

Approach

The approach below gives a guideline in how to solve the problems presented during this course. Correctly applying this approach will lead to a good understanding of the concepts presented in this course.

Analysis

- 1 Explain the problem: which physical phenomena are important in this problem?
- 2 Make a sketch of the problem
- 3 Give the known variables (with the appropriate units!)

Approach

- 1 Explain the assumptions you make to solve the problem
- 2 Show the solution method for solving the problem

Elaboration

- 1 Show the calculation steps and explain the equations you use
- 2 Give references if values are found online or in tables

Evaluation

- 1 Check the units of your solution
- 2 Is the answer realistic/expected?
- 3 Did you answer all the questions asked?
- 4 Iterate if this is required

Assignment 3

A aluminum conductor transmission line carries an electric current of 100 A, and has a resistance of 0.004 ohm per meter length. A transverse wind flow is passing the lines at a velocity of 40 km/h.

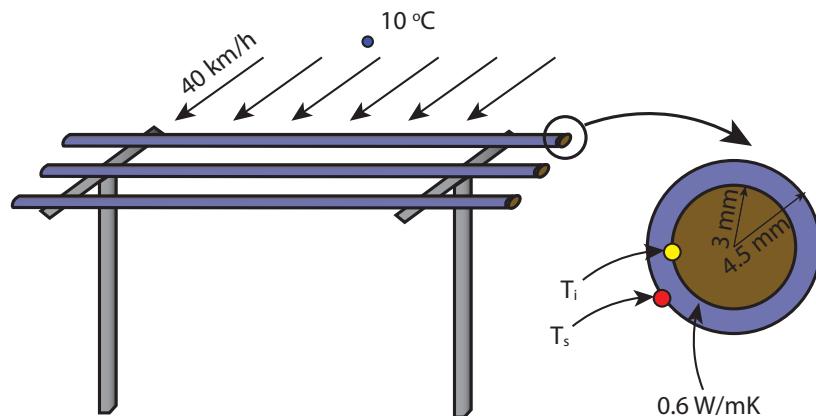


Figure 8: Transmission lines subjected to convection

- Determine the Reynolds number. Clearly state on the assumptions that are made.
- Determine the surface temperature T_s . Evaluate whether the assumptions made in question a) were correct. What can be done to increase the quality of the results?
- What will happen to the flow if the flow temperature increases? Assume T_s to remain constant.
- Determine the interface temperature T_i
- Provide a sketch of the temperature profile in radial direction of The transmission line.
Note: Also include the temperature profile outside the transmission line.

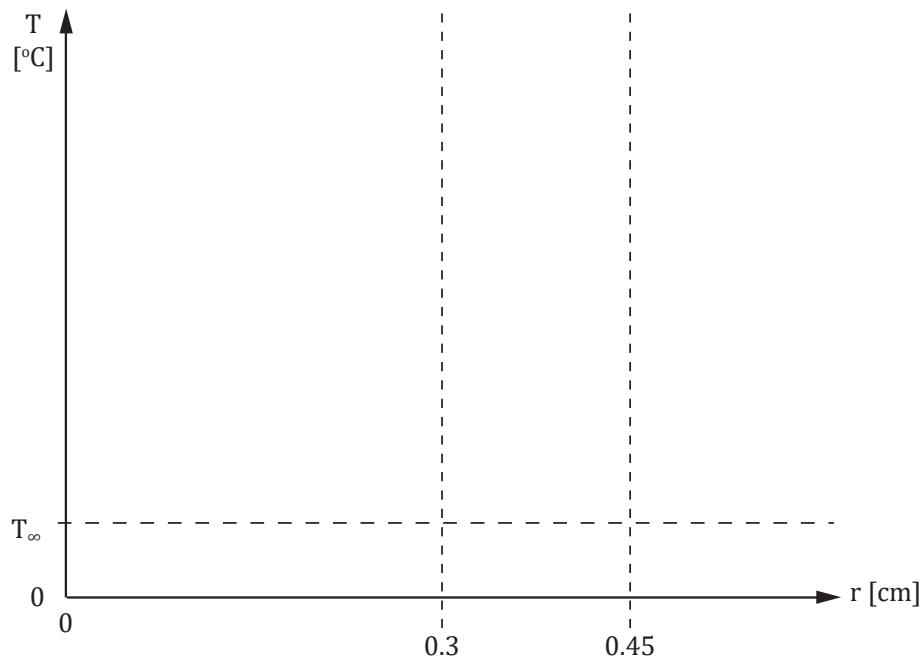


Figure 9: Transmission lines subjected to convection

Temperature °C	Density kg/m ³	Specific heat J/kgK	Thermal conductivity W/m · K	Thermal diffusivity m ² /s	Dynamic viscosity kg/m · s	Kinematic viscosity m ² /s	Prandtl number
0	1.292	1006	0.02364	$1.818 \cdot 10^{-5}$	$1.729 \cdot 10^{-5}$	$1.338 \cdot 10^{-5}$	0.7362
10	1.246	1006	0.02439	$1.944 \cdot 10^{-5}$	$1.778 \cdot 10^{-5}$	$1.426 \cdot 10^{-5}$	0.7336
20	1.204	1007	0.02514	$2.074 \cdot 10^{-5}$	$1.825 \cdot 10^{-5}$	$1.516 \cdot 10^{-5}$	0.7309
30	1.164	1007	0.02588	$2.208 \cdot 10^{-5}$	$1.872 \cdot 10^{-5}$	$1.608 \cdot 10^{-5}$	0.7282
40	1.127	1007	0.02662	$2.346 \cdot 10^{-5}$	$1.918 \cdot 10^{-5}$	$1.702 \cdot 10^{-5}$	0.7255

Table 2: Air properties at 1 atm pressure