Cambridge Part III Maths

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Hydrodynamic Stability

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Notes created based on Josh Kirklin's LATEX packages & classes. Please do not distribute these notes other than to fellow Part III students. Please send errors and suggestions to cwp29@cam.ac.uk.

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1 Introduction

We are typically interested in whether a given flow solution u(x,t) is 'stable', certainly to small (infinitesimal) disturbances and perhaps to larger perturbations too. We perturb u(x) to $u(x) + \hat{u}(x,t)$ and define the *perturbation energy* as

$$E(t) \equiv \int \frac{1}{2} \hat{\boldsymbol{u}}^2(\boldsymbol{x}, t) \, dV$$

A solution is said to be stable if

$$\lim_{t \to \infty} \frac{E(t)}{E(0)} = 0$$

for all perturbations $\hat{\boldsymbol{u}}$. Conversely, if there exists $\hat{\boldsymbol{u}}$ such that $E(t) \not\to 0$ then \boldsymbol{u} is unstable. The nature of E(0) determines the type of perturbation:

- If $E(0) \to 0$ we have an infinitesimal disturbance
- If $E(0) < \delta$ then we probe finite amplitude disturbances
- If $E(0) \to \infty$ this probes the global stability

In the first 9 lectures we focus on the first situation, which is linear stability analysis. Consider the Navier-Stokes equations

$$\frac{\partial \boldsymbol{u}}{\partial t} + \boldsymbol{u} \cdot \nabla \boldsymbol{u} + \nabla p = \frac{1}{\mathrm{Re}} \nabla^2 \boldsymbol{u}$$

If U(x) is a steady (basic) solution then

$$\boldsymbol{U} \cdot \nabla \boldsymbol{U} + \nabla P = \frac{1}{\mathrm{Re}} \nabla^2 \boldsymbol{U}$$

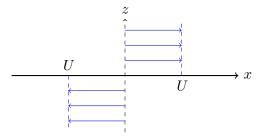
Let $\boldsymbol{u} = \boldsymbol{U}(\boldsymbol{x}) + \hat{\boldsymbol{u}}(\boldsymbol{x},t), p = P + \hat{p}$. Then

$$\frac{\partial \hat{\boldsymbol{u}}}{\partial t} + \boldsymbol{u} \cdot \nabla \boldsymbol{U} + \boldsymbol{U} \cdot \nabla \hat{\boldsymbol{u}} + \hat{\boldsymbol{u}} - \nabla \hat{\boldsymbol{u}} + \nabla \hat{\boldsymbol{p}} = \frac{1}{\mathrm{Re}} \nabla^2 \hat{\boldsymbol{u}}$$

The term $\hat{u} \cdot \nabla \boldsymbol{U}$ is stabilising whilst the term $\nabla^2 \hat{u}/\text{Re}$ is stabilising. Therefore, we expect stability as $\text{Re} \to 0$ as this term dominates, and instability as $\text{Re} \to \infty$. Thus there exists some value Re_{crit} at which instability arises. We will ask what this value is, and what is the form of initial instability/mode/pattern?

2 Kelvin-Helmholtz instability

See Drazen (2002), section 3.3, pages 47–50. Here we take a different approach and derive Rayleigh's equation (example 8.3, page 151 of Drazen).



Consider a flow $\boldsymbol{u} = U(z)\hat{\boldsymbol{x}}$ where

$$U(z) = \begin{cases} U & z > 0 \\ -U & z < 0 \end{cases}$$

The linearised, inviscid equation for perturbation $\hat{\boldsymbol{u}}$ is

$$\frac{\partial \hat{\boldsymbol{u}}}{\partial t} + \hat{\boldsymbol{w}}U'\hat{\boldsymbol{x}} + U\frac{\partial \hat{\boldsymbol{u}}}{\partial x} + \nabla \hat{\boldsymbol{p}} = 0$$
$$\nabla \cdot \hat{\boldsymbol{u}} = 0$$

The boundary conditions are $\hat{\boldsymbol{u}} \to 0$ as $z \to \pm \infty$, i.e. no energy radiated in from infinity. We will work in 2D $(\hat{u},\hat{w}) = (\psi_z, -\psi_x)$ and let $\psi(x,z,t) = \phi(z)e^{i\alpha(x-ct)}$ where c is a complex eigenvalue, currently unknown. Formally, this is equivalent to taking a Fourier transform. We have

$$i\alpha(U-c)\begin{pmatrix} \phi' \\ -i\alpha\phi \end{pmatrix} + \begin{pmatrix} -i\alpha U'\phi \\ 0 \end{pmatrix} + \begin{pmatrix} i\alpha p \\ \frac{\partial p}{\partial z} \end{pmatrix} = 0$$

We can eliminate p via $\partial_z(top) - i\alpha(bottom)$ to get

$$(U-c)(\phi''-\alpha^2\phi)-U''\phi=0$$

with boundary conditions $\phi \to 0$ as $z \to \pm \infty$. This is Rayleigh's equation. Note that c is the crucal eigenvalue. We wish to know when $c_i = \Im(c) > 0$ as a function of U(z), as c_i is the growth rate:

$$\hat{u} \propto e^{i\alpha(x-ct)} = e^{i\alpha(c-c_rt-ic_it)} = e^{i\alpha(x-c_rt)+\alpha c_it}$$

Note the following:

- There is a symmetry $\alpha \mapsto -\alpha$, so without loss of generality we consider $\alpha > 0$.
- The complex conjugate is also a solution with $c \mapsto c^*$. Hence an unstable mode has a damped partner, so we have stability only if all modes are 'neutral' i.e. $c_i = 0$.
- There is a possible singularity at y where U(y) = c, called the *critical layer*. If c is real, see later

We now solve Rayleigh's equation with U(z) defined as before. We solve above and below z = 0 and piece the solutions together. Since U'' = 0, we have

$$\phi'' = \alpha^2 \phi$$

which admits a solution satisfying the boundary conditions:

$$\phi = \begin{cases} A^{-\alpha z} & z > 0 \\ Be^{\alpha z} & z < 0 \end{cases}$$

The matching conditions at z = 0 are

1. Pressure \hat{p} continuous at z = 0, with \hat{p} given by:

$$\hat{p} = U'\phi - (U - c)\phi'$$

2. Kinematic condition at the surface:

$$\frac{\mathbf{D}}{\mathbf{D}t}\left(z - \zeta(x, t)\right) = 0$$

where $z = \zeta(x,t)$ is the position of the surface. After linearising, we have

$$w - \frac{\partial \zeta}{\partial t} - U \frac{\partial \zeta}{\partial x} = 0$$

Inserting the form of w and U we require that

$$\zeta = -\frac{\phi}{U-c}$$

is continuous across z = 0.

Requiring p continuous gives

$$-(U-c)A(-\alpha) = -(-U-c)B(\alpha)$$

Requiring ζ continuous gives

$$\frac{A}{U-c} = \frac{B}{-U-c}$$

Hence we have

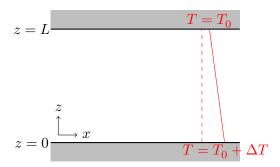
$$(U-c)^2 = -(U+c)^2$$

i.e. $c = \pm iU$ so the growth rate is αU . Thus the flow is unstable to waves of all wavelengths. The instability may be remedied

- by adding a density stratification, which stabilises long wavelengths (small α)
- by adding surface tension, which stabilises short wavelengths (large α), e.g. Drazen page 50 equation 3.21.

3 Thermal instabilities: convection

Consider two parallel plates separated by distance L with fluid subject to gravity and temperatue difference ΔT between the plates. The lower plate is heated.



The basic state consists of no motion, with heat transfer by conduction only.

Governing equations. The governing equations are

$$\rho \frac{\mathbf{D}\boldsymbol{u}}{\mathbf{D}t} + \nabla p = \mu \nabla^2 \boldsymbol{u} + \rho g \hat{\boldsymbol{z}}$$
$$\frac{\partial T}{\partial t} + \boldsymbol{u} \cdot \nabla T = \kappa \nabla^2 T$$
$$\frac{\mathbf{D}\rho}{\mathbf{D}t} + \rho \nabla \cdot \boldsymbol{u} = 0$$

We need a relationship between ρ and T. Most cases of interest have ΔT and $\Delta \rho$ small, i.e. $\Delta \rho \ll \rho_0, \Delta T \ll T_0$. Two consequences of this assumption are:

1. We can Taylor expand $\rho = \rho(T)$:

$$\rho \approx \rho(T_0) \left[1 - \alpha(T - T_0) \right]$$

where $\alpha > 0$ is the coefficient of thermal expansion, such that T increases when ρ decreases. We write $\rho_0 = \rho(T_0)$.

2. We can adopt a Boussinesq approximation: acknowledge density changes only in the buoyancy term $\rho g\hat{z}$. Importantly, we can assume the fluid is incompressible.

Define $\theta = T - T_0$. The governing equations are now

$$\begin{split} \rho_0 \frac{\mathrm{D} \boldsymbol{u}}{\mathrm{D} t} + \nabla p &= \mu \nabla^2 \boldsymbol{u} + \rho_0 (1 - \alpha \theta) g \hat{\boldsymbol{z}} \\ \frac{\partial \theta}{\partial t} + \boldsymbol{u} \cdot \nabla \theta &= \kappa \nabla^2 \theta \\ \nabla \cdot \boldsymbol{u} &= 0 \end{split}$$

The basic state is $u = 0, \theta = \Delta T(1 - z/L)$ and

$$\frac{\mathrm{d}p}{\mathrm{d}z} = -\rho_0(1-\alpha\Delta T(1-z/L))g$$

We now non-dimensionalise using scalings $t \sim L^2/\kappa, u \sim \kappa/L, \theta \sim \Delta T$, e.g. $\theta = \Delta T \theta^*$ where θ^* is the non-dimensionalised variable. We normalise the $\frac{\mathrm{D} \boldsymbol{u}^*}{\mathrm{D} t^*}$ term, to get:

$$\begin{split} \frac{\mathbf{D}\boldsymbol{u}^*}{\mathbf{D}t^*} + \nabla^*p^* &= \frac{\mu}{\rho_0\kappa}\nabla^{*2}\boldsymbol{u}^* + \frac{\alpha g\Delta TL^3}{\kappa^2}\theta^*\hat{\boldsymbol{z}}\\ \frac{\partial\theta^*}{\partial t^*} + \boldsymbol{u}^*\cdot\nabla^*\theta^* &= \nabla^{*2}\theta^* \end{split}$$

Define the Prandtl number

$$\sigma \equiv \frac{\nu}{\kappa} = \frac{\mu}{\rho_0 \kappa}$$

which is the ratio of viscous/momentum to thermal diffusivity. Typical values are 0.72 in air, 7 in water, 10^5 in magma. We also define the Rayleigh number

$$Ra \equiv \frac{\alpha \Delta T g L^3}{\kappa \nu}$$

which is the ratio of destabilising buoyancy to stabilising diffusion. Dropping the * notation, we have

$$\frac{\partial \boldsymbol{u}}{\partial t} + \boldsymbol{u} \cdot \nabla \boldsymbol{u} + \nabla p = \sigma \nabla^2 \boldsymbol{u} + \sigma \operatorname{Ra} \theta \hat{\boldsymbol{z}}$$
$$\frac{\partial \theta}{\partial t} + \boldsymbol{u} \cdot \nabla \theta = \nabla^2 \theta$$
$$\nabla \cdot \boldsymbol{u} = 0$$

Boundary conditions. There are three combinations of boundary condition available in this problem.

Fixed wall
$$v = 0$$
 Fixed wall $v = 0$ Free slip $v = \frac{\partial u}{\partial z} = \frac{\partial v}{\partial z} = 0$

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The double fixed wall case is easiest to replicate in a lab, whilst the double free slip case is the easiest analytically, which we shall use.

Basic state. In the basic state we have conductive profile $\mathbf{u}_0 = 0, \theta_0 = 1 - z$ and from integration $p_0 = \sigma \operatorname{Ra}(z - \frac{1}{2}z^2)$. We generate linearised equations for perturbations $\theta = \theta_0 + \theta', \mathbf{u} = \mathbf{u}_0 + \mathbf{u}', p = p_0 + p'$. As usual with linear stability analysis, we assume $(\theta, \mathbf{u}', p')$ are small.

$$\frac{\partial \mathbf{u}'}{\partial t} + \mathbf{u}' \nabla \mathbf{u}' + \nabla p' = \sigma \nabla^2 \mathbf{u}' + \sigma \operatorname{Ra} \theta' \hat{\mathbf{z}}$$
$$\frac{\partial \theta'}{\partial t} - w' + \mathbf{u}' \nabla \theta' = \nabla^2 \theta'$$
$$\nabla \cdot \mathbf{u}' = 0$$

Dropping the ' notation for clarity we have perturbation equations

$$\left(\frac{\partial}{\partial t} - \sigma \nabla^2\right) \boldsymbol{u} + \nabla p = \sigma \operatorname{Ra}\theta \hat{\boldsymbol{z}}$$
(1)

$$\nabla \cdot \boldsymbol{u} = 0 \tag{2}$$

$$\left(\frac{\partial}{\partial t} - \nabla^2\right)\theta = w\tag{3}$$

The perturbation boundary conditions also follow by inserting variables into the total boundary conditions, e.g. $\theta = \theta_0 + \theta' = 1$ at z = 0 combined with $\theta_0 = 1$ at z = 0 gives $\theta' = 0$. Similarly, $\theta' = 0$ at z = 1 and in fact all boundary conditions are homogeneous. To proceed further, we need to reduce the equations (1),(2) and (3) into a single equation. From $\nabla \times (1)$ we have

$$\left(\frac{\partial}{\partial t} - \sigma \nabla^2\right) \boldsymbol{\omega} = \sigma \text{Ra} \nabla \times \theta \hat{\boldsymbol{z}}$$

Taking the curl again and using $\nabla \times \boldsymbol{\omega} = \nabla \times (\nabla \times \boldsymbol{u}) = \nabla (\nabla \cdot \boldsymbol{u}) - \nabla^2 \boldsymbol{u}$ we have

$$\left(\frac{\partial}{\partial t} - \sigma \nabla^2\right) (-\nabla^2 \boldsymbol{u}) = \sigma \mathrm{Ra} \nabla \times (\nabla \times \theta \hat{\boldsymbol{z}}) = \sigma \mathrm{Ra} \left(\nabla \frac{\partial \theta}{\partial z} - \hat{\boldsymbol{z}} \nabla^2 \theta\right)$$

The z component is

$$\left(\frac{\partial}{\partial t} - \sigma \nabla^2\right) (-\nabla^2 w) = \sigma \text{Ra} \nabla_H^2 \theta \tag{4}$$

where $\nabla_H^2 = \partial_x^2 + \partial_y^2$. Now (3) can be used to eliminate θ by applying the operator $(\partial_t - \nabla^2)$:

$$\left(\frac{\partial}{\partial t} - \sigma \nabla^2\right) \left(\frac{\partial}{\partial t} - \nabla^2\right) \nabla^2 w = \sigma \text{Ra} \nabla_H^2 w \tag{5}$$

This is a $6^{\rm th}$ order PDE for w, hence we need three boundary conditions at each wall z=0,1. We use stress-free (i.e. free slip) at both walls to simplify analysis. Thus we have

$$\frac{\partial u}{\partial z} = \frac{\partial v}{\partial z} = w = 0$$
 at $z = 0, 1$

The second set of conditions comes from incompressibility. Taking $\partial_z(\nabla \cdot \boldsymbol{u})$ we have

$$\frac{\partial}{\partial x} \left(\frac{\partial u}{\partial z} \right) + \frac{\partial}{\partial y} \left(\frac{\partial v}{\partial z} \right) + \frac{\partial^2 w}{\partial z^2} = 0 \implies w_{zz} = 0$$

The third and final set of conditions comes from requiring $\theta = 0$ at z = 0, 1. From (4), $\nabla_H^2 \theta = 0$ implies

$$\left(\frac{\partial}{\partial t} - \sigma \nabla^2\right) \nabla^2 w = 0$$

We now have 6 boundary conditions to supplement the PDE.

Normal mode solution. Seek a solution $w(x,y,z,t)=W(z)e^{ik_1x+ik_2y+\lambda t}$ where k_1,k_2 are wavenumbers and $\lambda\in\mathbb{C}$ is the growth rate. Write $D=\mathrm{d}/\mathrm{d}z$ and $k=\sqrt{k_1^2+k_2^2}$ since the problem is rotationally symmetric in the (x,y) plane. Substituting into (5) we have

$$(\lambda - \left[D^2 - k^2\right])(\lambda - \sigma \left[D^2 - k^2\right])(D^2 - k^2)W = -\sigma \mathrm{Ra}k^2 W$$

with boundary conditions at z = 0, 1:

$$W(0) = W(1) = 0$$

$$D^2 W(0) = D^2 W(1) = 0$$

$$[\lambda - \sigma(D^2 - k^2)] [D^2 - k^2] W = 0 \implies D^4 W(0) = D^4 W(1) = 0$$

The objective is to find

$$\max_{k} \Re\{\lambda(k; \mathrm{Ra}, \sigma)\}$$

The onset of linear instability (for a given σ) at Ra = Ra_{crit} is defined by

$$\max_{k} \Re\{\lambda(k; \mathrm{Ra}_{\mathrm{crit}}, \sigma)\} = 0$$

In general, $\lambda \in \mathbb{C}$, but for this problem it can be proven that at marginality $\Im(\lambda) = 0$ as well as $\Re(\lambda) = 0$; a condition called the *principle of exchange of stabilities*. Hence setting $\lambda = 0$ in the above, we get

$$(D^2 - k^2)^3 W = -\operatorname{Ra} k^2 W \tag{6}$$

Note that σ drops out of the problem! It's easy to see $W(z) = \sin(n\pi z)$ solves (6) and satisfies the free-slip BCs. Hence

$$(n^2\pi^2 + k^2)^3 = \operatorname{Ra} k^2$$

Criticality is then given by

$$Ra_{crit} = \min_{n,k} \frac{(n^2 \pi^2 + k^2)^3}{k^2}$$

We find the minimum in the usual way:

$$\begin{split} \frac{\partial \text{Ra}}{\partial k} &= \frac{3(2k)(n^2\pi^2 + k^2)^2k^2 - 2k(n^2\pi^2 + k^2)^3}{k^4} \\ &= \frac{2k(n^2\pi^2 + k^2)^2(3k^2 - (n^2\pi^2 + k^2))}{k^4} = 0 \\ \implies 2k^2 &= n^2\pi^2 \\ \implies k &= \frac{n\pi}{\sqrt{2}} \end{split}$$

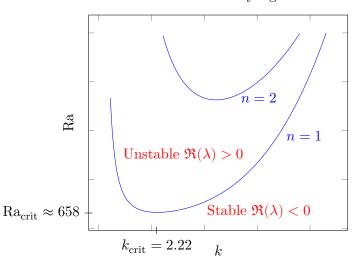
Given $k = n\pi/\sqrt{2}$ the Rayleigh number is

$$Ra(k = \frac{n\pi}{\sqrt{2}}) = \frac{(n^2\pi^2 + \frac{1}{2}n^2\pi^2)^3}{n^2\pi^2/2} = \frac{27}{4}n^4\pi^4$$

Clearly the critical Rayleigh number is given by n = 1, hence

$$Ra_{crit} = \frac{27}{4}\pi^4 \sim 658$$
$$k_{crit} = \frac{\pi}{\sqrt{2}} \sim 2.22$$

Thermal convection Rayleigh number



Results for other boundary conditions are:

- Free–rigid boundary: Ra $_{\rm crit} \sim 1101, k_c = 2.68$
- Rigid–rigid boundary: Ra $_{\rm crit} \sim 1708, k_c = 3.117$

Notice that at criticality only the size of k is specified, not its direction. Hence there are an infinite number of possibilities $\mathbf{k} = (k\cos\phi, k\sin\phi)$. Various different patterns seen which tesselate are as follows.

1. **2D rolls.** Orientate x-axis along k such that $k_2 = 0$. We have velocity components (w specified in problem, u follows from incompressibility)

$$w = W(z) \sin kx$$

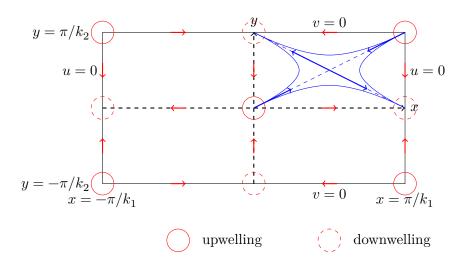
$$v = 0$$

$$u = \frac{\pi \cos \pi z \cos kx}{k}$$

side view z = 1top view $3\pi/2k \qquad 5\pi/2k$ $downwelling \quad upwelling$

2. Rectangles. Velocity components are

$$\begin{split} w &= W(z)\cos k_1 x \cos k_2 y \\ v &= -\frac{k_2}{k^2} W' \cos k_1 x \sin k_2 y \\ u &= -\frac{k_1}{k^2} W' \sin k_1 x \cos k_2 y \end{split}$$



3. Hexagons. Vertical velocity component

$$w = W(z) \left[\cos \frac{k}{2} (\sqrt{3}x + y) + \cos \frac{k}{2} (\sqrt{3}x - y) + \cos ky \right]$$

This is flow in a hexagon of side length $L = 4\pi/3k$.

