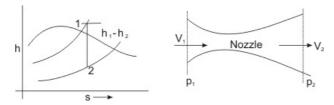
## Iscentropic Flow of a vapor or gas through a nozzle



First law of thermodynamics:

$$h_1 + \frac{V_1^2}{2} = h_2 + \frac{V_2^2}{2}$$
 
$$V_2 \approx \sqrt{2(h_1 - h_2)}$$
 (if  $V_1 \ll V_2$ )

Where  $(k_1-k_2)$  is enthalpy drop across the nozzle Again we know, Tds = dh -  $\nu$ dp

For the isentropic flow,  $dh = \nu dp$ 

or, 
$$\int_{1}^{2} dh = \int_{1}^{2} v dp$$
or, 
$$(h_{1} - h_{2}) = -\int_{1}^{2} v dp$$
 (20.1)

Assuming that the pressure and volume of steam during expansion obey the law  $p\nu^n$  = constant, where n is the isentropic index

$$-\int_{1}^{2} v dp = -\int_{1}^{2} (p_{1}v_{1}^{n})^{\frac{1}{n}} p^{-\frac{1}{n}} dp = -\int_{1}^{2} (p_{2}v_{2}^{n})^{\frac{1}{n}} p^{-\frac{1}{n}} dp$$

$$= -\left\{ p_{2}^{\frac{1}{n}} v_{2} \left[ \frac{p^{1-\frac{1}{n}}}{1-\frac{1}{n}} \right]_{1}^{2} \right\}$$

$$= -\frac{n}{n-1} \left\{ p_{2}^{\frac{1}{n}} v^{2} \left[ p_{2}^{\frac{n-1}{n}} - p_{1}^{\frac{n-1}{n}} \right] \right\}$$

$$= -\frac{n}{n-1} \left\{ p_{2}v_{2} - p_{1}^{\frac{1}{2}} v_{1}^{\frac{n-1}{n}} \right\}$$

$$= \frac{n}{n-1} \left\{ p_{1}v_{1} - p_{2}v_{2} \right\}$$

$$(20.2)$$

Now, mass flow rate

$$m = \rho_2 \ A_2 \ V_2$$

$$\frac{m}{A_2} = \rho_2 \ V_2 = \frac{V_2}{V_2}$$

Therefore, the mass flow rate at the exit of the nozzle

$$\frac{\frac{n}{M}}{A_2} = \frac{1}{\nu_2} \sqrt{\frac{2n}{n-1} p_1 \nu_1 \left(1 - \left(\frac{p_2}{p_1}\right)^{\frac{n-1}{n}}\right)}$$

$$= \sqrt{\frac{2n}{n-1} \frac{p_1}{\nu_1} \left[\left(\frac{p_2}{p_1}\right)^{\frac{2}{n}} - \left(\frac{p_2}{p_1}\right)^{\frac{n+1}{n}}\right)} \tag{20.3}$$

The exit pressure,  $p_2$  determines the m for a given inlet condition. The mass flow rate is maximum

when,

$$\frac{d}{dy}\left[y^{\frac{2}{n}} - y^{\frac{n+1}{n}}\right] = 0 \quad ; \quad y = \frac{p_2}{p_1}$$
$$y = \left[\frac{2}{n+1}\right]^{\frac{n}{n-1}}$$

For maximum m,

$$\frac{p_{or}}{p_1} = \frac{p^*}{p_1} = \frac{p_2}{p_1} = \left[\frac{2}{n+1}\right]^{\frac{n}{n-1}}$$
(20.4)

$$\begin{array}{ll} n = & \gamma = 1.4, & \text{for diatomic gases} \\ & = 1.3, & \text{for super saturated steam} \\ & = 1.135, & \text{for dry saturated steam} \\ & = 1.035 + 0.1x, & \text{for wet steam with dryness fraction x} \end{array}$$

For , 
$$n=1.4$$
,  $p^*=0.528p_1$  (50%drop in inlet pressure) 
$$n=1.3, \quad p^*=0.546p_1$$

If we compare this with the results of sonic properties, as described in the earlier section, we shall observe that the critical pressure occurs at the throat for Ma=1. The critical pressure ratio is defined as the ratio of pressure at the throat to the inlet pressure, for checked flow when Ma=1

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