TO TABLTTF
              ADG2TTF,1      #                2
STODL      TABLTTF +4      # A(2) = 6 ADG TO TABLTTF
              VGU      +4      #                2
DMP          DAD*
              3/4DP
              VDG2TTF,1
STODL*      TABLTTF +2      # A(1) = (6 VGU + 18 VDG )/8 TO TABLTTF
              RDG +4,1      #                2                2
DSU          DMP
              RGU +4
              3/8DP
STORE      TABLTTF      # A(0) = -24 (RGU - RDG )/64 TO TABLTTF
EXIT      #                2                2

      CA      BIT8
      TS      TABLTTF +10      # FRACTIONAL PRECISION FOR TTF TO TABLE

EXTEND
DCA          TTF/8
DXCH         MPAC      # LOADS TTF/8 (INITIAL GUESS) INTO MPAC
CAF          TWO      # DEGREE - ONE
TS           L
CAF          TABLTTF
TC           ROOTPSRS      # YIELDS TTF/8 IN MPAC
INDEX        WCHPHASE
```

TCF WHATALM

EXTEND

# GOOD RETURN

DCA MPAC

# FETCH TTF/8 KEEPING IT IN MPAC

DXCH TTF/8

# CORRECTED TTF/8

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TC TDISPSET

# (CONTINUE TO QUADGUID)

# \*\*\*\*\*

# MAIN GUIDANCE EQUATION

# \*\*\*\*\*

#

# AS PUBLISHED --

#

$$\text{ACG} = \text{ADG} + \frac{6(\text{VDG} + \text{VG})}{\text{TTF}} + \frac{12(\text{RDG} - \text{RG})}{(\text{TTF})(\text{TTF})}$$

# AS HERE PROGRAMMED --

#

$$\text{ACG} = \frac{3 \left( \frac{1}{4}(\text{RDG} - \text{RG}) - \left( \frac{\text{VDG} + \text{VG}}{\text{TTF}/8} \right) \right)}{\text{TTF}/8} + \text{ADG}$$

QUADGUID

CS TTF/8

AD LEADTIME

# LEADTIME IS A NEGATIVE NUMBER

AD POSMAX

# SAFEGUARD THE COMPUTATIONS THAT FOLLOW

TS L

# BY FORCING -TTF\*LEADTIME &gt; OR = ZERO

CS L

AD L

ZL

EXTEND

DV TTF/8

TS BUF

# - RATIO OF LAG-DIMINISHED TTF TO TTF

EXTEND

SQUARE

TS BUF +1

AD BUF

XCH BUF +1

# RATIO SQUARED - RATIO

AD BUF +1

TS MPAC

# COEFFICIENT FOR VGU TERM

AD BUF +1

INDEX FIXLOC

TS 26D

# COEFFICIENT FOR RDG-RGU TERM

AD BUF +1

INDEX FIXLOC

TS 28D

# COEFFICIENT FOR VDG TERM

AD BUF

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AD POSMAX

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	AD	BUF +1	
	AD	BUF +1	
	INDEX	FIXLOC	
	TS	30D	# COEFFICIENT FOR ADG TERM
	CAF	ZERO	
	TS	MODE	
	TC	INTPRETX	
	VXSC	PDDL	
		VGU	
		28D	
	VXSC*	PDVL*	
		VDG,1	
		RDG,1	
	VSU	V/SC	
		RGU	
		TTF/8	
	VSR2	VXSC	
		26D	
	VAD	VAD	
	V/SC	VXSC	
		TTF/8	
		3/4DP	
	PDDL	VXSC*	
		30D	
		ADG,1	
AFCCALC1	VAD		
	VXM	VSL1	# VERGUID COMES HERE
		CG	
	PDVL	V/SC	
		GDT/2	
		GSCALE	
	BVSU	STADR	
	STORE	UNFC/2	# UNFC/2 NEED NOT BE UNITIZED
	ABVAL		
AFCCALC2	STODL	/AFC/	# MAGNITUDE OF AFC FOR THROTTLE
		UNFC/2	# VERTICAL COMPONENT
	DSQ	PDDL	
		UNFC/2 +2	# OUT-OF-PLANE
	DSQ	PDDL	
		HIGHESTF	
	DDV	DSQ	
		MASS	#
	DSU	DSU	# AMAXHORIZ = SQRT(ATOTAL - A <sup>2</sup> - A <sup>2</sup> - A <sup>2</sup> )

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	BPL	DLOAD	#	1	0
		AFCCALC3			
		ZEROVECS			
AFCCALC3	SQRT	DAD			
		UNFC/2 +4			

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```

      BPL      BDSU
              AFCCLEND
              UNFC/2 +4
      STORE    UNFC/2 +4
AFCCLEND     EXIT
      TC       FASTCHNG

      CA       WCHPHASE      # PREPARE FOR PHASE SWITCHING LOGIC
      TS       WCHPHOLD
      INCR     FLPASSO      # INCREMENT PASS COUNTER

BRSPOT4      INDEX    WCHPHASE
              TCF      AFTRGUID

```

```

# *****
# ERECT GUIDANCE-STABLE MEMBER TRANSFORMATION MATRIX
# *****

```

```

CGCALC      CAF      EBANK5
              TS       EBANK
              EBANK=   TCGIBRAK
              EXTEND
              INDEX    WCHPHASE
              INDEX    TARGTDEX
              DCA       TCGFBRAK
              INCR     BBANK
              INCR     BBANK
              EBANK=   TTF/8
              AD       TTF/8
              XCH      L
              AD       TTF/8
              CCS      A
              CCS      L
              TCF      EXTLOGIC
              TCF      EXTLOGIC
              NOOP

              TC       INTPRETX
              VLOAD    UNIT
                      LAND
              STODL    CG
                      TTF/8
              DMP*     VXSC
                      GAINBRAK,1      # NUMERO MYSTERIOSO
                      ANGTERM

```

VAD

LAND

VSU

RTB

R

NORMUNIT



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```

      VXV      RTB
              LAND
              NORMUNIT
      STOVL     CG +6          # SECOND ROW
              CG
      VXV      VSL1
              CG +6
      STORE     CG +14
      EXIT

#             (CONTINUE TO EXTLOGIC)
#
# *****
# PREPARE TO EXIT
# *****
#
# DECIDE (1) HOW TO EXIT, AND (2) WHETHER TO SWITCH PHASES
#
EXTLOGIC      INDEX  WCHPHASE      # WCHPHASE = 1  APPRQUAD
              CA     TENDBRAK      # WCHPHASE = 0  BRAKQUAD
              AD     TTF/8

EXSPOT1       EXTEND
              INDEX  WCHPHASE
              BZMF   WHATEXIT

              TC     FASTCHNG

              CA     WCHPHOLD
              AD     ONE
              TS     WCHPHASE
              CA     ZERO
              TS     FLPASSO      # RESET FLPASSO

              INDEX  WCHPHOLD
              TCF    WHATEXIT

# *****
# ROUTINES FOR EXITING FROM LANDING GUIDANCE
# *****
#
# 1.    EXGSUB IS THE RETURN WHEN GUIDSUB IS CALLED BY THE IGNITION ALGORITHM.
# 2.    EXBRAK IN THE EXIT USED DURING THE BRAKING PHASE.  IN THIS CASE UNIT(R) IS THE WINDOW P
# 3.    EXNORM IS THE EXIT USED AT OTHER TIMES DURING THE BURN.
# (EXOVFLOW IS A SUBROUTINE OF EXBRAK AND EXNORM CALLED WHEN OVERFLOW OCCURRED ANYWHERE IN GUID

```

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EXGSUB TC INTPRET # COMPUTE TRIM VELOCITY CORRECTION TERM.

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	VLOAD	RTB	
		UNFC/2	
		NORMUNIT	
	VXSC	VXSC	
		ZOOMTIME	
		TRIMACCL	
	STORE	UNFC/2	
	EXIT		
	CCS	NGUIDSUB	
	TCF	GUIDSUB	
	CCS	NIGNLOOP	
	TCF	+3	
	TC	ALARM	
	OCT	01412	
+3	TC	POSTJUMP	
	CADR	DDUMCALC	
EXBRAK	TC	INTPRET	
	VLOAD		
		UNIT/R/	
	STORE	UNWC/2	
	EXIT		
	TCF	STEER?	
EXNORM	TC	INTPRET	
	VLOAD	VSU	
		LAND	
		R	
	RTB		
		NORMUNIT	
	STORE	UNWC/2	# UNIT(LAND - R) IS TENTATIVE CHOICE
	VXV	DOT	
		XNBPIP	
		CG +6	
	EXIT		# WITH PROJ IN MPAC 1/8 REAL SIZE
	CS	MPAC	# GET COEFFICIENT FOR CG +14
	AD	PROJMAX	
	AD	POSMAX	
	TS	BUF	
	CS	BUF	
	ADS	BUF	# RESULT IS 0 IF PROJMAX - PROJ NEGATIVE

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CS	PROJMIN	# GET COEFFICIENT FOR UNIT(LAND - R)
AD	MPAC	
AD	POSMAX	
TS	BUF +1	
CS	BUF +1	

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```

ADS      BUF +1      # RESULT IS 0 IF PROJ - PROJMIN NEGATIVE

UNWCLOOP  CAF      FOUR
          MASK     SIX
          TS       Q
          CA       EBANK5
          TS       EBANK
          EBANK=   CG
          CA       BUF
          EXTEND
          INDEX    Q
          MP       CG +14
          INCR     BBANK
          EBANK=   UNWC/2
          INDEX    Q
          DXCH     UNWC/2
          EXTEND
          MP       BUF +1
          INDEX    Q
          DAS      UNWC/2
          CCS      Q
          TCF      UNWCLOOP

          INCR     BBANK
          EBANK=   PIF

STEER?    CA       FLAGWRD2      # IF STEERSW DOWN NO OUTPUTS
          MASK     STEERBIT
          EXTEND
          BZF      RATESTOP

EXVERT    CA       OVFFIND      # IF OVERFLOW ANYWHERE IN GUIDANCE
          EXTEND      #          DON'T CALL THROTTLE OR FINDCDUW
          BZF      +13

EXOVFLOW  TC       ALARM        # SOUND THE ALARM NON-ABORTIVELY
          OCT       01410

RATESTOP  CAF      BIT13        # ARE WE IN ATTITUDE-HOLD?
          EXTEND
          RAND      CHAN31
          EXTEND
          BZF      DISPEXIT      # YES

          TC       BANKCALL      # NO: DO A STOPRATE
```

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CADR      STOPRATE

TCF      DISPEXIT

GDUMP1      TC      THROTTLE

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TC INTPRET  
CALL  
FINDCDUW -2  
EXIT

# (CONTINUE TO DISPEXIT)

# \*\*\*\*\*  
# GUIDANCE LOOP DISPLAYS  
# \*\*\*\*\*

DISPEXIT EXTEND # KILL GROUP 3: DISPLAYS WILL BE  
DCA NEGO # RESTORED BY NEXT GUIDANCE CYCLE.  
DXCH -PHASE3

+3 CS FLAGWRD8 # IF FLUNDISP IS SET, NO DISPLAY THIS PASS  
MASK FLUNDBIT  
EXTEND  
BZF ENDLLJOB # TO PICK UP THE TAG  
  
INDEX WCHPHOLD  
TCF WHATDISP

-2 TC PHASCHNG # KILL GROUP 5  
OCT 00035

P63DISPS CAF V06N63  
DISPCOMN TC BANKCALL  
CADR REGODSPR

ENDLLJOB TCF ENDOFJOB

P64DISPS CA TREDES # HAS TREDES REACHED ZERO?  
EXTEND  
BZF RED-OVER # YES: CLEAR REDESIGNATION FLAG  
  
CS FLAGWRD6 # NO: IS REDFLAG SET?  
MASK REDFLBIT  
EXTEND  
BZF REDES-OK # YES: DO STATIC DISPLAY  
  
CAF V06N64 # OTHERWISE USE FLASHING DISPLAY  
TC BANKCALL  
CADR REFLASHR  
TCF GOTOP00H # TERMINATE

```

                                TCF      P64CEED      # PROCEED      PERMIT REDESIGNATIONS
                                TCF      P64DISPS     # RECYCLE
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                                TCF      ENDLLJOB

P64CEED      CAF      ZERO
                                TS      ELINCR1
                                TS      AZINCR1

                                TC      UPFLAG      # ENABLE REDESIGNATION LOGIC
                                ADRES    REDFLAG

                                TCF      ENDOFJOB

RED-OVER      TC      DOWNFLAG
                                ADRES    REDFLAG
REDES-OK      CAF      V06N64
                                TCF      DISPCOMN

VERTDISP      CAF      V06N60
                                TCF      DISPCOMN

# *****
# GUIDANCE FOR P65
# *****

VERTGUID      CCS      WCHVERT
                                TCF      P67VERT      # POSITIVE NON-ZERO ---> P67
                                TCF      P66VERT      # +0

#
#      THE P65 GUIDANCE EQUATION IS AS FOLLOWS --
#
#              ----  ---
#              V2FG - VGU
#      ACG = -----
#              TAUVERT

P65VERT      TC      INTPRET
                                VLOAD    VSU
                                V2FG
                                VGU
                                V/SC     GOTO
                                TAUVERT
                                AFCCALC1

```



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```
# *****
# GUIDANCE FOR P66
# *****
```

```
P66VERT      TC      POSTJUMP
              CADR     P66VERTA
```

```
P67VERT      TC      PHASCHNG      # TERMINATE GROUP 3.
              OCT      00003
```

```
              TC      INTPRET
              VLOAD    GOTO
                      V
                      VHORCOMP
```

```
              SETLOC  P66LOC
              BANK
              COUNT*  $$/F2DPS
```

```
RODTASK      CAF      PRI022
              TC      FINDVAC
              EBANK=   DVCNTR
              2CADR    RODCOMP
```

```
              TCF      TASKOVER
```

```
P66VERTA     TC      PHASCHNG      # TERMINATE GROUP 3.
              OCT      00003
```

```
              CAF      1SEC
              TC      TWIDDLE
              ADRES     RODTASK
```

```
RODCOMP      INHINT
              CAF      ZERO
              XCH      RODCOUNT
              EXTEND
              MP      RODSCAL1
              DAS      VDGVERT
```

# UPDATE DESIRED ALTITUDE RATE.

# SET OLDPIPAX,Y,Z = PIPAX,Y,Z

```
              DCA      PIPAX
              DXCH     OLDPIPAX
              DXCH     RUPTREG1
              CA      PIPAZ
```

# SET RUPTREG1,2,3 = OLDPIPAX,Y,Z

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XCH OLDPIPAZ  
XCH RUPTREG3

EXTEND # SNAPSHOT TIME OF PIPA READING.  
DCA TIME2

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```

DXCH  THISTPIP

CA      OLDPIPAX
AD      PIPATMPX
TS      MPAC          # MPAC(X) = PIPAX + PIPATMPX
CA      OLDPIPAY
AD      PIPATMPY
TS      MPAC +3       # MPAC(Y) = PIPAY + PIPATMPY
CA      OLDPIPAZ
AD      PIPATMPZ
TS      MPAC +5       # MPAC(Z) = PIPAZ + PIPATMPZ

CS      OLDPIPAX
AD      TEMX
AD      RUPTREG1
TS      DELVROD
CS      OLDPIPAY
AD      TEMY
AD      RUPTREG2
TS      DELVROD +2
CS      OLDPIPAZ
AD      TEMZ
AD      RUPTREG3
TS      DELVROD +4

CAF     ZERO
TS      MPAC +1       # ZERO LO-ORDER MPAC COMPONENTS
TS      MPAC +4
TS      MPAC +6
TS      TEMX          # ZERO TEMX, TEMY, AND TEMZ SO WE WILL
TS      TEMY          #          KNOW WHEN READACCS CHANGES THEM.
TS      TEMZ
CS      ONE
TS      MODE
TC      INTERPRET

ITRPNT1 VXSC  PDDL          # SCALE MPAC TO M/CS *2(-7) AND PUSH      (6)
          KPIP1
          THISTPIP

DSU
          PIPTIME
STORE    30D          # 30-31D CONTAINS TIME IN CS SINCE PIPTIME
DDV      PDVL          #
          4SEC(28)
          GDT/2
VSU      VXSC          #
                                     (6)
```

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VSL2 VBIAS  
VAD  
V  
VAD STADR  
STOVL 24D

#  
# STORE UPDATED VELOCITY IN 24-29D (0)

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```
R
UNIT
STORE 14D
DOT SL1
      24D
STODL HDOTDISP # UPDATE HDOTDISP RATE FOR NOUN 63.
      30D
SL DMP
      11D
      HDOTDISP
DAD DSU
      36D
      /LAND/
STODL HCALC1 # UPDATE HCALC1 FOR NOUN 63.
      HDOTDISP
BDSU DDV
      VDGVERT
      TAUROD
PDVL ABVAL # (2)
      GDT/2
DDV SR2
      GSCALE
STORE 20D
DAD # (0)
PDVL CALL # (2)
      UNITX
      CDU*NBSM
DOT
      14D
STORE 22D
BDDV STADR # (0)
STOVL /AFC/
      DELVROD
VXSC VAD
      KPIP1
      VBIAS
ABVAL PDDL # (2)
      THISTPIP
DSU PDDL # (4)
      LASTTPIP
      THISTPIP
STODL LASTTPIP # (2)
DDV BDDV # (0)
      SHFTFACT
PDDL DMP # (2)
```

	FWEIGHT
	BIT1H
DDV	DDV
	MASS
	SCALEFAC

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	DAD	PDDL	#	(4)
		OD		
		20D		
	DDV	DSU	#	(2)
		22D		
	DMP	DAD		
		LAG/TAU		
		/AFC/		
	PDDL	DDV	#	(4)
		MAXFORCE		
		MASS		
	PDDL	DDV	#	(6)
		MINFORCE		
		MASS		
	PUSH	BDSU	#	(8)
		2D		
	BMN	DLOAD	#	(6)
		AFCSPOT		
	DLOAD	PUSH	#	(6)
	BDSU	BPL		
		2D		
		AFCSPOT		
	DLOAD		#	(4)
AFCSPOT	DLOAD		#	(2), (4), OR (6)
	SETPD		#	(2)
		2D		
	STODL	/AFC/	#	(0)
ITRPNT2	EXIT			
	DXCH	MPAC	#	MPAC = MEASURED ACCELERATION.
	TC	BANKCALL		
	CADR	THROTTLE +3		
	TC	INTPRET		
	VLOAD		#	PICK UP UPDATED VELOCITY VECTOR.
		24D		
VHORCOMP	VSL2	VAD		
		DELVS		
	VSR2	PDVL		
		R		
	UNIT	VXSC		
		HDOTDISP		
	VSL1	BVSU		
	ABVAL			
	STORE	VHORIZ		
	EXIT			
	TC	BANKCALL	#	PUT UP V06N60 DISPLAY BUT AVOID PHASCHNG

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	CADR	DISPEXIT +3
BIT1H	OCT	00001
SHFTFACT	2DEC	1 B-17



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BIASFACT 2DEC 655.36 B-28

# \*\*\*\*\*  
# REDESIGNATOR TRAP  
# \*\*\*\*\*

BANK 11  
SETLOC F2DPS\*11  
BANK

COUNT\* \$\$/F2DPS

PITFALL XCH BANKRUPT  
EXTEND  
QXCH QRUP

TC CHECKMM # IF NOT IN P64, NO REASON TO CONTINUE  
DEC 64  
TCF RESUME

EXTEND  
READ CHAN31  
COM  
MASK ALL4BITS  
TS ELVIRA  
CAF TWO  
TS ZERLINA  
CAF FIVE  
TC TWIDDLE  
ADRES REDESMON  
TCF RESUME

# REDESIGNATOR MONITOR (INITIATED BY PITFALL)

PREMON1 TS ZERLINA  
PREMON2 CAF SEVEN  
TC VARDELAY

REDESMON EXTEND  
READ 31  
COM  
MASK ALL4BITS  
XCH ELVIRA  
TS L  
CCS ELVIRA  
TCF PREMON2

# DO ANY BITS APPEAR THIS PASS?  
# Y: CONTINUE MONITOR

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CCS	L	# N:	ANY LAST PASS?
TCF	COUNT 'EM	#	Y: COUNT 'EM, RESET RUPT, TERMIN

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```

      CCS      ZERLINA      #      N:      HAS ZERLINA REACHED ZERO YET?
      TCF      PREMON1     #
RESETRPT CAF      BIT12     #      N:      DIMINISH ZERLINA, CONTINUE
      EXTEND
      WOR      CHAN13
      TCF      TASKOVER
      Y:      RESET RUPT. TERMINATE

COUNT'EM CAF      BIT13      # ARE WE IN ATTITUDE-HOLD?
      EXTEND
      RAND      CHAN31
      EXTEND
      BZF      RESETRPT      # YES: SKIP REDESIGNATION LOGIC.

      CA      L      # NO.
      MASK    -AZBIT
      CCS      A
-AZ      CS      AZEACH
      ADS      AZINCR1
      CA      L
      MASK    +AZBIT
      CCS      A
+AZ      CA      AZEACH
      ADS      AZINCR1
      CA      L
      MASK    -ELBIT
      CCS      A
-EL      CS      ELEACH
      ADS      ELINCR1
      CA      L
      MASK    +ELBIT
      CCS      A
+EL      CA      ELEACH
      ADS      ELINCR1
      TCF      RESETRPT
```

# THESE EQUIVALENCES ARE BASED ON GSOP CHAPTER 4, REVISION 16 OF P64LM

```

+ELBIT      =      BIT2      # -PITCH
-ELBIT      =      BIT1      # +PITCH
+AZBIT      =      BIT5
-AZBIT      =      BIT6
```

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```

ALL4BITS      OCT      00063
AZEACH        DEC      .03491      # 2 DEGREES
ELEACH        DEC      .00873      # 1/2 DEGREE

```

```

# *****
# R.O.D. TRAP
# *****

```

```

                                BANK      20
                                SETLOC    RODTRAP
                                BANK
                                COUNT*   $$/F2DPS      # *****
DESCBITS      MASK      BIT7          # COME HERE FROM MARKRUPT CODING WITH BIT
                                CCS       A            #          7 OR 6 OF CHANNEL 16 IN A; BIT 7 MEAN
                                CS        TWO          #          - RATE INCREMENT, BIT 6 + INCREMENT.
                                AD        ONE
                                ADS       RODCOUNT
                                TCF       RESUME        # TRAP IS RESET WHEN SWITCH IS RELEASED

                                BANK      31
                                SETLOC    F2DPS*31
                                BANK
                                COUNT*   $$/F2DPS

```

```

# *****
# DOUBLE PRECISION ROOT FINDER SUBROUTINE (BY ALLAN KLUMPP)
# *****
#
#                                N          N-1
#      ROOTPSRS FINDS ONE ROOT OF THE POWER SERIES A X  + A  X  + ... + A X + A
#                                N          N-1          1          0
# USING NEWTON'S METHOD STARTING WITH AN INITIAL GUESS FOR THE ROOT.  THE ENTERING DATA
#      A      SP      LOC-3          ADRES FOR REFERENCING PWR COF TABL
#      L      SP      N-1           N IS THE DEGREE OF THE POWER SERIES
#      MPAC   DP      X             INITIAL GUESS FOR ROOT
#
#      LOC-2N DP      A(0)
#      ...
#      LOC    DP      A(N)
#      LOC+2  SP      PRECROOT      PREC RQD OF ROOT (AS FRACT OF 1ST GUESS)
#

```

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```

# THE DP RESULT IS LEFT IN MPAC UPON EXIT, AND A SP COUNT OF THE ITERATIONS TO CONVERGENCE IS L
# RETURN IS NORMALLY TO LOC(TC ROOTPSRS)+3. IF ROOTPSRS FAILS TO CONVERGE TO IN 8 PASSES, RETU
# OUTPUTS ARE NOT TO BE TRUSTED.
#
# PRECAUTION: ROOTPSRS MAKES NO CHECKS FOR OVERFLOW OR FOR IMPROPER USAGE. IMPROPER USAGE COU
# PRECLUDE CONVERGENCE OR REQUIRE EXCESSIVE ITERATIONS. AS A SPECIFIC EXAMPLE, ROOTPSRS FORMS
# COEFFICIENT TABLE BY MULTIPLYING EACH A(I) BY I, WHERE I RANGES FROM 1 TO N. IF AN ELEMENT C
# COEFFICIENT TABLE = 1 OR >1 IN MAGNITUDE, ONLY THE EXCESS IS RETAINED. ROOTPSRS MAY CONVERGE
# ROOT NONETHELESS, BUT IT MAY TAKE AN EXCESSIVE NUMBER OF ITERATIONS. THEREFORE THE USER SHOU
#
# 1. USER'S RESPONSIBILITY TO ASSUR THAT I X A(I) < 1 IN MAGNITUDE FOR ALL I.
#
# 2. USER'S RESPONSIBILITY TO ASSURE OVERFLOW WILL NOT OCCUR IN EVALUATING EITHER THE RE
#
# POWER SERIES. THIS OVERFLOW WOULD BE PRODUCED BY SUBROUTINE POWRSERS, CALLED BY RO
#
# PRECLUDE EVENTUAL CONVERGENCE.
#
# 3. AT PRESENT, ERASABLE LOCATIONS ARE RESERVED ONLY FOR N UP TO 5. AN N IN EXCESS OF
#
# ALL ERASABLES USED BY ROOTPSRS ARE UNSWITCHED LOCATED IN THE REGION FROM MPAC-33 OC
#
# 4. THE ITERATION COUNT RETURNED IN MPAC+2 MAY BE USED TO DETECT ABNORMAL PERFORMANCE.

```

```

# STORE ENTERING DATA, INITIALIZE ERASABLES

ROOTPSRS      EXTEND
QXCH  RETROOT      # RETURN ADRES
TS    PWRPTR       # PWR TABLE POINTER
DXCH  MPAC +3      # PWR TABLE ADRES, N-1
CA    DERTABLL
TS    DERPTR       # DER TABL POINTER
TS    MPAC +5      # DER TABL ADRES
CCS   MPAC +4      # NO POWER SERIES DEGREE 1 OR LESS
TS    MPAC +6      # N-2
CA    ZERO         # MODE USED AS ITERATION COUNTER. MODE
TS    MODE         # MUST BE POS SO ABS WON'T COMP MPAC+3 ETC.

# COMPUTE CRITERION TO STOP ITERATING

EXTEND
DCA    MPAC        # FETCH ROOT GUESS, KEEPING IT IN MPAC
DXCH   ROOTPS      # AND IN ROOTPS
INDEX  MPAC +3     # PWR TABLE ADRES
CA     5           # PRECROOT TO A
TC     SHORTMP     # YIELDS DP PRODUCT IN MPAC
TC     USPRCADR
CADR   ABS         # YIELDS ABVAL OF CRITERION ON DX IN MPAC
DXCH   MPAC
DXCH   DXCRIT      # CRITERION

# SET UP DER COF TABL

```

342 (Page LM0824 342)≡

(296b 746)

```

                                EXTEND
                                INDEX  PWRPTR
                                DCA      3
                                DXCH    MPAC          # A(N) TO MPAC

                                CA      MPAC +4        # N-1 TO A

DERCLOOP                      TS      PWCNT          # LOOP COUNTER
                                AD      ONE
                                TC      DMPNSUB        # YIELDS DERCOF = I X A(I) IN MPAC
                                EXTEND
                                INDEX  PWRPTR
                                DCA      1
                                DXCH    MPAC          # (I-1) TO MPAC, FETCHING DERCOF
                                INDEX  DERPTR
                                DXCH    3             # DERCOF TO DER TABLE
                                CS      TWO
                                ADS     PWRPTR        # DECREMENT PWR POINTER
                                CS      TWO
                                ADS     DERPTR        # DECREMENT DER POINTER
                                CCS     PWCNT
                                TCF     DERCLOOP

                                # CONVERGE ON ROOT

ROOTLOOP                      EXTEND
                                DCA      ROOTPS        # FETCH CURRENT ROOT
                                DXCH    MPAC          # LEAVE IN MPAC
                                EXTEND
                                DCA      MPAC +5        # LOAD A, L WITH DER TABL ADRES, N-2
                                TC      POWRSERS        # YIELDS DERIVATIVE IN MPAC

                                EXTEND
                                DCA      ROOTPS
                                DXCH    MPAC          # CURRENT ROOT TO MPAC, FETCHING DERIVATIVE
                                DXCH    BUF           # LEAVE DERIVATIVE IN BUF AS DIVISOR
                                EXTEND
                                DCA      MPAC +3        # LOAD A, L WITH PWR TABL ADRES, N-1
                                TC      POWRSERS        # YIELDS RESIDUAL IN MPAC

                                TC      USPRCADR
                                CADR    DDV/BDDV      # YIELDS -DX IN MPAC

                                EXTEND
                                DCS     MPAC          # FETCH DX, LEAVING -DX IN MPAC
                                DAS     ROOTPS        # CORRECTED ROOT NOW IN ROOTPS

```

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TC	USPRCADR	
CADR	ABS	# YIELDS ABS(DX) IN MPAC
EXTEND		

344 (Page LM0825 344)≡

(296b 746)

```

          DCS      DXCRIT
          DAS      MPAC          # ABS(DX)-ABS(DXCRIT) IN MPAC

          CA      MODE
          MASK     BIT4          # KLUMPP SAYS GIVE UP AFTER EIGHT PASSES
          CCS      A
BADROOT   TC      RETROOT

          INCR     MODE          # INCREMENT ITERATION COUNTER
          CCS      MPAC          # TEST HI ORDER DX
          TCF      ROOTLOOP
          TCF      TESTLODX
          TCF      ROOTSTOR
TESTLODX  CCS      MPAC +1      # TEST LO ORDER DX
          TCF      ROOTLOOP
          TCF      ROOTSTOR
          TCF      ROOTSTOR
ROOTSTOR  DXCH     ROOTPS
          DXCH     MPAC
          CA      MODE
          TS      MPAC +2      # STORE SP ITERATION COUNT IN MPAC+2
          INDEX    RETROOT
          TCF      2

DERTABLL  ADRES    DERCOFN -3

# *****
# TRASHY LITTLE SUBROUTINES
# *****

INTPRETX  INDEX    WCHPHASE      # SET X1 ON THE WAY TO THE INTERPRETER
          CS      TARGTDEX
          INDEX    FIXLOC
          TS      X1
          TCF      INTPRET

TDISPSET  CA      TTF/8
          EXTEND
          MP      TSCALINV
          DXCH     TTFDISP

          CA      EBANK5          # TREDES BECOMES ZERO TWO PASSES
          TS      EBANK          # BEFORE TCGFAPPR IS REACHED
          EBANK=   TCGFAPPR
          CA      TCGFAPPR

```



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INCR BBANK  
INCR BBANK  
EBANK= TTF/8

346 (Page LM0826 346)≡

(296b 746)

```

AD      TTF/8
EXTEND
MP      TREDESCL
AD      -DEC103
AD      NEGMAX
TS      L
CS      L
AD      L
AD      +DEC99
AD      POSMAX
TS      TREDES
CS      TREDES
ADS     TREDES
TC      Q

1406P00 TC      P00D00
OCT     01406
1406ALM TC      ALARM
OCT     01406
TCF     RATESTOP

```

```

# *****
# SPECIALIZED "PHASCHNG" SUBROUTINE
# *****

```

```

FASTCHNG EBANK= PHSNAME2
CA      EBANK3      # SPECIALIZED 'PHASCHNG' ROUTINE
XCH     EBANK
DXCH    L
TS      PHSNAME3
LXCH    EBANK
EBANK=  E2DPS
TC      A

```

```

# *****
# PARAMETER TABLE INDIRECT ADDRESSES
# *****

```

```

RDG      =      RBRFG
VDG      =      VBRFG
ADG      =      ABRFG
VDG2TTF  =      VBRFG*
ADG2TTF  =      ABRFG*
JDG2TTF  =      JBRFG*

```

```
# *****
# LUNAR LANDING CONSTANTS
# *****
```

347a  $\langle$ Page LM0827 347a $\rangle \equiv$  (296b 746)

TABLTTFL	ADRES	TABLTTF +3	# ADDRESS FOR REFERENCING TTF TABLE
TTFSCALE	=	BIT12	
TSCALINV	=	BIT4	
-DEC103	DEC	-103	
+DEC99	DEC	+99	
TREDESCL	DEC	-.08	
180DEGS	DEC	+180	
1/2DEG	DEC	+.00278	
PROJMAX	DEC	.42262 B-3	# SIN(25')/8 TO COMPARE WITH PROJ
PROJMIN	DEC	.25882 B-3	# SIN(15')/8 TO COMPARE WITH PROJ
V06N63	VN	0663	# P63
V06N64	VN	0664	# P64
V06N60	VN	0660	# P65, P66, P67
	BANK	22	
	SETLOC	LANDCNST	
	BANK		
	COUNT*	\$\$/F2DPS	
HIGHESTF	2DEC	4.34546769 B-12	
GSCALE	2DEC	100 B-11	
3/8DP	2DEC	.375	
3/4DP	2DEC	.750	
DEPRCRIT	2DEC	-.02 B-1	

347b  $\langle$ Page LM0828 347b $\rangle \equiv$  (296b 746)

```
# *****
# *****
```

## 1.21 p70-p71 routines

$$\begin{aligned}
 348 \quad \langle p70-p71 \text{ routines } 348 \rangle &\equiv & (7) \\
 &\langle \text{Page } LM0829 \text{ } 349 \rangle \\
 &\langle \text{Page } LM0830 \text{ } 351 \rangle \\
 &\langle \text{Page } LM0831 \text{ } 353 \rangle \\
 &\langle \text{Page } LM0832 \text{ } 355 \rangle \\
 &\langle \text{Page } LM0833 \text{ } 357 \rangle \\
 &\langle \text{Page } LM0834 \text{ } 359 \rangle \\
 &\langle \text{Page } LM0835 \text{ } 361 \rangle \\
 &\langle \text{Page } LM0836 \text{ } 363 \rangle \\
 &\langle \text{Page } LM0837 \text{ } 364a \rangle
 \end{aligned}$$

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349 (Page LM0829 349)≡

(348 754)

```
BANK      21
SETLOC    R11
BANK

EBANK=    DVCNTR
COUNT*   $$/R11

R10,R11    CS      FLAGWRD7      # IS SERVICER STILL RUNNING?
           MASK     AVEGFBIT
           CCS      A
           TCF      TASKOVER      # LET AVGEN D TAKE CARE OF GROUP 2.
           CCS      PIPCTR
           TCF      +2
           TCF      LRHTASK        # LAST PASS. CALL LRHTASK.
+2         TS      PIPCTR1

PIPCTR1    =        LADQSAVE
PIPCTR     =        PHSPRDT2
           CAF      OCT31
           TC       TWIDDLE
           ADRES    R10,R11
R10,R11A   CS      IMODES33      # IF LAMP TEST, DO NTO CHANGE LR LITES
           MASK     BIT1
           EXTEND
           BZF      10,11

FLASHH?    MASK     FLGWRD11      # C(A) = 1 - HFLASH BIT
           EXTEND
           BZF      FLASHV?      # H FLASH OFF, SO LEAVE ALONE

           CA       HLITE
           TS       L
           TC       FLIP          # FLIP H LITE

FLASHV?    CA       VFLSHBIT      # VLASHBIT MUST BE BIT 2.
           MASK     FLGWRD11
           EXTEND
           BZF      10,11        # V FLASH OFF

           CA       VLITE
           TS       L
           TC       FLIP          # FLIP V LITE

10,11      CA       FLAGWRD9      # IS THE LETABORT FLAG SET ?
           MASK     LETABBIT
```

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EXTEND

BZF LANDISP

# NO. PROCEED TO R10.

P71NOW?

CS MODREG

# YES. ARE WE IN P71 NOW?

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```
351  (Page LM0830 351)≡ (348 754)
      AD      1DEC71
      EXTEND
      BZF     LANDISP      # YES.  PROCEED TO R10.

      EXTEND      # NO.  IS AN ABORT STAGE COMMANDED?
      READ      CHAN30
      COM
      TS        L
      MASK      BIT4
      CCS       A
      TCF       P71A      # YES.

P7ONOW?      CS      MODREG      # NO.  ARE WE IN P70 NOW?
      AD      1DEC70
      EXTEND
      BZF     LANDISP      # YES.  PROCEED TO R10.

      CA      L      # NO.  IS AN ABORT COMMANDED?
      MASK    BIT1
      CCS     A
      TCF     P70A      # YES.
      TCF     LANDISP   # NO.  PROCEED TO R10.

      COUNT*  $$/P70

P70      TC      LEGAL?
P70A     CS      ZERO
      TCF      +3
P71      TC      LEGAL?
P71A     CAF     TWO
      +3      TS      Q
      INHINT
      EXTEND
      DCA      CNTABTAD
      DTCB

      EBANK=   DVCNTR
CNTABTAD  2CADR  CONTABRT

1DEC70    DEC    70
1DEC71    DEC    71

      BANK     05
      SETLOC   ABORTS1
      BANK
```

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COUNT\* \$\$/P70

CONTABRT CAF ABRTJADR  
TS BRUPT  
RESUME



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353 (Page LM0831 353)≡

(348 754)

ABRTJADR	TCF	ABRTJASK	
ABRTJASK	CAF	OCTAL27	
	AD	Q	
	TS	L	
	COM		
	DXCH	-PHASE4	
	INDEX	Q	
	CAF	MODE70	
	TS	MODREG	
	TS	DISPDEX	# INSURE DISPDEX IS POSITIVE.
	CCS	Q	# SET APSFLAG IF P71.
	CS	FLGWRD10	# SET APSFLAG PRIOR TO THE ENEMA.
	MASK	APSFLBIT	
	ADS	FLGWRD10	
	CS	DAPBITS	# DAPBITS = OCT 640 = BITS 6, 8, 9
	MASK	DAPBOOLS	# (TURN OFF ULLAGE, DRIFT, AND KOVINHIB
	TS	DAPBOOLS	
	CS	FLAGWRD5	# SET ENGONFLG.
	MASK	ENGONBIT	
	ADS	FLAGWRD5	
	CS	PRI030	# INSURE THAT THE ENGINE IS ON, IF ARMED.
	EXTEND		
	RAND	DSALMOUT	
	AD	BIT13	
	EXTEND		
	WRITE	DSALMOUT	
	CAF	LRBYBIT	# TERMINATE R12.
	TS	FLGWRD11	
	CS	FLAGWRD0	# SET R10FLAG TO SUPPRESS OUTPUTS TO THE
	MASK	R10FLBIT	# CROSS-POINTER DISPLAY.
	ADS	FLAGWRD0	# THE FOLLOWING ENEMA WILL REMOVE THE
			# DISPLAY INERTIAL DATA OUTBIT.
	TC	CLRADMOD	# INSURE RADMODES PROPERLY SET FOR R29.
	EXTEND		# LOAD TEVENT FOR THE DOWNLINK.
	DCA	TIME2	
	DXCH	TEVENT	

EXTEND  
DCA SVEXITAD  
DXCH AVGEXIT

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355 (Page LM0832 355)≡

(348 754)

EXTEND  
DCA NEG0  
DXCH -PHASE1

EXTEND  
DCA NEG0  
DXCH -PHASE3

EXTEND  
DCA NEG0  
DXCH -PHASE6

CAF THREE # SET UP 4.3SPOT FOR GOABORT  
TS L  
COM  
DXCH -PHASE4

# the 3 in OCT37774 could be something else, garbled

CAF OCT37774 # SET T5RUPT TO CALL DAPIDLER IN  
TS TIME5 # 40 MILLISECONDS.

TC POSTJUMP  
CADR ENEMA

SVEXITAD EBANK= DVCNTR  
2CADR SERVEXIT

MODE70 DEC 70  
OCTAL27 OCT 27  
MODE71 DEC 71

DAPBITS OCT 00640

BANK 32  
SETLOC ABORTS  
BANK

COUNT\* \$\$/P70

GOABORT TC INTPRET  
CALL INITCDUW  
EXIT  
CAF FOUR  
TS DVCNTR

CAF	WHICHADR
TS	WHICH
TC	DOWNFLAG
ADRES	FLRCS

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357 (Page LM0833 357)≡

(348 754)

	TC	DOWNFLAG	
	ADRES	FLUNDISP	
	TC	DOWNFLAG	
	ADRES	IDLEFLAG	
	TC	UPFLAG	# INSURE 4-JET TRANSLATION CAPABILITY.
	ADRES	ACC4-2FL	
70DEC	TC	CHECKMM	
	DEC	70	
	TCF	P71RET	
P70INIT	TC	INTPRET	
	CALL		
		TGOCOMP	
	DLOAD	SL	
		MDOTDPS	
		4D	
	BDDV		
		MASS	
	STODL	TBUP	
		MASS	
	DDV	SR1	
		K(1/DV)	
	STORE	1/DV1	
	STORE	1/DV2	
	STORE	1/DV3	
	BDDV		
		K(AT)	
	STODL	AT	
		DTDECAY	
	DCOMP	SL	
		11D	
	STORE	TTO	
	SLOAD	DCOMP	
		DPSVEX	
	SR2		
	STORE	VE	# INITIALIZE DPS EXHAUST VELOCITY
	SET	CALL	
		FLAP	
		COMMINIT	
	AXC,1	GOTO	# RETURN HERE IN P70, SET X1 FOR DPS COEFF.
		OD	
		BOTHPOLY	

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INJTARG	AXC,1	8D	# RETURN HERE IN P71, SET X1 FOR APS COEFF.
BOTHPOLY	DLOAD*	DMP	# TGO D
		ABTCOF,1	
		TGO	

359 (Page LM0834 359)≡

(348 754)

```

DAD*    DMP
        ABTCOF +2,1    #      TGO(C+TGO )
        TGO
DAD*    DMP
        ABTCOF +4,1    # TGO(B+TGO d))
        TGO
DAD*    ABTCOF +6,1    # A+TGO(B+TGO(C+TGO D))
STORE   ZDOTD          # STORE TENTATIVELY IN ZDOTD
DSU     BPL            # CHECK AGAINST MINIMUM
        VMIN
        UPRATE        # IF BIG ENOUGH, LEAVE ZDOTD AS IS .
DLOAD
        VMIN
STORE   ZDOTD          # IF TOO SMALL, REPLCAE WITH MINIMUM.
DLOAD
        ABTRDOT
STCALL  RDOTD          # INITIALZE RDOTD.
        YCOMP         # COMPUTE Y
ABS     DSU
        YLIM          # /Y/-DYMAX
BMN     SIGN          # IF <0, XR<.5DEG, LEAVE YCO AT 0
        YOK           # IF >0, FIX SIGN OF DEFICIT.  THIS IS YCO.
        Y
STORE   YCO
DLOAD  DSU
        YCO
        Y             # COMPUTE XRANGE IN CASE ASTRONAUT WANTS
SR
        5D
STORE   XRANGE        # TO LOOK.
UPTHROT SET          EXIT
        FLVR
        TC            UPFLAG          # SET ROTFLAG
        ADRES         ROTFLAG
        TC            THROTUP
        TC            PHASCHNG
        OCT           04024
        TC            BANKCALL        # VERIFY THAT THE PANEL SWITCHES
        CADR          P40AUTO         # ARE PROPERLY SET.

```

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	TC	THROTUP	
UPTHROT1	EXTEND		# SET SERVICER TO CALL ASCENT GUIDANCE.
	DCA	ATMAGAD	
	DXCH	AVGEXIT	



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```
361  (Page LM0835 361)≡ (348 754)
      GRP4OFF          TC      PHASCHNG      # TERMINATE USE OF GROUP 4.
                        OCT      00004
                        TCF      ENDOFJOB
      P71RET          TC      DOWNFLAG
                        ADRES    LETABORT
                        CAF      THRESH2      # SET DVMON THRESHOLD TO THE ASCENT VALUE.
                        TS      DVTHRUSH
                        TC      INTPRET
                        BON      CALL
                        FLAP
                        OLDTIME
                        TGOCOMP      # IF FLAP=0, TGO=T-TIG
      SSP      GOTO
                        QPRET
      CADR      INJTARG
                        P12INIT      # WILL EXIT P12INIT TO INJTARG
      OLDTIME      DLOAD      SL1      # IF FLAP=1,GTO=2 TGO
                        TGO
      STCALL      TG01
                        P12INIT
      EXIT
      TC      PHASCHNG
      OCT      04024
      EXTEND
      DCA      TG01
      DXCH      TGO
      TCF      UPTHROT1 -3
      TG01          =      VGBODY
      # *****
      BANK      21
      SETLOC    R11
      BANK
      COUNT*    $$/P70
      LEGAL?    CS      MMNUMBER      # IS THE DESIRED PGM ALREADY IN PROGRESS?
                AD      MODREG
                EXTEND
                BZF      ABORTALM
```

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CS	FLAGWRD9	# ARE THE ABORTS ENABLED?
MASK	LETABBIT	
CCS	A	

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363 (Page LM0836 363)≡

(348 754)

```
TCF      ABORTALM

CA      FLAGWRD7      # IS SERVICER ON THE AIR?
MASK    AVEGFBIT
CCS     A
TC      Q      # YES. ALL IS WELL.
ABORTALM TC      FALTON
TC      RELDSP
TC      POSTJUMP
CADR    PINBRNCH

BANK    32
SETLOC  ABORTS
BANK

COUNT* $$/P70
```

# \*\*\*\*\*

```
TGOCOMP  RTB      DSU
          LOADTIME
          TIG
SL
          11D
STORE    TGO
RVQ
```

# \*\*\*\*\*

```
THROTUP  CAF      BIT13
          TS       THRUST
          CAF      BIT4
          EXTEND
          WOR      CHAN14
          TC       Q
```

# \*\*\*\*\*

```
10SECS   2DEC     1000
HINJECT  2DEC     18288 B-24      # 60,000 FEET EXPRESSED IN METERS.
(TGO)A   2DEC     37000 B-17
K(AT)    2DEC     .02      # SCALING CONSTANT
WHICHADR REMADR   ABRTABLE
```

# \*\*\*\*\*

364a  $\langle \text{Page LM0837 364a} \rangle \equiv$  (348 754)

	EBANK=	DVCNTR
ATMAGAD	2CADR	ATMAG
ORBMANAD	ADRES	ORBMANUV

## 1.22 p12 routine

364b  $\langle p12 \text{ routine 364b} \rangle \equiv$  (7)

$\langle \text{Page LM0838 365} \rangle$   
 $\langle \text{Page LM0839 367} \rangle$   
 $\langle \text{Page LM0840 369} \rangle$   
 $\langle \text{Page LM0841 371} \rangle$   
 $\langle \text{Page LM0842 372a} \rangle$

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365 (Page LM0838 365)≡

(364b 749)

	BANK	24	
	SETLOC	P12	
	BANK		
	EBANK=	DVCNTR	
	COUNT*	\$\$/P12	
P12LM	TC	PHASCHNG	
	OCT	04024	
	TC	BANKCALL	
	CADR	R02BOTH	# CHECK THE STATUS OF THE IMU.
	TC	UPFLAG	
	ADRES	MUNFLAG	
	TC	UPFLAG	# INSURE 4-JET TRANSLATION CAPABILITY.
	ADRES	ACC4-2FL	
	TC	UPFLAG	# PREVENT R10 FROM ISSUING CROSS-POINTER
	ADRES	R10FLAG	# OUTPUTS.
	TC	CLRADMOD	# INITIALIZE RADMODES FOR R29.
	TC	DOWNFLAG	# CLEAR RENDEZVOUS FLAG FOR P22
	ADRES	RNDVZFLG	
	CAF	THRESH2	# INITIALIZE DVMON
	TS	DVTHRUSH	
	CAF	FOUR	
	TS	DVCNTR	
	CA	ZERO	
	TS	TRKMKCNT	# SHOW THAT R29 DOWNLINK DATA IS NOT READY.
	CAF	V06N33A	
	TC	BANKCALL	# FLASH TIG
	CADR	GOFASH	
	TCF	GOTOP00H	
	TCF	+2	# PROCEED
	TCF	-5	# ENTER
	TC	PHASCHNG	
	OCT	04024	
	TC	INTPRET	

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```
CALL          # INITIALZE WM AND /LAND/
              GUIDINIT
SET           CALL
              FLPI
              P12INIT
```

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```
367      (Page LM0839 367)≡                                     (364b 749)
      P12LMB      DLOAD      (TGO)A      # SET TGO TO AN INITIAL NOMINAL VALUE.
      STODL      TGO
      TIG
      STCALL      TDEC1
      LEMPREC      # ROTATE THE STATE VECTORS TO THE
      VLOAD      MXV      # IGNITION TIME.
      VATT
      REFSMMAT
      VSL1
      STOVL      V1S      # COMPUTE V1S = VEL(TIG)*2(-7) M/CS.
      RATT
      MXV      VSL6
      REFSMMAT
      STCALL      R      # COMPUTE R = POS(TIG)*2(-24) M.
      MUNGRAV      # COMPUTE GDT1/2(TIG)*2(-T)M/CS.
      VLOAD      UNIT
      R
      STCALL      UNIT/R/      # COMPUTE UNIT/R/ FOR YCOMP.
      YCOMP
      SR      DCOMP
      5D
      STODL      XRANGE      # INITIALIZE XRANGE FOR NOUN 76
      VINJNOM
      STODL      ZDOTD
      RDOTDNOM
      STORE      RDOTD
      EXIT
      TC      PHASCHNG
      OCT      04024
      NEWLOAD      CAF      V06N76      # FLASH CROSS-RANGE, AND APOLUNE VALUES.
      TC      BANKCALL
      CADR      GOFLASH
      TCF      GOTOP00H
      TCF      +2      # PROCEED
      TCF      NEWLOAD      # ENTER NEW DATA.
      CAF      P12ADRES
      TS      WHICH
      TC      PHASCHNG
      OCT      04024
```

TC	INTPRET
DLOAD	SL
	XRANGE
	5D
DAD	



369 (Page LM0840 369)≡

(364b 749)

```

                                Y
                                YCO
                                UNIT/R/
                                VAD
                                49FPS
                                V1S
                                STORE V          # V(TIPOVER) = V(IGN) + 57FPS (UNIT/R/)
                                DOT   SL1
                                UNIT/R/
                                STOVL RDOT        # RDOT = 2(-7)
                                UNIT/R/
                                VXV   UNIT
                                QAXIS
                                STORE ZAXIS1
                                SETGO
                                FLVR
                                ASCENT
P12RET  DLOAD
                                ATP          # ATP(2)*2(18)
                                DSQ   PDDL
                                ATY          # ATY(2)*2(18)
                                DSQ   DAD
                                BZE   SQRT
                                YAWDUN
                                SL1   BDDV
                                ATY
                                ARCSIN
YAWDUN  STOVL YAW
                                UNFC/2
                                UNIT  DOT
                                UNIT/R/
                                SL1   ARCCOS
                                DCOMP
                                STORE PITCH
                                EXIT
                                TC    PHASCHNG
                                OCT   04024
                                TC    DOWNFLAG
                                ADRES FLPI
                                INHINT
                                TC    IBNKCALL
                                CADR  PFLITEDB
                                RELINT

```

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TC POSTJUMP  
CADR BURNBABY

P12INIT DLOAD # INITIALIZE ENGINE DATA. USED FOR P12 AND

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371 (Page LM0841 371)≡

(364b 749)

```
(1/DV)A
# P71.
STORE 1/DV3
STORE 1/DV2
STODL 1/DV1
(AT)A
STODL AT
(TBUP)A
STODL TBUP
ATDECAY
DCOMP SL
11D
STORE TTO
SLOAD DCOMP
APSVEX
SR2
STORE VE
BOFF RVQ
FLAP
COMMINIT
DLOAD DAD
HINJECT
/LAND/
STODL RCO
HI6ZEROS
STORE TXO
STORE YCO
STORE RDOTD
STOVL YDOTD
VRECTCSM
VXV MXV
RRECTCSM
REFSMMAT
UNIT
STORE QAXIS
RVQ
P12ADRES REMADR P12TABLE
SETLOC P12A
BANK
COUNT* $$/P12
GUIDINIT STQ SETPD
TEMPR60
OD
VLOAD PUSH
```

		UNITZ	
	RTB	PUSH	
		LOADTIME	
	CALL		
		RP-TO-R	
372a	$\langle \text{Page LM0842 372a} \rangle \equiv$		(364b 749)
	MXV	VXSC	
		REFSMMAT	
		MOONRATE	
	STOVL	WM	
		RLS	
	ABVAL	SL3	
	STCALL	/LAND/	
		TEMPR60	
	49FPS	2DEC .149352 B-6	# EXPECTED RDOT AT TIPOVER
	VINJNOM	2DEC 16.7924 B-7	# 5509.5 FPS(APO=30NM WITH RDOT=19.5FPS)
	RDOTDNOM	2DEC .059436 B-7	# 19.5 FPS

## 1.23 ascent guidance

372b  $\langle \text{ascent guidance 372b} \rangle \equiv$  (7)

$\langle \text{Page LM0843 373} \rangle$   
 $\langle \text{Page LM0844 374} \rangle$   
 $\langle \text{Page LM0845 375} \rangle$   
 $\langle \text{Page LM0846 377} \rangle$   
 $\langle \text{Page LM0847 379} \rangle$   
 $\langle \text{Page LM0848 381} \rangle$   
 $\langle \text{Page LM0849 383} \rangle$   
 $\langle \text{Page LM0850 385} \rangle$   
 $\langle \text{Page LM0851 387} \rangle$   
 $\langle \text{Page LM0852 389} \rangle$   
 $\langle \text{Page LM0853 391} \rangle$   
 $\langle \text{Page LM0854 393} \rangle$   
 $\langle \text{Page LM0855 394} \rangle$   
 $\langle \text{Page LM0856 395} \rangle$

373 (Page LM0843 373)≡ (372b 718)

	BANK	34
	SETLOC	ASCFILT
	BANK	
	EBANK=	DVCNTR
	COUNT*	\$\$/ASENT
ATMAG	TC	PHASCHNG
	OCT	00035
	TC	INTPRET
	BON	
		FLRCS
		ASCENT
	DLOAD	DSU
		ABDVCONV
		MINABDV
	BMN	CLEAR
		ASCTERM4
		SURFFLAG
	CLEAR	SLOAD
		RENDWFLG
		BIT3H
	DDV	EXIT
		ABDVCONV
	DXCH	MPAC
	DXCH	1/DV3
	DXCH	1/DV2
	DXCH	1/DV1
	DXCH	1/DV0
	TC	INTPRET
	DLOAD	DAD
		1/DV0
		1/DV1
	DAD	DAD
		1/DV2
		1/DV3
	DMP	DMP
		VE
		2SEC(9)
	SL3	PDDL
		TBUP
	SR1	DAD
	DSU	
		6SEC(18)

STODL TBUP  
VE  
SR1 DDV  
TBUP  
STCALL AT

374  $\langle \text{Page } LM0844 \text{ } 374 \rangle \equiv$

(372b 718)

BIT3H OCT ASCENT  
4

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375 (Page LM0845 375)≡

(372b 718)

	BANK	30	
	SETLOC	ASENT	
	BANK		
	COUNT*	\$\$/ASENT	
ASCENT	VLOAD	ABVAL	
		R	
	STOVL	/R/MAG	
		ZAXIS1	
	DOT	SL1	
		V	# Z.V = ZDOT*2(-8).
	STOVL	ZDOT	# ZDOT*2(-7)
		ZAXIS1	
	VXV	VSL1	
		UNIT/R/	# Z X UR = LAXIS*2(-2)
	STORE	LAXIS	# LAXIS*2(-1)
	DOT	SL1	
		V	# L.V = YDOT*2(-8).
	STCALL	YDOT	# YDOT * 2(-7)
		YCOMP	
	VLOAD		
		GDT1/2	# LOAD GDT1/2*2(-7) M/CS.
	V/SC	DOT	
		2SEC(18)	
		UNIT/R/	# G.UR*2(9) = GR*2(9).
	PDVL	VXV	# STORE IN PDL(0)
		UNIT/R/	# LOAD UNIT/R/ *2(-1)
		V	# UR*2(-1) X V*2(-7) = H/R*2(-8).
	VSQ	DDV	# H(2)/R(2)*2(-16).
		/R/MAG	# H(2)/R(3)*2(9).
	SL1	DAD	
	STADR		
	STODL	GEFF	# GEFF*2(10)m/CS/CS.
		ZDOTD	
	DSU		
		ZDOT	
	STORE	DZDOT	# DZDOT = (ZDOTD - ZDOT) * 2(7) M/CS.
	VXSC	PDDL	
		ZAXIS1	
		YDOTD	
	DSU		
		YDOT	
	STORE	DYDOT	# DYDOT = (YDOTD - YDOT) *2(7) M/CS.
	VXSC	PDDL	
		LAXIS	

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RDOTD



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```
377  (Page LM0846 377)≡ (372b 718)
DSU
RDOT
STORE DRDOT # DRDOT = (RDOTD - RDOT) * 2(7) M/CS.
VXSC VAD
UNIT/R/
VAD VSL1
STADR
STORE VGVECT # VG = (DRDOT)R + (DVDOT)L + (DZDOT)Z.
DLOAD DMP # LOAD TGO
TGO # TGO GEFF
GEFF
VXSC VSL1
UNIT/R/ # TGO GEFF UR
BVSU
VGVECT # COMPENSATED FOR GEFF
STORE VGVECT # STORE FOR DOWNLINK
MXV VSL1 # GET VGBODY FOR N85 DISPLAY
XNBPIP
STOVL VGBODY
VGVECT
ABVAL BOFF # MAGNITUDE OF VGVECT
FLRCS # IF FLRCS=0,DO NORMAL GUIDANCE
MAINENG
DDV # USE TGO=VG/AT WITH RCS
AT/RCS
STCALL TGO # THIS WILL BE USED ON NEXT CYCLE
ASCTERM2
MAINENG DDV PUSH # VG/VE IN PDL(0) (2)
VE
DMP BDSU # 1 - KT VG/VE
KT1
NEARONE
DMP DMP # TBUP VG(1-KT VG/VE)/VE (0)
TBUP # = TGO
DSU # COMPENSATE FOR TAILOFF
TTO
STORE TGO
SR DCOMP
11D
STODL TTOGO # TGO *2(-28) CS
TGO
BON DSU
IDLEFLAG
T2TEST
4SEC(17) # ( TGO - 4 ) *2(-17) CS.
```

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	BMN	
		ENG OFF
T2TEST	DLOAD	
		TGO
	DSU	BMN
		# IF TGO - T2 NEG., GO TO COMPONENT

379 (Page LM0847 379)≡

(372b 718)

```

                                T2A
                                CMPOENT
                                DLOAD DSU
                                TBUP
                                TGO
                                DDV CALL # 1- TGO/TBUP
                                TBUP
                                LOGSUB
                                SL PUSH # -L IN PDL(0) (2)
                                5
                                BDDV BDSU # -TGO/L*2(-17)
                                TGO
                                TBUP # TBUP + TGO/L = D12*2(-17)
                                PUSH BON # STORE IN PDL(2) (4)
                                FLPC # IF FLPC = 1, GO TO CONST
                                NORATES
                                DLOAD DSU
                                TGO
                                T3
                                BPL SET # FLPC=1
                                RATES
                                FLPC
                                NORATES DLOAD
                                HI6ZEROS
                                STORE PRATE # B = 0
                                STORE YRATE # D = 0
                                GOTO
                                RATES DLOAD DSU
                                TGO
                                O2D # TGO - D12 = D21*2(-17)
                                PUSH SL1 # IN PDL(4) (6)
                                BDSU SL3 # (1/2TGO - D21)*2(-13) = E * 2(-13)
                                TGO # (8)
                                PDDL DMP # IN PDL(6)
                                TGO
                                RDOT # RDOT TGO * 2(-24)
                                DAD DSU # R + RDOT TGO
                                /R/MAG # R + RDOT TGO - RCO
                                RCO # MPAC = -DR *2(-24).
                                PDDL DMP # -DR IN PDL(8) (10)
                                DRDOT
                                O4D # D21 DRDOT*2(-24)
                                DAD SL2 # (D21 DRDOT-DR)*2(-22) (8)
                                DDV DDV

```

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	06D	# (D21 DRDOT-DR)/E*2(-9)
	TGO	
STORE	PRATE	# B * 2(8)
BMN	DLOAD	# B>0 NOT PERMITTED
	CHKBAG	

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381 (Page LM0848 381)≡

(372b 718)

		HI6ZEROS	
	STCALL	PRATE	
		PROK	
CHKBMAG	SR4	DDV	# B*2(4)
		TBUP	# (B / TAU) * 2(21)
	DSU	BPL	
		PRLIMIT	# (B / TAU) = 2(21) MAX.
		PROK	
	DLOAD	DMP	
		PRLIMIT	
		TBUP	# B MAX. * 2(4)
	SL4		# BMAX*2(8)
	STORE	PRATE	
PROK	DLOAD		
		TGO	
	DMP	DAD	# YDOT TGO
		YDOT	
		Y	# Y + YDOT TGO
	DSU	PDDL	# Y + YDOT TGO - YCO
		YCO	# MPAC = - DY*(-24.) IN PDL(8) (10)
		DYDOT	
	DMP	DAD	# D21 DYDOT - DY (8)
		O4D	
	SL2	DDV	# (D21 DYDOT - DY)/E*2(-9)
	DDV	SETPD	# (D21 DYDOT - DY)/E TGO*2(8)
		TGO	# = D*2(8)
		O4	
	STORE	YRATE	
CONST	DLOAD	DMP	# LOAD B*2(8)
		PRATE	# B D12*2(-9)
		O2D	
	PDDL	DDV	# D12 B IN PDL(4) (6)
		DRDOT	# LOAD DRDOT*2(-7)
		O0D	# -DRDOT/L*2(-7)
	SR2	DSU	# (-DRDOT/L-D12 B)=A*2(-9) (4)
	STADR		
	STODL	PCONS	
		YRATE	# D*2(8)
	DMP	PDDL	# D12 D,EXCH WITH -L IN PDL(0) (2,2)
	BDDV	SR2	# -DYDOT/L*2(-9)
		DYDOT	
	DSU		# (-DYDOT/L-D12 D)=C*2(-9)
		O0D	
	STORE	YCONS	
CMPOENT	SETPD	DLOAD	

	OOD	
	100CS	
DMP		
	PRATE	# $B(T-T_0)*2(-9)$
DAD	DDV	# $(A+B(T-T_0))*2(-9)$

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383 (Page LM0849 383)≡

```

PCONS                                     (372b 718)
TBUP                                     # (A+B(T-T0))/TBUP*2(8)
SL1 DSU
GEFF                                     # ATR*2(9)
STODL ATR
100CS
DMP DAD
YRATE
YCONS                                     # (C+D(T-T0))*2(-9)
DDV SL1
TBUP
STORE ATY                                     # ATY*2(9)
VXSC PDDL                                     # ATY UY*2(8) (6)
LAXIS
ATR
VXSC VAD
UNIT/R/
VSL1 PUSH                                     # AH*2(9) IN PDL(0) (6)
ABVAL PDDL                                     # AH(2) IN PDL(34)
AT                                     # AHMAG IN PDL(6) (8)
DSQ DSU                                     # (AT(2)-AH(2))*2(18)
34D                                     # =ATP2*2(18)
PDDL PUSH                                     # (12)
AT
DSQ DSU                                     # (AT(2)KR(2)-AH(2))*2(18) (10)
34D                                     # =ATP3*2(18)
BMN DLOAD                                     # IF ATP3 NEG,GO TO NO-ATP
NO-ATP NO-ATP
8D
SQRT GOTO                                     # ATP*2(9)
AIMER
DLOAD BDDV                                     # KR AT/AH = KH (8)
6D
VXSC                                     # KH AG*2(9)
00D
STODL 00D                                     # STORE NEW AH IN PDL(0)
HI6ZEROS
AIMER SIGN
DZDOT
STORE ATP
VXSC
ZAXIS1                                     # ATP ZAXIS *2(8).
VSL1 VAD                                     # AT*2(0)
00D
STORE UNFC/2                               # WILL BE OVERWRITTEN IF IN VERT. RISE.

```

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SETPD BON  
OOD  
FLPI  
P12RET  
BON



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385 (Page LM0850 385)≡

(372b 718)

		FLVR	
		CHECKALT	
MAINLINE	VLOAD	VCOMP	
		UNIT/R/	
	STODL	UNWC/2	
		TXO	
	DSU	BPL	
		PIPTIME	
		ASCTERM	
	BON		
		ROTFLAG	
		ANG1CHEK	
CLRFLAG	CLEAR	CLEAR	
		NOR29FLG	# START r29 IN ASCENT PHASE.
		XOVINFLG	# ALLOW X-AXIS OVERRIDE
ASCTERM	EXIT		
	CA	FLAGWRD9	
	MASK	FLRCSBIT	
	CCS	A	
	TCF	ASCTERM3	
	TC	INTPRET	
	CALL		
		FINDCDUW -2	
ASCTERM1	EXIT		
+1	CA	FLAGWRD9	# INSURE THAT THE NOUN 63 DISPLAY IS
	MASK	FLRCSBIT	# BYPASSED IF WE ARE IN THE RCS TRIMMING
	CCS	A	# MODE OF OPERATION
	TCF	ASCTERM3	
	CA	FLAGWRD8	# BYPASS DISPLAYS IF ENGINE FAILURE IS
	MASK	FLUNDBIT	# INDICATED.
	CCS	A	
	TCF	ASCTERM3	
	CAF	V06N63*	
	TC	BANKCALL	
	CADR	GODSPR	
	TCF	ASCTERM3	
ASCTERM2	EXIT		
ASCTERM3	TCF	ENDOFJOB	
ASCTERM4	EXIT		
	INHINT		
	TC	IBNKCALL	# NO GUIDANCE THIS CYCLE -- HENCE ZERO
	CADR	ZATTEROR	# THE DAP COMMANDED ERRORS.
	TCF	ASCTERM1 +1	
CHECKALT	DLOAD	DSU	

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	/R/MAG	
	/LAND/	
DSU	BMN	# IF H LT 25K CHECK Z AXIS ORIENTATION
	25KFT	
	CHECKYAW	

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387 (Page LM0851 387)≡ (372b 718)

EXITVR	CLEAR	BON FLVR ROTFLAG MAINLINE
	DLOAD	DAD PIPTIME 10SECS
	STCALL	TXO MAINLINE
EXITVR1	CLRGO	ROTFLAG EXITVR
	SETLOC	ASENT1
	BANK	
	COUNT*	\$\$/ASENT
ANG1CHEK	VLOAD	DOT UNFC/2 XNBPIP
	DSU	BPL COSTHET1 OFFROT
	VLOAD	DOT XNBPIP UNIT/R/
	DSU	BMN COSTHET2 KEEPVR1
OFFROT	CLRGO	ROTFLAG CLRFLAG
	BANK	7
	SETLOC	ASENT2
	BANK	
	COUNT*	\$\$/ASENT
SETXFLAG	=	CHECKYAW
CHECKYAW	SET	
	DLOAD	XOVINFLG VXSC ATY LAXIS

# PROHIBIT X-AXIS OVERRIDE

PDDL	VXSC
	ATP
	ZAXIS1
VAD	UNIT
PUSH	DOT

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389 (Page LM0852 389)≡

(372b 718)

```

                                YNBPIP
                                ABS    DSU
                                SIN5DEG
                                BPL    DLOAD
                                KEEPVR
                                RDOT
                                DSU    BPL
                                40FPS
                                EXITVR1
                                GOTO
                                KEEPVR

                                BANK    5
                                SETLOC  ASENT3
                                BANK
                                COUNT*  $$/ASENT

SIN5DEG    2DEC    0.08716 B-2
40FPS      2DEC    0.12192 B-7

                                BANK    14
                                SETLOC  ASENT4
                                BANK
                                COUNT*  $$/ASENT

KEEPVR      VLOAD   STADR              # RECALL LOSVEC FROM PUSHLIST
STORE      UNWC/2
KEEPVR1     VLOAD
                                UNIT/R/
STCALL     UNFC/2
                                ASCTERM

ENGOFF      RTB
                                LOADTIME
                                DSU    DAD
                                PIPTIME
                                TTOGO
                                DCOMP  EXIT
                                TC      TPAGREE      # FORCE SIGN AGREEMENT ON MPAC, MPAC +1.
                                CAF      EBANK7
                                TS       EBANK
                                EBANK=  TGO
                                INHINT
                                CCS      MPAC +1
                                TCF      +3          # C(A) = DT - 1 BIT
```

TCF	+2	# C(A) = 0
CAF	ZERO	# C(A) = 0
AD	BIT1	# C(A) = 1 BIT OR DT.

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391 (Page LM0853 391)≡

(372b 718)

	TS	ENGOFFDT	
	TC	TWIDDLE	
	ADRES	ENGOFF1	
	TC	PHASCHNG	
	OCT	47014	
	-GENADR	ENGOFFDT	
	EBANK=	TGO	
	2CADR	ENGOFF1	
	TC	INTPRET	
	SET	GOTO	
		IDLEFLAG	# DISABLE DELTA-V MONITOR
		T2TEST	
ENG OFF1	TC	IBNKCALL	# SHUT OFF THE ENGINE.
	CADR	ENGNOF2	
	CAF	PRI017	# SET UP A JOB FOR THE ASCENT GUIDANCE
	TC	FINDVAC	# POSTBURN LOGIC.
	EBANK=	WHICH	
	2CADR	CUTOFF	
	TC	PHASCHNG	
	OCT	07024	
	OCT	17000	
	EBANK=	TGO	
	2CADR	CUTOFF	
	TCF	TASKOVER	
CUTOFF	TC	UPFLAG	# SET FLRCS FLAG.
	ADRES	FLRCS	
-5	CAF	V16N63	
	TC	BANKCALL	
	CADR	GOFLASH	
	TCF	+3	
	TCF	CUTOFF1	
	TCF	-5	
+3	TC	POSTJUMP	
	CADR	TERMASC	
CUTOFF1	INHINT		
	TC	IBNKCALL	# ZERO ATTITUDE ERRORS BEFORE REDUCINT DB.

CADR	ZATTEROR
TC	IBNKCALL
CADR	SETMINDB
TC	POSTJUMP
CADR	CUTOFF2



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393 (Page LM0854 393)≡

(372b 718)

V16N63 VN 1663  
BANK 30  
SETLOC ASENT5  
BANK  
COUNT\* \$\$/ASENT

CUTOFF2 TC PHASCHNG  
OCT 04024

CAF V16N85C  
TC BANKCALL  
CADR GOFLASH  
TCF TERMASC  
TCF +2  
TCF CUTOFF2

# PROCEED

TERMASC TC PHASCHNG  
OCT 04024

INHINT  
TC IBNKCALL  
CADR RESTORDB  
TC DOWNFLAG  
ADRES LETABORT  
TCF GOTOP00H

# RESTORE DEADBAND DESIRED BY ASTRONAUT.

# DISALLOW ABORTS AT THIS TIME.

V16N85C VN 1685  
  
BANK 27  
SETLOC ASENT1  
BANK  
COUNT\* \$\$/ASENT

YCOMP VLOAD DOT  
UNIT/R/  
QAXIS  
SL2 DMP  
RCO  
STORE Y  
RVQ

BANK 30  
SETLOC ASENT  
BANK

394  $\langle$ Page LM0855 394 $\rangle \equiv$ 

(372b 718)

100CS	EQUALS	2SEC(18)
T2A	EQUALS	2SEC(17)
4SEC(17)	2DEC	400 B-17
2SEC(17)	2DEC	200 B-17
T3	2DEC	1000 B-17
6SEC(18)	2DEC	600 B-18
BIT4H	OCT	10
2SEC(9)	2DEC	200 B-9
V06N63*	VN	0663
V06N76	VN	0676
V06N33A	VN	0633

BANK	33
SETLOC	ASENT6
BANK	
COUNT*	\$\$/ASENT

KT1	2DEC	0.5000
PRLIMIT	2DEC	-.0639
MINABDV	2DEC	.0356 B-5
1/DVO	=	MASS1

# (B/TBUP)MIN=-.1FT.SEC(-3)

# 10 PERCENT BIGGER THAN GRAVITY

395       $\langle \text{Page } LM0856 \text{ 395} \rangle \equiv$  (372b 718)

# THE LOGARITHM SUBROUTINE

BANK      24  
SETLOC   FLOGSUB  
BANK

# INPUT ..... X IN MPAC  
# OUTPUT ..... -LOG(X) IN MPAC

LOGSUB      NORM      BDSU  
                         MPAC +6  
                         NEARONE  
  
EXIT  
TC          POLY  
DEC          6  
2DEC        .0000000060  
2DEC        -.0312514377  
2DEC        -.0155686771  
2DEC        -.0112502068  
2DEC        -.0018545108  
2DEC        -.0286607906  
2DEC        .0385598563  
2DEC        -.0419361902

CAF          ZERO  
TS          MPAC +2  
EXTEND  
DCA          CLOG2/32  
DXCH        MPAC  
DXCH        BUF +1  
CA          MPAC +6  
TC          SHORTMP  
DXCH        MPAC +1  
DXCH        MPAC  
DXCH        BUF +1  
DAS        MPAC  
TC          INTERPRET  
DCOMP       RVQ

CLOG2/32      2DEC      .0216608494

## 1.24 servicer routine

396  $\langle \text{servicer routine 396} \rangle \equiv$  (7)

$\langle \text{Page LM0857 397} \rangle$   
 $\langle \text{Page LM0858 398} \rangle$   
 $\langle \text{Page LM0859 400} \rangle$   
 $\langle \text{Page LM0860 401} \rangle$   
 $\langle \text{Page LM0861 403} \rangle$   
 $\langle \text{Page LM0862 405} \rangle$   
 $\langle \text{Page LM0863 407} \rangle$   
 $\langle \text{Page LM0864 408} \rangle$   
 $\langle \text{Page LM0865 409} \rangle$   
 $\langle \text{Page LM0866 410} \rangle$   
 $\langle \text{Page LM0867 411} \rangle$   
 $\langle \text{Page LM0868 412a} \rangle$   
 $\langle \text{Page LM0869 412b} \rangle$   
 $\langle \text{Page LM0870 413} \rangle$   
 $\langle \text{Page LM0871 415} \rangle$   
 $\langle \text{Page LM0872 416} \rangle$   
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 $\langle \text{Page LM0874 418} \rangle$   
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 $\langle \text{Page LM0876 420} \rangle$   
 $\langle \text{Page LM0877 421} \rangle$   
 $\langle \text{Page LM0878 422} \rangle$   
 $\langle \text{Page LM0879 423} \rangle$   
 $\langle \text{Page LM0880 424} \rangle$   
 $\langle \text{Page LM0881 425} \rangle$   
 $\langle \text{Page LM0882 426} \rangle$   
 $\langle \text{Page LM0883 428} \rangle$   
 $\langle \text{Page LM0884 429} \rangle$   
 $\langle \text{Page LM0885 431} \rangle$   
 $\langle \text{Page LM0886 433} \rangle$   
 $\langle \text{Page LM0887 435} \rangle$   
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 $\langle \text{Page LM0890 441} \rangle$   
 $\langle \text{Page LM0891 442} \rangle$   
 $\langle \text{Page LM0892 443} \rangle$   
 $\langle \text{Page LM0893 445} \rangle$   
 $\langle \text{Page LM0894 447} \rangle$   
 $\langle \text{Page LM0895 449} \rangle$   
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 $\langle \text{Page LM0897 452a} \rangle$

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397 (Page LM0857 397)≡

(396 765)

BANK 37  
SETLOC SERV1  
BANK

EBANK= DVCNTR

# \*\*\*\*\* PREREAD \*\*\*\*\*

COUNT\* \$\$/SERV

PREREAD	CAF	SEVEN	# 5.7 SPOT TO SKIP LASTBIAS AFTER
	TC	GNUFAZE5	# RESTART.
	CAF	PRI021	
	TC	NOVAC	
	EBANK=	NBDX	
	2CADR	LASTBIAS	# DO LAST GYRO COMPENSATION IN FREE FALL
BIBIBIAS	TC	PIPASR +3	# CLEAR + READ PIPS LAST TIME IN FRE5+F133
			# DO NOT DESTROY VALUE OF PIPTIME1
	CS	FLAGWRD7	
	MASK	SUPER011	# SET V37FLAG AND AVEGFLAG (BITS 5 AND 6
	ADS	FLAGWRD7	# OF FLAGWRD7)
	CS	DRFTBIT	
	MASK	FLAGWRD2	# RESET DRIFTFLAG
	TS	FLAGWRD2	
	CAF	FOUR	# INITIALIZE DV MONITOR
	TS	PIPAGE	
	CAF	ENDJBCAD	# POINT OUTROUTE TO END-OF-JOB.
	TS	OUTROUTE	
	CAF	PRI022	
	TC	FINDVAC	# TO FIRST ENTRY TO AVERAGE G
	EBANK=	DVCNTR	
	2CADR	NORMLIZE	
GOREADAX	CA	TWO	# 5.2SPOT FOR REREADAC AND NORMLIZE
	TC	GNUTFAZ5	
	CA	2SECS	# WAIT TWO SECONDS FOR READACCS
	TC	VARDELAY	

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# \*\*\*\*\* READACCS \*\*\*\*\*

READACCS	CS	OCT37771	# THIS PIECE OF CODING ATTEMPTS TO
	AD	TIME5	# SYNCHRONIZE READACCS WITH THE DIGITAL
	CCS	A	# AUTOPILOT SO THAT A PAXIS RUPT WILL
	CS	ONE	# OCCUR APPROXIMATELY 70 MILLISECONDS
	TCF	+2	# FOLLOWING THE READACCS RUPT. THE 70 MS
	CA	ONE	# OFFSET WAS CHOSEN SO THAT THE PAXIS
+2	ADS	TIME5	# RUPT WOULD NOT OCCUR SIMULTANEOUSLY
			# WITH ANY OF THE 8 SUBSEQUENT R10,R11
			# INTERRUPTS -- THUS MINIMIZING THE POSS-
			# IBILITY OF LOSING DOWNRUPTS.
	TC	PIPASR	# READ THE PIPAS.
PIPSDONE	CA	FIVE	
	TC	GNUFAZE5	
RED05.5	CAF	ONE	
	TS	PIPAGE	
	CA	PRI020	
	TC	FINDVAC	
	EBANK=	DVCNTR	
	2CADR	SERVICER	# SET UP SERVICER JOB
	CA	BIT9	
	EXTEND		
	WOR	DSALMOUT	# TURN ON TEST CONNECTOR OUTBIT
	CA	FLAGWRD7	
	MASK	AVEGFBIT	
	EXTEND		
	BZF	AVEGOUT	# AVEGFLAG DOWN -- SET UP FINAL EXIT
	CA	FLAGWRD6	
	MASK	MUNFLBIT	
	EXTEND		
	BZF	MAKEACCS	# MUNFLAG CLEAR -- BYPASS LR AND DISP.
	CCS	PHASE2	
	TCF	MAKEACCS	# PHASE 2 ACTIVATED -- AVOID MULTIPLE R10.
	CAF	SEVEN	# SET PIPCTR FOR 4X/SEC RATE.
	TS	PIPCTR	

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CS	TIME1	# SET TBASE2 .05 SECONDS IN THE PAST.
AD	FIVE	
AD	NEG1/2	
AD	NEG1/2	
XCH	TBASE2	

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(396 765)

	CAF	DEC17	# 2.21SPOT FOR R10,R11
	TS	L	
	COM		
	DXCH	-PHASE2	
	CAF	OCT24	# FIRST R10,R11 IN .200 SECONDS
	TC	WAITLIST	
	EBANK=	UNIT/R/	
	2CADR	R10,R11	
MAKEACCS	CA	FOUR	
	TCF	GOREADAX	# DO PHASE CHANGE AND RECALL READACCS
AVEGOUT	EXTEND		
	DCA	AVOUTCAD	# SET UP FINAL SERVICER EXIT
	DXCH	AVGEXIT	
	CA	FOUR	# SET 5.4 SPOT FOR REREADAC AND SERVICER
	TC	GNUTFAZ5	# IF REREADAC IS CALLED, IT WILL EXIT
	TC	TASKOVER	# END TASK WITHOUT CALLING READACCS
GNUTFAZ5	TS	L	# SAVE INPUT IN L
	CS	TIME1	
	TS	TBASE5	# SET TBASE5
	TCF	+2	
GNUFAZE5	TS	L	# SAVE INPUT IN L
	CS	L	# -PHASE IN A, PHASE IN L.
	DXCH	-PHASE5	# SET -PHASE5,PHASE5
	TC	Q	
	EBANK=	DVCNTR	
AVOUTCAD	2CADR	AVGEND	
ENDJBCAD	CADR	SERVEXIT +2	
OCT37771	OCT	37771	
	BANK	33	
	SETLOC	SERVICES	
	BANK		
	COUNT*	\$\$/SERV	



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(396 765)

# \*\*\*\*\* SERVICER \*\*\*\*\*

```
SERVICER      TC      PHASCHNG      # RESTART REREADAC + SERVICER
               OCT      16035
               OCT      20000
               EBANK=   DVCNTR
               2CADR    GETABVAL

               CAF      PRI031      # INITIALIZE 1/PIPADT IN CASE RESTART HAS
               TS       1/PIPADT    # CAUSED LASTBIAS TO BE SKIPPED.

               TC       BANKCALL    # PIPA COMPENSATION CALL
               CADR      1/PIPA

GETABVAL      TC       INTPRET
               VLOAD    ABVAL
               DELV

               EXIT
               CA       MPAC
               TS       ABDELV      # ABDELV = CM/SEC*2(-14).
               EXTEND
               MP       KPIP
               DXCH     ABDVCONV    # ABDVCONV = M/CS * 2(-5).
               EXTEND
               DCA      MASS
               DXCH     MASS1      # INITIALIZE MASS1 IN CASE WE SKIP MASSMON
               CS       FLAGWRD8   # ARE WE ON THE SURFACE?
               MASK     SURFFBIT
               EXTEND
               BZF      MOONSPOT    # YES:  BYPASS MASS MESS

               CA       FLGWRD10    # NO:  WHICH VEX SHOULD BE USED?
               MASK     APSFLBIT
               CCS      A
               EXTEND      # IF EXTEND IS EXECUTED, APSVEX --> A,
               DCA      APSVEX      #      OTHERWISE DPSVEX --> A
               TS       Q

               EXTEND
               DCA      ABDVCONV

OCT10002      DV       Q          # WHERE APPROPRIATE VEX RESIDES
               EXTEND
               MP       MASS
               DAS      MASS1
```

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MOONSPOT	CA	KPIP1	# TP MPAC = ABDELV AT 2(14) CM/SEC
	TC	SHORTMP	# MULTIPLY BY KPIP1 TO GET

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```
403  (Page LM0861 403)≡
                                     (396 765)
                                     # ABDELV AT 2(7) M/CS
                                     # UPDATE DVTOTAL FOR DISPLAY

                                     DXCH  MPAC
                                     DAS   DVTOTAL

                                     TC     TMPTOSPT

                                     TC     BANKCALL
                                     CADR    QUICTRIG

                                     CAF     XNBPIPAD
                                     TC     BANKCALL
                                     CADR    FLESHPOT
                                     TC     INTPRET
AVERAGEG  BON     CALL
                                     MUNFLAG
                                     RVBOTH
                                     CALCRVG

                                     EXIT
GOSERV     TC     QUIKFAZ5

COPYCYCL   TC     COPYCYC

#          CA     ZERO          # A IS ZERO ON RETURN FROM COPYCYC
          TS     PIPATMPX
          TS     PIPATMPY
          TS     PIPATMPZ

          CS     STEERBIT      # CLEAR STEERSW PRIOR TO DVMON.
          MASK   FLAGWRD2
          TS     FLAGWRD2

          CAF     IDLEFBIT      # IS THE IDLE FLAG SET?
          MASK   FLAGWRD7
          CCS     A
          TCF     NODVMON1      # IDLEFLAG = 1, HENCE SET AUXFLAG TO 0.

          CS     FLAGWRD6
          MASK   AUXFLBIT
          CCS     A
          TCF     NODVMON2      # AUXFLAG = 0, HENCE SET AUXFLAG TO 1.

DVMON      CS     DVTHRUSH
          AD     ABDELV
          EXTEND
          BZMF    LOTHRUST
```

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	CS	FLAGWRD2	# SET STEERSW.
	MASK	STEERBIT	
	ADS	FLAGWRD2	
DVCNTSET	CAF	ONE	# ALLOW TWO PASSES MAXIMUM NOW THAT

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```
405  (Page LM0862 405)≡
                                     (396 765)
      TS      DVCNTR      # THRUST HAS BEEN DETECTED.

      CA      FLGWRD10    # BRANCH IF APSFLAG IS SET.
      MASK    APSFLBIT
      CCS     A
      TCF     USEJETS

      CA      BIT9        # CHECK GIMBAL FAIL BIT
      EXTEND
      RAND    CHAN32
      EXTEND
      BZF     USEJETS

      USEGTS   CS      USEQRJTS
      MASK    DAPBOOLS
      TS      DAPBOOLS
      TCF     SERVOUT

      NODVMON1 CS      AUXFLBIT      # SET AUXFLAG TO 0.
      MASK    FLAGWRD6
      TS      FLAGWRD6
      TCF     USEJETS

      NODVMON2 CS      FLAGWRD6      # SET AUXFLAG TO 1.
      MASK    AUXFLBIT
      ADS     FLAGWRD6
      TCF     USEJETS

      LOTHRUST TC      QUIKFAZ5
      CCS     DVCNTR
      TCF     DECCNTR

      CCS     PHASE4      # COMFAIL JOB ACTIVE?
      TCF     SERVOUT     # YES:  WON'T NEED ANOTHER.

      TC      PHASCHNG    # 4.37SPOT FOR COMFAIL.
      OCT     00374

      CAF     PRI025
      TC      NOVAC
      EBANK=  WHICH
      2CADR   COMFAIL

      TCF     SERVOUT

      DECCNTR TS      DVCNTR1
```

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TC QUIKFAZ5

CA DVCNTR1

TS DVCNTR

INHINT

TC IBNKCALL

# IF THRUST IS LOW, NO STEERING IS DONE

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	CADR	STOPRATE	# AND THE DESIRED RATES ARE SET TO ZERO.
USEJETS	CS	DAPBOOLS	
	MASK	USEQRJTS	
	ADS	DAPBOOLS	
SERVOUT	RELINT		
	TC	BANKCALL	
	CADR	1/ACCS	
	CA	PRIORITY	
	MASK	LOW9	
	TS	PUSHLOC	
	ZL		
	DXCH	FIXLOC	# FIXLOC AND DVFIND
	TC	QUIKFAZ5	
	EXTEND		# EXIT TO SELECTED ROUTINE WHETHER THERE
	DCA	AVGEXIT	# IS THRUST OR NOT. THE STATE OF STEERSW
	DXCH	Z	# WILL CONVEY THIS INFORMATION.
XNBPIPAD	ECADR	XNBPIP	
	BANK	32	
	SETLOC	SERV2	
	BANK		
	COUNT*	\$\$/SERV	
AVGEND	CA	PIPTIME +1	# FINAL AVERAGE G EXIT
	TS	1/PIPADT	# SET UP FREE FALL GYRO COMPENSATION.
	TC	UPFLAG	# SET DRIFT FLAG.
	ADRES	DRIFTFLG	
	TC	BANKCALL	
	CADR	PIPFREE	
	CS	BIT9	
	EXTEND		
	WAND	DSALMOUT	
	TC	2PHSCHNG	
	OCT	5	# GROUP 5 OFF
	OCT	05022	# GROUP 2 ON
	OCT	20000	
	TC	INTPRET	

SET	CLEAR	
	NOR29FLG	# SHUT OFF R29 WHEN SERVICER ENDS.
	SWANDISP	# SHUT OFF R10 WHEN SERVICER ENDS.
CLEAR	CALL	# RESET MUNFLAG.
	MUNFLAG	

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(396 765)

		AVETOMID	
	CLEAR	EXIT	
		V37FLAG	
AVERTRN	CA	OUTROUTE	# RETURN TO DESIRED POINT.
	TC	BANKJUMP	
OUTGOAVE	=	AVERTRN	
DVCNTR1	=	MASS1	



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(396 765)

```

SETLOC  SERV3
BANK
COUNT*  $$/SERV

SERVIDLE  EXTEND
DCA      SVEXTADR
DXCH     AVGEXIT

CS      FLAGWRD7
MASK    IDLEFBIT
ADS     FLAGWRD7

CAF     LRBYBIT
TS      FLGWRD11

EXTEND
DCA     NEG0
DXCH    -PHASE1

CA      FLAGWRD6
MASK    MUNFLBIT
CCS     A
TCF     +4

EXTEND
DCA     NEG0
DXCH    -PHASE2

+4      EXTEND
DCA     NEG0
DXCH    -PHASE3

EXTEND
DCA     NEG0
DXCH    -PHASE6

CAF     OCT33
TS      L
COM
DXCH    -PHASE4

TCF     WHIMPER

# DISCONNECT SERVICER FROM ALL GUIDANCE

# DISCONNECT THE DELTA-V MONITOR

# TERMINATE R12 IS RUNNING.

# DO NOT TURN OFF PHASE 2 IF MUNFLAG SET.

# 4.33SPOT FOR GOP00FIX

# PERFORM A SOFTWARE RESTART AND PROCEED
# TO GOTOP00H WHILE SERVICER CONTINUES TO
# RUN, ALBEIT IN A GROUND STATE WHERE
# ONLY STATE-VECTOR DEPENDENT FUNCTIONS
```

# ARE MAINTAINED.

	EBANK=	DVCNTR	
410	$\langle$ Page LM0866 410 $\rangle \equiv$		(396 765)
	SVEXTADR	2CADR	SERVEXIT
		BANK	32
		SETLOC	SERV
		BANK	
		COUNT*	\$\$/SERV
	SERVEXIT	TC	PHASCHNG
		OCT	00035
	+2	TCF	ENDOFJOB
		BANK	23
		SETLOC	NORMLIZ
		BANK	
		COUNT*	\$\$/SERV

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```
411  (Page LM0867 411)≡ (396 765)
      NORMLIZE          TC      INTPRET
                        VLOAD    BOFF
                                RN1
                                MUNFLAG
                                NORMLIZ1
                        VSL6     MXV
                                REFSMMAT
                        STCALL   R
                                MUNGRAV
                        VLOAD    VSL1
                                VN1
                        MXV
                                REFSMMAT
                        STOVL    V
                                V(CSM)
                        VXV      UNIT
                                R(CSM)
                        STORE    UHYP
ASCSPOT  EXIT
          EXTEND
          DCA      NEG0
          DXCH     -PHASE2

          TC      POSTJUMP
          CADR     NORMLIZ2

          BANK     33
          SETLOC   SERVICES
          BANK
          COUNT*   $$/SERV

NORMLIZ1  CALL
          CALCGRAV
          EXIT

NORMLIZ2  CA      EIGHTEEN
          TC      COPYCYC +1  # DO NOT COPY MASS IN NORMLIZE
          TC      ENDOFJOB

COPYCYC   CA      OCT24      # DEC 20
          +1     INHINT
          +2     MASK      NEG1  # REDUCE BY 1 IF ODD
                TS        ITEMP1
                EXTEND
                INDEX     ITEMP1
```

PIPASR	EXTEND
--------	--------

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```
413  (Page LM0870 413)≡ (396 765)
      DCA      TIME2
      DXCH     PIPTIME1      # CURRENT TIME POSITIVE VALUE
+3    CS      ZERO          # INITIALIZE THESE AT NEG. ZERO.
      TS      TEMX
      TS      TEMY
      TS      TEMZ

      CA      ZERO
      TS      DELVZ
      TS      DELVZ +1
      TS      DELVY
      TS      DELVY +1
      TS      DELVX +1
      TS      PIPAGE        # SHOW PIPA READING IN PROGRESS

REPIP1  EXTEND
      DCS     PIPAX          # X AND Y PIPS READ
      DXCH    TEMX
      DXCH    PIPAX          # PIPAS SET TO NEG ZERO AS READ.
      TS      DELVX
      LXCH    DELVY

REPIP3  CS      PIPAZ        # REPEAT PROCESS FOR Z PIP
      XCH     TEMZ
      XCH     PIPAZ

DODELVZ TS      DELVZ

REPIP4  EXTEND              # COMPUTE GUIDANCE PERIOD
      DCA     PIPTIME1
      DXCH    PGUIDE
      EXTEND
      DCS     PIPTIME
      DAS     PGUIDE

      CA      CDUX          # READ CDUS INTO HIGH ORDER CDUTEMPS
      TS      CDUTEMPX
      CA      CDUY
      TS      CDUTEMPY
      CA      CDUZ
      TS      CDUTEMPZ
      CA      DELVX
      TS      PIPATMPX
      CA      DELVY
      TS      PIPATMPY
      CA      DELVZ
```

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TS      PIPATMPZ

TC      Q

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REREADAC	CCS	PIPAGE	
	TCF	READACCS	# PIP READING NOT STARTED. GO TO BEGINNING
	CAF	DONEADR	# SET UP RETURN FROM PIPASR
	TS	Q	
	CCS	DELVZ	
	TCF	REPIP4	# Z DONE, GO DO CDUS
	TCF	+3	# Z NOT DONE, CHECK Y.
	TCF	REPIP4	
	TCF	REPIP4	
	ZL		
	CCS	DELVY	
	TCF	+3	
	TCF	CHKTEMX	# Y NOT DONE, CHECK X.
	TCF	+1	
	LXCH	PIPAZ	# Y DONE, ZERO Z PIP.
	CCS	TEMZ	
	CS	TEMZ	# TEMZ NOT = -0, CONTAINS -PIPAZ VALUE.
	TCF	DODELVZ	
	TCF	-2	
	LXCH	DELVZ	# TEMZ = -0, L HAS ZPIP VALUE.
	TCF	REPIP4	
CHKTEMX	CCS	TEMX	# HAS THIS CHANGED
	CS	TEMX	# YES
	TCF	+3	# YES
	TCF	-2	# YES
	TCF	REPIP1	# NO
	TS	DELVX	
	CS	TEMY	
	TS	DELVY	
	CS	ZERO	# ZERO X AND Y PIPS
	DXCH	PIPAX	# L STILL ZERO FROM ABOVE
	TCF	REPIP3	
DONEADR	GENADR	PIPSDONE	

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```

      BANK      33
      SETLOC    SERVICES
      BANK

      COUNT*    $$/SERV

      TMPTOSPT  CA      CDUTEMPY      # THIS SUBROUTINE, CALLED BY AN RTB FROM
                TS      CDUSPOTY      # INTERPRETIVE, LOADS THE CDUS CORRESPON-
                CA      CDUTEMPZ      # DING TO PIPTIME INTO THE CDUSPOT VECTOR.
                TS      CDUSPOTZ
                CA      CDUTEMPX
                TS      CDUSPOTX
                TC      Q

      # LRHTASK IS A WAITLIST TASK SET BY READACCS DURING THE DESCENT BRAKING
      # PHASE WHEN THE ALT TO THE LUNAR SURFACE IS LESS THAN 25,000 FT. THIS
      # TASK CLEARS THE ALTITUDE MEASUREMENT MADE DISCRETE AND INITIATES THE
      # LANDING RADAR MEASUREMENT JOB (LRHJOB) TO TAKE A ALTITUDE MEASUREMENT
      # 50 MS PRIOR TO THE NEXT READACCS TASK.

      BANK      21
      SETLOC    R10
      BANK

      COUNT*    $$/SERV

      LRHTASK   CS      FLGWRD11
                MASK    LRBYBIT
                EXTEND
                BZF     GRP20FF      # LR BYPASS SET -- BYPASS ALL LR READING.

                CA      READLBIT
                MASK    FLGWRD11    # IS READLR FLAG SET?
                EXTEND
                BZF     GRP20FF      # NO. BYPASS LR READ.

                CS      FLGWRD11
                MASK    NOLRRBIT    # IS LR READ INHIBITED?
                EXTEND
                BZF     GRP20FF      # YES. BYPASS LR READ.

                CA      PRI032      # LR READ OK. SET JOB TO DO IT
                TC      NOVAC        # ABOUT 50 MS. PRIOR TO PIPA READ.
                EBANK=   HMEAS
                2CADR    LRHJOB

```



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BANK 33  
SETLOC SERVICES  
BANK

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HIGATASK	INHINT	
	CS	PRI03 # SET HIGATE AND LR INHIBIT FLAGS
	MASK	FLGWRD11
	AD	PRI03
	TS	FLGWRD11
	CAF	PRI032
	TC	FINDVAC # SET LR POSITIONING JOB (POS2)
	EBANK=	HMEAS
	2CADR	HIGATJOB
	TCF	CONTSERV # CONTINUE SERVICER

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(396 765)

# MUNRETRN IS THE RETURN LOC FROM SPECIAL AVE G ROUTINE (MUNRVG)

MUNRETRN EXIT

CS FLGWRD11

MASK LRBYBIT

EXTEND

BZF COPYCYC1

# BYPASS LR LOGIC IF BIT15 IS SET.

CA READLBIT

MASK FLGWRD11

EXTEND

BZF 35KCHK

# SEE IF ALT &lt; 35000 FT LAST CYCLE

# ALT WAS &gt; 35000 FT LAST CYCLE CHK NOW

CAF XORFLBIT

MASK FLGWRD11

EXTEND

BZF XORCHK

# WERE WE BELOW 30000 FT LAST PASS?

# NO -- TEST THIS PASS

HITEST

CAF PSTHIBIT

MASK FLGWRD11

EXTEND

BZF HIGATCHK

# CHECK FOR HIGATE

# NOT AT HIGATE LAST CYCLE -- CHK THIS CYCLE

POS2CHK

CAF BIT7

EXTEND

RAND CHAN33

EXTEND

BZF UPDATCHK

CAF BIT13

EXTEND

RAND CHAN12

EXTEND

BZF LRPOSALM

TCF CONTSERV

# VERIFY LR IN POS2

# IT IS -- CHECK FOR LR UPDATE

# LR NOT IN POS2 OR REPOSITIONING -- BAD

# LR BEING REPOSITIONED -- CONTINUE SERV

HIGATCHK

CA TTF/8

AD RPCRTIME

EXTEND

BZMF POS1CHK

# IS TTF &gt; CRITERION? (TTF IS NEGATIVE)

# NO

CA EBANK4

XCH EBANK

TS L

# MUST SWITCH EBANKS

# SAVE IN L

EBANK= XNBPIP

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```
CS      XNBPIP      # UXBXP IN GSOP CH5
EBANK=  DVCNTR
LXCH    EBANK      # RESTORE EBANK
AD      RPCRTQSW    # QSW - UXBXP
```

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(396 765)

```
EXTEND
BZMF    HIGATASK    # IF UXBXP > QSW, THEN REPOSITION

POS1CHK  CAF      BIT6    # HIGATE NOT IN SIGHT -- DO POS1 CHK
EXTEND
RAND     33
EXTEND
BZF      UPDATCHK    # LR IN POS1 -- CHECK FOR LR UPDATE

LRPOSALM TC      ALARM    # LR NOT IN PROPER POS-ALARM-BYPASS UPDATE
OCT      511          # AND CONTINUE SERVICER

CONTSERV INHINT
CS       BITS4-7
MASK     FLGWRD11    # CLEAR LR MEASUREMENT MADE DISCRETES.
TS       FLGWRD11

TC       IBNKCALL    # SET LR LITES PROPERLY
CADR     R12LITES
```

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```

COPYCYC1 TC QUIKFAZ5

R29? CA FLAGWRD3
      MASK NR29&RDR
      CCS A # IS NOR29FLG OR READRFLG SET?
      TCF R29NODES # YES, SO DON'T DESIGNATE.

      CA RADMODES # NO, SO R29 IS CALLED FOR.
      MASK OCT10002 # IS THE RR NOT ZEROING ITS CDUS, AND
      CCS A # IS THE RENDEZVOUS RADAR IN AUTO MODE?
      TCF R29NODES # NO, SO DON'T DESIGNATE.

      CA RADMODES
      MASK PRI022
      CCS A # IS RR REPOSITIONING OR REMODING?
      TCF NOR29NOW # YES: COME BACK IN 2 SECONDS & TRY AGAIN.

      TCF R29

R29NODES INHINT # R29 NOT ALLOWED THIS CYCLE.
          CS DESIGBIT # SHOW THAT DESIGNATION IS OFF.
          MASK RADMODES
          TS RADMODES

NOR29NOW TC INTPRET # INTPRET DOES A RELINT.
          VLOAD ABVAL # MPAC = ABVAL( NEW SM. POSITION VECTOR )
          R1S
          PUSH DSU #
          /LAND/ (2)
          STORE HCALC # NEW HCALC*2(24)M.
          STORE HCALC1
          DMPR RTB
          ALTCONV
          SGNAGREE
          STOVL ALTBITS # ALTITUDE FOR R10 IN BIT UNITS.
          UNIT/R/
          VXV UNIT
          UHYP
          STOVL UHZP # DOWNRANGE HALF-UNIT VECTOR FOR R10.
          R1S
          VXM VSR4
          REFMMAT
          STOVL RN1 # TEMP. REF. POSITION VECTOR*2(29)M.
          V1S
          VXM VSL1

```

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```

                                REFSMMAT
                                VN1
                                UNIT/R/
                                ABVAL
                                VXV
                                ABVAL
                                V1S
                                SL1 DSQ
                                DDV
                                DMPR RTB
                                ARCONV1
                                SGNAGREE
                                EXIT
                                INHINT
                                CA UNIT/R/
                                TS RUNIT
                                CA UNIT/R/ +2
                                TS RUNIT +1
                                CA UNIT/R/ +4
                                TS RUNIT +2
                                CA MPAC
                                TS DALTRATE
                                EXTEND
                                DCA R1S
                                DXCH R
                                EXTEND
                                DCA R1S +2
                                DXCH R +2
                                EXTEND
                                DCA R1S +4
                                DXCH R +4
                                EXTEND
                                DCA V1S
                                DXCH V
                                EXTEND
                                DCA V1S +2
                                DXCH V +2
                                EXTEND
                                DCA V1S +4
                                DXCH V +4
                                TCF COPYCYCL
                                # COMPLETE THE COPYCYCL.

                                # TEMP. REF. VELOCITY VECTOR 2(7) M/CS.
                                # LEAVE ALTITUDE RATE COMPENSATION IN MPAC
                                # UPDATE RUNIT FOR R10.
                                # LOAD NEW DALTRATE FOR R10.
```

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422 (Page LM0878 422)≡

(396 765)

# ALTCHK COMPARES CURRENT ALTITUDE (IN HCALC) WITH A SPECIFIED ALTITUDE FROM A TABLE  
 # ITS CALLING SEQUENCE IS AS FOLLOWS:-  
 #  
 #       L       CAF       N  
 #       L+1     TC       BANKCALL  
 #       L+2     CADR     ALTCHK  
 #       L+3     RETURN HERE IF HCALC STILL > SPECIFIED CRITERION.   C(L) = +0.  
 #       L+4     RETURN HERE IF HCALC < OR = SPECIFIED CRITERION.   C(A) = C(L) = +0  
 #  
 # ALTCHK MUST BE BANKCALLED EVEN FROM ITS OWN BANK.   N IS THE LOCATION, RELATIVE TO  
 # OF THE BEGINNING OF THE DP CONSTANT TO BE USED AS A CRITERION.

ALTCHK	EXTEND		
	INDEX	A	
	DCA	ALTCRIT	
	DXCH	MPAC +1	
	EXTEND		
	DCS	HCALC	
	DAS	MPAC +1	
	TC	BRANCH +4	
	CAF	ZERO	# BETTER THAN A NOOP, PERHAPS
	INCR	BUF2	
	TCF	SWRETURN	
ALTCRIT	=	25KFT	
25KFT	2DEC	7620 B-24	# (0)
50KFT	2DEC	15240 B-24	# (2)
50FT	2DEC	15.24 B-24	# (4)
30KFT	2DEC	9144 B-24	# (6)
2KFT/SEC	DEC	6.096 B-7	# 2000 FT/SEC AT 2(7) M/CS

# (A remark was likely to be needed here to explain XORCHK) 4/Jun/09,FB

XORCHK	CAF	SIX	# ARE WE BELOW 30000 FT?
	TC	BANKCALL	
	CADR	ALTCHK	
	TCF	HITEST	# CONTINUE LR UPDATE
	TC	UPFLAG	# YES: INHIBIT X-AXIS OVERRIDE
	ADRES	XOVINFLG	

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	TC	UPFLAG	
	ADRES	XORFLG	
	TCF	HITEST	# CONTINUE LR UPDATE
35KCHK	CAF	TWO	# ARE WE BELOW 35000 FT?
423	$\langle \text{Page LM0879 423} \rangle \equiv$		(396 765)
	TC	BANKCALL	
	CADR	ALTCHK	
	TCF	CONTSERV	
	TC	UPFLAG	
	ADRES	READLR	# SET READLR FLAG TO ENABLE LR READING.
	TCF	CONTSERV	

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# \*\*\*\*\*

CALCGRAV	UNIT	PUSH	# SAVE UNIT/R/ IN PUSHLIST	(18)
	STORE	UNIT/R/		
	LXC,1	SLOAD	# RTX2 = 0 IF EARTH ORBIT, =2 IF LUNAR.	
		RTX2		
		RTX2		
	DCOMP	BMN		
		CALCGRV1		
	VLOAD	DOT	#	(12)
		UNITZ		
		UNIT/R/		
	SL1	PUSH	#	(14)
	DSQ	BDSU		
		DP1/20		
	PDDL	DDV		
		RESQ		
		34D	# (RN)SQ	
	STORE	32D	# TEMP FOR (RE/RN)SQ	
	DMP	DMP		
		20J		
	VXSC	PDDL		
		UNIT/R/		
	DMP	DMP		
		2J		
		32D		
	VXSC	VSL1		
		UNITZ		
	VAD	STADR		
	STORE	UNITGOBL		
	VAD	PUSH	# MPAC = UNIT GRAVITY VECTOR.	(18)
CALCGRV1	DLOAD	NORM	# PERFORM A NORMALIZATION ON RMAGSQ IN	
		34D	# ORDER TO BE ABLE TO SCALE THE MU FOR	
		X2	# MAXIMUM PRECISION.	
	BDDV*	SLR*		
		-MUDT,1		
		0 -21D,2		
	VXSC	STADR		
	STORE	GDT1/2	# SCALED AT 2(+7) M/CS	
	RVQ			
CALCRVG	VLOAD	VXM		
		DELV		
		REFSMMAT		
	VXSC	VSL1		



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≡

DP1/20

SHIFT11

2DEC

2DEC

0.05

1 B-11

STORE

VSR1

VAD

KPIP1

DELVREF

PUSH

PUSH

GDT/2

PDDL

VN

PGUIDE

VXSC

6D

STQ

RN

31D

STCALL

RN1

CALCGRAV

VAD

VAD

VAD

VN

STCALL

VN1

31D

# (DV-OLDGDT)/2 TO PD SCALED AT 2(+7) M/CS.

(396 765)

# TEMP STORAGE OF RN SCALED 2(+29) M

# TEMP STORAGE OF VN SCALED 2(+7) M/CS

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(396 765)

\*\*\*\*\*

# MUNRVG IS A SPECIAL AVERAGE G INTEGRATION ROUTINE USED BY THRUSTING  
 # PROGRAMS WHICH FUNCTION IN THE VICINITY OF AN ASSUMED SPHERICAL MOON.  
 # THE INPUT AND OUTPUT QUANTITIES ARE REFERENCED TO THE STABLE MEMBER  
 # COORDINATE SYSTEM.

RVBOTH	VLOAD	PUSH	
		G(CSM)	
	VAD	PDDL	
		V(CSM)	
		PGUIDE	
	DDV	VXSC	
		SHIFT11	
	VAD		
		R(CSM)	
	STCALL	R1S	
		MUNGRAV	
	VAD	VAD	
		V(CSM)	
	STADR		
	STORE	V1S	
	EXIT		
	TC	QUIKFAZ5	
	TC	INTPRET	
	VLOAD		
		GDT1/2	
	STOVL	G(CSM)	
		R1S	
	STOVL	R(CSM)	
		V1S	
	STORE	V(CSM)	
	EXIT		
	TC	QUIKFAZ5	
	TC	INTPRET	
MUNRVG	VLOAD	VXSC	
		DELV	
		KPIP2	
	PUSH	VAD	# 1ST PUSH: DELV IN UNITS OF 2(8) M/CS
		GDT/2	
	PUSH	VAD	# 2ND PUSH: (DELV + GDT)/2, UNITS OF 2(7)
		V	#
			(12)
	PDDL	DDV	
		PGUIDE	
		SHIFT11	
	VXSC		

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VAD

R

STCALL R1S

MUNGRAV

# STORE R SCALED AT 2(+24) M

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	VAD	VAD	
	VAD		
		V	
	STORE	V1S	# STORE V SCALED AT 2(+7) M/CS.
	ABVAL		
	STOVL	ABVEL	# STORE SPEED FOR LR AND DISPLAYS.
		UNIT/R/	
	DOT	SL1	
		V1S	
	STOVL	HDOTDISP	# HDOT = V. UNIT(R)*2(7) M/CS.
		R1S	
	VXV	VSL2	
		WM	
	STODL	DELVS	# LUNAR ROTATION CORRECTION TERM*2(5) M/CS.
		36D	
	DSU		
		/LAND/	
	STCALL	HCALC	# FOR NOW, DISPLAY WHETHER POS OR NEG
		MUNRETRN	
MUNGRAV	UNIT		# AT 36D HAVE ABVAL(R), AT 34D R.R
	STODL	UNIT/R/	
		34D	
	SL	BDDV	
		6D	
		-MUDTMUN	
	DMP	VXSC	
		SHIFT11	
		UNIT/R/	
	STORE	GDT1/2	# 1/2GDT SCALED AT 2(7) M/CS.
	RVQ		
1.95SECS	DEC	195	
7.5	2DEC	.02286 B-6	# 7.5 FT/SEC AT 2(6) M/CS
2SEC(18)	2DEC	200 B-18	
2SEC(28)	2OCT	0000000310	# 2SEC AT 2(28)
4SEC(28)	2DEC	400 B-28	
BITS4-7	OCT	110	

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UPDATCHK	CAF	NOLRRBIT	# SEE IF LR UPDATE INHIBITED.
	MASK	FLGWRD11	
	CCS	A	
	TCF	CONTSERV	# IT IS -- NO LR UPDATE
	CAF	RNGEDBIT	# NO INHIBIT -- SEE ALT MEAS. THIS CYCLE.
	MASK	FLGWRD11	
	EXTEND		
	BZF	VMEASCHK	# NO ALT MEAS THIS CYCLE -- CHECK FOR VEL
POSUPDAT	CA	FIXLOC	# SET PUSHLIST TO ZERO
	TS	PUSHLOC	
	TC	INTPRET	
	VLOAD	VXM	
		HBEAMNB	
		XNBPIP	# HBEAM SM AT 2(2)
	PDVL	VSL2	# STORE HBEAM IN PD 0-5
		V1S	# SCALE V AT 2(5) M/CS
	VAD	DOT	
		DELVS	# V RELATIVE TO SURFACE AT 2(5) M/CS
		OD	# V ALONG HBEAM AT 2(7) M/CS.
	DMP	EXIT	
		RADSKAL	# SCALE TO RADAR COUNTS X 5
	CS	FLGWRD12	# TEST LR ALTITUDE SCALE FACTOR
	MASK	ALTSCBIT	
	EXTEND		
	BZF	+3	# BRANCH IF HIGH SCALE
	CA	SKALSKAL	# RESCALE IF LOW SCALE
	TC	SHORTMP	
+3	TC	INTPRET	
	DAD	SL	# CORRECT HMEAS FOR DOPPLER EFFECT
		HMEAS	
		7D	
	DMP	VXSC	# SLANT RANGE AT 2(21), PUSH UP FOR HBEAM
		HSCAL	# SLANT RANGE VECTOR AT 2(23) M
	DOT	DSU	
		UNIT/R/	# ALTITUDE AT 2(24) M
		HCALC	# DELTA H AT 2(24) M
	STORE	DELTAH	
	EXIT		
	CA	FLGWRD11	

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MASK PSTHIBIT

EXTEND

BZF NOREASON

# DO NOT PERFORM DATA REASONABLENESS TEST

# UNTIL AFTER HIGATE

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(396 765)

```
TC      INTPRET
ABS     DSU
        DELQFIX      # ABS(DELTAH) - DQFIX    50 FT NOM
SL3     DSU          # SCALE TO 2(21)
        HCALC        # ABS(DELTAH) - (50 + HCALC/8) AT 2(21)
EXIT

INCR    LRLCTR
TC      BRANCH
TCF     HFAIL      # DELTA H TOO LARGE
TCF     HFAIL      # DELTA H TOO LARGE
TC      DOWNFLAG   # TURN OFF ALT FAIL LAMP
ADRES   HFLSHFLG

NOREASON CS    FLGWRD11
MASK     LRINHBIT
CCS      A
TCF      VMEASCHK      # UPDATE INHIBITED -- TEST VELOCITY ANYWAY

TC      INTPRET      # DO POSITION UPDATE
DLOAD   SR4
        HCALC        # RESCALE H TO 2(28)M
EXIT
EXTEND
DCA     DELTAH      # STORE DELTAH IN MPAC AND
DXCH    MPAC        # BRING HCALC INTO A,L
TC      ALSIGNAG
EXTEND      # IF HIGH PART OF HCALC IS NON-ZERO, THEN
BZF     +2          # HCALC > HMAX,
TCF     VMEASCHK    # SO UPDATE IS BYPASSED
TS      MPAC +2     #          FOR LATER SHORTMP

CS      L          # -H AT 2(14) M
AD      LRHMAX      # HMAX - H
EXTEND
BZMF    VMEASCHK    # IF H >HMAX, BYPASS UPDATE
EXTEND
MP      LRWH        # WH(HMAX - H)
EXTEND
DV      LRHMAX      # WH(1 - H/HMAX)
TS      MPTMP
TC      SHORTMP2    # DELTAH (WH)(1 - H/HMAX) IN MPAC
TC      INTPRET     # MODE IS DP FROM ABOVE
SL1
VXSC    VAD
```

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```
UNIT/R/      # DELTAR = DH(WH) (1 - H/HMAX) UNIT/R/
R1S
STCALL GNUR
MUNGRAV
EXIT
```



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	TC	QUIKFAZ5	
	CA	ZERO	
RUPDATED	TC	GNURVST	
VMEASCHK	TC	QUIKFAZ5	# RESTART AT NEXT LOCATION
	CS	FLGWRD11	
	MASK	VELDABIT	# IS V READING AVAILABLE?
	CCS	A	
	TCF	VALTCHK	# NO: SEE IF V READING TO BE TAKEN
VELUPDAT	CS	VSELECT	# PROCESS VELOCITY DATA
	TS	L	
	ADS	L	# -2 VSELECT IN L
	AD	L	
	AD	L	# -6 VSELECT IN A
	INDEX	FIXLOC	
	DXCH	X1	# X1 = -6 VSELECT, X2 = -2 VSELECT
	CA	EBANK4	
	TS	EBANK	
	EBANK=	LRXCDU	
	CA	LRYCDU	# STORE LRCDUS IN CDUSPOTS
	TS	CDUSPOT	
	CA	LRZCDU	
	TS	CDUSPOT +2	
	CA	LRXCDU	
	TS	CDUSPOT +4	
	TC	BANKCALL	
	CADR	QUICTRIG	# GET SINES AND COSINES FOR NBSM
	CA	FIXLOC	
	TS	PUSHLOC	# SET PD TO ZERO
	TC	INTPRET	
	VLOAD*	CALL	
		VZBEAMNB,1	# CONVERT VBEAM FROM NB TO SM
		*NBSM*	
	PDDL	SL	# STORE IN PD 0-5
		VMEAS	# LOAD VELOCITY MEASUREMENT
		12D	
	DMP*	PUSH	# SCALE TO M/CS AT 2(6)
		VZSCAL,2	# AND STORE IN PD 6-7

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EXIT

CS ONE

TS MODE # CHANGE STORE MODE TO VECTOR

CA PIPTEM # STORE DELV IN MPAC

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ZL		
DXCH	MPAC	
CA	PIPTM +1	
ZL		
DXCH	MPAC +3	
CA	PIPTM +2	
ZL		
DXCH	MPAC +5	
CA	EBANK7	
TS	EBANK	# RESTORE EBANK 7
EBANK=	DVCNTR	
TC	INTPRET	
VXSC	PDDL	
	KPIP1	# SCALE DELV TO 2(7) M/CS AND PUSH
	LRVTIME	# TIME OF DELV AT 2(28) CS
DSU	DDV	
	PIPTIME	# TU - T(N-1)
	2SEC(28)	
VXSC	VSL1	# G(N-1)(TU - T(N-1))
	GDT/2	# SCALED AT 2(7) M/CS
VAD	VAD	# PUSH UP FOR DELV
	V	# VU = V(N-1) + DELVU + G(N-1) DTU
VSL2	VAD	# SCALE TO 2(5) M/CS AND SUBTRACT
	DELVS	# MOON ROTATION.
PUSH	ABVAL	# STORE IN PD
SR4	DAD	# ABS(VM)/8 + 7.5 AT 2(6)
	7.5	
STOVL	20D	# STORE IN 20D AND PICK UP VM
DOT	BDSU	# V(EST) AT 2(6)
	0	# DELTAV = VMEAS - V(EST)
PUSH	ABS	
DSU	EXIT	# ABS(DV) - (7.5 + ABS(VM)/8))
	20D	
INCR	LRMCTR	
TC	BRANCH	
TCF	VFAIL	# DELTA V TOO LARGE.      ALARM
TCF	VFAIL	# DELTA V TOO LARGE.      ALARM
TC	DOWNFLAG	# TURN OFF VEL FAIL LAMP
ADRES	VFLSHFLG	

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CA FLGWRD11

MASK VXINHBIT

EXTEND

BZF VUPDAT

# IF VX INHIBIT RESET, INCORPORATE DATA.

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	TC	DOWNFLAG	
	ADRES	VXINH	# RESET VX INHIBIT
	CA	VSELECT	
	AD	NEG2	# IF VSELECT = 2 (X AXIS).
	EXTEND		# BYPASS UPDATE
	BZF	ENDVDAT	
VUPDAT	CS	FLGWRD11	
	MASK	LRINHBIT	
	CCS	A	
	TCF	VALTCHK	# UPDATE INHIBITED
	TS	MPAC +1	
	CA	ABVEL	# STORE E7 ERASABLES NEEDED IN TEMPS
	TS	ABVEL*	
	CA	VSELECT	
	TS	VSELECT*	
	CA	EBANK5	
	TS	EBANK	# CHANGE EBANKS
	EBANK=	LRVF	
	CS	LRVF	
	AD	ABVEL*	# IF V < VF, USE WVF
	EXTEND		
	BZMF	USEVF	
	CS	ABVEL*	
	AD	LRVMAX	# VMAX - V
	EXTEND		
	BZMF	WSTOR -1	# IF V > VMAX, W = 0
	EXTEND		
	INDEX	VSELECT*	
	MP	LRWVZ	# WV(VMAX - V)
	EXTEND		
	DV	LRVMAX	# WV( 1 - V/VMAX )
	TCF	WSTOR	
USEVF	INDEX	VSELECT*	
	CA	LRWVFZ	# USE APPROPRIATE CONSTANT WEIGHT
	TCF	WSTOR	

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	-1	CA	ZERO	
WSTOR		TS	MPAC	
		CS	BIT7	# (=64D)
		AD	MODREG	
		EXTEND		

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```

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                                     (396 765)
                                     # IF IN P65,P66,P67, USE ANOTHER CONSTANT
                                     BZMF      +3
                                     CA      LRWVFF
                                     TS      MPAC
+3   CA      EBANK7
     TS      EBANK      # CHANGE EBANKS
                                     EBANK=  ABVEL
                                     TC      INTPRET
                                     DMP     VXSC      # W(Delta V)(VBEAMSM) UP 6-7, 0-5
                                     VAD
                                     V1S      # ADD WEIGHTED Delta V TO VELOCITY
                                     STORE   GNUV
                                     EXIT
                                     TC      QUIKFAZ5      # DO NOT RE-UPDATE
VUPDATED   CA      SIX
ENDVDAT    TC      GNURVST      # STORE NEW VELOCITY VECTOR
           =      VALTCHK
VALTCHK    TC      QUIKFAZ5      # DO NOT REPEAT ABOVE
           CAF     READVBIT      # TEST READVEL TO SEE IF VELOCITY READING
           MASK    FLGWRD11      # IS DESIRED.
           CCS     A
           TCF     READV      # YES -- READ VELOCITY
           CS      ABVEL      # NO -- SEE IF VELOCITY < 2000 FT/SEC
           AD      2KFT/SEC
           EXTEND
           BZMF     CONTSERV      # V > 2000 FT/SEC DO NOT READ VEL
           TC      UPFLAG      # V < 2000 FT/SEC SET READVEL AND READ.
           ADRES   READVEL
READV      CAF     PRI032      # SET UP JOB TO READ VELOCITY BEAMS.
           TC      NOVAC
           EBANK=  HMEAS
           2CADR   LRVJOB
           TCF     CONTSERV      # CONTINUE WITH SERVICER
GNURVST    TS      BUF      # STORE GNUR (=GNUV) IN R1S OR V1S
           EXTEND      # A = 0 FOR R, A = 6 FOR V

```

DCA	GNUR
INDEX	BUF
DXCH	R1S
EXTEND	



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```

      DCA      GNUR +2
      INDEX    BUF
      DXCH     R1S +2
      EXTEND
      DCA      GNUR +4
      INDEX    BUF
      DXCH     R1S +4
      TC       Q

QUIKFAZ5      CA      EBANK3
              XCH     EBANK      # SET EBANK 3
              DXCH    L          # Q TO A, A TO L
              EBANK=  PHSNAME5
              TS      PHSNAME5
              LXCH    EBANK
              EBANK=  DVCNTR
              TC      A

HFAIL         CS      LRRCTR
              EXTEND
              BZF     NORLITE      # IF R = 0, DO NOT TURN ON TRK FAIL
              AD      LRLCTR
              MASK    NEG3
              EXTEND      # IF L-R LT 4, DO NOT TURN ON TRK FAIL
              BZF     +2
              TCF     NORLITE

              TC      UPFLAG      # AND SET BIT TO TURN ON TRACKER FAIL LITE
              ADRES   HFLSHFLG

NORLITE       CA      LRLCTR
              TS      LRRCTR      # SET R = L

              TCF     VMEASCHK

VFAIL         CS      LRSCTR      # DELTA Q LARGE
              EXTEND      # IF S = 0, DO NOT TURN ON TRACKER FAIL
              BZF     NOLITE
              AD      LRMCTR      # M-S
              MASK    NEG3        # TEST FOR M-S > 3
              EXTEND      # IF M-S > 3, THEN TWO OR MORE OF THE
              BZF     +2          #      LAST FOUR V READINGS WERE BAD,
              TCF     NOLITE      #      SO TURN ON VELOCITY FAIL LIGHT

              TC      UPFLAG      # AND SET BIT TO TURN ON TRACKER FAIL LITE
```

	ADRES	VFLSHFLG	
442	$\langle \text{Page LM0891 442} \rangle \equiv$		(396 765)
	NOLITE	CA	LRMCTR
		TS	LRSCTR
		CCS	VSELECT
		TCF	ENDVDAT
		TC	UPFLAG
		ADRES	VXINH
		TCF	ENDVDAT
			# SET S = M
			# TEST FOR Z COMPONENT
			# NOT Z, DO NOT SET VX INHIBIT
			# Z COMPONENT - SET FLAG TO SKIP X
			# COMPONENT, AS ERROR MAY BE DUE TO CROSS
			# LOBE LOCK UP NOT DETECTED ON X AXIS.

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(396 765)

```
# *****
# LRVJOB IS SET WHEN THE LEM IS BELOW 15000 FT DURING THE LANDING PHASE
# THIS JOB INITIALIZES THE LANDING RADAR READ ROUTINE FOR 5 VELOCITY
# SAMPLES AND GOES TO SLEEP WHILE THE SAMPLING IS DONE -- ABOUT 500 MS.
# WITH A GOODEND RETURN THE DATA IS STORED IN VMEAS AND BIT7 OF LRSTAT
# IS SET. THE GIMBAL ANGLES ARE READ ABOUT MIDWAY IN THE SAMPLINGS.
```

```
170MS          EQUALS  ND1
```

```
LRVJOB          CA      170MS          # SET TASK TO READ CDUS + PIPAS
                TC      WAITLIST
                EBANK=  LRVTIME
                2CADR   RDGIMS

                CCS      VSELECT        # SEQUENCE LR VEL BEAM SELECTOR
                TCF      +2
                CAF      TWO            # IF ZERO, RESET TO TWO
                DOUBLE   # 2XVSELECT USED FOR VBEAM INDEX IN LRVEL
                TC      BANKCALL        # GO INITIALIZE LR VEL READ ROUTINE
                CADR     LRVEL
                TC      BANKCALL        # PUT LRVJOB TO SLEEP ABOUT 500 MS
                CADR     RADSTALL
                TCF      VBAD
                CCS      STILBADV       # IS DATA GOOD JUST PRESENT?
                TCF      VSTILBAD       # JUST GOOD -- MUST WAIT 4 SECONDS.

                INHINT
                EXTEND                  # GOOD RETURN -- STOW AWAY VMEAS
                DCA      SAMPLSUM
                DXCH      VMEAS
                CA      EBANK4          # FOR DOWNLINK
                TS      EBANK
                EBANK=  LRVTIME

                EXTEND
                DCA      LRVTIME
                DXCH      LRVTIMDL
                EXTEND
                DCA      LRXCDU
                DXCH      LRXCDUDL
                CA      LRZCDU
                TS      LRZCDUDL
                CA      EBANK7
                TS      EBANK
                EBANK=  VSELECT
```

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CS	FLGWRD11	# SET BIT TO INDICATE VELOCITY
MASK	VELDABIT	# MEASUREMENT MADE

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```

      ADS      FLGWRD11
ENDLRV      CCS      VSELECT      # UPDATE VSELECT
            TCF      +2
            CA       TWO
            TS       VSELECT
            TCF      ENDOFJOB

VBAD        CAF      TWO          # SET STILBAD TO WAIT 4 SECONDS
VSTILBAD    TS       STILBADV
            TCF      ENDLRV

# LRHJOB IS SET BY LRHTASK WHEN LEM IS BELOW 25000 FT.  THIS JOB
# INITIALIZES THE LR READ ROUTINE FOR AN ALT MEASUREMENT AND GOES TO
# SLEEP WHILE THE SAMPLING IS DONE -- ABOUT 95 MS.  WITH A GOODEND RETURN
# THE ALT DATA IS STORED IN HMEAS AND BIT7 OF LRSTAT IS SET.

      BANK      34
      SETLOC    R12STUFF
      BANK

COUNT*     $$/SERV

LRHJOB      TC      BANKCALL      # INITIATE LR ALT MEASUREMENT
            CADR     LRALT
            TC      BANKCALL      # LRHJOB TO SLEEP ABOUT 95MS
            CADR     RADSTALL
            TCF      HBAD
            CCS      STILBADH      # IS DATA GOOD JUST PRESENT?
            TCF      HSTILBAD      # JUST GOOD -- MUST WAIT 4 SECONDS.

      INHINT
      EXTEND
      DCA      SAMPLSUM      # GOOD RETURN -- STORE AWAY LRH DATA
      DXCH     HMEAS         # LRH DATA 1.079 FT/BIT
      EXTEND
      DCA      PIPTIME1
      DXCH     MKTIME

      EXTEND
      DCA      CDUTEMPY      # CDUY,Z = AIG,AMG
      DXCH     AIG

      CA       CDUTEMPX      # CDUX = AOG
      TS       AOG
```

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	CS	FLGWRD11	# SET BIT TO INDICATE RANGE
	MASK	RNGEDBIT	# MEASUREMENT MADE.
	ADS	FLGWRD11	
ENDLRH	TC	ENDOFJOB	# TERMATE LRHJOB

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(396 765)

```
HBAD      CA      FLAGWRD5
          MASK     RNGSCBIT      # IS BAD RETURN DUE TO SCALE CHANGE?
          EXTEND
          BZF      HSTILBAD -1    # NO  RESET HSTILBAD
          TC       DOWNFLAG      # YES  RESET SCALE CHANGE BIT AND IGNORE
          ADRES    RNGSCFLG
          TC       ENDOFJOB

HSTILBAD  CAF      TWO           # SET STILBAD TO WAIT 4 SECONDS
          TS       STILBADH
          TC       ENDOFJOB

          BANK     34
          SETLOC   SERV4
          BANK

          COUNT*   $$/SERV
```

```
# RDGIMS IS A TASK SET UP BY LRVJOB TO PICK UP THE IMU CDUS AND TIME
# AT ABOUT THE MIDPOINT OF THE LR VEL READ ROUTINE WHEN 5 VEL SAMPLES
# ARE SPECIFIED.
```

```
RDGIMS    EBANK=   LRVTIME
          EXTEND
          DCA      TIME2          # PICK UP TIME2, TIME1
          DXCH     LRVTIME        #          AND SAVE IN LRVTIME

          EXTEND
          DCA      CDUX           # PICK UP CDUX AND CDUY
          DXCH     LRXCDU         #          AND SAVE IN LRXCDU AND LRYCDU

          CA       CDUZ
          TS       LRZCDU        # SAVE CDUZ IN LRZCDU

          CA       PIPAX
          TS       PIPTM         # SAVE PIPAX IN PIPTM

          EXTEND
          DCA      PIPAY          # PICK UP PIPAY AND PIPAZ
          DXCH     PIPTM +1       #          AND SAVE IN PIPTM +1 AND PIPTM +2
          TC       TASKOVER

          BANK     33
          SETLOC   SERVICES
          BANK
```

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COUNT\* \$\$/SERV

EBANK= DVCNTR



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```
# HIGATJOB IS SET APPROXIMATELY 6 SECONDS PRIOR TO HIGH GATE DURING
# THE DESCENT BURN PHASE OF LUNAR LANDING. THIS JOB INITIATES THE
# LANDING RADAR REPOSITIONING ROUTINE AND GOES TO SLEEP UNTIL THE
# LR ANTENNA MOVES FROM POSITION 1 TO POSITION 2. IF THE LR ANTENNA
# ACHIEVES POSITION 2 WITHIN 22 SECONDS THE ALTITUDE AND VELOCITY
# BEAM VECTORS ARE RECOMPUTED TO REFLECT THE NEW ORIENTATION WITH
# RESPECT TO THE NB. BIT10 OF LRSTAT IS CLEARED TO ALLOW LR
# MEASUREMENTS AND THE JOB TERMINATES.
```

```
HIGATJOB      TC      BANKCALL      # START LRPOS2 JOB
               CADR      LRPOS2
               TC      BANKCALL      # PUT HIGATJOB TO SLEEP UNTIL JOB IS DONE
               CADR      RADSTALL
               TCF      POSALARM      # BAD END      ALARM

POSGOOD        CA      PRI023      # REDUCE PRIORITY FOR INTERPRETIVE COMPS.
               TC      PRIOCHNG

               TC      SETPOS2      # LR IN POS2 -- SET UP TRANSFORMATIONS

               TC      DOWNFLAG
ADRES          NOLRREAD      # RESET NOLRREAD FLAG TO ENABLE LR READING
TC             ENDOFJOB

POSALARM        CA      OCT523
               TC      BANKCALL
               CADR      PRIOLARM      # FLASH ALARM CODE
               TCF      GOTOPOOH      # TERMINATE
               TCF      +3            # PROCEED -- TRY AGAIN
               TCF      ENDOFJOB      # V 32 E      TERMINATE R12
               TC      ENDOFJOB

               +3      CA      BIT7      # SEE IF IN POS2 YET
               EXTEND
               RAND      CHAN33
               EXTEND
               BZF      POSGOOD      # POS2 ACHIEVED  SET UP ANTENNA BEAMS
               TCF      POSALARM      # STILL DIDN'T MAKE IT  REALARM

OCT523          OCT      00523

SETPOS1         TC      MAKECADR      # MUST BE CALLED BY BANKCALL
               TS      LRADRET1      # SAVE RETURN CADR.  SINCE BUP2 CLOBBBERED
```

CAF	TWO	
TS	STILBADH	# INITIALIZE STILBAD
TS	STILBADV	# INITIALIZE STILBAD
CA	ZERO	# INDEX FOR LRALPHA, LRBETA IN POS 1.

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```
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      TS      LRLCTR      # SET L,M,R, ANS S TO ZERO
      TS      LRMCTR
      TS      LRRCTR
      TS      LRSCTR
      TS      VSELECT      # INITIALIZE VSELECT

      TC      SETPOS      # CONTINUE WITH COMPUTATIONS.

      CA      LRADRET1
      TC      BANKJUMP      # RETURN TO CALLER

      SETPOS2      CA      TWO      # INDEX FOR POS2
      SETPOS      XCH      Q      # SAVE INDEX IN Q
      TS      LRADRET      # SAVE RETURN

      CA      EBANK5
      TS      EBANK
      EBANK=    LRALPHA

      EXTEND
      INDEX      Q
      DCA      LRALPHA      # LRALPHA IN A, LRBETA IN L
      TS      CDUSPOT +4      # ROTATION ABOUT X
      LXCH      CDUSPOT      # ROTATION ABOUT Y
      CA      ZERO
      TS      CDUSPOT +2      # ZERO ROTATION ABOUT Z.

      CA      EBANK7
      TS      EBANK
      EBANK=    LRADRET

      TC      INTPRET
      VLOAD      CALL
      UNITY      # CONVERT UNITY(ANTENNA) TO NB
      TRG*SMNB
      STOVL      VYBEAMNB
      UNITX      # CONVERT UNITX(ANTENNA) TO NB
      CALL
      *SMNB*
      STORE      VXBEAMNB
      VXV      VSL1
      VYBEAMNB
      STOVL      VZBEAMNB      # Z = X * Y
      HBEAMANT
      CALL
```

```

                                *SMNB*
                                # CONVERT TO NB
                                STORE HBEAMNB
                                EXIT

```

452a  $\langle \text{Page LM0897 452a} \rangle \equiv$  (396 765)

TC LRADRET

## 1.25 landing analog displays

452b  $\langle \text{landing analog displays 452b} \rangle \equiv$  (7)

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 $\langle \text{Page LM0902 461} \rangle$   
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```

      BANK      21
      SETLOC    R10
      BANK

      EBANK=    UNIT/R/
      COUNT*    $$/R10

LANDISP      LXCH    PIPCTR1      # UPDATE TBASE2 AND PIPCTR SIMULTANEOUSLY.
              CS      TIME1
              DXCH    TBASE2

              CS      FLAGWRD7      # IS LANDING ANALOG DISPLAYS FLAG SET?
              MASK    SWANDBIT
              CCS      A
              TCF      DISPRSET      # NO.
              CA      IMODES33      # BIT 7 = 0 (DO ALTRATE), =1 (DO ALT.)
              MASK    BIT7
              CCS      A
              TCF      ALTOUT

ALTROUT      TC      DISINDAT      # CHECK MODE SELECT SWITCH AND DIDFLG.
              CS      IMODES33
              MASK    BIT7
              ADS      IMODES33      # ALTERNATE ALTITUDE RATE WITH ALTITUDE.
              CAF      BIT2          # RATE COMMAND IS EXECUTED BEFORE RANGE.
              EXTEND

              WOR      CHAN14      # ALTRATE (BIT2 = 1), ALTITUDE (BIT2 = 0).
              CA      RUNIT        # COMPUTE ALTRATE = RUNIT.VVECT M/CS *(-6).
              EXTEND

              MP      VVECT        # MULTIPLY X-COMPONENTS.
              XCH      RUPTREG1    # SAVE SINGLE PRECISION RESULT M/CS*2(-6)
              CA      RUNIT +1     # MULTIPLY Y-COMPONENTS.
              EXTEND

              MP      VVECT +1
              ADS      RUPTREG1    # ACCUMULATE PARTIAL PRODUCTS.
              CA      RUNIT +2     # MULTIPLY Z-COMPONENTS.
              EXTEND

              MP      VVECT +2
              ADS      RUPTREG1    # ALTITUDE RATE IN M/CS *2(-6).
              CA      ARCONV      # CONVERT ALTRATE TO BIT UNITS (.5FPS/BIT)
              EXTEND

              MP      RUPTREG1
              DDOUBL
              DDOUBL
              XCH      RUPTREG1    # ALTITUDE RATE IN BIT UNITS*2(-14).
              CA      DALTRATE    # ALTITUDE RATE COMPENSATION FACTOR.

```

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EXTEND

MP DT

AD RUPTREG1

TS ALTRATE

CS ALTRATE

# ALTITUDE RATE IN BIT UNITS\*2(-14).

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(452b 743)

```

DATAOUT  EXTEND      # CHECK POLARITY OF ALTITUDE RATE.
          BZMF      +2
          TCF      DATAOUT      # NEGATIVE -- SEND POS. PULSES TO ALTM REG.
          CA      ALTRATE      # POSITIVE OR ZERO -- SET SIGN BIT = 1 AND
          AD      BIT15      # SEND TO ALTM REGISTER. *DO NOT SEND +0*
          TS      ALTM      # ACTIVATE THE LANDING ANALOG DISPLAYS
          CAF      BIT3
          EXTEND
          WOR      CHAN14      # BIT3 DRIVES THE ALT/ALTRATE METER.
          TCF      TASKOVER      # EXIT

ALTOUT    TC      DISINDAT      # CHECK MODE SELECT SWITCH AND DIDFLG.
          CS      BIT7
          MASK     IMODES33
          TS      IMODES33      # ALTERNATE ALTITUDE RATE WITH ALTITUDE.
          CS      BIT2
          EXTEND
          WAND     CHAN14
          CCS      ALTBITS      # = -1 IF OLD ALT. DATA TO BE EXTRAPOLATED.
          TCF      +4
          TCF      +3
          TCF      OLDDATA
          TS      ALTBITS      # SET ALTBITS FROM -0 TO +0.
          CS      ONE
          DXCH     ALTBITS      # SET ALTBITS = -1 FOR SWITCH USE NEXT PASS.
          DXCH     ALTSAVE
          CA      BIT10      # NEW ALTITUDE EXTRAPOLATION WITH ALTRATE.
          XCH      Q
          LXCH     7      # ZL
          CA      DT
          EXTEND
          DV      Q      # RESCALE DT*2(-14) TO *2(-9) TIME IN CS.
          EXTEND
          MP      ARTOA2      # .0021322 *2(+8)
          TCF      OLDDATA +1      # RATE APPLIES FOR DT CS.

ZDATA2    DXCH     ALTSAVE
          TCF      NEWDATA

OLDDATA    CA      ARTOA      # RATE APPLIES FOR .5 SEC. (4X/SEC. CYCLE)
          EXTEND
          MP      ALTRATE      # EXTRAPOLATE WITH ALTITUDE RATE.
          DDOUBL
          AD      ALTSAVE +1
          TS      ALTSAVE +1
          CAF      ZERO
```

ADS	ALTSAVE	
CAF	POSMAX	# FORCE SIGN AGREEMENT ASSUMING A
AD	ONE	# NON-NEGATIVE ALTSAVE.
AD	ALTSAVE +1	# IF ALTSAVE IS NEGATIVE, ZERO ALTSAVE
TS	ALTSAVE +1	# AND ALTSAVE +1 AT ZERODATA.



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(452b 743)

	CAF	ZERO	
	AD	POSMAX	
	AD	ALTSAVE	
	TS	ALTSAVE	# POSSIBLY SKIP TO NEWDATA.
	TCF	ZERODATA	
NEWDATA	CCS	ALTSAVE +1	
	TCF	+4	
	TCF	+3	
	CAF	ZERO	# SET NEGATIVE ALTSAVE +1 TO +0.
	TS	ALTSAVE +1	
	CCS	ALTSAVE	# PROVIDE A 15 BIT UNSIGNED OUTPUT.
	CAF	BIT15	# THE HI-ORDER PART IS +1 OR +0.
	AD	ALTSAVE +1	
	TCF	DATAOUT	# DISPATCH UNSIGNED BITS TO ALTM REG.
DISINDAT	EXTEND		
	QXCH	LADQSAVE	# SAVE RETURN TO ALTROUT +1 OR ALTOUT +1
	CAF	BIT6	
	EXTEND		# WISHETH THE ASTRONAUT THE ANALOG
	RAND	CHAN30	# DISPLAYS? I.E.,
	CCS	A	# IS THE MODE SELECT SWITCH IN PGNC?
	TCF	DISPRSET	# NO. ASTRONAUT REQUESTS NO INERTIAL DATA
	CS	FLAGWRD1	# YES. CHECK STATUS OF DIDFLAG.
	MASK	DIDFLBIT	
	EXTEND		
	BZF	SPEEDRUN	# SET. PERFORM DATA DISPLAY SEQUENCE.
	CS	FLAGWRD1	# RESET. PERFORM INITIALIZATION FUNCTIONS.
	MASK	DIDFLBIT	
	ADS	FLAGWRD1	# SET DIDFLAG.
	CS	BIT7	
	MASK	IMODES33	# TO DISPLAY ALTRATE FIRST AND ALT. SECOND
	TS	IMODES33	
	CS	FLAGWRD0	# ARE WE IN DESCENT TRAJECTORY?
	MASK	R10FLBIT	
	EXTEND		
	BZF	TASKOVER	# NO
	CAF	BIT8	# YES.
	EXTEND		
	WOR	CHAN12	# SET DISPLAY INERTIAL DATA OUTBIT.
	CAF	ZERO	
	TS	TRAKLATV	# LATERAL VELOCITY MONITOR FLAG
	TS	TRAKFWDV	# FORWARD VELOCITY MONITOR FLAG
	TS	LATVMETR	# LATVEL MONITOR METER
	TS	FORVMETR	# FORVEL MONITOR METER
	CAF	BIT4	
	TC	TWIDDLE	

	ADRES	INTLZE	
	TCF	TASKOVER	
INTLZE	CAF	BIT2	
	EXTEND		
	WOR	CHAN12	# ENABLE RR ERROR COUNTER.

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(452b 743)

```

      CS      IMODES33
      MASK    BIT8
      ADS     IMODES33      # SET INERTIAL DATA FLAG.
      TCF     TASKOVER

SPEEDRUN      CS      PIPTIME +1      # UPDATE THE VELOCITY VECTOR
              AD      TIME1          # COMPUTE T - TN
              AD      HALF          # CORRECT FOR POSSIBLE OVERFLOW OF TIME1.
              AD      HALF
              XCH     DT              # SAVE FOR LATER USE
              CA      1SEC
              TS      ITEMP5          # INITIALIZE FOR DIVISION LATER
      EXTEND
      DCA      GDT/2                # COMPUTE THE X-COMPONENT OF VELOCITY.
      DDOUBL
      DDOUBL
      EXTEND
      MP      DT
      EXTEND
      DV      ITEMP5
      XCH     VVECT                # VVECT = G(T-TN) M/CS *2(-5)
      EXTEND
      DCA      V                    # M/CS *2(-7)
      DDOUBL                # RESCALE TO 2(-5)
      DDOUBL
      ADS     VVECT                # VVECT = VN + G(T-TN) M/CS *2(-5)
      CA      PIPAX                # DELV CM/SEC *2(-14)
      AD      PIPATMPX            # IN CASE PIPAX HAS BEEN ZEROED
      EXTEND
      MP      KPIP1(5)            # DELV M/CS *2(-5)
      ADS     VVECT                # VVECT = VN + DELV + GN(T-TN) M/CS *2(-5)
      EXTEND
      DCA      GDT/2 +2            # COMPUTE THE Y-COMPONENT OF VELOCITY.
      DDOUBL
      DDOUBL
      EXTEND
      MP      DT
      EXTEND
      DV      ITEMP5
      XCH     VVECT +1
      EXTEND
      DCA      V +2
      DDOUBL
      DDOUBL
      ADS     VVECT +1

```

CA	PIPAY
AD	PIPATMPY
EXTEND	
MP	KPIP1(5)
ADS	VVECT +1

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(452b 743)

```
EXTEND
DCA      GDT/2 +4      # COMPUTE THE Z-COMPONENT OF VELOCITY.
DDOUBL
DDOUBL
EXTEND
MP      DT
EXTEND
DV      ITEMP5
XCH      VVECT +2
EXTEND
DCA      V +4
DDOUBL
DDOUBL
ADS      VVECT +2
CA      PIPAZ
AD      PIPATMPZ
EXTEND
MP      KPIP1(5)
ADS      VVECT +2

CAF      BIT3          # PAUSE 40 MS TO LET OTHER RUPTS IN.
TC      VARDELAY

CS      FLAGWRDO      # ARE WE IN DESCENT TRAJECTORY?
MASK     R10FLBIT
CCS      A
TCF      +2          # YES.
TC      LADQSAVE      # NO.

CA      DELVS          # HI X OF VELOCITY CORRECTION TERM.
AD      VVECT          # HI X OF UPDATED VELOCITY VECTOR.
TS      ITEMP1          # = VX - DVX M/CS *2(-5).
CA      DELVS +2        # Y
AD      VVECT +1        # Y
TS      ITEMP2          # = VY - DVY M/CS *2(-5)
CA      DELVS +4        # Z
AD      VVECT +2        # Z
TS      ITEMP3          # = VZ - DVZ M/CS *2(-5)
CA      ITEMP1          # COMPUTE VHY, VELOCITY DIRECTED ALONG THE
EXTEND          # Y-COORDINATE.
MP      UHYP          # HI X OF CROSS-RANGE HALF-UNIT VECTOR
XCH      RUPTREG1
CA      ITEMP2
EXTEND
MP      UHYP +2        # Y
```

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ADS	RUPTREG1	# ACCUMULATE PARTIAL PRODUCTS.
CA	ITEMP3	
EXTEND		
MP	UHYP +4	# Z
ADS	RUPTREG1	

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(452b 743)

```

CA      RUPTREG1
DOUBLE
XCH     VHYP
CA      ITEMP1      # VHYP=VMP.UHYP M/CS*2(-5).
EXTEND  # NO COMPUTE VHZ, VELOCITY DIRECTED ALONG
MP      UHYP        # THE Z-COORDINATE.
XCH     RUPTREG1    # HI X OF DOWN-RANGE HALF-UNIT VECTOR.
CA      ITEMP2
EXTEND
MP      UHYP +2      # Y
ADS     RUPTREG1    # ACCUMULATE PARTIAL PRODUCTS.
CA      ITEMP3
EXTEND
MP      UHYP +4      # Z
ADS     RUPTREG1
CA      RUPTREG1
DOUBLE
XCH     VHZ          # VHZ = VMP.UHYP M/CS*2(-5).
CAF     EBANK6        # GET SIN(AOG),COS(AOG) FROM GPMATRIX.
TS      EBANK
EBANK=  M22
CA      M22
TS      ITEMP3
CA      M32
TS      ITEMP4
CAF     EBANK7
TS      EBANK
EBANK=  UNIT/R/
LATFWDV CA      ITEMP4      # COMPUTE LATERAL AND FORWARD VELOCITIES.
EXTEND
MP      VHYP
XCH     RUPTREG1
CA      ITEMP3
EXTEND
MP      VHZ
ADS     RUPTREG1    # = VHYP(COS)AOG+VHZ(SIN)AOG M/CS *2(-5)
CA      VELCONV     # CONVERT LATERAL VELOCITY TO BIT UNITS.
EXTEND
MP      RUPTREG1
DDOUBL
XCH     LATVEL      # LATERAL VELOCITY IN BIT UNITS *2(-14).
CA      ITEMP4      # COMPUTE FORWARD VELOCITY.
EXTEND
MP      VHZ
XCH     RUPTREG1

```

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CA ITEMP3  
EXTEND  
MP VHY  
CS A  
ADS RUPTREG1

# =VHZ(COS)AOG-VHY(SIN)AOG M/CS \*2(-5).



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```

CA      VELCONV      # CONVERT FORWARD VELOCITY TO BIT UNITS.
EXTEND
MP      RUPTREG1
DDOUBL
XCH     FORVEL      # FORWARD VELOCITY IN BIT UNITS *2(-14).

CS      MAXVBITS     # ACC.=-199.9989 FT./SEC.
TS      ITEMP6      # -547 BIT UNITS (OCTAL) AT 0.5571 FPS/BIT

VMONITOR
CAF     ONE          # LOOP TWICE.
TS      ITEMP5       # FORWARD AND LATERAL VELOCITY LANDING
INDEX  ITEMP5       # ANALOG DISPLAYS MONITOR.
CCS     LATVEL
TCF     +4
TCF     LVLIMITS
TCF     +8D
TCF     LVLIMITS
INDEX  ITEMP5
CS      LATVEL
AD      MAXVBITS     # +199.9989 FT.SEC.
EXTEND
BZMF    CHKLASTY
TCF     LVLIMITS
INDEX  ITEMP5
CA      LATVEL
AD      MAXVBITS
EXTEND
BZMF    +2
TCF     LVLIMITS
CHKLASTY
INDEX  ITEMP5
CCS     LATVMETR
TCF     +4
TCF     LASTOK
TCF     +7
TCF     LASTOK
INDEX  ITEMP5
CA      LATVEL
EXTEND
BZMF    LASTPOSY +5
TCF     +5
INDEX  ITEMP5
CS      LATVEL
EXTEND
BZMF    LASTNEGY +4
LASTOK
INDEX  ITEMP5

```

CCS	TRAKLATV
TCF	LASTPOSY
TCF	+2
TCF	LASTNEGY
INDEX	ITEMP5

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	CA	LATVEL
	EXTEND	
	BZMF	NEGVMAXY
	TCF	POSVMAXY
LASTPOSY	INDEX	ITEMP5
	CA	LATVEL
	EXTEND	
	BZMF	+2
	TCF	POSVMAXY
	CS	MAXVBITS
	TCF	ZEROLSTY
POSVMAXY	INDEX	ITEMP5
	CS	LATVMETR
	AD	MAXVBITS
	INDEX	ITEMP5
	XCH	RUPTREG3
	CAF	ONE
	TCF	ZEROLSTY +3
LASTNEGY	INDEX	ITEMP5
	CA	LATVEL
	EXTEND	
	BZMF	NEGVMAXY
	CA	MAXVBITS
	TCF	ZEROLSTY
NEGVMAXY	INDEX	ITEMP5
	CA	LATVMETR
	AD	MAXVBITS
	COM	
	INDEX	ITEMP5
	XCH	RUPTREG3
	CS	ONE
	TCF	ZEROLSTY +3
LVLIMITS	INDEX	ITEMP5
	CCS	TRAKLATV
	TCF	LATVPOS
	TCF	+2
	TCF	LATVNEG
	INDEX	ITEMP5
	CS	LATVMETR
	EXTEND	
	BZMF	+2
	TCF	NEGLMLV
	INDEX	ITEMP5
	CS	LATVEL

```
EXTEND
BZMF    LVMINLM
AD      ITEMP6
INDEX   ITEMP5
AD      LATVMETR
EXTEND
```

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(452b 743)

	BZMF	LVMINLM
	INDEX	ITEMP5
	AD	LATVEL
	EXTEND	
	INDEX	ITEMP5
	SU	LATVMETR
	TCF	ZEROLSTY
LATVPOS	INDEX	ITEMP5
	CS	LATVEL
	EXTEND	
	BZMF	LVMINLM
	TCF	+5
LATVNEG	INDEX	ITEMP5
	CA	LATVEL
	EXTEND	
	BZMF	LVMINLM
	INDEX	ITEMP5
	CS	LATVMETR
	TCF	ZEROLSTY
NEGLMLV	INDEX	ITEMP5
	CA	LATVEL
	EXTEND	
	BZMF	LVMINLM
	CA	MAXVBITS
	INDEX	ITEMP5
	AD	LATVMETR
	COM	
	INDEX	ITEMP5
	AD	LATVEL
	EXTEND	
	BZMF	LVMINLM
	EXTEND	
	INDEX	ITEMP5
	SU	LATVEL
	INDEX	ITEMP5
	AD	LATVMETR
	COM	
	TCF	ZEROLSTY
LVMINLM	INDEX	ITEMP5
	CS	LATVMETR
	INDEX	ITEMP5
	AD	LATVEL
ZEROLSTY	INDEX	ITEMP5
	XCH	RUPTREG3
	CAF	ZERO

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INDEX ITEMP5  
TS TRAKLATV  
INDEX ITEMP5  
CA RUPTREG3  
AD NEG0

# AVOIDS +0 DINC HARDWARE MALFUNCTION

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(452b 743)

```

INDEX  ITEMP5
TS      CDUTCMD
INDEX  ITEMP5
CA      RUPTREG3
INDEX  ITEMP5
ADS     LATVMETR
CCS     ITEMP5      # FIRST MONITOR FORWARD THEN LATERAL VEL.
TCF     VMONITOR

CAF     BITSET      # DRIVE THE X-POINTER DISPLAY.
EXTEND
WOR     CHAN14
TC      LADQSAVE    # GO TO ALTROUT +1 OR TO ALTOUT +1
ZERODATA CAF ZERO    # ZERO ALTSAVE AND ALTSAVE +1
TS      L           # NO NEGATIVE ALTITUDES ALLOWED.
TCF     ZDATA2
```

# \*\*\*\*\*

```
DISPRSET CS  FLAGWRDO      # ARE WE IN DESCENT TRAJECTORY?
MASK     R10FLBIT
EXTEND
BZF      ABORTON          # NO.
CAF      BIT8             # YES.
MASK     IMODES33         # CHECK IF INERTIAL DATA JUST DISPLAYED.
CCS      A
CAF      BIT2             # YES. DISABLE RR ERROR COUNTER
AD       BIT8             # NO. REMOVE DISPLAY INERTIAL DATA
COM
EXTEND
WAND     CHAN12
ABORTON  CS  BITS8/7      # RESET INERTIAL DATA, INTERLEAVE FLAGS.
MASK     IMODES33
TS       IMODES33
CS       DIDFLBIT
MASK     FLAGWRD1
TS       FLAGWRD1        # RESET DIDFLAG.
TCF      TASKOVER
```

# \*\*\*\*\*

```
BITS8/7 OCT 00300      # INERTIAL DATA AND INTERLEAVE FLAGS.
BITSET   =    PRI06
```

# \*\*\*\*\*

## 1.26 findcduw-guidap interface

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```
# PROGRAM NAME:  FINDCDUW
# MOD NUMBER:   1      68-07-15
# MOD AUTHOR:   KLUMPP
#
# OBJECTS OF MOD:      1.      TO SUPPLY COMMANDED GIMBAL ANGLES FOR NOUN 22.
#                      2.      TO MAINTAIN CORRECT AND CURRENT THRUST
#                      DIRECTION DATA IN ALL MODES.  THIS IS DONE BY
#                      FETCHING FOR THE THRUST DIRECTION FILTER THE
#                      CDUD'S IN PNGCS-AUTO, THE CDU'S IN ALL OTHER
#                      MODES.
#                      3.      TO SUBSTITUTE A STOPRATE FOR THE NORMAL
#                      AUTOPILOT COMMANDS WHENEVER
#                      1) NOT IN PNGCS-AUTO, OR
#                      2) ENGINE IS OFF.
#
# FUNCTIONAL DESCRIPTION:
#
# FINDCDUW PROVIDES THE INTERFACES BETWEEN THE VARIOUS POWERED FLITE GUIDANCE PROGRAMS
# AND THE DIGITAL AUTOPILOT.  THE INPUTS TO FINDCDUW ARE THE THRUST COMMAND VECTOR
# AND THE WINDOW COMMAND VECTOR, AND THE OUTPUTS ARE THE GIMBAL ANGLE
# INCREMENTS, THE COMMANDED ATTITUDE ANGLE RATES, AND THE COMMANDED
# ATTITUDE LAG ANGLES (WHICH ACCOUNT FOR THE ANGLES BY WHICH THE BODY WILL
# LAG BEHIND A RAMP COMMAND IN ATTITUDE ANGLE DUE TO THE FINITE ANGULAR
# ACCELERATIONS AVAILABLE).
#
# FINDCDUW ALIGNS THE ESTIMATED THRUST VECTOR FROM THE THRUST DIRECTION
# FILTER WITH THE THRUST COMMAND VECTOR, AND, WHEN XDVINHIB SET,
# ALIGNS THE +Z HALF OF THE LM ZX PLANE WITH THE WINDOW COMMAND VECTOR.
#
```

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# SPECIFICATIONS:

#

# INITIALIZATION:

A SINGLE INTERPRETIVE CALL TO INITCDUW IS REQUIRED  
BEFORE EACH GUIDED MANEUVER USING FINDCDUW.

#

# CALL:

INTERPRETIVE CALL TO FINDCDUW WITH THE THRUST COMMAND  
VECTOR IN MPAC. INTERPRETIVE CALL TO FINDCDUW -2 WITH  
THE THRUST COMMAND VECTOR IN UNFC/2 AND NOT IN MPAC.

#

# RETURNS:

NORMAL INTERPRETIVE IN ALL CASES

#

1. NORMALLY ALL AUTOPILOT CMDS ARE ISSUED.

#

2. IF NOT PNGCS AUTO, DO STOPRATE AND RETURN  
WITHOUT ISSUING AUTOPILOT CMDS.

#

#

3. IF ENGINE OFF, DO STOPRATE AND RETURN WITHOUT  
ISSUING AUTOPILOT CMDS.

#

# ALARMS:

00401 IF INPUTS DETERMINE AN ATTITUDE IN GIMBAL LOCK.  
FINDCDUW DRIVES CDUXD AND CDUYD TO THE RQD VALUES,  
BUT DRIVES CDUZD ONLY TO THE GIMBAL LOCK CONE.

#

#

#

00402 IF UNFC/2 OR UNWC/2 PRODUCE OVERFLOW WHEN  
UNITIZED USING NORMUNIT. FINDCDUW ISSUES  
STOPRATE AS ONLY INPUT TO AUTOPILOT.

#

#

# INPUTS:

UNFC/2 THRUST COMMAND VECTOR, NEED NOT BE SEMI-UNIT  
UNWC/2 WINDOW COMMAND VECTOR, NEED NOT BE SEMI-UNIT  
OGABIAS POSSIBLE BIAS FOR OUTER GIMBAL ANGLE (ZEROED  
XOVINHIB FLAG DENOTING X AXIS OVERRIDE INHIBITED.  
CSMDOCKD FLAG DENOTING CSM DOCKED.  
STEERSW FLAG DENOTING INSUFF THRUST FOR THRUST DIR FI

#

# OUTPUTS:

DELCDEX,Y,Z  
OMEGAPD,+1,+2  
DELPOR,+1,+2  
CPHI,+1,+2 FOR NOUN22

#

#

# DEBRIS:

FINDCDUW DESTROYS SINCDUX,Y,Z AND COSCDUX,Y,Z BY  
WRITING INTO THESE LOCATIONS THE SINES AND COSINES  
OF THE CDUD'S IN PNGCS-AUTO, OF THE CDU'S OTHERWISE.

#

#

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# INITIALIZATION FOR FINDCDUW

BANK 30  
SETLOC FCDUW  
BANK  
  
EBANK= ECDUW  
COUNT\* \$\$/FCDUW

INITCDUW VLOAD  
UNITX  
STORE UNFV/2  
STORE UNWC/2  
RVQ

# FINDCDUW PRELIMINARIES

VLOAD  
UNFC/2  
FINDCDUW BOV SETPD  
FINDCDUW 22  
STQ EXIT  
QCDUWUSR  
  
# FINDCDUW -2: ENTRY WHEN UNFC/2 PRE-STORD  
# INPUT VECTORS NEED NOT BE SEMI-UNIT  
# FINDCDUW: ENTRY WHEN UNFC/2 IN MPAC  
# INTERPRETER NOW INITIALIZED  
# LOCS 0 THRU 21 FOR DIRECTION COSINE MAT  
# SAVE RETURN ADDRESS

# MORE HAUSKEEPING

CA ECDUWL  
XCH EBANK  
TS ECDUWUSR  
  
# SET EBANK  
# SAVE USER'S EBANK

CA DAPBOOLS  
MASK CSMDOCKD  
CCS A  
CA ONE  
TS NDXCDUW  
  
# CSMDOCKD MUST NOT BE BIT15  
# INDEX IF CSM DOCKED

CA XOVINHIB  
TS FLPAUTNO  
  
# XOVINHIB MUST NOT BE BIT15  
# SET TO POS-NON-ZERO FLAG PNGCS AUTO NOT

MASK DAPBOOLS  
TS FLAGOODW  
  
# FLAGOODW = ANY PNZ NUMBER IF XOVI INHIBTD

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# FETCH BASIC DATA

INHINT

# RELINT AT PAUTNO (TC INTPRET)

CA CDUX  
 TS CDUSPOTX  
 CA CDUY  
 TS CDUSPOTY  
 CA CDUZ  
 TS CDUSPOTZ

# FETCH CDUX,CDUY,CDUZ IN ALL CASES, BUT  
 # REPLACE BELOW IF PNGCS AUTO

CA BIT10  
 EXTEND  
 RAND CHAN30  
 CCS A  
 TCF PAUTNO

# PNGCS CONTROL BIT

CA BIT14  
 EXTEND  
 RAND CHAN31  
 CCS A  
 TCF PAUTNO

# NOT PNGCS (BITS INVERTED)

# AUTO MODE BIT

# NOT AUTO (BITS INVERTED)

TS FLPAUTNO

# RESET FLAG PNGCS AUTO NOT

CA CDUXD  
 TS CDUSPOTX  
 CA CDUYD  
 TS CDUSPOTY  
 CA CDUZD  
 TS CDUSPOTZ

# PNGCS AUTO: FETCH CDUXD,CDUYD,CDUZD

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# FETCH INPUTS

PAUTNO

TC

INTPRET

# ENTERING THRUST CMD STILL IN MPAC

RTB

STOVL UNX/2

# SEMI-UNIT THRUST CMD AS INITIAL UNX/2

UNWC/2

RTB RTB

NORMUNIT

QUICTRIG

# ALWAYS RQD TO OBTAIN TRIGS OF CDUD'S

STOVL UNZ/2

# SEMI-UNIT WINDOW CMD AS INITIAL UNZ/2

DELV

BOVB UNIT

NOATTCNT

# AT LEAST ONE ENTERING CMD VCT ZERO

BOV CALL

AFTNFLTR

# IF UNIT DELV OVERFLOWS SKIP FILTER

\*SMNB\*

# YIELDS UNIT(DELV) IN VEH COORDS FOR FLTR

# THRUST DIRECTION FILTER

EXIT

CA UNFVY/2

# FOR RESTARTS, UNFV/2 ALWAYS INTACT, MPAC

LXCH MPAC +3

# RENEWD AFTER RETURN FROM CALLER,

TC FLTRSUB

# TWO FILTER UPDATES MAY BE DONE.

TS UNFVY/2

# UNFV/2 NEED NOT BE EXACTLY SEMI-UNIT.

CA UNFVZ/2

LXCH MPAC +5

TC FLTRSUB

TS UNFVZ/2

TC INTERPRET

# COMPLETES FILTER

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# FIND A SUITABLE WINDOW POINTING VECTOR

AFTRFLTR	SLOAD	BHIZ	# IF XOY NOT INHIBITED, GO FETCH ZNB
		FLAGOODW	
		FETCHZNB	
	VLOAD	CALL	
		UNZ/2	
		UNWCTEST	

FETCHZNB	VLOAD	
		ZNBPIP
	STCALL	UNZ/2
		UNWCTEST

VLOAD	VCOMP	# Z AND -X CAN'T BOTH PARALLEL UNFC/2
	XNBPIP	
STORE	UNZ/2	

# COMPUTE THE REQUIRED DIRECTION COSINE MATRIX

DCMCL	VLOAD	VXV	
		UNZ/2	
		UNX/2	
	UNIT	PUSH	# UNY/2 FIRST ITERATION
	VXV	VSL1	
		UNX/2	
	STORE	UNZ/2	# -UNZ/2 FIRST ITERATION
	VXSC	PDVL	# EXCHANGE -UNFVZ/2 UNZ/2 FOR UNY/2
		UNFVZ/2	# MUST BE SMALL
	VXSC	BVSU	# YIELDS -UNFVY/2 UNY/2-UNFVZ/2 UNZ/2
		UNFVY/2	# MUST BE SMALL
	VSL1	VAD	
		UNX/2	
	UNIT		# TOTALLY ELIMINATES THRUST POINTING ERROR
	STORE	UNX/2	# UNX/2
	VXV	VSL1	
		UNZ/2	# -UNZ/2 WAS STORED HERE REMEMBER
	STORE	UNY/2	# UNY/2
	VCOMP	VXV	
		UNX/2	
	VSL1		
	STORE	UNZ/2	# UNZ/2

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# COMPUTES THE REQUIRED GIMBAL ANGLES

CALL

NB2CDUSP

# YIELDS THE RQD GIMBAL ANGLES, 2'S, PI

EXIT

# LIMIT THE MIDDLE GIMBAL ANGLE & COMPUTE THE UNLIMITED GIMBAL ANGLE CHGS

CA MPAC +2

# LIMIT THE MGA

TS L

# CAN'T LXCH: NEED UNLIMITED MGA FOR ALARM

CA CDUZDLIM

TC LIMITSUB

# YIELDS LIMITED MGA. 1 BIT ERROR POSSIBLE

XCH MPAC +2

# BECAUSE USING 2'S COMP. WHO CARES?

EXTEND

MSU MPAC +2

# THIS BETTER YIELD ZERO

EXTEND

BZF +2

TCF ALARMMGA

MGARET

INHINT

# RELINT AT TC INTPRET AFTER TCQCUDW

ZL

CA TWO

DELGMBLP

TS TEM2

CA L

# TO PREVENT FALSE STARTS ABOUT X, ZERO

EXTEND

# FLAGOODW IF DELGMBZ OR Y TOO BIG.

SQUARE

AD HI5

# WITHIN 1 BIT OF -(45 DEG SQUARED)

EXTEND

BZMF +3

CA ZERO

TS FLAGOODW

INDEX TEM2

CA MPAC

INDEX TEM2

TS CPHI

# OUTPUTS TO NOUN22

EXTEND

INDEX TEM2

MSU CDUXD

# NO MATTER THAT THESE SLIGHTLY DIFFERENT

COM

# FROM WHEN WE INITIALLY FETCHED THEM

INDEX TEM2

TS -DELGMB

# -UNLIMITED GIMBAL ANGLE CHGS, 1'S, PI

TS L

# FOR PRECEDING TEST ON NEXT LOOP PASS

CCS TEM2  
TCF DELGMBLP

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# BRANCHES TO NOATTCNT

CCS FLPAUTNO  
TCF NOATTCNT +2 # NO PNGCS AUTO

CA FLAGWRD5  
MASK ENGONBIT  
EXTEND  
BZF NOATTCNT +2 # ENGINE NOT ON



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(472 731)

# LIMIT THE ATTITUDE ANGLE CHANGES

#

# THIS SECTION LIMITS THE ATTITUDE ANGLE CHANGES ABOUT A SET OF ORTHOGONAL VEHICLE AXES X,YPRIME

# THESE AXES COINCIDE WITH THE COMMANDED VEHICLE AXES IF AND ONLY IF CDUXD IS ZERO. THE PRIME

# THE COMMANDED VEHICLE SYSTEM ROTATED ABOUT THE X AXIS TO BRING THE Z AXIS INTO ALIGNMENT WITH

# AXIS. ATTITUDE ANGLE CHANGES IN THE PRIME SYSTEM ARE RELATED TO SMALL GIMBAL ANGLE CHANGES BY

#

# [ -DELATTX ] [ 1 SIN(CDUZD) 0 ] [ -DELGMBX ]

# [ ] [ ] [ ]

# [ -DELATTYPRIME ] = [ 0 COS(CDUZD) 0 ] [ -DELGMBY ]

# [ ] [ ] [ ]

# [ -DELATTZPRIME ] [ 0 0 1 ] [ -DELGMBZ ]

LXCH -DELGMB +2 # SAME AS -DELATTZPRIME UNLIMITED

INDEX NDXCDUW

CA DAZMAX

TC LIMITSUB

TS -DELGMB +2 # -DELGMBZ

CA -DELGMB +1

EXTEND

MP COSCDUZ # YIELDS -DELATTYPRIME/2 UNLIMITED

TS L

INDEX NDXCDUW

CA DAY/2MAX

TC LIMITSUB

EXTEND

DV COSCDUZ

XCH -DELGMB +1 # -DELGMBY, FETCHING UNLIMITED VALUE

EXTEND

MP SINCDUZ

DDOUBL

COM

EXTEND # YIELDS +DELATTX UNLIMITD, MAG &lt; 180 DEG.

MSU -DELGMB # BASED ON UNLIMITED DELGMBV.

TS L # ONE BIT ERROR IF OPERANDS IN MSU

INDEX NDXCDUW # OF MIXED SIGNS. WHO CARES?

CA DAXMAX

TC LIMITSUB

TS -DELGMB # SAVE LIMITED +DELATTX

CCS FLAGOODW

CS -DELGMB # FETCH IT BACK CHGNG SIGN IF WINDOW GOOD

TS -DELGMB # OTHERWISE USE ZERO FOR -DELATTX

CS -DELGMB +1

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EXTEND

MP SINCDUZ

DDOUBL

ADS -DELGMB

# YIELDS -CNTRIB TO -DELATTX FROM -DELGMBY

# -DELGMBX. NO OVERFLOW SINCE LIMITED TO

#  $20\text{DEG}(1+\sin(70\text{DEG})/\cos(70\text{DEG})) < 180\text{DEG}$

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# COMPUTE COMMANDED ATTITUDE RATES

#

# [ OMEGAPD ] [ -2 -4 SINCDUZ +0 ] [ -DELGMBZ ]

# [ ] [ ] [ ]

# [ OMEGAQD ] = [ +0 -8 COSCDUZ COSCDUX -4 SINCDUX ] [ -DELGMBY ]

# [ ] [ ] [ ]

# [ OMEGARD ] [ +0 +8 COSCDUZ SINCDUX -4 COSCDUX ] [ -DELGMBZ ]

#

# ATTITUDE ANGLE RATES IN UNITS OF PI/4 RAD/SEC = K TRIG FCNS IN UNITS OF 2 X GIMBAL ANGLE RATE

# PI/2 RAD/SEC. THE CONSTANTS ARE BASED ON DELGMB BEING THE GIMBAL ANGLE CHANGES IN UNITS OF P

# AND 2 SECONDS BEING THE COMPUTATION PERIOD (THE PERIOD BETWEEN SUCCESSIVE PASSES THRU FINDCDU

CS -DELGMB

TS OMEGAPD

CS -DELGMB +1

EXTEND

MP SINCDUZ

DDOUBL

ADS OMEGAPD

ADS OMEGAPD

CS -DELGMB +1

EXTEND

MP COSCDUX

DDOUBL

EXTEND

MP COSCDUZ

TS OMEGAQD

CS -DELGMB +2

EXTEND

MP SINCDUX

ADS OMEGAQD

ADS OMEGAQD

ADS OMEGAQD

CA -DELGMB +1

EXTEND

MP SINCDUX

DDOUBL

EXTEND

MP COSCDUZ

TS OMEGARD

CS -DELGMB +2

EXTEND

MP COSCDUX

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ADS	OMEGARD
ADS	OMEGARD
ADS	OMEGARD

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# FINAL TRANSFER

(472 731)

```
CDUWXFR      CA      TWO
              TS      TEM2
              INDEX   TEM2
              CA      -DELGMB
              EXTEND
              MP      DT/DELT      # RATIO OF DAP INTERVAL TO CDUW INTERVAL
              TC      ONESTO2S
              INDEX   TEM2
              TS      DELCDUX      # ANGLE INTERFACE

              INDEX   TEM2
              CCS      OMEGAPD
              AD      ONE
              TCF      +2
              AD      ONE
              EXTEND      # WE NOW HAVE ABS(OMEGAPD,QD,RD)
              INDEX   TEM2
              MP      OMEGAPD
              EXTEND
              MP      BIT11      # 1/16
              EXTEND
              INDEX   TEM2      #
              DV      1JACC      # UNITS PI/4 RAD/SEC
              TS      L
              CA      DELERLIM
              TC      LIMITSUB
              INDEX   TEM2
              TS      DELPEROR      # LAG ANGLE = OMEGA ABS(OMEGA)/2 ACCEL
              CCS      TEM2
              TCF      CDUWXFR

# HAUSKEEPING AND RETURN

TCQCDUW      CA      ECDUWUSR
              TS      EBANK      # RETURN USER'S EBANK

              TC      INTPRET
              SETPD   GOTO
                      0
                      QCDUWUSR      # NORMAL AND ABNORMAL RETURN TO USER
```

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(472 731)

# THRUST VECTOR FILTER SUBROUTINE

```

FLTRSUB      EXTEND
              QXCH   TEM2
              TS      TEM3      # SAVE ORIGINAL OFFSET
              COM     # ONE MCT, NO WDS, CAN BE SAVED IF NEG OF
              AD      L        # ORIG OFFSET ARRIVES IN A, BUT IT'S
              EXTEND  # NOT WORTH THE INCREASED OBSCURITY.
              INDEX   NDXCDUW
              MP      GAINFLTR
              TS      L        # INCR TO OFFSET, UNLIMITED
              CA      DUNFVLIM # SAME LIMIT FOR Y AND Z
              TC      LIMITSUB # YIELDS INCR TO OFFSET, LIMITED
              AD      TEM3     # ORIGINAL OFFSET
              TS      L        # TOTAL OFFSET, UNLIMITED
              CA      UNFVLIM  # SAME LIMIT FOR Y AND Z
              TC      LIMITSUB # YIELDS TOTAL OFFSET, LIMITED
              TC      TEM2

```

# SUBR TO TEST THE ANGLE BETWEEN THE PROPOSED WINDOW AND THRUST CMD VCTS

```

UNWCTEST     DOT      DSQ
              UNX/2
              DSU      BMN
              DOTSWFMX
              DCMCL
              SSP      RVQ      # RVQ FOR ALT CHOICE IF DOT MAGN TOO LARGE
              FLAGOODW # ZEROING WINDOW GOOD FLAG
              0

```

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# NB2CDUSP RETURNS THE 2'S COMPLEMENT, PI, SP CDU ANGLES X,Y,Z IN MPAC,+1,+2 GIVEN THE MATRIX W  
 # ARE THE SEMI-UNIT NAV BASE VECTORS X,Y,X EXPRESSED IN STABLE MEMBER COORDINATES, LOCATED AT C  
 # NB2CDUSP USES THE ARCTRGSP WHICH HAS A MAXIMUM ERROR OF +-4 BITS.

```

NB2CDUSP      DLOAD   DSQ
                2
      BDSU     BPL
                DP1/4TH
                +3
      DLOAD
      ZEROVECS      # IN CASE SIN WAS SLIGHTLY > 1/2
      SQRT      EXIT      # YIELDS COS(CDUZ) IN UNITS OF 2

      EXTEND
      DCA      MPAC
      DDOUBL
      TS      TEM5
      TCF      +3
      CA      POSMAX      # OVERFLOW.  FETCH POSMAX, MPAC ALWAYS POS
      TS      TEM5      # COS(CDUZ) IN TEM5, UNITS 1

      INDEX   FIXLOC
      CA      2
      LXCH    MPAC
      TC      ARCTRGSP
      TS      MPAC +2      # CDUZ

      CA      ZERO
      TC      DVBYCOSM
      CA      FOUR
      TC      DVBYCOSM
      CS      TEM1
      TC      ARCTRGSP
      TS      MPAC +1      # CDUY

      CA      BIT4
      TC      DVBYCOSM
      CA      16OCT
      TC      DVBYCOSM
      CS      TEM1
      TC      ARCTRGSP
      TS      MPAC      # CDUX

      TC      INTERPRET
      RVQ

```

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(472 731)

# THE ELEMENTS OF THE NAV BASE MATRIX WHICH WE MUST DIVIDE BY COS(MGA)  
 # ALREADY CONTAIN COS(MGA)/2 AS A FACTOR. THEREFORE THE QUOTIENT SHOULD  
 # ORDINARILY NEVER EXCEED 1/2 IN MAGNITUDE. BUT IF THE MGA IS NEAR PI/2  
 # THEN COS(MGA) IS NEAR ZERO, AND THERE MAY BE SOME CHAFF IN THE OTHER  
 # ELEMENTS OF THE MATRIX WHICH WOULD PRODUCE CHAOS UNDER DIVISION.  
 # BEFORE DIVIDING WE MAKE SURE COS(MGA) IS AT LEAST ONE BIT LARGER  
 # THAN THE MAGNITUDE OF THE HIGH ORDER PART OF THE OPERAND.  
 #  
 # IF ONE OR MORE DIVIDES CANNOT BE PERFORMED, THIS MEANS THAT THE  
 # REQUIRED MGA IS VERY NEARLY  $\pm\pi/2$  AND THEREFORE THE OTHER GIMBAL  
 # ANGLES ARE INDETERMINATE. THE INNER AND OUTER GIMBAL ANGLES RETURNED  
 # IN THIS CASE WILL BE RANDOM MULTIPLES OF  $\pi/2$ .

DVBYCOSM	AD	FIXLOC	
	TS	ADDRWD	# ADRES OF OPERAND
	INDEX	ADDRWD	# FETCH NEG ABS OF OPERAND, AD TEM5, AND
	CA	0	# SKIP DIVIDE IF RESULT NEG OR ZERO
	EXTEND		
	BZMF	+2	
	COM		
	AD	TEM5	# C(A) ZERO OR NEG, C(TEM5) ZERO OR POS
	EXTEND		
	BZMF	TSL&TCQ	# DIFFERENCE ALWAYS SMALL IF BRANCH
	EXTEND		# TEM5 EXCEEDS ABS HIGH ORDER PART OF
	INDEX	ADDRWD	# OPERAND BY AT LEAST ONE BIT.
	DCA	0	# THEREFORE IT EXCEEDS THE DP OPERAND
	EXTEND		# AND DIVISION WILL ALWAYS SUCCEED.
	DV	TEM5	
TSL&TCQ	TS	L	
	LXCH	TEM1	
	TC	Q	



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(472 731)

# ARCTRGSP RETURNS THE 2'S COMPLEMENT, PI, SP ANGLE IN THE A REGISTER GIVEN ITS SINE IN A AND I  
 # UNITS OF 2. THE RESULT IS AN UNAMBIGUOUS ANGLE ANYWHERE IN THE CIRCLE, WITH A MAXIMUM ERROR  
 # THE ERROR IS PRODUCED BY THE SUBROUTINE SPARCSIN WHICH IS USED ONLY IN THE REGION +-45 DEGREE

```

ARCTRGSP      EXTEND
               BZF      SINZERO      # TO AVOID DIVIDING BY ZERO

               EXTEND
               QXCH     TEM4
               TS       TEM2
               CA       L
               TS       TEM3
               CA       ZERO
               EXTEND
               DV       TEM2
               EXTEND
               BZF      USECOS

               CCS      TEM3          # SIN IS SMALLER OR EQUAL
               CA       ZERO
               TCF      +4
               CS       TEM2          # IF COS NEG, REVERSE SIGN OF SIN,
               TS       TEM2          #      ANGLE = PI-ARCSIN(SIN)
               CA       NEGMAX        # PICK UP PI, 2'S COMPLEMENT
               TS       TEM3          # WE NO LONGER NEED COS
               CA       TEM2
               TC       SPARCSIN -1
               TC       ONESTO2S
               EXTEND
               MSU      TEM3
1T02&TCQ      TC       ONESTO2S
               TC       TEM4

USECOS        CS       TEM3          # COS IS SMALLER
               TC       SPARCSIN -1  # ANGLE = SIGN(SIN)(FI/2-ARCSIN(COS))
               AD       HALF
               TS       TEM3          # WE NO LONGER NEED COS
               CCS      TEM2
               CA       TEM3
               TCF      1T02&TCQ
               CS       TEM3
               TCF      1T02&TCQ

SINZERO       CCS      L
               CA       ZERO

```

TC	Q	
CA	NEGMAX	# PI, 2'S COMP
TC	Q	

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(472 731)

# SPARCSIN TAKES AN ARGUMENT SCALED UNITY IN A AND RETURNS AN ANGLE SCALED  
 # 180 DEGREES IN A. IT HAS BEEN UNIT TESTED IN THE REGION  $\pm .94$  ( $\pm 70$   
 # DEGREES) AND THE MAXIMUM ERROR IS  $\pm 5$  BITS WITH AN AVERAGE TIME OF  
 # 450 MICROSECONDS. SPARCSIN -1 TAKES THE ARGUMENT SCALED TWO. (BOB CRISP)

	DOUBLE	
SPARCSIN	TS	SR
	TCF	+4
	INDEX	A
	CS	LIMITS
	TS	SR
	EXTEND	
	MP	A
	TS	TEM1
	EXTEND	
	MP	DPL9
	AD	DPL7
	EXTEND	
	MP	TEM1
	AD	DPL5
	EXTEND	
	MP	TEM1
	AD	DPL3
	EXTEND	
	MP	TEM1
	AD	DPL1
	EXTEND	
	MP	SR
	TC	Q
DPL1	DEC	10502
DPL3	DEC	432
DPL5	DEC	7300
DPL7	DEC	-11803
DPL9	DEC	8397

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```

# LIMITSUB LIMITS THE MAGNITUDE OF THE POSITIVE OR NEGATIVE VARIABLE
# ARRIVING IN L TO THE POSITIVE LIMIT ARRIVING IN A.
# THE SIGNED LIMITED VARIABLE IS RETURNED IN A.
#
# VERSION COUTESY HUGH BLAIR-SMITH

```

```

LIMITSUB      TS      TEM1
               CA      ZERO
               EXTEND
               DV      TEM1
               CCS      A
               LXCH     TEM1
               TCF      +2
               TCF      +3
               CA      L
               TC      Q
               CS      TEM1
               TC      Q

```

```

# SUBROUTINE TO CONVERT 1'S COMP SP TO 2'S COMP

```

```

ONESTO2S      CCS      A
               AD      ONE
               TC      Q
               CS      A
               TC      Q

```

```

# NO ATTITUDE CONTROL

```

```

NOATTCNT      TC      ALARM
               OCT     00402

```

```

# NO ATTITUDE CONTROL

```

```

+2            INHINT
               TC      IBNKCALL
               FCADR   STOPRATE
               TCF     TCQCDUW

```

```

# COME HERE FOR NOATTCNT WITHOUT ALARM

```

```

# RELINT AT TC INTPRET AFTER TCQCDUW

```

```

# RETURN TO USER SKIPPING AUTOPILOT CMDS

```

```

# MIDDLE GIMBAL ANGLE ALARM

```

```

ALARMMGA      TC      ALARM
               OCT     00401
               TCF     MGARET

```

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\*\*\*\*\*

## # CONSTANTS

\*\*\*\*\*

## # ADDRESS CONSTANTS

ECDUWL            ECADR    ECDUW

## # THRUST DIRECTION FILTER CONSTANTS

GAINFLTR	DEC	.2	# GAIN FILTER SANS CSM
	DEC	.1	# GAIN FILTER WITH CSM

DUNFVLIM	DEC	.007 B-1	# 7 MR MAX CHG IN F DIR IN VEH IN 2 SECS.
			# THIS DOES NOT ALLOW FOR S/C ROT RATE.

UNFVLIM	DEC	.129 B-1	# 129 MR MAX THRUST OFFSET. 105 MR TRAVEL
			# +10MR DEFL+5MR MECH MOUNT+9MR ABLATION.

## # CONSTANT RELATED TO GIMBAL ANGLE COMPUTATIONS

DOTSWFMX	DEC	.93302 B-4	# LIM COLNRTY OF UNWC/2 & UNFC/2 TO 85 DEG
			# LOWER PART COMES FROM NEXT CONSTANT

DAXMAX	DEC	.1111111111	# DELATTX LIM TO 20 DEG IN 2 SECS, 1'S, PI
	DEC	.0111111111	# 2 DEG WHEN CSM DOCKED

DAY/2MAX	DEC	.0555555555	# LIKEWISE FOR DELATTY
	DEC	.0055555555	

DAZMAX	=	DAXMAX	# LIKEWISE FOR DELATTZ
--------	---	--------	------------------------

CDUZDLIM	DEC	.3888888888	# 70 DEG LIMIT FOR MGA, 1'S, PI
----------	-----	-------------	---------------------------------

## # CONSTANTS FOR DATA TRANSFER

DT/DELT	DEC	.05	# .1 SEC/2 SEC WHICH IS THE AUTOPILOT
			# CONTROL SAMPLE PERIOD/COMPUTATION PERIOD

DELERLIM	=	DAY/2MAX	# 0 DEG LIMIT FOR LAG ANGLES, 1'S, PI
----------	---	----------	---------------------------------------

# *** END OF FLY		.132 ***	
------------------	--	----------	--

1.27 lm down-telemetry program

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```

# PROGRAM NAME -- DOWN TELEMETRY PROGRAM
# MOD NO. -- 0          TO COMPLETELY REWRITE THE DOWN TELEMETRY PROGRAM AND DOWNLINK
#                      PURPOSE OF SAVING APPROXIMATELY 150 WORDS OF CORE STORAGE.
#                      THIS CHANGE REQUIRES AN ENTIRELY NEW METHOD OF SPECIFYING DOWN
#                      LISTS LOG SECTION FOR MORE DETAILS.  HOWEVER THIS CHANGE WILL
#                      OF DOWN TELEMETRY DATA.
# MOD BY -- KILROY, SMITH, DEWITT
# DATE -- 02 OCT 67
# AUTHORS -- KILROY, SMITH, DWWITT, DEWOLF, FAGIN
# LOG SECTION -- DOWN-TELEMETRY PROGRAM
#
# FUNCTIONAL DESCRIPTION -- THIS ROUTINE IS INITIATED BY TELEMETRY END
# PULSE FROM THE DOWNLINK TELEMETRY CONVERTER.  THIS PULSE OCCURS
# AT 50 TIMES PER SEC (EVERY 20 MS) THEREFORE DODOWNTM IS
# EXECUTED AT THESE RATES.  THIS ROUTINE SELECTS THE APPROPRIATE
# AGC DATA TO BE TRANSMITTED DOWNLINK AND LOADS IT INTO OUTPUT
# CHANNELS 34 AND 35.  THE INFORMATION IS THEN GATED OUT FROM THE
# LGC IN SERIAL FASHION.
#
# THIS PROGRAM IS CODED FOR A 2 SECOND DOWNLIST.  SINCE DOWNRUPTS
# OCCUR EVERY 20 MS AND 2 AGC COMPUTER WORDS CAN BE PLACED IN
# CHANNELS 34 AND 35 DURING EACH DOWNRUPT THE PROGRAM IS CAPABLE
# OF SENDING 200 AGC WORDS EVERY 2 SECONDS.
#
# CALLING SEQUENCE -- NONE
# PROGRAM IS ENTERED VIA TCF DODOWNTM WHICH IS EXECUTED AS A
# RESULT OF A DOWNRUPT.  CONTROL IS RETURNED VIA TCF RESUME WHICH
# IN EFFECT IS A RESUME.
#
# SUBROUTINES CALLED -- NONE
#
# NORMAL EXIT MODE -- TCF RESUME
#
# ALARM OR ABORT EXIT MODE -- NONE
#
# RESTART PROTECTION:
# ON A FRESH START AND RESTART THE 'STARTSUB' SUBROUTINE WILL INITIALIZE THE DOWN
# DNTMGOTO) TO THE BEGINNING OF THE CURRENT DOWNLIST (I.E., CURRENT CONTENTS OF
# EFFECT OF IGNORING THE REMAINDER OF THE DOWNLIST WHICH THE DOWN-TELEMETRY PROGRAM
# THE RESTART (OR FRESH START) OCCURRED AND RESUME DOWN TELEMETRY FROM THE BEGINNING
# DOWNLIST.
#
# ALSO OF INTEREST IS THE FACT THAT ON A RESTART THE AGC WILL ZERO DOWNLINK CHANNELS
#
# DOWNLINK LIST SELECTION:

```

```
# THE APPROPRIATE DOWNLINK LISTS ARE SELECTED BY THE FOLLOWING:
# 1. FRESH START
# 2. V37EXXE WHERE XX = THE MAJOR MODE BEING SELECTED.
# 3. UPDATE PROGRAM (P27)
# 4. NON-V37 SELECTABLE TYPE PROGRAMS (E.G., AGS INITIALIZATION (SUNDANCE, LUMINARY)
# TRANSITION (COLOSSUS) ETC.).
#
# DOWNLINK LIST RULES AND LIMITATIONS:
# READ SECTION(S) WHICH FOLLOW 'DEBRIS' WRITEUP.
#
# OUTPUT -- EVERY 2 SECONDS 100 DOUBLE PRECISION WORDS (I.E., 200 LGC
# COMPUTER WORDS) ARE TRANSMITTED VIA DOWNLINK.
#
# ERASABLE INITIALIZATION REQUIRED -- NONE
# 'DNTMGOTO' AND 'DNLSTADR' ARE INITIALIZED BY THE FRESH START PROGRAM.
#
# DEBRIS (ERASABLE LOCATIONS DESTROYED BY THIS PROGRAM) --
# LDATA1ST, DNTMBUFF TO DNTMBUFF +21D, TMINDEX, DNQ.
```

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# DODOWNTM IS ENTERED EVERY 20 MS BY AN INTERRUPT TRIGGERED BY THE
# RECEIPT OF AN ENDPULSE FROM THE SPACECRAFT TELEMETRY PROGRAMMER.
#
# NOTES REGARDING DOWNLINK LISTS ASSOCIATED WITH THIS PROGRAM:
# 1.  DOWNLISTS.  DOWNLISTS MUST BE COMPILED IN THE SAME BANK AS THE
#     DOWN-TELEMETRY PROGRAM.  THIS IS DONE FOR EASE OF CODING, FASTER
#     EXECUTION.
# 2.  EACH DOWNLINK LIST CONSISTS OF A CONTROL LIST AND A NUMBER OF
#     SUBLISTS.
# 3.  A SUBLIST REFERS TO A SNAPSHOT OR DATA COMMON TO THE SAME OR OTHER
#     DOWNLINK LISTS.  ANY SUBLIST CONTAINING COMMON DATA NEEDS TO BE
#     CODED ONLY ONCE FOR THE APPLICABLE DOWNLINK LISTS.
# 4.  SNAPSHOT SUBLISTS REFER SPECIFICALLY TO HOMOGENEOUS DATA WHICH MUST BE
#     SAVED IN A BUFFER DURING ONE DOWNRUPT.
# 5.  THE 1DNADR FOR THE 1ST WORD OF SNAPSHOT DATA IS FOUND AT THE END
#     OF EACH SNAPSHOT SUBLIST, SINCE THE PROGRAM CODING SENDS THIS DP WORD
#     IMMEDIATELY AFTER STORING THE OTHERS IN THE SNAPSHOT BUFFER.
# 6.  ALL LISTS ARE COMBINATIONS OF CODED ERASABLE ADDRESS CONSTANTS
#     CREATED FOR THE DOWNLIST PROGRAM.
#     A.      1DNADR              1-WORD DOWNLIST ADDRESS.
#             SAME AS ECADR, BUT USED WHEN THE WORD ADDRESSED IS THE LEFT
#             HALF OF A DOUBLE-PRECISION WORD FOR DOWN TELEMETRY.
#     B.      2DNADR - 6DNADR      N-WORD DOWNLIST ADDRESS, N = 2 - 6.
#             SAME AS 1DNADR, BUT WITH THE 4 UNUSED BITS OF THE ECADR FORMAT
#             FILLED IN WITH 0001-0101.  USED TO POINT TO A LIST OF N DOUBLE-
#             PRECISION WORDS, STORED CONSECUTIVELY, FOR DOWN TELEMETRY.
#     C.      DNCHAN              DOWNLIST CHANNEL ADDRESS.
#             SAME AS 1DNADR, BUT WITH PREFIX BITS 0111.  USED TO POINT TO
#             A PAIR OF CHANNELS FOR DOWN TELEMETRY.
#     D.      DNPTR              DOWN-TELEMETRY SUBLIST POINTER.
#             SAME AS CAF BUT TAGGED AS A CONSTANT.  USED IN CONTROL LIST TO POINT
#             CAUTION --- A DNPTR CANNOT BE USED IN A SUBLIST.
# 7.  THE WORD ORDER CODE IS SET TO ZERO AT THE BEGINNING OF EACH DOWNLIST (I.E.,
#     A '1DNADR TIME2' IS DETECTED IN THE CONTROL LIST (ONLY).
# 8.  IN THE SNAPSHOT SUBLIST ONLY, THE DNADR'S CANNOT POINT TO THE FIRST WORD OF A
#
# DOWNLIST LIST RESTRICTIONS:
# (THE FOLLOWING POINTS MAY BE LISTED ELSEWHERE BUT ARE LISTED HERE SO IT IS CLEAR TH
# DONE)
# 1.  SNAPSHOT DOWNLIST:
#     (A) CANNOT CONTAIN THE FOLLOWING ECADRS (I.E., 1DNADR'S): Q, 400, 1000, 1400.
#     (B) CAN CONTAIN ONLY 1DNADR'S
# 2.  ALL DOWNLINKED DATA (EXCEPT CHANNELS) IS PICKED UP BY A DCA SO DOWNLINK LISTS
#     EQUIVALENT OF THE FOLLOWING ECADRS (I.E., 1DNADRS): 377, 777, 1377, 1777, 2377
#     (NOTE: THE TERM 'EQUIVALENT' MEANT THAT THE 1DNADR TO 6DNADR WILL BE PROCESSI

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# 3. CONTROL LISTS AND SUBLISTS CANNOT HAVE ENTRIES = OCTAL 00000 OR OCTAL 77777

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# 4. THE '1DNADR TIME2' WHICH WILL CAUSE THE DOWNLINK PROGRAM TO SET THE WORDER C
# CONTROL SECTION OF THE DOWNLIST.
# 5. 'DNCHAN 0' CANNOT BE USED.
# 6. 'DNPTR 0' CANNOT BE USED.
# 7. DNPTR CANNOT APPEAR IN A SUBLIST.
#
# EBANK SETTINGS
# IN THE PROCESS OF SETTING THE EBANK (WHEN PICKING UP DOWNLINK DATA) THE DOWN
# 'GARBAGE' INTO BITS15-12 OF EBANK. HUGH BLAIR-SMITH WARNS US THAT BITS15-12
# SIGNIFICANT SOMEDAY IN THE FUTURE. IF/WHEN THAT HAPPENS, THE PROGRAM SHOULD
# THAT BITS 15-12 OF EBANK ARE ZERO.
#
# INITIALIZATION REQUIRED -- TO INTERRUPT CURRENT LIST AND START A NEW ONE.
# 1. ADRES OF DOWNLINK LIST INTO DNLSTADR
# 2. NEGONE INTO SUBLIST
# 3. NEGONE INTO DNECADR

BANK 22
SETLOC DOWNTLM
BANK

EBANK= DNTMBUFF

COUNT* $$/DPROG
DODOWNTM TS BANKRUPT
EXTEND
QXCH QRUPT # SAVE Q
TCF WOTEST
W01 EXTEND # SET WORD ORDER BIT TO 1 ONLY IF IT
WOR CHAN13 # ALREADY ISN'T
TC DNTMGOTO # GOTO APPROPRIATE PHASE OF PROGRAM

DNPHASE1 CA NEGONE # INITIALIZE ALL CONTROL WORDS
TS SUBLIST # WORDS TO MINUS ONE
TS DNECADR
CA LDNPHAS2 # SET DNTMGOTO = 0 ALL SUSEQUENT DOWRUPTS
TS DNTMGOTO # GO TO DNPHASE2
TCF NEWLIST
DNPHASE2 CCS DNECADR # SENDING OF DATA IN PROGRESS
DODNADR TC FETCH2WD # YES -- THEN FETCH THE NEXT 2 SP WORDS
MINTIME2 -1DNADR TIME2 # NEGATIVE OF TIME2 1DNADR
TCF +1 # (ECADR OF 3776 + 74001 = 77777)

CCS SUBLIST # IS THE SUBLIST IN CONTROL
TCF NEXTINSL # YES

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      DNADRDCR      OCT      74001      # DNADR COUNT AND ECADR DECREMENTER

      CHKLIST      CA      CTLIST
      EXTEND
      BZMF      NEWLIST      # IT WILL BE NEGATIVE AT END OF LIST
      TCF      NEXTINCL
      NEWLIST      INDEX      DNLSTCOD
      CA      DNTABLE      # INITIALIZE CTLIST WITH
      TS      CTLIST      #      STARTING ADDRESS OF NEW LIST
      CS      DNLSTCOD
      TCF      SENDID      +3
      NEXTINCL      INDEX      CTLIST
      CA      0
      CCS      A
      INCR      CTLIST      # SET POINTER TO PICK UP NEXT CTLIST WORD
      TCF      +4      # ON NEXT ENTRY TO PROG. (A SHOULD NOT =0)
      XCH      CTLIST      # SET CTLIST TO NEGATIVE AND PLACE(CODING)
      COM      # UNCOMPLEMENTED DNADR INTO A. (FOR LA)
      XCH      CTLIST      # (ST IN )
      +4      INCR      A      # (CTLIST)
      TS      DNECADR      # SAVE DNADR
      AD      MINTIME2      # TEST FOR TIME2 (NEG. OF ECADR)
      CCS      A
      TCF      SETWO      +1      # DON'T SET WORD ORDER CODE
      MINB1314      OCT      47777      # MINUS BIT 13 AND 14 (CAN'T GET HERE)
      TCF      SETWO      +1      # DON'T SET WORD ORDER CODE
      SETWO      TC      WOZERO      # GO SET WORD ORDER CODE TO ZERO.
      +1      CA      DNECADR      # RELOAD A WITH THE DNADR.
      +2      AD      MINB1314      # IS THIS A REGULAR DNADR?
      EXTEND
      BZMF      FETCH2WD      # YES. (A MUST NEVER BE ZERO)
      AD      MINB12      # NO. IS IT A POINTER (DNPTR) OR A
      EXTEND      # CHANNEL(DNCHAN)
      BZMF      DODNPTR      # IT'S A POINTER. (A MUST NEVER BE ZERO)

      DODNCHAN      TC      6      # (EXECUTED AS EXTEND) IT'S A CHANNEL
      INDEX      DNECADR
      INDEX      0      -4000      # (EXECUTED AS READ)
      TS      L
      TC      6      # (EXECUTED AS EXTEND)
      INDEX      DNECADR
      INDEX      0      -4001      # (EXECUTED AS READ)
      TS      DNECADR      # SET DNECADR
      CA      NEGONE      # TO MINUS
      XCH      DNECADR      # WHILE PRESERVING A.

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	TCF	DNTMEXIT	# GO SEND CHANNELS
WOZERO	CS	BIT7	
	EXTEND		
	WAND	CHAN13	# SET WORD ORDER CODE TO ZERO

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TC      Q      # RETURN TO CALLER

DODNPTR  INDEX  DNECADR  # DNECADR CONTAINS ADRES OF SUBLIST
          0      0      # CLEAR AND ADD LIST ENTRY INTO A.
          CCS    A      # IS THIS A SNAPSHOT SUBLIST
          CA     DNECADR # NO, IT IS A REGULAR SUBLIST.
          TCF    DOSUBLST # A MUST NOT BE ZERO.

          XCH     DNECADR # YES. IT IS A SNAPSHOT SUBLIST.
          TS      SUBLIST # C(DNECADR) INTO SUBLIST
          CAF     ZERO    # A INTO A
          XCH     TMINDEX # (NOTE: TMINDEX = DNECADR)

# THE FOLLOWING CODING (FROM SNAPLOOP TO SNAPEND) IS FOR THE PURPOSE OF TAKING A SNAPSHOT OF 12
# THIS IS DONE BY SAVING 11 DP REGISTERS IN DNTMBUFF AND SENDING THE FIRST DP WORD IMMEDIATELY.
# THE SNAPSHOT PROCESSING IS THE MOST TIME CONSUMING AND THEREFORE THE CODING AND LIST STRUCTURE
# TO MINIMIZE TIME. THE TIME OPTIMIZATION RESULTS IN RULES UNIQUE TO THE SNAPSHOT PORTION OF T
# THESE RULES ARE .....
# 1. ONLY 1DNADR'S CAN APPEAR IN THE SNAPSHOT SUBLIST
# 2. THE 1DNADR'S CANNOT REFER TO THE FIRST LOCATION IN ANY BANK.

SNAPLOOP TS      EBANK      # SET EBANK
          MASK    LOW8      # ISOLATE RELATIVE ADDRESS
          EXTEND
          INDEX   A
          EBANK=  1401
          DCA     1401      # PICK UP 2 SNAPSHOT WORDS.
          EBANK=  DNTMBUFF
          INDEX   TMINDEX
          DXCH    DNTMBUFF  # STORE 2 SNAPSHOT WORDS IN BUFFER
          INCR    TMINDEX  # SET BUFFER INDEX FOR NEXT 2 WORDS.
          INCR    TMINDEX
          SNAPAGN INCR    SUBLIST # SET POINTER TO NEXT 2 WORDS OF SNAPSHOT
          INDEX   SUBLIST
          0      0      # = CA SSSS (SSSS = NEXT ENTRY IN SUBLIST)
          CCS     A      # TEST FOR LAST TWO WORDS OF SNAPSHOT.
          TCF     SNAPLOOP # NOT LAST TWO.

LDNPHAS2 GENADR  DNPHASE2
          TS      SUBLIST  # YES, LAST. SAVE A.
          CA      NEGONE   # SET DNECADR AND
          TS      DNECADR  # SUBLIST POINTERS
          XCH     SUBLIST  # TO NEGATIVE VALUES
          TS      EBANK
          MASK    LOW8
          EXTEND

```

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INDEX A  
EBANK= 1401  
DCA 1401

# PICK UP FIRST 2 WORDS OF SNAPSHOT.

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```

SNAPEND      EBANK= DNTMBUFF
              TCF    DNTMEXIT      # NOW TO SEND THEM.

FETCH2WD     CA      DNECADR
              TS      EBANK        # SET EBANK
              MASK    LOW8        # ISOLATE RELATIVE ADDRESS
              TS      L
              CA      DNADRDRCR    # DECREMENT COUNT AND ECADR
              ADS     DNECADR
              EXTEND
              INDEX   L
              EBANK=  1400
              DCA     1400        # PICK UP 2 DATA WORDS
              EBANK=  DNTMBUFF
              TCF     DNTMEXIT    # NOW GO SEND THEM.

DOSUBLST     TS      SUBLIST      # SET SUBLIST POINTER
NEXTINSL     INDEX   SUBLIST
              0       0          # = CA SSSS (SSSS = NEXT ENTRY IN SUBLIST)
              CCS     A          # IS IT THE END OF THE SUBLIST
              INCR    SUBLIST    # NO --
              TCF     +4
              TS      SUBLIST    # SAVE A.
              CA      NEGONE     # SET SUBLIST TO MINUS
              XCH     SUBLIST    # RETRIEVE A.
              +4      INCR      A
              TS      DNECADR    # SAVE DNADR
              TCF     SETWO +2   # GO USE COMMON CODING (PROLEMS WOULD
                                # OCCUR IF THE PROGRAM ENCOUNTERED A
                                # DNPTR NOW)

DNTMEXIT     EXTEND            # DOWN-TELEMETRY EXIT
              WRITE   DNTM1    # TO SEND A + L TO CHANNELS 34 + 35
              CA      L        # RESPECTIVELY

TMEXITL      EXTEND
              WRITE   DNTM2

TMRESUME     TCF     RESUME    # EXIT TELEMETRY PROGRAM VIA RESUME.

MINB12       EQUALS  -1/8
DNECADR      EQUALS  TMINDEX
CTLIST       EQUALS  LDATALST
SUBLIST      EQUALS  DNQ
```

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```

# SUBROUTINE NAME -- DNDUMP
#
# FUNCTIONAL DESCRIPTION -- TO SEND (DUMP) ALL ERASABLE STORAGE 'N' TIMES. (N=1 TO 4)
#     EACH BANK IS PRECEDED BY AN ID WORD, SYNCH BITS, ECADR AND TIME1 FOLLOWED BY
#     EBANK. EBANKS ARE DUMPED IN ORDER (I.E., EBANK 0 FIRST, THEN EBANK1 ETC.)
#
# CALLING SEQUENCE -- THE GROUND OR ASTRONAUT BY KEYING V74E CAN INITIALIZE THE DUMP
#     AFTER KEYING IN V74E THE CURRENT DOWNLIST WILL BE IMMEDIATELY TERMINATED AND
#     WILL BEGIN.
#
#     ONCE INITIATED THE DOWNLINK ERASABLE DUMP CAN BE TERMINATED (AND INTERRUPTED
#     BY THE FOLLOWING:
#
#     1.      A FRESH START
#     2.      COMPLETION OF ALL DOWNLINK DUMPS REQUESTED (ACCORDING TO BITS SET IN
#             CAN BE ALTERED BY A V21N01.
#     3.      AND INVOLUNTARILY BY A RESTART.
#
# NORMAL EXIT MODE -- TCF DNPHASE1
#
# ALARM OR ABORT MODE -- NONE
#
# *SUBROUTINES CALLED -- NONE
#
# ERASABLE INITIALIZATION REQUIRED --
#     DUMPCNT      OCT 20000      IF 4 COMPLETE ERASABLE DUMPS ARE DESIRED
#     DUMPCNT      OCT 10000      IF 2 COMPLETE ERASABLE DUMPS ARE DESIRED
#     DUMPCNT      OCT 04000      IF 1 COMPLETE ERASABLE DUMP  IS  DESIRED
#
# DEBRIS -- DUMPLOC, DUMPSW, DNTMGOTO, EBANK, AND CENTRAL REGISTERS
#
# TIMING --      TIME (IN SECS) = ((NO.DUMPS)*(NO.EBANKS)*(WDSPEREBANK + NO.IDWDS)) /
#     TIME (IN SECS) = (  4  )*(  8  )*(  256  +  4  ) /
#     THUS TIME (IN SECS TO SEND DUMP OF ERASABLE 4 TIMES VIA DOWNLINK) = 8
#
# STRUCTURE OF ONE EBANK AS IT IS SENT BY DOWNLINK PROGRAM --
#     (REMINDER -- THIS ONLY DESCRIBES ONE OF THE 8 EBANKS X 4 (DUMPS) = 32 EBANKS
#
#     DOWNLIST
#
#     WORD  TAKEN FROM CONTENTS OF  EXAMPLE 0      COMMENTS
#     1     ERASID                    0177X 0      DOWNLIST I.D. FOR DOWNLINK ER
#     2     LOWIDCOD                  77340 1      DOWNLINK SYNCH BITS. (SAME 0
#     3     DUMPLOC                   13400 1      (SEE NOTES ON DUMPLOC) 1 = 3F
#     4     TIME1                     14120 1      TIME IN CENTISECONDS
#     5     FIRST WORD OF EBANK X     03400 1      IN THIS EXAMPLE THIS WORD = 0

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```
#          6  2ND  WORD OF EBANK X    00142  1      IN THIS EXAMPLE THIS WORD = CONTENTS OF
#          7  3RD  WORD OF EBANK X    00142  1      IN THIS EXAMPLE THIS WORD = CONTENTS OF
#          .
#          .
#          .
#          260D 256TH WORD OF EBANK X    03777  1      IN THIS EXAMPLE THIS WORD = CONTENTS OF
#
# NOTE --      DUMPLOC CONTAINS THE COUNTER AND ECADR FOR EACH WORD BEING SENT.
#              THE BIT STRUCTURE OF DUMPLOC IS FOLLOW --
#
#              X = NOT USED
#              X ABC EEE RRRRRRRR      ABC = ERASABLE DUMP COUNTER (I.E. ABC = 0,1,2, OR
#              COMPLETE ERASABLE DUMP NUMBER 1,2,3, OR 4 R
#              EEE = EBANK BITS
#              RRRRRRRR = RELATIVE ADDRESS WITHIN AN EBANK
```

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DNDUMPI      CA      ZERO      # INITIALIZE DOWNLINK
              TS      DUMPLOC   # ERASABLE DUMP
              +2      TC      SENDID # GO SEND ID AND SYNCH BITS
              CA      LDNDUMP1  # SET DNTMGOTO
              TS      DNTMGOTO  # TO LOCATION FOR NEXT PASS
              CA      TIME1     # PLACE TIME1
              XCH     L         # INTO L
              CA      DUMPLOC   # AND ECADR OF THIS EBANK INTO A
              TCF     DNTMEXIT  # SEND DUMPLOC AND TIME1

LDNDUMP      ADRES    DNDUMP
LDNDUMP1     ADRES    DNDUMP1

DNDUMP       CA      TWO      # INCREMENT ECADR IN DUMPLOC
              ADS     DUMPLOC  # TO NEXT DP WORD TO BE
              MASK    LOW8     # DUMPED AND SAVE IT.
              CCS     A        # IS THIS THE BEGINNING OF A NEW EBANK
              TCF     DNDUMP2  # NO -- THEN CONTINUE DUMPING
              CA      DUMPLOC  # YES -- IS THIS THE END OF THE
              MASK    DUMPCNT  # N TH (N = 1 TO 4) COMPLETE ERASABLE
              MASK    PRI034   # DUMP (BIT14 FOR 4, BIT13 FOR 2 OR BIT12
              CCS     A        # FOR 1 COMPLETE ERASABLE DUMP(S)).
              TCF     DNPHASE1 # YES -- START SENDING INTERRUPTED DOWNLIST
              # AGAIN
              TCF     DNDUMPI +2 # NO -- GO BACK AND INITIALIZE NEXT BANK

DNDUMP1      CA      LDNDUMP   # SET DNTMGOTO
              TS      DNTMGOTO # FOR WORDS 3 TO 256D OF CURRENT EBANK

DNDUMP2      CA      DUMPLOC   # SET EBANK
              TS      EBANK    # ISOLATE RELATIVE ADDRESS.
              MASK    LOW8     # (NOTE: MASK INSTRUCTION IS USED TO PICK
              TS      Q        # UP ERASABLE REGISTERS SO THAT EDITING
              CA      NEGO     # REGISTERS 20-23 WILL NOT BE ALTERED.)
              TS      L
              INDEX   Q
              EBANK=  1400     # PICK UP LOW ORDER REGISTER OF PAIR
              MASK    1401     # OF ERASABLE REGISTERS.
              XCH     L
              INDEX   Q        # PICK UP HIGH ORDER REGISTER OF PAIR
              MASK    1400     # OF ERASABLE REGISTERS.
              EBANK=  DNTMBUFF
              TCF     DNTMEXIT # GO SEND THEM

SENDID       EXTEND          # ** ENTRANCE USED BY ERASABLE DUMP PROG. **

```

		QXCH	DNTMGOTO	# SET DNTMGOTO SO NEXT TIME PROG WILL GO
		CAF	ERASID	# TO LOCATION FOLLOWING 'TC SENDID'
		TS	L	# ** ENTRANCE USED BY REGULAR DOWNLINK PG **
507a	$\langle$ Page LM0997 507a $\rangle \equiv$			(493 729)
		TC	WOZERO	# GO SET WORD ORDER CODE TO ZERO
		CAF	LOWIDCOD	# PLACE SPECIAL ID CODE INTO L
		XCH	L	# AND ID BACK INTO A
		TCF	DNTMEXIT	# SEND DOWNLIST ID CODE(S).
	WOTEST	CA	BIT7	# AT THE BEGINNING OF THE LIST THE WORD
		EXTEND		# ORDER BIT WILL BE SET BACK TO ZERO
		RAND	CHAN13	
		CCS	A	
		TC	DNTMGOTO	
		CA	BIT7	
		TCF	W01	

1.28 interpretive constant

507b	$\langle$ interpretive constant 507b $\rangle \equiv$	(7)
	$\langle$ Page LM1100 508 $\rangle$	
	$\langle$ Page LM1101 509a $\rangle$	

508  $\langle$ Page LM1100 508 $\rangle \equiv$  (507b 740)

	SETLOC	INTPRET1
	BANK	
	COUNT*	\$\$/ICONS
DP1/4TH	2DEC	.25
UNITZ	2DEC	0
UNITY	2DEC	0
UNITX	2DEC	.5
ZEROVECS	2DEC	0
	2DEC	0
	2DEC	0
DPHALF	=	UNITX
DPPOSMAX	OCT	37777
	OCT	37777

509a      *<Page LM1101 509a>≡* (507b 740)  
# INTERPRETIVE CONSTANTS IN THE OTHER HALF-MEMORY

	SETLOC	INTPRET2	
	BANK		
	COUNT*	\$\$/ICONS	
ZUNIT	2DEC	0	
YUNIT	2DEC	0	
XUNIT	2DEC	.5	
ZEROVEC	2DEC	0	
	2DEC	0	
	2DEC	0	
	OCT	77777	# -0, -6, -12 MUST REMAIN IN THIS ORDER
DFC-6	DEC	-6	
DFC-12	DEC	-12	
LODPMAX	2OCT	3777737777	# THESE TWO CONSTANTS MUST REMAIN
LODPMAX1	2OCT	3777737777	# ADJACENT AND THE SAME FOR INTEGRATION
ZERODP	=	ZEROVEC	
HALFDP	=	XUNIT	

1.29    agc block two self-check

509b      *<lm agc block two self check 509b>≡* (7)  
    *<Page LM1284 510>*  
    *<Page LM1285 512>*  
    *<Page LM1286 514>*  
    *<Page LM1287 516>*  
    *<Page LM1288 518>*  
    *<Page LM1289 520>*  
    *<Page LM1290 522>*  
    *<Page LM1291 524>*  
    *<Page LM1292 526>*  
    *<Page LM1293 527a>*

510 (Page LM1284 510)≡

(509b 711)

```

# PROGRAM DESCRIPTION
# PROGRAM NAME -- SELF-CHECK
# MOD NO -- 1
# MOD BY -- GAUNTT
#
# FUNCTIONAL DESCRIPTION
#     PROGRAM HAS TWO MAIN PARTS.  THE FIRST IS SELF-CHECK WHICH RUNS AS A ZERO PR
#     PART OF THE BACK-UP IDLE LOOP.  THE SECOND IS SHOW-BANKSUM WHICH RUNS AS A R
#     STARTING VERB.
#
#     THE PURPOSE OF SELF-CHECK IS TO CHECK OUT VARIOUS PARTS OF THE COMPUTER AS O
#
#     THE PURPOSE OF SHOW-BANKSUM IS TO DISPLAY THE SUM OF EACH BANK, ONE AT A TIME
#
#     IN ALL THERE ARE 7 POSSIBLE OPTIONS IN THIS BLOCK II VERSION OF SELF-CHECK.
#     FOUND IN E-2065 BLOCK II AGC SELF-CHECK AND SHOW BANKSUM BY EDWIN D. SMALLY I
#
#     THE DIFFERENT OPTIONS ARE CONTROLLED BY PUTTING DIFFERENT NUMBERS IN THE SMO
#     A DESCRIPTION OF WHAT PARTS OF THE COMPUTER THAT ARE CHECKED BY THE OPTIONS,
#     OCTAL, TO LOAD INTO SMODE.
#         +-4             ERASABLE MEMORY
#         +-5             FIXED MEMORY
#         +-1,2,3,6,7,10  EVERYTHING IN OPTIONS 4 AND 5.
#         -0              SAME AS +-10 UNTIL AN ERROR IS DETECTED.
#         +0              NO CHECK, PUTS COMPUTER INTO THE BACKUP IDLE LOOP.
#
# WARNINGS
#     USE OF E MEMORY RESERVED FOR SELF-CHECK (EVEN IN IDLE LOOP) AS TEMP STORAGE I
#     SMODE SET GREATER THAN OCT 10 PUTS COMPUTER INTO BACKUP IDLE LOOP.
#
# CALLING SEQUENCE
#     TO CALL SELF-CHECK KEY IN
#         V 21 N 27 E      OPTION NUMBER E
#     TO CALL SHOW-BANKSUM KEY IN
#         V 91 E           DISPLAYS FIRST BANK
#         V 33 E           PROCEED, DISPLAYS NEXT BANK
#
# EXIT MODES, NORMAL AND ALARM
#     SELF-CHECK NORMALLY CONTINUES INDEFINITELY UNLESS THERE IS AN ERROR DETECTED
#     COMPUTER INTO BACKUP IDLE LOOP, - OPTIONS NUMBERS RESTART THE OPTION.
#
#     THE -0 OPTION PROCEEDS FROM THE LINE FOLLOWING THE LINE WHERE THE ERROR WAS I
#
#     SHOW-BANKSUM PROCEEDS UNTIL A TERMINATE IS KEYED IN (V 34 E).  THE COMPUTER I
#

```

```

DATE: 20 DECEMBER 1967
LOG SECTION: AGC BLOCK TWO SELF-CHECK
ASSEMBLY SUBROUTINE UTILITYM REV 25

```

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# OUTPUT

512 (Page LM1285 512)≡

(509b 711)

```

#      SELF-CHECK UPON DETECTING AN ERROR LOADS THE SELF-CHECK ALARM CONSTANT (01102
#      TURNS ON THE ALARM LIGHT.  THE OPERATOR MAY THEN DISPLAY THE THREE FAILREGS I
#      INFORMATION HE MAY KEY IN V 05 N 08 E, THE DSKY DISPLAY IN R1 WILL BE ADDRESS
#      IN R2 THE BBCON OF SELF-CHECK, AND IN R3 THE TOTAL NUMBER OF ERRORS DETECTED
#      INITIATED FRESH START (SLAP1).
#
#      SHOW-BANKSUM STARTING WITH BANK 0 DISPLAYS IN R1 THE BANK SUM (A +-NUMBER EQ
#      THE BANK NUMBER, AND IN R3 THE BUGGER WORD.
#
# ERASABLE INITIALIZATION REQUIRED
#      ACCOMPLISHED BY FRESH START
#      SMODE SET TO +0
#
# DEBRIS
#      ALL EXITS FROM THE CHECK OF ERASABLE (ERASCHK) RESTORE ORIGINAL CONTENTS TO B
#      EXCEPTION IS A RESTART.  RESTART THAT OCCURS DURING ERASCHK RESTORES ERASABLE
#      E MEMORY, IN WHICH CASE PROGRAM THEN DOES A FRESH START (DOFSTART).

      BANK      25
      SETLOC    SELFCHEC
      BANK

      COUNT*    $$/SELF
SBIT1      EQUALS  BIT1
SBIT2      EQUALS  BIT2
SBIT3      EQUALS  BIT3
SBIT4      EQUALS  BIT4
SBIT5      EQUALS  BIT5
SBIT6      EQUALS  BIT6
SBIT7      EQUALS  BIT7
SBIT8      EQUALS  BIT8
SBIT9      EQUALS  BIT9
SBIT10     EQUALS  BIT10
SBIT11     EQUALS  BIT11
SBIT12     EQUALS  BIT12
SBIT13     EQUALS  BIT13
SBIT14     EQUALS  BIT14
SBIT15     EQUALS  BIT15

S+ZERO     EQUALS  ZERO
S+1        EQUALS  BIT1
S+2        EQUALS  BIT2
S+3        EQUALS  THREE
S+4        EQUALS  FOUR
S+5        EQUALS  FIVE

```



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S+6

EQUALS SIX

514 (Page LM1286 514)≡

(509b 711)

S+7	EQUALS	SEVEN	
S8BITS	EQUALS	LOW8	# 00377
CNTRCON	=	OCT50	# USED IN CNTRCHK
ERASCON1	OCTAL	00061	# USED IN ERASCHK
ERASCON2	OCTAL	01373	# USED IN ERASCHK
ERASCON6	=	OCT1400	# USED IN ERASCHK
ERASCON3	OCTAL	01461	# USED IN ERASCHK
ERASCON4	OCTAL	01773	# USED IN ERASCHK
S10BITS	EQUALS	LOW10	# 01777, USED IN ERASCHK
SBNK03	EQUALS	PRI06	# 06000, USED IN ROPECHK
-MAXADRS	=	HI5	# FOR ROPECHK
SIXTY	OCTAL	00060	
SUPRCON	OCTAL	60017	# USED IN ROPECHK
S13BITS	OCTAL	17777	
CONC+S1	OCTAL	25252	# USED IN CYCLSHFT
CONC+S2	OCTAL	52400	# USED IN CYCLSHFT
ERASCON5	OCTAL	76777	
S-7	=	OCT77770	
S-4	EQUALS	NEG4	
S-3	EQUALS	NEG3	
S-2	EQUALS	NEG2	
S-1	EQUALS	NEGONE	
S-ZERO	EQUALS	NEGO	
	EBANK=	LST1	
ADRS1	ADRES	SKEEP1	
SELFADRS	ADRES	SELFCHK	# SELFCHK RETURN ADDRESS. SHOULD BE PUT # IN SELFRET WHEN GOING FROM SELFCHK TO # SHOWSUM AND PUT IN SKEEP1 WHEN GOING # FROM SHOWSUM TO SELF-CHECK.
PRERRORS	CA	ERESTORE	# IS IT NECESSARY TO RESTORE ERASABLE
	EXTEND		
	BZF	ERRORS	# NO
	EXTEND		
	DCA	SKEEP5	
	INDEX	SKEEP7	
	DXCH	0000	# RESTORE THE TWO ERASABLE REGISTERS
	CA	S+ZERO	
	TS	ERESTORE	
ERRORS	INHINT		
	CA	Q	
	TS	SFAIL	# SAVE Q FOR FAILURE LOCATION
	TS	ALMCADR	# FOR DISPLAY WITH BBANK AND ERCOUNT
	INCR	ERCOUNT	# KEEP TRACK OF NUMBER OF MALFUNCTIONS.

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TCALARM2	TC	ALARM2	
	OCT	01102	# SELF-CHECK MALFUNCTION INDICATOR
	CCS	SMODE	
SIDLOOP	CA	S+ZERO	
	TS	SMODE	

516 (Page LM1287 516)≡

(509b 711)

	TC	SELFCHK	# GO TO IDLE LOOP
	TC	SFAIL	# CONTINUE WITH SELF-CHECK
-1CHK	CCS	A	
	TCF	PRERRORS	
	TCF	PRERRORS	
	CCS	A	
	TCF	PRERRORS	
	TC	Q	
SMODECHK	EXTEND		
	QXCH	SKEEP1	
	TC	CHECKNJ	# CHECK FOR NEW JOB
	CCS	SMODE	
	TC	SOPTIONS	
	TC	SMODECHK +2	# TO BACKUP IDLE LOOP
	TC	SOPTIONS	
	INCR	SCOUNT	
	TC	SKEEP1	# CONTINUE WITH SELF-CHECK
SOPTIONS	AD	S-7	
	EXTEND		
	BZMF	+2	# FOR OPTIONS BELOW NINE.
BNKOPTN	TC	SIDLOOP	# ILLEGAL OPTION. GO TO IDLE LOOP.
	INCR	SCOUNT	# FOR OPTIONS BELOW NINE.
	AD	S+7	
	INDEX	A	
	TC	SOPTION1	
SOPTION1	TC	SKEEP1	# WAS TC+TCF
SOPTION2	TC	SKEEP1	# WAS IN:OUT1
SOPTION3	TC	SKEEP1	
SOPTION4	TC	ERASCHK	
SOPTION5	TC	ROPECHK	
SOPTION6	TC	SKEEP1	
SOPTION7	TC	SKEEP1	
SOPTON10	TC	SKEEP1	# CONTINUE WITH SELF-CHECK
CHECKNJ	EXTEND		
	QXCH	SELFRET	# SAVE RETURN ADDRESS WHILE TESTING NEWJOB
	TC	POSTJUMP	# TO SEE IF ANY JOBS HAVE BECOME ACTIVE.
	CADR	ADVAN	
SELFCHK	TC	SMODECHK	# ** CHARLEY, COME IN HERE

```
# SKEEP7 HOLDS LOWEST OF TWO ADDRESSES BEING CHECKED.  
# SKEEP6 HOLDS B(X+1).  
# SKEEP5 HOLDS B(X).  
# SKEEP4 HOLDS C(EBANK) DURING ERASLOOP AND CHECKNJ  
# SKEEP3 HOLDS LAST ADDRESS BEING CHECKED (HIGHEST ADDRESS).
```

518 &lt;Page LM1288 518&gt;≡

(509b 711)

# SKEEP2 CONTROLS CHECKING OF NON-SWITCHABLE ERASABLE MEMORY WITH BANK NUMBERS IN EB  
#  
# ERASCHK TAKES APPROXIMATELY 7 SECONDS.

ERASCHK	CA	S+1	
	TS	SKEEP2	
OEBANK	CA	S+ZERO	
	TS	EBANK	
	CA	ERASCON3	# 01461
	TS	SKEEP7	# STARTING ADDRESS
	CA	S10BITS	# 01777
	TS	SKEEP3	# LAST ADDRESS CHECKED
	TC	ERASLOOP	
E134567B	CA	ERASCON6	# 01400
	TS	SKEEP7	# STARTING ADDRESS
	CA	S10BITS	# 01777
	TS	SKEEP3	# LAST ADDRESS CHECKED
	TC	ERASLOOP	
2EBANK	CA	ERASCON6	# 01400
	TS	SKEEP7	# STARTING ADDRESS
	CA	ERASCON4	# 01773
	TS	SKEEP3	# LAST ADDRESS CHECKED
	TC	ERASLOOP	
NOEBANK	TS	SKEEP2	# +0
	CA	ERASCON1	# 00061
	TS	SKEEP7	# STARTING ADDRESS
	CA	ERASCON2	# 01373
	TS	SKEEP3	# LAST ADDRESS CHECKED
ERASLOOP	INHINT		
	CA	EBANK	# STORES C(EBANK)
	TS	SKEEP4	
	EXTEND		
	NDX	SKEEP7	
	DCA	0000	
	DXCH	SKEEP5	# STORES C(X) AND C(X+1) IN SKEEP6 AND 5.
	CA	SKEEP7	
	TS	ERESTORE	# IF RESTART, RESTORE C(X) AND C(X+1)
	TS	L	
	INCR	L	
	NDX	A	
	DXCH	0000	# PUTS OWN ADDRESS IN X AND X +1

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NDX	SKEEP7	
CS	0001	# CS X+1
NDX	SKEEP7	
AD	0000	# AD X
TC	-1CHK	
CA	ERESTORE	# HAS ERASABLE BEEN RESTORED
EXTEND		

520 (Page LM1289 520)≡

(509b 711)

```

BZF      ELOOPFIN      # YES, EXIT ERASLOOP.
EXTEND
NDX      SKEEP7
DCS      0000          # COMPLEMENT OF ADDRESS OF X AND X+1
NDX      SKEEP7
DXCH     0000          # PUT COMPLEMENT OF ADDRESS OF X AND X+1
NDX      SKEEP7
CS       0000          # CS X
NDX      SKEEP7
AD       0001          # AD X+1
TC       -1CHK
CA       ERESTORE      # HAS ERASABLE BEEN RESTORED
EXTEND
BZF      ELOOPFIN      # YES, EXIT ERASLOOP.
EXTEND
DCA      SKEEP5
NDX      SKEEP7
DXCH     0000          # PUT B(X) AND B(X+1) BACK INTO X AND X+1
CA       S+ZERO
TS       ERESTORE      # IF RESTART, DO NOT RESTORE C(X), C(X+1)
ELOOPFIN RELINT
TC       CHECKNJ       # CHECK FOR NEW JOB
CA       SKEEP4        # REPLACES B(EBANK)
TS       EBANK
INCR     SKEEP7
CS       SKEEP7
AD       SKEEP3
EXTEND
BZF      +2
TC       ERASLOOP      # GO TO NEXT ADDRESS IN SAME BANK
CCS      SKEEP2
TC       NOEBANK
INCR     SKEEP2        # PUT +1 IN SKEEP2.
CA       EBANK
AD       SBIT9
TS       EBANK
AD       ERASCON5      # 76777, CHECK FOR BANK E2
EXTEND
BZF      2EBANK
CCS      EBANK
TC       E134567B      # GO TO EBANKS 1,3,4,5,6, AND 7
CA       ERASCON6      # END OF ERASCHK
TS       EBANK

```

# CNTRCHK PERFORMS A CS OF ALL REGISTERS FROM OCT. 60 THROUGH OCT. 10.



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# INCLUDED ARE ALL COUNTERS, T6-1, CYCLE AND SHIFT, AND ALL RUPT REGISTERS

CNTRCHK	CA	CNTRCON	# 00050
CNTRLOOP	TS	SKEEP2	
	AD	SBIT4	# +10 OCTAL
	INDEX	A	
	CS	0000	

522 (Page LM1290 522)≡

(509b 711)

CCS SKEEP2  
TC CNTRLOOP

## # CYCLSHFT CHECKS THE CYCLE AND SHIFT REGISTERS

CYCLSHFT	CA	CONC+S1	# 25252	
	TS	CYR	# C(CYR) = 12525	
	TS	CYL	# C(CYL) = 52524	
	TS	SR	# C(SR) = 12525	
	TS	EDOP	# C(EDOP) = 00125	
	AD	CYR	# 37777	C(CYR) = 45252
	AD	CYL	# 00-12524	C(CYL) = 25251
	AD	SR	# 00-25251	C(SR) = 05252
	AD	EDOP	# 00-25376	C(EDOP) = +0
	AD	CONC+S2	# C(CONC+S2) = 52400	
	TC	-1CHK		
	AD	CYR	# 45252	
	AD	CYL	# 72523	
	AD	SR	# 77775	
	AD	EDOP	# 77775	
	AD	S+1	# 77776	
	TC	-1CHK		
	INCR	SCOUNT +1		
	TC	SMODECHK		

# SKEEP1 HOLDS SUM

# SKEEP2 HOLDS PRESENT CONTENTS OF ADDRESS IN ROPECHK AND SHOWSUM ROUTINES

# SKEEP2 HOLDS BANK NUMBER IN LOW ORDER BITS DURING SHOWSUM DISPLAY

# SKEEP3 HOLDS PRESENT ADDRESS (00000 TO 01777 IN COMMON FIXED BANKS)

# (04000 TO 07777 IN FXFX BANKS)

# SKEEP3 HOLDS BUGGER WORD DURING SHOWSUM DISPLAY

# SKEEP4 HOLDS BANK NUMBER AND SUPER BANK NUMBER

# SKEEP5 COUNTS 2 SUCCESSIVE TC SELF WORDS

# SKEEP6 CONTROLS ROPECHK OR SHOWSUM OPTION

# SKEEP7 CONTROLS WHEN ROUTINE IS IN COMMON FIXED OR FIXED FIXED BANKS

ROPECHK	CA	S-ZERO	# *
	TS	SKEEP6	# * -0 FOR ROPECHK
STSHOSUM	CA	S+ZERO	# * SHOULD BE ROPECHK
	TS	SKEEP4	# BANK NUMBER
	CA	S+1	
COMMFx	TS	SKEEP7	
	CA	S+ZERO	

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	TS	SKEEP1	
	TS	SKEEP3	
	CA	S+1	
	TS	SKEEP5	# COUNTS DOWN 2 TC SELF WORDS
COMADRS	CA	SKEEP4	
	TS	L	# TO SET SUPER BANK
	MASK	HI5	

524 (Page LM1291 524)≡

(509b 711)

	AD	SKEEP3	
	TC	SUPDACAL	# SUPER DATA CALL
	TC	ADSUM	
	AD	SBIT11	# 02000
	TC	ADRSCHK	
FXFX	CS	A	
	TS	SKEEP7	
	EXTEND		
	BZF	+3	
	CA	SBIT12	# 04000, STARTING ADDRESS OF BANK 02
	TC	+2	
	CA	SBNK03	# 06000, STARTING ADDRESS OF BANK 03
	TS	SKEEP3	
	CA	S+ZERO	
	TS	SKEEP1	
	CA	S+1	
	TS	SKEEP5	# COUNTS DOWN 2 TC SELF WORDS
FXADRS	INDEX	SKEEP3	
	CA	0000	
	TC	ADSUM	
	TC	ADRSCHK	
ADSUM	TS	SKEEP2	
	AD	SKEEP1	
	TS	SKEEP1	
	CAF	S+ZERO	
	AD	SKEEP1	
	TS	SKEEP1	
	CS	SKEEP2	
	AD	SKEEP3	
	TC	Q	
ADRSCHK	LXCH	A	
	CA	SKEEP3	
	MASK	LOW10	# RELATIVE ADDRESS
	AD	-MAXADRS	# SUBTRACT MAX RELATIVE ADDRESS = 1777.
	EXTEND		
	BZF	SOPTION	# CHECKSUM FINISHED IF LAST ADDRESS.
	CCS	SKEEP5	# IS CHECKSUM FINISHED
	TC	+3	# NO
	TC	+2	# NO
	TC	SOPTION	# GO TO ROPECHK SHOWSUM OPTION
	CCS	L	# -0 MEANS A TC SELF WORD.
	TC	CONTINU	

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TC	CONTINU
TC	CONTINU
CCS	SKEEP5
TC	CONTINU +1
CA	S-1

526 (Page LM1292 526)≡

(509b 711)

	TC	CONTINU +1	# AD IN THE BUGGER WORD
CONTINU	CA	S+1	# MAKE SURE TWO CONSECUTIVE TC SELF WORDS
	TS	SKEEP5	
	CCS	SKEEP6	# *
	CCS	NEWJOB	# * +1, SHOWSUM
	TC	CHANG1	# *
	TC	+2	# *
	TC	CHECKNJ	# -0 IN SKEEP6 FOR ROPECHK
ADRS+1	INCR	SKEEP3	
	CCS	SKEEP7	
	TC	COMADRS	
	TC	COMADRS	
	TC	FXADRS	
	TC	FXADRS	
NXTBNK	CS	SKEEP4	
	AD	LSTBNKCH	# LAST BANK TO BE CHECKED
	EXTEND		
	BZF	ENDSUMS	# END OF SUMMING OF BANKS.
	CA	SKEEP4	
	AD	SBIT11	
	TS	SKEEP4	# 37 TO 40 INCRMTS SKEEP4 BY END RND CARRY
	TC	CHKSUPR	
17T020	CA	SBIT15	
	ADS	SKEEP4	# SET FOR BANK 20
	TC	GONXTBNK	
CHKSUPR	MASK	HI5	
	EXTEND		
	BZF	NXTSUPR	# INCREMENT SUPER BANK
27T030	AD	S13BITS	
	EXTEND		
	BZF	+2	# BANK SET FOR 30
	TC	GONXTBNK	
	CA	SIXTY	# FIRST SUPER BANK
	ADS	SKEEP4	
	TC	GONXTBNK	
NXTSUPR	AD	SUPRCON	# SET BNK 30 + INCR SUPR BNK AND CANCEL
	ADS	SKEEP4	# ERC BIT OF THE 37 TO 40 ADVANCE.
GONXTBNK	CCS	SKEEP7	
	TC	COMMFY	
	CA	S+1	
	TC	FXFX	
	CA	SBIT7	# HAS TO BE LARGER THAN NO OF FXSW BANKS.
	TC	COMMFY	

	SOPTION	CA	SKEEP4	
		MASK	HI5	# = BANK BITS
		TC	LEFT5	
		TS	L	# BANK NUMBER BEFORE SUPER BANK
527a	<i>&lt;Page LM1293 527a&gt;</i> ≡			(509b 711)
		CA	SKEEP4	
		MASK	S8BITS	# = SUPER BANK BITS
		EXTEND		
		BZF	SOPT	# BEFORE SUPER BANK
		TS	SR	# SUPER BANK NECESSARY
		CA	L	
		MASK	SEVEN	
		AD	SR	
		TS	L	# BANK NUMBER WITH SUPER BANK
	SOPT	CA	SKEEP6	# *
		EXTEND		# *
		BZF	+2	# * ON -0 CONTINUE WITH ROPE CHECK.
		TC	SDISPLAY	# * ON +1 GO TO DISPLAY OF SUM.
		CCS	SKEEP1	# FORCE SUM TO ABSOLUTE VALUE.
		TC	+2	
		TC	+2	
		AD	S+1	
		TS	SKEEP1	
	BNKCHK	CS	L	# = - BANK NUMBER
		AD	SKEEP1	
		AD	S-1	
		TC	-1CHK	# CHECK SUM
		TC	NXTBNK	
		EBANK=	NEWJOB	
	LSTBNKCH	BBCON*		# * CONSTANT, LAST BANK.

### 1.30 rtb op codes

527b	<i>&lt;rtb op codes 527b&gt;</i> ≡			(7)
	<i>&lt;Page LM1397 528&gt;</i>			
	<i>&lt;Page LM1398 530&gt;</i>			
	<i>&lt;Page LM1399 531a&gt;</i>			
	<i>&lt;Page LM1400 531b&gt;</i>			
	<i>&lt;Page LM1401 532&gt;</i>			
	<i>&lt;Page LM1402 533a&gt;</i>			

528 *(Page LM1397 528)*≡

(527b 763)

BANK 22  
 SETLOC RTBCODES  
 BANK

EBANK= XNB  
 COUNT\* \$\$/RTB

# LOAD TIME2, TIME1 INTO MPAC:

LOADTIME EXTEND  
 DCA TIME2  
 TCF SLOAD2

# CONVERT THE SINGLE PRECISION 2'S COMPLEMENT NUMBER ARRIVING IN MPAC (SCALED IN HALF-REVOLUTIONS)  
 # DP 1'S COMPLEMENT NUMBER SCALED IN REVOLUTIONS.

CDULOGIC CCS MPAC  
 CAF ZERO  
 TCF +3  
 NOOP  
 CS HALF  
  
 TS MPAC +1  
 CAF ZERO  
 XCH MPAC  
 EXTEND  
 MP HALF  
 DAS MPAC  
 TCF DANZIG # MODE IS ALREADY AT DOUBLE-PRECISION

# FORCE TP SIGN AGREEMENT IN MPAC:

SGNAGREE TC TPAGREE  
 TCF DANZIG

# CONVERT THE DP 1'S COMPLEMENT ANGLE SCALED IN REVOLUTIONS TO A SINGLE PRECISION 2'S COMPLEMENT  
 # SCALED IN HALF-REVOLUTIONS.

1ST02S TC 1T02SUB  
 CAF ZERO  
 TS MPAC +1  
 TCF NEWMODE

# DO 1ST02S ON A VECTOR OF ANGLES:



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V1ST02S

TC

1T02SUB

# ANSWER ARRIVES IN A AND MPAC.

DXCH

MPAC +5

DXCH

MPAC

TC

1T02SUB

530 (Page LM1398 530)≡

(527b 763)

TS MPAC +2

DXCH MPAC +3

DXCH MPAC

TC 1T02SUB

TS MPAC +1

CA MPAC +5

TS MPAC

TPMODE CAF ONE # MODE IS TP.  
TCF NEWMODE

# V1ST02S FOR 2 COMPONENT VECTOR. USED BY RR.

2V1ST02S TC 1T02SUB

DXCH MPAC +3

DXCH MPAC

TC 1T02SUB

TS L

CA MPAC +3

TCF SLOAD2

# SUBROUTINE TO DO DOUBLING AND 1'S TO 2'S CONVERSION:

1T02SUB DXCH MPAC # FINAL MPAC +1 UNSPECIFIED.

DDOUBL

CCS A

AD ONE

TCF +2

COM # THIS WAS REVERSE OF MSU.

TS MPAC

TC Q # AND SKIP ON OVERFLOW.

INDEX A # OVERFLOW UNCORRECT AND IN MSU.

CAF LIMITS

ADS MPAC

TC Q

# THE FOLLOWING ROUTINE INCREMENTS IN 2S COMPLEMENT THE REGISTER WHOSE ADDRESS IS IN  
# QUANTITY FOUND IN TEM2. THIS MAY BE USED TO INCREMENT DESIRED IMU AND OPTICS CDU A  
# (+0 UNEQUAL TO -0) QUANTITY. MAY BE CALLED BY BANKCALL/SWCALL.

CDUINC TS TEM2 # 1S COMPL.QUANT. ARRIVES IN ACC. STORE IT

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	INDEX	BUF	
	CCS	0	# CHANGE 2S COMPL. ANGLE(IN BUF) INTO 1S
	AD	ONE	
	TCF	+4	
	AD	ONE	
531a	<i>(Page LM1399 531a)</i> ≡		
			(527b 763)
	AD	ONE	# OVERFLOW HERE IF 2S COMPL. IS 180 DEG.
	COM		
	AD	TEM2	# SULT MOVES FROM 2ND TO 3D QUAD. (OR BACK)
	CCS	A	# BACK TO 2S COMPL.
	AD	ONE	
	TCF	+2	
	COM		
	TS	TEM2	# STORE 14BIT QUANTITY WITH PRESENT SIGN
	TCF	+4	
	INDEX	A	# SIGN.
	CAF	LIMITS	# FIX IT, BY ADDING IN 37777 OR 40000
	AD	TEM2	
	INDEX	BUF	
	TS	0	# STORE NEW ANGLE IN 2S COMPLEMENT.
	TC	Q	
531b	<i>(Page LM1400 531b)</i> ≡		
			(527b 763)
	# RTB TO TORQUE GYROS, EXCEPT FOR THE CALL TO IMUSTALL. ECADR OF COMMANDS ARRIVES IN X1.		
PULSEIMU	INDEX	FIXLOC	# ADDRESS OF GYRO COMMANDS SHOULD BE IN X1
	CA	X1	
	TC	BANKCALL	
	CADR	IMUPULSE	
	TCF	DANZIG	

532 (Page LM1401 532)≡

(527b 763)

```

# THE SUBROUTINE SIGNMPAC SETS C(MPAC, MPAC +1) TO SIGN(MPAC).
# FOR THIS, ONLY THE CONTENTS OF MPAC ARE EXAMINED.  ALSO +0 YIELDS POSMAX AND -0 YI
#
# ENTRY MAY BE BY EITHER OF THE FOLLOWING:
#     1.      LIMIT THE SIZE OF MPAC ON INTERPRETIVE OVERFLOW:
#             ENTRY:      BOVB
#                               SIGNMPAC
#     2.      GENERATE IN MPAC THE SIGNUM FUNCTION OF MPAC:
#             ENTRY:      RTB
#                               SIGNMPAC
# IN EITHER CASE, RETURN IS TO  THE NEXT INTERPRETIVE INSTRUCTION IN THE CALLING SEQ

```

```

SIGNMPAC      EXTEND
               DCA      DPOSMAX
               DXCH     MPAC
               CCS      A
DPMODE        CAF      ZERO          # SETS MPAC +2 TO ZERO IN THE PROCESS
               TCF      SLOAD2 +2
               TCF      +1
               EXTEND
               DCS      DPOSMAX
               TCF      SLOAD2

```

```

# RTB OP CODE NORMUNIT IS LIKE INTERPRETIVE INSTRUCTION UNIT, EXCEPT THAT IT CAN BE I
# UP WHEN THE VECTOR BEING UNITIZED IS VERY SMALL -- IT WILL BLOW UP WHEN ALL COMPON
# IS USED AND THE UPPER ORDER HALVES OF ALL COMPONENTS ARE ZERO, THE MAGNITUDE RETUR
# BY A FACTOR OF 2(13) AND THE SQUARED MAGNITUDE RETURNED ATE 34D WILL BE TOO BIG BY

```

```

NORMUNX1      CAF      ONE
               TCF      NORMUNIT +1
NORMUNIT      CAF      ZERO
               AD       FIXLOC
               TS       MPAC +2
               TC       BANKCALL      # GET SIGN AGREEMENT IN ALL COMPONENTS
               CADR     VECAGREE
               CCS      MPAC
               TCF      NOShift
               TCF      +2
               TCF      NOShift
               CCS      MPAC +3
               TCF      NOShift
               TCF      +2
               TCF      NOShift
               CCS      MPAC +5
               TCF      NOShift

```

```

TCF      +2
TCF      NOSHIFT

533a      <Page LM1402 533a>≡
                                         (527b 763)
          CA      MPAC +1      # SHIFT ALL COMPONENTS LEFT 13
          EXTEND
          MP      BIT14
          DAS      MPAC      # DAS GAINS A LITTLE ACCURACY
          CA      MPAC +4
          EXTEND
          MP      BIT14
          DAS      MPAC +3
          CA      MPAC +6
          EXTEND
          MP      BIT14
          DAS      MPAC +5
          CAF      THIRTEEN
          INDEX    MPAC +2
          TS      37D
          OFFTUNIT TC      POSTJUMP
          CADR     UNIT +1      # SKIP THE "TC VECAGREE" DONE AT UNIT

          NOSHIFT  CAF      ZERO
          TCF      OFFTUNIT -2

          # RTB VECSGNAG ... FORCES SIGN AGREEMENT OF VECTOR IN MPAC.

          VECSGNAG TC      BANKCALL
                   CADR     VECAGREE
                   TC      DANZIG

          # *** END OF SKIPPER .087 ***
```

1.31 t6 rupt programs

```

533b      <t6 rupt programs 533b>≡
          <Page LM1403 534>
          <Page LM1404 536>
          <Page LM1405 537>
                                         (7)
```

534 (Page LM1403 534)≡

(533b 769)

# PROGRAM NAMES: (1) T6JOBCHK MOD. NO. 5 OCTOBER 2, 1967

# (2) DOT6RUPT

# MODIFICATION BY: LOWELL G. HULL (A.C.ELECTRONICS)

#

# THESE PROGRAMS ENABLE THE LM DAP TO CONTROL THE THRUST TIMES OF THE REACTION CONTROL

# SINCE THE LM DAP MAINTAINS EXCLUSIVE CONTROL OVER TIME6 AND ITS INTERRUPTS, THE FOLLOWING

# ESTABLISHED AND MUST NOT BE TAMPERED WITH:

# 1. NO NUMBER IS EVER PLACED INTO TIME6 EXCEPT BY LM DAP.

# 2. NO PROGRAM OTHER THAN LM DAP ENABLES THE TIME6 COUNTER.

# 3. TO USE TIME6, THE FOLLOWING SEQUENCE IS ALWAYS EMPLOYED:

# A. A POSITIVE (NON-ZERO) NUMBER IS STORED IN TIME6.

# B. THE TIME6 CLOCK IS ENABLED.

# C. TIME6 IS INTERROGATED AND IS:

# I. NEVER FOUND NEGATIVE (NON-ZERO) OR +0.

# II. SOMETIMES FOUND POSITIVE (BETWEEN 1 AND 240D) INDICATING THAT IT IS INACTIVE.

# III. SOMETIMES FOUND POSMAX INDICATING THAT IT IS INACTIVE.

# IV. SOMETIMES FOUND NEGATIVE ZERO INDICATING THAT:

# A. A T6RUPT IS ABOUT TO OCCUR AT THE NEXT DINC,

# B. A T6RUPT IS WAITING IN THE PRIORITY CHAIN, OR

# C. A T6RUPT IS IN PROCESS NOW.

# 4. ALL PROGRAMS WHICH OPERATE IN EITHER INTERRUPT MODE OR WITH INTERRUPT MODE

# EVERY 5 MILLISECONDS TO PROCESS A POSSIBLE WAITING T6RUPT BEFORE IT C

# (5. PROGRAM JTLST, IN Q,R-AXES, HANDLES THE INPUT LIST.)

#

# T6JOBCHK CALLING SEQUENCE:

# L TC T6JOBCHK

# L+1 (RETURN)

#

# DOT6RUPT CALLING SEQUENCE:

# DXCH ARUPT # T6RUPT LEAD IN AT LOCATION 4004.

# EXTEND

# DCA T6ADR

# DTCB

#

# SUBROUTINES CALLED: DOT6RUPT CALLS T6JOBCHK.

#

# NORMAL EXIT MODES: T6JOBCHK RETURNS TO L +1.

# DOT6RUPT TRANSFERS CONTROL TO RESUME.

#

# ALARM/ABORT MODES: NONE.

#

# INPUT: TIME6 NXT6ADR OUTPUT: TIME6 NXT6

# T6NEXT T6NEXT +1 T6NEXT T6NEXT

# T6FURTHA T6FURTHA +1 T6FURTHA T6FURTHA

#

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# DEBRIS: T6JOBCHK CLOBBERS A. DOT6RUPT CLOBBERS NOTHING.  
BLOCK 02

536 (Page LM1404 536)≡

(533b 769)

```

BANK      17
SETLOC    DAPS2
BANK
EBANK=    T6NEXT
COUNT*   $$/DAPT6

T6JOBCHK  CCS      TIME6      # CHECK TIME6 FOR WAITING T6RUPT:
          TC       Q          # NONE: CLOCK COUNTING DOWN.
          TC       CCSHOLE
          TC       T6JOBCHK +3

```

# CONTROL PASSES TO T6JOB ONLY WHEN C(TIME6) = -0 (I.E., WHEN A T6RUPT MUST BE PROCESSED)

```

T6JOB      CAF      POSMAX      # DISABLE CLOCK: NEEDED SINCE RUPT OCCURS
          EXTEND
          WAND      CHAN13      # 1 DINC AFTER T6 = 77777. FOR 625 MUSECS
                                # MUST NOT HAVE T6 = +0 WITH ENABLE SET

```

```

          CA      POSMAX
          ZL
          DXCH     T6FURTHA
          DXCH     T6NEXT
          LXCH     NXT6ADR
          TS       TIME6

          AD      PRI037
          TS      A
          TCF     ENABLET6
          CA      POSMAX
          TS      TIME6
          TCF     GOCH56
ENABLET6   CA      BIT15
          EXTEND
          WOR     CHAN13
          CA      T6NEXT
          AD      PRI037
          TS      A
          TCF     GOCH56
          CA      POSMAX
          TS      T6NEXT
GOCH56    INDEX   L
          TCF     WRITEP -1

```

```

BLOCK     02
SETLOC    FFTAG9
BANK

```



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```

EBANK= CDUXD
COUNT* $$/DAPT6

WRITEP CA NEXTP
EXTEND
WRITE CHAN6

537 <Page LM1405 537>≡ (533b 769)
TC Q

WRITEU CA NEXTU
TS L
CS 00314OCT
EXTEND
RAND CHAN5
AD L
EXTEND
WRITE CHAN5
TC Q

WRITEV CA NEXTV
TS L
CA 00314OCT
TCF -9D
00314OCT OCT 00314

BANK 17
SETLOC DAPS2
BANK

EBANK= T6NEXT
COUNT* $$/DAPT6

DOT6RUPT LXCH BANKRUPT # (INTERRUPT LEAD INS CONTINUED)
EXTEND
QXCH QRUPT

TC T6JOBCHK # CALL T6JOBCHK.

TCF RESUME # END TIME6 INTERRUPT PROCESSOR.
```

### 1.32 dap interface subroutines

538a *<dap interface subroutines 538a>*≡ (7)  
*<Page LM1406 538b>*  
*<Page LM1407 539>*  
*<Page LM1408 541>*  
*<Page LM1409 542a>*

538b *<Page LM1406 538b>*≡ (538a 728)

BANK 20  
 SETLOC DAPS3  
 BANK  
  
 EBANK= CDUXD  
 COUNT\* \$\$/DAPIF

# MOD 0 DATE 11/15/66 BY GEORGE W. CHERRY  
 # MOD 1 1/23/67 MODIFICATION BY PETER ADLER

#  
 # FUNCTIONAL DESCRIPTION  
 # HEREIN IS A COLLECTION OF SUBROUTINES WHICH ALLOW MISSION CONTROL PROGRAMS TO  
 # AND INTERFACE WITH THE DAP.  
 #

# CALLING SEQUENCES  
 # IN INTERRUPT OR WITH INTERRUPT INHIBITED  
 # TC IBNKCALL  
 # FCADR ROUTINE  
 # IN A JOB WITHOUT INTERRUPT INHIBITED  
 # INHINT  
 # TC IBNKCALL  
 # FCADR ROUTINE  
 # RELINT  
 #

# OUTPUT  
 # SEE INDIVIDUAL ROUTINES BELOW  
 #

# DEBRIS  
 # A, L, AND SOMETIMES MDUETEMP ODE NOT IN PULSES MODE

539

(Page LM1407 539)≡

(538a 728)

```

# SUBROUTINE NAMES:
#     SETMAXDB, SETMINDB, RESTORDB, PFLITEDB
# MODIFIED:      30 JANUARY 1968 BY P. S. WEISSMAN TO CREATE RESTORDB.
# MODIFIED:      1 MARCH 1968 BY P. S. WEISSMAN TO SAVE EBANK AND CREATE PFLITEDB
#
# FUNCTIONAL DESCRIPTION:
#     SETMAXDB -- SET DEADBAND TO 5.0 DEGREES
#     SETMINDB -- SET DEADBAND TO 0.3 DEGREE
#     RESTORDB -- SET DEADBAND TO MAX OR MIN ACCORDING TO SETTINGS OF DBSELECT BIT OF DAPBOOL
#     PFLITEDB -- SET DEADBAND TO 1.0 DEGREE AND ZERO THE COMMANDED ATTITUDE CHANGE AND COMMA
#
#     ALL ENTRIES SET UP A NOVAC JOB TO DO 1/ACCS SO THAT THE TJETLAW SWITCH CURVES ARE POSIT
#     REFLECT THE NEW DEADBAND. IT SHOULD BE NOTED THAT THE DEADBAND REFERS TO THE ATTITUDE
#
# SUBROUTINE CALLED:    NOVAC
#
# CALLING SEQUENCE:     SAME AS ABOVE
#                       OR      TC RESTORDB +1      FROM ALLCOAST
#
# DEBRIS:               A, L, Q, RUPTREG1, (ITEMPS IN NOVAC)

RESTORDB      CAE      DAPBOOLS      # DETERMINE CREW-SELECTED DEADBAND.
              MASK     DBSELECT
              EXTEND
              BZF      SETMINDB

SETMAXDB      CAF      WIDEDB      # SET 5 DEGREE DEADBAND.
      +1      TS      DB

              EXTEND      # SET UP JOB TO RE-POSITION SWITCH CURVES.
              QXCH      RUPTREG1
CALLACCS      CAF      PRIO27
              TC      NOVAC
              EBANK=    AOSQ
              2CADR    1/ACCJOB

              TC      RUPTREG1      # RETURN TO CALLER.

SETMINDB      CAF      NARROWDB      # SET 0.3 DEGREE DEADBAND.
              TCF      SETMAXDB +1

PFLITEDB      EXTEND      # THE RETURN FROM CALLACCS IS TO RUPTREG1.
              QXCH      RUPTREG1
              TC      ZATTEROR      # ZERO THE ERRORS AND COMMANDED RATES.
              CAF      POWERDB      # SET DB TO 1.0 DEG.

```

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	TS	DB	
	TCF	CALLACCS	# SET UP 1/ACCS AND RETURN TO CALLER.
NARROWDB	OCTAL	00155	# 0.3 DEGREE SCALED AT 45.

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541 (Page LM1408 541)≡

(538a 728)

WIDEDB OCTAL 03434 # 5.0 DEGREES SCALED AT 45.  
POWERDB DEC .02222 # 1.0 DEGREE SCALED AT 45.

ZATTEROR CAF EBANK6  
XCH EBANK  
TS L # SAVE CALLERS EBANK IN L.  
CAE CDUX  
TS CDUXD  
CAE CDUY  
TS CDUYD  
CAE CDUZ  
TS CDUZD  
TCF STOPRATE +3

STOPRATE CAF EBANK6  
XCH EBANK  
TS L # SAVE CALLERS EBANK IN L.  
+3 CAF ZERO  
TS OMEGAPD  
TS OMEGAQD  
TS OMEGARD  
TS DELCDUX  
TS DELCDUY  
TS DELCDUZ  
TS DELPEROR  
TS DELQEROR  
TS DELREROR  
LXCH EBANK # RESTORE CALLERS EBANK.  
TC Q

# SUBROUTINE NAME: ALLCOAST  
# WILL BE CALLED BY FRESH STARTS AND ENGINE OFF ROUTINES.  
#  
# CALLING SEQUENCE: (SAME AS ABOVE)  
#  
# EXIT: RETURN TO Q.  
#  
# SUBROUTINES CALLED: STOPRATE, RESTORDB, NOVAC  
#  
# ZERO: (FOR ALL AXES) AOS, ALPHA, AOSTERM, OMEGAD, DELCDU, DELEROR  
#  
# OUTPUT: DRIFTBIT/DAPBOOLS, OE, JOB TO DO 1/ACCS  
#  
# DEBRIS: A, L, Q, RUPTREG1, RUPTREG2, (ITEMPS IN NOVAC)

	ALLCOAST	EXTEND		# SAVE Q FOR RETURN
		QXCH	RUPTREG2	
542a	$\langle \text{Page LM1409 542a} \rangle \equiv$			(538a 728)
		TC	STOPRATE	# CLEAR RATE INTERFACE. RETURN WITH A=0
		LXCH	EBANK	# AND L=EBANK6. SAVE CALLER'S EBANK.
		TS	AOSQ	
		TS	AOSQ +1	
		TS	AOSR	
		TS	AOSR +1	
		TS	ALPHAQ	# FOR DOWNLIST.
		TS	ALPHAR	
		TS	AOSQTERM	
		TS	AOSRTERM	
		LXCH	EBANK	# RESTORE EBANK (EBANK6 NO LONGER NEEDED)
		CS	DAPBOOLS	# SET UP DRIFTBIT
		MASK	DRIFTBIT	
		ADS	DAPBOOLS	
		TC	RESTORDB +1	# RESTORE DEADBANK TO CREW-SELECTED VALUE.
		TC	RUPTREG2	# RETURN.

### 1.33 dapidler program

542b	$\langle \text{dapidler program 542b} \rangle \equiv$	(7)
	$\langle \text{Page LM1410 543} \rangle$	
	$\langle \text{Page LM1411 544} \rangle$	
	$\langle \text{Page LM1412 546} \rangle$	
	$\langle \text{Page LM1413 548} \rangle$	
	$\langle \text{Page LM1414 549} \rangle$	
	$\langle \text{Page LM1415 551} \rangle$	
	$\langle \text{Page LM1416 552} \rangle$	
	$\langle \text{Page LM1417 553} \rangle$	
	$\langle \text{Page LM1418 554} \rangle$	
	$\langle \text{Page LM1419 556} \rangle$	
	$\langle \text{Page LM1420 557a} \rangle$	

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543 (Page LM1410 543)≡

(542b 726)

# THE DAPIDLER PROGRAM IS STARTED BY FRESH START AND RESTART. THE DAPIDLER PROGRAM IS DONE 10  
# PER SECOND UNTIL THE ASTRONAUT DESIRES THE DAP TO WAKE UP, AND THE IMU AND CDUS ARE READY FOR  
# THE NECESSARY INITIALIZATION OF THE DAP IS DONE BY THE DAPIDLER PROGRAM.

BANK 16  
SETLOC DAPS1  
BANK

EBANK= AOSQ

COUNT\* \$\$/DAPID

CHEKBITS EXTEND  
READ CHAN31 # IF BOTH BIT13 AND BIT14 ARE ONE, THEN  
COM # THE MODE SELECT SWITCH IS IN THE OFF  
MASK BIT13-14 # POSITION, AND SO THE DAP SHOULD BE OFF,  
EXTEND # WITH NO ATTITUDE ERROR DISPLAY.  
BZF MOREIDLE

CS IMODES33  
MASK BIT6  
CCS A  
TCF JUMPDSP  
CS RCSFLAGS # IMU NOT USABLE. SET UP INITIALIZATION  
MASK BIT3 # FLAG FOR ATT ERROR DISPLAY ROUTINE.  
ADS RCSFLAGS  
TCF SHUTDOWN

CHEKMORE CAF BIT10 # BIT 10 OF 30 IS PGNC S CONTROL OF S/C  
EXTEND  
RAND CHAN30 # BITS IN 30 ARE INVERTED  
CCS A  
TCF MOREIDLE

RETURN

544 <Page LM1411 544>=  
# DAPIDLER ENTRY.

(542b 726)

DAPIDLER	LXCH	BANKRUPT	# INTERRUPT LEAD INS (CONTINUED)
	EXTEND		
	QXCH	QRUPT	
	CA	RCSFLAGS	
	MASK	BIT13	
	CCS	A	# CHECK IF 1/ACCJOB HAS BEEN SET UP SINCE
	TCF	CHECKUP	# THE LAST FRESH START OR RESTART.
	CA	BIT13	
	ADS	RCSFLAGS	# BIT 13 IS 1.
	CAF	PRI027	
	TC	NOVAC	# SET UP JOB TO DO A LITTLE INITIALIZATION
	EBANK=	AOSQ	# AND EXECUTE 1/ACCS.
	2CADR	1/ACCSET	# (WILL BRANCH TO MOREIDLE ON ACCSOKAY)
CHECKUP	TC	CHEKBITS	# CHECK TO SEE IF LM DAP IS TO GO ON AND
			# DO ERROR DISPLAY.
	CAE	DAPBOOLS	# IF 1/ACCS HAS NOT BEEN COMPLETED, IDLE.
	MASK	ACCSOKAY	# NOTE: ONLY FRESH START AND RESTART
	EXTEND		# KNOCK THIS BIT DOWN.
	BZF	MOREIDLE	
STARTDAP	TC	IBNKCALL	# ZERO ATTITUDE ERROR AND DESIRED RATES.
	FCADR	ZATTEROR	
	CAF	ZERO	# ***** INITIALIZE: *****
	TS	TJP	
	TS	TJU	
	TS	TJV	
	TS	OMEGAP	# RATES IN BODY (PILOT) COORDINATES.
	TS	OMEGAQ	
	TS	OMEGAR	
	TS	TRAPEDP	
	TS	TRAPEDQ	
	TS	TRAPEDR	
	TS	AOSQ	# OFFSET ACCELERATION ESTIMATES.
	TS	AOSQ +1	
	TS	AOSR	
	TS	AOSR +1	
	TS	ALPHAQ	# COPIES OF OFFSET ESTIMATES FOR DOWNLIST.
	TS	ALPHAR	
	TS	NEGUQ	
	TS	NEGUR	



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TS	AOSQTERM	# QRAXIS RATE DERIVATION TERMS AND KALMAN
TS	AOSRTERM	# FILTER INITIALIZATION TERMS.
TS	QACCDOT	# DESCENT ACCELERATION DERIVATIVE EST.
TS	RACCDOT	

546 (Page LM1412 546)≡

(542b 726)

	TS	ALLOWGTS	# AOSTASK FLAG FOR QRAXIS RCS CONTROL USE.
	TS	COTROLER	# DO TRYGTS ON FIRST PASS (WILL GO TO RCS)
	TS	INGTS	# RECOGNIZE FIRST GTS PASS AS SUCH.
	TS	QGIMTIMR	# STOP GIMBAL DRIVES. (PROBABLY WOULD BE
	TS	RGIMTIMR	# GOOD ENOUGH JUST TO INACTIVATE TIMERS
	TS	OLDPMIN	# MINIMUM IMPULSE MODE ERASABLES
	TS	OLDQRMIN	
	TS	PJETCTR	# INITIALIZE DOCKED JET INHIBITION
	TS	UJETCTR	# COUNTERS
	TS	VJETCTR	
CALLGMBL	EQUALS	BIT5	# RCSFLAGS INITIALIZATION.
	CS	MANFLAG	
	MASK	RCSFLAGS	# NEGUQ(R) HAVE BEEN GENERATED.
	TS	RCSFLAGS	

# SET UP "OLD" MEASURED CDU ANGLES:

EXTEND

DCA CDUX # OLDXFORP AND OLDYFORP

DXCH OLDXFORP

CA CDUZ

TS OLDZFORQ

CS RCSFLAGS

MASK BIT12

ADS RCSFLAGS # BIT 12 SET TO 1.

CA FOUR

TS SKIPU

TS SKIPV

CA POSMAX

TS TIME6

TS T6NEXT

TS T6FURTHA

CA ZERO

TS T6NEXT +1

TS T6FURTHA +1

TS NXT6ADR

TS NEXTP

TS NEXTU

TS NEXTV

CS TEN

TS DAPZRUPT # JASK NOT IN PROGRESS, INITIALIZE NEG.

CA TWO

TS NPTRAPS

TS NQTRAPS

TS NRTRAPS

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	EXTEND	
	DCA	PAXADIDL
	DXCH	T5ADR
SETTIME5	CAF	MS100
	TS	TIME5

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(542b 726)

	TCF	RESUME	
	EBANK=	AOSQ	
IDLERADR	2CADR	DAPIDLER	
MOREIDLE	TC	IBNKCALL	# CALCULATE Q,R-AXES ATTITUDE ERRORS.
	CADR	QERRCALC	
	TC	IBNKCALL	
	CADR	CALCPERR	# CALCULATE P AXIS ATTITUDE ERRORS.
SHUTDOWN	EXTEND		
	DCA	IDLERADR	
	DXCH	T5ADR	
	CAF	ZERO	# KILL ANY POSSIBLE JET REQUESTS
	TS	NEXTP	
	TS	NEXTU	
	TS	NEXTV	
	EXTEND		# COMMAND JETS OFF.
	WRITE	CHAN5	
	EXTEND		
	WRITE	CHAN6	
	CS	BGIM23	# TURN TRIM GIMBAL OFF
	EXTEND		
	WAND	CHAN12	
	TCF	SETTIME5	# RETURN IN 100 MSEC.
MANFLAG	OCT	03021	
BGIM23	OCTAL	07400	
	EBANK=	OMEGAP	
PAXADIDL	2CADR	PAXIS	
MS100	=	OCT37766	
COSMG	=	ITEMP1	
JUMPDSP	EXTEND		# TRANSFER TO BANK 20
	DCA	DSPCADR	# FOR ATTITUDE ERROR DISPLAYS
	DTCB		
	EBANK=	AK	
DSPCADR	2CADR	ALTDSPLY	

549 (Page LM1414 549)≡

(542b 726)

```

BANK      20
SETLOC    DAPS3
BANK
COUNT*   $$/NEEDL

```

```

# PROGRAM:      ALTDSPY
# MOD 0.        6 DEC 1967
# AUTHOR:       CRAIG WORK, DON KEENE, MIT IL
# MOD 3 BY DON KEENE AUG 1, 1968 -- MOVED PROGRAM TO BANK 20
#

```

## # PROGRAM DESCRIPTION:

```

#      ALTDSPY REVERSES THE DSPYALT BIT OF RCSFLAGS EACH TIME IT IS CALLED, WHICH IS PRESUMED
#      IF THE REVERSED BIT IS ONE, NEEDLER IS CALLED TO DISPLAY ATTITUDE ERRORS.  IF THE BIT IS
#      ZERO, NEEDLER IS CALLED TO DISPLAY ATTITUDE ERRORS.  IF THE BIT IS ONE, NEEDLER IS CALLED TO
#      DISPLAY ATTITUDE ERRORS AS 1) DAP FOLLOWING ERRORS, IF NEEDLFLG = 0, AND 2) TOTAL ATTITUDE ERROR
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# WARNING:      ALTDSPY MAY ONLY BE CALLED WITH INTERRUPT INHIBITED
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# WARNING:      EBANK MUST BE SET TO 6 WHEN USING THIS ROUTINE.
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# INPUT:        RCSFLAGS AND      1) IF NEEDLFLG = 0, INPUT PERROR, QERROR, RERROR.
#                                     2) IF NEEDLFLG = 1, INPUT CPHI,CTHETA,CPSI,CDUX,CDUY,CDUZ,M11,M12,M13,M14,M15,M16,M17,M18,M19,M20,M21,M22,M23,M24,M25,M26,M27,M28,M29,M30,M31,M32,M33,M34,M35,M36,M37,M38,M39,M40,M41,M42,M43,M44,M45,M46,M47,M48,M49,M50,M51,M52,M53,M54,M55,M56,M57,M58,M59,M60,M61,M62,M63,M64,M65,M66,M67,M68,M69,M70,M71,M72,M73,M74,M75,M76,M77,M78,M79,M80,M81,M82,M83,M84,M85,M86,M87,M88,M89,M90,M91,M92,M93,M94,M95,M96,M97,M98,M99,M100,M101,M102,M103,M104,M105,M106,M107,M108,M109,M110,M111,M112,M113,M114,M115,M116,M117,M118,M119,M120,M121,M122,M123,M124,M125,M126,M127,M128,M129,M130,M131,M132,M133,M134,M135,M136,M137,M138,M139,M140,M141,M142,M143,M144,M145,M146,M147,M148,M149,M150,M151,M152,M153,M154,M155,M156,M157,M158,M159,M160,M161,M162,M163,M164,M165,M166,M167,M168,M169,M170,M171,M172,M173,M174,M175,M176,M177,M178,M179,M180,M181,M182,M183,M184,M185,M186,M187,M188,M189,M190,M191,M192,M193,M194,M195,M196,M197,M198,M199,M200,M201,M202,M203,M204,M205,M206,M207,M208,M209,M210,M211,M212,M213,M214,M215,M216,M217,M218,M219,M220,M221,M222,M223,M224,M225,M226,M227,M228,M229,M230,M231,M232,M233,M234,M235,M236,M237,M238,M239,M240,M241,M242,M243,M244,M245,M246,M247,M248,M249,M250,M251,M252,M253,M254,M255,M256,M257,M258,M259,M260,M261,M262,M263,M264,M265,M266,M267,M268,M269,M270,M271,M272,M273,M274,M275,M276,M277,M278,M279,M280,M281,M282,M283,M284,M285,M286,M287,M288,M289,M290,M291,M292,M293,M294,M295,M296,M297,M298,M299,M300,M301,M302,M303,M304,M305,M306,M307,M308,M309,M310,M311,M312,M313,M314,M315,M316,M317,M318,M319,M320,M321,M322,M323,M324,M325,M326,M327,M328,M329,M330,M331,M332,M333,M334,M335,M336,M337,M338,M339,M340,M341,M342,M343,M344,M345,M346,M347,M348,M349,M350,M351,M352,M353,M354,M355,M356,M357,M358,M359,M360,M361,M362,M363,M364,M365,M366,M367,M368,M369,M370,M371,M372,M373,M374,M375,M376,M377,M378,M379,M380,M381,M382,M383,M384,M385,M386,M387,M388,M389,M390,M391,M392,M393,M394,M395,M396,M397,M398,M399,M400,M401,M402,M403,M404,M405,M406,M407,M408,M409,M410,M411,M412,M413,M414,M415,M416,M417,M418,M419,M420,M421,M422,M423,M424,M425,M426,M427,M428,M429,M430,M431,M432,M433,M434,M435,M436,M437,M438,M439,M440,M441,M442,M443,M444,M445,M446,M447,M448,M449,M450,M451,M452,M453,M454,M455,M456,M457,M458,M459,M460,M461,M462,M463,M464,M465,M466,M467,M468,M469,M470,M471,M472,M473,M474,M475,M476,M477,M478,M479,M480,M481,M482,M483,M484,M485,M486,M487,M488,M489,M490,M491,M492,M493,M494,M495,M496,M497,M498,M499,M500,M501,M502,M503,M504,M505,M506,M507,M508,M509,M510,M511,M512,M513,M514,M515,M516,M517,M518,M519,M520,M521,M522,M523,M524,M525,M526,M527,M528,M529,M530,M531,M532,M533,M534,M535,M536,M537,M538,M539,M540,M541,M542,M543,M544,M545,M546,M547,M548,M549,M550,M551,M552,M553,M554,M555,M556,M557,M558,M559,M560,M561,M562,M563,M564,M565,M566,M567,M568,M569,M570,M571,M572,M573,M574,M575,M576,M577,M578,M579,M580,M581,M582,M583,M584,M585,M586,M587,M588,M589,M590,M591,M592,M593,M594,M595,M596,M597,M598,M599,M600,M601,M602,M603,M604,M605,M606,M607,M608,M609,M610,M611,M612,M613,M614,M615,M616,M617,M618,M619,M620,M621,M622,M623,M624,M625,M626,M627,M628,M629,M630,M631,M632,M633,M634,M635,M636,M637,M638,M639,M640,M641,M642,M643,M644,M645,M646,M647,M648,M649,M650,M651,M652,M653,M654,M655,M656,M657,M658,M659,M660,M661,M662,M663,M664,M665,M666,M667,M668,M669,M670,M671,M672,M673,M674,M675,M676,M677,M678,M679,M680,M681,M682,M683,M684,M685,M686,M687,M688,M689,M690,M691,M692,M693,M694,M695,M696,M697,M698,M699,M700,M701,M702,M703,M704,M705,M706,M707,M708,M709,M710,M711,M712,M713,M714,M715,M716,M717,M718,M719,M720,M721,M722,M723,M724,M725,M726,M727,M728,M729,M730,M731,M732,M733,M734,M735,M736,M737,M738,M739,M740,M741,M742,M743,M744,M745,M746,M747,M748,M749,M750,M751,M752,M753,M754,M755,M756,M757,M758,M759,M760,M761,M762,M763,M764,M765,M766,M767,M768,M769,M770,M771,M772,M773,M774,M775,M776,M777,M778,M779,M780,M781,M782,M783,M784,M785,M786,M787,M788,M789,M790,M791,M792,M793,M794,M795,M796,M797,M798,M799,M800,M801,M802,M803,M804,M805,M806,M807,M808,M809,M810,M811,M812,M813,M814,M815,M816,M817,M818,M819,M820,M821,M822,M823,M824,M825,M826,M827,M828,M829,M830,M831,M832,M833,M834,M835,M836,M837,M838,M839,M840,M841,M842,M843,M844,M845,M846,M847,M848,M849,M850,M851,M852,M853,M854,M855,M856,M857,M858,M859,M860,M861,M862,M863,M864,M865,M866,M867,M868,M869,M870,M871,M872,M873,M874,M875,M876,M877,M878,M879,M880,M881,M882,M883,M884,M885,M886,M887,M888,M889,M890,M891,M892,M893,M894,M895,M896,M897,M898,M899,M900,M901,M902,M903,M904,M905,M906,M907,M908,M909,M910,M911,M912,M913,M914,M915,M916,M917,M918,M919,M920,M921,M922,M923,M924,M925,M926,M927,M928,M929,M930,M931,M932,M933,M934,M935,M936,M937,M938,M939,M940,M941,M942,M943,M944,M945,M946,M947,M948,M949,M950,M951,M952,M953,M954,M955,M956,M957,M958,M959,M960,M961,M962,M963,M964,M965,M966,M967,M968,M969,M970,M971,M972,M973,M974,M975,M976,M977,M978,M979,M980,M981,M982,M983,M984,M985,M986,M987,M988,M989,M990,M991,M992,M993,M994,M995,M996,M997,M998,M999,M1000,M1001,M1002,M1003,M1004,M1005,M1006,M1007,M1008,M1009,M1010,M1011,M1012,M1013,M1014,M1015,M1016,M1017,M101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TCF NEEDLER

CAE FLAGWRDO # NEEDLFLG WILL INDICATE TOTAL OR DAP AT-

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      MASK    NEEDLBIT    # TITUDE ERROR DISPLAY REQUEST.
      CCS      A
      TCF      DSPLYTOT    # TOTAL ERROR IS NEEDED IN AK, AK +1, AK +2

      CS      QERROR      # YES.  DISPLAY ATT ERRORS ON THE -BALL.
      TS      AK +1        # ERROR COMPLEMENTS ARE INPUT TO NEEDLER.
      CS      RERROR
      TS      AK +2
      CS      PERROR
      XCH      AK

      TCF      RETNMORE    # DISPLAY THESE THE NEXT TIME THROUGH

# CALCULATE GIMBAL ANGLE TOTAL ERRORS, RESOLVE INTO PILOT AXES, STORE TOTAL ERRORS FOR NEEDLER.

DSPLYTOT      EXTEND
              QXCH      ITEMP1    # SAVE Q FOR CHEKBITS RETURN.

              CA      CTHETA      # DESIRED ATTITUDE, Y-AXIS, 2'S COMP.
              EXTEND      # SUBTRACT CURRENT ATTITUDE.
              MSU      CDUY      # DIFFERENCE SCALED AT PI, 1'S COMP.
              TS      AK      # SAVE FOR R-ERROR CALCULATION.
              EXTEND
              MP      M21      # (CTHETA-CDUY)*M21 SCALED AT PI RADIANS.
              XCH      AK +1    # STORE FIRST TERM OF Q ERROR.
              CA      CPSI      # DESIRED ATTITUDE, Z-AXIS, 2'S COMP.
              EXTEND      # SUBTRACT CURRENT ATTITUDE.
              MSU      CDUZ      # DIFFERENCE SCALED AT PI, 1'S COMP.
              TS      AK +2    # SAVE Z-AXIS TERM FOR R ERROR CALCULATION
              EXTEND
              MP      M22      # (CPSI-CDUZ)*M22, SCALED AT PI RADIANS.
              AD      AK +1    # Q ERROR COMPLETE , AT PI RAD.
              TC      OVERSUB2  # PIN NEEDLES IN CASE OF OVERFLOW
              TS      AK +1

# R ERROR CALCULATION NEXT.

              CA      AK      # Y-AXIS DIFFERENCE STORED BY Q-AXIS CALC.
              EXTEND
              MP      M31      # (CTHETA-CDUY)*M31, SCALED AT PI RADIANS.
              XCH      AK +2    # FIRST TERM OF R ERROR.
              # Z-AXIS DIFFERENCE, STORED BY A CALC. IS
              EXTEND      # RECOVERED BY THE EXCHANGE.
              MP      M32      # (CPSI-CDUZ)*M32, SCALED AT PI RADIANS.
              AD      AK +2    # R ERROR COMPLETE , AT PI RAD.
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TC      OVERSUB2      # PIN NEEDLES IN CASE OF OVERFLOW.
TS      AK +2

# NOW CALCULATE P ERROR.  (NOTE THAT M13 = 1, SCALED AT 1, SO THE MULTIPLICATION IS 1)

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      CA      AK      # Y-AXIS DIFFERENCE STORED BY Q AXIS CALC.
      EXTEND
      MP      M11      # (CTHETA-CDUY)*M11 SCALED AT PI RADIANS.
      XCH      AK      # FIRST TERM OF P ERROR IN AK, AT PI RAD.
      CAE      CPHI      # DESIRED ATTITUDE, X-AXIS, 2'S COMP.
      EXTEND      # SUBTRACT CURRENT X ATTITUDE.
      MSU      CDUX      # X-AXIS DIFFERENCE, 1'S COMP, AT PI RAD.

# M13 = 1, SO BYPASS THE MULTIPLICATION.
#      EXTEND
#      MP      M13      # (CPHI-CDUX)*M13 SCALED AT PI RADIANS.

      AD      AK      # P ERROR COMPLETE      , SCALED AT PI RAD
      TC      OVERSUB2      # PIN NEEDLES IN CASE OF OVERFLOW.
      TS      AK

      EXTEND
      QXCH      ITEMP1      # RESTORE Q FOR CHEKBITS RETURN.

      TCF      RETNMORE      # DISPLAY THESE THE NEXT TIME THROUGH

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# FDAI ATTITUDE ERROR DISPLAY SUBROUTINE
#
# PROGRAM DESCRIPTION:          D. KEENE          5/24/67
# MOD 1 BY CRAIG WORK, 12 DEC 67
# MOD 2 BY CRAIG WORK, 6 APRIL 68, CONVERTS ATTITUDE ERROR DISPLAY SCALING FROM 16 7/8 DEG. TO
#
# THIS SUBROUTINE IS USED TO DISPLAY ATTITUDE ERRORS ON THE FDAI VIA THE DIGITAL TO ANALOG CONV
# IN THE CDUS. CARE IS TAKEN TO METER OUT THE APPROPRIATE NUMBER OF PULSES TO THE IMU ERROR CO
# OVERFLOW, TO CONTROL THE RELAY SEQUENCING, AND TO AVOID INTERFERENCE WITH THE COARSE ALIGN LO
# THE DACS.
#
# CALLING SEQUENCE:
#   DURING THE INITIALIZATION SECTION OF THE USER'S PROGRAM, BIT3 OF RCSFLAGS SHOULD BE SET
#   TURN-ON SEQUENCE WITHIN THE NEEDLES PROGRAM:
#       CS      RCSFLAGS      # IN EBANK6
#       MASK    BIT3
#       ADS     RCSFLAGS
#   THEREAFTER, THE ATTITUDE ERRORS GENERATED BY THE USER SHOULD BE TRANSFERRED TO THE FOLL
#       AK      SCALED 180 DEGREES      NOTE:  THESE LOCATIONS ARE SUBJECT
#       AK1     SCALED 180 DEGREES      TO CHANGE
#       AK2     SCALED 180 DEGREES
#   FULL SCALED DEFLECTION OF THE NEEDLES CORRESPONDS TO 5 1/16 DEGREES, WHILE 384 BITS IN
#   CORRESPONDS TO 42 3/16 DEGREES. (DAC MAXIMUM CAPACITY IS 384 BITS.) 46 BITS EFFECTIVE
#
#   A CALL TO NEEDLER WILL THE UPDATE THE DISPLAY:
#       INHINT
#       TC      IBNKCALL      # NOTE:  EBANK SHOULD BE SET TO E6
#       CADR    NEEDLER
#       RELINT
#   THIS PROCESS SHOULD BE REPEATED EACH TIME THE ERRORS ARE UPDATED. AT LEAST 3 PASSES TH
#   REQUIRED BEFORE ANYTHING IS ACTUALLY DISPLAYED ON THE ERROR METERS.
#   NOTE:  EACH CALL TO NEEDLER MUST BE SEPARATED BY AT LEAST 50 MS. TO ASSURE PROPER RELAY
#
# ERASABLES USED:
#       AK      CDUXCMD
#       AK1     CDUYCMD
#       AK2     CDUZCMD
#       EDRIVEX A,L,Q
#       EDRIVEY T5TEMP
#       EDRIVEZ DINDX

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#
# SWITCHES:                RCSFLAGS  BITS 3,2
#
# I/O CHANNELS:           CHAN12  BIT 4 (COARSE ALIGN -- READ ONLY)
#                           CHAN12  BIT 6 (IMU ERROR COUNTER ENABLE)
#                           CHAN14  BIT 13,14,15 (DAC ACTIVITY)
#
# SIGN CONVENTION:        AK = THETAC - THETA
#                           WHERE  THETAC = COMMAND ANGLE
#                           THETA  = PRESENT ANGLE

NEEDLER                    CA      RCSFLAGS
                           MASK    SIX
                           EXTEND
                           BZF     NEEDLES3
                           MASK    BIT3
                           EXTEND
                           BZF     NEEDLER2          # BIT3 = 0, BIT2 = 1

                           CS      BIT6              # FIRST PASS BIT3 = 1
                           EXTEND                    # DISABLE IMU ERROR COUNTER TO ZERO DACS
WAND                       CHAN12                    # MUST WAIT AT LEAST 60 MS BEFORE
NEEDLE11                   CS      ZERO              # ENABLING COUNTERS.
                           TS      AK                # ZERO THE INPUTS ON FIRST PASS
                           TS      AK1
                           TS      AK2
                           TS      EDRIXEX           # ZERO THE DISPLAY REGISTERS
                           TS      EDRIYEX
                           TS      EDRIVEZ
                           TS      CDUXCMD           # ZERO THE OUT COUNTERS
                           TS      CDUYCMD
                           TS      CDUZCMD
                           CS      SIX              # RESET RCSFLAGS FOR PASS2
                           MASK    RCSFLAGS
                           AD      BIT2
                           TS      RCSFLAGS
                           TCF     RETNMORE

NEEDLER2                   CAF     BIT6              # ENABLE IMU ERROR COUNTERS
                           EXTEND
WOR                       CHAN12
CS                         SIX                    # RESET RCSFLAGS TO DISPLAY ATTITUDE
MASK                      RCSFLAGS                # ERRORS. WAIT AT LEAST 4 MS FOR
TS                        RCSFLAGS                # RELAY CLOSURE.
TCF                       RETNMORE

```

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NEEDLES3	CAF	BIT6	# CHECK TO SEE IF IMU ERROR COUNTER
	EXTEND		# IS ENABLED
	RAND	CHAN12	

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(542b 726)

	CCS	A	# IF NOT, RE-INITIALIZE NEEDLER.
	TCF	NEEDLES	
	CS	RCSFLAGS	# SET UP INITIALIZATION FLAG IN RCSFLAGS.
	MASK	BIT3	
	ADS	RCSFLAGS	
	TCF	RETNMORE	
NEEDLES	CAF	TWO	
DACLOOP	TS	DINDX	
	CS	ONETENTH	# RESCALE INPUTS TO + OR - 1800 DEGREES.
	EXTEND		
	INDEX	DINDX	
	MP	AK	
	TS	L	
	CCS	A	
	CA	DACLIMIT	
	TCF	+2	
	CS	DACLIMIT	
	AD	L	
	TS	T5TEMP	# OVFL0 CHK
	TCF	+4	
	INDEX	A	# ON OVERFLOW LIMIT OUTPUT TO +-384
	CAF	DACLIMIT	
	TS	L	
	INDEX	DINDX	
	CS	EDRIVEX	# CURRENT VALUE OF DAC
	AD	L	
	INDEX	DINDX	
	ADS	CDUXCMD	
	INDEX	DINDX	
	LXCH	EDRIVEX	
	CCS	DINDX	
	TCF	DACLOOP	
	CAF	13,14,15	
	EXTEND		
	WOR	CHAN14	# SET DAC ACTIVITY BITS
	TCF	RETNMORE	
	DEC	-384	
DACLIMIT	DEC	16000	
	DEC	384	
ONETENTH	OCT	03146	# DECIMAL +0.1, SCALED AT 1.
DSPLYALT	EQUALS	BIT4	# 100 MS ALTERNATION BIT IN RCSFLAGS

	OVERSUB2	TS	7	# RETURNS A UNCHANGED OR LIMITED TO
		TC	Q	# POSMAX OR NEGMAX IF A HAS OVERFLOW
		INDEX	A	
557a	$\langle \text{Page LM1420 557a} \rangle \equiv$			(542b 726)
		CS	LIMITS	# DUPLICATE CODING IN BANK 16
		TC	Q	
	RETNMORE	EXTEND		# RETURN TO CHEKMORE
		DCA	MORECADR	
		DTCB		
		EBANK=	AOSQ	
	MORECADR	2CADR	CHEKMORE	

### 1.34 p axis rcs autopilot

557b  $\langle p \text{ axis rcs autopilot 557b} \rangle \equiv$  (7)

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 $\langle \text{Page LM1422 560} \rangle$   
 $\langle \text{Page LM1423 562} \rangle$   
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 $\langle \text{Page LM1441 595} \rangle$

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(557b 756)

BANK 16  
 SETLOC DAPS1  
 BANK

EBANK= PERROR  
 COUNT\* \$\$/DAPP

# THE FOLLOWING T5RUPT ENTRY BEGINS THE PROGRAM WHICH CONTROLS THE P-AXIS ACTION OF T  
 # THE NOMINAL TIME BETWEEN THE P-AXIS RUPTS IS 100 MS IN ALL NON-IDLING MODES OF THE

PAXIS CA MS100  
 ADS TIME5 # \*\*\* NECESSARY IN ORDER TO ALLOW  
 # SYNCHRONIZATION WITH OTHER INTERRUPTS \*\*\*  
 LXCH BANKRUPT # INTERRUPT LEAD IN (CONTINUED)  
 EXTEND  
 QXCH QRUPT

# CHECK IF DAP PASS IS PERMISSIBLE

CCS DAPZRUPT # IF DAPZRUPT POSITIVE, DAP (JASK) IS  
 TC BAILOUT # STILL IN PROGRESS AND A RESTART IS  
 OCT 02000 # CALLED FOR. IT IS NEVER ZERO  
 TC CHEKBITS # RETURN IS TC I+1 IF DAP SHOULD STAY ON.  
 CA CDUX # READ AND STORE CDU'S  
 TS DAPTREG4  
 CA CDUY  
 TS DAPTREG5  
 CA CDUZ  
 TS DAPTREG6

# \*\*\*\*\* KALCMANU-DAP AND "RATE-HOLD"-DAP INTERFACE \*\*\*\*\*

#

# THE FOLLOWING SECTION IS EXECUTED EVERY 100 MS (10 TIMES A SECOND) WITHIN THE P-AXIS  
 # AUTOPILOT (WHENEVER THE DAP IS IN OPERATION).

CA CDUXD  
 EXTEND  
 MSU DELCDUX  
 TC 1STOTWOS  
 TS CDUXD  
 CA CDUYD  
 EXTEND

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MSU	DELCDUY
TC	1STOTWOS
TS	CDUYD
CA	CDUZD
EXTEND	
MSU	DELCDUZ

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(557b 756)

```

TC      1STOTWOS
TS      CDUZD
EXTEND
DIM      TCP      # DIMINISH MANUAL CONTROL DIRECT RATE
EXTEND
DIM      TCQR     # TIME COUNTERS.

# RATFLOOP COMPUTES JETRATER, JRATER, AND 1JACC*NO. PJEETS IN ITEMP1.
# RETURNS TO BACKP.
#
# JETRATER = 1JACC*NO.PJETS*TJP      (NOTE TJ IS THE TIME FIRED DURING CSP)
# JETRATER = 1JACCQ(TJU*NO.UJETS - TJV*NO.VJETS)
# JETRATER = 1JACCQ(TJU*NO.UJETS + TJV*NO.VJETS)

TCF      PAXFILT      # PROCEEDS TO RATELOOP AFTER SUPERJOB
1STOTWOS CCS      A
AD      ONE
TC      Q
CS      A
TC      Q
SUBDIVIDE EXTEND      # OVERFLOW PROTECTION ROUTINE TO GIVE
MP      DAPTEMP3      # POSMAX OR NEGMAX IF THE DIVIDE WOULD
DAS      OMEGAU        # OVERFLOW

+3      EXTEND
DCA      OMEGAU
DXCH     DAPTEMP5
CCS      OMEGAU
TCF      +2
TCF      DIVIDER
AD      -OCT630
EXTEND
BZMF     DIVIDER

CCS      OMEGAU
CA      POSMAX      # 45 DEG/SEC
TC      Q
CS      POSMAX
TC      Q

DIVIDER  DXCH      OMEGAU
EXTEND
DV      DAPTREG4
TC      Q

```



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OVERSUB	TS	7		# RETURNS A UNCHANGED OR LIMITED TO
	TC	Q		# POSMAX OR NEGMAX IF A HAS OVERFLOW
	INDEX	A		
	CS	BIT15	-1	

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(557b 756)

	TC	Q
-OCT630	OCT	77147
BACKP	CA	DAPTEMP1
	EXTEND	
	MP	1JACC
	TS	JETRATE

```

# BEGINNING OF THE RATE DERIVATION
#      OMEGAP,Q,R      BODY RATES SCALED AT PI/4
#      TRAPEDP,Q,R     BODY ANGLE ERRORS FROM PREDICTED ANGLE (PI/40)
#      NP(QR)TRAPS     NUMBER OF TIMES ANGLE ERROR HAS BEEN ACCUMULATED
#      AOSQ(R)TERM     CHANGE IN RATE DUE TO OFFSET ACCELERATION. (PI/4)
#      JETRATE,Q,R     CHANGE IN RATE DUE TO JET ACCELERATION. (PI/4)
#      TRAPSIZE        NEGATIVE LIMIT OF MAGNITUDE OF TRAPEDP, ETC.
#      OMEGAU          DP-TEMPORARY STORAGE
# OMEGA = OMEGA + JETRATE + AOSTERM (+TRAPED/NTRAPS IF TRAPED BIG)

```

CAE	DAPTREG4	# CDUX IS STORED HERE
TS	L	
EXTEND		
MSU	OLDXFORP	# SCALED AT PI
LXCH	OLDXFORP	
TS	DAPTEMP1	
CA	1/40	
TS	DAPTREG4	
CS	JETRATE	
EXTEND		
MP	BIT14	
ADS	TRAPEDP	
CA	JETRATER	
AD	AOSQTERM	
EXTEND		
MP	-BIT14	
ADS	TRAPEDQ	
CA	JETRATER	
AD	AOSRTERM	
EXTEND		
MP	-BIT14	
ADS	TRAPEDR	
CA	DAPTREG5	# CDUY IS STORED HERE
TS	L	
EXTEND		

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MSU	OLDYFORP	# SCALED AT PI
LXCH	OLDYFORP	
TS	DAPTEMP2	
EXTEND		
MP	M11	# M11 SCALED AT 1

564 *(Page LM1424 564)*≡

(557b 756)

AD DAPTEMP1

DXCH OMEGAU

TC SUBDIVDE +3 # RETURNS WITH CDU-RATE AT PI/4

EXTEND

SU OMEGAP

ADS TRAPEDP

TC OVERSUB

TS TRAPEDP

EXTEND

DCA DAPTEMP5

DAS DXERROR

CS PLAST

EXTEND

MP 1/40

DAS DXERROR

# MANUAL MODE X-ATTITUDE ERROR (DP)

CA DAPTREG6

# CDUZ IS STORED HERE

TS L

EXTEND

MSU OLDZFORQ

TS DAPTEMP3

LXCH OLDZFORQ

CA M21

EXTEND

MP DAPTEMP2

DXCH OMEGAU

CA M22

TC SUBDIVDE

EXTEND

SU OMEGAQ

ADS TRAPEDQ

TC OVERSUB

TS TRAPEDQ

EXTEND

DCA DAPTEMP5

DAS DYERROR

CS QLAST

EXTEND

MP 1/40

DAS DYERROR

# MANUAL MODE Y-ATTITUDE ERROR (DP)

CA M31

EXTEND

MP DAPTEMP2

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DXCH	OMEGAU
CA	M32
TC	SUBDIVDE

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(557b 756)

```

EXTEND
SU      OMEGAR
ADS     TRAPEDR
TC      OVERSUB
TS      TRAPEDR      # TRAPEDS HAVE ALL BEEN COMPUTED

EXTEND
DCA     DAPTEMP5
DAS     DZERROR
CS      RLAST
EXTEND
MP      1/40
DAS     DZERROR      # MANUAL MODE Z-ATTITUDE ERROR (DP)
CA      DAPBOOLS     # PICK UP PAD LOADED STATE ESTIMATOR GAINS
MASK    CSMDOCKD
EXTEND
BZF     LMONLY
EXTEND      # DOCKED
DCA     DKOMEGAN
DXCH    DAPTREG4
CA      DKTRAP
TCF     +5
LMONLY  EXTEND      # UNDOCKED
DCA     LMOMEGAN
DXCH    DAPTREG4
CA      LMTRAP
+5      TS      DAPTREG6
CCS     TRAPEDP
TCF     +2
TCF     SMALPDIF
AD      DAPTREG6      # TRAPSIZE > ABOUT 77001 %-1.4DEG/SEC"
EXTEND
BZMF    SMALPDIF
ZL
LXCH    TRAPEDP
CA      ZERO
EXTEND
DV      NPTRAPS
ADS     OMEGAP
TC      OVERSUB
TS      OMEGAP
CA      DAPTREG4      ABOUT 10 OR 0 FOR DOCKED OR UNDOCKED
TS      NPTRAPS
SMALPDIF INCR      NPTRAPS
P-RATE  CA      JETRATE

```

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ADS	OMEGAP
TC	OVERSUB
TS	OMEGAP
CCS	TRAPEDQ

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(557b 756)

```

TCF      +2
TCF      Q-RATE
AD       DAPTREG6      # TRAPSIZE > ABOUT 77001 %-1.4DEG/SEC"
EXTEND
BZMF     Q-RATE
ZL
LXCH     TRAPEDQ
CA       ZERO
EXTEND
DV       NQTRAPS
TS       DAPTEMP1      # SAVE FOR OFFSET ESTIMATE
ADS      OMEGAQ
TC       OVERSUB
TS       OMEGAQ
CA       DAPTREG4      # ABOUT 10 OR 0 FOR DOCKED OR UNDOCKED
XCH      NQTRAPS
AD       DAPTREG5      # KAOS > ABOUT 60D %N/N_60"
XCH      DAPTEMP1
EXTEND
MP       FIVE
EXTEND
DV       DAPTEMP1
ADS      AOSQ
Q-RATE   INCR          NQTRAPS
CA       JETRATEREQ
AD       AOSQTERM
ADS      OMEGAQ
TC       OVERSUB
TS       OMEGAQ

CCS      TRAPEDR
TCF      +2
TCF      R-RATE
AD       DAPTREG6      # TRAPSIZE > ABOUT 77001 %-1.4DEG/SEC"
EXTEND
BZMF     R-RATE
ZL
LXCH     TRAPEDR
CA       ZERO
EXTEND
DV       NRTRAPS
TS       DAPTEMP2      # SAVE FOR OFFSET ESTIMATE
ADS      OMEGAR
TC       OVERSUB
TS       OMEGAR

```



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CA	DAPTREG4	# ABOUT 10 OR 0 FOR DOCKED OR UNDOCKED
XCH	NRTRAPS	
AD	DAPTREG5	# KAOS > ABOUT 60D %N/N_60"
XCH	DAPTEMP2	
EXTEND		

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(557b 756)

```

R-RATE      MP      FIVE
            EXTEND
            DV      DAPTEMP2
            ADS      AOSR
            INCR     NRTRAPS
            CA      JETRATER
            AD      AOSRTERM
            ADS      OMEGAR
            TC      OVERSUB
            TS      OMEGAR

# END OF RATE DERIVATION
# BEGIN OFFSET ESTIMATOR
# IN POWERED FLIGHT, AOSTASK WILL BE CALLED EVERY 2 SECONDS.
# AOS = AOS + K*SUMRATE

CS      DAPBOOLS
MASK    DRIFTBIT
CCS     A
TCF     WORKTIME
TS      ALPHAQ      # ZERO THE OFFSET ACCELERATION VALUES.
TS      ALPHAR
TS      AOSQTERM
TS      AOSRTERM
TS      AOSQ
TS      AOSR
TCF     PRETIMCK
KAOS    DEC      60
WORKTIME CA      QACCDOT
EXTEND
MP      CALLCODE    # OCTAL 00032 IS DECIMAL .1 AT 2(6).
DAS     AOSQ
CA      AOSQ
TS      ALPHAQ
EXTEND
MP      200MS      # .2 AT 1
TS      AOSQTERM
CA      RACCDOT
EXTEND
MP      CALLCODE    # OCTAL 00032 IS DECIMAL .1 AT 2(6).
DAS     AOSR
CA      AOSR
TS      ALPHAR
EXTEND
MP      200MS      # .2 AT 1

```

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TS	AOSRTERM
TCF	PRETIMCK

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(557b 756)

```

PAXFILT      CA      CALLGMBL      # EXECUTE ACDT+C12, IF NEEDED.
              MASK    RCSFLAGS
              CCS     A      # CALLGMBL IS NOT BIT15, SO THIS TEST IS
              TC      ACDT+C12    # VALID.

              DXCH     ARUPT
              DXCH     DAPARUPT
              CA      SUPERJOB      # SETTING UP THE SUPERJOB
              XCH      BRUPT
              LXCH     QRUPT
              DXCH     DAPBQRPT
              CA      SUPERADR
              DXCH     ZRUPT
              DXCH     DAPZRUPT
              TCF      NOQBRSM +1    # RELINT (JUST IN CASE) AND RESUME, IN THE
                                      # FORM OF A JASK, AT SUPERJOB.

SUPERADR      GENADR  SUPERJOB +1

# COUNT DOWN GIMBAL DRIVE TIMERS AND TURN OFF DRIVES IF REQUIRED.

SUPERJOB      TCF      RATELOOP
PRETIMCK      CCS      QGIMTIMR
              TCF      DECQTIMR      # POSITIVE -- COUNTING DOWN
              TCF      TURNOFFQ      # NEGATIVE -- DRIVE SHOULD BE ENDED
CHKRTIMR      CCS      RGIMTIMR      # NEGATIVE -- INACTIVE
              TCF      DECRTIMR      # (NEG ZERO -- IMPOSSIBLE)
              TCF      TURNOFFR      # REPEATED (ABOVE) FOR R AXIS.

              EXTEND
              DIM      PJETCTR      # DECREMENT DOCKED JET INHIBITION COUNTERS
              EXTEND
              DIM      UJETCTR
              EXTEND
              DIM      VJETCTR
              CA      BIT12
              MASK    RCSFLAGS
              EXTEND
              BZF      SKIPPAXS
              TC      CHKVISFZ
DECQTIMR      TS      QGIMTIMR      # COUNT TIMERS DOWN TO POS ZERO.
              TCF      CHKRTIMR
DECRTIMR      TS      RGIMTIMR
              TCF      CHKRTIMR +3

```

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TURNOFFQ

TS      NEGUQ  
TS      QACCDOT  
CS      QGIMBITS  
EXTEND

# HALT DRIVES.

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(557b 756)

```

TURNOFFR      WAND      CHAN12
                CAF       NEGMAX
                TS        QGIMTIMR
                TCF       CHKRTIMR
                TS        NEGUR
                TS        RACCDOT
                CS        RGIMBITS
                EXTEND
                WAND      CHAN12
                CAF       NEGMAX
                TS        RGIMTIMR
                TCF       CHKRTIMR +3
QGIMBITS      EQUALS    OCT1400      # BITS 9 AND 10 (OF CHANNEL 12).
RGIMBITS      EQUALS    PRI06        # BITS 11 AND 12 (OF CHANNEL 12).

SKIPPAXS      CS        RCSFLAGS
                MASK      BIT12
                ADS       RCSFLAGS    # BIT 12 SET TO 1.
                TCF       QRAXIS      # GO TO QRAXIS OR TO CTS.

```

# Y-X TRANSLATION

#

# INPUT: BITS 9-12 OF CH31 (FROM TRANSLATION CONTROLLER)

#

# OUTPUT: NEXTP

#

# NEXTP IS THE CHANNEL 6 CODE OF JETS FOR THE DESIRED TRANSLATION.

# IF THERE ARE FAILURES IN THE DESIRED POLICY, THEN

# (1) FOR DIAGONAL TRANS: UNFAILED PAIR

# ALARM (IF NO PAIR)

# (2) FOR PRINCIPAL TRANS: TRY TO TACK WITH DIAGONAL PAIRS

# ALARM (IF DIAGONAL PAIRS ARE FAILED)

CHKVISFZ

EXTEND

READ CHAN31

CS A

MASK 07400OCT

EXTEND

BZF TSNEXTP

EXTEND

MP BIT7

INDEX A

CA INDXYZ

TS ROTINDEX

TRYUORV

CA SIX

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TC SELECTYZ  
CS SIX  
AD NUMBERT  
EXTEND

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(557b 756)

```

      BZF      TSNEXTP -1
      CS       FIVE
      AD       ROTINDEX
      EXTEND
      BZMF     ALTERYZ
      CS       NUMBERT
      AD       FOUR
      EXTEND
      BZMF     TSNEXTP -1
ABORTYZ      TC       ALARM
      OCT      02001
      CA       BIT1          # INVERT BIT 1 OF RCSFLAGS.
      LXCH     RCSFLAGS
      EXTEND
      RXOR     1
      TS       RCSFLAGS
      CA       ZERO
      TCF      TSNEXTP
      CA       BIT1          # INVERT BIT 1 OF RCSFLAGS.
      LXCH     RCSFLAGS
      EXTEND
      RXOR     1
      TS       RCSFLAGS
      MASK     BIT1
      AD       FOUR
      ADS      ROTINDEX
      TCF      TRYUORV
      CA       POLYTEMP
TSNEXTP      TS       NEXTP

# STATE LOGIC
#      CHECK IN ORDER:          IF ON
#      LPDPHASE                 GO TO PURGENCY
#      PULSES                   MINIMUM PULSE LOTIC
#      DETENT(BIT15 CH31)       RATE COMMAND
#      GOTO TO PURGENCY

      CA       BIT13          # CHECK STICK IF IN ATT. HOLD.
      EXTEND
      RAND     CHAN31
      EXTEND
      BZF      MANMODE

      CA       DAPBOOLS
      MASK     XOVINHIB

```



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	CCS	A	
	TCF	PURGENCY	# ATTITUDE STEER DURING VISIBILITY PHASE
MANMODE	TCF	DETENTCK	
	CA	PULSES	# PULSES IS ONE FOR PULSE MODE
	MASK	DAPBOOLS	

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(557b 756)

```

EXTEND
BZF      DETENTCK      # BRANCH FOR RATE COMMAND

CA       ZERO
TS       PERROR

# MINIMUM IMPULSE MODE

CA       CDUX
TS       CDUXD

CCS      OLDPMIN
TCF      CHECKP

FIREP    CA       BIT3
EXTEND
RAND     CHAN31
EXTEND
BZF      +XMIN

CA       BIT4
EXTEND
RAND     CHAN31
EXTEND
BZF      -XMIN

TCF      JETSOFF

CHECKP    EXTEND
READ     CHAN31
CS       A
MASK     OCT14
TS       OLDPMIN
TCF      JETSOFF

-XMIN    CS       TEN      # ANYTHING LESS THAN 14MS. CORRECTED
TCF      +2          #          IN JET SELECTION ROUTINE

+XMIN    CA       TEN
TS       TJP
CA       ONE
TS       OLDPMIN
TCF      PJETSLEC -6

#
#          MANUAL RATE COMMAND MODE
=====

```

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```
#
#                                     BY ROBERT F. STENGEL
#
# THIS MODE PROVIDES RCAF MANUAL CONTROL THRU 2 CONTROL LAWS:  1) DIRECT RATE AND 2) PSEUDO-AUTO
# THE DIRECT RATE MODE AFFORDS IMMEDIATE CONTROL WITHOUT OVERSHOOT.  THE PSEUDO-AUTO MODE PROVIDES
# RATE CONTROL AND ATTITUDE HOLD.
#
```

580 (Page LM1432 580)≡

(557b 756)

```

# IN DIRECT RATE, JETS ARE FIRED WHEN STICK POSITION CHANGES BY A FIXED NUMBER OF IN
# THE 'BREAKOUT LEVEL' IS .6 D/S FOR LM-ONLY AND .3 D/S FOR CSM-DOCKED. THIS LAW NU
# THE 'TARGET DEADBAND', WHICH EQUALS THE BREAKOUT LEVEL.
#
# IN PSEUDO-AUTO, BODY-FIXED RATE AND ATTITUDE ERRORS ARE SUPPLIED TO TJETLAW, WHICH
# CONTROL SWITCHES FROM DIRECT RATE TO PSEUDO-AUTO IF THE TARGET DB IS ACHIEVED OR IF
# IF THE INITIAL COMMAND DOES NOT EXCEED THE BREAKOUT LEVEL, CONTROL GOES TO PSEUDO-A
#
# SINCE P-AXIS CONTROL IS SEPARATE FROM Q,R AXES CONTROL, IT IS POSSIBLE TO USE (1)
# OR VICE VERSA. THIS ALLOWS A DEGREE OF ATTITUDE HOLD IN UNCONTROLLED AXES. DUE TO
# R AXES ARE COUPLED AND MUST USE THE SAME CONTROL LAW.
#
# HAND CONTROLLER COMMANDS ARE SCALED BY A LINEAR/QUADRATIC LAW. FOR THE LM-ALONE, M
# AND 4 D/S IN NORMAL AND FINE SCALING; HOWEVER, STICK SENSITIVITY AT ZERO COUNTS (OR
# OF 2 DEGREES FROM THE CENTERED POSITION) IS .5 OR .1 D/S PER DEGREE. NORMAL AND F
# CASE IS AUTOMATICALLY SET TO 1/10 THE ABOVE VALUES. SCALING IS DETERMINED IN ROUT
#
# ZEROENBL      ENABLES COUNTERS SO THEY CAN BE READ NEXT TIME
# JUSTOUT       FIRST DETECTION OF OUT OF DETENT (BY OURRCBIT)

DETENTCK      EXTEND
               READ    CHAN31
               TS      CH31TEMP
               MASK    BIT15          # CHECK OUT-OF-DETENT BIT.
               EXTEND
               BZF     RHCMOVED        # BRANCH IF OUT OF DETENT.
               CAF     OURRCBIT        # IN DETENT. CHECK THE RATE COMMAND BIT.
               MASK    DAPBOOLS
               EXTEND
               BZF     PURGENCY        # BRANCH IF NOT IN RATE COMMAND LAST PASS.

# .....

               CA      BIT9           # JUST IN DETENT??
               MASK    RCSFLAGS
               EXTEND
               BZF     RUTH
               CAF     BIT13          # CHECK FOR ATTITUDE HOLD.
               EXTEND
               RAND    CHAN31
               EXTEND
               BZF     RATEDAMP        # BRANCH IF IN ATTITUDE HOLD.

               CS      BITS9,11       # IN AUTO.
               MASK    RCSFLAGS       # (X-AXIS OVERRIDE)

```

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	TS	RCSFLAGS	# ZERO ORBIT (BIT 11) AND JUST-IN BIT (9).
	TCF	RATEDAMP	
RUTH	CA	RCSFLAGS	
	MASK	PBIT	# IN ATTITUDE HOLD.
	EXTEND		
	BZF	+2	# BRANCH IF P-RATE DAMPING IS FINISHED.
	TCF	RATEDAMP	

582 (Page LM1433 582)≡

(557b 756)

```

CA      RCSFLAGS
MASK    QRBIT
EXTEND
BZF     RATEDONE      # BRANCH IF Q,R RATE DAMPING IS FINISHED.
TCF     RATEDAMP

```

# =====

```

1/10SEC      OCT      1
40CYC        OCT      50
PQRBIT       OCT      74777
BITS9,11     EQUALS   EBANK5
LINRATP      DEC      46

```

# =====

```

RATEDONE     CS      OURRCBIT      # MANUAL COMMAND AND DAMPING COMPLETED IN
              INHINT                # ALL AXES.
              MASK    DAPBOOLS
              TS      DAPBOOLS

```

# READ CDUS INTO CDU DESIRED REGISTERS

```

CAF      BIT13
EXTEND
RAND     CHAN31
EXTEND
BZF      +4
CA       CDUX      # (X-AXIS OVERRIDE)
TS       CDUXD
TC       +3
TC       IBNKCALL
FCADR    ZATTEROR
RELINT
TCF      PURGENCY

```

```

JUSTOUT     TS      PERROR
            CA      OURRCBIT      # INITIALIZATION -- FIRST MANUAL PASS.
            ADS     DAPBOOLS
            CA      ZERO
            TS      DXERROR
            TS      DXERROR +1
            TS      DYERROR
            TS      DYERROR +1
            TS      DZERROR

```

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TS	DZERROR +1	
TS	PLAST	
TS	QLAST	
TS	RLAST	
TS	Q-RHCCTR	
TS	R-RHCCTR	
CA	PQRBIT	
MASK	RCSFLAGS	
TS	RCSFLAGS	# BITS 10 AND 11 OF RCSFLAGS ARE 0.

584    *(Page LM1434 584)*≡ (557b 756)

	CS	RCSFLAGS	# SET 'JUST-IN' BIT TO 1.
	MASK	BIT9	
	ADS	RCSFLAGS	
	TC	ZEROENBL	
	TCF	JETSOFF	
ZEROENBL	LXCH	R-RHCCTR	
	CA	Q-RHCCTR	
	DXCH	SAVEHAND	
	CA	ZERO	
	TS	P-RHCCTR	
	TS	Q-RHCCTR	
	TS	R-RHCCTR	
	CA	BITS8,9	
	EXTEND		
	WOR	CHAN13	# COUNTERS ZEROED AND ENABLED
	TC	Q	
RATEDAMP	CA	ZERO	
	TS	P-RHCCTR	
	TCF	RATERROR	
RHCMOVED	CA	OURRCBIT	# P CONTROL
	MASK	DAPBOOLS	
	EXTEND		
	BZF	JUSTOUT -1	
RATERROR	CA	CDUX	# FINDCDUW REQUIRES THAT CDUXD=CDUX DURING
	TS	CDUXD	# X-AXIS OVERRIDE
	CCS	P-RHCCTR	
	TCF	+3	
	TCF	+2	
	TCF	+1	
	DOUBLE		# LINEAR/QUADRATIC CONTROLLER SCALING
	DOUBLE		# (SEE EXPLANATION OF Q,R-AXES RCS
	AD	LINRATP	# AUTOPILOT)
	EXTEND		
	MP	P-RHCCTR	
	CA	L	
	EXTEND		
	MP	STIKSENS	
	XCH	PLAST	
	COM		
	AD	PLAST	
	TS	DAPTEMP1	
	TC	ZEROENBL	# INTERVAL. ZERO AND ENABLE ACA COUNTERS.
	CS	PLAST	
	AD	OMEGAP	



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TS	EDOTP	
CCS	DAPTEMP1	# IF P COMMAND CHANGE EXCEEDS BREAKOUT
TCF	+3	# LEVEL, GO TO DIRECT RATE CONTROL. IF NOT
TCF	+8D	# CHECK FOR DIRECT RATE CONTROL LAST TIME.
TCF	+1	

586 (Page LM1435 586)≡

(557b 756)

```

AD      -RATEDB
EXTEND
BZMF    +4
CA      40CYC
TS      TCP
TC      PEGI
CA      RCSFLAGS      # CHECK FOR DIRECT RATE COMMAND LAST TIME.
MASK    PBIT
EXTEND
BZF     +2
TC      PEGI          # TO PURE RATE COMMAND
CA      DXERROR        # PSEUDO-AUTO CONTROL.
TS      E              # X-ATTITUDE ERROR (SP)
TS      PERROR         # LOAD P-AXIS ERROR FOR MODE1 FDAI DISPLAY
TC      PURGENCY +4
PEGI    CA      CDUX    # DIRECT RATE CONTROL.
TS      CDUXD
CA      ZERO
TS      DXERROR
TS      DXERROR +1
TS      PERROR        # ZERO P-AXIS ERROR FOR MODE1 FDAI DISPLAY
CCS     EDOTP
TC      +3
TC      +2
TC      +1
TS      ABSEDOTP
AD      TARGETDB
EXTEND      # IF RATE ERROR IS LESS THAN DEADBANK,
BZMF    LAST      # FIRE, AN SWITCH TO PSEUDO-AUTO.
CA      TCP
EXTEND      # IF TIME IN RATE COMMAND EXCEEDS 4 SEC.
BZMF    LAST
CS      RCSFLAGS
MASK    PBIT
ADS     RCSFLAGS      # BIT 10 IS 1.
TCF     +4
LAST    CS      PBIT
MASK    RCSFLAGS
TS      RCSFLAGS      # BIT 10 IS 0.
CS      EDOTP
EXTEND
MP      1/ANETP      # 1/2JTACC SCALED AT 2EXP(7)/PI
DAS     A
TC      OVERSUB
EXTEND

```

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MP	25/32	# A CONTAINS TJET SCALED AT 2EXP(4)(16/25)
TS	TJP	# 4.JET TIME
CA	ABSEDOTP	
AD	-2JETLIM	# COMPARING DELTA RATE WITH 2 JET LIMIT
EXTEND		

588  $\langle$ Page LM1436 588 $\rangle \equiv$  (557b 756)

```

      BZMF      +3

      CA        SIX
      TCF       +8D
      CA        TJP
      ADS       TJP

# GOES TO PJETSLEC FOR TWO JETS
# P-JET-SELECTION-ROUTINE (ROTATION)
#
# INPUT:        NUMBERT          4,5,6 FOR WHICH PAIR OR 4 JETS
#              TJP              + FOR +P ROTATION
#
# OUTPUT:       CHANNEL 6
#              PJUMPADR         FOR P-AXIS SKIP
#              (JTLST CALL)     (SMALL TJP)
#
# ORDER OF POLICIES TRIED IN CASE OF FAILURE.
#      +P      -P
#      7,15    8,16
#      4,12    3,11
#      4,7     8,11
#      7,12    11,16
#      12,15   3,16
#      4,15    3,8
#      ALARM   ALARM

      CA        AORBSYST
      MASK      DAPBOOLS
      CCS       A
      CA        ONE
      AD        FOUR
      TS        NUMBERT
PJETSLEC      CA        ONE
              TS        L
              CCS       TJP
              TCF       +5
              TCF       JETSOFF
              TCF       +2
              TCF       JETSOFF
              ZL
              AD        ONE
              TS        ABSTJ
              LXCH      ROTINDEX
              TC        SELECTP

```

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CS	SIX
AD	NUMBERT
EXTEND	
BZF	+2
CS	TWO

590 *(Page LM1437 590)*≡ (557b 756)

```

AD      FOUR
TS      NO.PJETS
CA      POLYTEMP
TC      WRITEP
CS      ABSTJ
AD      +150MST6
EXTEND
BZMF    QRAXIS      # GO TO QRAXIS OR TO GTS.

AD      -136MST6
EXTEND
BZMF    +5

ADS     ABSTJ
INDEX   ROTINDEX
CA      MINTIMES
TS      TJP

CA      ABSTJ
ZL
INHINT
DXCH    T6FURTHA
TC      IBNKCALL
CADR    JTLST
CS      BIT12
MASK    RCSFLAGS
TS      RCSFLAGS    # BIT 12 SET TO 0.
TC      ALTSYST
TCF     QRAXIS

ALTSYST CA      DAPBOOLS    # ALTERNATE P-AXIS JETS
        TS      L
        CA      AORBSYST
        EXTEND
        RXOR    LCHAN
        TS      DAPBOOLS
        RELINT
        TC      Q

DKALT   TC      ALTSYST

JETSOFF TC      WRITEP  -1
        CA      ZERO
        TS      TJP
        TCF     QRAXIS

```

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# (NOTE -- M13 = 1 IDENTICALLY IMPLIES NULL MULTIPLICATION.)

CALCPERR	CA	CDUY	# P-ERROR CALCULATION.
	EXTEND		
	MSU	CDUYD	# CDU VALUE -- ANGLE DESIRED (Y-AXIS)

592 (Page LM1438 592)≡

(557b 756)

```

EXTEND
MP      M11      # (CDUY-CDUYD)M11 SCALED AT PI RADIANS
XCH     E        # SAVE FIRST TERM (OF TWO)
CA      CDUX     # THIRD COMPONENT
EXTEND
MSU     CDUXD    # CDU VALUE -- ANGLE DESIRED (X-AXIS)
#
# EXTEND
# MP      M13
# AD      DELPEROR # KALCMANU INTERFACE ERROR.
# ADS     E        # SAVE SUM OF TERMS. COULD BE OVERFLOW.
# XCH     PERROR   # SAVE P-ERROR FOR EIGHT-BALL DISPLAY.
# TC      Q        # RETURN TO CALLER

# P-AXIS URGENCY FUNCTION CALCULATION.

PURGENCY TC      CALCPERR # CALCULATE P-AXIS ERRORS.
          CS      OMEGAPD  # THIS CODING IS COMMON TO BOTH LM DAP AND
          AD      OMEGAP   # SPS-BACKUP MODE.
          TS      EDOTP    # EDOTP = OMEGAP - OMEGAPD AT PI/4 RAD/SEC

          CS      ONE
          TS      AXISCTR
          CA      DAPBOOLS
          MASK    CSMDOCKD
          EXTEND
          BZF     HEADTJET
          INHINT
          TC      IBNKCALL # IF CSMDOCKD = 1, GOT TO DOCKED RCS LOGIC
          CADR    SPSRCS

          CA      TJP
          EXTEND
          BZF     DKALT    # IF TJP = ZERO, CHANGE AORBSYST.
          RELINT
          TCF     PJETSLEC -6 # SELECT AORBSYST AND USE TWO JETS.
HEADTJET CA      ZERO
          TS      SENSETYP
          INHINT
          TC      IBNKCALL
          CADR    TJETLAW
          RELINT

          CS      FIREFCT
          AD      -FOURDEG
          EXTEND

```



BZMF PJETSLEC -6  
CCS TJP  
TCF +2  
TCF JETSOFF

593  $\langle \text{Page LM1439 593} \rangle \equiv$  (557b 756)

	AD	-160MST6
	EXTEND	
	BZMF	PJETSLEC -6
	CA	SIX
	TCF	PJETSLEC -1
-160MST6	DEC	-256
-FOURDEG	DEC	-.08888

594 (Page LM1440 594)≡

(557b 756)

```

# JET POLICY CONSTRUCTION SUBROUTINE
#
# INPUT:          ROTINDEX, NUMBERT
#
# OUTPUT:         POLYTEMP (JET POLICY)
#
# THIS SUBROUTINE SELECT A SUBSET OF THE DESIRED JETS WHICH HAS NO FAILURE

SELECTP          CA      SIX
                  TS      TEMPNUM
                  INDEX   NUMBERT
                  CA      TYPEP
                  INDEX   ROTINDEX
                  MASK     JETSALL
                  TS      POLYTEMP
                  MASK     CH6MASK
                  CCS      A
                  TCF      +2
                  TC       Q
                  CCS      TEMPNUM
                  TCF      +4
                  TC       ALARM
                  OCT      02003
                  TCF      JETSOFF      # ***** TCF ALARMJET *****
SELECTYZ          TS      NUMBERT
                  TCF      SELECTP +1
                  -1      TCF      ABORTYZ +2
JETSALL          OCT      00252
                  OCT      00125      # +P
                  OCT      00140      # -Y
                  OCT      00006      # -Z
                  OCT      00220      # +Y
                  OCT      00011      # +Z
                  OCT      00151      # +V
TYPEP            OCT      00146      # -U
                  OCT      00226      # -V
                  OCT      00231      # +U
                  OCT      00151      # +V
                  OCT      00132      # 1-3
                  OCT      00245      # 2-4
                  OCT      00377      # ALL
INDXYZ           =      -136MST6
-136MST6        DEC      -218
                  DEC      4
                  DEC      2

```

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OCT 07776  
DEC 5  
DEC 9  
DEC 10  
OCT 07776  
DEC 3

595 <Page LM1441 595>≡

(557b 756)

DEC 8  
DEC 7  
OCT 07776  
OCT 07776  
OCT 07776  
OCT 07776  
OCT 07776  
+150MST6 DEC 240  
07400OCT OCT 07400

# THESE INDEXES OF MASK JETSALL WILL  
# CHANGE THE INSTRUCTION AT SELECTP +4  
# TO BE TC JETSALL -1  
# ONLY USED FOR TRANSLATION FAILURE

# T-JET LAW FIXED CONSTANTS

NORMSCL OCT 266  
-100MS DEC -.1  
200MS DEC .2  
25/32 = PRI031  
BITS8,9 OCTAL 00600  
1/40 DEC .02500  
MINTIMES DEC -22  
DEC 22  
PSKIPADR GENADR SKIPPAXS

# (DEC .78125)

# GOES TO Q,R-AXES RCS AUTOPILOT

QRAXIS CS OMEGARD  
AD OMEGAR  
TC OVERSUB  
TS EDOTR  
CS OMEGAQD  
AD OMEGAQ  
TC OVERSUB  
TS EDOTQ  
EXTEND  
DCA QERRCALL  
DTCB

EBANK= AOSQ  
QERRCALL 2CADR CALLQERR

### 1.35 q-r axis rcs autopilot

$$\begin{aligned}
 596 \quad \langle q\text{-}r \text{ axis rcs autopilot } 596 \rangle \equiv & \quad (7) \\
 & \langle \text{Page LM1442 } 597 \rangle \\
 & \langle \text{Page LM1443 } 599 \rangle \\
 & \langle \text{Page LM1444 } 601 \rangle \\
 & \langle \text{Page LM1445 } 603 \rangle \\
 & \langle \text{Page LM1446 } 605 \rangle \\
 & \langle \text{Page LM1447 } 607 \rangle \\
 & \langle \text{Page LM1448 } 609 \rangle \\
 & \langle \text{Page LM1449 } 611 \rangle \\
 & \langle \text{Page LM1450 } 612 \rangle \\
 & \langle \text{Page LM1451 } 613 \rangle \\
 & \langle \text{Page LM1452 } 614 \rangle \\
 & \langle \text{Page LM1453 } 616 \rangle \\
 & \langle \text{Page LM1454 } 618 \rangle \\
 & \langle \text{Page LM1455 } 620 \rangle \\
 & \langle \text{Page LM1456 } 622 \rangle \\
 & \langle \text{Page LM1457 } 624 \rangle \\
 & \langle \text{Page LM1458 } 626 \rangle \\
 & \langle \text{Page LM1459 } 627\text{a} \rangle
 \end{aligned}$$

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597 (Page LM1442 597)≡

(596 758)

```
BANK      17
SETLOC    DAPS2
BANK

EBANK=    CDUXD

COUNT*   $$/DAPQR

CALLQERR   CA      BIT13      # CALCULATE Q,R ERRORS UNLESS THESE AXES
          EXTEND      # ARE IN MANUAL RATE COMMAND.
          RAND    CHAN31
          CCS      A
          TCF      +5          # IN AUTO COMPUTE Q,R ERRORS
          CS       DAPBOOLS    # IN MANUAL RATE COMMAND?
          MASK     OURRCBIT
          EXTEND
          BZF      Q,RORGTS    # IF SO BYPASS CALCULATION OF ERRORS.
          TC       QERRCALC

Q,RORGTS   CCS      COTROLER   # CHOOSE CONTROL SYSTEM FOR THIS DAP PASS:
          TCF      GTOGTGS     #       GTS (ALTERNATES WITH RCS WHEN DOCKED)
          TCF      TRYGTGS     #       GTS IF ALLOWED, OTHERWISE RCS
RCS        CAF      ZERO       #       RCS (TRYGTGS MAY BRANCH TO HERE)
          TS       COTROLER

          DXCH     EDOTQ
          TC       ROT-TOUV
          DXCH     OMEGAU

# X - TRANSLATION
#
# INPUT:    BITS 7,8 OF CH31 (TRANSLATION CONTROLLER)
#           ULLAGER
#           APSFLAG, DRIFTBIT
#           ACC40R2X, ACRBTRAN
#
# OUTPUT:   NEXTU, NEXTV      CODES OF TRANSLATION FOR AFTER ROTATION
#           SENSETYP          TELL ROTATION DIRECTION AND DESIRE
#
# X-TRANS POLICIES ARE EITHER 4 JETS OR A DIAGONAL PAIR.  IN 2-JET TRANSLATION THE SYSTEM IS SP
# WILL OVERRIDE THIS SPECIFICATION.  AN ALARM RESULTS WHEN NO POLICY IS AVAILABLE BECAUSE OF FA

SENSEGET   CA      BIT7        # INPUT BITS OVERRIDE THE INTERNAL BITS
          EXTEND      # SENSETYP WILL NOT OPPOSE ANYTRANS
          RAND    CHAN31
```

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EXTEND  
BZF +XORULGE

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599 (Page LM1443 599)≡

(596 758)

	CA	BIT8	
	EXTEND		
	RAND	CHAN31	
	EXTEND		
	BZF	-XTRANS	
	CA	ULLAGER	
	MASK	DAPBOOLS	
	CCS	A	
	TCF	+XORULGE	
	TS	NEXTU	# STORE NULL TRANSLATION POLICIES
	TS	NEXTV	
	CS	DAPBOOLS	# BURNING OR DRIFTING?
	MASK	DRIFTBIT	
	EXTEND		
	BZF	TSENSE	
	CA	FLGWRD10	# DPS (INCLUDING DOCKED) OR APS?
	MASK	APSFLBIT	
	CCS	A	
	CAF	TWO	# FAVOR +X JETS DURING AN APS BURN.
TSENSE	TS	SENSETYP	
	TCF	QRCONTRL	
+XORULGE	CAF	ONE	
-XTRANS	AD	FOUR	
	TS	ROTINDEX	
	AD	NEG3	
	TS	SENSETYP	# FAVOR APPROPRIATE JETS DURING TRANS.
	CA	DAPBOOLS	
	MASK	ACC40R2X	
	CCS	A	
	TCF	TRANS4	
	CA	DAPBOOLS	
	MASK	AORBTRAN	
	CCS	A	
	CA	ONE	# THREE FOR B
	AD	TWO	# TWO FOR A SYSTEM 2 JET X TRANS
TSNUMBRT	TS	NUMBERT	
	TC	SELCTSUB	
	CCS	POLYTEMP	
	TCF	+3	

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	TC	ALARM
	OCT	02002
	CA	00314OCT
	MASK	POLYTEMP
TSNEXTS	TS	NEXTU



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(596 758)

CS 003140CT  
MASK POLYTEMP  
TS NEXTV

# Q,R-AXES RCS CONTROL MODE SELECTION  
# SWITCHES INDICATION WHEN SET  
# BIT13/CHAN31 AUTO, GO TO ATTSTEER  
# PULSES MINIMUM IMPULSE MODE  
# (OTHERWISE) RATE COMMAND/ATTITUDE HOLD MODE

QRCONTRL CA BIT13 # CHECK MODE SELECT SWITCH.  
EXTEND  
RAND CHAN31 # BITS INVERTED  
CCS A  
TCF ATTSTEER  
CHKBIT10 CAF PULSES # PULSES = 1 FOR MIN IMP USE OF RHC  
MASK DAPBOOLS  
EXTEND  
BZF CHEKSTIK # IN ATT-HOLD/RATE-COMMAND IF BIT10=0

# MINIMUM IMPULSE MODE

INHINT  
TC IBNKCALL  
CADR ZATTEROR  
CA ZERO  
TS QERROR  
TS RERROR # FOR DISPLAYS  
RELINT

EXTEND  
READ CHAN31  
TS TEMP31 # IS EQUAL TO DAPTEMP1  
CCS OLDQRMIN  
TCF CHECKIN

FIREQR CA TEMP31  
MASK BIT1  
EXTEND  
BZF +QMIN  
  
CA TEMP31  
MASK BIT2  
EXTEND  
BZF -QMIN

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CA	TEMP31
MASK	BIT5

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603 (Page LM1445 603)≡

(596 758)

```
EXTEND
BZF      +RMIN

CA      TEMP31
MASK    BIT6
EXTEND
BZF      -RMIN

TCF      XTRANS

CHECKIN  CS      TEMP31
          MASK    OCT63
          TS      OLDQRMIN
          TCF      XTRANS

+QMIN    CA      14MS
          TS      TJU
          CS      14MS
          TCF      MINQR
-QMIN    CS      14MS
          TS      TJU
          CA      14MS
          TCF      MINQR
+RMIN    CA      14MS
          TCF      +2
-RMIN    CS      14MS
          TS      TJU
MINQR    TS      TJV
          CA      MINADR
          TS      RETJADR
          CA      ONE
          TS      OLDQRMIN
MINRTN   TS      AXISCTR
          CA      DAPBOOLS
          MASK    CSMDOCKD
          EXTEND
          BZF      MIMRET
          INDEX   AXISCTR
          CCS      TJU
          CA      60MS
          TCF      +2
          CS      60MS
          INDEX   AXISCTR
          TS      TJU
MIMRET   CA      DAPBOOLS
```

# IF DOCKED, USE 60MS MINIMUM IMPULSE

MASK	AORBTRAN
CCS	A
CA	ONE
AD	TWO
TS	NUMBERT

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605 (Page LM1446 605)≡

(596 758)

TCF AFTERTJ  
  
60MS DEC 96  
MINADR GENADR MINRTN  
OCT63 OCT 63  
14MS = +TJMINT6  
  
TRANS4 CA FOUR  
TCF TSNMBRT

# RSB 2009 -- was 96.0.

# RATE COMMAND MODE:

#

# DESCRIPTION (SAME AS P-AXIS)

CHEKSTIK TS INGTS  
CS ONE  
TS COTROLER  
CA BIT15  
MASK CH31TEMP  
EXTEND  
BZF RHCACTIV  
CA OURRCBIT  
MASK DAPBOOLS  
EXTEND  
BZF STILLRCS  
CS BIT9  
MASK RCSFLAGS  
TS RCSFLAGS  
TCF DAMPING  
  
40CYCL OCT 50  
1/10S OCT 1  
LINRAT DEC 46

# NOT IN GTS WHEN IN ATT HOLD  
# 1/ACCS WILL DO THE NULLING DRIVES  
# COME BACK TO RCS NEXT TIME

# BRANCH IF OUT OF DETENT.  
# \*\*\*\*\*  
# \*IN DETENT\* CHECK FOR MANUAL CONTROL  
# \*\*\*\*\* LAST TIME.

# BIT 9 IS 0.

# =====

DAMPING CA ZERO  
TS SAVEHAND  
TS SAVEHAND +1  
  
RHCACTIV CCS SAVEHAND  
TCF +3  
TCF +2  
TCF +1  
DOUBLE  
DOUBLE  
AD LINRAT

# \*\*\*\*\*  
# Q,R MANUAL CONTROL WC = A\*(B+|D|)\*D  
# \*\*\*\*\*  
# WHERE  
#  
# WC = COMMANDED ROTATIONAL RATE

EXTEND		#	A	=	QUADRATIC SENSITIVITY FACTOR
MP	SAVEHAND	#	B	=	LINEAR/QUADRATIC SENSITIVITY
CA	L	#	D	=	ABS. VALUE OF DEFLECTION
EXTEND		#	D	=	HAND CONTROLLER DEFLECTION
MP	STIKSENS				
XCH	QLAST	#	COMMAND Q	RATE, SCALED 45 DEG/SEC	
COM					

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607 (Page LM1447 607)≡

(596 758)

```
AD      QLAST
TS      DAPTEMP3
CCS     SAVEHAND +1
TCF     +3
TCF     +2
TCF     +1
DOUBLE
DOUBLE
AD      LINRAT
EXTEND
MP      SAVEHAND +1
CA      L
EXTEND
MP      STIKSENS
XCH     RLAST
COM
AD      RLAST
TS      DAPTEMP4
CS      QLAST      # INTERVAL.
AD      OMEGAQ
TS      QRATEDIF
CS      RLAST
AD      OMEGAR
TS      RRATEDIF
ENTERQR DXCH     QRATEDIF      # TRANSFORM RATES FROM Q,R TO U,V AXES
TC      ROT-TOUV
DXCH    URATEDIF
CCS     DAPTEMP3      # CHECK IF Q COMMAND CHANGE EXCEEDS
TC      +3            # BREAKOUT LEVEL.  IF NOT, CHECK R.
TC      +2
TC      +1
AD      -RATEDB
EXTEND
BZMF    +2
TCF     ENTERUV -2    # BREAKOUT LEVEL EXCEEDED.  DIRECT RATE.
CCS     DAPTEMP4      # R COMMAND BREAKOUT CHECK.
TC      +3
TC      +2
TC      +1
AD      -RATEDB
EXTEND
BZMF    +2
TCF     ENTERUV -2    # BREAKOUT LEVEL EXCEEDED.  DIRECT RATE.
CA      RCSFLAGS      # BREAKOUT LEVEL NOT EXCEEDED.  CHECK FOR
MASK    QRBIT         # DIRECT RATE CONTROL LAST TIME.
```

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EXTEND

BZF +2

TCF ENTERUV

TCF STILLRCS

CA 40CYCL

# CONTINUE DIRECT RATE CONTROL.

# PSEUDO-AUTO CONTROL.



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```
609  (Page LM1448 609)≡ (596 758)
      TS      TCQR
ENTERUV INHINT      # DIRECT RATE CONTROL
      TC      IBNKCALL
      FCADR    ZATTEROR
      RELINT
      CA      ZERO
      TS      DYERROR
      TS      DYERROR +1
      TS      DZERROR
      TS      DZERROR +1
      CCS      URATEDIF
      TCF      +3
      TCF      +2
      TCF      +1
      AD      TARGETDB      # IF TARGET DB IS EXCEEDED, CONTINUE
EXTEND      # DIRECT RATE CONTROL.
      BZMF     VDB
      CCS      VRATEDIF
      TCF      +3
      TCF      +2
      TCF      +1
      AD      TARGETDB
EXTEND
      BZMF     +2
      TCF      QRTIME
      CA      ZERO
      TS      VRATEDIF
      TCF      QRTIME
VDB      CCS      VRATEDIF
      TC      +3
      TC      +2
      TC      +1
      AD      TARGETDB      # IF TARGET DB IS EXCEEDED, CONTINUE
EXTEND      # DIRECT RATE CONTROL. IF NOT, FIRE AND
      BZMF     TOPSEUDO      # SWITCH TO PSEUDO-AUTO CONTROL ON NEXT
      CA      ZERO          # PASS.
      TS      URATEDIF
QRTIME     CA      TCQR      # DIRECT RATE TIME CHECK.
EXTEND
      BZMF     +5          # BRANCH IF TIME EXCEEDS 4 SEC.
      CS      RCSFLAGS
      MASK     QRBIT
      ADS      RCSFLAGS      # BIT 11 IS 1.
      TC      +4
TOPSEUDO   CS      QRBIT
```

MASK	RCSFLAGS	
TS	RCSFLAGS	# BIT 11 IS 0.
CA	HANDADR	
TS	RETJADR	
CA	ONE	

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```
611  (Page LM1449 611)≡ (596 758)
      BACKHAND      TS      AXISCTR
                        CA      FOUR
                        TS      NUMBERT
                        INDEX  AXISCTR
                        INDEX  SKIPU
                        TCF     +1
                        CA      FOUR
                        INDEX  AXISCTR
                        TS      SKIPU
                        TCF     LOOPER
                        INDEX  AXISCTR
                        CCS     URATEDIF
                        CA      ZERO
                        TCF     +2
                        CA      ONE
                        INDEX  AXISCTR
                        AD      AXISDIFF
                        #      INDEX  AXIS  QUANTITY
                        #      0      -U    1/JETACC-AOSU
                        #      1      +U    1/JETACC+AOSU
                        #      16     -V    1/JETACC-AOSV
                        #      17     +V    1/JETACC+AOSV
                        # JETACC = 2 JET ACCELERATION (1 FOR FAIL)
                        INDEX  A
                        CS      1/ANET2 +1
                        EXTEND
                        INDEX  AXISCTR
                        MP      URATEDIF
                        TS      Q
                        DAS     A
                        AD      Q
                        TS      A
                        TCF     +2
                        CA      Q
                        # RIGHT SIGN AND BIGGER THAN 150MS
SETTIME  INDEX  AXISCTR
                        TS      TJU
                        TCF     AFTERTJ
                        # SCALED AT 10.67 WHICH IS CLOSE TO 10.24
ZEROTJ   CA      ZERO
                        TCF     SETTIME
HANDADR   GENADR  BACKHAND
# GTS WILL BE TRIED IF
# 1. USEQRJTS = 0,
# 2. ALLOWGTS POS,
# 3. JETS ARE OFF (Q,R-AXES)
```

TRYGTS	CAF	USEQRJTS	# IS JET USE MANDATORY. (AS LONG AS
	MASK	DAPBOOLS	# USEQRJTS BIT IS NOT BIT 15, CCS IS SAFE.)
	CCS	A	
	TCF	RCS	
	CCS	ALLOWGTS	# NO. DOES AOSTASK OK CONTROL FOR GTS?

612    *<Page LM1450 612>*≡ (596 758)

	TCF	+2	
	TCF	RCS	
	EXTEND		
	READ	CHAN5	
	CCS	A	
	TCF	CHKINGTS	
GOTOGTS	EXTEND		
	DCA	GTSCADR	
	DTCB		

CHKINGTS	CCS	INGTS	# WAS THE TRIM GIMBAL CONTROLLING
	TCF	+2	# YES. SET UP A DAMPED NULLING DRIVE.
	TCF	RCS	# NO. NULLING WAS SET UP BEFORE. DO F
	INHINT		
	TC	IBNKCALL	
	CADR	TIMEGMBL	
	RELINT		
	CAF	ZERO	
	TS	INGTS	
	TCF	RCS	

	EBANK=	CDUXD	
GTSCADR	2CADR	GTS	

613 (Page LM1451 613)≡

(596 758)

# SUBROUTINE TO COMPUTE Q,R-AXES ATTITUDE ERRORS FOR USE IN THE RCS AND GTS CONTROL LAWS AND TH

```

QERRCALC      CAE      CDUY      # Q-ERROR CALCULATION
EXTEND
MSU      CDUYD      # CDU ANGLE -- ANGLE DESIRED (Y-AXIS)
TS      DAPTEMP1      # SAVE FOR RERRCALC
EXTEND
MP      M21      # (CDUY-CDUYD)*M21 SCALED AT PI RADIANS
TS      E
CAE      CDUZ      # SECOND TERM CALCULATION:
EXTEND
MSU      CDUZD      # CDU ANGLE -ANGLE DESIRED (Z-AXIS)
TS      DAPTEMP2      # SAVE FOR RERRCALC
EXTEND
MP      M22      # (CDUZ-CDUZD)*M22 SCALED AT PI RADIANS
AD      DELQEROR      # KALCMANU INERFACE ERROR
AD      E
XCH      QERROR      # SAVE Q-ERROR FOR EIGHT-BALL DISPLAY.

RERRCALC      CAE      DAPTEMP1      # R-ERROR CALCULATION:
EXTEND      # CDU ANGLE -ANGLE DESIRED (Y-AXIS)
MP      M31      # (CDUY-CDUYD)*M31 SCALED AT PI RADIANS
TS      E
CAE      DAPTEMP2      # SECOND TERM CALCULATION:
EXTEND      # CDU ANGLE -ANGLE DESIRED (Z-AXIS)
MP      M32      # (CDUZ-CDUZD)*M32 SCALED AT PI RADIANS
AD      DELREROR      # KALCMANU INERFACE ERROR
AD      E
XCH      RERROR      # SAVE R-ERROR FOR EIGHT-BALL DISPLAY.
TC      Q

```

614 &lt;Page LM1452 614&gt;≡

(596 758)

# "ATTSTEER" IS THE ENTRY POINT FOR Q,R-AXES (U,V-AXES) ATTITUDE CONTROL USING THE R

ATTSTEER           EQUALS   STILLRCS           # "STILLRCS" IS THE RCS EXIT FROM TRYGTS.

STILLRCS           CA       RERROR  
                   LXCH     A  
                   CA       QERROR  
                   TC       ROT-TOUV  
                   DXCH     UERROR

# PREPARES CALL TO TJETLAW (OR SPSRCS(DOCKED))

# PREFORMS SKIP LOGIC ON U OR Y AXIS IF NEEDED.

TJLAW           CA       TJLAWADR  
                   TS       RETJADR  
                   CA       ONE  
                   TS       AXISCTR  
                   INDEX   AXISCTR  
                   INDEX   SKIPU  
                   TCF      +1  
                   CA       FOUR  
                   INDEX   AXISCTR  
                   TS       SKIPU  
                   TCF      LOOPER  
                   INDEX   AXISCTR  
                   CA       UERROR  
                   TS       E  
                   INDEX   AXISCTR  
                   CA       OMEGAU  
                   TS       EDOT  
                   CA       DAPBOOLS  
                   MASK     CSMDOCKD  
                   CCS      A  
                   TCF      +3  
                   TC       TJETLAW  
                   TCF      AFTERTJ  
                   +3      CS       DAPBOOLS  
                           MASK    USEQRJTS  
                           CCS     A  
                           TS       COTROLER  
                           INHINT  
                           TC       IBNKCALL  
                           CADR    SPSRCS  
                           RELINT  
                           CAF     FOUR

# DOCKED. IF GIMBAL USABLE DO GTS CONTROL  
 #       ON THE NEXT PASS.  
 # USEQRJTS BIT MUST NOT BE BIT 15.  
 # GIMBAL USABLE. STORE POSITIVE VALUE.

# DETERMINE RCS CONTROL

# ALWAYS CALL FOR 2-JET CONTROL ABOUT U,V.

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```

          TS      NUMBERT      # FALL THROUGH TO JET SELECTION, ETC.

# Q,R-JET-SELECTION-LOGIC
#
# INPUT:      AXISCTR      0,1 FOR U,V
#            SNUFFBIT      ZERO TJETU,V AND TRANS. ONLY IF SET IN A DPS BURN
```

616  $\langle$ Page LM1453 616 $\rangle \equiv$ 

(596 758)

```

#          TJU,TJV          JET TIME SCALED 10.24 SEC.
#          NUMBERT          INDICATES NUMBER OF JETS AND TYPE OF POLICY
#          RETJADR          WHERE TO RETURN TO
#
# OUTPUT:    NO.U(V)JETS    RATE DERIVATION FEEDBACK
#            CHANNEL 5
#            SKIPU,SKIPV    FOR LESS THAN 150MS FIRING
#
# NOTES:     IN CASE OF FAILURE IN DESIRED ROTATION POLICY, "ALL" UNFAILED
#            JETS OF THE DESIRED POLICY ARE SELECTED.  SINCE THERE ARE ONLY
#            TWO JETS, THIS MEANS THE OTHER ONE OR NONE.  THE ALARM IS SENT
#            IF NONE CAN BE FOUND.
#
#            TIMES LESS THAN 14 MSEC ARE TAKEN TO CALL FOR A SINGLE-JET
#            MINIMUM IMPULSE, WITH THE JET CHOSEN SEMI-RANDOMLY.

AFTERTJ      CA      FLAGWRD5      # IF SNUFFBIT SET DURING A DPS BURN GO TO
            MASK     SNUFFBIT      # XTRANS; THAT IS, INHIBIT CONTROL.
            EXTEND
            BZF      DOROTAT
            CS       FLGWRD10
            MASK     APSFLBIT
            EXTEND
            BZF      DOROTAT
            CA       DAPBOOLS
            MASK     DRIFTBIT
            EXTEND
            BZF      XTRANS

DOROTAT      CAF      TWO
            TS       L
            INDEX    AXISCTR
            CCS      TJU
            TCF      +5
            TCF      NOROTAT
            TCF      +2
            TCF      NOROTAT
            ZL
            AD       ONE
            TS       ABSTJ

            CA       AXISCTR
            AD       L
            TS       ROTINDEX      # 0 1 2 3 = -U -V +U +V

```



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CA	ABSTJ
AD	-150MS
EXTEND	
BZMF	DOSKIP

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(596 758)

	TC	SELCTSUB	
	INDEX	AXISCTR	
	CA	INDEXES	
	TS	L	
	CA	POLYTEMP	
	INHINT		
	INDEX	L	
	TC	WRITEP	
	RELINT		
	TCF	FEEDBACK	
NOROTAT	INDEX	AXISCTR	
	CA	INDEXES	
	INHINT		
	INDEX	A	
	TC	WRITEP	-1
	RELINT		
LOOPER	CCS	AXISCTR	
	TC	RETJADR	
	TCF	CLOSEOUT	
DOSKIP	CS	ABSTJ	
	AD	+TJMINT6	# 14MS
	EXTEND		
	BZMF	NOTMIN	
	ADS	ABSTJ	
	INDEX	AXISCTR	
	CCS	TJU	
	CA	+TJMINT6	
	TCF	+2	
	CS	+TJMINT6	
	INDEX	AXISCTR	
	TS	TJU	
	CCS	SENSETYP	# ENSURE MIN-IMPULSE NOT AGAINST TRANS
	TCF	NOTMIN	-1
	EXTEND		
	READ	LOSCALAR	
	MASK	ONE	
	TS	NUMBERT	

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NOTMIN	TC	SELCTSUB
	INDEX	AXISCTR
	CA	INDEXES
	INHINT	

620  $\langle$ Page LM1455 620 $\rangle \equiv$ 

(596 758)

TS T6FURTHA +1  
 CA POLYTEMP  
 INDEX T6FURTHA +1  
 TC WRITEP

CA ABSTJ  
 TS T6FURTHA  
 TC JTLST

# IN QR BANK BY NOW

RELINT

CA ZERO  
 INDEX AXISCTR  
 TS SKIPU

FEEDBACK

CS THREE  
 AD NUMBERT  
 EXTEND  
 BZMF +3

CA TWO  
 TCF +2  
 CA ONE  
 INDEX AXISCTR  
 TS NO.UJETS  
 TCF LOOPER

XTRANS

CA ZERO  
 TS TJU  
 TS TJV  
 CA FOUR  
 INHINT  
 XCH SKIPU  
 EXTEND  
 BZF +2  
 TC WRITEU -1  
 CA FOUR  
 XCH SKIPV  
 RELINT

EXTEND  
 BZF CLOSEOUT  
 INHINT  
 TC WRITEV -1  
 RELINT

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	TCF	CLOSEOUT
INDEXES	DEC	4
	DEC	13
+TJMINT6	DEC	22

622 (Page LM1456 622)≡

(596 758)

```

-150MS      DEC      -240
BIT8,9      OCT      00600
SCLNORM     OCT      266
TJLAWADR    GENADR   TJLAW   +3      # RETURN ADDRESS FOR RCS ATTITUDE CONTROL

```

# THE JET LIST:

# THIS IS A WAITLIST FOR T6RUPTS.

#

# CALLED BY:

```

#          CA      TJ          # TIME WHEN NEXT JETS WILL BE WRITTEN
#          TS      T6FURTHA
#          CA      INDEX      # AXIS TO BE WRITTEN AT TJ (FROM NOW)
#          TS      T6FURTHA +1
#          TC      JTLST
#

```

# EXAMPLE -- U-AXIS AUTOPILOT WILL WRITE ITS ROTATION CODE OF

# JETS INTO CHANNEL 5. IF IT DESIRES TO TURN OFF THIS POLICY WITHIN

# 150MS AND THEN FIRE NEXTU, A CALL TO JTLST IS MADE WITH T6FURTHA

# CONTAINING THE TIME TO TURN OFF THE POLICY, T6FURTHA +1 THE INDEX

# OF THE U-AXIS(4), AND NEXTU WILL CONTAIN THE "U-TRANS" POLICY OR ZERO.

#

# THE LIST IS EXACTLY 3 LONG. (THIS LEADS UP TO SKIP LOGIC AND 150MS LIMIT)

# THE INPUT IS THE LAST MEMBER OF THE LIST.

#

# RETURNS BY:

# + TC Q

#

# DEFINITIONS: (OUTPUT)

```

#      TIME6      TIME OF NEXT RUPT
#      T6NEXT      DELTA TIME TO NEXT RUPT
#      T6FURTHA     DELTA TIME FROM 2ND TO LAST RUPT
#      NXT6ADR      AXIS INDEX      0 -- P-AXIS
#      T6NEXT +1     AXIS INDEX      4 -- U-AXIS
#      T6FURTHA +1   AXIS INDEX      13 -- V-AXIS

```

JTLST CS T6FURTHA

AD TIME6

EXTEND

BZMF MIDORLST # TIME6 -- TI IS IN A

LXCH NXT6ADR

DXCH T6NEXT

DXCH T6FURTHA

TS TIME6

LXCH NXT6ADR

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TURNON

CA BIT15  
EXTEND  
WOR CHAN13  
TC Q

624 &lt;Page LM1457 624&gt;≡

(596 758)

```

MIDORLST      AD      T6NEXT
               EXTEND
               BZMF     LASTCHG      # TIME6 + T6NEXT - T IS IN A

               LXCH     T6NEXT  +1
               DXCH     T6FURTHA
               EXTEND
               SU       TIME6
               DXCH     T6NEXT

               TC       Q

LASTCHG        CS      A
               AD      NEGO
               TS      T6FURTHA

               TC      Q

```

```

# ROT-TOUV IS ENTERED WITH THE Q-COMPONENT OF THE QUANTITY TO BE TRANSFORMED IN A AND
# ROT-TOUV TRANSFORMS THE QUANTITY INTO THE NON-ORTHOGONAL U-V AXIS SYSTEM.  IN THE U
# PRODUCED FROM RCS JET FIRINGS.  AT THE COMPLETION OF ROT-TOUV, THE U-COMPONENT OF T
# A AND THE V-COMPONENT IS IN L.

```

```

ROT-TOUV      LXCH     ROTEMP2      # (R) IS PUT INTO ROTEMP2
               EXTEND
               MP       COEFFQ
               XCH      ROTEMP2      # (R) GOES TO A AND COEFFQ.(Q) TO ROTEMP2
               EXTEND
               MP       COEFFR
               TS       L            # COEFFR.(R) IS PUT INTO L
               AD       ROTEMP2
               TS       ROTEMP1      # COEFFQ.(Q)+COEFFR.(R) IS PUT IN ROTEMP1
               TCF      +4
               INDEX    A            # COEFFQ.(Q) + COEFFR.(R) HAS OVERFLOWED
               CS       LIMITS       # AND IS LIMITED TO POSMAX OR NEGMAX
               TS       ROTEMP1
               CS       ROTEMP2
               AD       L            # -COEFFQ.(Q) + COEFFR.(R) IS NOW IN A
               TS       7
               TCF      +3
               INDEX    A            # -COEFFQ.(Q) + COEFFR.(R) HAS OVERFLOWED
               CS       LIMITS       # AND IS LIMITED TO POSMAX OR NEGMAX
               LXCH     ROTEMP1      # COEFFQ.(Q) + COEFFR.(R) IS PUT INTO L
               TC       Q
SELCTSUB       INDEX    ROTINDEX

```



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CA	ALLJETS
INDEX	NUMBERT
MASK	TYPEPOLY
TS	POLYTEMP

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(596 758)

	MASK	CH5MASK			
	CCS	A			
	TCF	+2			
	TC	Q			
FAILLOOP	CA	THREE			
	TS	NUMBER			
	INDEX	ROTINDEX			
	CA	ALLJETS			
	INDEX	NUMBER			
	MASK	TYPEPOLY			
	TS	POLYTEMP			
	MASK	CH5MASK			
	EXTEND				
	BZF	FAILLOOP -2			
	CCS	NUMBER			
	TCF	FAILLOOP			
	INDEX	AXISCTR			
	TS	TJU			
	TC	ALARM			
	OCT	02004			
	TCF	NOROTAT			
ALLJETS	OCT	00110	#	-U	6 13
	OCT	00022	#	-V	2 9
	OCT	00204	#	+U	5 14
	OCT	00041	#	+V	1 10
TYPEPOLY	OCT	00125	#	-X	1 5 9 13
	OCT	00252	#	+X	2 6 10 14
	OCT	00146	#	A	2 5 10 13
	OCT	00231	#	B	1 6 9 14
	OCT	00377	#	ALL	1 2 5 6 9 10 13 14

# THE FOLLOWING SETS THE INTERRUPT FLIP-FLOP AS SOON AS POSSIBLE, WHICH PERMITS A RE

CLOSEOUT	CA	ADRRUPT
	TC	MAKERUPT

ADRRUPT	ADRES	ENDJASK
---------	-------	---------

ENDJASK	DXCH	DAPARUPT
	DXCH	ARUPT
	DXCH	DAPBQRPT
	XCH	BRUPT
	LXCH	Q

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CAF        NEGMAX        # NEGATIVE DAPZRUPT SIGNALS JASK IS OVER.  
DXCH       DAPZRUPT  
DXCH       ZRUPT  
TCF        NOQRSM

627a         $\langle$ Page LM1459 627a $\rangle \equiv$  (596 758)

BLOCK     3  
SETLOC    FFTAG6  
BANK  
  
COUNT\*   \$\$/DAP  
  
MAKERUPT   EXTEND  
EDRUPT    MAKERUPT

1.36    tjet law

627b         $\langle$ tjet law 627b $\rangle \equiv$  (7)

$\langle$ Page LM1460 628 $\rangle$   
 $\langle$ Page LM1461 630 $\rangle$   
 $\langle$ Page LM1462 632 $\rangle$   
 $\langle$ Page LM1463 634 $\rangle$   
 $\langle$ Page LM1464 636 $\rangle$   
 $\langle$ Page LM1465 638 $\rangle$   
 $\langle$ Page LM1466 640 $\rangle$   
 $\langle$ Page LM1467 641 $\rangle$   
 $\langle$ Page LM1468 642 $\rangle$   
 $\langle$ Page LM1469 644 $\rangle$

628 (Page LM1460 628)≡

(627b 772)

```

# PROGRAM DESCRIPTION
# DESIGNED BY:  R. D. GOSS AND P. S. WEISSMAN
# CODED BY:  P. S. WEISSMAN, 28 FEBRUARY 1968
#
# TJETLAW IS CALLED AS A SUBROUTINE WHEN THE LEM IS NOT DOCKED AND THE AUTOPILOT IS IN
# ATTITUDE-HOLD MODE TO CALCULATE THE JET-FIRING-TIME (TJET) REQUIRED FOR THE AXIS IN
#
#      -1      INDICATES THE P-AXIS
#      +0      INDICATES THE U-AXIS
#      +1      INDICATES THE V-AXIS
#
# THE REGISTERS E AND EDOT CONTAIN THE APPROPRIATE ATTITUDE ERROR AND ERROR RATE AND
# UNBALANCED COUPLES ARE PREFERRED.  TJETLAW ALSO USES VARIOUS FUNCTIONS OF ACCELERATION
# COMPUTED IN THE 1/ACCONT SECTION OF 1/ACCS AND ARE STORED IN SUCH AN ORDER THAT THEY
# ACCESSED BY INDEXING.
#
# THE SIGN OF THE REQUIRED ROTATION IS CARRIED THROUGH TJETLAW AS ROTSENSE AND IS FIDUCIAL
# PREVIOUS TO ITS STORAGE IN THE LOCATION CORRESPONDING TO THE AXIS (TJP, TJU, OR TJV)
# TJETLAW ASSUMES WILL BE USED AS INDICATED BY THE SETTING OF NUMBERT FOR THE U- OR V-
# ASSUMED FOR THE P-AXIS ALTHOUGH FOUR JETS WILL BE FIRED WHEN FIREFCT IS MORE NEGATIVE
# (FIREFCT IS THE DISTANCE TO A SWITCH CURVE IN THE PHASE PLANE) AND A LONG FIRING IS
#
# IN ORDER TO AVOID SCALING DIFFICULTIES, SIMPLE ALGORITHMS TAGGED RUFLAW1, -2 AND -3
# ERROR AND/OR ERROR RATE ARE LARGE.
#
# CALLING SEQUENCE:
#
#      TC      TJETLAW      # (MUST BE IN JASK)
#
#      OR
#
#      INHINT      # (MUST BE IN JASK)
#
#      TC      IBNKCALL
#      CADR      TJETLAW
#      RELINT
#
# EXIT:      RETURN TO Q.
#
# INPUT:
#
#      FROM THE CALLER:  E, EDOT, AXISCTR, SENSETYP, TJP, -U, -V.
#      FROM 1/ACCONT:  48 ERASABLES BEGINNING AT BLOCKTOP (INCLUDING FLAT, ZONE3LIMIT)
#
# OUTPUT:
#
#      TJP, -U OR -V, NUMBERT (DAPTEMP5), FIREFCT (DAPTEMP3).
#
# DEBRIS:
#
#      A, L, Q, E, EDOT, DAPTEMP1-6, DAPTEMP1-4.
#
# ALARM:  NONE

```

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BANK 17  
SETLOC DAPS2  
BANK  
EBANK= TJP

630 (Page LM1461 630)≡

(627b 772)

COUNT\* \$\$/DAPTJ

TJETLAW            EXTEND                            # SAVE Q FOR RETURN.  
                      QXCH        HOLDQ

# SET INDEXERS TO CORRESPOND TO THE AXIS AND TO THE SIGN OF EDOT

INDEX    AXISCTR                            # AXISDIFF(-1)=NO OF LOCATIONS BET P AND U  
 CAF      AXISDIFF                          # AXISDIFF(0)=0  
 TS       ADRSDIF1                          # AXISDIFF(+1)=NO OF LOCATIONS BET V AND U  
  
 CAE      EDOT                              # IF EDOT NEGATIVE, PICK UP SET OF VALUES  
 EXTEND                                      #        THAT ALLOW USE OF SAME CODING AS FOR  
 BZMF     NEGEDOT                          #        POSITIVE EDOT.  
 CAE      ADRSDIF1                          # SET A SECOND INDEXER WHICH MAY BE  
 TS       ADRSDIF2                          #        MODIFIED BY A DECISION FOR MAX JETS.  
 CAF      SENSOR                           # FOR POSITIVE EDOT, ROTSENSE IS  
 TCF      SETSENSE                          #        INITIALIZED POSITIVE.

NEGEDOT        CS        E                            # IN ORDER FOR NEG EDOT CASE TO USE CODING  
                  TS        E                            #        OF POS EDOT, MUST MODIFY AS FOLLOWS:  
                  CS        EDOT                          #        1. COMPLEMENT E AND EDOT.  
                  TS        EDOT                          #        2. SET SENSE OF ROTATION TO NEGATIVE  
                  CAF      BIT1                           #        (REVERSED LATER IF NECESSARY).  
                  ADS      ADRSDIF1                      #        3. INCREMENT INDEXERS BY ONE SO THAT  
                  TS        ADRSDIF2                      #        THE PROPER PARAMETERS ARE ACCESE  
                  CS        SENSOR  
 SETSENSE       TS        ROTSENSE

# TEST MAGNITUDE OF E (ATTITUDE ERROR, SINGLE-PRECISION, SCALED AT PI RADIANS):  
 #        IF GREATER THAN (OR EQUAL TO) PI/16 RADIANS, GO TO THE SIMPLIFIED TJET ROUTIN  
 #        IF LESS THAN PI/16 RADIANS, RESCALE TO PI/4

CAE      E                                    # PICK UP ATTITUDE ERROR FOR THIS AXIS  
 EXTEND  
 MP       BIT5                                # SHIFT RIGHT TEN BITS: IF A-REGISTER IS  
 CCS       A                                    #        ZERO, RESCALE AND TEST EDOT.  
 TCF       RUFLAW2  
 TCF       SCALEE  
 TCF       RUFLAW1  
 SCALEE    CAF      BIT13                      # ERROR IS IN L SCALED AT PI/16. RESCALE  
                  EXTEND                            #        IT TO PI/4 AND SAVE IT.  
                  MP        L  
                  TS        E

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```
# TEST MAGNITUDE OF EDOT (ERROR RATE SCALED AT PI/4 RADIANS/SECOND)
#     IF GREATER THAN (OR EQUAL TO) PI/32 RADIANS/SECOND, GO TO THE SIMPLIFIED TJET ROUTINE.
#     IF LESS THAN PI/32 RADIANS/SECOND, THEN RESCALE TO PI/32 RADIANS/SECOND.

      CAE      EDOT      # PICK UP SINGLE-PRECISION ERROR-RATE
```

632 (Page LM1462 632)≡

(627b 772)

```

EXTEND
MP      BIT4
EXTEND
BZF     SCALEDOT
TCF     RUFLAW3

# *** FINELAW STARTS HERE ***

SCALEDOT    LXCH    EDOT    # EDOT IS SCALED AT PI/32 RADIANS/SECOND.

                CAE     EDOT    # COMPUTE (EDOT)(EDOT)
EXTEND
SQUARE
EXTEND
MP      BIT13
TS      EDOTSQ    # SHIFT RIGHT TWO BITS TO RESCALE TO EDOTSQ
                # TO PI(2)/2(8) RAD(2)/SEC(2).

ERRTEST     CCS      E      # DOES BIG ERROR (THREE DEG BEYOND THE
AD          -3DEG    # DEADBAND) REQUIRE MAXIMUM JETS?
TCF        +2
AD          -3DEG
EXTEND
INDEX      ADRSDIF1
SU         FIREDDB
EXTEND
BZMF       SENSTEST
MAXJETS    CAF      TWO     # IF NOT: ARE UNBALANCED JETS PREFERRED?
AD         ADRSDIF2    # IF YES: INCREMENT ADDRESS LOCATOR AND
CAF        FOUR      # SET SWITCH FOR JET SELECT LOGIC TO
TCF        TJCALC    # (ALWAYS DO THIS FOR P-AXIS)

SENSTEST   CCS      SENSETYP # DOES TRANSLATION PREFER MIN JETS.
TCF        TJCALC    # YES. USE MIN-JET PARAMETERS
TCF        MAXJETS   # NO. GET THE MAX-JET PARAMETERS.
TJCALC     TS        NUMBERT # SET TO +0,1,4 FOR (U,V-AXES) JET SELECT.

# BEGINNING OF TJET CALCULATIONS:

CS         EDOTSQ    # SCALED AT PI(2)/2(8).
EXTEND
INDEX      ADRSDIF2
MP         1/ANET1   # .5/ACC SCALED AT 2(6)/PI SEC(2)/RADIAN.
INDEX      ADRSDIF1
AD         FIREDDB   # DEADBAND SCALED AT PI/4 RADIAN.
EXTEND
SU         E         # ATTITUDE ERROR SCALED AT PI/4 RADIAN.

```



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	TS	FIREFCT	# -E-.5(EDOTSQ)/ACC-DB AT PI/4 Radian.
	EXTEND		
	BZMF	ZON1,2,3	
ZONE4,5	INDEX	ADRSDIF1	
	CAE	1/ACOAST	# .5/ACC SCALED AT 2(6)/PI WHERE

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```

EXTEND      # ACC = MAX(AMIN, AOS-).
MP          # SCALED AT PI/2(8).
AD          # SCALED AT PI/4
INDEX      ADRSDIF1
AD          COASTDB      # SCALED AT PI/4 POS. FOR NEG. INTERCEPT.
EXTEND      # TEST E+.5(EDOTSQ)/ACC+DB AT PI/4 RADIAN.
BZMF       ZONE5      # IF FUNCTION NEGATIVE, FIND TJET.
                # IF FUNCTION POSITIVE, IN ZONE 4.

# ZONE 4 IS THE COAST REGION.  HOWEVER, IF THE JETS ARE ON AND DRIVING TOWARD
#     A. THE AXIS WITHIN + OR - (DB + FLAT) FOR DRIFTING FLIGHT, OR
#     B. THE USUAL TARGET PARABOLA FOR POWERED FLIGHT
# THEN THE THRUSTERS ARE KEPT ON.

ZONE4      INDEX  AXISCTR      # IS THE CURRENT VALUE IN TJET NON-ZERO
          CS      TJETU        #     WITH SENSE OPPOSITE TO EDOT,
EXTEND      #     (I.E., ARE JETS ON AND FIRING TOWARD
MP          ROTSENSE          #     THE DESIRABLE STATE).
EXTEND
BZMF       COASTTJ      # NO.  COAST.

JETSON      CCS      FLAT      # YES.  IS THIS DRIFTING OR POWERED FLIGHT?
          TCF      DRIFT/ON    # DRIFTING.  GO MAKE FURTHER TEST.

          CS      FIREFCT      # POWERED (OR ULLAGE).  CAN TARGET PARABOLA
INDEX      ADRSDIF1          #     BE REACHED FROM THIS POINT IN THE
AD          AXISDIST          #     PHASE PLANE?
EXTEND
BZMF       COASTTJ      # NO.  SET TJET = 0.
TC          Z123COMP      # YES.  CALCULATE TJET AS THOUGH IN ZONE 1
CAE        FIREFCT      #     AFTER COMPUTING THE REQUIRED
TCF        ZONE1        #     PARAMETERS.

DRIFT/ON    INDEX      ADRSDIF1      # CAN TARGET STRIP OF AXIS BE REACHED FROM
          CS          FIREDB        #     THIS POINT IN THE PHASE PLANE?
DOUBLE
AD          FIREFCT
EXTEND
BZMF       +3
COASTTJ     CAF        ZERO      # NO.  SET TJET = 0.
          TCF        RETURNJT

          TC          Z123COMP      # YES.  CALCULATE TJET AS THOUGH IN ZONE 2
          TCF        ZONE2,3      #     OR 3 AFTER COMPUTING REQUIRED VALUES

```

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ZONE5	TS	L	# TEMPORARILY STORE FUNCTION IN L.
	CCS	ROTSENSE	# MODIFY ADRSDIF2 FOR ACCESSING 1/ANET2
	TCF	+4	# AND ACCFCTZ5, WHICH MUST BE PICKED UP
	TC	CCSHOLE	# FROM THE NEXT LOWER REGISTER IF THE
	CS	TWO	# (ACTUAL) ERROR RATE IS NEGATIVE.

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```

        ADS      ADRSDIF2

+4      CAE      L
        EXTEND
        INDEX    ADRSDIF2      # TTOAXIS AND HH ARE THE PARAMETERS UPON
        MP       ACCFCTZ5      # WHICH THE APPROXIMATIONS TO TJET ARE
        DDOUBL   #              ABASED.
        DDOUBL
        DXCH     HH             # DOUBLE PRECISION H SCALED AT 8 SEC(2).
        INDEX    ADRSDIF2
        CAE      1/ANET2        # SCALED AT 2(7)/PI SEC(2)/RAD.
        EXTEND
        MP       EDOT           # SCALED AT PI/2(5)
        TS       TTOAXIS        # SCALED AT 4 SEC.

# TEST WHETHER TJET GREATER THAN 50 MSEC.

        EXTEND
        MP       -.05AT2        # H - .05 TTOAXIS - .00125 G.T. ZERO
        AD       HH             # (SCALED AT 8 SEC(2) ).
        AD       NEG2
        EXTEND
        BZMF     FORMULA1

# TEST WHETHER TJET GREATER THAN 150 MSEC.

        CAE      TTOAXIS
        EXTEND
        MP       -.15AT2        # H - .15 TTOAXIS - .01125 G.T. ZERO
        AD       HH             # (SCALED AT 8 SEC(2) )
        AD       -.0112A8
        EXTEND
        BZMF     FORMULA2

# IF TJET GREATER THAN 150 MSEC, ASSIGN IT VALUE OF 250 MSEC, SINCE THIS
# IS ENOUGH TO ASSURE NO SKIP NEXT CSP (100 MSEC).

FULLTIME      CAF      BIT11      # 250 MSEC SCALED AT 4 SEC.

# RETURN TO CALLING PROGRAM WITH JET TIME SCALED AS TIME6 AND SIGNED.

RETURN TJ      EXTEND            # ALL BRANCHES TERMINATE HERE WITH TJET
        MP       ROTSENSE        # (SCALED AT 4 SEC) IN THE ACCUMULATOR
        INDEX    AXISCTR        # ROTSENSE APPLIES SIGN AND CHANGES SCALE.
        TS       TJETU

```

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EXTEND  
INDEX AXISCTR  
MP ACCSWU  
CAE L  
EXTEND  
BZMF +3  
CAF FOUR

# SET SWITCH FOR JET SELECT IF ROTATION IS  
# IN A SENSE FOR WHICH 1/ACCS HAS FORCED  
# A MAX-JET CALCULATION.

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```

        TS      NUMBERT
        TC      HOLDQ          # RETURN VIA SAVED Q.

# TJET = H/(.025 + TTOAXIS)    FOR TJET LESS THAN 50 MSEC.

FORMULA1      CS      -.025AT4      # .025 SEC SCALED AT 4.
               AD      TTOAXIS      # SCALED AT 4 SECONDS.
               DXCH    HH            # STORE DENOMINATOR IN FIRST WORD OF H,
               EXTEND   #            WHICH NEED NOT BE PRESERVED. PICK UP
               DV      HH            #            DP H AND DIVIDE BY DENOMINATOR.
               EXTEND
               MP      BIT14          # RESCALE TJET FROM 2 TO USUAL 4 SEC.
               TCF     CHKMINTJ      # CHECK THAT TJET IS NOT LESS THAN MINIMUM

# TJET = (H + .00375)/(0.1 + TTOAXIS)  FOR TJET GREATER THAN 50 MSEC.

FORMULA2      EXTEND
               DCA      .00375A8     # .00375 SEC(2) SCALED AT 8.
               DAS      HH            # STORE NUMERATOR IN DP H, WHICH NEED NOT
               #            BE PRESERVED.
               CAE      TTOAXIS      # SCALED AT 4 SEC.
               AD      .1AT4         # 0.1 SEC SCALED AT 4.
               DXCH    HH            # STORE DENOMINATOR IN FIRST WORD OF H,
               EXTEND   #            WHICH NEED NOT BE PRESERVED. PICK UP
               DV      HH            #            DP NUMERATOR AND DIVIDE BY DENOMINATOR
               EXTEND
               MP      BIT14          # RESCALE TJET FROM 2 TO USUAL 4 SEC.
               TCF     RETURN TJ      # END SUBROUTINE.

# SUBROUTINIZED COMPUTATIONS REQUIRED FOR ALL ENTRIES INTO CODING FOR ZONES 1, 2, AND
# REACHED BY TC FROM 3 POINTS IN TJETLAW.

Z123COMP      CS      ROTSENSE      # USED IN RETURN TJ SECTION TO RESCALE TJET
               TS      ROTSENSE      #            AS TIME6 AND GIVE IT PROPER SIGN.
               CAE      EDOT          # SCALED AT PI/2(5) RAD/SEC.
               EXTEND
               INDEX    ADRSDIF2
               MP      1/ANET1        # SCALED AT 2(7)/PI SEC(2)/RAD.
               TS      TTOAXIS        # STORE TIME-TO-AXIS SCALED AT 4 SECONDS.
               AD      -TJMAX
               EXTEND                # IS TIME TO AXIS LESS THAN 150 MSEC.
               BZMF     +2
               TCF     FULLTIME       # NO. FIRE JETS, DO NOT CALCULATE TJET.
               RETURN                # YES. GO ON TO FIND TJET

```

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ZON1,2,3 TC Z123COMP # SUBROUTINIZED PREPARATION FOR ZONE1,2,3.

# IF THE (NEG) DISTANCE BEYOND PARABOLA IS LESS THAN FLAT, USE SPECIAL

# LOGIC TO ACQUIRE MINIMUM IMPULSE LIMIT CYCLE. DURING POWERED FLIGHT

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# OR ULLAGE, FLAT = 0

```

CAE    FIREFCT    # SCALED AT PI/4 RAD.
AD      FLAT
EXTEND
BZMF    ZONE1      # NOT IN SPECIAL ZONES.

```

# FIRE FOR AXIS OR, IF CLOSE, FIRE MINIMUM IMPULSE. IF ON AXIS, COAST.

```

ZONE2,3  CS      ZONE3LIM    # HEIGHT OF MIN-IMPULSE ZONE SET BY 1/ACCS
AD      TTOAXIS    #          35 MSEC IN DRIFTING FLIGHT
EXTEND    #          ZERO WHEN TRYING TO ENTER GTS CONTROL
BZMF    ZONE3

```

ZONE2 CAE TTOAXIS # FIRE TO AXIS.

TCF RETURNTJ

ZONE3 CCS EDOT # CHECK IF EDOT IS ZERO.

CAF BIT6 # FIRE A ONE-JET MINIMUM IMPULSE.

TCF RETURNTJ # TJET = +0.

TC CCSHOLE # CANNOT BE BECAUSE NEG EDOT COMPLEMENTED.

TCF RETURNTJ # TJET = +0.

ZONE1

EXTEND

INDEX ADRSDIF1

SU AXISDIST

# SCALED AT PI/4 RAD.

EXTEND

INDEX ADRSDIF2

MP ACCFCTZ1

# SCALED AT 2(7)/PI SEC(2)/RAD.

DDOUBL

DDOUBL

DXCH HH

# DOUBLE PRECISION H SCALED AT 8 SEC(2).

# TEST WHETHER TOTAL TIME REQUIRED GREATER THAN 150 MSEC:

```

#          2          2
#      IS .5(.150 - TTOAXIS) - H  NEGATIVE (SCALED AT 8 SECONDS )

```

CAE TTOAXIS # TTOAXIS SCALED AT 4 SECONDS.

AD -TJMAX # -.150 SECOND SCALED AT 4.

EXTEND

SQUARE

EXTEND

SU

HH

# HIGH WORD OF H SCALED AT 8 SEC(2).

EXTEND

BZMF

FULLTIME

# YES. NEED NOT CALCULATE TJET.

# TEST WHETHER TIME BEYOND AXIS GREATER THAN 50 MSEC TO DETERMINE WHICH APPROXIMATION



```

CAE      HH
AD        NEG2
EXTEND
BZMF     FORMULA3

```

641  $\langle \text{Page LM1467 641} \rangle \equiv$ 

(627b 772)

# TJET = H/0.1 + TTOAXIS + .0375

FOR APPROXIMATION OVER MORE THAN 50 MSEC.

```

CAF      .1AT2      # STORE .1 SEC SCALED AT 2 FOR DIVISION.
DXCH     HH          # DP H SCALED AT 8 SEC(2) NEED NOT BE
EXTEND    #          # PRESERVED.
DV        HH          # QUOTIENT SCALED AT 4 SECONDS.
AD        TTOAXIS     # SCALED AT 4 SEC.
AD        .0375AT4    # .0375 SEC SCALED AT 4.
TCF       RETURN TJ   # END COMPUTATION.

```

# TJET - H/.025 + TTOAXIS FOR APPROXIMATION OVER LESS THAN 50 MSEC.

```

FORMULA3  CS      -.025AT2    # STORE +.25 SEC SCALED AT 2 FOR DIVISION
          DXCH     HH          # PICK UP DP H AT 8, WHICH NEED NOT BE
          EXTEND    #          # PRESERVED.
          DV        HH          # QUOTIENT SCALED AT 4 SECONDS.
          AD        TTOAXIS     # SCALED AT 4 SEC.

```

# IF COMPUTED JET TIME IS LESS THAN TJMIN, TJET IS SET TO ZERO.

# MINIMUM IMPULSES REQUIRED IN ZONE 3 ARE NOT SUBJECT TO THIS CONSTRAINT, NATURALLY.

```

CHKMINTJ  AD      -TJMIN      # IS COMPUTED TIME LESS THAN THE MINIMUM.
          EXTEND
          BZMF     COAST TJ    # YES, SET TIME TO ZERO.
          AD        TJMIN      # NO, RESTORE COMPUTED TIME.
          TCF       RETURN TJ  # END COMPUTATION.

```

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(627b 772)

# \*\*\* ROUGHLAW \*\*\*

#

# BEFORE ENTRY TO RUFLAW:

# 1. INDEXERS ADRSDIF1 AND ADRSDIF2 ARE SET ON BASIS OF AXIS, AND SIGN OF EDOT

# 2. IF EDOT WAS NEGATIVE, E AND EDOT ARE ROTATED INTO UPPER HALF-PLANE AND ROT

# 3. E IS SCALED AT PI RADIANS AND EDOT AT PI/4 RAD/SEC.

# (EXCEPT THE RUFLAW3 ENTRY WHEN E IS AT PI/4)

#

# RUFLAW1: ERROR MORE NEGATIVE THAN PI/16 RAD. FIRE TO A RATE OF 6.5 DEG/SEC (

# RUFLAW2: ERROR MORE POSITIVE THAN PI/16 RAD. FIRE TO AN OPPOSING RATE OF 6.5

# RUFLAW3: ERROR RATE GREATER THAN PI/32 RAD/SEC AND ERROR WITHIN BOUNDS. COAST

RUFLAW1	CS	RUFRATE	# DECREMENT EDOT BY .1444 RAD/SEC AT PI/4
	ADS	EDOT	# WHICH IS THE TARGET RATE
	EXTEND		
	BZMF	SMALRATE	# BRANCH IF RATE LESS THAN TARGET.
	TC	RUFSETUP	# REVERSE ROTSENSE AND INDICATE MAX JETS.
	CAE	EDOT	# PICK UP DESIRED RATE CHANGE.

RUFLAW12	EXTEND		# COMPUTE TJET
	INDEX	ADRSDIF2	# = (DESIRED RATE CHANGE)/(2-JET ACCEL
	MP	1/ANET1 +2	
	AD	-1/8	# IF TJET, SCALED AT 32 SEC, EXCEEDS
	EXTEND		# 4 SECONDS, SET TJET TO TJMAX.
	BZMF	+2	
	TCF	FULLTIME	
	EXTEND		
	BZF	FULLTIME	
	AD	BIT12	# RESTORE COMPUTED TJET TO ACCUMULATOR
	DAS	A	
	DAS	A	
	DAS	A	# RESCALED TJET AT 4 SECONDS.
	TCF	CHKMINTJ	# RETURN AS FROM FINELAW.

SMALRATE	TC	RUFSETUP +2	# SET NUMBERT AND FIREFCT FOR MAXIMUM JETS
	CCS	ROTSSENSE	
	CAF	ONE	# MODIFY INDEXER TO POINT TO 1/ANET
	TCF	+2	# CORRESPONDING TO THE PROPER SENSE.
	CAF	NEGONE	
	ADS	ADRSDIF2	
	CS	EDOT	# (.144 AT PI/4 - EDOT) = DESIRED RATE CHNG.
	TCF	RUFLAW12	

RUFLAW2	TC	RUFSETUP	# REVERSE ROTSENSE AND INDICATE MAX JETS.
---------	----	----------	---

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CAF	RUFRATE	
AD	EDOT	# (.144 AT PI/4 + EDOT) = DESIRED RATE CHNG.
TS	A	# IF OVERFLOW SKIP, FIRE FOR FULL TIME.
TCF	RUFLAW12	# OTHERWISE, COMPUTE JET TIME.
TCF	FULLTIME	

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```

RUFLAW3      TC      RUFSETUP      # EXECUTE COMMON RUFLAW SUBROUTINE.
              INDEX   ADRSDIF1
              CS      FIREDDB      # CALCULATE DISTANCE FROM SWITCH CURVE
              AD      E              # 1/ANET1*EDOT*EDOT +E - FIREDDB = 0
              EXTEND      #
              MP      BIT11          # SCALED AT 4 PI RADIANS
              XCH     EDOT
              EXTEND
              SQUARE
              EXTEND
              INDEX   ADRSDIF1
              MP      1/ANET1 +2
              AD      EDOT
              EXTEND
              BZMF    COASTTJ      # COAST IF BELOW IT.
              TCF     FULLTIME     # FIRE FOR FULL PERIOD IF ABOVE IT.

```

# SUBROUTINE USED IN ALL ENTRIES TO ROUGHLAW.

```

RUFSETUP      CS      ROTSENSE      # REVERSE ROTSENSE WHEN ENTER HERE.
              TS      ROTSENSE
              +2     CAF    FOUR      # REQUIRE MAXIMUM (2) JETS IN U,V-AXES.
              TS      NUMBERT
              CAF     NEGMAX          # SUGGEST MAXIMUM (4) JETS IN P-AXIS.
              TS      FIREFCT
              TC      Q

```

# CONSTANTS FOR TJETLAW

```

              DEC     -16            # AXISDIFF(INDEX) = NUMBER OF REGISTERS
AXISDIFF      DEC     +0            # BETWEEN STORED 1/ACCS PARAMETERS FOR
              DEC     16             # THE INDEXED AXIS AND THE U-AXIS.
SENSOR        OCT     14400         # RATIO OF TJET SCALING WITHIN TJETLAW
              # (4 SEC) TO SCALING FOR T6 (10.24 SEC)
-3DEG         DEC     -.06667       # -3.0 DEGREES SCALED AT 45.
-.0112A8      DEC     -.00141       # -.01125 SEC(2) SCALED AT 8.
.1AT4         DEC     .025          # 0.1 SECOND SCALED AT 4.
.1AT2         DEC     .05           # .1 SEC SCALED AT 2.
.0375AT4      DEC     .00938        # .0375 SEC SCALED AT 4.
-.025AT2      DEC     -.0125        # -.025 SEC SCALED AT 2.
-.025AT4      DEC     -.00625
-.05AT2       DEC     -.025
-.15AT2       DEC     -.075
.00375A8      2DEC    .00375 B-3

```

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-TJMAX	DEC	-.0375	# LARGEST CALCULATED TIME. .150 SEC AT 4.
TJMIN	DEC	.005	# SMALLEST ALLOWABLE TIME. .020 SEC AT 4.
-TJMIN	DEC	-.005	
RUFRATE	DEC	.1444	# CORRESPONDS TO TARGET RATE OF 6.5 DEG/S.

1.37 kalman filter

645

$\langle kalman\ filter\ 645 \rangle \equiv$   
 $\langle Page\ LM1470\ 646 \rangle$   
 $\langle Page\ LM1471\ 647a \rangle$

(7)

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(645 741)

```

EBANK= NO.UJETS
BANK 16
SETLOC DAPS1
BANK

COUNT* $$/DAP

RATELOOP CA TWO
TS DAPTEMP6
DOUBLE
TS Q
INDEX DAPTEMP6
CCS TJP
TCF +2
TCF LOOPRATE
AD -100MST6
EXTEND
BZMF SMALLTJU
INDEX DAPTEMP6
CCS TJP
CA -100MST6
TCF +2
CS -100MST6
INDEX DAPTEMP6
ADS TJP
INDEX DAPTEMP6
CCS TJP
CS -100MS # 0.1 AT 1
TCF +2
CA -100MS
LOOPRATE EXTEND
INDEX DAPTEMP6
MP NO.PJETS
CA L
INDEX DAPTEMP6
TS DAPTEMP1 # SIGNED TORQUE AT 1 JET-SEC FOR FILTER
EXTEND
MP BIT10 # RESCALE TO 32; ONE BIT ABOUT 2 JET-MSEC
EXTEND
BZMF NEGTOCK
STORTORK INDEX Q # INCREMENT DOWNLIST REGISTER.
ADS DOWNTORK # NOTE: NOT INITIALIZED; OVERFLOWS.

CCS DAPTEMP6
TCF RATELOOP +1

```

	SMALLTJU	TCF	ROTORQUE		
		CA	ZERO		
		INDEX	DAPTEMP6		
		XCH	TJP		
		EXTEND			
647a	$\langle \text{Page LM1471 647a} \rangle \equiv$				
		MP	ELEVEN	# 10.24 PLUS	(645 741)
		CA	L		
	ROTORQUE	TCF	LOOPRATE		
		CA	DAPTEMP2		
		AD	DAPTEMP3		
		EXTEND			
		MP	1JACCR		
		TS	JETRATER		
		CS	DAPTEMP3		
		AD	DAPTEMP2		
		EXTEND			
		MP	1JACCQ		
		TS	JETRATEQ		
		TCF	BACKP		
	-100MST6	DEC	-160		
	NEGTORK	COM			
		INCR	Q		
		TCF	STORTORK		

1.38 trim gimbal cntrol system

647b	$\langle \text{trim gimbal cntrol system 647b} \rangle \equiv$				
	$\langle \text{Page LM1472 648} \rangle$				
	$\langle \text{Page LM1473 649} \rangle$				
	$\langle \text{Page LM1474 651} \rangle$				
	$\langle \text{Page LM1475 653} \rangle$				
	$\langle \text{Page LM1476 655} \rangle$				
	$\langle \text{Page LM1477 656} \rangle$				
	$\langle \text{Page LM1478 657} \rangle$				
	$\langle \text{Page LM1479 659} \rangle$				
	$\langle \text{Page LM1480 660} \rangle$				
	$\langle \text{Page LM1481 661} \rangle$				
	$\langle \text{Page LM1482 663} \rangle$				
	$\langle \text{Page LM1483 665} \rangle$				
	$\langle \text{Page LM1484 666} \rangle$				

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BANK 21  
 EBANK= QDIFF  
 SETLOC DAPS4  
 BANK

COUNT\* \$\$/DAPGT

# CONTROL REACHES THIS POINT UNDER EITHER OF THE FOLLOWING TWO CONDITIONS ONCE THE D  
 # AUTOPILOT ARE BOTH ON:  
 # A) THE TRIM GIMBAL CONTROL LAW WAS ON DURING THE PREVIOUS Q,R-AXIS TIME5 INT  
 # INITIALIZATION WAS SET FOR TRIM GIMBAL CONTROL AND THIS IS THE FIRST PASS  
 # B) THE Q,R-AXES RCS AUTOPILOT DETERMINED THAT THE VEHICLE WAS ENTERING (OR HA  
 # ZONE WITH A SMALL OFFSET ANGULAR ACCELERATION.  
 # GTS IS THE ENTRY TO THE GIMBAL TRIM SYSTEM FOR CONTROLLING ATTITUDE ERRORS AND RATE

GTS CAF NEGONE # MAKE THE NEXT PASS THROUGH THE DAP BE  
 TS COTROLER # THROUGH RCS CONTROL,  
 CAF FOUR # AND ENSURE THAT IT IS NOT A SKIP.  
 TS SKIPU  
 TS SKIPV  
  
 CAF TWO  
 TS INGTS # SET INDICATOR OF GTS CONTROL POSITIVE.  
 TS QGIMTIMR # SET TIMERS TO 200 MSEC TO AVOID BOTH  
 TS RGIMTIMR # RUNAWAY AND INTERFERENCE BY NULLING.

# THE DRIVE SETTING ALGORITHM

#  
 # DEL = SGN(OMEGA + ALPHA\*ABS(ALPHA)/(2\*K))  
 #  
 # NEGUSUM = ERROR\*K + ALPHA\*(DEL\*OMEGA + ALPHA / (3\*K)) + DEL\*K (DEL\*OMEGA + A  
 #  
 # DRIVE = -SGN(NEGUSUM)

CA SR # SAVE THE SR. SHIFT IT LEFT TO CORRECT  
 AD A # FOR THE RIGHT SHIFT DUE TO EDITING.  
 TS SAVESR

GTSGO+DN CAF TWO # SET INDEXER FOR R-AXIS CALCULATIONS.  
 TCF GOQTRIMG +1

GOQTRIMG CAF ZERO # SET INDEXER FOR Q-AXIS CALCULATIONS  
 TS QRCNTR



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```

# RSB 2009 -----
# Everything between this line and the similar line below was simply filled-in
# as-is from Luminary 131, and then verified to assemble to the proper binary
# values. This area is blank on the Luminary 099 print-out, as if the
# printer ribbon had run out.
      INDEX  QRCNTR      # AOS SCALED AT PI/2
      CA      AOSQ
      EXTEND
      MP      BIT2      # RESCALE AOS TO PI/4
      EXTEND
      BZF     GTSQAXIS -3  # USE FULL SCALE FOR LARGER AOS ESTIMATES.

      INDEX  A
      CS     LIMITS      # LIMITS +1 CONTAINS NEGMAX.
      XCH    L           # LIMITS -1 CONTAINS POSMAX.

      CCS    QRCNTR      # PICK UP RATE FOR THIS AXIS.  RATE CELLS
      INDEX  A           # USE ADJACENT, NOT SEPARATED.  AT PI/4
      CA     EDOTQ
      GTSQAXIS DXCH    WCENTRAL

      INDEX  QRCNTR      # COLLECT K FOR THIS AXIS
      CA     KQ
      TS     KCENTRAL

      EXTEND
      BZF    POSDRIVE +1  # CONTROL AUTHORITY ZERO.  AVOID DRIVING
                        # ENGINE BELL TO THE STOPS.

      INDEX  QRCNTR      # QDIFF, RDIFF ARE STORED IN D.P.
      CA     QDIFF

      ALGORITHM EXTEND    # Q(R)DIFF IS THETA (ERROR) SCALED AT PI.
      MP      KCENTRAL    # FORM K*ERROR AT PI(2)/2(8), IN D.P.
      LXCH    K2THETA
      EXTEND
      MP      BIT5        # RESCALE TO 4*PI(2)
      DXCH    K2THETA
      EXTEND
      MP      BIT5        # FIRST TERM OF NEGUSUM IN K2THETA.
      ADS     K2THETA +1  # NO CARRY NEEDED          D.P. AT 4*PI(2)

      CS      ACENTRAL    # FORM ALPHA(2)/(2*K) AT 16*PI, IN D.P.,
      EXTEND    # LIMITING QUOTIENT TO AVOID OVERFLOW.
      MP      BIT14      # -ALPHA/2 IN A, SCALED AT PI/4
      EXTEND

```

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```
MP      ACENTRAL      # -ALPHA(2)/2 IN A,L, SCALED AT PI(2)/16)
AD      KCENTRAL
EXTEND
BZMF    HUGEQUOT      # K-ALPHA(2)/2 SHOULD BE PNZ FO DIVISION

EXTEND
DCS     A              # ALPHA(2)/2 - K
AD      KCENTRAL
```

```
# RSB 2009 -----
EXTEND
DV      KCENTRAL      # HIGH ORDER OF QUOTIENT.
XCH     A2CENTRAL
CA      L              # SHIFT UP THE REMAINDER.
LXCH    7              # ZERO LOW-ORDER DIVIDEND.
EXTEND
```

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```
651  (Page LM1474 651)≡ (647b 774)
      DV      KCENTRAL
      XCH      A2CNTRAL +1      # QUOTIENT STORED AT 16*PI, D.P.
      TCF      HAVEQUOT

      HUGEQUOT      CA      POSMAX
      TS      L
      DXCH      A2CNTRAL      # LIMITED QUOTIENT STORED AT 16*PI, D.P.

      HAVEQUOT      CA      WCENTRAL
      EXTEND
      MP      BIT9      # RESCALE OMEGA AT 16*PI IN D.P.
      DXCH      K2CNTRAL      # LOWER WORD OVERLAYS OMEGA IN WCENTRAL

      EXTEND
      DCA      K2CNTRAL
      DXCH      FUNCTION

      CA      ACENTRAL      # GET ALPHA*ABS(ALPHA)/(2*K)
      EXTEND
      BZMF      +4

      EXTEND
      DCA      A2CNTRAL
      TCF      +3

      EXTEND
      DCS      A2CNTRAL

      DAS      FUNCTION      # OMEGA + ALPHA*ABS(ALPHA)/(2*K) AT 16*PI

      CCS      FUNCTION      # DEL = +1 FOR FUNCT1 GREATER THAN ZERO.
      TCF      POSFNCT1      # OTHERWISE DEL = -1
      TCF      +2
      TCF      NEGFNCT1

      CCS      FUNCTION +1      # USE LOW ORDER WORD SINCE HIGH IS ZERO
      POSFNCT1      CAF      BIT1
      TCF      +2
      NEGFNCT1      CS      BIT1
      TS      DEL

      CCS      DEL      # REPLACE OMEGA BY DEL*OMEGA
      TCF      FUNCT2      # POSITIVE DEL VALUE. PROCEED.
      TCF      DEFUNCT
      TCF      NEGFNCT2
```

DEFUNCT	TS	K2CNTRAL
	TS	K2CNTRAL +1
	TCF	FUNCT2

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653  $\langle \text{Page LM1475 653} \rangle \equiv$  (647b 774)

NEG1/3 DEC -.33333

NEGFUNCT2  
EXTEND  
DCS K2CNTRAL  
DXCH K2CNTRAL

FUNCT2  
EXTEND  
DCA A2CNTRAL  
DAS K2CNTRAL # DEL\*OMEGA + ALPHA(2)/(2\*K) AT 16\*PI, D.P.

FUNCT3  
CA A2CNTRAL  
EXTEND  
MP NEG1/3  
DXCH A2CNTRAL  
CA L  
EXTEND  
MP NEG1/3  
ADS A2CNTRAL +1  
TS L  
TCF +2 # A2CNTRAL NOW CONTAINS -ALPHA(2)/(6\*K),  
ADS A2CNTRAL # SCALED AT 16\*PI, IN D.P.

EXTEND  
DCA K2CNTRAL # DEL\*OMEGA + ALPHA(2)/(3\*K) IN A2CNTRAL,  
DAS A2CNTRAL # SCALED AT 16\*PI, D.P.

CA A2CNTRAL  
EXTEND  
MP ACENTRAL  
DAS K2THETA  
CA A2CNTRAL +1  
EXTEND  
MP ACENTRAL # ACENTRAL MAY NOW BE OVERLAID.  
ADS K2THETA +1  
TS L  
TCF +2 # TWO TERMS OF NEGUSUM ACCUMULATED, SO FAR  
ADS K2THETA # SCALED AT 4\*PI(2), IN D.P.

GETROOT  
CA K2CNTRAL # K\*(DEL\*OMEGA + ALPHA(2)/(2\*K)) IS THE  
EXTEND # TERM FOR WHICH A SQUARE ROOT IS NEEDED.  
MP KCENTRAL # K AT PI/2(8)  
DXCH FUNCTION  
CA K2CNTRAL +1  
EXTEND  
MP KCENTRAL

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ADS FUNCTION +1

TS L

TCF +2

ADS FUNCTION # DESIRED TERM IN FUNCTION, AT  $\text{PI}(2)/16$

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655 (Page LM1476 655)≡

(647b 774)

CCS DEL  
TCF RSTOFGTS  
TCF NEGUSUM  
TCF NEGATE  
TCF NEGUSUM

NEGATE EXTEND  
DCS K2CNTRAL  
DXCH K2CNTRAL  
TCF RSTOFGTS

BANK 16  
EBANK= NEGUQ  
SETLOC DAPS1  
BANK

# THE WRCHN12 SUBROUTINE SETS BITS 9,10,11,12 OF CHANNEL 12 ON THE BASIS OF THE CONTENTS OF NEG  
# THE NEGATIVES OF THE DESIRED ACCELERATION CHANGES. ACDT+C12 SETS Q(R)ACCDOT TO REFLECT THE N  
#  
# WARNING: ACDT+C12 AND WRCHN12 MUST BE CALLED WITH INTERRUPT INHIBITED.

BGIM OCTAL 07400  
CHNL12 EQUALS ITEMP6  
ACDT+C12 CS NEGUQ

EXTEND # GIMBAL DRIVE REQUESTS.

MP ACCDOTQ  
LXCH QACCDOT  
CS NEGUR  
EXTEND  
MP ACCDOTR  
LXCH RACCDOT

CCS NEGUQ  
CAF BIT10  
TCF +2  
CAF BIT9  
TS CHNL12

CCS NEGUR  
CAF BIT12  
TCF +2  
CAF BIT11  
ADS CHNL12

# (STORED RESULT NOT USED AT PRESENT)

CS BGIM

```
EXTEND
RAND   CHAN12
AD     CHNL12
EXTEND
WRITE  CHAN12
```

656  $\langle$ Page LM1477 656 $\rangle \equiv$ 

(647b 774)

```
CS      CALLGMBL
MASK    RCSFLAGS
TS      RCSFLAGS
```

# TURN OFF REQUEST FOR ACDT+C12 EXECUTION.

```
TC      Q
```

# RETURN TO CALLER.

```
BANK    21
EBANK=  QDIFF
SETLOC  DAPS4
BANK
```



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(Page LM1478 657)≡

(647b 774)

```

# SUBROUTINE TIMEGMBL:  MOD 0,  OCTOBER 1967,  CRAIG WORK
#
# TIMEGMBL COMPUTES THE DRIVE TIME NEEDED FOR THE TRIM GIMBAL TO POSITION THE DESCENT ENGINE NO
# THE OFFSET ANGULAR ACCELERATION ABOUT THE Q (OR R) AXIS.  INSTEAD OF USING AOSQ(R), TIMEGMBL
# SCALED AT PI/8.  FOR EACH AXIS, THE DRIVE TIME IS COMPUTED AS ABS(ALPHA/ACCDOT).  A ZERO
# ALPHA OR ACCDOT OR A ZERO QUOTIENT TURNS OFF THE GIMBAL DRIVE IMMEDIATELY.  OTHERWISE, THE G
# DRIVING IN THE CORRECT DIRECTION.  THE Q(R)GIMTIMR IS SET TO TERMINATE THE DRIVE AND Q(R)ACCD
# IS STORED TO REFLECT THE NEW ACCELERATION DERIVATIVE.  NEGUQ(R) WILL CONTAIN +1,+0,-1 FOR A Q
# WHICH IS NEGATIVE, ZERO, OR POSITIVE.
#
# INPUTS:          AOSQ,AOSR, SCALED AT P1/2, AND ACCDOTQ, ACCDOTR AT PI/2(7).      PI/2(7).
#
# OUTPUTS:         NEW GIMBAL DRIVE BITS IN CHANNEL 12, NEGUQ, NEGUR, QACCDOT, AND RACCDOT, THE LA
#                  Q(R)GIMTIMR WILL BE SET TO TIME AND TERMINATE GIMBAL DRIVE(S).
#
# DEBRIS:          A, L, Q, ITEMPS 2, 3, 6, AND RUPTREG2 AND ACDT+C12 DEBRIS.
#
# EXITS:           VIA TC Q.
#
# ALARMS, ABORTS:  NONE.
#
# SUBROUTINES:     ACDT+C12, IBNKCALL
#
# WARNING:         THIS SUBROUTINE WRITES INTO CHANNEL 12 AND USES THE ITEMPS.  THEREFORE IT MAY C
#                  INTERRUPT INHIBITED.
#
# ERASABLE STORAGE CONFIGURATION (NEEDED BY THE INDEXING METHODS):
#   NEGUQ           ERASE   +2           # NEGATIVE OF Q-AXIS GIMBAL DRIVE
#   (SPWORD)        EQUALS  NEGUQ +1      # ANY S.P. ERASABLE NUMBER, NOW THRSTCMD
#   NEGUR           EQUALS  NEGUQ +2      # NEGATIVE OF R-AXIS GIMBAL DRIVE
#   ACCDOTQ         ERASE   +2           # Q-JERK TERM SCALED AT PI/2(7) RAD/SEC(3)
#   (SPWORD)        EQUALS  ACCDOTQ +1    # ANY S.P. ERASABLE NUMBER NOW QACCDOT
#   ACCDOTR         EQUALS  ACCDOTQ +2    # R-JERK TERM SCALED AT PI/2(7) RAD/SEC(3)
#                                     # ACCDOTQ, ACCDOTR ARE MAGNITUDES.
#   AOSQ            ERASE   +4           # Q-AXIS ACC., D.P. AT PI/2 R/SEC(2)
#   AOSR            EQUALS  AOSQ +2       # R-AXIS ACCELERATION SCALED AT PI/2 R/S2
#
QRNDXER           EQUALS  ITEMP6
OCT23146          OCTAL   23146          # DECIMAL .6
NZACCDOT          EQUALS  ITEMP3
#
TIMEGMBL          CAF     ONE            # INITIALZE ALLOWGTS.
                  TS      ALLOWGTS
#
                  CAF     TWO            # SET UP LOOP FOR R AXIS.

```

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LXCH Q  
LXCH RUPTREG2

# SAVE RETURN ADDRESS.

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```

659      (Page LM1479 659)≡
                                (647b 774)
TIMQGMBL      TCF      +2
               CAF      ZERO      # NOW DO THE Q-AXIS
               TS      QRNDXER
               INDEX    QRNDXER
               CA      ACCDOTQ      # ACCDOT IS PRESUMED TO BE AT PI/2(7).
               EXTEND
               BZMF      TGOFFNOW      # IS ACCDOT LESS THAN OR EQUAL TO 0?
               TS      NZACCDOT      # NO.  STORE NON-ZERO, POSITIVE ACCDOT.

ALPHATRY      INDEX    QRNDXER
               CS      AOSQ
               EXTEND
               BZF      TGOFFNOW      # IS ALPHA ZERO?

               TS      Q      # SAVE A COPY OF -AOS.
               EXTEND      # NO.  RESCALE FOR TIMEGMBL USE.
               MP      OCT23146      # OCTAL 23146 IS DECIMAL .6
               AD      Q      # -1.6*AOS AT PI/2 = -.4*AOS AT PI/8
               TS      L      # WAS THERE OVERFLOW?
               TCF      SETNEGU      # NO.  COMPUTE DRIVE TIME.

               CS      A      # RECOVER -SGN(AOS) IN THE A REGISTER.
               INDEX    QRNDXER      # YES.  START DRIVE WITHOUT WAITLIST.
               XCH      NEGUQ
               TCF      NOTALLOW      # KNOCK DOWN THE ALLOWGTS FLAG.

SETNEGU      EXTEND
               BZMF      POSALPH

               COM
               TS      ITEMP2      # STORE -ABS(.4*AOS) SCALED AT PI/8.
               CS      BIT1
               TCF      POSALPH +2
POSALPH      TS      ITEMP2      # STORE -ABS(.4*AOS) SCALED AT PI/8.
               CA      BIT1
               +2      INDEX    QRNDXER      # SGN(AOS) INTO NEGU
               TS      NEGUQ      # STORE SGN(ALPHA) AS NEGU

               CA      NZACCDOT
               EXTEND
               MP      BIT12      # 2*ACCDOT, SCALED AT PI/8.
               AD      ITEMP2      # -ABS(ALPHS) + 2*ACCDOT, AT PI/8.
               EXTEND
               BZMF      NOTALLOW      # IS DRIVE TIME MORE THAN TWO SECONDS?
               CS      ITEMP2      # NO.  COMPUTE DRIVE TIME.

```

	EXTEND		# ABS(ALPHA) AT PI/8.
	MP	OCT00240	# DECIMAL 10/1024
	EXTEND		# QUOTIENT IS DRIVE TIME AT WAITLIST
	DV	NZACCDOT	# ABS(ALPHA)/ACCDOT AT 2(14)/100
660	$\langle \text{Page LM1480 660} \rangle \equiv$		(647b 774)
	EXTEND		
	BZF	TGOFFNOW	# DRIVE TIME MUST BE GREATER THAN ZER
	TCF	DRIVEON	
TGOFFNOW	CAF	ZERO	# TURN OFF GIMBAL NOW.
	INDEX	QRNDXER	
	TS	NEGUQ	
	TCF	DONEYET	
NOTALLOW	CAF	OCT31	
	INDEX	QRNDXER	
	TS	QGIMTIMR	
	CAF	ZERO	# DRIVE TIME IS MORE THAN 2 SECONDS,
	TS	ALLOWGTS	# DO NOT PERMIT FURTHER GTS ATTITUDE.
	TCF	DONEYET	# CONTROL UNTIL AOSTASK APPROVES.
			# NO WAITLIST CALL IS MADE.
DRIVEON	INDEX	QRNDXER	
	TS	QGIMTIMR	# CHOOSE Q OR R AXIS.
DONEYET	CCS	QRNDXER	
	TCF	TIMQGMBL	
	DXCH	RUPTREG3	# PROTECT IBNKCALL ERASABLES. ACDT+
	DXCH	ITEMP2	# LEAVES ITEMPS2,3 ALONE.
	TC	IBNKCALL	# TURN OFF CHANNEL BITS, SET Q(R)ACCD
	CADR	ACDT+C12	
	DXCH	ITEMP2	# RESTORE ERASABLES FOR IBNKCALL.
	DXCH	RUPTREG3	
	TC	RUPTREG2	# RETURN TO CALLER.
OCT00240	OCTAL	00240	# DECIMAL 10/1024

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(647b 774)

# THE FOLLOWING SECTION IS A CONTINUATION OF THE TRIM GIMBAL CONTROL FROM THE LAST GTS ENTRY.  
 # IS COMPUTED FOR EACH AXIS (Q,R), .707\*DEL\*FUNCTION(3/2) + K2THETA = NEGUSUM. NEW DRIVES ARE  
 #  
 # THE SUBROUTINE GTSQRT ACCEPTS A DOUBLE PRECISION VALUE IN FUNCTION, FUNCTION +1 AND RETURNS A  
 # SQUARE ROOT OF THE FOURTEEN MOST SIGNIFICANT BITS OF THE ARGUMENT. ALSO, THE CELL SHFTFLAG C  
 # EXPONENT S, SUCH THAT THE SQUARE ROOT (RETURNED IN THE A REGISTER) MUST BE SHIFTED RIGHT (MUL  
 # POWER (-S)) IN ORDER TO BE THE TRUE SQUARE ROOT OF THE FOURTEEN MOST SIGNIFICANT BITS OF FUNC  
 # SQUARE ROOT ERROR IS NOT MORE THAN 2 IN THE 14TH SIGNIFICANT BIT. CELLS CLOBBED ARE A, L,  
 # HALFARG, SCRATCH, SR, FUNCTION, FUNCTION +1. GTSQRT IS CALLED BY TC GTSQRT AND RETURNS VIA T  
 # ZERO OR NEGATIVE ARGUMENTS YIELD ZERO FOR SQUARE ROOTS.

GTSQRT	CCS	FUNCTION	
	TCF	GOODARG	# FUNCTION IS POSITIVE. TAKE SQUARE ROOT.
	TCF	+2	# HIGH ORDER WORD IS ZERO. TRY THE LOWER.
	TCF	ZEROOT	# NEGATIVE. USE ZERO FOR 1/2 POWER.
	CA	FUNCTION +1	
	EXTEND		
	BZMF	ZEROOT	
	TCF	ZEROHIGH	# PROCEED.
ZEROOT	CA	ZERO	
	TS	SHFTFLAG	
	TC	Q	
ZEROHIGH	XCH	FUNCTION	# 14 MOST SIGNIFICANT BITS ARE IN THE
	XCH	FUNCTION +1	# LOWER WORD. EXCHANGE THEM.
	CA	SEVEN	
	TCF	GOODARG +1	
GOODARG	CA	ZERO	
	TS	SHFTFLAG	
	CA	TWELVE	# INITIALIZE THE SCALING LOOP.
	TS	ININDEX	
	TCF	SCALLOOP	
SCALSTRT	CA	FUNCTION	
	TCF	SCALDONE	
MULBUSH	CA	NEG2	# IF ARG IS NOT LESS THAN 1/4, INDEX IS
	ADS	ININDEX	# ZERO, INDICATING NO SHIFT NEEDED.
	EXTEND		# BRANCH IF ARG IS NOT LESS THAN 1/4.
	BZMF	SCALSTRT	# OTHERWISE COMPARE ARG WITH A REFERENCE
			# WHICH IS 4 TIMES LARGER THAN THE LAST.
SCALLOOP	CS	FUNCTION	

INDEX ININDEX  
AD BIT15  
EXTEND  
BZMF MULBUSH

# REFERENCE MAGNITUDE LESS OR EQUAL TO 1/4

# IF ARG IS NOT LESS THAN REFERENCE, GO  
# AROUND THE MULBERRY BUSH ONCE MORE.

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```

      INDEX  ININDEX
      CA     BIT15      # THIS IS THE SCALE MAGNITUDE
      XCH    HALFARG    # 2**(-ININDEX) IS THE SHIFT DIVISOR.
      EXTEND      # RESCALE ARGUMENT.
      DCA     FUNCTION
      EXTEND
      DV      HALFARG

      # ININDEX AND SHFTFLAG PRESERVE INFO FOR
      # RESCALING AFTER ROOT PROCESS.

SCALDONE      EXTEND
      QXCH    FUNCTION +1  # SAVE Q FOR RETURN
      EXTEND
      MP      BIT14
      TS      HALFARG
      MASK    BIT13
      CCS     A
      CA      OCT11276
      AD      Roothalf    # INITIAL GUESS IS ROOT 1/2 OR POSMAX
      TC      ROOTCYCL
      TC      ROOTCYCL
      TC      ROOTCYCL
      TC      FUNCTION +1

# *****

RSTOFGTS      TC      GTSQRT
PRODUCT       XCH     K2CNTRAL
      EXTEND
      MP      K2CNTRAL
      DXCH    K2CNTRAL
      EXTEND      # THE PRODUCT OF
      MP      L        # 1/2      2      1/2
      ADS     K2CNTRAL +1 # K      *(DEL*OMEGA + ALPHA /(2*K))
      TS      L        # AND
      TCF     +2        # 2
      ADS     K2CNTRAL  # DEL*(DEL*OMEGA + ALPHA /(2*K)) NOW IN
      # K2CNTRAL

DOSHIPT       CA      ININDEX
      EXTEND      # MULTIPLY IN THE FACTOR 2(-S), RETURNED
      MP      BIT14  # BY THE GTSQRT SUBROUTINE
      ADS     SHFTFLAG
      EXTEND
      BZF     ADDITIN
      INDEX   SHFTFLAG
```

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CA BIT15



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(647b 774)

	XCH	K2CNTRAL	
	EXTEND		
	MP	K2CNTRAL	
	DAS	K2THETA	
	XCH	K2CNTRAL	
	EXTEND		
	MP	K2CNTRAL +1	
	ADS	K2THETA +1	
	TS	L	
	TCF	+2	
	ADS	K2THETA	
	TCF	NEGUSUM	
ADDITIN	EXTEND		
	DCA	K2CNTRAL	
	DAS	K2THETA	# NO ADD IN THE K2THETA TERM.
NEGUSUM	CCS	K2THETA	# TEST SIGN OF HIGH ORDER PART.
	TCF	NEGDRIVE	
	TCF	+2	
	TCF	POSDRIVE	
	CCS	K2THETA +1	# SIGN TEST FOR LOW ORDER PART.
NEGDRIVE	CA	BIT1	
	TCF	+2	# STOP GIMBAL DRIVE FOR A ZERO NEGUSUM.
POSDRIVE	CS	BIT1	
	TS	L	# SAVE FOR DRIVE REVERSAL TEST.
	INDEX	QRCNTR	
	XCH	NEGUQ	
	EXTEND		
	MP	L	# MULTIPLY OLD NEGU AND NEW NEGU.
	CCS	L	
	TCF	LOUPE	# NON-ZERO GIMBAL DRIVE BEING CONTINUED.
	TCF	ZEROLOUP	# NO REVERSAL PROBLEM HERE.
	TCF	REVERSAL	# NON-ZERO GIMBAL DRIVE BEING REVERSED.
	TCF	ZEROLOUP	# NO REVERSAL PROBLEM HERE.
REVERSAL	INDEX	QRCNTR	# A ZERO-DRIVE PAUSE IS NEEDED HERE. ZERO
	TS	QACCDOT	# IS IN A REGISTER FROM CCS ON (-1).
	INDEX	QRCNTR	
	CS	GMBLBITA	
	EXTEND		

```

                                WAND    CHAN12

ZEROLOUP    CS    RCSFLAGS    # SET UP REQUEST FOR ACDT+C12 CALL.
            MASK    CALLGMBL
            ADS    RCSFLAGS

666  <Page LM1484 666>≡                                (647b 774)
      LOUPE    CCS    QRCNTR    # HAVE BOTH AXES BEEN PROCESSED?
            TCF    GOQTRIMG    # NO. DO Q AXIS NEXT.

            CA    SAVESR    # RESTORE THE SR
            TS    SR

GOCLOSE    EXTEND    # TERMINATE THE JASK.
            DCA    CLOSEADR
            DTCB

CLOSEADR    EBANK=  AOSQ
            2CADR    CLOSEOUT    # TERMINATE THE JASK.

TWELVE    EQUALS    OCT14
ROOTHALF    OCTAL    26501    # SQUARE ROOT OF 1/2
GMBLBITA    OCTAL    01400    # INDEXED WRT GMBLBITB DO NOT MOVE *****
OCT11276    OCTAL    11276    # POSMAX -- ROOTHALF
GMBLBITB    OCTAL    06000    # INDEXED WRT GMBLBITA DO NOT MOVE *****

# SUBROUTINE ROOTCYCL: BY CRAIG WORK, 3 APRIL 68
#
# ROOTCYCL IS A SUBROUTINE WHICH EXECUTES ONE NEWTON SQUARE ALGORITHM ITERATION. THE
# SQUARE ROOT IS PRESUMED TO BE IN THE A REGISTER AND ONE-HALF THE SQUARE IS TAKEN FR
# TO THE SQUARE ROOT IS RETURNED IN THE A REGISTER. DEBRIS: A, L, SR, SCRATCH. RO
# LOCATION (LOC) BY A TC ROOTCYCL, AND RETURNS (TC Q) TO LOC +1.
#
# WARNING: IF THE INITIAL GUESS IS NOT GREATER THAN THE SQUARE, DIVIDE OR ADD OVERF

ROOTCYCL    TS    SCRATCH    # STORE X
            TS    SR    # X/2 NOW IN SR
            CA    HALFARG    # ARG/2 IN THE A REG
            ZL    # PREPARE FOR DIVISION
            EXTEND
            DV    SCRATCH    # (ARG/X)/2
            AD    SR    # (X + ARG/X)/2 IN THE A REG
            TC    Q

```

## 1.39 aostask and aosjob

667  $\langle aostask \text{ and } aosjob \text{ 667} \rangle \equiv$  (7)

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 $\langle Page \text{ LM1506 } 704 \rangle$

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(667 714)

```

# PROGRAM NAME:          1/ACCS
# PROGRAM WRITTEN BY:    BOB COVELLI AND MIKE HOUSTON
# LAST MODIFICATION:     FEB. 14, 1969 BY G. KALAN
#
# PROGRAM DESCRIPTION:
#     1/ACCS PROVIDES THE INTERFACE BETWEEN THE GUIDANCE PROGRAMS AND THE DIGITAL A
#     CHANGE IN THE MASS OF THE VEHICLE, IN THE DEADBAND SELECTED, IN THE VEHICLE C
#     DOCKED), AND DURING A FRESH START OR A RESTART, 1/ACCS IS CALLED TO COMMUNICA
#
#     THE INPUTS TO 1/ACCS ARE MASS, ACCELERATION (ABDELV), DEADBAND (DB), OFFSET A
#     STAGE VERIFY BIT (CHAN30, BIT2), DOCKED BIT (DAPBOOLS, BIT13), DRIFT BIT (DA
#     BIT14), AND SURFACE FLAG (FLAGWRDB, BIT8), AND CH5MASK.
#
#     1/ACCS COMPUTES THE JET ACCELERATIONS (1JACC, 1JACCQ, 1JACCR) AS FUNCTIONS OF
#     FORMED BY RESOLVING 1JACCQ AND 1JACCR.  IN THE DESCENT CASE, THE DESCENT ENGI
#     COMPUTED AS A FUNCTION OF MASS.  THE RATE OF CHANGE OF ACCELERATION DUE TO RO
#     ACCDOTR) IS ALSO COMPUTED IN THE DESCENT CASE.
#
#     AFTER THE ABOVE COMPUTATIONS, THE PROGRAM 1/ACCONT COMPUTES THE RECIPROCAL N
#     AND V AXES (2 JETS FOR P-AXIS, BOTH 1 AND 2 JETS FOR U AND V AXES), AND THE P
#     THE P, U, AND V AXES.  THE ACCELERATION FUNCTIONS (ACCFCTZ1 AND ACCFCTZ5) ARE
#     FIRE AND COAST DEADBANDS AND AXISDIST ARE COMPUTED FOR EACH AXIS.  FLAT AND 2
#     MINIMUM IMPULSE ZONE, ARE COMPUTED.  1/ACCONT ALSO SETS ACCSWU AND ACCSWV, W
#     IS NOT SUFFICIENT TO PRODUCE MINIMUM ACCELERATION.  AT THE COMPLETION OF 1/A
#
# SUBROUTINES CALLED:
#     TIMEGMBL
#     MAKECADR
#     ROT45DEG
#
# CALLING SEQUENCE:
#         TC          BANKCALL          # (1/ACCS MUST BE CALLED BY BANKCALL)
#         CADR        1/ACCS
#
# NORMAL EXIT:  VIA BANKJUMP
#
# ALARM AND EXIT MODES:  NONE
#
# INPUT/OUTPUT:  SEE PROGRAM DESCRIPTION.
#
# DEBRIS:
#     ALL OF THE EXECUTIVE TEMPORARY REGISTERS, EXCEPT FIXLOC AND OVFINDD, AND THE C
#
# RESTRICTIONS:
#     1/ACCS MUST BE CALLED BY BANKCALL

```

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# EBANK IS SET TO 6, BUT NOT RESTORED.

BANK 20  
 SETLOC DAPS3  
 BANK

COUNT\* \$\$/DAPAO

EBANK= AOSQ

# ENTRY IS THROUGH 1/ACCJOB OR 1/ACCSIT WHEN 1/ACCS IS TO BE DONE AS A SEPARATE NOVAC  
 # IT IS POSSIBLE FOR MORE THAN ONE OF THESE JOBS TO BE SET UP CONCURRENTLY. HOWEVER,  
 # NEWJOB, A SECOND MANIFESTATION CANNOT BE STARTED UNTIL THE FIRST IS COMPLETED.

1/ACCSET CAF ZERO # ENTRY FROM FRESH START/RESTART CODING.  
 TS AOSQ # NULL THE OFFSET ESTIMATES FOR 1/ACCS  
 TS AOSR  
 TS ALPHAQ # NULL THE OFFSET ESTIMATES FOR DOWNLIS  
 TS ALPHAR

1/ACCJOB TC BANKCALL # 1/ACCS ASSUMES ENTRY VIA BANKCALL.  
 CADR 1/ACCS +2 # SKIP EBANK SETTING.

TC ENDOFJOB

1/ACCS CA EBANK6 # \*\*\*\*\* EBANK SET BUT NOT RESTORED \*\*\*\*\*  
 TS EBANK

TC MAKECADR # SAVE RETURN SO THAT BUF2 MAY BE USED  
 TS ACCRETRN

# DETERMINE MASS OF THE LEM.

CA DAPBOOLS # IS THE CSM DOCKED  
 MASK CSMDOCKD  
 TS DOCKTEMP # STORE RECORD OF STATE IN TEMP (MPAC +3).  
 CCS A  
 CS CSMMASS # DOCKED: LEMMAS = MASS - CSMMASS  
 AD MASS # LEM ALONE: LEMMASS = MASS  
 TS LEMMASS

# ON THE BASIS OF APSFLAG:

# SET THE P-AXIS RATE COMMAND LIMIT FOR 2-JET/2-JET CONTROL

# SET MPAC, WHICH INDICATES THE PROPER SET OF COEFFICIENTS FOR THE LEM-ALONE F

# ENSURE THAT THE LEM MASS VALUE IS WITHIN THE ACCEPTABLE RANGE

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INHINT

CAE FLGWRD10

MASK APSFLBIT

EXTEND

BZF DPSFLITE

# DETERMINE WHETHER STAGED.

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```

        CS      POSMAX      # ASCENT (OR ON LUNAR SURFACE)
        TS      -2JETLIM    # ALWAYS 2 JETS FOR P-AXIS RATE COMMAND
        CAF      OCT14      # INITIALIZE INDEX AT 12.
        TS      MPAC
        CS      LEMMASS     # CHECK IF MASS TOO HIGH.  CATCH STAGING.
        AD      HIASCENT
        EXTEND
        BZMF     MASSFIX
        CS      LEMMASS     # CHECK IF MASS TOO LOW.  THIS LIMITS THE
        AD      LOASCENT   #      DECREMENTING BY MASSMON.
        EXTEND
        BZMF     F(MASS)

MASSFIX      ADS      LEMMASS      # STORE THE VIOLATED LIMIT AS LEMMASS.
              ZL              #      ALSO CORRECT TOTAL MASS, ZEROING THE
              CCS      DOCKTEMP    #      LOW-ORDER WORD.
              CAE      CSMASS      #      DOCKED:  MASS = LEMMASS + CS
              AD      LEMMASS      #      LEM ALONE:  MASS = LEMMASS
              DXCH     MASS
              TCF      F(MASS)

DPSFLITE     CS      BIT10      # FOUR JETS FOR P-AXIS RATE COMMAND ERRORS
              TS      -2JETLIM   #      EXCEEDING 1.4 DEG/SEC (SCALED AT 45)
              CAF      SIX       # INITIALIZE INDEX AT 6.
              TS      MPAC
              CS      LEMMASS     # CHECK IF MASS TOO HIGH.  SHOULD NEVER
              AD      HIDESCNT    #      OCCUR EXCEPT PERHAPS BEFORE THE PAD
              EXTEND            #      LOAD IS DONE.
              BZMF     MASSFIX
              CS      LEMMASS     # CHECK IF MASS TOO LOW.  THIS LIMITS THE
              AD      LODESCNT   #      DECREMENTING BY MASSMON.
              AD      HIASCENT
              EXTEND
              BZMF     F(MASS)
              TCF      MASSFIX

# COMPUTATION OF FUNCTIONS OF MASS

F(MASS)      RELINT
              CCS      DOCKTEMP
              TCF      DOCKED    # DOCKED:  USE SEPARATE COMPUTATION.
              CA      TWO
STCTR        TS      MPAC      +1 # J=2,1,0 FOR 1JACCR,1JACCQ,1JACC
              CS      TWO

```



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	ADS	MPAC		# JX=10,8,6 OR 4,2,0 TO INDEX COEFS.
STCTR1	CAE	LEMMASS		
	INDEX	MPAC		
	AD	INERCONC		
	TS	MPAC	+2	# MASS + C

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```

EXTEND
INDEX    MPAC
DCA      INERCONA
EXTEND
DV       MPAC      +2
INDEX    MPAC
AD       INERCONB
INDEX    MPAC      +1      # 1JACC(J)=A(JX)/(MASS+C(JX) + B(JX)
TS       1JACC            # 1JACC(-1)=L,PVT-CG   SCALED AT 8 FEET

CCS      MPAC      +1
TCF      STCTR
TCF      COMMEQS
TCF      LRESC

# COEFFQ AND COEFFR ARE COMPUTED IN THIS SECTION.  THEY ARE USED TO RESOLVE Q-R COMPO
# U AND V COMPONENTS (SEE ROT-TOUV SECTION).

COMMEQS   CS      1JACCR
          AD      1JACCQ
          EXTEND
          BZMF    BIGIQ
          EXTEND
          DV      1JACCQ      # EPSILON IS A MEASURE OF COUPLING AND IS
          TS      EPSILON     # DEFINED=1-IQ/IR FOR IR GREATER THAN IQ.
          AD      -EPSMAX     # THE COMPUTED EXPRESSION IS EQUIVALENT
          EXTEND
          BZMF    GOODEPS1
          CS      -EPSMAX
          TS      EPSILON     # EPSILON IS LIMITED TO A MAX. OF .42265
GOODEPS1  CA      EPSILON
          EXTEND
          MP      0.35356
          AD      .7071
          TS      COEFFR      # IN THIS CASE WHERE IR IS GREATER THAN
          CS      POSMAX     # IQ, COEFFQ=-.707(1+.5EPSILON)(1-EPSILON)
          AD      EPSILON     # AND COEFFR=.707(1+.5EPSILON)
          EXTEND
          MP      COEFFR
          TS      COEFFQ
          TCF     JACCUV
BIGIQ     EXTEND
          DV      1JACCR      # EPSILON IS DEFINED AS 1-IR/IQ FOR IQ
          TS      -EPSILON    # GREATER THAN IR.  -EPSILON IS COMPUTED
          CS      -EPSILON    # RATHER THAN EPSILON FOR CONVENIENCE

```

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AD -EPSMAX  
EXTEND  
BZMF GOODEPS2  
CA -EPSMAX  
TS -EPSILON

# EPSILON IS LIMITED TO A MAX. OF .42265

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(667 714)

```

GOODEPS2      CA      -EPSILON
EXTEND
MP            0.35356
AD            -.7071
TS            COEFFQ      # IN THIS CASE WHERE IQ IS GREATER THAN
CS            -EPSILON    # IR, COEFFQ=-.707(1+.5EPSILON) AND
AD            NEGMX       # COEFFR=.707(1+.5EPSILON)(1-EPSILON)
EXTEND
MP            COEFFQ
TS            COEFFR
JACCUV        CS            COEFFQ
EXTEND
MP            1JACCQ      # 1JACCQ IS SCALED AT PI/4
TS            1JACCU      # 1JACCU USED AS TEMPORARY STORAGE
CA            COEFFR
EXTEND
MP            1JACCR
AD            1JACCU
EXTEND
MP            BIT14       # SCALING CHANGED FROM PI/4 TO PI/2
TS            1JACCU
TS            1JACCV      # SCALED AT PI/2 RADIANS/SEC(2)
CCS           MPAC        # COMPUTE L,PVT-CG IF IN DESCENT
CAF           ZERO        # ZERO SWITCHES AND GO TO 1/ACCONT IN
TS            ALLOWGTS     #      ASCENT
TCF           1/ACCONT -1

CS            TWO
TS            MPAC
CS            ONE
TS            MPAC      +1
TCF           STCTR1

```

```

# THIS SECTION COMPUTES THE RATE OF CHANGE OF ACCELERATION DUE TO THE ROTATION OF THE
# IMPLEMENTED IN BOTH THE Y-X PLANE AND THE Z-X PLANE IS -- D(ALPHA)/DT = TL/I*D(DELTA)
#      T = ENGINE THRUST FORCE
#      L = PIVOT TO CG DISTANCE OF ENGINE
#      I = MOMENT OF INERTIA

```

```

LRESC          CAE      ABDELV      # SCALED AT 2(13) CM/SEC(2)
EXTEND
MP            MASS      # SCALED AT B+16 KGS
TC            DVOVSUB   # GET QUOTIENT WITH OVERFLOW PROTECTION
ADRES         GFACTM

```

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# MASS IS DIVIDED BY ACCELERATION OF GRAVITY IN ORDER TO MATCH THE UNITS OF IXX,IYY,IZZ, WHICH  
# THE RATIO OF ACCELERATION FROM PIPAS TO ACCELERATION OF GRAVITY IS THE SAME IN METRIC OR ENGI  
# THAT IS UNCONVERTED. 2.20462 CONVERTS KG. TO LB. NOW T IN IN A SCALED AT 2(14).

EXTEND

MP L,PVT-CG # SCALED AT 8 FEET.

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```

                                INHINT
                                TS      MPAC
                                EXTEND
                                MP      1JACCR
                                TC      DVOVSUB      # GET QUOTIENT WITH OVERFLOW PROTECTION
                                ADRES   TORKJET1

                                TS      ACCDOTR      # SCALED AT PI/2(7)
                                CA      MPAC
                                EXTEND
                                MP      1JACCQ
                                TC      DVOVSUB      # GET QUOTIENT WITH OVERFLOW PROTECTION
                                ADRES   TORKJET1

SPSCONT      TS      ACCDOTQ      # SCALED AT PI/2(7)
              EXTEND
              MP      DGBF      # .3ACCDOTQ SCALED AT PI/2(8)
              TS      KQ
              CAE     ACCDOTR      # .3ACCDOTR AT PI/2(8)
              EXTEND
              MP      DGBF
              TS      KRDA
              EXTEND
              READ    CHAN12      # NOW COMPUTE QACCDOT, RACCDOT, THE SIGNED
                                # JERK TERMS. STORE CHANNEL 12. WITH GIMBAL
              TS      MPAC      +1 # DRIVE BITS 9 THROUGH 12 SET LOOP
              CAF     BIT2      # INDEX TO COMPUTE RACCDOT, THEN QACCDOT.
              TCF     LOOP3
              CAF     ZERO      # ACCDOTQ AND ACCDOTR ARE NOT NEGATIVE,
LOOP3      TS      MPAC      # BECAUSE THEY ARE MAGNITUDES
              CA      MPAC      +1
              INDEX   MPAC      # MASK CHANNEL IMAGE FOR ANY GIMBAL MOTION
              MASK    GIMBLBTS
              EXTEND
              BZF     ZACCDOT      # IF NONE, Q(R)ACCDOT IS ZERO.
              CA      MPAC      +1
              INDEX   MPAC      # GIMBAL IS MOVING. IS ROTATION POSITIVE.
              MASK    GIMBLBTS +1
              EXTEND
              BZF     FRSTZERO      # IF NOT POSITIVE, BRANCH
              INDEX   MPAC      # POSITIVE ROTATION, NEGATIVE Q(R)ACCDOT.
              CS      ACCDOTQ
              TCF     STACCDOT
FRSTZERO      INDEX   MPAC      # NEGATIVE ROTATION, POSITIVE Q(R)ACCDOT.
              CA      ACCDOTQ
              TCF     STACCDOT

```

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ZACCDOT	CAF	ZERO	
STACCDOT	INDEX	MPAC	
	TS	QACCDOT	# STORE Q(R)ACCDOT.
	CCS	MPAC	
	TCF	LOOP3 -1	# NOW DO QACCDOT.

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CS	DAPBOOLS	# IS GIMBAL USABLE?
MASK	USEQRJTS	
EXTEND		
BZF	DOWNGTS	# NO. BE SURE THE GIMBAL SWITCHES ARE DOWN
CS	T5ADR	# YES. IS THE DAP RUNNINT?
AD	PAXISADR	
EXTEND		
BZF	+2	
TCF	DOWNGTS	# NO. BE SURE THE GIMBAL SWITCHES ARE DOWN
CCS	INGTS	# YES. IS GTS IN CONTROL?
TCF	DOCKTEST	# YES. PROCEED WITH 1/ACCS.
TC	IBNKCALL	# NO. NULL OFFSET AND FIND ALLOWGTS
CADR	TIMEGMBL	

DOCKTEST	CCS	DOCKTEMP	# BYPASS 1/ACCONT WHEN DOCKED.
	TCF	1/ACCET	
	TCF	1/ACCONT	

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(667 714)

```

# SUBROUTINE:  DVOVSUB
# AUTHOR:      C. WORK, MOD 0, 12 JUNE 68
# PURPOSE:     THIS SUBROUTINE PROVIDES A SINGLE-PRECISION MACHINE LANGUAGE DIVISION
#              (1) THE QUOTIENT, IF THE DIVISION WAS NORMAL.
#              (2) NEGMAX, IF THE QUOTIENT WAS IMPROPER AND NEGATIVE.
#              (3) POSMAX, IF THE QUOTIENT WAS IMPROPER AND POSITIVE OR IF THERE WAS
#              THE CALLING PROGRAM IS PRESUMED TO BE A JOB IN THE F BANK WHICH CONTAINS
#              THE DIVISOR FOR THIS ROUTINE MAY BE IN EITHER FIXED OR ERASABLE STORAGE
#              ASSUMED BETWEEN THE TWO HALVES OF THE DIVIDEND. (THIS IS CERTAIN IF THE
#              RESULT OF A MULTIPLICATION OPERATION.)
# CALL SEQUENCE:  L      TC      DVOVSUB
#                 L +1    ADRES    (DIVISOR)
#                 L +2    RETURN HERE, WITH RESULT IN A,L
# INPUT:          DIVIDEND IN A,L (SIGN AGREEMENT ASSUMED), DIVISOR IN LOCATION DESIGNATED BY
#                 DIVISOR MAY BE IN THE DVOVSUB FBANK, FIXED-FIXED FBANK, EBANK 6, OR UNASSIGNED FBANK
# OUTPUT:         QUOTIENT AND REMAINDER, OR POSMAX (NEGMAX), WHICHEVER IS APPROPRIATE
# DEBRIS:         SCRATCHX, SCRATCHY, SCRATCHZ, A, L (NOTE: SCRATCHX, Y, Z ARE EQUATED TO MEMORY)
# ABORTS OR ALARMS:  NONE
# EXITS:          TO THE CALL POINT +2.
# SUBROUTINES CALLED:  NONE.

```

```

DVOVSUB      TS      SCRATCHY      # SAVE UPPER HALF OF DIVIDEND
              TS      SCRATCHX
              INDEX   Q              # OBTAIN ADDRESS OF DIVISOR.
              CA      0
              INCR    Q              # STEP Q FOR PROPER RETURN SEQUENCE.
              INDEX   A
              CA      0              # PICK UP THE DIVISOR.
              EXTEND
              BZF     MAXPLUS        # RETURN POSMAX FOR A ZERO DIVISOR.

              TS      SCRATCHZ      # STORE DIVISOR.

              CCS     A              # GET ABS(DIVISOR) IN THE A REGISTER.
              AD      BIT1
              TCF     ZEROPLUS
              AD      BIT1

ZEROPLUS     XCH      SCRATCHY      # STORE ABS(DIVISOR).  PICK UP TOP HALF OF
              EXTEND
              BZMF    GOODNEG        # DIVIDEND.
              # GET -ABS(DIVIDEND)

```



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```
CS      A

GOODNEG  AD      SCRATCHY      # ABS(DIVISOR) - ABS(DIVIDEND)
          EXTEND
          BZMF     MAKEMAX      # BRANCH IF DIVISION IS NOT PROPER.

          CA      SCRATCHX      # RE-ESTABLISH THE DIVIDEND
          EXTEND
          DV      SCRATCHZ      # QUOTIENT IN THE A, REMAINDER IN L.
          TC      Q             # RETURN TO CALLER.

MAKEMAX   CCS      SCRATCHX      # DETERMINE THE SIGN OF THE QUOTIENT.
          CCS      SCRATCHZ      # SCRATCHX AND SCRATCHZ ARE NON-ZERO.
          TCF      MAXPLUS
          CCS      SCRATCHZ
          CAF      NEGMAX      # +,- OR -,+
          TC      Q
MAXPLUS   CAF      POSMAX      # -,- OR +,+
          TC      Q
```

# COEFFICIENTS FOR THE JET ACCELERATION CURVE FITS

# THE CURVE FITS ARE OF THE FORM --

#

# 1JACC = A/(MASS + C) + B

#

# A IS SCALED AT PI/4 RAD/SEC\*\*2 B+16KG, B IS SCALED AT PI/4 RAD/SEC\*\*2, AND C IS SCALED AT B +

#

# THE CURVE FIT FOR L,PVT-CG IS OF THE SAME FORM, EXCEPT THAT A IS SCALED AT 8 FT B+16 KG, B IS

# AND C IS SCALED AT B+16 KG.

```
2DEC     +.0410511917    # L      A      DESCENT

INERCONA 2DEC     +.0059347674    # 1JACP  A      DESCENT

2DEC     +.0014979264    # 1JACCQ  A      DESCENT

2DEC     +.0010451889    # 1JACCR  A      DESCENT

2DEC     +.0065443852    # 1JACP  A      ASCENT

2DEC     +.0035784354    # 1JACCQ  A      ASCENT

2DEC     +.0056946631    # 1JACCR  A      ASCENT

DEC       +.155044       # L      B      DESCENT
```

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DEC

-.025233

# L

C

DESCENT

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```
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      INERCONB DEC +.002989 # 1JACCP B DESCENT
      INERCONC DEC +.008721 # 1JACCP C DESCENT
      DEC +.018791 # 1JACCQ B DESCENT
      DEC -.068163 # 1JACCQ C DESCENT
      DEC +.021345 # 1JACCR B DESCENT
      DEC -.066027 # 1JACCR C DESCENT

      DEC +.000032 # 1JACCP B ASCENT
      DEC -.006923 # 1JACCP C ASCENT
      DEC +.162862 # 1JACCQ B ASCENT
      DEC +.002588 # 1JACCQ C ASCENT
      DEC +.009312 # 1JACCR B ASCENT
      DEC -.023608 # 1JACCR C ASCENT

      GIMBLBTS OCTAL 01400
      OCTAL 01000
      OCTAL 06000
      OCTAL 04000
      DGBF DEC 0.6 # .3 SCALED AT 1/2
      0.35356 DEC 0.35356 # .70711 SCALED AT 2
      GFACTM OCT 337 # 979.24/2.20462 AT B+15
      .7071 DEC .70711
      -.7071 DEC -.70711
      -EPSMAX DEC -.42265

# CSM-DOCKED INERTIA COMPUTATIONS

DOCKED CA ONE # COEFTR = 1 FOR INERTIA COEFFICIENTS
SPSLOOP1 TS COEFCTR # = 7 FOR CG COEFFICIENTS
CA ONE # MASSCTR = 1 FOR CSM
TS MASSCTR # = 0 FOR LEM

INDEX COEFCTR
CA COEFF -1 # COEFF -1 = C
EXTEND
MP LEMASS
EXTEND
MP CSMMASS # LET X = CSMMASS AND Y = LEMASS

INDEX COEFCTR
AD COEFF # COEFF = F
TS MPAC # MPAC = C X Y + F
TCF +4

SPSLOOP2 TS MASSCTR # LOOP TWICE THROUGH HERE TO OBTAIN
```

```
EXTEND      # MPAC = MPAC + (A X +D)X + (B Y +E)Y
DIM COEFCTR  #
INDEX COEFCTR
CA COEFF +2  # COEFF +2 = A OR B
EXTEND
```

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```

      INDEX  MASSCTR
      MP      LEMMASS
      INDEX  COEFCTR
      AD      COEFF  +4      # COEFF +4 = E OR D
      EXTEND
      INDEX  MASSCTR
      MP      LEMMASS
      ADS     MPAC

      CCS     MASSCTR
      TCF     SPSLOOP2
      CCS     COEFCTR      # IF COEFCTR IS POS, EXIT FROM LOOP WITH
      TCF     +7           # CG X DELDOT = MPAC X 4 PI RAD-CM/SEC
      TORQCONS 2DEC      0.51443 B-14 # CORRESPONDS TO 500 LB-FT

      CA      MPAC
      TS      MPAC  +1      # INERTIA = (MPAC +1) X 2(38) KG-CM(2)
      CA      SEVEN
      TCF     SPSLOOP1

      CA      1JACCCON      # 1JACC=1JACCCON/MASS
      ZL
      TC      DVOVSUB
      ADRES   MASS
      TS      1JACC         # SCALED AT PI/4

      CA      POSMAX        # SET INVERSE JET ACCELERATIONS TO POSMAX,
      TS      1/ANETP        # WHICH CORRESPONDS TO ACCEL. OF 1.4 D/SS.
      TS      1/ANET2 +1
      TS      1/ANET2 +2
      TS      1/ANET2 +17D
      TS      1/ANET2 +18D
      EXTEND
      DCA     TORQCONS
      EXTEND
      DV      MPAC  +1
      INHINT
      TS      1JACCQ        # SCALED AT PI/4
      TS      1JACCR

      CA      -.7071
      TS      COEFFQ        # COEFFQ AND COEFFR ARE CHOSEN TO MAKE U-
      CA      .7071         # AND V-AXES ORTHOGONAL FOR DOCKED CASE
      TS      COEFFR
      CA      MASS         # SCALED AT 2(16) KG
```

```

EXTEND
MP      MPAC      # SCALED AT 4 PI RAD-CM/SEC
EXTEND
MP      ABDELV    # SCALED AT 2(13) CM/SEC(2)
TC      DVOVSUB   # GET QUOTIENT WITH OVERFLOW PROTECTION

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      ADRES      MPAC      +1

      TS         ACCDOTR
      TCF        SPSCONT   # CONTINUE K, KSQ CALCULATIONS

1JACCON      OCT      00167      # SCALED AT PI/4X2(16) RAD/SEC(2)-KG

#                                     2      2
# COEFFICIENTS FOR CURVE FIT OF THE FORM Z = A X  +B Y  +C X Y +D X +E Y +F

COEFF      DEC      .19518      # C COEFFICIENT OF INERTIA
           DEC      -.00529      # F      "
           DEC      -.17670      # B      "
           DEC      -.03709      # A      "
           DEC      .06974      # E      "
           DEC      .02569      # D      "

           DEC      .20096      # C COEFFICIENT OF CG
           DEC      .13564      # F      "
           DEC      .75704      # B      "
           DEC      -.37142      # A      "
           DEC      -.63117      # E      "
           DEC      .41179      # D      "

# ASSIGNMENT OF TEMPORARIES FOR 1/ACCS (EXCLUDING 1/ACCONT)
# MPAC, MPAC +1, MPAC +2 USED EXPLICITLY
COEFCTR      EQUALS      MPAC      +4
MASSCTR      EQUALS      MPAC      +5
SCRATCHX     EQUALS      MPAC      +4      # SCRATCH AREA FOR DVOVSUB ROUTINE.
SCRATCHY     EQUALS      SCRATCHX +1
SCRATCHZ     EQUALS      SCRATCHX +2

DOCKTEMP     EQUALS      MPAC      +3      # RECORD OF CSMDOCKED BIT OF DAPBOOLS
EPSILON      EQUALS      MPAC      +1
-EPILON      EQUALS      EPSILON
-.1875       DEC      -.18750

```

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```

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                                BANK      20
                                SETLOC    DAPS3
                                BANK

                                EBANK=    AOSQ

                                COUNT*    $$/DAPAO

-1      TS      INGTS      # ZERO INGTS IN ASCENT
1/ACCONT CA      DB        # INITIALIZE DBVAL1,2,3
      EXTEND
      MP      BIT13
      TS      L            # 0.25 DB
      AD      A
      TS      DBVAL3      # 0.50 DB
      CS      DBVAL1
      AD      L
      TS      DBVAL2      # -.75 DB

GETAOSUV INHINT
      CAE      AOSR      # COMPUTE ASOU AND AOSV BY ROTATING
      TS      L            #
      CAE      AOSQ
      TC      IBNKCALL
      CADR     ROT-TOUV
      DXCH     AOSU

      RELINT
      CA      DAPBOOLS
      MASK     DRIFTBIT   # ZERO DURING ULLAGE AND POWERED FLIGHT.
      CCS      A          # IF DRIFTING LIGHT,
      CA      ONE         # SET DRIFTER TO 1
      TS      DRIFTER     # SAVE TO TEST FOR DRIFTING FLIGHT LATER
      AD      ALLOWGTS    # NON-ZERO IF DRIFT OR GTS NEAR
      CCS      A
      CA      FLATVAL     # DRIFTING FLIGHT, STORE .8 IN FLAT
      TS      FLATEMP     # IN POWERED FLIGHT, STORE ZERO IN FLAT
      EXTEND
      BZF      DOPAXIS    # IF POWERED AND NO GTS, START P AXIS,
      CCS      DRIFTER    # OTHERWISE SET ZONE3LIM
      CA      ZONE3MAX    # 17.5 MS, SCALED AT 4 SECONDS.
      TS      Z3TEM

DOPAXIS CA      1JACC     # 1JACC AT PI/4 = 2JACC AT PI/2 =
                                # ANET AT PI/2 = ANET/ACOAOST AT 2(6).

```

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AD BIT9  
TS FUNTEM  
CA 1JACC

# 1 + ANET/ACOAST AT 2(6)



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```
TC      INVERT
INHINT
TS      1/ANETP      # P AXIS DATA MUST BE CONSISTENT
TS      1/ANETP +1   # SCALED AT 2(7)/PI.

CS      BIT9         # -1 AT 2(6)
EXTEND
MP      1/ANETP      # -1/ANET AT 2(13)/PI
EXTEND
DV      FUNTEM       # -1/(ANET + ANET**2/ACOAST) AT 2(7)/PI
TS      PACCFUN
TS      PACCFUN +1

CA      1/.03        # NO AOS FOR P AXIS, ACOAST = AMIN
TS      1/ACOSTP
TS      1/ACOSTP +1
RELINT

ZL
CCS     DRIFTER
DXCH    AOSU         # ZERO AOSU,V IF IN DRIFT, JUST TO BE SURE

UAXIS   CA      ZERO      # DO U AXIS COMPUTATIONS
        TS      UV        # ZERO FOR U AXIS, ONE FOR V AXIS.

BOTHAXES TS      SIGNAOS   # CODING COMMON TO U,V AXES
        INDEX   UV
        CCS     AOSU       # PICK UP ABS(AOSU OR AOSV)
        AD      ONE       # RESTORE TO PROPER VALUE
        TCF     +3        # AND LEAVE SIGNAOS AT ZERO
        AD      ONE       # NEGATIVE, RESTORE TO PROPER VALUE
        INCR    SIGNAOS   # AND SET SIGNAOS TO ONE TO SHOW AOS NEG
        TS      ABSAOS    # SAVE ABS(AOS)
        CS      SIGNAOS
        TS      -SIGNAOS  # USED AS AN INDEX

        CA      DBVAL1    # SET DB1, DB2 TO DBVAL1 (= DB)
        TS      DBB1
        TS      DBB2

        CA      ABSAOS    # TEST MAGNITUDE OF ABS(AOS)
        AD      -.03R/S2
        EXTEND
        BZMF    NOTMUCH   # ABS(AOS) LESS THAN AMIN
        BIGAOS  CCS      FLATEMP # AGS(AOS) GREATER THAN AMIN
```

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TCF	SKIPDB1	# I DRIFT OR GTS, DO NOT COMPUTE DB
CA	DBVAL1	
INDEX	-SIGNAOS	

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```

ADS      DBB2      # DB2(1) = 2 DB
INDEX    SIGNAOS
TS        DBB4      # DB4(3) = 1 DB
CA        -.1875    # -.1875 PI/2 RAD/SEC(2) SCALED AT PI/2
AD        ABSAOS     # ABSAOS IS SCALED AT PI/2
EXTEND
BZMF      +3
CS        DBVAL3     # -.5 DB
TCF       DBONE
CS        ABSAOS
DOUBLE
DOUBLE
AD        BIT14
DOUBLE    # 1-8 ABSAOS. (8 IS 16/PI SCALED AT 2/PI)
EXTEND
MP        DB
DBONE     INDEX     SIGNAOS      # DB1(2)=(1-8 ABSAOS) DB. IF ABSAOS IS
TS        DBB1      # GREATER THAN .1875 THEN DB1(2) = -.5 DB
CA        DBVAL2
INDEX     -SIGNAOS
TS        DBB3      # DB3(4) = -.75 DB

SKIPDB1   CA        ABSAOS      # ABS(AOS) GREATER THAN AMIN, SO IT IS
EXTEND
MP        BIT12
AD        ABSAOS      # (9/8) ABSAOS.
TC        INVERT      # ALL RIGHT TO DIVIDE
INDEX     -SIGNAOS
TS        1/ACOSTT +1  # 1/ACOSTPOS(NET) = 1/ABS(AOS)
CA        1/.03
INDEX     SIGNAOS
TS        1/ACOSTT     # 1/ACOSTNEG(POS) = 1/AIN

CA        ABSAOS
AD        1JACCU
AD        1JACCU      # 2 JACC + ABS(AOS)
AD        BIT9        # MAXIMUM VALUE IN COMPUTATIONS
TS        A           # TEST FOR OVERFLOW
TCF       SKIPDB2     # NO OVERFLOW, DO NORMAL COMPUTATION

CA        ABSAOS      # RESCALE TO PI TO PREVENT OVERFLOW
EXTEND
MP        BIT14
AD        1JACCU      # 1 JACC AT PI/2 = 2JACC AT PI
TS        ANET        # ANETPOS(NEG) MAX SCALED AT PI =
```

AD BIT8  
XCH ANET  
TC INVERT  
EXTEND

# ANETPOS(NEG) MAX/ACOASTNEG(POS) AT 2(7)  
# 1 + ANETPOS/ACOASTNEG AT 2(7)  
# SAVE IN ANET, WHILE PICKING UP ANET

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693  (Page LM1500 693)≡
                                     (667 714)
MP      BIT14      # SCALE 1/ANET AT 2(7)/PI
TS      1/ANET
CA      ACCHERE    # SET UP RETURN FROM COMPUTATION ROUTINE
TS      ARET
CS      BIT8        # -1 AT 2(7)
TCF     DOACCFUN    # FINISH ACCFUN COMPUTATION

ACCHERE    TCF     ACCTHERE

NOTMUCH    TS      L      # ABS(AOS) LESS THAN AMIN, SAVE IN L
CA         1/.03      # ACOASTPOS,NEG = AMIN
TS         1/ACOSTT
TS         1/ACOSTT +1

CCS        FLATEMP
TCF        SKIPDB2    # DO NOT COMPUTE DB IF DRIFT OR GTS

CA         .0125RS    # AMIN/2
AD         L          # L HAS ABS(AOS) - AMIN
EXTEND
BZMF      NOAOS       # RESULT IS ABS(AOS)- AMIN/2
                                     # ABS(AOS) LESS THAN AMIN/2

SOMEAOS    CA      DBVAL3    # AMIN/2 LT ABS(AOS) LT AMIN
INDEX     -SIGNAOS
TS        DBB3        # DB3(4) = DB/2
AD        A
INDEX     SIGNAOS
TS        DBB4        # DB4(3) = DB
TCF       SKIPDB2

NOAOS      CA      DBVAL1
TS        DBB3        # DB3,4 = DB
TS        DBB4

SKIPDB2    CA      ABSAOS    # ANETPOS(NEG) MAX = 2 JACC + ABS(AOS)
AD        1JACCU
AD        1JACCU
TS        ANET        # CANNOT OVERFLOW HERE
CL1/NET+   TC      DO1/NET+  # COMPUTE 1/ANET, ACCFUN

ACCTHERE   INDEX    -SIGNAOS
TS         Z5TEM    +2      # STORE ACCFUN IN TEMPORARY BUFFER
CA         1/ANET
INDEX     -SIGNAOS

```

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TS	1/ATEM2 +2	# STORE 1/ANET IN TEMPORARY BUFFER
CA	ABSAOS	# SEE IF OVERFLOW IN MIN CASE
AD	1JACCU	

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```

AD      BIT9      # MAXIMUM POSSIBLE VALUE
TS      A          # OVERFLOW POSSIBLE BUT REMOTE
TCF     +2
CA      POSMAX     # IF OVERFLOW, TRUNCATE TO PI/2
AD      -.03R/S2   # RESTORE TO CORRECT VALUE
TS      ANET
TC      DO1/NET+   # COMPUTE 1/ANET, ACCFUN

INDEX   -SIGNAOS   # STORE MIN VALUES JUST AS MAX VALUES
TS      Z5TEM
CA      1/ANET
INDEX   -SIGNAOS
TS      1/ATEM2

CS      ABSAOS     # NOW DO NEG(POS) CASES
AD      1JACCU
AD      1JACCU     # ANETNEG(POS) MAX
TC      1/ANET-    # COMPUTE 1/ANET, ACCFUN, AND ACCSW
INDEX   SIGNAOS    # STORE NEG(POS) VALUES JUST AS POS(NEG)
TS      Z1TEM      +2
TS      L          # SAVE IN L FOR POSSIBLE FUTURE USE
CA      1/ANET
INDEX   SIGNAOS
TS      1/ATEM1    +2
CS      ABSAOS
AD      1JACCU     # 1/ANETNEG(POS) MIN
TS      ANET
AD      -.03R/S2   # TEST FOR AMIN
EXTEND                      # IF ANET LESS THAN AMIN, STORE MAX JET
BZMF    FIXMIN     # VALUES FOR MIN JETS AND SET ACCSW

TC      1/NETMIN   # OTHERWISE DO MIN JET COMPUTATIONS
INDEX   SIGNAOS   # STORE VALUES
TS      Z1TEM
CA      1/ANET
INDEX   SIGNAOS
TS      1/ATEM1

INDEX   UV
CA      +UMASK
MASK    CH5MASK    # TEST FOR +U (+V) JET FAILURES
EXTEND
BZF     FAIL-
CA      1/ATEM2    # REPLACE FUNCTION VALUES DEPENDING ON THE
TS      1/ATEM2    +2 # FAILED JET PAIR WITH CORRESPONDING ONE-

```

STMIN-

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	CA	Z5TEM		# JET (OR AMIN) FUNCTION VALUES
	TS	Z5TEM	+2	
FAIL-	INDEX	UV		



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```

      CA      -UMASK
      MASK     CH5MASK      # TEST FOR -U (-V) JET FAILURES
      EXTEND
      BZF      DBFUN
      CA      1/ATEM1      # REPLACE FUNCTION VALUES DEPENDING ON THE
      TS      1/ATEM1 +2    # FAILED JET PAIR WITH CORRESPONDING ONE-
      CA      Z1TEM        # JET (OR AMIN) FUNCTION VALUES
      TS      Z1TEM +2

      DBFUN      CS      DBB3      # COMPUTE AXISDIST
                  AD      DBB1
                  AD      FLATEMP
                  TS      AXDSTEM
                  CS      DBB4
                  AD      DBB2
                  AD      FLATEMP
                  TS      AXDSTEM +1

                  INHINT
                  CCS      UV      # TEST FOR U OR V AXIS
                  TCF      STORV    # V AXIS      STORE V VALUES

                  CA      ACCSW     # U AXIS      STORE U VALUES
                  TS      ACCSWU

                  CA      NINE      # TRANSFER 10 WORDS VIA GENTRAN
                  TC      GENTRAN +1
                  ADRES    1/ATEM1  # TEMPORARY BUFFER
                  ADRES    1/ANET1  # THE REAL PLACE

                  RELINT
                  DXCH     DBB1      # SAVE U DBS FOR LATER STORING
                  DXCH     UDB1
                  DXCH     DBB4
                  DXCH     UDB4

                  DXCH     AXDSTEM
                  DXCH     UAXDIST

                  CA      ONE      # NOW DO V AXIS
                  TS      UV
                  CA      ZERO
                  TCF      BOTHAXES # AND DO IT AGAIN

      STORV      CA      ACCSW     # STORE V AXIS VALUES
```

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TS	ACCSWV
CA	NINE
TC	GENTRAN +1

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```
699      (Page LM1503 699)≡(667 714)
      ADRES 1/ATEM1      # TEMPORARY BUFFER
      ADRES 1/ANET1 +16D # THE REAL PLACE

      DXCH FLATEMP      # NOW STORE DEADBANDS FOR ALL AXES
      DXCH FLAT         # FLAT AND ZONE3LIM

      CA DBVAL1         # COMPUTE P AXIS DEADBANDS
      TS PDB1
      TS PDB2
      AD FLAT
      TS PDB3
      TS PDB4
      CA ZERO
      TS PAXDIST
      TS PAXDIST +1

      CCS FLAT
      TCF DRFDB         # DRIFT OR GTS -- COMPUTE DBS

      DXCH UDB1         # STORE U DEADBANDS
      DXCH FIREDDB      # CANNOT USE GENTRAN BECAUSE OF RELINT
      DXCH UDB4
      DXCH COASTDB
      DXCH UAXDIST
      DXCH AXISDIST
      DXCH DBB1         # STORE V AXIS DEADBANDS
      DXCH FIREDDB +16D # COULD USE GENTRAN IF DESIRED
      DXCH DBB4
      DXCH COASTDB +16D
      DXCH AXDSTEM
      DXCH AXISDIST +16D

      TCF 1/ACCRET +1   # ALL DONE
DRFDB CA DBVAL1        # DRIFT DEADBANDS
      TS FIREDDB
      TS FIREDDB +1
      TS FIREDDB +16D
      TS FIREDDB +17D
      AD FLAT
      TS COASTDB
      TS COASTDB +1
      TS COASTDB +16D
      TS COASTDB +17D
      CA ZERO
```

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TS	AXISDIST
TS	AXISDIST +1
TS	AXISDIST +16D
TS	AXISDIST +17D

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```
701  (Page LM1504 701)≡ (667 714)
      1/ACCRET          INHINT
                        CS      DAPBOOLS      # SET BIT TO INDICATE DATA GOOD.
                        MASK     ACCSOKAY
                        ADS      DAPBOOLS
                        RELINT
                        CA       ACCRETRN
                        TC       BANKJUMP      # RETURN TO CALLER

      INVERT            TS      HOLD          # ROUTINE TO INVERT -INPUT AT PI/2
                        CA       BIT9          # 1 AT 2(6)
                        ZL
                        EXTEND                # ZERO L FOR ACCURACY AND TO PREVENT OVFL0
                        DV       HOLD
                        TC       Q             # RESULT AT 2(7)/PI

      DOWNGTS           CAF      ZERO          # ZERO SWITCHES WHEN USEQRJTS BIT IS UP
                        TS       ALLOWGTS      # OR DAP IS OFF
                        TS       INGTS
                        TCF      DOCKTEST

      1/ANET-           ZL
                        LXCH     ACCSW         # ZERO ACCSW
                        TS       ANET          # SAVE ANET
                        AD       -.03R/S2      # TEST FOR MIN VALUE
                        EXTEND
                        BZMF      NETNEG        # ANET LESS THAN AMIN, SO FAKE IT
      1/NETMIN          CA       ANET
                        EXTEND
                        INDEX     -SIGNAOS
                        MP        1/ACOSTT +1  # ANETNEG(POS)/ACOSTPOS(NEG) AT 2(6)

      # THE FOLLOWING CODING IS VALID FOR BOTH POS OR NEG
      #      VALUES OF AOS

      DO1/NET+          AD       BIT9          # 1 + ANET/ACOST AT 2(6)
                        XCH      ANET          # SAVE AND PICK UP ANET
                        EXTEND
                        QXCH     ARET          # SAVE RETURN
                        TC       INVERT
                        TS       1/ANET        # 1/ANET AT 2(7)/PI
                        CS       BIT9          # -1 AT 2(6)
      DOACCFUN          EXTEND
                        MP        1/ANET        # -1/ANET AT 2(13)/PI
                        EXTEND
                        DV       ANET          # ACCFUN AT 2(7)/PI
```

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	TC	ARET	# RETURN
NETNEG	CS	-.03R/S2	# ANET LESS THAN AMIN -- SET EQUAL TO AMIN
	TS	ANET	

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703 (Page LM1505 703)≡

(667 714)

```
TCF      1/NETMIN +1      # CONTINUE AS IF NOTHING HAPPENED.

FIXMIN    CCS      SIGNAOS
CA        TWO           # IF AOS NEG, ACCSW = +1
AD        NEGONE        # IF AOS POS, ACCSW = -1
TS        ACCSW
AD        UV            # IF ACCSW = +1, TEST FOR +U (+V) JET FAIL
INDEX     A             # IF ACCSW = -1, TEST FOR -U (-V) JET FAIL
CA        -UMASK  +1
MASK      CH5MASK
EXTEND
BZF       +4
CS        -.03R/S2      # JET FAILURE -- CANNOT USE 2-JET VALUES
TS        ANET           # ANET = AMIN
TCF       STMIN-  -1     # CALCULATE FUNCTIONS USING AMIN
CA        L              # L HAS ACCFUN
TCF       STMIN-        # STORE MAX VALUES FOR MIN JETS

# ERASABLE ASSIGNMENTS FOR 1/ACCONT

1/ANETP    EQUALS  BLOCKTOP +2
1/ACOSTP    EQUALS  BLOCKTOP +4
PACCFUN     EQUALS  BLOCKTOP +8D
PDB1        EQUALS  BLOCKTOP +10D
PDB2        EQUALS  BLOCKTOP +11D
PDB4        EQUALS  BLOCKTOP +12D
PDB3        EQUALS  BLOCKTOP +13D
PAXDIST     EQUALS  BLOCKTOP +14D

ACCSW       EQUALS  VBUF      # EXECUTIVE TEMPORARIES
                                # CANNOT DO CCS NEWJOB DURING 1/ACCS
1/ATEM1     EQUALS  ACCSW  +1  # TEMP BUFFER FOR U AND V AXES
1/ATEM2     EQUALS  1/ATEM1 +1
1/ACOSTT     EQUALS  1/ATEM1 +4
Z1TEM       EQUALS  1/ATEM1 +6
Z5TEM       EQUALS  1/ATEM1 +7

UDB1        EQUALS  1/ATEM1 +10D # UAXIS DEADBAND BUFFER
UDB2        EQUALS  1/ATEM1 +11D
UDB4        EQUALS  1/ATEM1 +12D
UDB3        EQUALS  1/ATEM1 +13D
UAXDIST     EQUALS  1/ATEM1 +14D

DBB1        EQUALS  1/ATEM1 +16D # TEMP DEADBAND BUFFER, ALSO V AXIS
DBB2        EQUALS  1/ATEM1 +17D
```

DBB4	EQUALS	1/ATEM1 +18D
DBB3	EQUALS	1/ATEM1 +19D
AXDSTEM	EQUALS	1/ATEM1 +20D

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FLATEMP	EQUALS	1/ATEM1 +22D	
Z3TEM	EQUALS	1/ATEM1 +23D	# MUST FOLLOW FLATEMP

DBVAL1	EQUALS	DB
DBVAL2	EQUALS	INTB15+
DBVAL3	EQUALS	INTB15+ +1

DRIFTER	EQUALS	INTB15+ +2
---------	--------	------------

UV	EQUALS	MPAC
ANET	EQUALS	MPAC +3
FUNTEM	EQUALS	MPAC +3
1/ANET	EQUALS	MPAC +4
ARET	EQUALS	MPAC +5
ABSAOS	EQUALS	MPAC +6
SIGNAOS	EQUALS	MPAC +7
-SIGNAOS	EQUALS	MPAC +8D
HOLD	EQUALS	MPAC +9D
ACCRETRN	EQUALS	FIXLOC -1

ZONE3MAX	DEC	.004375	# 17.5 MS (35 MS FOR 1 JET) AT 4 SECONDS
FLATVAL	DEC	.01778	# .8 AT PI/4 RAD
-.03R/S2	OCT	77377	# -PI/2(7) AT PI/2

.0125RS	EQUALS	BIT8	# PI/2(+8) AT PI/2
1/.03	EQUALS	POSMAX	# 2(7)/PI AT 2(7)/PI

PAXISADR	GENADR	PAXIS
----------	--------	-------

			# THE FOLLOWING 4 CONSTANTS ARE JET
			# FAILURE MASKS AND ARE INDEXED
-UMASK	OCT	00110	# -U
	OCT	00022	# -V
+UMASK	OCT	00204	# +U
	OCT	00041	# +V



1.40 sps back up rcs control

705  $\langle \textit{sps back up rcs control 705} \rangle \equiv$  (7)  
 $\langle \textit{Page LM1507 706} \rangle$   
 $\langle \textit{Page LM1508 707} \rangle$   
 $\langle \textit{Page LM1509 709} \rangle$   
 $\langle \textit{Page LM1510 710} \rangle$

706 (Page LM1507 706)≡

(705 768)

```

# PROGRAM NAME:          SPSRCS
# AUTHOR:                EDGAR M. OSHIKA (AC ELECTRONICS)
# MODIFIED:              TO RETURN TO ALL AXES VIA Q BY P. S. WEISSMAN, OCT 7, 1968
# MODIFIED TO IMPROVE BENDING STABILITY BY G. KALAN, FEB. 14, 1969
#
# FUNCTIONAL DESCRIPTION:
#   THE PROGRAM CONTROLS THE FIRING OF ALL RCS JETS IN THE DOCKED CONFIGURATION AND
#   PLANE LOGIC.
#
#   1. JET SENSE TEST (SPSRCS)
#       IF JETS ARE FIRING NEGATIVELY, SET OLDSENSE NEGATIVE AND CONTINUE
#       IF JETS ARE FIRING POSITIVELY, SET OLDSENSE POSITIVE AND CONTINUE
#       IF JETS ARE NOT FIRING, SET OLDSENSE TO ZERO AND GO TO OUTER RATE LIMIT TEST
#
#   2. RATE DEAD BAND TEST
#       IF JETS ARE FIRING NEGATIVELY AND RATE IS GREATER THAN TARGET RATE, INHIBIT
#       JETS ON AND GO TO INHIBITION LOGIC. OTHERWISE, CONTINUE.
#       IF JETS ARE FIRING POSITIVELY AND RATE IS LESS THAN TARGET RATE, INHIBIT
#       JETS ON AND GO TO INHIBITION LOGIC. OTHERWISE, CONTINUE.
#
#   3. OUTER RATE LIMIT TEST (SPSSTART)
#       IF MAGNITUDE OF EDOT IS GREATER THAN 1.73 DEG/SEC SET JET FIRING TIME TO ZERO
#       TO REDUCE RATE AND GO TO INHIBITION LOGIC. OTHERWISE, CONTINUE.
#
#   4. COAST ZONE TEST
#       IF STATE (E,EDOT) IS BELOW LINE E + 4 X EDOT > -1.4 DEG AND EDOT IS POSITIVE
#       POSITIVE AND CONTINUE. OTHERWISE, SET JET FIRING TIME TO ZERO AND GO TO INHIBITION LOGIC.
#       IF STATE IS ABOVE LINE E + 4 X EDOT > +1.4 DEG AND EDOT IS GREATER THAN 1.73 DEG/SEC
#       AND CONTINUE. OTHERWISE, SET JET FIRING TIME TO ZERO AND GO TO INHIBITION LOGIC.
#
#   5. INHIBITION LOGIC
#       IF OLDSENSE IS NON-ZERO:
#           A) RETURN IF JET TIME AS THE SAME SIGN AS OLDSENSE
#           B) SET INHIBITION COUNTER* AND RETURN IF JET TIME IS ZERO
#           C) SET INHIBITION COUNTER,* SET JET TIME TO ZERO AND RETURN IF JET TIME IS
#              OF JET TIME IS OPPOSITE TO THAT OF OLDSENSE
#       IF OLDSENSE IS ZERO:
#           A) RETURN IF INHIBITION COUNTER IS NOT POSITIVE
#           B) SET JET TIME TO ZERO AND RETURN IF INHIBITION COUNTER IS POSITIVE
#
# *NOTE: INHIBITION COUNTERS CAN BE SET TO 4 OR 10 FOR THE P AND UV AXES
# RESPECTIVELY, IN SPSRCS. THEY ARE DECREMENTED BY ONE AT THE BEGINNING OF EACH

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```

#          EACH DAP PASS.
#
#          THE MINIMUM PULSE WIDTH OF THIS CONTROLLER IS DETERMINED BY THE REPETITION RATE AT WHICH
#          AND IS NOMINALLY 100 MS FOR ALL AXES IN DRIFTING FLIGHT.  DURING POWERED FLIGHT THE MINIMUM
#          P AXIS AND 200 MS FOR THE CONTROL OF THE U AND V AXES.
#
# CALLING SEQUENCE:
#          INHINT
#          TC      IBNKCALL
#          CADR    SPSRCE
#
# EXIT:
#          TC      Q
#
# ALARM/ABORT MODE:  NONE
#
# SUBROUTINES CALLED:  NONE
#
# INPUT:
#          E, EDOT
#          TJP, TJV, TJU          TJ MUST NOT BE NEGATIVE ZERO
#
# OUTPUT:
#          TJP, TJV, TJU
#
#          BANK    21
#          SETLOC  DAPS4
#          BANK
#
#          COUNT*  $$/DAPBU
#
#          EBANK=  TJU
#          RATELIM2 OCT    00632          # 1.125 DEG/SEC
#          POSTHRST CA     HALF
#
#          NDX     AXISCTR
#          TS      TJU
#          CCS     OLDSENSE
#          TCF     POSCHECK          # JETS FIRING POSITIVELY
#          TCF     CTRCHECK          # JETS OFF.  CHECK INHIBITION CTR
#          NEGCHECK INDEX  AXISCTR    # JETS FIRING NEGATIVELY
#          CS      TJU
#          CCS     A
#          TC      Q          # RETURN
#          TCF     +2
#          TCF     +1          # JETS COMMANDED OFF.  SET CTR AND RETURN
#          SETCTR  INDEX  AXISCTR    # JET FIRING REVERSAL COMMANDED.  SET CTR,

```

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CA UTIME

# SET JET TIME TO ZERO, AND RETURN

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```
709  (Page LM1509 709)≡ (705 768)
      INDEX  AXISCTR
      TS      UJETCTR
      ZAPTJ    CA      ZERO
      INDEX  AXISCTR
      TS      TJU
      TC      Q
      POSCHECK INDEX  AXISCTR
      CA      TJU
      TCF     NEGCHECK +2
      CTRCHECK INDEX  AXISCTR      # CHECK JET INHIBITION COUNTER
      CCS      UJETCTR
      TCF      +2
      TC      Q      # CTR IS NOT POSITIVE.  RETURN
      TCF     ZAPTJ    # CTR IS POSITIVE.  INHIBIT FIRINGS
      TC      Q      # CTR IS NOT POSITIVE.  RETURN
      OCT     00004
      UTIME   OCT     00012
      OCT     00012
      OLDSENSE EQUALS  DAPTREG1
      NEGFIRE CS      ONE      # JETS FIRING NEGATIVELY
      TS      OLDSENSE
      CA      EDOT
      TCF     +4
      PLUSFIRE CA      ONE
      TS      OLDSENSE
      CS      EDOT      # RATE DEAD BAND TEST
      LXCH    A
      CS      DAPBOOLS    # IF DRIFTBIT = 1, USE ZERO TARGET RATE
      MASK    DRIFTBIT    # IF DRIFTBIT = 0, USE 0.10 RATE TARGET
      CCS     A
      CA      RATEDB1
      AD      L
      EXTEND
      BZMF    SPSSTART
      TCF     POSTHRST +3
      SPSRCS  INDEX  AXISCTR      # JET SENSE TEST
      CCS     TJU
      TCF     PLUSFIRE    # JETS FIRING POSITIVELY
      TCF     +2
      TCF     NEGFIRE     # JETS FIRING NEGATIVELY
      TS      OLDSENSE    # JETS OFF
      SPSSTART CA      EDOT      # OUTER RATE LIMIT TEST
      EXTEND
      MP      RATELIM1
```

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```

710      <Page LM1510 710>≡
                                     (705 768)
                                     AD      E
                                     EXTEND
                                     MP      DKDB      # PAD LOADED DEADBAND.  FRESHSTART: 1.4 DEG
                                     EXTEND
                                     BZF      TJZERO

                                     EXTEND
                                     BZMF      +7
                                     CA      EDOT
                                     AD      RATELIM2
                                     EXTEND
                                     BZMF      TJZERO
NEGTHRST      CS      HALF
                                     TCF      POSTHRST +1
                                     CS      RATELIM2
                                     AD      EDOT
                                     EXTEND
                                     BZMF      POSTHRST
TJZERO      CA      ZERO
                                     TCF      POSTHRST +1

RATELIM1      =      CALLCODE      # = 00032, CORRESPONDING TO 1.73 DEG/SEC
RATEDB1      =      TBUILDFX      # = 00045, CORRESPONDS TO 0.101 DEG/SEC

# *** END OF LMDAP .015 ***

```

## Chapter 2

# Original Files

### 2.1 AGC BLOCK TWO SELF-CHECK

```
711  <src/Luminary099/AGC-BLOCK-TWO-SELF-CHECK.agc 711>≡
      # Copyright:    Public domain.
      # Filename:     AGC_BLOCK_TWO_SELF_CHECK.agc
      # Purpose:      Part of the source code for Luminary 1A build 099.
      #               It is part of the source code for the Lunar Module's (LM)
      #               Apollo Guidance Computer (AGC), for Apollo 11.
      # Assembler:    yaYUL
      # Contact:       Onno Hommes <ohommes@cmu.edu>.
      # Website:       www.ibiblio.org/apollo.
      # Pages:         1284-1293
      # Mod history:   2009-05-27 OH   Transcribed from page images.
      #
      # This source code has been transcribed or otherwise adapted from
      # digitized images of a hardcopy from the MIT Museum. The digitization
      # was performed by Paul Fjeld, and arranged for by Deborah Douglas of
      # the Museum. Many thanks to both. The images (with suitable reduction
      # in storage size and consequent reduction in image quality as well) are
      # available online at www.ibiblio.org/apollo. If for some reason you
      # find that the images are illegible, contact me at info@sandroid.org
      # about getting access to the (much) higher-quality images which Paul
      # actually created.
      #
      # Notations on the hardcopy document read, in part:
      #
      #       Assemble revision 001 of AGC program LMY99 by NASA 2021112-61
      #       16:27 JULY 14, 1969
      #
      # Page 1284
```

*⟨Page LM1284 510⟩*  
# Page 1285  
*⟨Page LM1285 512⟩*  
# Page 1286  
*⟨Page LM1286 514⟩*  
# Page 1287  
*⟨Page LM1287 516⟩*  
# Page 1288  
*⟨Page LM1288 518⟩*  
# Page 1289  
*⟨Page LM1289 520⟩*  
# Page 1290  
*⟨Page LM1290 522⟩*  
# Page 1291  
*⟨Page LM1291 524⟩*  
# Page 1292  
*⟨Page LM1292 526⟩*  
# Page 1293  
*⟨Page LM1293 527a⟩*

This code is written to file `src/Luminary099/AGC-BLOCK-TWO-SELF-CHECK.agc`.



## 2.2 AGS INITIALIZATION

```

713  <src/Luminary099/AGS-INITIALIZATION.agc 713>≡
      # Copyright:   Public domain.
      # Filename:    AGS-INITIALIZATION.agc
      # Purpose:     Part of the source code for Luminary 1A build 099.
      #              It is part of the source code for the Lunar Module's (LM)
      #              Apollo Guidance Computer (AGC), for Apollo 11.
      # Assembler:   yaYUL
      # Contact:      Hartmuth Gutsche <hgutsche@explornet.com>.
      # Website:      www.ibiblio.org/apollo.
      # Pages:        206-210
      # Mod history:  2009-05-19 HG   Transcribed from page images.
      #
      # This source code has been transcribed or otherwise adapted from
      # digitized images of a hardcopy from the MIT Museum.  The digitization
      # was performed by Paul Fjeld, and arranged for by Deborah Douglas of
      # the Museum.  Many thanks to both.  The images (with suitable reduction
      # in storage size and consequent reduction in image quality as well) are
      # available online at www.ibiblio.org/apollo.  If for some reason you
      # find that the images are illegible, contact me at info@sandroid.org
      # about getting access to the (much) higher-quality images which Paul
      # actually created.
      #
      # Notations on the hardcopy document read, in part:
      #
      #       Assemble revision 001 of AGC program LMY99 by NASA 2021112-61
      #       16:27 JULY 14, 1969

      # Page 206
      <Page LM0206 71>
      # Page 207
      <Page LM0207 73>
      # Page 208
      <Page LM0208 75>
      # Page 209
      <Page LM0209 77>
      # Page 210
      <Page LM0210 78a>

```

This code is written to file `src/Luminary099/AGS-INITIALIZATION.agc`.

## 2.3 AOSTASK AND AOSJOB

```

714  <src/Luminary099/AOSTASK-AND-AOSJOB.agc 714>≡
      # Copyright:   Public domain.
      # Filename:    AOSTASK_AND_AOSJOB.agc
      # Purpose:     Part of the source code for Luminary 1A build 099.
      #              It is part of the source code for the Lunar Module's (LM)
      #              Apollo Guidance Computer (AGC), for Apollo 11.
      # Assembler:   yaYUL
      # Contact:      Ron Burkey <info@sandroid.org>.
      # Website:      www.ibiblio.org/apollo.
      # Pages:        1485-1506
      # Mod history:  2009-05-27 RSB   Adapted from the corresponding
      #              Luminary131 file, using page
      #              images from Luminary 1A.
      #              2009-06-05 RSB   Corrected a memory-bank error type.
      #              2009-06-07 RSB   Corrected a typo.
      #
      # This source code has been transcribed or otherwise adapted from
      # digitized images of a hardcopy from the MIT Museum. The digitization
      # was performed by Paul Fjeld, and arranged for by Deborah Douglas of
      # the Museum. Many thanks to both. The images (with suitable reduction
      # in storage size and consequent reduction in image quality as well) are
      # available online at www.ibiblio.org/apollo. If for some reason you
      # find that the images are illegible, contact me at info@sandroid.org
      # about getting access to the (much) higher-quality images which Paul
      # actually created.
      #
      # Notations on the hardcopy document read, in part:
      #
      #      Assemble revision 001 of AGC program LMY99 by NASA 2021112-61
      #      16:27 JULY 14, 1969
      #
      # Page 1485
      # <Page LM1485 668>
      # Page 1486
      # <Page LM1486 670>
      # Page 1487
      # <Page LM1487 672>
      # Page 1488
      # <Page LM1488 674>
      # Page 1489
      # <Page LM1489 676>
      # Page 1490
      # <Page LM1490 678>
      # Page 1491

```

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# Page 1492  
*⟨Page LM1492 680⟩*  
# Page 1493  
*⟨Page LM1493 681⟩*  
# Page 1494  
*⟨Page LM1494 683⟩*  
# Page 1495  
*⟨Page LM1495 685⟩*  
# Page 1496  
*⟨Page LM1496 686⟩*  
# Page 1497  
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# Page 1500  
*⟨Page LM1500 693⟩*  
# Page 1501  
*⟨Page LM1501 695⟩*  
# Page 1502  
*⟨Page LM1502 697⟩*  
# Page 1503  
*⟨Page LM1503 699⟩*  
# Page 1504  
*⟨Page LM1504 701⟩*  
# Page 1505  
*⟨Page LM1505 703⟩*  
# Page 1506  
*⟨Page LM1506 704⟩*

This code is written to file `src/Luminary099/AOSTASK-AND-AOSJOB.agc`.

## 2.4 AOTMARK

```

716  <src/Luminary099/AOTMARK.agc 716>≡
      # Copyright:   Public domain.
      # Filename:    AOTMARK.agc
      # Purpose:     Part of the source code for Luminary 1A build 099.
      #              It is part of the source code for the Lunar Module's (LM)
      #              Apollo Guidance Computer (AGC), for Apollo 11.
      # Assembler:   yaYUL
      # Contact:      Ron Burkey <info@sandroid.org>.
      # Website:      www.ibiblio.org/apollo.
      # Pages:        244-261
      # Mod history:  2009-05-10 SN    (Sergio Navarro).  Started adapting
      #              from the Luminary131/ file of the same
      #              name, using Luminary099 page images.
      #
      # This source code has been transcribed or otherwise adapted from
      # digitized images of a hardcopy from the MIT Museum.  The digitization
      # was performed by Paul Fjeld, and arranged for by Deborah Douglas of
      # the Museum.  Many thanks to both.  The images (with suitable reduction
      # in storage size and consequent reduction in image quality as well) are
      # available online at www.ibiblio.org/apollo.  If for some reason you
      # find that the images are illegible, contact me at info@sandroid.org
      # about getting access to the (much) higher-quality images which Paul
      # actually created.
      #
      # Notations on the hardcopy document read, in part:
      #
      #      Assemble revision 001 of AGC program LMY99 by NASA 2021112-61
      #      16:27 JULY 14, 1969

      # Page 244
      <Page LM0244 79>
      # Page 245
      <Page LM0245 80>
      # Page 246
      <Page LM0246 81>
      # Page 247
      <Page LM0247 82>
      # Page 248
      <Page LM0248 83>
      # Page 249
      <Page LM0249 84>
      # Page 250
      <Page LM0250 86>
      # Page 251

```

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# Page 252  
*⟨Page LM0252 88⟩*  
# Page 253  
*⟨Page LM0253 90⟩*  
# Page 254  
*⟨Page LM0254 92⟩*  
# Page 255  
*⟨Page LM0255 94⟩*  
# Page 256  
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This code is written to file `src/Luminary099/AOTMARK.agc`.

## 2.5 ASCENT GUIDANCE

```

718  <src/Luminary099/ASCENT-GUIDANCE.agc 718>≡
      # Copyright:   Public domain.
      # Filename:    ASCENT_GUIDANCE.agc
      # Purpose:     Part of the source code for Luminary 1A build 099.
      #              It is part of the source code for the Lunar Module's (LM)
      #              Apollo Guidance Computer (AGC), for Apollo 11.
      # Assembler:   yaYUL
      # Contact:      Hartmuth Gutsche <hgutsche@explornet.com>.
      # Website:      www.ibiblio.org/apollo.
      # Pages:        843-856
      # Mod history:  2009-05-23 HG   Transcribed from page images.
      #              2009-06-05 RSB   Fixed a couple of typos.
      #              2009-06-07 RSB   Corrected a typo.
      #
      # This source code has been transcribed or otherwise adapted from
      # digitized images of a hardcopy from the MIT Museum. The digitization
      # was performed by Paul Fjeld, and arranged for by Deborah Douglas of
      # the Museum. Many thanks to both. The images (with suitable reduction
      # in storage size and consequent reduction in image quality as well) are
      # available online at www.ibiblio.org/apollo. If for some reason you
      # find that the images are illegible, contact me at info@sandroid.org
      # about getting access to the (much) higher-quality images which Paul
      # actually created.
      #
      # Notations on the hardcopy document read, in part:
      #
      #      Assemble revision 001 of AGC program LMY99 by NASA 2021112-61
      #      16:27 JULY 14, 1969
      #
      # Page 843
      <Page LM0843 373>
      # Page 844
      <Page LM0844 374>
      # Page 845
      <Page LM0845 375>
      # Page 846
      <Page LM0846 377>
      # Page 847
      <Page LM0847 379>
      # Page 848
      <Page LM0848 381>
      # Page 849
      <Page LM0849 383>
      # Page 850

```

*⟨Page LM0850 385⟩*  
# Page 851  
*⟨Page LM0851 387⟩*  
# Page 852  
*⟨Page LM0852 389⟩*  
# Page 853  
*⟨Page LM0853 391⟩*  
# Page 854  
*⟨Page LM0854 393⟩*  
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# Page 856  
*⟨Page LM0856 395⟩*

This code is written to file `src/Luminary099/ASCENT-GUIDANCE.agc`.

## 2.6 ATTITUDE MANEUVER ROUTINE

```

720  <src/Luminary099/ATTITUDE-MANEUVER-ROUTINE.agc 720>≡
      # Copyright:   Public domain.
      # Filename:    ATTITUDE_MANEUVER_ROUTINE.agc
      # Purpose:     Part of the source code for Luminary 1A build 099.
      #              It is part of the source code for the Lunar Module's (LM)
      #              Apollo Guidance Computer (AGC), for Apollo 11.
      # Assembler:   yaYUL
      # Contact:      Ron Burkey <info@sandroid.org>.
      # Website:      www.ibiblio.org/apollo.
      # Pages:        342-363
      # Mod history:  2009-05-16 RSB   Adapted from the corresponding
      #              Luminary131 file, using page
      #              images from Luminary 1A.
      #
      # This source code has been transcribed or otherwise adapted from
      # digitized images of a hardcopy from the MIT Museum. The digitization
      # was performed by Paul Fjeld, and arranged for by Deborah Douglas of
      # the Museum. Many thanks to both. The images (with suitable reduction
      # in storage size and consequent reduction in image quality as well) are
      # available online at www.ibiblio.org/apollo. If for some reason you
      # find that the images are illegible, contact me at info@sandroid.org
      # about getting access to the (much) higher-quality images which Paul
      # actually created.
      #
      # Notations on the hardcopy document read, in part:
      #
      #       Assemble revision 001 of AGC program LMY99 by NASA 2021112-61
      #       16:27 JULY 14, 1969
      #
      # Page 342
      <Page LM0342 113>
      # Page 343
      <Page LM0343 115>
      # Page 344
      <Page LM0344 117>
      # Page 345
      <Page LM0345 119>
      # Page 346
      <Page LM0346 121>
      # Page 347
      <Page LM0347 123>
      # Page 348
      <Page LM0348 125>
      # Page 349

```



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# Page 350  
*⟨Page LM0350 127⟩*  
# Page 351  
*⟨Page LM0351 128⟩*  
# Page 352  
*⟨Page LM0352 130⟩*  
# Page 353  
*⟨Page LM0353 132⟩*  
# Page 354  
*⟨Page LM0354 134⟩*  
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*⟨Page LM0355 135⟩*  
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*⟨Page LM0356 136⟩*  
# Page 357  
*⟨Page LM0357 138⟩*  
# Page 358  
*⟨Page LM0358 140⟩*  
# Page 359  
*⟨Page LM0359 142⟩*  
# Page 360  
*⟨Page LM0360 144⟩*  
# Page 361  
*⟨Page LM0361 146⟩*  
# Page 362  
*⟨Page LM0362 148⟩*  
# Page 363  
*⟨Page LM0363 149a⟩*

This code is written to file `src/Luminary099/ATTITUDE-MANEUVER-ROUTINE.agc`.

## 2.7 BURN BABY BURN—MASTER IGNITION ROUTINE

```

722  <src/Luminary099/BURN-BABY-BURN-MASTER-IGNITION-ROUTINE.agc 722>≡
      # Copyright:    Public domain.
      # Filename:     BURN_BABY_BURN--MASTER_IGNITION_ROUTINE.agc
      # Purpose:      Part of the source code for Luminary 1A build 099.
      #               It is part of the source code for the Lunar Module's (LM)
      #               Apollo Guidance Computer (AGC), for Apollo 11.
      # Assembler:    yaYUL
      # Contact:       Ron Burkey <info@sandroid.org>.
      # Website:       www.ibiblio.org/apollo.
      # Pages:         731-751
      # Mod history:   2009-05-19 RSB  Adapted from the corresponding
      #               Luminary131 file, using page
      #               images from Luminary 1A.
      #               2009-06-07 RSB  Corrected 3 typos.
      #               2009-07-23 RSB  Added Onno's notes on the naming
      #               of this function, which he got from
      #               Don Eyles.
      #
      # This source code has been transcribed or otherwise adapted from
      # digitized images of a hardcopy from the MIT Museum. The digitization
      # was performed by Paul Fjeld, and arranged for by Deborah Douglas of
      # the Museum. Many thanks to both. The images (with suitable reduction
      # in storage size and consequent reduction in image quality as well) are
      # available online at www.ibiblio.org/apollo. If for some reason you
      # find that the images are illegible, contact me at info@sandroid.org
      # about getting access to the (much) higher-quality images which Paul
      # actually created.
      #
      # Notations on the hardcopy document read, in part:
      #
      #       Assemble revision 001 of AGC program LMY99 by NASA 2021112-61
      #       16:27 JULY 14, 1969
      #
      # Page 731
      ## At the get-together of the AGC developers celebrating the 40th anniversary
      ## of the first moonwalk, Don Eyles (one of the authors of this routine along
      ## with Peter Adler) has related to us a little interesting history behind the
      ## naming of the routine.
      ##
      ## It traces back to 1965 and the Los Angeles riots, and was inspired
      ## by disc jockey extraordinaire and radio station owner Magnificent Montague.
      ## Magnificent Montague used the phrase "Burn, baby! BURN!" when spinning the

```

## hottest new records. Magnificent Montague was the charismatic voice of  
## soul music in Chicago, New York, and Los Angeles from the mid-1950s to  
## the mid-1960s.

*<Page LM0731 242>*

# Page 732

*<Page LM0732 244>*

# Page 733

*<Page LM0733 245>*

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*<Page LM0734 246>*

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*<Page LM0735 248>*

# Page 736

*<Page LM0736 250>*

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*<Page LM0737 252>*

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*<Page LM0738 254>*

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*<Page LM0739 256>*

# Page 740

*<Page LM0740 258>*

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*<Page LM0741 260>*

# Page 742

*<Page LM0742 262>*

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*<Page LM0743 264>*

# Page 744

*<Page LM0744 266>*

# Page 745

*<Page LM0745 267>*

# Page 746

*<Page LM0746 269>*

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*<Page LM0747 271>*

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*<Page LM0748 273>*

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*<Page LM0749 274>*

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*<Page LM0750 275>*

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*<Page LM0751 277>*

This code is written to file `src/Luminary099/BURN-BABY-BURN--MASTER-IGNITION-ROUTINE.agc`.

## 2.8 CONTROLLED CONSTANTS

```

724  <src/Luminary099/CONTROLLED-CONSTANTS.agc 724>≡
      # Copyright:   Public domain.
      # Filename:    CONTROLLED_CONSTANTS.agc
      # Purpose:     Part of the source code for Luminary 1A build 099.
      #              It is part of the source code for the Lunar Module's (LM)
      #              Apollo Guidance Computer (AGC), for Apollo 11.
      #
      # Assembler:   yaYUL
      # Contact:      Jim Lawton <jim.lawton@gmail.com>
      # Website:      www.ibiblio.org/apollo.
      # Pages:        038-053
      # Mod history:  2009-05-16      JVL      Transcribed from page images.
      #
      # This source code has been transcribed or otherwise adapted from digitized
      # images of a hardcopy from the MIT Museum. The digitization was performed
      # by Paul Fjeld, and arranged for by Deborah Douglas of the Museum. Many
      # thanks to both. The images (with suitable reduction in storage size and
      # consequent reduction in image quality as well) are available online at
      # www.ibiblio.org/apollo. If for some reason you find that the images are
      # illegible, contact me at info@sandroid.org about getting access to the
      # (much) higher-quality images which Paul actually created.
      #
      # Notations on the hardcopy document read, in part:
      #
      # Assemble revision 001 of AGC program LMY99 by NASA 2021112-061
      # 16:27 JULY 14, 1969

      # Page 38
      <Page LM0038 9>
      # Page 39
      <Page LM0039 10>
      # Page 40
      <Page LM0040 11>
      # Page 41
      <Page LM0041 12>
      # Page 42
      <Page LM0042 13>
      # Page 43
      <Page LM0043 14>
      # Page 44
      <Page LM0044 15>
      # Page 45
      <Page LM0045 16>
      # Page 46

```

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# Page 47  
*⟨Page LM0047 18⟩*  
# Page 48  
*⟨Page LM0048 19⟩*  
# Page 49  
*⟨Page LM0049 20⟩*  
# Page 50  
*⟨Page LM0050 21⟩*  
# Page 51  
*⟨Page LM0051 22⟩*  
# Page 52  
*⟨Page LM0052 23a⟩*  
# Page 53  
*⟨Page LM0053 23b⟩*

This code is written to file `src/Luminary099/CONTROLLED-CONSTANTS.agc`.

## 2.9 DAPIDLER PROGRAM

```

726  <src/Luminary099/DAPIDLER-PROGRAM.agc 726>≡
      # Copyright:   Public domain.
      # Filename:    DAPIDLER_PROGRAM.agc
      # Purpose:     Part of the source code for Luminary 1A build 099.
      #              It is part of the source code for the Lunar Module's (LM)
      #              Apollo Guidance Computer (AGC), for Apollo 11.
      # Assembler:   yaYUL
      # Contact:      Ron Burkey <info@sandroid.org>.
      # Website:      www.ibiblio.org/apollo.
      # Pages:        1410-1420
      # Mod history:  2009-05-10 SN    (Sergio Navarro).  Started adapting
      #              from the Luminary131/ file of the same
      #              name, using Luminary099 page images.
      #
      # This source code has been transcribed or otherwise adapted from
      # digitized images of a hardcopy from the MIT Museum.  The digitization
      # was performed by Paul Fjeld, and arranged for by Deborah Douglas of
      # the Museum.  Many thanks to both.  The images (with suitable reduction
      # in storage size and consequent reduction in image quality as well) are
      # available online at www.ibiblio.org/apollo.  If for some reason you
      # find that the images are illegible, contact me at info@sandroid.org
      # about getting access to the (much) higher-quality images which Paul
      # actually created.
      #
      # Notations on the hardcopy document read, in part:
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      #      Assemble revision 001 of AGC program LMY99 by NASA 2021112-61
      #      16:27 JULY 14, 1969

      # Page 1410
      <Page LM1410 543>
      # Page 1411
      <Page LM1411 544>
      # Page 1412
      <Page LM1412 546>
      # Page 1413
      <Page LM1413 548>
      # Page 1414
      <Page LM1414 549>
      # Page 1415
      <Page LM1415 551>
      # Page 1416
      <Page LM1416 552>
      # Page 1417

```

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*⟨Page LM1418 554⟩*

**# Page 1419**

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**# Page 1420**

*⟨Page LM1420 557a⟩*

This code is written to file `src/Luminary099/DAPIDLER-PROGRAM.agc`.

## 2.10 DAP INTERFACE SUBROUTINES

```

728  <src/Luminary099/DAP-INTERFACE-SUBROUTINES.agc 728>≡
      # Copyright:   Public domain.
      # Filename:    DAP_INTERFACE_SUBROUTINES.agc
      # Purpose:     Part of the source code for Luminary 1A build 099.
      #              It is part of the source code for the Lunar Module's (LM)
      #              Apollo Guidance Computer (AGC), for Apollo 11.
      # Assembler:   yaYUL
      # Contact:      Ron Burkey <info@sandroid.org>.
      # Website:      www.ibiblio.org/apollo.
      # Pages:        1406-1409
      # Mod history:  2009-05-10 SN    (Sergio Navarro).  Started adapting
      #                               from the Luminary131/ file of the same
      #                               name, using Luminary099 page images.
      #
      # This source code has been transcribed or otherwise adapted from
      # digitized images of a hardcopy from the MIT Museum.  The digitization
      # was performed by Paul Fjeld, and arranged for by Deborah Douglas of
      # the Museum.  Many thanks to both.  The images (with suitable reduction
      # in storage size and consequent reduction in image quality as well) are
      # available online at www.ibiblio.org/apollo.  If for some reason you
      # find that the images are illegible, contact me at info@sandroid.org
      # about getting access to the (much) higher-quality images which Paul
      # actually created.
      #
      # Notations on the hardcopy document read, in part:
      #
      #      Assemble revision 001 of AGC program LMY99 by NASA 2021112-61
      #      16:27 JULY 14, 1969
      #
      # Page 1406
      <Page LM1406 538b>
      # Page 1407
      <Page LM1407 539>
      # Page 1408
      <Page LM1408 541>
      # Page 1409
      <Page LM1409 542a>

```

This code is written to file `src/Luminary099/DAP-INTERFACE-SUBROUTINES.agc`.



## 2.11 DOWN TELEMETRY PROGRAM

```

729  <src/Luminary099/DOWN-TELEMETRY-PROGRAM.agc 729>≡
    # Copyright:    Public domain.
    # Filename:     DOWN-TELEMETRY-PROGRAM.agc
    # Purpose:      Part of the source code for Luminary 1A build 099.
    #               It is part of the source code for the Lunar Module's (LM)
    #               Apollo Guidance Computer (AGC), for Apollo 11.
    # Assembler:    yaYUL
    # Contact:       Ron Burkey <info@sandroid.org>.
    # Website:       www.ibiblio.org/apollo.
    # Pages:         988-997
    # Mod history:   2009-05-24 RSB   Adapted from the corresponding
    #               Luminary131 file, using page
    #               images from Luminary 1A.
    #
    # This source code has been transcribed or otherwise adapted from
    # digitized images of a hardcopy from the MIT Museum. The digitization
    # was performed by Paul Fjeld, and arranged for by Deborah Douglas of
    # the Museum. Many thanks to both. The images (with suitable reduction
    # in storage size and consequent reduction in image quality as well) are
    # available online at www.ibiblio.org/apollo. If for some reason you
    # find that the images are illegible, contact me at info@sandroid.org
    # about getting access to the (much) higher-quality images which Paul
    # actually created.
    #
    # Notations on the hardcopy document read, in part:
    #
    #       Assemble revision 001 of AGC program LMY99 by NASA 2021112-61
    #       16:27 JULY 14, 1969

    # Page 988
    <Page LM0988 494>
    # Page 989 (empty page) <Page LM0989 495>
    # Page 990
    <Page LM0990 496>
    # Page 991
    <Page LM0991 498>
    # Page 992
    <Page LM0992 499>
    # Page 993
    <Page LM0993 501>
    # Page 994
    <Page LM0994 503>
    # Page 995
    <Page LM0995 504>

```

# Page 996

⟨Page LM0996 506⟩

# Page 997

⟨Page LM0997 507a⟩

This code is written to file `src/Luminary099/DOWN--TELEMETRY-PROGRAM.agc`.

## 2.12 FINDCDUW–GUIDAP INTERFACE

```

731  <src/Luminary099/FINDCDUW-GUIDAP-INTERFACE.agc 731>≡
      # Copyright:   Public domain.
      # Filename:    FINDCDUW--GUIDAP_INTERFACE.agc
      # Purpose:     Part of the source code for Luminary 1A build 099.
      #              It is part of the source code for the Lunar Module's (LM)
      #              Apollo Guidance Computer (AGC), for Apollo 11.
      # Assembler:   yaYUL
      # Contact:      Hartmuth Gutsche <hgutsche@explornet.com>.
      # Website:      www.ibiblio.org/apollo.
      # Pages:        908-925
      # Mod history:  2009-05-28 HG   Transcribed from page images.
      #
      # This source code has been transcribed or otherwise adapted from
      # digitized images of a hardcopy from the MIT Museum. The digitization
      # was performed by Paul Fjeld, and arranged for by Deborah Douglas of
      # the Museum. Many thanks to both. The images (with suitable reduction
      # in storage size and consequent reduction in image quality as well) are
      # available online at www.ibiblio.org/apollo. If for some reason you
      # find that the images are illegible, contact me at info@sandroid.org
      # about getting access to the (much) higher-quality images which Paul
      # actually created.
      #
      # Notations on the hardcopy document read, in part:
      #
      #       Assemble revision 001 of AGC program LMY99 by NASA 2021112-61
      #       16:27 JULY 14, 1969

      # Page 908
      <Page LM0908 473>
      # Page 909
      <Page LM0909 474>
      # Page 910
      <Page LM0910 475>
      # Page 911
      <Page LM0911 476>
      # Page 912
      <Page LM0912 477>
      # Page 913
      <Page LM0913 478>
      # Page 914
      <Page LM0914 479>
      # Page 915
      <Page LM0915 480>
      # Page 916

```

*⟨Page LM0916 481⟩*  
# Page 917  
*⟨Page LM0917 483⟩*  
# Page 918  
*⟨Page LM0918 485⟩*  
# Page 919  
*⟨Page LM0919 486⟩*  
# Page 920  
*⟨Page LM0920 487⟩*  
# Page 921  
*⟨Page LM0921 488⟩*  
# Page 922  
*⟨Page LM0922 489⟩*  
# Page 923  
*⟨Page LM0923 490⟩*  
# Page 924  
*⟨Page LM0924 491⟩*  
# Page 925  
*⟨Page LM0925 492⟩*

This code is written to file `src/Luminary099/FINDCDUW--GUIDAP-INTERFACE.agc`.

## 2.13 FLAGWORD ASSIGNMENTS

```

733  <src/Luminary099/FLAGWORD-ASSIGNMENTS.agc 733>≡
      # Copyright:   Public domain.
      # Filename:    FLAGWORD_ASSIGNMENTS.agc
      # Purpose:     Part of the source code for Luminary 1A build 099.
      #              It is part of the source code for the Lunar Module's (LM)
      #              Apollo Guidance Computer (AGC), for Apollo 11.
      # Assembler:   yaYUL
      # Contact:      Onno Hommes <ohommes@cmu.edu>.
      # Website:      www.ibiblio.org/apollo.
      # Pages:        0061-0089
      # Mod history:  2009-05-15 OH   Transcribed from page images.
      #              2009-05-17 RSB   Extended to (blank) p. 89.
      #
      # This source code has been transcribed or otherwise adapted from
      # digitized images of a hardcopy from the MIT Museum. The digitization
      # was performed by Paul Fjeld, and arranged for by Deborah Douglas of
      # the Museum. Many thanks to both. The images (with suitable reduction
      # in storage size and consequent reduction in image quality as well) are
      # available online at www.ibiblio.org/apollo. If for some reason you
      # find that the images are illegible, contact me at info@sandroid.org
      # about getting access to the (much) higher-quality images which Paul
      # actually created.
      #
      # Notations on the hardcopy document read, in part:
      #
      #      Assemble revision 001 of AGC program LMY99 by NASA 2021112-61
      #      16:27 JULY 14, 1969

      # Page 61
      <Page LM0061 32>
      # Page 62
      <Page LM0062 33>
      # Page63
      <Page LM0063 35>
      # Page 64
      <Page LM0064 37>
      # Page 65
      <Page LM0065 39>
      # Page 66
      <Page LM0066 40>
      # Page 67
      <Page LM0067 41>
      # Page 68

```

*<Page LM0068 42>*  
# Page 69  
*<Page LM0069 43>*  
# Page 70  
*<Page LM0070 44>*  
# Page 71  
*<Page LM0071 45>*  
# Page 72  
*<Page LM0072 46>*  
# Page 73  
*<Page LM0073 48>*  
# Page 74  
*<Page LM0074 49>*  
# Page 75  
*<Page LM0075 50>*  
# Page 76  
*<Page LM0076 51>*  
# Page 77  
*<Page LM0077 52>*  
# Page 78  
*<Page LM0078 53>*  
# Page 79  
*<Page LM0079 54>*  
# Page 80  
*<Page LM0080 55>*  
# Page 81  
*<Page LM0081 56>*  
# Page 82  
*<Page LM0082 57>*  
# Page 83  
*<Page LM0083 59>*  
# Page 84  
*<Page LM0084 60>*  
# Page 85  
*<Page LM0085 61>*  
# Page 86  
*<Page LM0086 62>*  
# Page 87  
*<Page LM0087 63>*  
# Page 88  
*<Page LM0088 64a>*  
# Page 89 (nothing on this page)  
*<Page LM0089 64b>*

This code is written to file `src/Luminary099/FLAGWORD-ASSIGNMENTS.agc`.

## 2.14 IMU PERFORMANCE TEST 2

```

735  <src/Luminary099/IMU-PERFORMANCE-TEST-2.agc 735>≡
    # Copyright:    Public domain.
    # Filename:     IMU_PERFORMANCE_TEST_2.agc
    # Purpose:      Part of the source code for Luminary 1A build 099.
    #               It is part of the source code for the Lunar Module's (LM)
    #               Apollo Guidance Computer (AGC), for Apollo 11.
    # Assembler:   yaYUL
    # Contact:      Ron Burkey <info@sandroid.org>.
    # Website:      www.ibiblio.org/apollo.
    # Pages:        373-381
    # Mod history:  2009-05-17 RSB   Adapted from the corresponding
    #               Luminary131 file, using page
    #               images from Luminary 1A.
    #
    # This source code has been transcribed or otherwise adapted from
    # digitized images of a hardcopy from the MIT Museum. The digitization
    # was performed by Paul Fjeld, and arranged for by Deborah Douglas of
    # the Museum. Many thanks to both. The images (with suitable reduction
    # in storage size and consequent reduction in image quality as well) are
    # available online at www.ibiblio.org/apollo. If for some reason you
    # find that the images are illegible, contact me at info@sandroid.org
    # about getting access to the (much) higher-quality images which Paul
    # actually created.
    #
    # Notations on the hardcopy document read, in part:
    #
    #       Assemble revision 001 of AGC program LMY99 by NASA 2021112-61
    #       16:27 JULY 14, 1969

    # Page 373
    <Page LM0373 150>
    # Page 374
    <Page LM0374 152>
    # Page 375
    <Page LM0375 154>
    # Page 376
    <Page LM0376 156>
    # Page 377
    <Page LM0377 157>
    # Page 378
    <Page LM0378 158>
    # Page 379
    <Page LM0379 159>
    # Page 380

```

*⟨Page LM0380 160⟩*

**# Page 381**

*⟨Page LM0381 162⟩*

This code is written to file `src/Luminary099/IMU-PERFORMANCE-TEST-2.agc`.



## 2.15 IMU PERFORMANCE TESTS 4

```

737  <src/Luminary099/IMU-PERFORMANCE-TESTS-4.agc 737>≡
    # Copyright:    Public domain.
    # Filename:     IMU_PERFORMANCE_TESTS_4.agc
    # Purpose:      Part of the source code for Luminary 1A build 099.
    #               It is part of the source code for the Lunar Module's (LM)
    #               Apollo Guidance Computer (AGC), for Apollo 11.
    # Assembler:    yaYUL
    # Contact:       Ron Burkey <info@sandroid.org>.
    # Website:       www.ibiblio.org/apollo.
    # Pages:         382-389
    # Mod history:   2009-05-17 RSB   Adapted from the corresponding
    #               Luminary131 file, using page
    #               images from Luminary 1A.
    #
    # This source code has been transcribed or otherwise adapted from
    # digitized images of a hardcopy from the MIT Museum. The digitization
    # was performed by Paul Fjeld, and arranged for by Deborah Douglas of
    # the Museum. Many thanks to both. The images (with suitable reduction
    # in storage size and consequent reduction in image quality as well) are
    # available online at www.ibiblio.org/apollo. If for some reason you
    # find that the images are illegible, contact me at info@sandroid.org
    # about getting access to the (much) higher-quality images which Paul
    # actually created.
    #
    # Notations on the hardcopy document read, in part:
    #
    #       Assemble revision 001 of AGC program LMY99 by NASA 2021112-61
    #       16:27 JULY 14, 1969

    # Page 382
    <Page LM0382 163b>
    # Page 383
    <Page LM0383 164>
    # Page 384
    <Page LM0384 165>
    # Page 385
    <Page LM0385 166>
    # Page 386
    <Page LM0386 168>
    # Page 387
    <Page LM0387 170>
    # Page 388
    <Page LM0388 172>
    # Page 389

```

*⟨Page LM0389 173a⟩*

This code is written to file `src/Luminary099/IMU-PERFORMANCE-TESTS-4.agc`.

## 2.16 INPUT OUTPUT CHANNEL BIT DESCRIPTIONS

```

739  <src/Luminary099/INPUT-OUTPUT-CHANNEL-BIT-DESCRIPTIONS.agc 739>≡
    # Copyright:    Public domain.
    # Filename:     INPUT_OUTPUT_CHANNEL_BIT_DESCRIPTIONS.agc
    # Purpose:      Part of the source code for Luminary 1A build 099.
    #               It is part of the source code for the Lunar Module's (LM)
    #               Apollo Guidance Computer (AGC), for Apollo 11.
    # Assembler:    yaYUL
    # Contact:       Onno Hommes <ohommes@cmu.edu>.
    # Website:       www.ibiblio.org/apollo.
    # Pages:         0054-0060
    # Mod history:   2009-05-14 OH   Transcribed from page images.
    #
    # This source code has been transcribed or otherwise adapted from
    # digitized images of a hardcopy from the MIT Museum. The digitization
    # was performed by Paul Fjeld, and arranged for by Deborah Douglas of
    # the Museum. Many thanks to both. The images (with suitable reduction
    # in storage size and consequent reduction in image quality as well) are
    # available online at www.ibiblio.org/apollo. If for some reason you
    # find that the images are illegible, contact me at info@sandroid.org
    # about getting access to the (much) higher-quality images which Paul
    # actually created.
    #
    # Notations on the hardcopy document read, in part:
    #
    #       Assemble revision 001 of AGC program LMY99 by NASA 2021112-61
    #       16:27 JULY 14, 1969
    #
    # Page 54
    <Page LM0054 25>
    # Page 55
    <Page LM0055 26>
    # Page 56
    <Page LM0056 27>
    # Page 57
    <Page LM0057 28>
    # Page 58
    <Page LM0058 29>
    # Page 59
    <Page LM0059 30a>
    # Page 60
    <Page LM0060 30b>

```

This code is written to file `src/Luminary099/INPUT-OUTPUT-CHANNEL-BIT-DESCRIPTIONS.agc`.

## 2.17 INTERPRETIVE CONSTANT

```

740  <src/Luminary099/INTERPRETIVE-CONSTANT.agc 740>≡
      # Copyright:   Public domain.
      # Filename:    INTERPRETIVE_CONSTANT.agc
      # Purpose:     Part of the source code for Luminary 1A build 099.
      #              It is part of the source code for the Lunar Module's (LM)
      #              Apollo Guidance Computer (AGC), for Apollo 11.
      # Assembler:   yaYUL
      # Contact:      Ron Burkey <info@sandroid.org>.
      # Website:      www.ibiblio.org/apollo.
      # Pages:        1100-1101
      # Mod history:  2009-05-25 RSB   Adapted from the corresponding
      #              Luminary131 file, using page
      #              images from Luminary 1A.
      #
      # This source code has been transcribed or otherwise adapted from
      # digitized images of a hardcopy from the MIT Museum. The digitization
      # was performed by Paul Fjeld, and arranged for by Deborah Douglas of
      # the Museum. Many thanks to both. The images (with suitable reduction
      # in storage size and consequent reduction in image quality as well) are
      # available online at www.ibiblio.org/apollo. If for some reason you
      # find that the images are illegible, contact me at info@sandroid.org
      # about getting access to the (much) higher-quality images which Paul
      # actually created.
      #
      # Notations on the hardcopy document read, in part:
      #
      #       Assemble revision 001 of AGC program LMY99 by NASA 2021112-61
      #       16:27 JULY 14, 1969
      #
      # Page 1100
      <Page LM1100 508>
      # Page 1101
      <Page LM1101 509a>

```

This code is written to file `src/Luminary099/INTERPRETIVE-CONSTANT.agc`.

## 2.18 KALMAN FILTER

```

741  <src/Luminary099/KALMAN-FILTER.agc 741>≡
      # Copyright:   Public domain.
      # Filename:    KALMAN_FILTER.agc
      # Purpose:     Part of the source code for Luminary 1A build 099.
      #              It is part of the source code for the Lunar Module's (LM)
      #              Apollo Guidance Computer (AGC), for Apollo 11.
      # Assembler:  yaYUL
      # Contact:     Ron Burkey <info@sandroid.org>.
      # Website:     www.ibiblio.org/apollo.
      # Pages:       1470-1471
      # Mod history: 2009-05-27 RSB   Adapted from the corresponding
      #              Luminary131 file, using page
      #              images from Luminary 1A.
      #
      # This source code has been transcribed or otherwise adapted from
      # digitized images of a hardcopy from the MIT Museum. The digitization
      # was performed by Paul Fjeld, and arranged for by Deborah Douglas of
      # the Museum. Many thanks to both. The images (with suitable reduction
      # in storage size and consequent reduction in image quality as well) are
      # available online at www.ibiblio.org/apollo. If for some reason you
      # find that the images are illegible, contact me at info@sandroid.org
      # about getting access to the (much) higher-quality images which Paul
      # actually created.
      #
      # Notations on the hardcopy document read, in part:
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      #       Assemble revision 001 of AGC program LMY99 by NASA 2021112-61
      #       16:27 JULY 14, 1969
      #
      # Page 1470
      <Page LM1470 646>
      # Page 1471
      <Page LM1471 647a>

```

This code is written to file `src/Luminary099/KALMAN-FILTER.agc`.

## 2.19 LAMBERT AIMPOINT GUIDANCE

```

742  <src/Luminary099/LAMBERT-AIMPOINT-GUIDANCE.agc 742>≡
      # Copyright:   Public domain.
      # Filename:    LAMBERT_AIMPOINT_GUIDANCE.agc
      # Purpose:     Part of the source code for Luminary 1A build 099.
      #              It is part of the source code for the Lunar Module's (LM)
      #              Apollo Guidance Computer (AGC), for Apollo 11.
      # Assembler:   yaYUL
      # Contact:      Ron Burkey <info@sandroid.org>.
      # Website:      www.ibiblio.org/apollo.
      # Pages:        651-653
      # Mod history:  2009-05-18 RSB   Transcribed from Luminary 099
      #                  page images.
      #                2009-06-05 RSB   Corrected 4 typos.
      #                2009-06-07 RSB   Fixed a typo.
      #
      # This source code has been transcribed or otherwise adapted from
      # digitized images of a hardcopy from the MIT Museum. The digitization
      # was performed by Paul Fjeld, and arranged for by Deborah Douglas of
      # the Museum. Many thanks to both. The images (with suitable reduction
      # in storage size and consequent reduction in image quality as well) are
      # available online at www.ibiblio.org/apollo. If for some reason you
      # find that the images are illegible, contact me at info@sandroid.org
      # about getting access to the (much) higher-quality images which Paul
      # actually created.
      #
      # Notations on the hardcopy document read, in part:
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      # Assemble revision 001 of AGC program LMY99 by NASA 2021112-61
      # 16:27 JULY 14, 1969
      #
      # Page 651
      <Page LM0651 237>
      # Page 652
      <Page LM0652 239>
      # Page 653
      <Page LM0653 240>

```

This code is written to file `src/Luminary099/LAMBERT-AIMPOINT-GUIDANCE.agc`.

## 2.20 LANDING ANALOG DISPLAYS

743 *<src/Luminary099/LANDING-ANALOG-DISPLAYS.agc 743>≡*

```
# Copyright:    Public domain.
# Filename:     LANDING_ANALOG_DISPLAYS.agc
# Purpose:      Part of the source code for Luminary, build 099. It
#               is part of the source code for the Lunar Module's
#               (LM) Apollo Guidance Computer (AGC), Apollo 11.
# Assembler:   yaYUL
# Reference:    pp. 898-907
# Contact:      Ron Burkey <info@sandroid.org>,
#               Fabrizio Bernardini <fabrizio@spacecraft.it>
# Website:      http://www.ibiblio.org/apollo.
# Mod history:  05/06/09 FB      Transcription Batch 4 Assignment.
#
# The contents of the "Luminary099" files, in general, are transcribed
# from scanned documents.
#
# Assemble revision 001 of AGC program Luminary099 by NASA
# 2021112-061.  July 14, 1969.
#
# Prepared by
#
#               Massachusetts Institute of Technology
#               75 Cambridge Parkway
#               Cambridge, Massachusetts
#
# under NASA contract NAS 9-4065.
#
# Refer directly to the online document mentioned above for further
# information. Please report any errors to info@sandroid.org.

# Page 898
<Page LM0898 453>
# Page 899
<Page LM0899 455>
# Page 900
<Page LM0900 457>
# Page 901
<Page LM0901 459>
# Page 902
<Page LM0902 461>
# Page 903
<Page LM0903 463>
# Page 904
<Page LM0904 465>
# Page 905
```

*⟨Page LM0905 467⟩*

**# Page 906**

*⟨Page LM0906 469⟩*

**# Page 907**

*⟨Page LM0907 471⟩*

This code is written to file `src/Luminary099/LANDING-ANALOG-DISPLAYS.agc`.



## 2.21 LEM GEOMETRY

```

745  <src/Luminary099/LEM-GEOMETRY.agc 745>≡
# Copyright:    Public domain.
# Filename:     LEM-GEOMETRY.agc
# Purpose:      Part of the source code for Luminary 1A build 099.
#              It is part of the source code for the Lunar Module's (LM)
#              Apollo Guidance Computer (AGC), for Apollo 11.
# Assembler:   yaYUL
# Contact:      Ron Burkey <info@sandroid.org>.
# Website:      www.ibiblio.org/apollo.
# Pages:        320-325
# Mod history:  2009-05-16 RSB   Adapted from the corresponding
#              Luminary131 file, using page
#              images from Luminary 1A.
#
# This source code has been transcribed or otherwise adapted from
# digitized images of a hardcopy from the MIT Museum. The digitization
# was performed by Paul Fjeld, and arranged for by Deborah Douglas of
# the Museum. Many thanks to both. The images (with suitable reduction
# in storage size and consequent reduction in image quality as well) are
# available online at www.ibiblio.org/apollo. If for some reason you
# find that the images are illegible, contact me at info@sandroid.org
# about getting access to the (much) higher-quality images which Paul
# actually created.
#
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#       16:27 JULY 14, 1969
#
# Page 320
# <page LM0320 102>
# Page 321
# <page LM0321 103>
# Page 322
# <page LM0322 104>
# Page 323
# <page LM0323 105>
# Page 324
# <page LM0324 106>
# Page 325
# <page LM0325 107a>

```

This code is written to file `src/Luminary099/LEM-GEOMETRY.agc`.

## 2.22 LUNAR LANDING GUIDANCE EQUATIONS

```

746  <src/Luminary099/LUNAR-LANDING-GUIDANCE-EQUATIONS.agc 746>≡
      # Copyright:    Public domain.
      # Filename:     LUNAR_LANDING_GUIDANCE_EQUATIONS.agc
      # Purpose:      Part of the source code for Luminary 1A build 099.
      #               It is part of the source code for the Lunar Module's (LM)
      #               Apollo Guidance Computer (AGC), for Apollo 11.
      # Assembler:    yaYUL
      # Contact:       HARTMUTH GUTSCHE <hgutsche@explornet.com>.
      # Website:       www.ibiblio.org/apollo.
      # Pages:         798-828
      # Mod history:   2009-05-23 HG   Transcribed from page images.
      #               2009-06-05 RSB   Fixed a goofy thing that was apparently
      #                               legal in GAP but not in yaYUL.  Eliminated
      #                               a couple of lines of code that shouldn't
      #                               have survived from Luminary 131 to here.
      #               2009-06-07 RSB   Fixed a typo.
      #
      # This source code has been transcribed or otherwise adapted from
      # digitized images of a hardcopy from the MIT Museum.  The digitization
      # was performed by Paul Fjeld, and arranged for by Deborah Douglas of
      # the Museum.  Many thanks to both.  The images (with suitable reduction
      # in storage size and consequent reduction in image quality as well) are
      # available online at www.ibiblio.org/apollo.  If for some reason you
      # find that the images are illegible, contact me at info@sandroid.org
      # about getting access to the (much) higher-quality images which Paul
      # actually created.
      #
      # Notations on the hardcopy document read, in part:
      #
      #       Assemble revision 001 of AGC program LMY99 by NASA 2021112-61
      #       16:27 JULY 14, 1969

      # Page 798
      <Page LM0798 297>
      # Page 799
      <Page LM0799 298>
      # Page 800
      <Page LM0800 300>
      # Page 801
      <Page LM0801 302>
      # Page 802
      <Page LM0802 304>

```

# Page 803  
⟨Page LM0803 306⟩  
# Page 804  
⟨Page LM0804 307⟩  
# Page 805  
⟨Page LM0805 309⟩  
# Page 806 actually starts one line earlier but that would separate the markers from their vari  
⟨Page LM0806 311⟩  
# Page 807  
⟨Page LM0807 313⟩  
# Page 808  
⟨Page LM0808 315⟩  
# Page 809  
⟨Page LM0809 317⟩  
# Page 810  
⟨Page LM0810 319⟩  
# Page 811  
⟨Page LM0811 321⟩  
# Page 812  
⟨Page LM0812 323⟩  
# Page 813  
⟨Page LM0813 325⟩  
# Page 814  
⟨Page LM0814 327⟩  
# Page 815  
⟨Page LM0815 328⟩  
# Page 816  
⟨Page LM0816 329⟩  
# Page 817  
⟨Page LM0817 331⟩  
# Page 818  
⟨Page LM0818 333⟩  
# Page 819  
⟨Page LM0819 335⟩  
# Page 820  
⟨Page LM0820 337⟩  
# Page 821  
⟨Page LM0821 339⟩  
# Page 822  
⟨Page LM0822 340⟩  
# Page 823  
⟨Page LM0823 341⟩  
# Page 824  
⟨Page LM0824 342⟩  
# Page 825  
⟨Page LM0825 344⟩

# Page 826

*⟨Page LM0826 346⟩*

# Page 827

*⟨Page LM0827 347a⟩*

# Page 828

*⟨Page LM0828 347b⟩*

This code is written to file `src/Luminary099/LUNAR-LANDING-GUIDANCE-EQUATIONS.agc`.

## 2.23 P12

```

749  <src/Luminary099/P12.agc 749>≡
      # Copyright:   Public domain.
      # Filename:    P12.agc
      # Purpose:     Part of the source code for Luminary 1A build 099.
      #              It is part of the source code for the Lunar Module's (LM)
      #              Apollo Guidance Computer (AGC), for Apollo 11.
      # Assembler:   yaYUL
      # Contact:      Hartmuth Gutsche <hgutsche@explornet.com>.
      # Website:      www.ibiblio.org/apollo.
      # Pages:        838-842
      # Mod history:  2009-05-23 HG   Transcribed from page images.
      #
      # This source code has been transcribed or otherwise adapted from
      # digitized images of a hardcopy from the MIT Museum. The digitization
      # was performed by Paul Fjeld, and arranged for by Deborah Douglas of
      # the Museum. Many thanks to both. The images (with suitable reduction
      # in storage size and consequent reduction in image quality as well) are
      # available online at www.ibiblio.org/apollo. If for some reason you
      # find that the images are illegible, contact me at info@sandroid.org
      # about getting access to the (much) higher-quality images which Paul
      # actually created.
      #
      # Notations on the hardcopy document read, in part:
      #
      #      Assemble revision 001 of AGC program LMY99 by NASA 2021112-61
      #      16:27 JULY 14, 1969
      #
      # Page 838
      <Page LM0838 365>
      # Page 839
      <Page LM0839 367>
      # Page 840
      <Page LM0840 369>
      # Page 841
      <Page LM0841 371>
      # Page 842
      <Page LM0842 372a>

```

This code is written to file `src/Luminary099/P12.agc`.

## 2.24 P30 P37

```

750  <src/Luminary099/P30-P37.agc 750>≡
      # Copyright:   Public domain.
      # Filename:    P30_P37.agc
      # Purpose:     Part of the source code for Luminary 1A build 099.
      #              It is part of the source code for the Lunar Module's (LM)
      #              Apollo Guidance Computer (AGC), for Apollo 11.
      # Assembler:   yaYUL
      # Contact:      Ron Burkey <info@sandroid.org>.
      # Website:      www.ibiblio.org/apollo.
      # Pages:        614-617
      # Mod history:  2009-05-17 RSB   Adapted from the corresponding
      #              Luminary131 file, using page
      #              images from Luminary 1A.
      #              2009-06-05 RSB   Removed 4 lines of code that shouldn't
      #              have survived from Luminary 131.
      #
      # This source code has been transcribed or otherwise adapted from
      # digitized images of a hardcopy from the MIT Museum. The digitization
      # was performed by Paul Fjeld, and arranged for by Deborah Douglas of
      # the Museum. Many thanks to both. The images (with suitable reduction
      # in storage size and consequent reduction in image quality as well) are
      # available online at www.ibiblio.org/apollo. If for some reason you
      # find that the images are illegible, contact me at info@sandroid.org
      # about getting access to the (much) higher-quality images which Paul
      # actually created.
      #
      # Notations on the hardcopy document read, in part:
      #
      #      Assemble revision 001 of AGC program LMY99 by NASA 2021112-61
      #      16:27 JULY 14, 1969
      #
      # Page 614
      <Page LM0614 182>
      # Page 615
      <Page LM0615 183>
      # Page 616
      <Page LM0616 184>
      # Page 617
      <Page LM0617 186>

```

This code is written to file `src/Luminary099/P30--P37.agc`.

## 2.25 P32-P35 P72-P75

```

751  <src/Luminary099/P32-P35-P72-P75.agc 751>≡
      # Copyright:   Public domain.
      # Filename:    P32-P35_P72-P75.agc
      # Purpose:     Part of the source code for Luminary 1A build 099.
      #              It is part of the source code for the Lunar Module's (LM)
      #              Apollo Guidance Computer (AGC), for Apollo 11.
      # Assembler:   yaYUL
      # Contact:      Ron Burkey <info@sandroid.org>.
      # Website:      www.ibiblio.org/apollo.
      # Pages:        618-650
      # Mod history:  2009-05-18 RSB   Adapted from the Luminary 131 file of the
      #              same name, as corrected from Luminary 099
      #              page images.
      #
      # This source code has been transcribed or otherwise adapted from
      # digitized images of a hardcopy from the MIT Museum. The digitization
      # was performed by Paul Fjeld, and arranged for by Deborah Douglas of
      # the Museum. Many thanks to both. The images (with suitable reduction
      # in storage size and consequent reduction in image quality as well) are
      # available online at www.ibiblio.org/apollo. If for some reason you
      # find that the images are illegible, contact me at info@sandroid.org
      # about getting access to the (much) higher-quality images which Paul
      # actually created.
      #
      # Notations on the hardcopy document read, in part:
      #
      #       Assemble revision 001 of AGC program LMY99 by NASA 2021112-61
      #       16:27 JULY 14, 1969

      # Page 618
      <Page LM0618 188>
      # Page 619
      <Page LM0619 190>
      # Page 620
      <Page LM0620 192>
      # Page 621
      <Page LM0621 194>
      # Page 622
      <Page LM0622 196>
      # Page 623
      <Page LM0623 197>
      # Page 624
      <Page LM0624 198>
      # Page 625

```

*⟨Page LM0625 200⟩*  
# Page 626  
*⟨Page LM0626 202⟩*  
# Page 627  
*⟨Page LM0627 204⟩*  
# Page 628  
*⟨Page LM0628 206⟩*  
# Page 629  
*⟨Page LM0629 207⟩*  
# Page 630  
*⟨Page LM0630 208⟩*  
# Page 631  
*⟨Page LM0631 209⟩*  
# Page 632  
*⟨Page LM0632 210⟩*  
# Page 633  
*⟨Page LM0633 212⟩*  
# Page 634  
*⟨Page LM0634 214⟩*  
# Page 635  
*⟨Page LM0635 216⟩*  
# Page 636  
*⟨Page LM0636 218⟩*  
# Page 637  
*⟨Page LM0637 220⟩*  
# Page 638  
*⟨Page LM0638 222⟩*  
# Page 639  
*⟨Page LM0639 224⟩*  
# Page 640  
*⟨Page LM0640 226⟩*  
# Page 641  
*⟨Page LM0641 228⟩*  
# Page 642  
*⟨Page LM0642 229⟩*  
# Page 643  
*⟨Page LM0643 230a⟩*  
# Page 644  
*⟨Page LM0644 230b⟩*  
# Page 645  
*⟨Page LM0645 231a⟩*  
# Page 646  
*⟨Page LM0646 231b⟩*  
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**# Page 649**

*⟨Page LM0649 235a⟩*

**# Page 650**

*⟨Page LM0650 235b⟩*

This code is written to file `src/Luminary099/P32-P35-P72-P75.agc`.

## 2.26 P70-P71

```

754  <src/Luminary099/P70-P71.agc 754>≡
      # Copyright:   Public domain.
      # Filename:    P70-P71.agc
      # Purpose:     Part of the source code for Luminary 1A build 099.
      #              It is part of the source code for the Lunar Module's (LM)
      #              Apollo Guidance Computer (AGC), for Apollo 11.
      # Assembler:   yaYUL
      # Contact:      Hartmuth Gutsche <hgutsche@explornet.com>.
      # Website:      www.ibiblio.org/apollo.
      # Pages:        829-837
      # Mod history:  2009-05-23 HG   Transcribed from page images.
      #              2009-06-05 RSB   Fixed a typo.
      #
      # This source code has been transcribed or otherwise adapted from
      # digitized images of a hardcopy from the MIT Museum. The digitization
      # was performed by Paul Fjeld, and arranged for by Deborah Douglas of
      # the Museum. Many thanks to both. The images (with suitable reduction
      # in storage size and consequent reduction in image quality as well) are
      # available online at www.ibiblio.org/apollo. If for some reason you
      # find that the images are illegible, contact me at info@sandroid.org
      # about getting access to the (much) higher-quality images which Paul
      # actually created.
      #
      # Notations on the hardcopy document read, in part:
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      #      Assemble revision 001 of AGC program LMY99 by NASA 2021112-61
      #      16:27 JULY 14, 1969
      #
      # Page 829
      <Page LM0829 349>
      # Page 830
      <Page LM0830 351>
      # Page 831
      <Page LM0831 353>
      # Page 832
      <Page LM0832 355>
      # Page 833
      <Page LM0833 357>
      # Page 834
      <Page LM0834 359>
      # Page 835
      <Page LM0835 361>
      # Page 836
      <Page LM0836 363>

```

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# Page 837

*⟨Page LM0837 364a⟩*

This code is written to file `src/Luminary099/P70-P71.agc`.

## 2.27 P-AXIS RCS AUTOPILOT

```

756  <src/Luminary099/P-AXIS-RCS-AUTOPILOT.agc 756>≡
      # Copyright:   Public domain.
      # Filename:    P-AXIS_RCS_AUTOPILOT.agc
      # Purpose:     Part of the source code for Luminary 1A build 099.
      #              It is part of the source code for the Lunar Module's (LM)
      #              Apollo Guidance Computer (AGC), for Apollo 11.
      # Assembler:   yaYUL
      # Contact:      Ron Burkey <info@sandroid.org>.
      # Website:      www.ibiblio.org/apollo.
      # Pages:        1421-1441
      # Mod history:  2009-05-27 RSB   Adapted from the corresponding
      #              Luminary131 file, using page
      #              images from Luminary 1A.
      #              2009-06-05 RSB   Corrected a relative jump from
      #              +8 to +8D.
      #              2009-06-07 RSB   Corrected a typo.
      #
      # This source code has been transcribed or otherwise adapted from
      # digitized images of a hardcopy from the MIT Museum. The digitization
      # was performed by Paul Fjeld, and arranged for by Deborah Douglas of
      # the Museum. Many thanks to both. The images (with suitable reduction
      # in storage size and consequent reduction in image quality as well) are
      # available online at www.ibiblio.org/apollo. If for some reason you
      # find that the images are illegible, contact me at info@sandroid.org
      # about getting access to the (much) higher-quality images which Paul
      # actually created.
      #
      # Notations on the hardcopy document read, in part:
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      #       16:27 JULY 14, 1969
      #
      # Page 1421
      <Page LM1421 558>
      # Page 1422
      <Page LM1422 560>
      # Page 1423
      <Page LM1423 562>
      # Page 1424
      <Page LM1424 564>
      # Page 1425
      <Page LM1425 566>
      # Page 1426
      <Page LM1426 568>

```

# Page 1427  
⟨Page LM1427 570⟩  
# Page 1428  
⟨Page LM1428 572⟩  
# Page 1429  
⟨Page LM1429 574⟩  
# Page 1430  
⟨Page LM1430 576⟩  
# Page 1431  
⟨Page LM1431 578⟩  
# Page 1432  
⟨Page LM1432 580⟩  
# Page 1433  
⟨Page LM1433 582⟩  
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# Page 1435  
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# Page 1438  
⟨Page LM1438 592⟩  
# Page 1439  
⟨Page LM1439 593⟩  
# Page 1440  
⟨Page LM1440 594⟩  
# Page 1441  
⟨Page LM1441 595⟩

This code is written to file `src/Luminary099/P-AXIS-RCS-AUTOPILOT.agc`.

## 2.28 Q R-AXIS RCS AUTOPILOT

```

758  <src/Luminary099/Q-R-AXIS-RCS-AUTOPILOT.agc 758>≡
      # Copyright:   Public domain.
      # Filename:    Q_R-AXIS-RCS-AUTOPILOT.agc
      # Purpose:     Part of the source code for Luminary 1A build 099.
      #              It is part of the source code for the Lunar Module's (LM)
      #              Apollo Guidance Computer (AGC), for Apollo 11.
      # Assembler:   yaYUL
      # Contact:      Ron Burkey <info@sandroid.org>.
      # Website:      www.ibiblio.org/apollo.
      # Pages:        1442-1459
      # Mod history:  2009-05-27 RSB   Adapted from the corresponding
      #              Luminary131 file, using page
      #              images from Luminary 1A.
      #              2009-06-07 RSB   Corrected "DEC 96.0" to "DEC 96", since
      #              the former is not compatible with yaYUL.
      #
      # This source code has been transcribed or otherwise adapted from
      # digitized images of a hardcopy from the MIT Museum. The digitization
      # was performed by Paul Fjeld, and arranged for by Deborah Douglas of
      # the Museum. Many thanks to both. The images (with suitable reduction
      # in storage size and consequent reduction in image quality as well) are
      # available online at www.ibiblio.org/apollo. If for some reason you
      # find that the images are illegible, contact me at info@sandroid.org
      # about getting access to the (much) higher-quality images which Paul
      # actually created.
      #
      # Notations on the hardcopy document read, in part:
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      # Assemble revision 001 of AGC program LMY99 by NASA 2021112-61
      # 16:27 JULY 14, 1969
      #
      # Page 1442
      <Page LM1442 597>
      # Page 1443
      <Page LM1443 599>
      # Page 1444
      <Page LM1444 601>
      # Page 1445
      <Page LM1445 603>
      # Page 1446
      <Page LM1446 605>
      # Page 1447
      <Page LM1447 607>
      # Page 1448

```

*⟨Page LM1448 609⟩*  
# Page 1449  
*⟨Page LM1449 611⟩*  
# Page 1450  
*⟨Page LM1450 612⟩*  
# Page 1451  
*⟨Page LM1451 613⟩*  
# Page 1452  
*⟨Page LM1452 614⟩*  
# Page 1453  
*⟨Page LM1453 616⟩*  
# Page 1454  
*⟨Page LM1454 618⟩*  
# Page 1455  
*⟨Page LM1455 620⟩*  
# Page 1456  
*⟨Page LM1456 622⟩*  
# Page 1457  
*⟨Page LM1457 624⟩*  
# Page 1458  
*⟨Page LM1458 626⟩*  
# Page 1459  
*⟨Page LM1459 627a⟩*

This code is written to file `src/Luminary099/Q-R-AXIS-RCS-AUTOPILOT.agc`.

## 2.29 R63

```

760  <src/Luminary099/R63.agc 760>≡
      # Copyright:   Public domain.
      # Filename:    R63.agc
      # Purpose:     Part of the source code for Luminary 1A build 099.
      #              It is part of the source code for the Lunar Module's (LM)
      #              Apollo Guidance Computer (AGC), for Apollo 11.
      # Assembler:   yaYUL
      # Contact:      Ron Burkey <info@sandroid.org>.
      # Website:      www.ibiblio.org/apollo.
      # Pages:        338-341
      # Mod history:  2009-05-16 RSB   Adapted from the corresponding
      #              Luminary131 file, using page
      #              images from Luminary 1A.
      #
      # This source code has been transcribed or otherwise adapted from
      # digitized images of a hardcopy from the MIT Museum. The digitization
      # was performed by Paul Fjeld, and arranged for by Deborah Douglas of
      # the Museum. Many thanks to both. The images (with suitable reduction
      # in storage size and consequent reduction in image quality as well) are
      # available online at www.ibiblio.org/apollo. If for some reason you
      # find that the images are illegible, contact me at info@sandroid.org
      # about getting access to the (much) higher-quality images which Paul
      # actually created.
      #
      # Notations on the hardcopy document read, in part:
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      #      Assemble revision 001 of AGC program LMY99 by NASA 2021112-61
      #      16:27 JULY 14, 1969
      #
      # Page 338
      # <Page LM0338 108>
      # Page 339
      # <Page LM0339 110>
      # Page 340
      # <Page LM0340 111a>
      # Page 341
      # <Page LM0341 111b>

```

This code is written to file `src/Luminary099/R63.agc`.



## 2.30 RADAR LEADIN ROUTINES

```

761  <src/Luminary099/RADAR-LEADIN-ROUTINES.agc 761>≡
      # Copyright:   Public domain.
      # Filename:    RADAR_LEADIN_ROUTINES.agc
      # Purpose:    Part of the source code for Luminary 1A build 099.
      #             It is part of the source code for the Lunar Module's (LM)
      #             Apollo Guidance Computer (AGC), for Apollo 11.
      # Assembler:  yaYUL
      # Contact:    Ron Burkey <info@sandroid.org>.
      # Website:    www.ibiblio.org/apollo.
      # Pages:      490-491
      # Mod history: 2009-05-17 RSB   Adapted from the corresponding
      #             Luminary131 file, using page
      #             images from Luminary 1A.
      #
      # This source code has been transcribed or otherwise adapted from
      # digitized images of a hardcopy from the MIT Museum. The digitization
      # was performed by Paul Fjeld, and arranged for by Deborah Douglas of
      # the Museum. Many thanks to both. The images (with suitable reduction
      # in storage size and consequent reduction in image quality as well) are
      # available online at www.ibiblio.org/apollo. If for some reason you
      # find that the images are illegible, contact me at info@sandroid.org
      # about getting access to the (much) higher-quality images which Paul
      # actually created.
      #
      # Notations on the hardcopy document read, in part:
      #
      # Assemble revision 001 of AGC program LMY99 by NASA 2021112-61
      # 16:27 JULY 14, 1969
      #
      # Page 490
      <Page LM0490 180>
      # Page 491
      <Page LM0491 181a>

```

This code is written to file `src/Luminary099/RADAR-LEADIN-ROUTINES.agc`.

## 2.31 RCS FAILURE MONITOR

```

762  <src/Luminary099/RCS-FAILURE-MONITOR.agc 762>≡
      # Copyright:   Public domain.
      # Filename:    RCS_FAILURE_MONITOR.agc
      # Purpose:     Part of the source code for Luminary 1A build 099.
      #              It is part of the source code for the Lunar Module's (LM)
      #              Apollo Guidance Computer (AGC), for Apollo 11.
      # Assembler:   yaYUL
      # Contact:      Hartmuth Gutsche <hgutsche@explornet.com>.
      # Website:      www.ibiblio.org/apollo.
      # Pages:        190-192
      # Mod history:  2009-05-19 HG   Transcribed from page images.
      #
      # This source code has been transcribed or otherwise adapted from
      # digitized images of a hardcopy from the MIT Museum. The digitization
      # was performed by Paul Fjeld, and arranged for by Deborah Douglas of
      # the Museum. Many thanks to both. The images (with suitable reduction
      # in storage size and consequent reduction in image quality as well) are
      # available online at www.ibiblio.org/apollo. If for some reason you
      # find that the images are illegible, contact me at info@sandroid.org
      # about getting access to the (much) higher-quality images which Paul
      # actually created.
      #
      # Notations on the hardcopy document read, in part:
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      #      Assemble revision 001 of AGC program LMY99 by NASA 2021112-61
      #      16:27 JULY 14, 1969
      #
      # Page 190
      <Page LM0190 65>
      # Page 191
      <Page LM0191 67>
      # Page 192
      <Page LM0192 69>

```

This code is written to file `src/Luminary099/RCS-FAILURE-MONITOR.agc`.

## 2.32 RTB OP CODES

```

763  <src/Luminary099/RTB-OP-CODES.agc 763>≡
      # Copyright:   Public domain.
      # Filename:    RTB_OP_CODES.agc
      # Purpose:     Part of the source code for Luminary 1A build 099.
      #              It is part of the source code for the Lunar Module's (LM)
      #              Apollo Guidance Computer (AGC), for Apollo 11.
      # Assembler:   yaYUL
      # Contact:      Ron Burkey <info@sandroid.org>.
      # Website:      www.ibiblio.org/apollo.
      # Pages:        1397-1401
      # Mod history:  2009-05-10 SN    (Sergio Navarro).  Started adapting
      #              from the Luminary131/ file of the same
      #              name, using Luminary099 page images.
      #
      # This source code has been transcribed or otherwise adapted from
      # digitized images of a hardcopy from the MIT Museum.  The digitization
      # was performed by Paul Fjeld, and arranged for by Deborah Douglas of
      # the Museum.  Many thanks to both.  The images (with suitable reduction
      # in storage size and consequent reduction in image quality as well) are
      # available online at www.ibiblio.org/apollo.  If for some reason you
      # find that the images are illegible, contact me at info@sandroid.org
      # about getting access to the (much) higher-quality images which Paul
      # actually created.
      #
      # Notations on the hardcopy document read, in part:
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      #      16:27 JULY 14, 1969

      # Page 1397
      <Page LM1397 528>
      # Page 1398
      <Page LM1398 530>
      # Page 1399
      <Page LM1399 531a>
      # Page 1400
      <Page LM1400 531b>
      # Page 1401
      <Page LM1401 532>
      # Page 1402
      <Page LM1402 533a>

```

This code is written to file `src/Luminary099/RTB-OP-CODES.agc`.

## 2.33 S-BAND ANTENNA FOR LM

```

764  <src/Luminary099/S-BAND-ANTENNA-FOR-LM.agc 764>≡
      # Copyright:   Public domain.
      # Filename:    S-BAND_ANTENNA_FOR_LM.agc
      # Purpose:     Part of the source code for Luminary 1A build 099.
      #              It is part of the source code for the Lunar Module's (LM)
      #              Apollo Guidance Computer (AGC), for Apollo 11.
      # Assembler:   yaYUL
      # Contact:      Ron Burkey <info@sandroid.org>.
      # Website:      www.ibiblio.org/apollo.
      # Pages:        486-489
      # Mod history:  2009-05-17 RSB   Adapted from the corresponding
      #              Luminary131 file, using page
      #              images from Luminary 1A.
      #              2009-06-07 RSB   Corrected a misprint.
      #
      # This source code has been transcribed or otherwise adapted from
      # digitized images of a hardcopy from the MIT Museum. The digitization
      # was performed by Paul Fjeld, and arranged for by Deborah Douglas of
      # the Museum. Many thanks to both. The images (with suitable reduction
      # in storage size and consequent reduction in image quality as well) are
      # available online at www.ibiblio.org/apollo. If for some reason you
      # find that the images are illegible, contact me at info@sandroid.org
      # about getting access to the (much) higher-quality images which Paul
      # actually created.
      #
      # Notations on the hardcopy document read, in part:
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      # Assemble revision 001 of AGC program LMY99 by NASA 2021112-61
      # 16:27 JULY 14, 1969
      #
      # Page 486
      <Page LM0486 174>
      # Page 487
      <Page LM0487 176>
      # Page 488
      <Page LM0488 178>
      # Page 489
      <Page LM0489 179a>

```

This code is written to file `src/Luminary099/S-BAND-ANTENNA-FOR-LM.agc`.

## 2.34 SERVICER

```
765  <src/Luminary099/SERVICER.agc 765>≡
# Copyright:    Public domain.
# Filename:     SERVICER.agc
# Purpose:      Part of the source code for Luminary, build 099. It
#               is part of the source code for the Lunar Module's
#               (LM) Apollo Guidance Computer (AGC), Apollo 11.
# Assembler:   yaYUL
# Reference:    pp. 857-897
# Contact:      Ron Burkey <info@sandroid.org>,
#               Fabrizio Bernardini <fabrizio@spacecraft.it>
# Website:      http://www.ibiblio.org/apollo.
# Mod history:  2009-06-01 FB   Transcription Batch 4 Assignment.
#               2009-06-05 RSB   Fixed a couple of typos, plus a goofy relative
#                               label reference from the original source.
#
# The contents of the "Luminary099" files, in general, are transcribed
# from scanned documents.
#
#       Assemble revision 001 of AGC program Luminary099 by NASA
#       2021112-061.   July 14, 1969.
#
#       Prepared by
#
#               Massachusetts Institute of Technology
#               75 Cambridge Parkway
#               Cambridge, Massachusetts
#
#       under NASA contract NAS 9-4065.
#
# Refer directly to the online document mentioned above for further
# information. Please report any errors to info@sandroid.org.

# Page 857
<Page LM0857 397>
# Page 858
<Page LM0858 398>
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*⟨Page LM0896 451⟩*  
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*⟨Page LM0897 452a⟩*

This code is written to file `src/Luminary099/SERVICER.agc`.

## 2.35 SPS BACK-UP RCS CONTROL

```

768  <src/Luminary099/SPS-BACK-UP-RCS-CONTROL.agc 768>≡
      # Copyright:   Public domain.
      # Filename:    SPS_BACK-UP_RCS_CONTROL.agc
      # Purpose:     Part of the source code for Luminary 1A build 099.
      #              It is part of the source code for the Lunar Module's (LM)
      #              Apollo Guidance Computer (AGC), for Apollo 11.
      # Assembler:   yaYUL
      # Contact:      Ron Burkey <info@sandroid.org>.
      # Website:      www.ibiblio.org/apollo.
      # Pages:        1507-1510
      # Mod history:  2009-05-27 RSB   Adapted from the corresponding
      #              Luminary131 file, using page
      #              images from Luminary 1A.
      #
      # This source code has been transcribed or otherwise adapted from
      # digitized images of a hardcopy from the MIT Museum. The digitization
      # was performed by Paul Fjeld, and arranged for by Deborah Douglas of
      # the Museum. Many thanks to both. The images (with suitable reduction
      # in storage size and consequent reduction in image quality as well) are
      # available online at www.ibiblio.org/apollo. If for some reason you
      # find that the images are illegible, contact me at info@sandroid.org
      # about getting access to the (much) higher-quality images which Paul
      # actually created.
      #
      # Notations on the hardcopy document read, in part:
      #
      #       Assemble revision 001 of AGC program LMY99 by NASA 2021112-61
      #       16:27 JULY 14, 1969
      #
      # Page 1507
      <Page LM1507 706>
      # Page 1508
      <Page LM1508 707>
      # Page 1509
      <Page LM1509 709>
      # Page 1510
      <Page LM1510 710>

```

This code is written to file `src/Luminary099/SPS-BACK-UP-RCS-CONTROL.agc`.



## 2.36 T6-RUPT PROGRAMS

```

769  <src/Luminary099/T6-RUPT-PROGRAMS.agc 769>≡
      # Copyright:   Public domain.
      # Filename:    T6-RUPT_PROGRAMS.agc
      # Purpose:     Part of the source code for Luminary 1A build 099.
      #              It is part of the source code for the Lunar Module's (LM)
      #              Apollo Guidance Computer (AGC), for Apollo 11.
      # Assembler:   yaYUL
      # Contact:      Ron Burkey <info@sandroid.org>.
      # Website:      www.ibiblio.org/apollo.
      # Pages:        1403-1405
      # Mod history:  2009-05-10 SN    (Sergio Navarro).  Started adapting
      #              from the Luminary131/ file of the same
      #              name, using Luminary099 page images.
      #
      # This source code has been transcribed or otherwise adapted from
      # digitized images of a hardcopy from the MIT Museum.  The digitization
      # was performed by Paul Fjeld, and arranged for by Deborah Douglas of
      # the Museum.  Many thanks to both.  The images (with suitable reduction
      # in storage size and consequent reduction in image quality as well) are
      # available online at www.ibiblio.org/apollo.  If for some reason you
      # find that the images are illegible, contact me at info@sandroid.org
      # about getting access to the (much) higher-quality images which Paul
      # actually created.
      #
      # Notations on the hardcopy document read, in part:
      #
      #      Assemble revision 001 of AGC program LMY99 by NASA 2021112-61
      #      16:27 JULY 14, 1969

      # Page 1403
      <Page LM1403 534>
      # Page 1404
      <Page LM1404 536>
      # Page 1405
      <Page LM1405 537>

```

This code is written to file `src/Luminary099/T6-RUPT-PROGRAMS.agc`.

## 2.37 THE LUNAR LANDING

770 *<src/Luminary099/THE-LUNAR-LANDING.agc 770>≡*

```
# Copyright:   Public domain.
# Filename:    THE_LUNAR_LANDING.agc
# Purpose:    Part of the source code for Luminary 1A build 099.
#             It is part of the source code for the Lunar Module's (LM)
#             Apollo Guidance Computer (AGC), for Apollo 11.
# Assembler:  yaYUL
# Contact:    Hartmuth Gutsche<hgutsche@explornet.com>.
# Website:    www.ibiblio.org/apollo.
# Pages:      785-792
# Mod history: 2009-05-20 HG   Transcribed from page images.
#
# This source code has been transcribed or otherwise adapted from
# digitized images of a hardcopy from the MIT Museum. The digitization
# was performed by Paul Fjeld, and arranged for by Deborah Douglas of
# the Museum. Many thanks to both. The images (with suitable reduction
# in storage size and consequent reduction in image quality as well) are
# available online at www.ibiblio.org/apollo. If for some reason you
# find that the images are illegible, contact me at info@sandroid.org
# about getting access to the (much) higher-quality images which Paul
# actually created.
#
# Notations on the hardcopy document read, in part:
#
#       Assemble revision 001 of AGC program LMY99 by NASA 2021112-61
#       16:27 JULY 14, 1969
```

# Page 785

*<Page LM0785 279>*

# Page 786

*<Page LM0786 281>*

# Page 787 new page is actually one line earlier but this would put the indices on a

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# Page 788

*<Page LM0788 285>*

# Page 789

*<Page LM0789 287a>*

# Page 790

*<Page LM0790 287b>*

# Page 791

*<Page LM0791 288>*

# Page 792

*<Page LM0792 289a>*

This code is written to file `src/Luminary099/THE-LUNAR-LANDING.agc`.

## 2.38 THROTTLE CONTROL ROUTINES

```

771  <src/Luminary099/THROTTLE-CONTROL-ROUTINES.agc 771>≡
      # Copyright:   Public domain.
      # Filename:    THROTTLE_CONTROL_ROUTINES.agc
      # Purpose:     Part of the source code for Luminary 1A build 099.
      #              It is part of the source code for the Lunar Module's (LM)
      #              Apollo Guidance Computer (AGC), for Apollo 11.
      # Assembler:   yaYUL
      # Contact:      HARTMUTH GUTSCHE <hgutsche@explornet.com>.
      # Website:      www.ibiblio.org/apollo.
      # Pages:        793-797
      # Mod history:  2009-05-20 HG   Transcribed from page images.
      #
      # This source code has been transcribed or otherwise adapted from
      # digitized images of a hardcopy from the MIT Museum. The digitization
      # was performed by Paul Fjeld, and arranged for by Deborah Douglas of
      # the Museum. Many thanks to both. The images (with suitable reduction
      # in storage size and consequent reduction in image quality as well) are
      # available online at www.ibiblio.org/apollo. If for some reason you
      # find that the images are illegible, contact me at info@sandroid.org
      # about getting access to the (much) higher-quality images which Paul
      # actually created.
      #
      # Notations on the hardcopy document read, in part:
      #
      #       Assemble revision 001 of AGC program LMY99 by NASA 2021112-61
      #       16:27 JULY 14, 1969
      #
      # Page 793
      <Page LM0793 290>
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      <Page LM0795 293>
      # Page 796
      <Page LM0796 295>
      # Page 797
      <Page LM0797 296a>

```

This code is written to file `src/Luminary099/THROTTLE-CONTROL-ROUTINES.agc`.

## 2.39 TJET LAW

```

772  <src/Luminary099/TJET-LAW.agc 772>≡
      # Copyright:   Public domain.
      # Filename:    TJET_LAW.agc
      # Purpose:     Part of the source code for Luminary 1A build 099.
      #              It is part of the source code for the Lunar Module's (LM)
      #              Apollo Guidance Computer (AGC), for Apollo 11.
      # Assembler:   yaYUL
      # Contact:      Ron Burkey <info@sandroid.org>.
      # Website:      www.ibiblio.org/apollo.
      # Pages:        1460-1469
      # Mod history:  2009-05-27 RSB   Adapted from the corresponding
      #              Luminary131 file, using page
      #              images from Luminary 1A.
      #              2009-06-06 RSB   Eliminated a stray instruction that had crept
      #              in somehow.
      #
      # This source code has been transcribed or otherwise adapted from
      # digitized images of a hardcopy from the MIT Museum. The digitization
      # was performed by Paul Fjeld, and arranged for by Deborah Douglas of
      # the Museum. Many thanks to both. The images (with suitable reduction
      # in storage size and consequent reduction in image quality as well) are
      # available online at www.ibiblio.org/apollo. If for some reason you
      # find that the images are illegible, contact me at info@sandroid.org
      # about getting access to the (much) higher-quality images which Paul
      # actually created.
      #
      # Notations on the hardcopy document read, in part:
      #
      #       Assemble revision 001 of AGC program LMY99 by NASA 2021112-61
      #       16:27 JULY 14, 1969
      #
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      <Page LM1465 638>
      # Page 1466

```

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**# Page 1467**

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**# Page 1468**

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This code is written to file `src/Luminary099/TJET-LAW.agc`.

## 2.40 TRIM GIMBAL CNTROL SYSTEM

```

774  <src/Luminary099/TRIM-GIMBAL-CNTROL-SYSTEM.agc 774>≡
      # Copyright:   Public domain.
      # Filename:    TRIM_GIMBAL_CNTROL_SYSTEM.agc
      # Purpose:     Part of the source code for Luminary 1A build 099.
      #              It is part of the source code for the Lunar Module's (LM)
      #              Apollo Guidance Computer (AGC), for Apollo 11.
      # Assembler:   yaYUL
      # Contact:      Ron Burkey <info@sandroid.org>.
      # Website:      www.ibiblio.org/apollo.
      # Pages:        1472-1485
      # Mod history:  2009-05-27 RSB   Adapted from the corresponding
      #              Luminary131 file, using page
      #              images from Luminary 1A.
      #
      # This source code has been transcribed or otherwise adapted from
      # digitized images of a hardcopy from the MIT Museum. The digitization
      # was performed by Paul Fjeld, and arranged for by Deborah Douglas of
      # the Museum. Many thanks to both. The images (with suitable reduction
      # in storage size and consequent reduction in image quality as well) are
      # available online at www.ibiblio.org/apollo. If for some reason you
      # find that the images are illegible, contact me at info@sandroid.org
      # about getting access to the (much) higher-quality images which Paul
      # actually created.
      #
      # Notations on the hardcopy document read, in part:
      #
      #       Assemble revision 001 of AGC program LMY99 by NASA 2021112-61
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      # Page 1479

```

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This code is written to file `src/Luminary099/TRIM-GIMBAL-CNTRL-SYSTEM.agc`.





## Chapter 3

# Notes, Bibliography and Indexes

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