

O Flight Phase

$$\begin{array}{c}
X \\
\zeta = \begin{pmatrix} x \\ x \\ \dot{x} \\ \dot{z} \end{pmatrix}
\end{array}$$

$$\dot{X}f = \begin{pmatrix} \dot{x} \\ \dot{z} \\ 0 \\ -g \end{pmatrix}$$

@ Stance Phase

$$X_{s} = \begin{pmatrix} r \\ \theta \\ r \end{pmatrix}$$

$$\begin{pmatrix} X \\ Z \end{pmatrix} = \begin{pmatrix} -\Upsilon \cdot Sin\theta \\ \Upsilon \cdot \cos\theta \end{pmatrix}$$

Kinetic energy

$$T = \frac{2}{M} \cdot (\dot{\gamma}^2 + \dot{\gamma}^2 \cdot \dot{\theta}^2)$$

Potential energy

N=m·g· ·· · cθ + k (lo-r)2

⇒ Lagrange formulation

$$\dot{\gamma} = \gamma \cdot \dot{\theta}^2 - g \cdot c\theta + \frac{k}{m} \cdot (l_0 - r)$$

$$\theta = \frac{-2}{r} \cdot r \cdot \theta + \frac{9}{r} \cdot 5\theta$$

(1) → (2) Flight → Stance

Assuming the spring is ideal, so no energy loss

$$\gamma = 10$$

$$\dot{\theta} = -\frac{1}{r} \cdot (\dot{x} \cdot (\theta + Z \cdot S\theta))$$

Raibert style controller





accelerate de accelerate

· leg's neutral point

$$Xf = \frac{\dot{x} \cdot T_S}{2} + R \cdot err + \bar{l} \cdot esum$$

$$\theta = a \tan \left(\frac{x_f}{l_0} \right)$$

Ts: time of stance Phase

Ref

M. Raibert, M. Chepponis and H. Brown, "Running on four legs as though they were one," in *IEEE Journal on Robotics and Automation*, vol. 2, no. 2, pp. 70-82, June 1986, doi: 10.1109/JRA.1986.1087044.

http://www.cim.mcgill.ca/~aki/research/SLIP_Hopper.htm

http://underactuated.mit.edu/simple_legs.html#section3