

# Stars and Planets Problem Set6

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## Exercise VI.1 Synodical timescale

(a)

The  $t_{syn}$  is the least common multiple of  $P_1$  and  $P_2$ :

$$t_{syn} = [P_1, P_2]$$

(b)

how to calculate?

## Exercise VI.2 Epicycle approximation

(a)

Assuming that  $e \ll 1$ , from the Kepler equation  $M = E - e \sin E$  we can have:

$$\begin{aligned}\sin E &= \sin(M + e \sin E) = \sin M \cos(e \sin E) + \cos M \sin(e \sin E) \\ &= \sin M \left(1 + \frac{1}{2}(e \sin E)^2 + \cdots\right) + \cos M (e \sin E + \cdots) \\ &= \sin M + \mathcal{O}(e)\end{aligned}$$

and:

$$\begin{aligned}\cos E &= \cos(M + e \sin E) = \cos M \cos(e \sin E) - \sin M \sin(e \sin E) \\ &= \cos M \left(1 + \frac{1}{2}(e \sin E)^2 + \cdots\right) - \sin M (e \sin E + \cdots) \\ &= \cos M - e \sin M \sin E + \mathcal{O}(e^2) = \cos M - e \sin^2 M + \mathcal{O}(e^2) \\ &= \cos M + \mathcal{O}(e)\end{aligned}$$

Firstly consider  $r$ :

$$\begin{aligned}\cos E &= e + \frac{r}{a} \cos \nu = \frac{e + \cos \nu}{1 + e \cos \nu} \\ \cos \nu &= \frac{\cos E - e}{1 - e \cos E} \\ r &= \frac{a(\cos E - e)}{\cos \nu} = a(1 - e \cos E)\end{aligned}$$

Plug in  $\cos E = \cos M + \mathcal{O}(e)$  and we have:

$$r \approx a(1 - e \cos M) + \mathcal{O}(e^2)$$

Secondly consider  $\nu$ :

$$\cos \nu = \frac{\cos E - e}{1 - e \cos E} = (\cos E - e)(1 + e \cos E) = \cos E - e \sin^2 E + \mathcal{O}(e^2)$$

Plug in  $\sin E = \sin M + \mathcal{O}(e)$  and  $\cos E = \cos M - e \sin^2 M + \mathcal{O}(e^2)$ , and we have:

$$\cos \nu = \cos M - 2e \sin^2 M + \mathcal{O}(e^2)$$

If we assume that  $\nu \approx M + 2e \sin M + \mathcal{O}(e^2)$ , we can have:

$$\begin{aligned}\cos \nu &= \cos(M + 2e \sin M + \mathcal{O}(e^2)) \\ &= \cos M \cos(2e \sin M) - \sin M \sin(2e \sin M) + \mathcal{O}(e^2) \\ &= \cos M - 2e \sin^2 M + \mathcal{O}(e^2)\end{aligned}$$

Therefore, we can conclude that  $\nu \approx M + 2e \sin M + \mathcal{O}(e^2)$ .

(b)

how to prove?

(c)

$$\begin{aligned}\phi'_{\text{eff}}(r_o) &= Anr_o^{n-1} - \frac{l_z^2}{2} \frac{2}{r_o^3} = 0 \\ r_o &= \left( \frac{l_z^2}{An} \right)^{\frac{1}{n+2}}\end{aligned}$$

(d)

Expanding the potential around  $r = r_o$ :

$$\phi_{\text{eff}} = \phi_{\text{eff}}(r_o) + \phi'_{\text{eff}}(r_o)x + \frac{1}{2}\phi''_{\text{eff}}(r_o)x^2$$

where  $x = r - r_o$  and  $\phi'_{\text{eff}}(r_o) = 0$ . Therefore the equation of motion (3) becomes:

$$\ddot{x} = -x\phi''_{\text{eff}}(r_o) = -x(An(n-1)r_o^{n-2} + 3l_z^2r_o^{-4}) = -(n+2)l_z^2r_o^{-4}x$$

Compared to  $\ddot{x} = -\kappa^2x$ , we have:

$$\kappa = \frac{\sqrt{n+2}l_z}{r_o^2} = \sqrt{n+2}\left(\frac{l_z^{\frac{2-n}{2}}}{An}\right)^{-\frac{2}{n+2}}$$

In a Keplerian potential ( $n=-1$ ):

$$\kappa = A^2/l_z^3 = \Omega$$

where  $\Omega$  is the orbital frequency. **why the orbital frequency**

(e)

**For which values of  $n$  does the circular orbit solution become unstable? What is the physical reason?**

## Exercise VI.3 The Trojans

(a)

From the law of cos, we have:

$$\begin{aligned} r_1^2 &= m^2 + 2mr \cos \theta + r^2 \\ r_2^2 &= (1-m)^2 - 2(1-m)r \cos \theta + r^2 \end{aligned}$$

Because  $m \ll 1$ , we can expand  $r_1^{-1}$  as:

$$\begin{aligned} r_1^{-1} &= (m^2 + 2mr \cos \theta + r^2)^{-1/2} \\ &\approx (1 + 2\Delta + \Delta^2 + 2m \cos \theta)^{-1/2} \\ &\approx 1 - 1/2(2\Delta + \Delta^2 + 2m \cos \theta) + 3/8(2\Delta + \Delta^2 + 2m \cos \theta)^2 \\ &\approx 1 - \Delta + \Delta^2 - m \cos \theta \end{aligned}$$

We can expand  $r_2^{-1}$  as:

$$\begin{aligned} r_2^{-1} &= ((1-m)^2 - 2(1-m)r \cos \theta + r^2)^{-1/2} \\ &\approx (1 - 2 \cos \theta + 1)^{-1/2} \\ &= \frac{1}{\sqrt{2(1 - \cos \theta)}} \end{aligned}$$

We can expand  $r^2$  as:

$$r^2 = (1 + \Delta)^2 = 1 + 2\Delta + \Delta^2$$

Plug the three above relations into the effective potential:

$$\begin{aligned} \phi_{\text{eff}} &= -\frac{1-m}{r_1} - \frac{m}{r_2} - \frac{1}{2}r^2 \\ \phi_{\text{eff}} &= m(\cos \theta - \frac{1}{\sqrt{2(1 - \cos \theta)}}) - \frac{3}{2}\Delta^2 + m - 3/2 \end{aligned}$$

Ignore the constants in the potential and we get:

$$\phi_{\text{eff}} = m(\cos \theta - \frac{1}{\sqrt{2(1 - \cos \theta)}}) - \frac{3}{2}\Delta^2$$

**(b)**

The vector form of the equation of motion is:

$$\ddot{\mathbf{r}} + 2(\boldsymbol{\omega} \times \dot{\mathbf{r}}) = -\nabla \phi_{\text{eff}}$$

The radial component of the equation of motion is:

$$\ddot{r} - r\dot{\theta}^2 - 2r\dot{\theta} = -\frac{\partial \phi_{\text{eff}}}{\partial r}$$

Plug in the expression of the effective potential and we have:

$$\ddot{\Delta} - (1 + \Delta)\dot{\theta}^2 - 2(1 + \Delta)\dot{\theta} = 3\Delta$$

**(c)**

Assume  $\Delta \ll 1$  and we get:

$$\ddot{\Delta} - \dot{\theta}^2 - 2\dot{\theta} = 3\Delta$$

And  $\dot{\theta}^2$  must be the higher order small value compared to  $\dot{\theta}$ . Therefore, the above equation can be reduced to:

$$\ddot{\Delta} - 2\dot{\theta} = 3\Delta$$

If the objects are confined in a small range of radius, then the motion along the radial direction can not be too large, so  $\ddot{\Delta}$  is also a higher order small value. Equation (8) can be reduced to:

$$2\dot{\theta} + 3\Delta = 0$$

(d)

The azimuthal component of the equation of motion is:

$$r\ddot{\theta} + 2\dot{r}\dot{\theta} + 2\dot{r} = -\frac{1}{r} \frac{\partial \phi_{\text{eff}}}{\partial \theta}$$

Plug in  $r = 1 + \Delta$  and we have:

$$(1 + \Delta)\ddot{\theta} + 2\dot{\Delta}\dot{\theta} + 2\dot{\Delta} = -\frac{1}{1 + \Delta} \frac{\partial \phi_{\text{eff}}}{\partial \theta}$$

Ignore the higher order small value:

$$\ddot{\theta} + 2\dot{\Delta} = -\frac{\partial \phi_{\text{eff}}}{\partial \theta}$$

(e)

The effective potential can be rewritten as:

$$\phi_{\text{eff}} = -\frac{m}{2} \left( 4 \sin^2 \frac{\theta}{2} + \frac{1}{\sin \theta/2} \right) - \frac{3}{2} \Delta^2 + m$$

From the radial component of the equation of motion we have  $\Delta = -2/3\dot{\theta}$ . Plug the above two expressions into the azimuthal component and we can have:

$$-\frac{1}{3}\ddot{\theta} = \frac{m}{2} \frac{\partial}{\partial \theta} \left( 4 \sin^2 \frac{\theta}{2} + \frac{1}{\sin \theta/2} \right) + 3\Delta \frac{\partial \Delta}{\partial \theta}$$

Ignore the higher order small value  $3\Delta \frac{\partial \Delta}{\partial \theta}$  and times  $\dot{\theta}$  on both sides:

$$\begin{aligned} \dot{\theta}\ddot{\theta} &= \frac{3}{2}m \frac{\partial}{\partial \theta} \left( 4 \sin^2 \frac{\theta}{2} + \frac{1}{\sin \theta/2} \right) \dot{\theta} \\ 0 &= \frac{d}{dt} \left( \frac{1}{2} \dot{\theta}^2 + \frac{3}{2}m \left( 4 \sin^2 \frac{\theta}{2} + \frac{1}{\sin \theta/2} \right) \right) \end{aligned}$$

The integral of motion under this approximation gives a conserved quantity  $I$ :

$$I = \frac{1}{2} \dot{\theta}^2 + \frac{3}{2}m \left( 4 \sin^2 \frac{\theta}{2} + \frac{1}{\sin \theta/2} \right)$$

(f)

The potential component is:

$$U = \frac{3}{2}m(4\sin^2\frac{\theta}{2} + \frac{1}{\sin\theta/2})$$

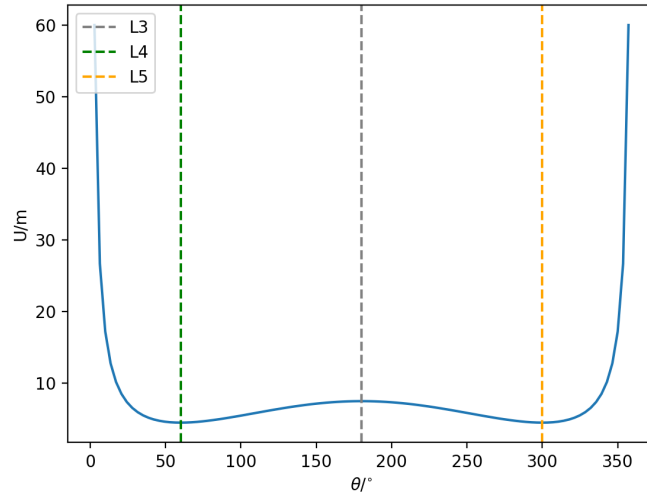
Take the first derivative of the potential and we can get the Lagrange points:

$$U'/m = 4\sin\frac{\theta}{2}\cos\frac{\theta}{2} - \frac{1}{2}\frac{\cos\theta/2}{\sin^2\theta/2}$$

$$\cos\frac{\theta}{2} = 0 \text{ or } \sin\frac{\theta}{2} = 1/2$$

$$\theta = \pi/3, \pi/2 \text{ or } 5\pi/3$$

The potential and the Lagrange points are shown as below.



(g)

the extent of the widest possible Trojan orbit?

(h)

Expand the potential to the second order at L4  $\theta = \pi/3$ :

$$U''/m = 2\cos\theta + \frac{1}{2}\sin^{-3}\frac{\theta}{2}\cos^2\frac{\theta}{2} + \frac{1}{4}\sin^{-1}\frac{\theta}{2}$$

$$\omega_{\text{lib}} = \sqrt{k/m} = \sqrt{U''/m} = \sqrt{4.5} = 2.12 \omega$$

$$t_{\text{lib}} = 3t$$

The orbital period for Jupiter is  $t = 12$  years, so the orbital period for Jupiter's Trojans around L4 is  $t_{\text{lib}} = 36$  years.

## Exercise VI.4 Tides

(a)

For the Earth-Moon system, the value for  $n$  (the mean motion) is  $n = \frac{2\pi}{28\text{days}} = 0.22\text{day}^{-1}$ .

(b)

The spin-down timescale is:

$$t_{\text{de-spin, p}}^{-1} = \frac{\dot{\Omega}_p}{\Omega_p} = -\frac{\Gamma}{\Omega_p I_p} = -\frac{3k_{2p}}{2QC_I} \frac{m_s^2}{(m_s + m_p)m_p} \left(\frac{R_p}{d}\right)^3 \frac{n}{\Omega_p}$$

For the Moon to spin-down the Earth, the love number  $k_{2p} =$ , the Quality factor  $Q =$ , the inertia factor  $C_I =$ ,  $m_s = 7.3477 \times 10^{25}$  g,  $m_p = 5.974 \times 10^{27}$  g,  $R_p = 6.378 \times 10^8$  cm,  $d = 384,401$  km,  $n = \frac{2\pi}{28\text{days}}$ , and  $\Omega_P = \frac{2\pi}{1 \text{ day}}$ . Therefore the timescale for the Moon to spin down the Earth is 4.4 billion years .

For the Sun to spin-down the Earth, the love number  $k_{2p} = 0.3$ , the Quality factor  $Q = 12$ , the inertia factor  $C_I = 0.33$ ,  $m_s = 1.99 \times 10^{33}$  g,  $m_p = 5.974 \times 10^{27}$  g,  $R_p = 6.378 \times 10^8$  cm,  $d = 1 \text{ au}$ ,  $n = \frac{2\pi}{365\text{days}}$ , and  $\Omega_P = \frac{2\pi}{1 \text{ day}}$ . Therefore the timescale for the Sun to spin down the Earth is 19.9 billion years.

(c)

The time taken for the Moon to "crash" into the Earth is:

$$t_{\text{orbit}}^{-1} = \frac{9k}{2Q} \frac{m_s}{m_p} \left(\frac{R_p}{d}\right)^5 n$$

Plug in the values from (b) and we can get  $t_{\text{orbit}} = 7 \text{ Gyr}$ .

(d)

how to motivate?

## Exercise VI.5 Hot Jupiter migration by tides

(a)

The total energy of the original orbit is:

$$E_0 = -\frac{Gm}{2a_0}$$

where the  $m$  is the total mass of the star and the hot jupiter. The kinetic energy of the original orbit is:

$$K_0 = \frac{Gm}{2a_0}$$

The gravitational potential energy of the orbit is:

$$U_0 = -\frac{Gm}{a_0}$$

When the magnitude of the orbit velocity is suddenly changed by a factor of  $f$ , the kinetic energy changes to:

$$K_1 = f^2 K_0 = \frac{f^2}{2} \frac{Gm}{a_0}$$

The gravitational energy remains the same:

$$U_1 = U_0 = -\frac{Gm}{a_0}$$

Therefore, the total energy changes to:

$$E_1 = U_1 + K_1 = \left(\frac{f^2}{2} - 1\right) \frac{Gm}{a_0}$$

Compared to the energy expression of the Kelperian orbit  $E_1 = -\frac{Gm}{2a_1}$  we can get:

$$a_1 = \frac{a_0}{2 - f^2}$$

When the magnitude of the orbit velocity is suddenly changed by a factor of  $f$ , the total angular momentum changes to:

$$l_1 = fl_0 = f\sqrt{Gma_0(1 - e_0^2)}$$

Compared to the angular momentum expression of the Kelperian orbit  $l_1 = \sqrt{Gma_1(1 - e_1^2)}$  we can get:

$$e_1 = 1 - f^2$$

The pericenter  $r_{p1}$  is:

$$r_{p1} = a_1(1 - e_1) = \frac{f^2}{2 - f^2} a_0$$



(b)

During this tidal dissipation step, the hot Jupiter's angular momentum is roughly conserved as it circularizes to a final, close-in semimajor axis  $e_2 = 0$ :

$$l_1 = \sqrt{Gma_1(1 - e_1^2)} = l_2 = \sqrt{Gma_2}$$
$$a_2 = a_1(1 - e_1^2) = a_1 f^2 (2 - f^2) = a_0 f^2$$

If  $a_0 = 5$  au and  $a_2 = 0.05$  au, then  $f = 0.1$ .

(c)

I don't think the existence of these planets can be explained by this mechanism. **explain my answer**

## Exercise VI.6 Geometry of resonances

(a)

(b)

(c)

(d)

## Exercise VI.7 Planet trapping

(a)

(b)

(c)

(d)