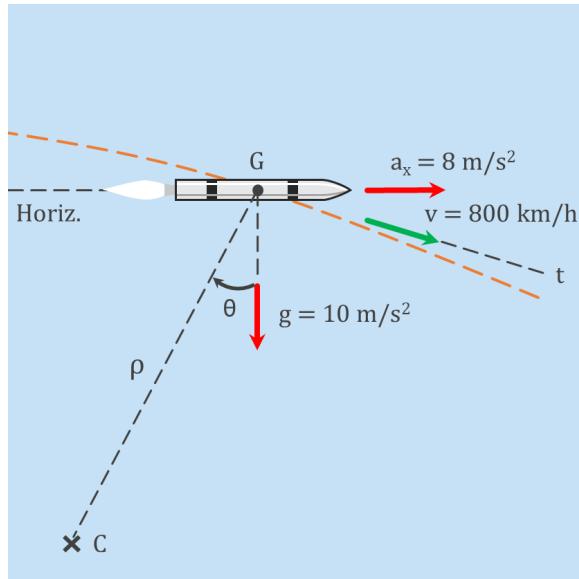


# Rocket Accelerates



A rocket maintains at horizontal attitude of its axis during the powered phase of its flight. The acceleration due to horizontal thrust is  $8 \text{ m/s}^2$ , and the downward acceleration due to gravity is  $g = 10 \text{ m/s}^2$ . At the instant represented, the velocity of the mass centre  $G$  of the rocket along the ( $\theta$ )  $15^\circ$  direction of its trajectory is  $800 \text{ km/h}$ . Determine the radius of curvature  $\rho$  in meters of the flight trajectory. Round to the nearest hundred (e.g. 8700 m).

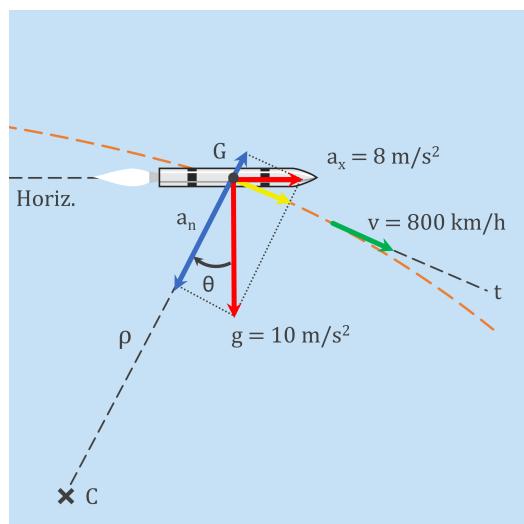


Figure 1: Rocket Accelerates

The normal acceleration  $a_n$  points towards the centre of curvature  $C$ . Figure 1 shows the acceleration vectors  $a_x$  and  $g$  deconstructed in the normal direction (blue) and the tangential direction (yellow). It can be easily seen that  $g$  and  $a_x$  deconstructed in the normal-direction are equal to  $g \cdot \cos(\theta)$  and  $a_x \cdot \sin(\theta)$  respectively. However, in this case  $a_x \cdot \sin(\theta)$  points in the opposite way of  $a_n$  (points ↗ instead of ↘). This means that to determine the final value of  $a_n$ , the term  $a_x \cdot \sin(\theta)$  should be subtracted. Resulting in the final answer:

$$a_n = g \cdot \cos(\theta) - a_x \cdot \sin(\theta) \quad (1)$$

*Given:*

Angle:  $\theta = 15^\circ$

Gravitational acceleration:  $g = 10m/s^2$

Horizontal acceleration:  $a_x = 8m/s^2$

Velocity:  $v = 800km/h = 222.22m/s$

Inserting  $\theta$ ,  $g$  and  $a_x$  into Equation 1 results in:

$$a_n = g \cdot \cos(\theta) - a_x \cdot \sin(\theta) \Rightarrow a_n = 10 \cdot \cos(15) - 8 \cdot \sin(15) = 7.59m/s^2 \quad (2)$$

This results in:

$$\rho = \frac{v^2}{a_n} \Rightarrow \rho = \frac{222.22^2}{7.59} = 6507.40m \approx 6500m \quad (3)$$