
```
% HW 8
% Q1

clc
close all

% a
% Define the matrices A and B
A = [1 2 1; 0 4 3; 0 0 2];
B = [1; 1; 0];

% Number of states (rows of A)
n = size(A, 1);

% Initialize the controllability matrix C
C = B;

% Construct the controllability matrix
for i = 1:n-1
    C = [C, A^i * B];
end

% Check if the system is controllable by verifying the rank of C
isControllable = rank(C) == n;

% Display result
if isControllable
    disp('The system is controllable.');
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else
    disp('The system is not controllable.');
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end

% b
% Extract the first two columns of C, which span the controllable subspace
V = C(:, 1:2);

% Initialize the orthonormal basis vectors
orthonormalBasis = zeros(size(V));

% First orthonormal vector (normalized first vector of V)
orthonormalBasis(:, 1) = V(:, 1) / norm(V(:, 1));

% Second vector orthogonalized against the first
v2_orthogonal = V(:, 2) - proj(V(:, 2), orthonormalBasis(:, 1));

% Normalizing the second orthogonal vector
orthonormalBasis(:, 2) = v2_orthogonal / norm(v2_orthogonal);

% Display the orthonormal basis
disp('Orthonormal basis for the controllable subspace:');
disp(orthonormalBasis);
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% c and d
% Compute e^(At) and integral_part (common for both parts c and d)
t = 1; % Time of transition
eAt = expm(A*t);
integral_part = integral(@(tau) expm(A*(t-tau)) * B, 0, t, 'ArrayValued',
    true);

% c
% Define the initial and final states for part c
x0_c = [1; 1; 1];
xf_c = [0; 0; 0];

% Solve for u(t) for part c
u_t_c = integral_part \ (xf_c - eAt*x0_c);

% Display the input for part c
disp('Input u(t) that drives the state from x(0) to x(1) for part c:');
disp(u_t_c);

% d
% Define the initial and final states for part d
x0_d = [0; 0; 0];
xf_d = [1; 1; 0];

% Solve for u(t) for part d
u_t_d = integral_part \ (xf_d - eAt*x0_d);

% Display the input for part d
disp('Input u(t) that drives the state from x(0) to x(1) for part d:');
disp(u_t_d);

% Function to calculate the projection of v onto u
function p = proj(v, u)
    p = (dot(v, u) / dot(u, u)) * u;
end

The system is not controllable.
Orthonormal basis for the controllable subspace:
    0.7071    -0.7071
    0.7071     0.7071
         0         0

Input u(t) that drives the state from x(0) to x(1) for part c:
    -9.0380

Input u(t) that drives the state from x(0) to x(1) for part d:
     0.0849

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% HW 8
% Q2

clc
close all

% a
% Define the matrices A and B
A = [-1 0 -1; 0 -3 1; 0 0 -2];
B = [1; 0; 2];

% Number of states (rows of A)
n = size(A, 1);

% Initialize the controllability matrix C
C = B;

% Construct the controllability matrix
for i = 1:n-1
    C = [C, A^i * B];
end

% Check if the system is controllable by verifying the rank of C
isControllable = rank(C) == n;

% Display result
if isControllable
    disp('The system is controllable.');
```

```
else
    disp('The system is not controllable.');
```

```
end

% b
% Extract the columns of C that span the controllable subspace
V = C(:, 1:rank(C));

% Initialize the orthonormal basis vectors
orthonormalBasis = zeros(size(V));

% First orthonormal vector (normalized first vector of V)
orthonormalBasis(:, 1) = V(:, 1) / norm(V(:, 1));

% Second vector orthogonalized against the first and normalized
v2_orthogonal = V(:, 2) - proj(V(:, 2), orthonormalBasis(:, 1));
orthonormalBasis(:, 2) = v2_orthogonal / norm(v2_orthogonal);

% Third vector orthogonalized against the first two and normalized
v3_orthogonal = V(:, 3) - proj(V(:, 3), orthonormalBasis(:, 1)) - proj(V(:, 3), orthonormalBasis(:, 2));
orthonormalBasis(:, 3) = v3_orthogonal / norm(v3_orthogonal);

% Display the orthonormal basis
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disp('Orthonormal basis for the controllable subspace:');
disp(orthonormalBasis);

% c
% Define the time interval
t0 = 0;
t1 = 1;

% Calculate the controllability Gramian
Wc = integral(@(tau) expm(A*(t1-tau))*B*B'*expm(A'*(t1-tau)), t0,
    t1, 'ArrayValued', true);

% Display the controllability Gramian
disp('Controllability Gramian from t = 0 to t = 1:');
disp(Wc);

% d
% Define the initial and final states
x0 = [1; 1; 1];
xf = [0; 0; 0];

% Time of transition
t = 1;

% Compute e^(At)
eAt = expm(A*t);

% Solving for the integral part (as in previous part)
integral_part = integral(@(tau) expm(A*(t-tau)) * B, 0, t, 'ArrayValued',
    true);

% Solve for u(t)
u_t = integral_part \ (xf - eAt*x0);

% Display the input
disp('Input u(t) that drives the state from x(0) to x(1):');
disp(u_t);

% e
% Define Q as B*B'
Q = B * B';

% Solve the Lyapunov equation A'P + PA + Q = 0 for P
P = lyap(A, Q);

% Display the P matrix
disp('Matrix P that solves the Lyapunov equation:');
disp(P);

% f
% Compute the infinite-time controllability Gramian (which should be equal to
    P)
% AW + WA' + BBt = 0
Wc_infinite = lyap(A, B*B');

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% Display the infinite-time controllability Gramian
disp('Controllability Gramian from t = 0 to t = infinity:');
disp(Wc_infinite);
```

```
% Function to calculate the projection of v onto u
function p = proj(v, u)
    p = (dot(v, u) / dot(u, u)) * u;
end
```

The system is controllable.

Orthonormal basis for the controllable subspace:

0.4472	-0.3651	-0.8165
0	0.9129	-0.4082
0.8944	0.1826	0.4082

Controllability Gramian from t = 0 to t = 1:

0.1471	0.0444	0.3482
0.0444	0.0575	0.1871
0.3482	0.1871	0.9817

Input u(t) that drives the state from x(0) to x(1):

-0.2102

Matrix P that solves the Lyapunov equation:

0.1667	0.0333	0.3333
0.0333	0.0667	0.2000
0.3333	0.2000	1.0000

Controllability Gramian from t = 0 to t = infinity:

0.1667	0.0333	0.3333
0.0333	0.0667	0.2000
0.3333	0.2000	1.0000

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Q3

DT sys w/ $x_0 = x(0)$ $n=3$

$$x(k+1) = Ax(k) + Bu(k), \quad A \in \mathbb{R}^{3 \times 3}, \quad B \in \mathbb{R}^3$$

* Sys is controllable. Given some desired state x_f , we want to find a sequence of inputs $u(0), u(1), \dots$ that drives the sys to x_f @ some time $k = k_f$ 100%

a) for general $A, B, x_0 \neq x_f$ - fastest that you can drive the sys. to x_f ? What's the smallest possible value of k_f ?

$$C = [B \quad AB \quad A^2B] \quad \text{Since the sys. is controllable } \text{rank}(C) = 3$$

So the smallest value of $k_f = 3$

b) Sequence of inputs for shortest time?

$$x_f, k_f = 3 \quad C \begin{bmatrix} u_{k-1} \\ u_{k-2} \\ u_0 \end{bmatrix} = -A^k x_0$$

$$[B \quad AB \quad A^2B] \begin{bmatrix} u(2) \\ u(1) \\ u(0) \end{bmatrix} = x_f - Ax(0)$$

$$\begin{bmatrix} u(2) \\ u(1) \\ u(0) \end{bmatrix} = [B \quad AB \quad A^2B]^{-1} (x_f - Ax(0))$$

represents state change

general A, B

c) Do there exist specific values of x_0 and x_f , fewer steps than (a)

Yes, example: $Ax_0 = x_f$ - no input needed & the sys. transitions to x_f in one step.

d) general x_0, x_f - specific A , fewer steps than (a)

Yes, certain configurations of A can facilitate faster transitions. Especially,

$$\text{ex: } Ax_0 = x_f \quad \text{one step}$$

true if A has certain props align sys's dyn. favorably w/ desired

e) Assume $B^{3 \times 2} \rightarrow$ the sys's controllability must be re-evaluated.

$$C = [B \quad AB \quad A^2B \quad A^{n-1}B] \rightarrow \text{w/ } B \in \mathbb{R}^{3 \times 2} \text{ sys has 2 inputs @ each step}$$

Yes, there can be. 2 inputs can increase control capacity.

For instance: A has evals that significantly amplify certain state vars. & B may allow direct influence on these var.

Q4 The sys: $\dot{x} = Ax + Bu$, $A \in \mathbb{R}^{n \times n}$, $B \in \mathbb{R}^{n \times m}$.

If there exists a vector $v \in \mathbb{R}^n$ s.t. $v^T A = \lambda v^T$, $v^T B = 0$, then the sys. is not controllable.

$C = [B \ AB \ A^2 B \ \dots \ A^{n-1} B]$; the sys is controllable iff $\text{rank}(C) = n$

(1) - $v^T B = 0 \rightarrow v^T$ is 1st col. of C

(2) - $v^T A = \lambda v^T \rightarrow$ more general: $v^T A^k = \lambda^k v^T$ for $k = 1, \dots, (n-1)$

(1) & (2): $v^T A^k B = \lambda^k v^T B = 0$ for $k = 1, \dots, (n-1)$

So if we multiply v^T w/ any col. block of C we'll get zero vector.

The existence of v implies C does not have full rank.

So if v exists then the sys is not controllable.

Q5 $\dot{x} = Ax + Bu$ (1) Define a new state w s.t. $w = Q^{-1}x$ for some invertible matrix Q (2)

a) rewrite the sys: $\dot{w} = \tilde{A}w + \tilde{B}u$. Find \tilde{A}, \tilde{B} in terms of A, B, Q .

$w = Q^{-1}x$

$\dot{w} = \frac{d}{dt}(Q^{-1}x) = Q^{-1}\dot{x}$
 $\stackrel{(1)}{=} Q^{-1}(Ax + Bu)$

(2) $\rightarrow x = Qw$

$\dot{w} = Q^{-1}(AQw + Bu) = \underbrace{Q^{-1}AQ}_{\tilde{A}}w + \underbrace{Q^{-1}B}_{\tilde{B}}u$

b) show original sys controllable iff second sys. controllable

original sys: Cont. matrix: $C = [B \ AB \ A^2 B \ \dots \ A^{n-1} B]$ - controllable iff $\text{rank}(C) = n$

2nd sys: $\tilde{C} = [\tilde{B} \ \tilde{A}\tilde{B} \ \tilde{A}^2\tilde{B} \ \dots \ \tilde{A}^{n-1}\tilde{B}]$ - substitute \tilde{A} & \tilde{B}

$= [Q^{-1}B, Q^{-1}AQQ^{-1}B, Q^{-1}AQQ^{-1}AQQ^{-1}B, \dots]$

$= [Q^{-1}B, Q^{-1}AB, Q^{-1}A^2B, \dots]$

$\tilde{C} = Q^{-1}[B, AB, A^2B, \dots] = Q^{-1}C$

$\tilde{C} = Q^{-1}C$

$\xrightarrow{\text{invertible}} \text{rank}(\tilde{C}) = \text{rank}(C)$ - so statement is true.