

# Materials: Elasticity Theory

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University of Twente

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# Introduction

## Elasticitstheorie – Elasticity Theory module 6: 2024-202000131-1B

script & sheets, based on:

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## Overview:

Index notation, matrix-vector notation, vectors and tensors

Stress, strain, (material behavior: elasticity, plasticity, viscosity)

## 1) Index notation and summation convention

Set of quantities:  $a_1, a_2, a_3, \dots, a_n \longrightarrow$   
 $a_i$ , with  $i = 1, 2, 3, \dots, n$

In 2- or 3-dim. space  $a_i$  is the  $i^{\text{th}}$  component  
of the vector  $\underline{a}$ , with  $n = 2$  or  $3$

also:  $a_i$  is the  $i^{\text{th}}$  element of the (column)-vector  $\{a\}$   
(just a column of numbers?)

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(just a column of numbers?)

### Difference:

vector as geometric, physical quantity  
(e.g. distance or velocity with norm [unit] and direction)

$\underline{a} = \vec{a}$  - or vector as matrix with one column  $\{a\}$

*summation convention*

inner product (scalar product) of two vectors:

$$\{x\}^T \{y\} = x_1 y_1 + x_2 y_2 + x_3 y_3 + \cdots + x_n y_n = \sum_{i=1}^n x_i y_i$$

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shorter (Einstein) notation:

$$\{x\}^T \{y\} = (\underline{x} \cdot \underline{y} =) x_i y_i = x_p y_p = x_q y_q$$

where  $i$ ,  $p$  and  $q$  are so-called *dummy-indices*.

Those are double, in products, and have to be summed over.

So called *free indices* occur only once in product-terms.

System of equations:

$$y_1 = A_{11} x_1 + A_{12} x_2 + \cdots + A_{1n} x_n$$

$$y_2 = A_{21} x_1 + A_{22} x_2 + \cdots + A_{2n} x_n$$

$$\vdots \quad \vdots \quad \vdots \quad \vdots$$

$$y_m = A_{m1} x_1 + A_{m2} x_2 + \cdots + A_{mn} x_n$$

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The  $i^{\text{th}}$  equation is:

$$y_i = A_{i1} x_1 + A_{i2} x_2 + \cdots + A_{in} x_n = \left( \sum_{j=1}^n A_{ij} x_j \right)$$

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$$\begin{aligned}y_1 &= A_{11} x_1 + A_{12} x_2 + \cdots + A_{1n} x_n \\y_2 &= A_{21} x_1 + A_{22} x_2 + \cdots + A_{2n} x_n \\&\vdots \quad \vdots \quad \vdots \quad \vdots \\y_m &= A_{m1} x_1 + A_{m2} x_2 + \cdots + A_{mn} x_n\end{aligned}$$

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shortly:  $y_i = A_{ij} x_j \quad \begin{cases} i = 1, 2, \dots, m \\ j = 1, 2, \dots, n \end{cases}$

or  $\{y\} = [A] \{x\}$

or  $\underline{y} = \underline{\underline{A}} \cdot \underline{x}$

Matrix  $[A]$  with entries  $A_{ij}$

(where  $i$  is the row-index, and  $j$  is the column-index)

Example:  $S = \{x\}^T \cdot \{x\} = \left( \sum_{i=1}^n \right) x_i x_i$

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Kronecker delta:  $\delta_{ij} = \begin{cases} 0 & \text{if } i \neq j \\ 1 & \text{if } i = j \end{cases}$

Thus:  $S = x_i x_i = \delta_{ij} x_i x_j = \{x\}^T [I] \{x\}$

Also:  $x_i = \delta_{ij} x_j = \delta_{ip} x_p$

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Also:  $x_i = \delta_{ij} x_j = \delta_{ip} x_p$

BUT:  $\delta_{ii} \neq 1$

instead, note:  $\delta_{ii} = \delta_{11} + \delta_{22} + \cdots + \delta_{nn} = 1 + 1 + \cdots + 1 = n$

## Differentiation:

Consider a function:  $F(x_1, x_2, \dots, x_n)$

Partial derivative:  $\frac{\partial F}{\partial x_i} = F_{,i}$

The index  $, i$  means the partial derivative with respect to  $x_i$ !

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Total differential of  $F$ :

$$dF = \frac{\partial F}{\partial x_1} dx_1 + \frac{\partial F}{\partial x_2} dx_2 + \dots + \frac{\partial F}{\partial x_n} dx_n = \left( \sum_{i=1}^n \right) \frac{\partial F}{\partial x_i} dx_i = F_{,i} dx_i$$

## 2) Scalars, vectors and Cartesian tensors

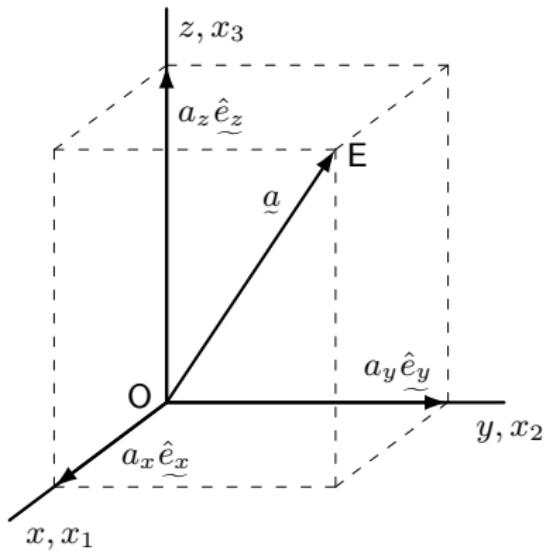
Scalar fields, functions of position, only a number with [unit].  
e.g. density, temperature

## 2) Scalars, vectors and Cartesian tensors

Scalar fields, functions of position, only a number with [unit].  
e.g., density, temperature, ...

Vector fields, functions of position, quantity with norm [unit]  
and a direction, e.g., force, displacement, velocity, acceleration, ...

## Cartesian coordinate system



$$\underline{a} = a_x \hat{e}_x + a_y \hat{e}_y + a_z \hat{e}_z$$

$$= a_1 \hat{e}_1 + a_2 \hat{e}_2 + a_3 \hat{e}_3$$

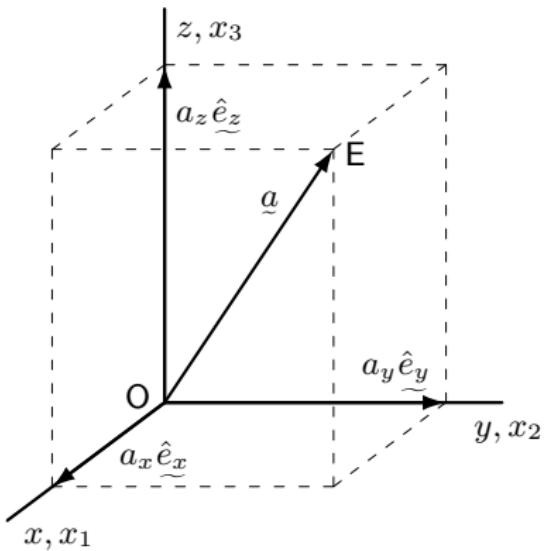
$$= \sum_{i=1}^3 a_i \hat{e}_i = a_i \hat{e}_i$$

... alternatives ...

$$= a_1 \hat{x} + a_2 \hat{y} + a_3 \hat{z}$$

$$= a_1 \hat{i} + a_2 \hat{j} + a_3 \hat{k}$$

## Cartesian coordinate system



$$\begin{aligned}\underline{a} &= a_x \hat{e}_x + a_y \hat{e}_y + a_z \hat{e}_z \\ &= a_1 \hat{e}_1 + a_2 \hat{e}_2 + a_3 \hat{e}_3 \\ &= \sum_{i=1}^3 a_i \hat{e}_i = a_i \hat{e}_i\end{aligned}$$

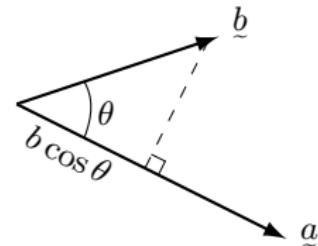
Norm (length) of  $\underline{a}$ :

$$|\underline{a}| = a = \sqrt{a_1^2 + a_2^2 + a_3^2} = \sqrt{a_i a_i}$$

## inner product

(scalar product, “dot”-product):

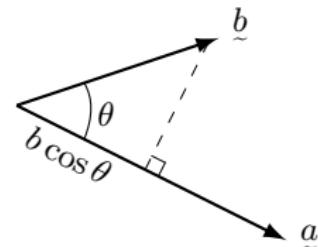
$$\underline{a} \cdot \underline{b} = a b \cos \theta$$



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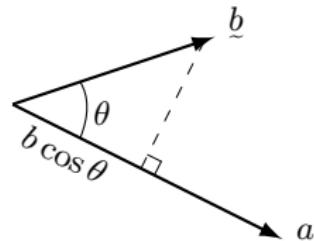
Also:

$$\underline{a} \cdot \underline{b} = a_x b_x + a_y b_y + a_z b_z = a_1 b_1 + a_2 b_2 + a_3 b_3 = a_i b_i$$

## inner product

(scalar product, “dot”-product):

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Also:

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For unit-direction-vectors:

$$\hat{\underline{e}}_x \cdot \hat{\underline{e}}_x = \dots = 1 \quad \text{or} \quad \hat{\underline{e}}_1 \cdot \hat{\underline{e}}_1 = \hat{\underline{e}}_2 \cdot \hat{\underline{e}}_2 = \hat{\underline{e}}_3 \cdot \hat{\underline{e}}_3 = 1$$

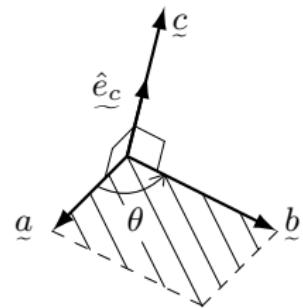
$$\hat{\underline{e}}_x \cdot \hat{\underline{e}}_y = \dots = 0 \quad \text{or} \quad \hat{\underline{e}}_1 \cdot \hat{\underline{e}}_2 = \hat{\underline{e}}_2 \cdot \hat{\underline{e}}_3 = \hat{\underline{e}}_3 \cdot \hat{\underline{e}}_1 = 0$$

$$\Rightarrow \hat{\underline{e}}_i \cdot \hat{\underline{e}}_j = \delta_{ij} \quad \text{in a way that e.g.:}$$

$$\underline{a} \cdot \underline{b} = a_i \hat{\underline{e}}_i \cdot b_j \hat{\underline{e}}_j = a_i b_j \hat{\underline{e}}_i \cdot \hat{\underline{e}}_j = a_i b_j \delta_{ij} = a_i b_i$$

Outer product or vector-product  
(“cross”-product)

$$\underline{a} * \underline{b} = (|\underline{a}| |\underline{b}| \sin \theta) \hat{e}_c = \underline{c}$$



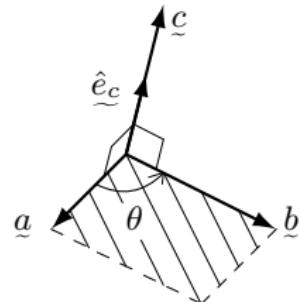
## Outer product or vector-product

(“cross”-product)

$$\underline{a} * \underline{b} = \underline{a} \times \underline{b} = (|\underline{a}| |\underline{b}| \sin \theta) \hat{\underline{e}}_c = \underline{c}$$

Also:

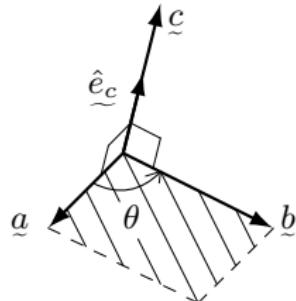
$$\begin{aligned}\underline{a} * \underline{b} &= \det \begin{pmatrix} \hat{\underline{e}}_1 & \hat{\underline{e}}_2 & \hat{\underline{e}}_3 \\ a_1 & a_2 & a_3 \\ b_1 & b_2 & b_3 \end{pmatrix} = \\ &= (a_2 b_3 - a_3 b_2) \hat{\underline{e}}_1 + (a_3 b_1 - a_1 b_3) \hat{\underline{e}}_2 + (a_1 b_2 - a_2 b_1) \hat{\underline{e}}_3 = \\ &= c_1 \hat{\underline{e}}_1 + c_2 \hat{\underline{e}}_2 + c_3 \hat{\underline{e}}_3 = c_i \hat{\underline{e}}_i = \underline{c}\end{aligned}$$



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Also:

$$\begin{aligned}
 \underline{a} * \underline{b} &= \det \left( \begin{bmatrix} \hat{\underline{e}}_1 & \hat{\underline{e}}_2 & \hat{\underline{e}}_3 \\ a_1 & a_2 & a_3 \\ b_1 & b_2 & b_3 \end{bmatrix} \right) = \\
 &= (a_2 b_3 - a_3 b_2) \hat{\underline{e}}_1 + (a_3 b_1 - a_1 b_3) \hat{\underline{e}}_2 + (a_1 b_2 - a_2 b_1) \hat{\underline{e}}_3 = \\
 &= c_1 \hat{\underline{e}}_1 + c_2 \hat{\underline{e}}_2 + c_3 \hat{\underline{e}}_3 = c_i \hat{\underline{e}}_i = \underline{c}
 \end{aligned}$$

permutation symbol:  $c_i = \varepsilon_{ijk} a_j b_k$  ( $\underline{c} = \varepsilon_{ijk} a_j b_k \hat{\underline{e}}_i$ ) with:

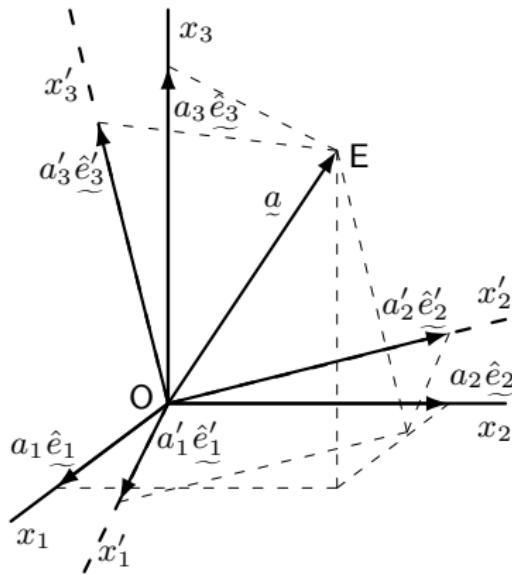
$$\varepsilon_{123} = \varepsilon_{231} = \varepsilon_{312} = 1 ; \quad \varepsilon_{213} = \varepsilon_{321} = \varepsilon_{132} = -1 ; \quad \text{note: } \varepsilon_{ijk} = 0$$

*Rotation of axes:* (Cartesian coordinates)

Rotations from “old”  $x_i$ -coordinates  
to “new”  $x'_p$ -coordinates.

For an arbitrary vector holds:

$$\underline{a} = a_1 \hat{e}_1 + a_2 \hat{e}_2 + a_3 \hat{e}_3 = \\ = a'_1 \widetilde{\hat{e}'_1} + a'_2 \widetilde{\hat{e}'_2} + a'_3 \widetilde{\hat{e}'_3}$$



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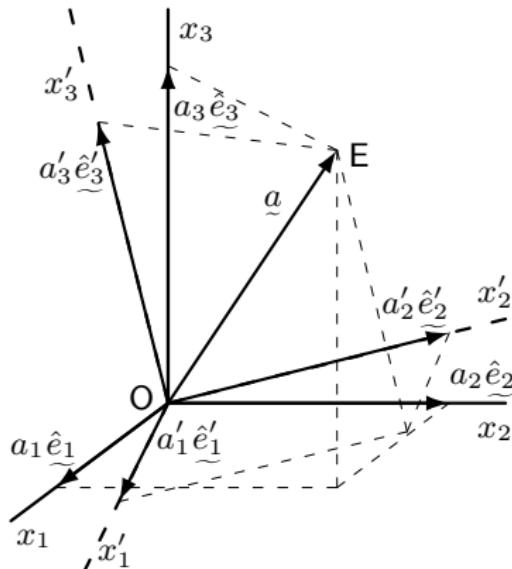
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In index notation:  $\underline{\underline{a}} = a_i \hat{e}_i = a'_p \hat{e}'_p$

Direction-cosines:

$$R_{pi} = \cos(x'_p, x_i) \Rightarrow$$

$$\begin{aligned}a'_1 &= a_1 \cos(x'_1, x_1) + a_2 \cos(x'_1, x_2) + a_3 \cos(x'_1, x_3) = \\ &= a_i \cos(x'_1, x_i) = R_{1i} a_i \quad (\text{change now } 1 \rightarrow p)\end{aligned}$$



For  $p = 1, 2, 3$ :

$$\begin{aligned} a'_p &= a_1 \cos(x'_p, x_1) + a_2 \cos(x'_p, x_2) + a_3 \cos(x'_p, x_3) = \\ &= a_i \cos(x'_p, x_i) = R_{pi} a_i \end{aligned}$$

Summation over *second* index of  $R$ .

For  $p = 1, 2, 3$ :

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Summation over *second* index of  $R$ .

Inverse relation, from “new” to “old”:

$$\begin{aligned} a_j &= a'_1 \cos(x_j, x'_1) + a'_2 \cos(x_j, x'_2) + a'_3 \cos(x_j, x'_3) = \\ &= a'_q \cos(x_j, x'_q) = R_{qj} a'_q \end{aligned}$$

*Note*: summation over the *first* index of  $R$ !

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*Note*: summation over the *first* index of  $R$ !

If this transformation rule holds, it is a vector (line with a length and direction): a so-called first-order-tensor!

Written in matrix notation  $a'_p = R_{pi} a_i$  becomes :

$$\begin{bmatrix} a'_1 \\ a'_2 \\ a'_3 \end{bmatrix} = \begin{bmatrix} R_{11} & R_{12} & R_{13} \\ R_{21} & R_{22} & R_{23} \\ R_{31} & R_{32} & R_{33} \end{bmatrix} \begin{bmatrix} a_1 \\ a_2 \\ a_3 \end{bmatrix}$$

In *matrix-vector-notation* :  
 $\{a'\} = [R]\{a\}$

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The inverse transform  $a_j = R_{qj} a'_q$  in matrix notation:

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In *matrix-vector-notation* :  
 $\{a\} = [R]^T\{a'\}$

With  $[R]^T$  being the transposed of  $[R]$ . Matrix mirroring with respect to the main diagonal (top-left  $\rightarrow$  bottom-right).

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Properties of the (transformation matrix) rotation matrix  $[R]$ .

Substitute  $\{a\} = [R]^T\{a'\}$  in  $\{a'\} = [R]\{a\}$ . This gives:

$$\{a'\} = [R]\{a\} = [R][R]^T\{a'\} \implies [R][R]^T = [I]$$

Also:  $[R][R]^{-1} = [I]$  (definition of inverted matrix), so:

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$$\begin{bmatrix} R_{11} & R_{12} & R_{13} \\ R_{21} & R_{22} & R_{23} \\ R_{31} & R_{32} & R_{33} \end{bmatrix} \begin{bmatrix} R_{11} & R_{21} & R_{31} \\ R_{12} & R_{22} & R_{32} \\ R_{13} & R_{23} & R_{33} \end{bmatrix} = \begin{bmatrix} 1 & 0 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 1 \end{bmatrix}$$

$\Rightarrow$  *Normality condition* and *orthogonality condition*

Orthogonality rotation matrix  $[R]$  using index notation:

Derived before:  $a'_p = R_{pi} a_i$  and  $a_j = R_{qj} a'_q$

So it holds:  $a'_p = R_{pi} a_i = R_{pi} R_{qi} a'_q$  also:  $a'_p = \delta_{pq} a'_q$

From which follows:  $R_{pi} R_{qi} = \delta_{pq}$  or  $R_{ip} R_{jp} = \delta_{ij}$

Equivalent with:  $[R] [R]^T = [I]$

(Note:  $[A] [B] = [C] \iff A_{ip} B_{pj} = C_{ij}$ )

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Reverse:  $a_j = R_{qj} a'_q = R_{qj} R_{qi} a_i$  also:  $a_j = \delta_{ij} a_i$

From which follows:  $R_{qi} R_{qj} = \delta_{ij}$  or  $R_{pi} R_{pj} = \delta_{ij}$

Equivalent with:  $[R]^T [R] = [I]$

*Second order tensor*

Consider 2 vectors,  $\underline{a}$  and  $\underline{b}$  with components  $a_i$  and  $b_j$

Dyadic product (per definition):     $\underline{S} = \underline{a} \underline{b}$     with     $S_{ij} = a_i b_j$

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Dyadic product (per definition):  $\underline{S} = \underline{a} \underline{b}$  with  $S_{ij} = a_i b_j$

Rotation of axis:  $S'_{ij} = a'_i b'_j$

Substitute transformation rule for vectors:

$$S'_{ij} = a'_i b'_j = R_{ip} a_p R_{jq} b_q = R_{ip} R_{jq} a_p b_q = R_{ip} R_{jq} S_{pq}$$

### *Second order tensor*

Consider 2 vectors,  $\underline{a}$  and  $\underline{b}$  with components  $a_i$  and  $b_j$

Dyadic product (per definition):  $\underline{S} = \underline{a} \underline{b}$  with  $S_{ij} = a_i b_j$

Rotation of axis:  $S'_{ij} = a'_i b'_j$

Substitute transformation rule for vectors:

$$S'_{ij} = a'_i b'_j = R_{ip} a_p R_{jq} b_q = R_{ip} R_{jq} a_p b_q = R_{ip} R_{jq} S_{pq}$$

Quantities that satisfy this transformation rule are called *second order tensors*. NOTE, not every second order tensor can be written as a dyadic product of two vectors.

Second order tensor:  $S'_{ij} = R_{ip} R_{jq} S_{pq}$

Can be written as:  $S'_{ij} = R_{ip} S_{pq} R_{jq} = R_{ip} S_{pq} R_{qj}^T$

In matrix-vector-notation:  $[S'] = [R] [S] [R]^T$

Second order tensor:  $S'_{ij} = R_{ip} R_{jq} S_{pq}$

Can be written as:  $S'_{ij} = R_{ip} S_{pq} R_{jq} = R_{ip} S_{pq} R_{qj}^T$

In matrix-vector-notation:  $[S'] = [R] [S] [R]^T$

Inverse:  $S_{ij} = R_{pi} R_{qj} S'_{pq} = R_{ip}^T S'_{pq} R_{qj}$

In matrix-vector-notation:  $[S] = [R]^T [S'] [R]$

Overview tensors ( $\Rightarrow$  means transforms to)

- 0. If  $\Phi(x_r) \Rightarrow \Phi'(x_r)$  by  $\Phi = \Phi'$   
then *zero order tensor field or scalar field*

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2. If  $S_{ij}(x_r) \Rightarrow S'_{pq}(x_r)$  by  $S'_{pq} = R_{pi} R_{qj} S_{ij}$

then *second order tensor field*

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0. If  $\Phi(x_r) \Rightarrow \Phi'(x_r)$  by  $\Phi = \Phi'$

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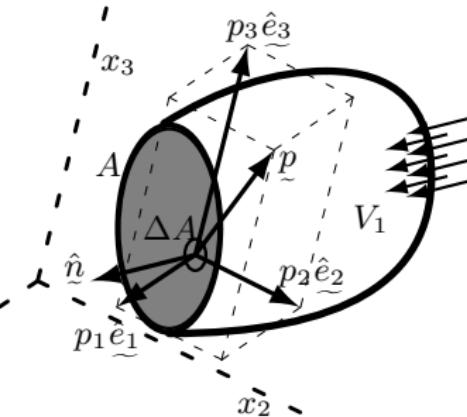
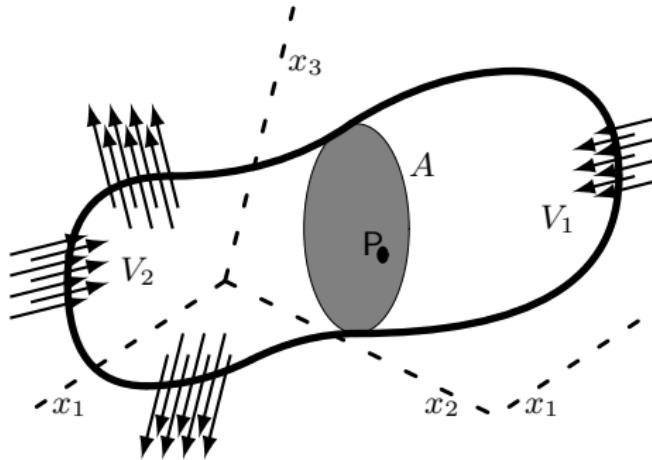
3. If  $S_{ijk}(x_r) \Rightarrow S'_{pqt}(x_r)$  by  $S'_{pqt} = R_{pi} R_{qj} R_{tk} S_{ijk}$

then *third order tensor field*

4. etc. etc. etc.

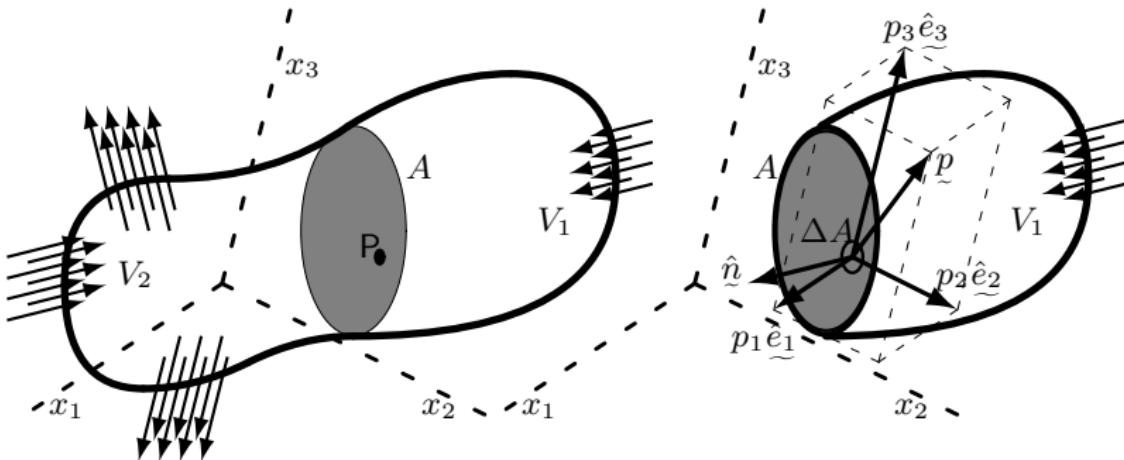
### 3) Stresses

#### 3.1) Stress state in a point



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Per definition, stress vector  $\tilde{p} = \lim_{\Delta A \rightarrow 0} \left( \frac{\Delta F}{\Delta \tilde{A}} \right)$

A few remarks:

- ▶ A stress vector  $\underline{p}$  has in general an arbitrary orientation

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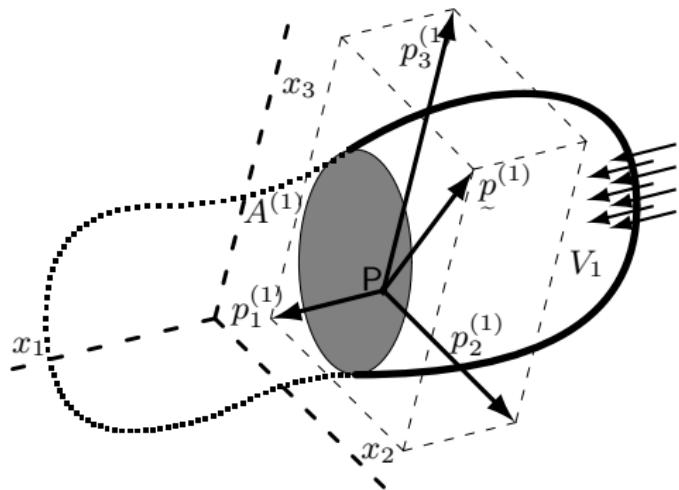
- ▶ A stress vector  $\underline{\underline{p}}$  has in general an arbitrary orientation
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- ▶ A stress vector  $\underline{\underline{p}}$  can be decomposed in a normal and a tangential component w.r.t. the orientation of  $\Delta A$
- ▶ A stress vector  $\underline{\underline{p}}$  can be decomposed along a Cartesian coordinate system     $\underline{\underline{p}} = p_i \underline{\hat{e}_i}$

Cartesian  
stress- components  $\sigma_{ij}$ :

$$\sigma_{11} = p_1^{(1)}$$

$$\sigma_{12} = p_2^{(1)}$$

$$\sigma_{13} = p_3^{(1)}$$



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$$\sigma_{11} = p_1^{(1)}$$

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Cross-section

perpendicular to  $x_2$ -axis:

$$\sigma_{21} = p_1^{(2)}$$

$$\sigma_{22} = p_2^{(2)}$$

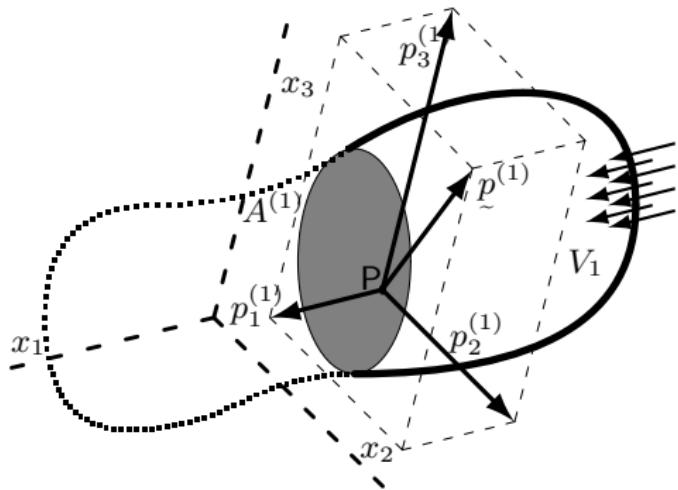
$$\sigma_{23} = p_3^{(2)}$$

Cross-section perpendicular to  $x_3$ -axis:

$$\sigma_{31} = p_1^{(3)}$$

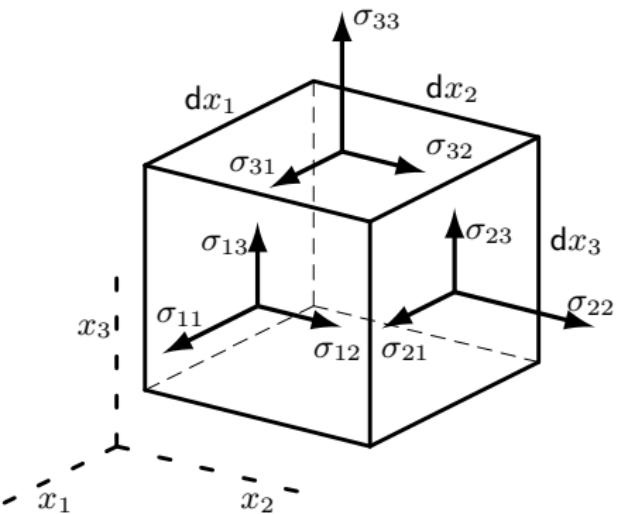
$$\sigma_{32} = p_2^{(3)}$$

$$\sigma_{33} = p_3^{(3)}$$



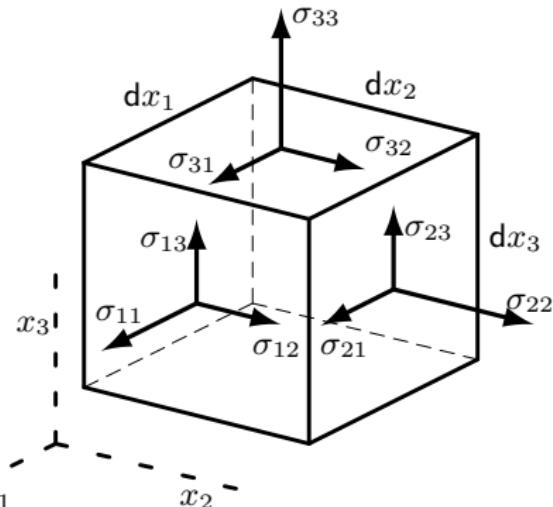
Where

$\sigma_{ij}$  is the stress component in the directions of  $x_j$  acting on a surface with its normal pointing in  $x_i$ -direction.



Where

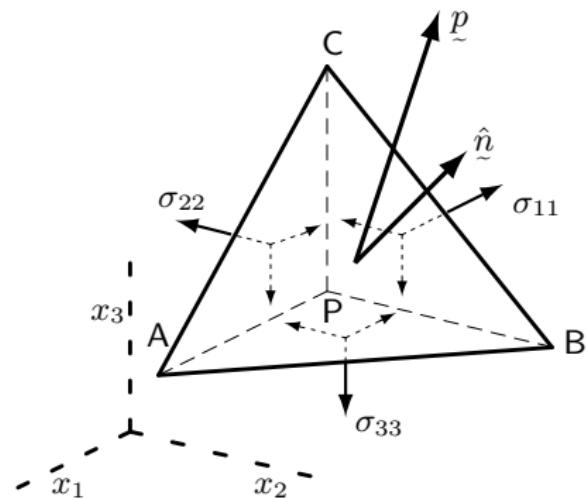
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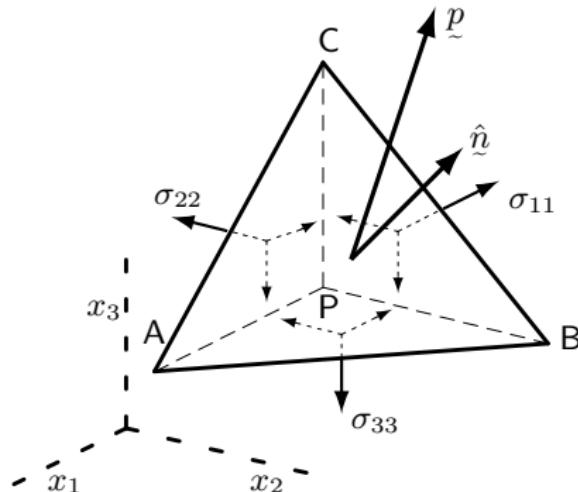
Displayed in a stress matrix:

$$[\sigma] = \begin{bmatrix} \sigma_{11} & \sigma_{12} & \sigma_{13} \\ \sigma_{21} & \sigma_{22} & \sigma_{23} \\ \sigma_{31} & \sigma_{32} & \sigma_{33} \end{bmatrix} = \begin{bmatrix} \sigma_{xx} & \sigma_{xy} & \sigma_{xz} \\ \sigma_{yx} & \sigma_{yy} & \sigma_{yz} \\ \sigma_{zx} & \sigma_{zy} & \sigma_{zz} \end{bmatrix} = \begin{bmatrix} \sigma_x & \tau_{xy} & \tau_{xz} \\ \tau_{yx} & \sigma_y & \tau_{yz} \\ \tau_{zx} & \tau_{zy} & \sigma_z \end{bmatrix}$$

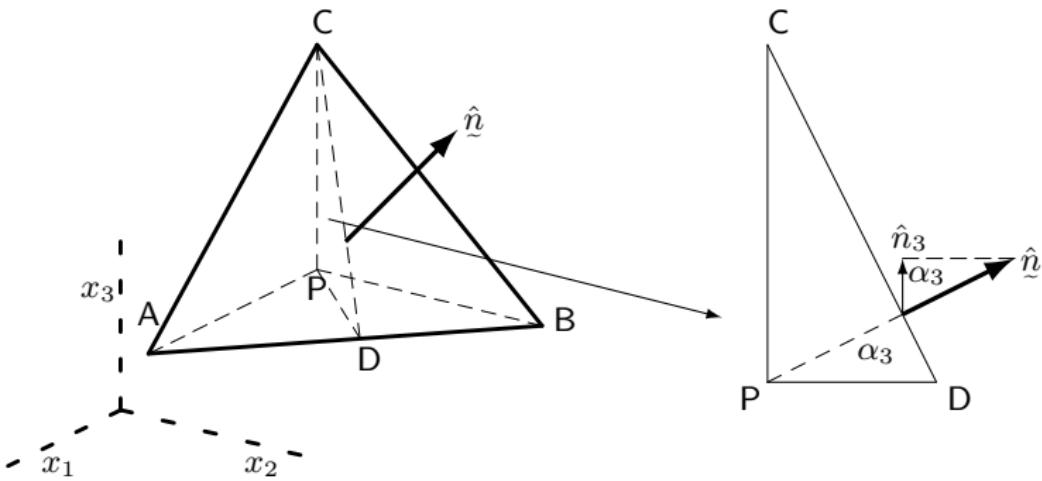
### 3.2) Stress-vector on arbitrary cross-section

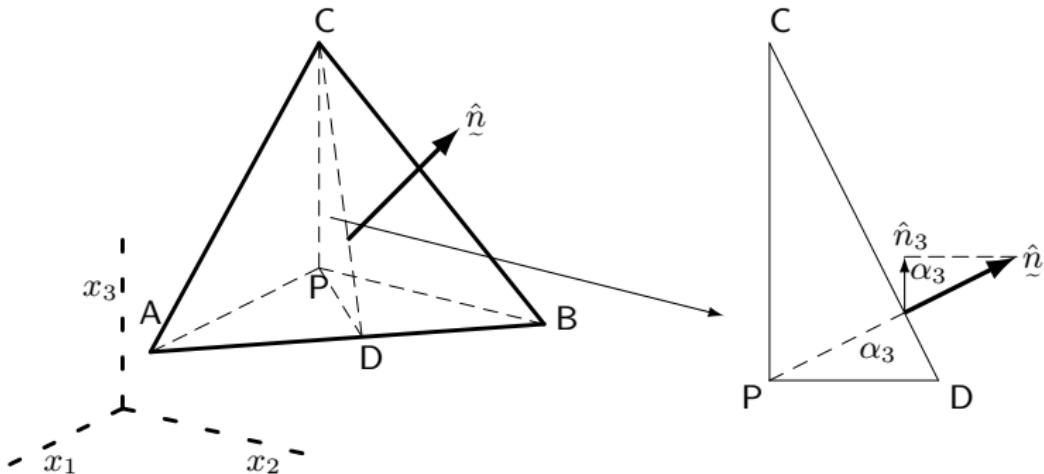


### 3.2) Stress-vector on arbitrary cross-section



Equilibrium:  $p_1 \Delta A_{ABC} - \sigma_{11} \Delta A_{PBC} - \sigma_{21} \Delta A_{PCA} - \sigma_{31} \Delta A_{PAB} = 0$   
 $\Rightarrow p_1 = \sigma_{11} \frac{\Delta A_{PBC}}{\Delta A_{ABC}} + \sigma_{21} \frac{\Delta A_{PCA}}{\Delta A_{ABC}} + \sigma_{31} \frac{\Delta A_{PAB}}{\Delta A_{ABC}}$





Now it holds:

$$\frac{\Delta A_{PAB}}{\Delta A_{ABC}} = \frac{\frac{1}{2} |PD| |AB|}{\frac{1}{2} |CD| |AB|} = \frac{|PD|}{|CD|} = \cos \alpha_3 = \frac{\hat{n}_3}{|\hat{n}|} = \hat{n}_3 \quad (|\hat{n}| = 1)$$

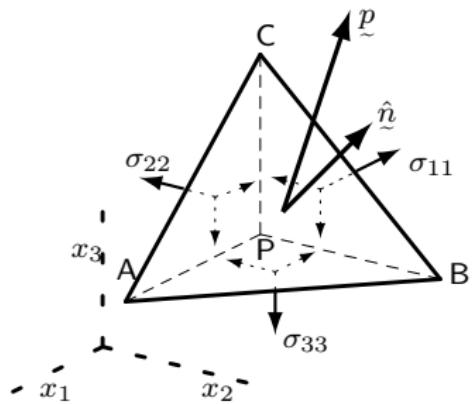
And similar for the other two area ratios.

Combined this gives:

$$\frac{\Delta A_{PBC}}{\Delta A_{ABC}} = \cos(\hat{n}, x_1) = \hat{n}_1$$

$$\frac{\Delta A_{PCA}}{\Delta A_{ABC}} = \cos(\hat{n}, x_2) = \hat{n}_2$$

$$\frac{\Delta A_{PAB}}{\Delta A_{ABC}} = \cos(\hat{n}, x_3) = \hat{n}_3$$

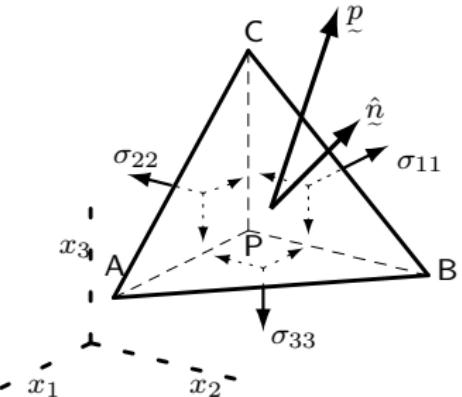


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$$\frac{\Delta A_{PAB}}{\Delta A_{ABC}} = \cos(\hat{n}, x_3) = \hat{n}_3$$



Equilibrium in  $x_i$ -direction ( $i = 1, 2, 3$ ):

$$p_1 \Delta A = \sigma_{11} \Delta A_{PBC} + \sigma_{21} \Delta A_{PCA} + \sigma_{31} \Delta A_{PAB}$$

$$\Rightarrow p_1 = \sigma_{11} \hat{n}_1 + \sigma_{21} \hat{n}_2 + \sigma_{31} \hat{n}_3 = \sigma_{j1} \hat{n}_j$$

Equilibrium in  $x_i$ -direction:

$$\Rightarrow p_i = \sigma_{1i} \hat{n}_1 + \sigma_{2i} \hat{n}_2 + \sigma_{3i} \hat{n}_3 = \sigma_{ji} \hat{n}_j$$

*Cauchy's formula:*  $p_i = \sigma_{ji} \hat{n}_j$ . The summation takes place over the index  $j$ , and index  $i$  can be either 1, 2 or 3.

$$\underline{p} = \underline{\hat{n}} \cdot \underline{\sigma} = \underline{\sigma}^T \cdot \underline{\hat{n}} \quad \text{or} \quad \{p\} = \{\hat{n}\}^T [\sigma] = [\sigma]^T \{\hat{n}\}$$

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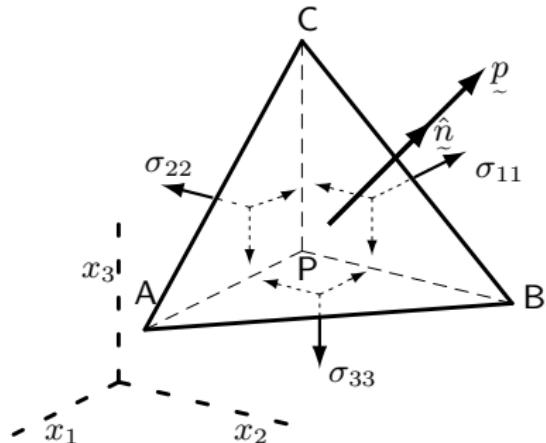
It will be shown later that the stress matrix is symmetric, so  $\sigma_{ij} = \sigma_{ji}$  or  $\underline{\sigma} = \underline{\sigma}^T$  or  $[\sigma] = [\sigma]^T$ . From that:

$$\begin{aligned} p_i &= \sigma_{ij} \hat{n}_j \\ \underline{p} &= \underline{\sigma} \cdot \underline{\hat{n}} \quad \text{or} \quad \begin{bmatrix} p_1 \\ p_2 \\ p_3 \end{bmatrix} = \begin{bmatrix} \sigma_{11} & \sigma_{12} & \sigma_{13} \\ \sigma_{21} & \sigma_{22} & \sigma_{23} \\ \sigma_{31} & \sigma_{32} & \sigma_{33} \end{bmatrix} \begin{bmatrix} \hat{n}_1 \\ \hat{n}_2 \\ \hat{n}_3 \end{bmatrix} \\ \{p\} &= [\sigma] \{\hat{n}\} \end{aligned}$$

### 3.3) Principal stresses and principal directions

Question:

Is it possible to orientate the area ABC in such a way that the direction of the stress vector is the same as the direction of the normal acting on this area.



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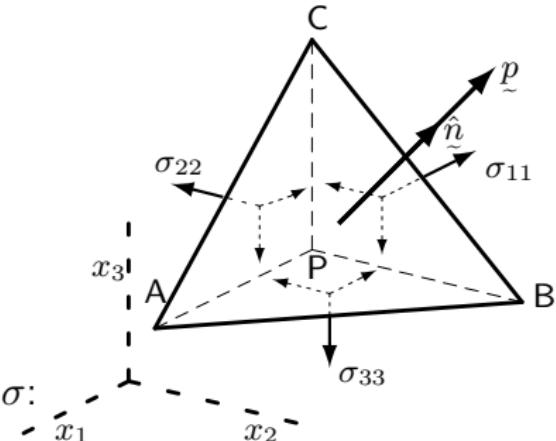
Is it possible to orientate the area ABC in such a way that the direction of the stress vector is the same as the direction of the normal acting on this area.

In that case it must hold that:

$$\tilde{p} = \text{factor} \times \hat{n}.$$

Substitute “factor” with the symbol  $\sigma$ :

$$\tilde{p} \stackrel{?}{=} \sigma \hat{n} \quad \text{or} \quad p_i \stackrel{?}{=} \sigma \hat{n}_i$$



Are there multiple orientations of area ABC possible such that:

$$\underline{\underline{p}} \stackrel{?}{=} \sigma \hat{\underline{\underline{n}}} \quad \text{or} \quad p_i \stackrel{?}{=} \sigma \hat{n}_i$$

Cauchy's formula gives:  $p_i = \sigma_{ij} \hat{n}_j$

Combined:  $p_i = \sigma_{ij} \hat{n}_j \stackrel{?}{=} \sigma \hat{n}_i = \sigma \delta_{ij} \hat{n}_j$

Written as a matrix ( $[\sigma]\{\hat{n}\} = \sigma\{\hat{n}\}$ ):

$$\{p\} = \begin{bmatrix} \sigma_{11} & \sigma_{12} & \sigma_{13} \\ \sigma_{21} & \sigma_{22} & \sigma_{23} \\ \sigma_{31} & \sigma_{32} & \sigma_{33} \end{bmatrix} \begin{bmatrix} \hat{n}_1 \\ \hat{n}_2 \\ \hat{n}_3 \end{bmatrix} \stackrel{?}{=} \sigma \begin{bmatrix} \hat{n}_1 \\ \hat{n}_2 \\ \hat{n}_3 \end{bmatrix}$$

$[\sigma]\{\hat{n}\} = \sigma\{\hat{n}\}$  can be written as:  $([\sigma] - \sigma[I])\{\hat{n}\} = \{0\}$

$$\begin{bmatrix} (\sigma_{11} - \sigma) & \sigma_{12} & \sigma_{13} \\ \sigma_{21} & (\sigma_{22} - \sigma) & \sigma_{23} \\ \sigma_{31} & \sigma_{32} & (\sigma_{33} - \sigma) \end{bmatrix} \begin{bmatrix} \hat{n}_1 \\ \hat{n}_2 \\ \hat{n}_3 \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \\ 0 \end{bmatrix}$$

This is a homogeneous system of algebraic equations.

$[\sigma]\{\hat{n}\} = \sigma\{\hat{n}\}$  can be written as:  $([\sigma] - \sigma[I])\{\hat{n}\} = \{0\}$

$$\begin{bmatrix} (\sigma_{11} - \sigma) & \sigma_{12} & \sigma_{13} \\ \sigma_{21} & (\sigma_{22} - \sigma) & \sigma_{23} \\ \sigma_{31} & \sigma_{32} & (\sigma_{33} - \sigma) \end{bmatrix} \begin{bmatrix} \hat{n}_1 \\ \hat{n}_2 \\ \hat{n}_3 \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \\ 0 \end{bmatrix}$$

This is a homogeneous system of algebraic equations.

This also yields a solution  $\{\hat{n}\} \neq \{0\}$  when the determinant of the coefficient matrix is equal to zero.

$$\det \left( \begin{bmatrix} (\sigma_{11} - \sigma) & \sigma_{12} & \sigma_{13} \\ \sigma_{21} & (\sigma_{22} - \sigma) & \sigma_{23} \\ \sigma_{31} & \sigma_{32} & (\sigma_{33} - \sigma) \end{bmatrix} \right) = 0$$

$$\det \begin{pmatrix} (\sigma_{11} - \sigma) & \sigma_{12} & \sigma_{13} \\ \sigma_{21} & (\sigma_{22} - \sigma) & \sigma_{23} \\ \sigma_{31} & \sigma_{32} & (\sigma_{33} - \sigma) \end{pmatrix} = 0$$

Written as *characteristic equation*:

$$\boxed{\sigma^3 - I_1 \sigma^2 + I_2 \sigma - I_3 = 0} \quad \text{with}$$

$$I_1 = \sigma_{11} + \sigma_{22} + \sigma_{33} = \sigma_{ii}$$

$$I_2 = \sigma_{11} \sigma_{22} - \sigma_{12} \sigma_{21} + \dots = \frac{1}{2} (\sigma_{ii} \sigma_{jj} - \sigma_{ij} \sigma_{ji})$$

$$I_3 = \det([\sigma])$$

From this the *principal stresses*:  $\sigma_1$ ,  $\sigma_2$  and  $\sigma_3$  are obtained for which holds that:  $\sigma_1 \geq \sigma_2 \geq \sigma_3$ .

Principal stresses are the *eigenvalues* of the stress matrix  $[\sigma]$ .

With eigenvalues comes *eigenvectors*. Here the *principal directions*.

Obtain first principal directions, related to  $\sigma_1$ :

$$\begin{bmatrix} (\sigma_{11} - \sigma_1) & \sigma_{12} & \sigma_{13} \\ \sigma_{21} & (\sigma_{22} - \sigma_1) & \sigma_{23} \\ \sigma_{31} & \sigma_{32} & (\sigma_{33} - \sigma_1) \end{bmatrix} \begin{bmatrix} \hat{n}_1 \\ \hat{n}_2 \\ \hat{n}_3 \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \\ 0 \end{bmatrix}$$

This is a *dependent* system of equations!!! e.g.:

$$\begin{bmatrix} (\sigma_{11} - \sigma_1) & \sigma_{12} \\ \sigma_{21} & (\sigma_{22} - \sigma_1) \end{bmatrix} \begin{bmatrix} \hat{n}_1 \\ \hat{n}_2 \end{bmatrix} = \begin{bmatrix} -\sigma_{13} \\ -\sigma_{23} \end{bmatrix} \hat{n}_3 \Rightarrow \begin{aligned} \hat{n}_1 &= a \hat{n}_3 \\ \hat{n}_2 &= b \hat{n}_3 \end{aligned}$$

Found:  $\hat{n}_1 = a \hat{n}_3$  and  $\hat{n}_2 = b \hat{n}_3$ .

Normal vector is an unit vector, so  $\hat{n}_1^2 + \hat{n}_2^2 + \hat{n}_3^2 = 1$

With this the first principal direction is determined.

The same could be done for the second and third principal directions

All three principal directions are perpendicular to each other!

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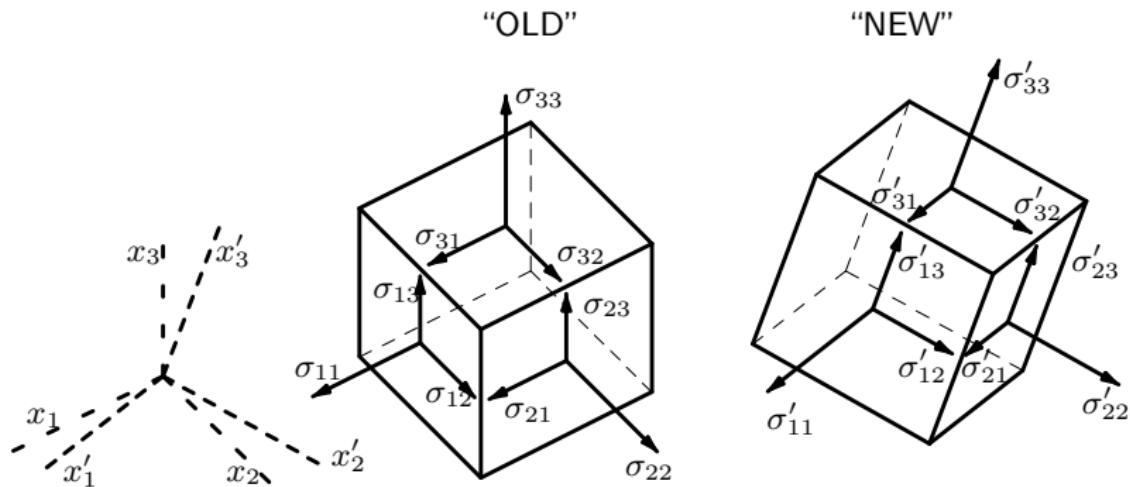
The same could be done for the second and third principal directions

**All three principal directions are perpendicular to each other!**

Preferably a coordinate system coinciding with the principal directions:

$$[\sigma]_{\text{ps}} = \begin{bmatrix} \sigma_1 & 0 & 0 \\ 0 & \sigma_2 & 0 \\ 0 & 0 & \sigma_3 \end{bmatrix} \Rightarrow \begin{array}{l} \text{Stress invariants:} \\ I_1 = \sigma_1 + \sigma_2 + \sigma_3 \\ I_2 = \sigma_1 \sigma_2 + \sigma_2 \sigma_3 + \sigma_3 \sigma_1 \\ I_3 = \sigma_1 \sigma_2 \sigma_3 \end{array}$$

### 3.4) Rotation matrix



Directional cosines:  $R_{pi} = \cos(x'_p, x_i)$

Surface

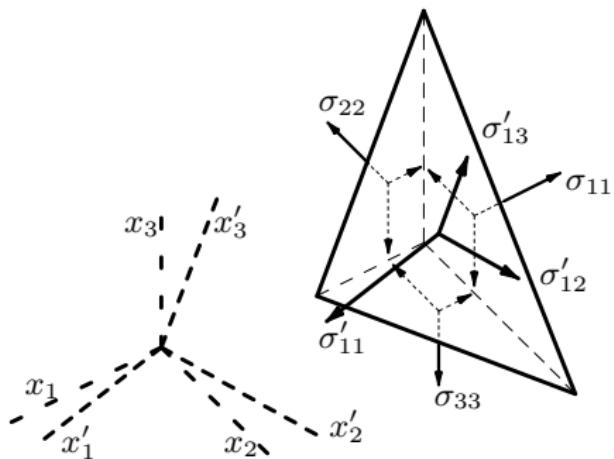
perpendicular to  $x'_1$ -axis.

Normal vector  $\hat{n}^{(1)}$

with:  $\hat{n}_i^{(1)} = \cos(x'_1, x_i) = R_{1i}$

Cauchy's formula:

$p_j^{(1)} = \sigma_{ij} \hat{n}_i^{(1)} = \sigma_{ij} R_{1i}$



Surface

perpendicular to  $x'_1$ -axis.

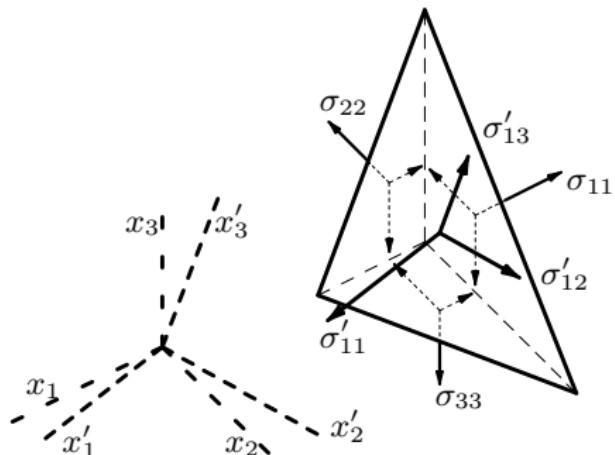
Normal vector  $\hat{n}^{(1)}$

with:  $\hat{n}_i^{(1)} = \cos(x'_1, x_i) = R_{1i}$

Cauchy's formula:

$$p_j^{(1)} = \sigma_{ij} \hat{n}_i^{(1)} = \sigma_{ij} R_{1i}$$

After rotation:



$$\sigma'_{1q} = p'^{(1)}_q = R_{qj} p_j^{(1)} = R_{qj} \sigma_{ij} R_{1i} = R_{1i} R_{qj} \sigma_{ij}$$

Same for the  $x'_2$ - and  $x'_3$ -directions. So:

$$\sigma'_{pq} = R_{pi} R_{qj} \sigma_{ij}$$

The stress components  $\sigma_{ij}$  turn out to be the components of a second order tensor. During rotation of coordinate axes holds:

$$\sigma'_{pq} = R_{pi} R_{qj} \sigma_{ij} \quad \text{or} \quad [\sigma'] = [R] [\sigma] [R]^T$$

Elaborated:

$$\begin{aligned}\sigma'_{pq} = & R_{p1} R_{q1} \sigma_{11} + R_{p1} R_{q2} \sigma_{12} + R_{p1} R_{q3} \sigma_{13} + \\& + R_{p2} R_{q1} \sigma_{21} + R_{p2} R_{q2} \sigma_{22} + R_{p2} R_{q3} \sigma_{23} + \\& + R_{p3} R_{q1} \sigma_{31} + R_{p3} R_{q2} \sigma_{32} + R_{p3} R_{q3} \sigma_{33}\end{aligned}$$

The stress components  $\sigma_{ij}$  turn out to be the components of a second order tensor. During transformation it holds that:

$$\sigma'_{pq} = R_{pi} R_{qj} \sigma_{ij} \quad \text{or} \quad [\sigma'] = [R] [\sigma] [R]^T$$

Elaborated:

$$\begin{aligned}\sigma'_{pq} &= R_{p1} R_{q1} \sigma_{11} + R_{p1} R_{q2} \sigma_{12} + R_{p1} R_{q3} \sigma_{13} + \\ &+ R_{p2} R_{q1} \sigma_{21} + R_{p2} R_{q2} \sigma_{22} + R_{p2} R_{q3} \sigma_{23} + \\ &+ R_{p3} R_{q1} \sigma_{31} + R_{p3} R_{q2} \sigma_{32} + R_{p3} R_{q3} \sigma_{33}\end{aligned}$$

Also the inverse transformation holds (“new” to “old”):

$$\sigma_{ij} = R_{pi} R_{qj} \sigma'_{pq} \quad \text{or} \quad \sigma_{pq} = R_{ip} R_{jq} \sigma'_{ij} \quad \text{or} \quad [\sigma] = [R]^T [\sigma'] [R]$$

### 3.5) Equilibrium equations

In reader 3-dim., here 2-dim.:

Instead

of a single point now a *volume*.

Stresses are a function of  $x_i$ .

Coordinates of the points:

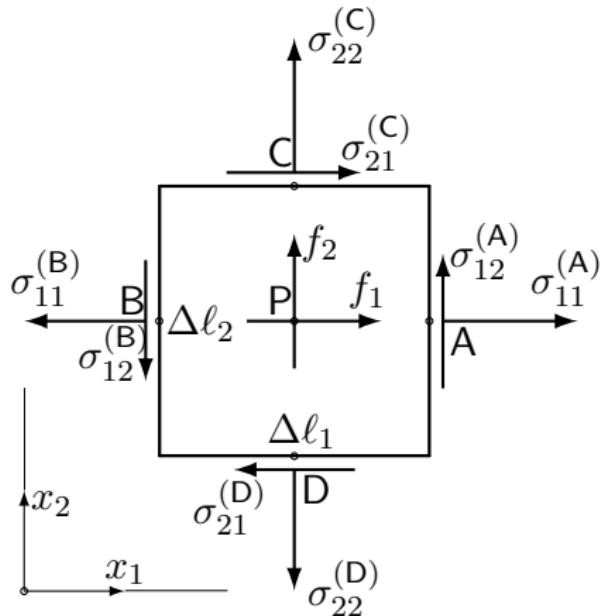
$$P : (x_1, x_2)$$

$$A : (x_1 + \frac{1}{2}\Delta\ell_1, x_2)$$

$$B : (x_1 - \frac{1}{2}\Delta\ell_1, x_2)$$

$$C : (x_1, x_2 + \frac{1}{2}\Delta\ell_2)$$

$$D : (x_1, x_2 - \frac{1}{2}\Delta\ell_2)$$



$f_1$  and  $f_2$  are volume forces or body forces, forces per unit volume.

### 3.5) Equilibrium equations

In reader 3-dim., here 2-dim.:

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Stresses are a function of  $x_i$ .

Coordinates of the points:

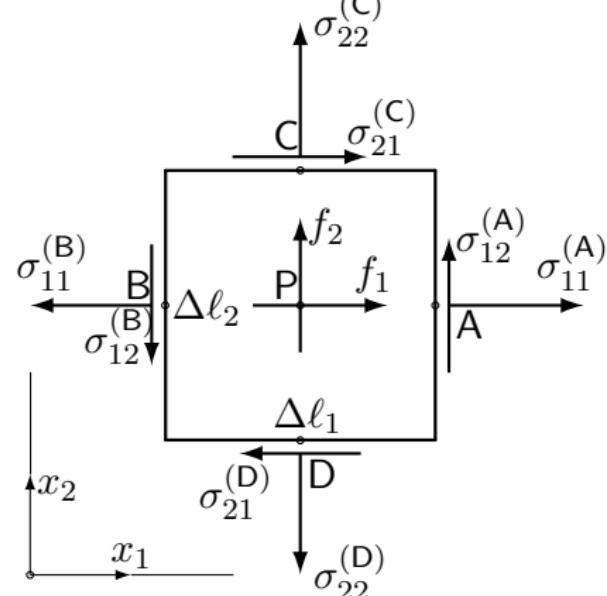
$$P : (x_1, x_2)$$

$$A : (x_1 + \frac{1}{2}\Delta\ell_1, x_2)$$

$$B : (x_1 - \frac{1}{2}\Delta\ell_1, x_2)$$

$$C : (x_1, x_2 + \frac{1}{2}\Delta\ell_2)$$

$$D : (x_1, x_2 - \frac{1}{2}\Delta\ell_2)$$

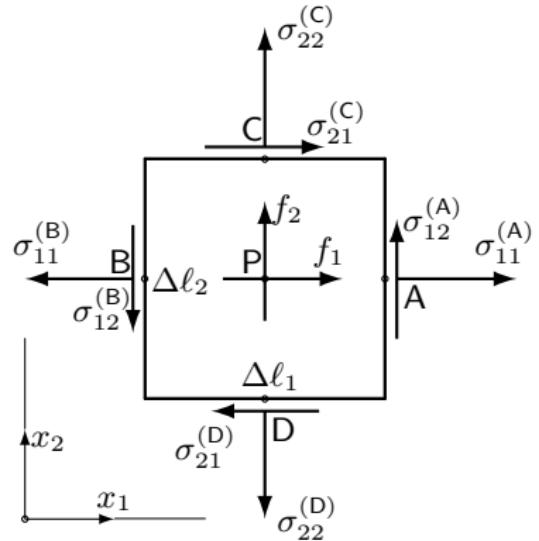


**Force equilibrium** in  $x_1$ -direction (volume size in  $x_3$ -direction is  $\Delta\ell_3$ ):

$$(\sigma_{11}^{(A)} - \sigma_{11}^{(B)}) \Delta\ell_2 \Delta\ell_3 + (\sigma_{21}^{(C)} - \sigma_{21}^{(D)}) \Delta\ell_1 \Delta\ell_3 + f_1 \Delta\ell_1 \Delta\ell_2 \Delta\ell_3 = 0$$

$$(\sigma_{11}^{(A)} - \sigma_{11}^{(B)}) \Delta\ell_2 \Delta\ell_3 + (\sigma_{21}^{(C)} - \sigma_{21}^{(D)}) \Delta\ell_1 \Delta\ell_3 + f_1 \Delta\ell_1 \Delta\ell_2 \Delta\ell_3 = 0$$

$$\begin{aligned}\sigma_{11}^{(A)} &= \sigma_{11}(x_1 + \frac{1}{2}\Delta\ell_1, x_2) = \\&= \sigma_{11}(x_1, x_2) + \frac{1}{2}\Delta\ell_1 \left( \frac{\partial \sigma_{11}}{\partial x_1} \right) + \text{H.O.T.} \\ \sigma_{11}^{(B)} &= \sigma_{11}(x_1 - \frac{1}{2}\Delta\ell_1, x_2) = \\&= \sigma_{11}(x_1, x_2) - \frac{1}{2}\Delta\ell_1 \left( \frac{\partial \sigma_{11}}{\partial x_1} \right) + \text{H.O.T.}\end{aligned}$$



$$(\sigma_{11}^{(A)} - \sigma_{11}^{(B)}) \Delta\ell_2 \Delta\ell_3 + (\sigma_{21}^{(C)} - \sigma_{21}^{(D)}) \Delta\ell_1 \Delta\ell_3 + f_1 \Delta\ell_1 \Delta\ell_2 \Delta\ell_3 = 0$$

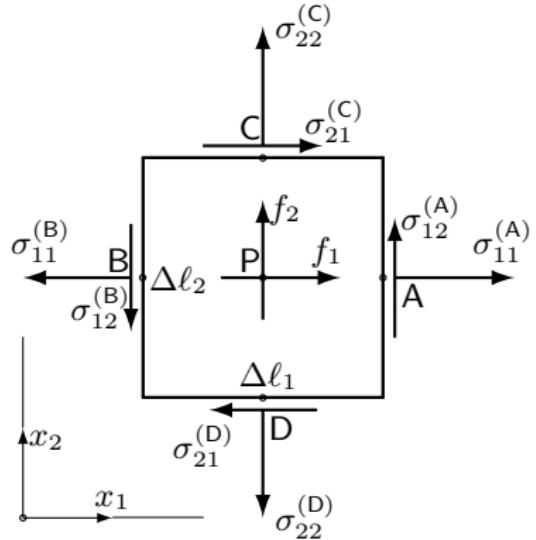
$$\begin{aligned}\sigma_{11}^{(A)} &= \sigma_{11}(x_1 + \frac{1}{2}\Delta\ell_1, x_2) = \\&= \sigma_{11}(x_1, x_2) + \frac{1}{2}\Delta\ell_1 \left( \frac{\partial \sigma_{11}}{\partial x_1} \right) + \text{H.O.T.}\end{aligned}$$

$$\begin{aligned}\sigma_{11}^{(B)} &= \sigma_{11}(x_1 - \frac{1}{2}\Delta\ell_1, x_2) = \\&= \sigma_{11}(x_1, x_2) - \frac{1}{2}\Delta\ell_1 \left( \frac{\partial \sigma_{11}}{\partial x_1} \right) + \text{H.O.T.}\end{aligned}$$

So (neglecting Higher Order Terms):

$$\sigma_{11}^{(A)} - \sigma_{11}^{(B)} = \Delta\ell_1 \left( \frac{\partial \sigma_{11}}{\partial x_1} \right)$$

$$\text{Idem: } \sigma_{21}^{(C)} - \sigma_{21}^{(D)} = \Delta\ell_2 \left( \frac{\partial \sigma_{21}}{\partial x_2} \right)$$



## Equilibrium of forces in $x_1$ -direction:

$$(\sigma_{11}^{(A)} - \sigma_{11}^{(B)}) \Delta\ell_2 \Delta\ell_3 + (\sigma_{21}^{(C)} - \sigma_{21}^{(D)}) \Delta\ell_1 \Delta\ell_3 + f_1 \Delta\ell_1 \Delta\ell_2 \Delta\ell_3 = 0$$

Derived:  $\sigma_{11}^{(A)} - \sigma_{11}^{(B)} = \Delta\ell_1 \frac{\partial\sigma_{11}}{\partial x_1}$  and  $\sigma_{21}^{(C)} - \sigma_{21}^{(D)} = \Delta\ell_2 \frac{\partial\sigma_{21}}{\partial x_2}$

Results in (after dividing by  $\Delta\ell_1 \Delta\ell_2 \Delta\ell_3 = \Delta V$ ):

$$\frac{\partial\sigma_{11}}{\partial x_1} + \frac{\partial\sigma_{21}}{\partial x_2} + f_1 = 0$$

Shorter:  $\sum_{j=1}^{\dots} \frac{\partial\sigma_{j1}}{\partial x_j} + f_1 = 0 \Rightarrow \sigma_{j1,j} + f_1 = 0$

## Equilibrium of forces in $x_1$ -direction:

$$(\sigma_{11}^{(A)} - \sigma_{11}^{(B)}) \Delta\ell_2 \Delta\ell_3 + (\sigma_{21}^{(C)} - \sigma_{21}^{(D)}) \Delta\ell_1 \Delta\ell_3 + f_1 \Delta\ell_1 \Delta\ell_2 \Delta\ell_3 = 0$$

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Results in (after dividing by  $\Delta\ell_1 \Delta\ell_2 \Delta\ell_3 = \Delta V$ ):

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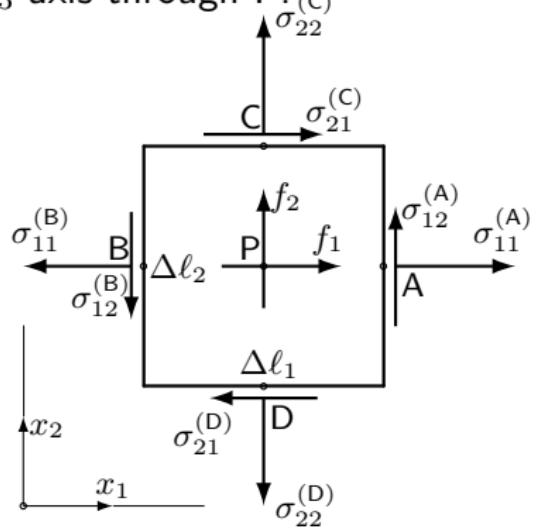
Shorter:  $\sum_{j=1}^{\dots} \frac{\partial\sigma_{j1}}{\partial x_j} + f_1 = 0 \Rightarrow \sigma_{j1,j} + f_1 = 0$

The same for other directions, expanded to 3-dim.:

$$\boxed{\sigma_{ji,j} + f_i = 0} \quad \left( \sum_{j=1}^3 \frac{\partial\sigma_{ji}}{\partial x_j} + f_i = 0 \text{ for } i = 1, 2, 3 \right)$$

**Equilibrium of moments** around the  $x_3$ -axis through P:

$$\begin{aligned} & (\sigma_{12}^{(A)} + \sigma_{12}^{(B)}) \Delta\ell_2 \Delta\ell_3 \frac{1}{2} \Delta\ell_1 + \\ & - (\sigma_{21}^{(C)} + \sigma_{21}^{(D)}) \Delta\ell_1 \Delta\ell_3 \frac{1}{2} \Delta\ell_2 = 0 \\ \implies & (\sigma_{12}^{(A)} + \sigma_{12}^{(B)}) - (\sigma_{21}^{(C)} + \sigma_{21}^{(D)}) = 0 \end{aligned}$$



Equilibrium of moments around the  $x_3$ -axis through P:

$$(\sigma_{12}^{(A)} + \sigma_{12}^{(B)}) \Delta\ell_2 \Delta\ell_3 \frac{1}{2} \Delta\ell_1 + \\ - (\sigma_{21}^{(C)} + \sigma_{21}^{(D)}) \Delta\ell_1 \Delta\ell_3 \frac{1}{2} \Delta\ell_2 = 0 \\ \Rightarrow (\sigma_{12}^{(A)} + \sigma_{12}^{(B)}) - (\sigma_{21}^{(C)} + \sigma_{21}^{(D)}) = 0$$

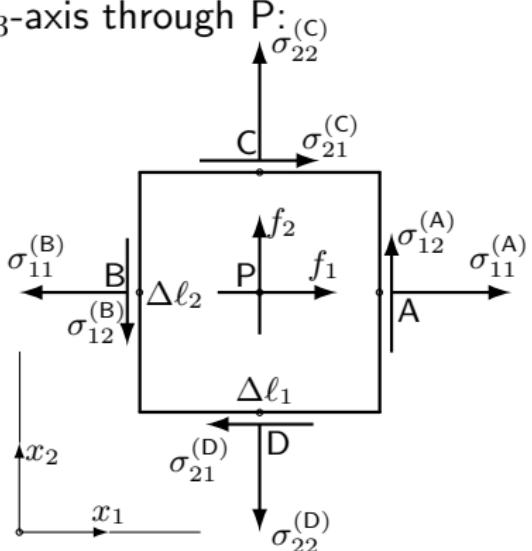
$$\sigma_{12}^{(A)} = \sigma_{12} + \frac{1}{2} \Delta\ell_1 \left( \frac{\partial \sigma_{12}}{\partial x_1} \right) + \text{H.O.T.}$$

$$\sigma_{12}^{(B)} = \sigma_{12} - \frac{1}{2} \Delta\ell_1 \left( \frac{\partial \sigma_{12}}{\partial x_1} \right) + \text{H.O.T.}$$

$$\sigma_{21}^{(C)} = \sigma_{21} + \frac{1}{2} \Delta\ell_2 \left( \frac{\partial \sigma_{21}}{\partial x_2} \right) + \text{H.O.T.}$$

$$\sigma_{21}^{(D)} = \sigma_{21} - \frac{1}{2} \Delta\ell_2 \left( \frac{\partial \sigma_{21}}{\partial x_2} \right) + \text{H.O.T.}$$

$$\Rightarrow 2\sigma_{12} - 2\sigma_{21} = 0 \Rightarrow \sigma_{12} = \sigma_{21}$$



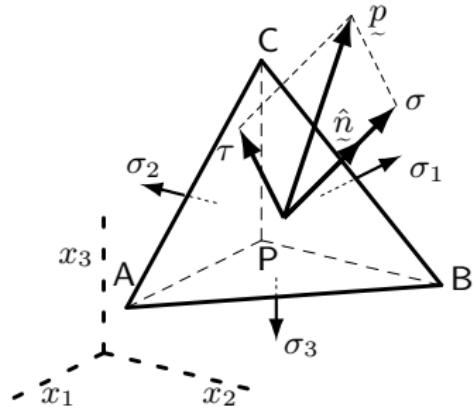
in general:

$$\boxed{\sigma_{ij} = \sigma_{ji}}$$

### 3.6) Maximum shear stresses, extrema

How to obtain maximum shear stress  $\tau_{\max}$ .

Consider the stress state in a point. Calculate the principal stresses  $\sigma_i$  with corresponding principal directions. Orientate the  $x_i$ -coordinate system that it will coincide with the principal directions.



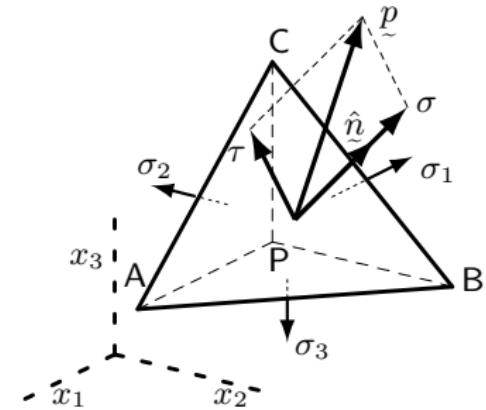
### 3.6) Maximum shear stresses, extrema

How to obtain maximum shear stress  $\tau_{\max}$ .

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The stress matrix will be:

$$[\sigma_h] = [\sigma]_{x_i} = \begin{bmatrix} \sigma_1 & 0 & 0 \\ 0 & \sigma_2 & 0 \\ 0 & 0 & \sigma_3 \end{bmatrix}$$



Apply an arbitrary orientated surface, with normal vector  $\hat{n}$ .

The components of the stress vector  $\underline{p}$  to the surface (**Cauchy**):

$$\{p\} = [\sigma_h]\{\hat{n}\} = \begin{bmatrix} \sigma_1 & 0 & 0 \\ 0 & \sigma_2 & 0 \\ 0 & 0 & \sigma_3 \end{bmatrix} \begin{bmatrix} \hat{n}_1 \\ \hat{n}_2 \\ \hat{n}_3 \end{bmatrix} = \begin{bmatrix} \sigma_1 \hat{n}_1 \\ \sigma_2 \hat{n}_2 \\ \sigma_3 \hat{n}_3 \end{bmatrix}$$

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The magnitude,  $\sigma$ , of the normal stress on the surface is:

$$\sigma = \{\hat{n}\}^T \{p\} = [ \hat{n}_1 \ \hat{n}_2 \ \hat{n}_3 ] \begin{bmatrix} \sigma_1 \hat{n}_1 \\ \sigma_2 \hat{n}_2 \\ \sigma_3 \hat{n}_3 \end{bmatrix} = \sigma_1 \hat{n}_1^2 + \sigma_2 \hat{n}_2^2 + \sigma_3 \hat{n}_3^2$$

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$$\sigma = \{\hat{n}\}^T \{p\} = [\hat{n}_1 \ \hat{n}_2 \ \hat{n}_3] \begin{bmatrix} \sigma_1 \hat{n}_1 \\ \sigma_2 \hat{n}_2 \\ \sigma_3 \hat{n}_3 \end{bmatrix} = \sigma_1 \hat{n}_1^2 + \sigma_2 \hat{n}_2^2 + \sigma_3 \hat{n}_3^2$$

The magnitude,  [\$\tau\$](#) , of the shear stress on the surface:

$$\tau^2 = |\underline{\underline{p}}|^2 - \sigma^2 = \sigma_1^2 \hat{n}_1^2 + \sigma_2^2 \hat{n}_2^2 + \sigma_3^2 \hat{n}_3^2 - (\sigma_1 \hat{n}_1^2 + \sigma_2 \hat{n}_2^2 + \sigma_3 \hat{n}_3^2)^2$$

Found:  $\tau^2 = \sigma_1^2 \hat{n}_1^2 + \sigma_2^2 \hat{n}_2^2 + \sigma_3^2 \hat{n}_3^2 - (\sigma_1 \hat{n}_1^2 + \sigma_2 \hat{n}_2^2 + \sigma_3 \hat{n}_3^2)^2$

Eliminate the directional coefficient  $\hat{n}_3$  with:  $\hat{n}_3^2 = 1 - \hat{n}_1^2 - \hat{n}_2^2 \implies$

$$\begin{aligned}\tau^2 &= (\sigma_1^2 - \sigma_3^2) \hat{n}_1^2 + (\sigma_2^2 - \sigma_3^2) \hat{n}_2^2 + \sigma_3^2 + \\ &\quad - \{(\sigma_1 - \sigma_3) \hat{n}_1^2 + (\sigma_2 - \sigma_3) \hat{n}_2^2 + \sigma_3\}^2\end{aligned}$$

Extrema can be found with:  $\frac{\partial \tau^2}{\partial \hat{n}_1} = \frac{\partial \tau^2}{\partial \hat{n}_2} = 0$

Found:  $\tau^2 = \sigma_1^2 \hat{n}_1^2 + \sigma_2^2 \hat{n}_2^2 + \sigma_3^2 \hat{n}_3^2 - (\sigma_1 \hat{n}_1^2 + \sigma_2 \hat{n}_2^2 + \sigma_3 \hat{n}_3^2)^2$   
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Extrema can be found with:  $\frac{\partial \tau^2}{\partial \hat{n}_1} = \frac{\partial \tau^2}{\partial \hat{n}_2} = 0$

The following solutions are found:

$$\hat{n}_1 = 0 \quad \text{with} \quad \begin{cases} 1) & \hat{n}_2 = 0 \quad \hat{n}_3 = \pm 1 \\ 2) & \hat{n}_2 = \pm \frac{1}{2}\sqrt{2} \quad \hat{n}_3 = \mp \frac{1}{2}\sqrt{2} \end{cases} \quad \circlearrowright$$

The first solution is  $\tau = 0$ ; this is trivial and not interesting

The second solution,  $\hat{n}_1 = 0$  ;  $\hat{n}_2 = \pm \frac{1}{2}\sqrt{2}$  ;  $\hat{n}_3 = \mp \frac{1}{2}\sqrt{2}$  gives:

$$\tau^2 = (\sigma_2^2 - \sigma_3^2) \frac{1}{2} + \sigma_3^2 - \{(\sigma_2 - \sigma_3) \frac{1}{2} + \sigma_3\}^2 = \frac{1}{4} (\sigma_2 - \sigma_3)^2$$

An extreme case for  $\tau$  is found at:  $\frac{1}{2} |\sigma_2 - \sigma_3|$ .

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In a similar way the following solutions can be found:

$$\hat{n}_1 = 0 \quad \hat{n}_2 = \pm \frac{1}{2}\sqrt{2} \quad \hat{n}_3 = \mp \frac{1}{2}\sqrt{2} \quad \tau = \frac{1}{2} |\sigma_2 - \sigma_3|$$

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$$\begin{array}{llll} \hat{n}_1 = 0 & \hat{n}_2 = \pm \frac{1}{2}\sqrt{2} & \hat{n}_3 = \mp \frac{1}{2}\sqrt{2} & \tau = \frac{1}{2} |\sigma_2 - \sigma_3| \\ \hat{n}_1 = \pm \frac{1}{2}\sqrt{2} & \hat{n}_2 = \mp \frac{1}{2}\sqrt{2} & \hat{n}_3 = 0 & \tau = \frac{1}{2} |\sigma_1 - \sigma_2| \\ \hat{n}_1 = \pm \frac{1}{2}\sqrt{2} & \hat{n}_2 = 0 & \hat{n}_3 = \mp \frac{1}{2}\sqrt{2} & \tau = \frac{1}{2} |\sigma_1 - \sigma_3| \end{array}$$

If  $\sigma_1 \geq \sigma_2 \geq \sigma_3$  then  $\boxed{\tau_{\max} = \frac{1}{2} (\sigma_1 - \sigma_3)}$

This  $\tau_{\max}$  originates from a  $45^\circ$  rotation around the  $x_2$ -axis

$45^\circ$  rotation around  $x_2$ -axis:

$$[R] = \begin{bmatrix} \frac{1}{2}\sqrt{2} & 0 & -\frac{1}{2}\sqrt{2} \\ 0 & 1 & 0 \\ \frac{1}{2}\sqrt{2} & 0 & \frac{1}{2}\sqrt{2} \end{bmatrix}$$

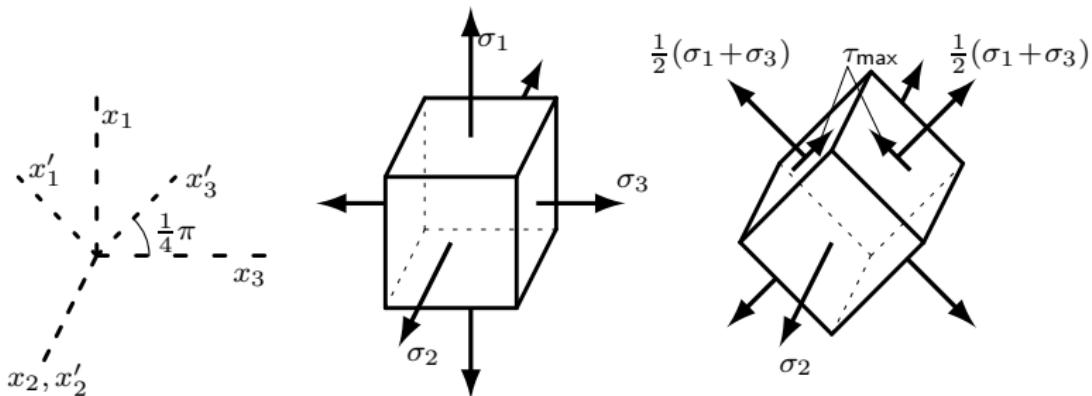
Then:

$$[\sigma'] = [R] [\sigma_h] [R]^T = \begin{bmatrix} \frac{1}{2}(\sigma_1 + \sigma_3) & 0 & \frac{1}{2}(\sigma_1 - \sigma_3) \\ 0 & \sigma_2 & 0 \\ \frac{1}{2}(\sigma_1 - \sigma_3) & 0 & \frac{1}{2}(\sigma_1 + \sigma_3) \end{bmatrix}$$

$45^\circ$  rotation around  $x_2$ -axis:

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Extrema from the **normal stress** on a surface.

Found previously:  $\sigma = \sigma_1 \hat{n}_1^2 + \sigma_2 \hat{n}_2^2 + \sigma_3 \hat{n}_3^2$

Elimination of  $\hat{n}_3$  using  $\hat{n}_3^2 = 1 - \hat{n}_1^2 - \hat{n}_2^2$

$$\sigma = (\sigma_1 - \sigma_3) \hat{n}_1^2 + (\sigma_2 - \sigma_3) \hat{n}_2^2 + \sigma_3$$

Extrema when:  $\frac{\partial \sigma}{\partial \hat{n}_1} = \frac{\partial \sigma}{\partial \hat{n}_2} = 0$

$$\text{Resulting in: } (\sigma_1 - \sigma_3) 2\hat{n}_1 = (\sigma_2 - \sigma_3) 2\hat{n}_2 = 0$$

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Extrema when:  $\frac{\partial \sigma}{\partial \hat{n}_1} = \frac{\partial \sigma}{\partial \hat{n}_2} = 0$

$$\text{Resulting in: } (\sigma_1 - \sigma_3) 2\hat{n}_1 = (\sigma_2 - \sigma_3) 2\hat{n}_2 = 0$$

Solution (general case):  $\hat{n}_1 = \hat{n}_2 = 0 ; \hat{n}_3 = 1$

This is exactly the 3<sup>th</sup> principal direction corresponding to  $\sigma_3$ .  $\sigma_1$  and  $\sigma_2$  can be found with elimination of  $\hat{n}_1$  resp.  $\hat{n}_2$ . So, the principal stresses are the extrema of the normal stress.

When  $\sigma_1 \geq \sigma_2 \geq \sigma_3$  it must hold that:  $\sigma_1 \geq \sigma \geq \sigma_3$ .

### 3.7) Miscellaneous, including Mohr's circle(s)

Consider the following special stress state:  $[\sigma] = \begin{bmatrix} \sigma_{11} & \sigma_{12} & 0 \\ \sigma_{21} & \sigma_{22} & 0 \\ 0 & 0 & \sigma_{33} \end{bmatrix}$

NOTE, that  $\sigma_{21} = \sigma_{12}$  and that  $\sigma_{33}$  must be a **Principal stress** (because  $\sigma_{13} = \sigma_{31} = 0$  and  $\sigma_{23} = \sigma_{32} = 0$ ).

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NOTE, that  $\sigma_{21} = \sigma_{12}$  and that  $\sigma_{33}$  must be a **Principal stress** (because  $\sigma_{13} = \sigma_{31} = 0$  and  $\sigma_{23} = \sigma_{32} = 0$ ).

Calculate principal stresses:

$$\begin{aligned} & \det \left( \begin{bmatrix} \sigma_{11} - \sigma & \sigma_{12} & 0 \\ \sigma_{21} & \sigma_{22} - \sigma & 0 \\ 0 & 0 & \sigma_{33} - \sigma \end{bmatrix} \right) = 0 \\ \Rightarrow & (\sigma_{33} - \sigma) \{(\sigma_{11} - \sigma)(\sigma_{22} - \sigma) - \sigma_{12}^2\} = 0 \\ \Rightarrow & (\sigma_{33} - \sigma) \{\sigma^2 - (\sigma_{11} + \sigma_{22})\sigma + \sigma_{11}\sigma_{22} - \sigma_{12}^2\} = 0 \end{aligned}$$

$$\text{Found: } (\sigma_{33} - \sigma) \{ \sigma^2 - (\sigma_{11} + \sigma_{22})\sigma + \sigma_{11}\sigma_{22} - \sigma_{12}^2 \} = 0$$

Solutions:

$$\begin{aligned}\sigma_{1,2} &= \frac{1}{2}(\sigma_{11} + \sigma_{22}) \pm \frac{1}{2}\sqrt{(\sigma_{11} + \sigma_{22})^2 - 4\sigma_{11}\sigma_{22} + 4\sigma_{12}^2} = \\ &= \frac{1}{2}(\sigma_{11} + \sigma_{22}) \pm \sqrt{\frac{1}{4}(\sigma_{11} - \sigma_{22})^2 + \sigma_{12}^2} \\ \sigma_3 &= \sigma_{33}\end{aligned}$$

The order of the principal stresses is STILL random!

$$\text{Found: } (\sigma_{33} - \sigma) \{ \sigma^2 - (\sigma_{11} + \sigma_{22}) \sigma + \sigma_{11}\sigma_{22} - \sigma_{12}^2 \} = 0$$

Solutions:

$$\begin{aligned}\sigma_{1,2} &= \frac{1}{2}(\sigma_{11} + \sigma_{22}) \pm \frac{1}{2}\sqrt{(\sigma_{11} + \sigma_{22})^2 - 4\sigma_{11}\sigma_{22} + 4\sigma_{12}^2} = \\ &= \frac{1}{2}(\sigma_{11} + \sigma_{22}) \pm \sqrt{\frac{1}{4}(\sigma_{11} - \sigma_{22})^2 + \sigma_{12}^2} \\ \sigma_3 &= \sigma_{33}\end{aligned}$$

The order of the principal stresses is STILL random!

The principal directions can be found with:

$$\begin{bmatrix} \sigma_{11} - \sigma_i & \sigma_{12} & 0 \\ \sigma_{21} & \sigma_{22} - \sigma_i & 0 \\ 0 & 0 & \sigma_{33} - \sigma_i \end{bmatrix} \begin{bmatrix} \hat{n}_1 \\ \hat{n}_2 \\ \hat{n}_3 \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \\ 0 \end{bmatrix} \quad \text{for resp. } i = 1, 2, 3$$

$$\begin{bmatrix} \sigma_{11} - \sigma_i & \sigma_{12} & 0 \\ \sigma_{21} & \sigma_{22} - \sigma_i & 0 \\ 0 & 0 & \sigma_{33} - \sigma_i \end{bmatrix} \begin{bmatrix} \hat{n}_1 \\ \hat{n}_2 \\ \hat{n}_3 \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \\ 0 \end{bmatrix} \quad \text{for resp. } i = 1, 2, 3$$

Consecutively resulting in, for  $i = 1, 2, 3$ :

$$\{\hat{n}\}^{(1)} = \begin{bmatrix} a_1 \\ a_2 \\ 0 \end{bmatrix} ; \quad \{\hat{n}\}^{(2)} = \begin{bmatrix} a_2 \\ -a_1 \\ 0 \end{bmatrix} ; \quad \{\hat{n}\}^{(3)} = \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix}$$

$$a_1^2 + a_2^2 = 1 \Rightarrow a_1 = \frac{\sigma_{12}}{w} ; \quad a_2 = \frac{\sigma_i - \sigma_{11}}{w} ; \quad w = \sqrt{(\sigma_i - \sigma_{11})^2 + \sigma_{12}^2}$$

What happens with a rotation of angle  $\alpha$  around the  $x_3$ -axis?

Rotation matrix:  $[R] = \begin{bmatrix} \cos \alpha & \sin \alpha & 0 \\ -\sin \alpha & \cos \alpha & 0 \\ 0 & 0 & 1 \end{bmatrix}$

What happens with a rotation of angle  $\alpha$  around the  $x_3$ -axis?

Rotatiematrix:  $[R] = \begin{bmatrix} \cos \alpha & \sin \alpha & 0 \\ -\sin \alpha & \cos \alpha & 0 \\ 0 & 0 & 1 \end{bmatrix}$

Calculation (using symmetry,  $\sigma_{21} = \sigma_{12}$ ):

$$\begin{aligned} [\sigma'] &= \begin{bmatrix} \sigma'_{11} & \sigma'_{12} & 0 \\ \sigma'_{21} & \sigma'_{22} & 0 \\ 0 & 0 & \sigma_{33} \end{bmatrix} = [R][\sigma][R]^T = \\ &= \begin{bmatrix} \cos \alpha & \sin \alpha & 0 \\ -\sin \alpha & \cos \alpha & 0 \\ 0 & 0 & 1 \end{bmatrix} \begin{bmatrix} \sigma_{11} & \sigma_{12} & 0 \\ \sigma_{21} & \sigma_{22} & 0 \\ 0 & 0 & \sigma_{33} \end{bmatrix} \begin{bmatrix} \cos \alpha & -\sin \alpha & 0 \\ \sin \alpha & \cos \alpha & 0 \\ 0 & 0 & 1 \end{bmatrix} \end{aligned}$$

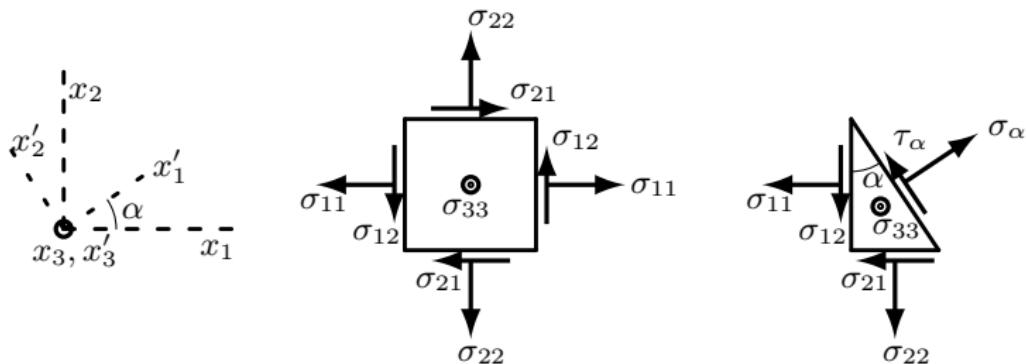
Solving matrices:

$$\begin{aligned}\sigma_\alpha &= \sigma'_{11} = \sigma_{11}(\cos \alpha)^2 + 2\sigma_{12} \sin \alpha \cos \alpha + \sigma_{22}(\sin \alpha)^2 = \\&= \frac{1}{2}(\sigma_{11} + \sigma_{22}) + \frac{1}{2}(\sigma_{11} - \sigma_{22}) \cos 2\alpha + \sigma_{12} \sin 2\alpha \\ \tau_\alpha &= \sigma'_{12} = (\sigma_{22} - \sigma_{11}) \sin \alpha \cos \alpha + \sigma_{12} \{(\cos \alpha)^2 - (\sin \alpha)^2\} = \\&= -\frac{1}{2}(\sigma_{11} - \sigma_{22}) \sin 2\alpha + \sigma_{12} \cos 2\alpha\end{aligned}$$

Solving matrices:

$$\begin{aligned}\sigma_\alpha &= \sigma'_{11} = \sigma_{11}(\cos \alpha)^2 + 2\sigma_{12} \sin \alpha \cos \alpha + \sigma_{22}(\sin \alpha)^2 = \\ &= \frac{1}{2}(\sigma_{11} + \sigma_{22}) + \frac{1}{2}(\sigma_{11} - \sigma_{22}) \cos 2\alpha + \sigma_{12} \sin 2\alpha \\ \tau_\alpha &= \sigma'_{12} = (\sigma_{22} - \sigma_{11}) \sin \alpha \cos \alpha + \sigma_{12} \{(\cos \alpha)^2 - (\sin \alpha)^2\} = \\ &= -\frac{1}{2}(\sigma_{11} - \sigma_{22}) \sin 2\alpha + \sigma_{12} \cos 2\alpha\end{aligned}$$

Visualisation:



$$\begin{aligned}\sigma_\alpha &= \frac{1}{2}(\sigma_{11} + \sigma_{22}) + \frac{1}{2}(\sigma_{11} - \sigma_{22}) \cos 2\alpha + \sigma_{12} \sin 2\alpha \\ \tau_\alpha &= -\frac{1}{2}(\sigma_{11} - \sigma_{22}) \sin 2\alpha + \sigma_{12} \cos 2\alpha\end{aligned}$$

The equation of a circle is hidden in the expressions of  $\sigma_\alpha$  and  $\tau_\alpha$ :

$$\{\sigma_\alpha - \frac{1}{2}(\sigma_{11} + \sigma_{22})\}^2 + \tau_\alpha^2 = \frac{1}{4}(\sigma_{11} - \sigma_{22})^2 + \sigma_{12}^2$$

$$\sigma_\alpha = \frac{1}{2}(\sigma_{11} + \sigma_{22}) + \frac{1}{2}(\sigma_{11} - \sigma_{22}) \cos 2\alpha + \sigma_{12} \sin 2\alpha$$

$$\tau_\alpha = -\frac{1}{2}(\sigma_{11} - \sigma_{22}) \sin 2\alpha + \sigma_{12} \cos 2\alpha$$

The equation of a circle is hidden in the expressions of  $\sigma_\alpha$  and  $\tau_\alpha$ :

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### Mohr's circle

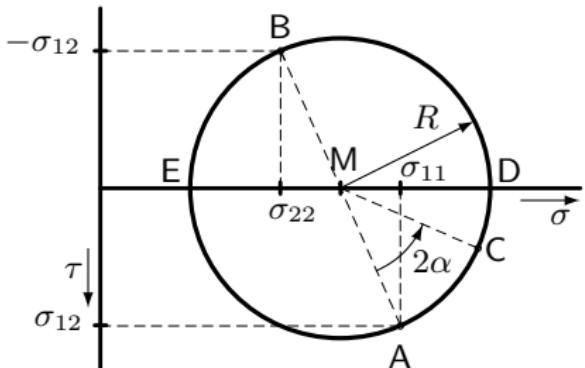
Midpoint M:  $\left\{ \frac{1}{2}(\sigma_{11} + \sigma_{22}), 0 \right\}$

Radius:  $R = \sqrt{\frac{1}{4}(\sigma_{11} - \sigma_{22})^2 + \sigma_{12}^2}$

Omit index  $\alpha$  at  $\sigma$  and  $\tau$ !

A:  $\alpha = 0$   $(\sigma_{11}, \sigma_{12})$

B:  $\alpha = \frac{1}{2}\pi$   $(\sigma_{22}, \sigma_{21})$



$$\sigma_\alpha = \frac{1}{2}(\sigma_{11} + \sigma_{22}) + \frac{1}{2}(\sigma_{11} - \sigma_{22}) \cos 2\alpha + \sigma_{12} \sin 2\alpha$$

$$\tau_\alpha = -\frac{1}{2}(\sigma_{11} - \sigma_{22}) \sin 2\alpha + \sigma_{12} \cos 2\alpha$$

The equation of a circle is hidden in the expressions of  $\sigma_\alpha$  and  $\tau_\alpha$ :

$$\{\sigma_\alpha - \frac{1}{2}(\sigma_{11} + \sigma_{22})\}^2 + \tau_\alpha^2 = \frac{1}{4}(\sigma_{11} - \sigma_{22})^2 + \sigma_{12}^2$$

### Mohr's circle - ALTERNATIVE

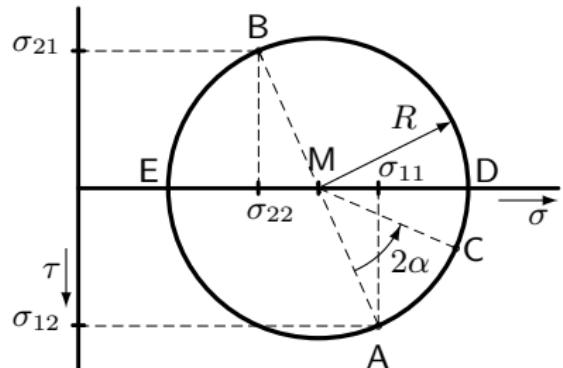
Midpoint M:  $\left\{\frac{1}{2}(\sigma_{11} + \sigma_{22}), 0\right\}$

Radius:  $R = \sqrt{\frac{1}{4}(\sigma_{11} - \sigma_{22})^2 + \sigma_{12}^2}$

Omit index  $\alpha$  at  $\sigma$  and  $\tau$ !

A:  $\alpha = 0$   $(\sigma_{11}, \sigma_{12} \downarrow)$

B:  $\alpha = \frac{1}{2}\pi$   $(\sigma_{22}, \sigma_{21} \uparrow)$



C:  $\alpha = ??$   $(\sigma_\alpha, \tau_\alpha)$

D:  $\alpha = \alpha_1$   $(\sigma_1, 0)$

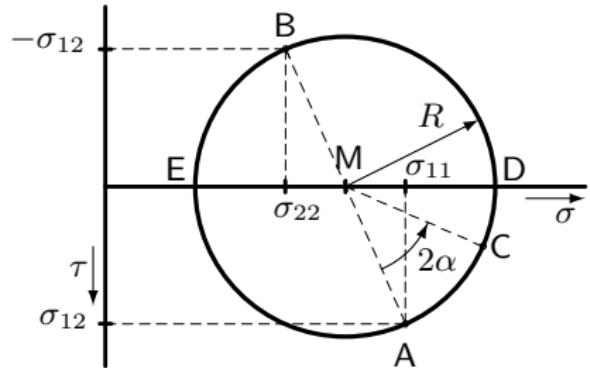
E:  $\alpha = \alpha_2$   $(\sigma_2, 0)$

( $2\alpha_1$  = angle AMD,

$2\alpha_2$  = angle AME)

$$\tan(2\alpha_1) = \frac{\sigma_{12}}{\frac{1}{2}(\sigma_{11} - \sigma_{22})}$$

$$\alpha_2 = \alpha_1 \pm \frac{1}{2}\pi$$



C:  $\alpha = ??$   $(\sigma_\alpha, \tau_\alpha)$

D:  $\alpha = \alpha_1$   $(\sigma_1, 0)$

E:  $\alpha = \alpha_2$   $(\sigma_2, 0)$

( $2\alpha_1$  = angle AMD,

$2\alpha_2$  = angle AME)

$$\tan(2\alpha_1) = \frac{\sigma_{12}}{\frac{1}{2}(\sigma_{11} - \sigma_{22})}$$

$$\alpha_2 = \alpha_1 \pm \frac{1}{2}\pi$$

(i.e. eigenvectors in real space are perpendicular!)

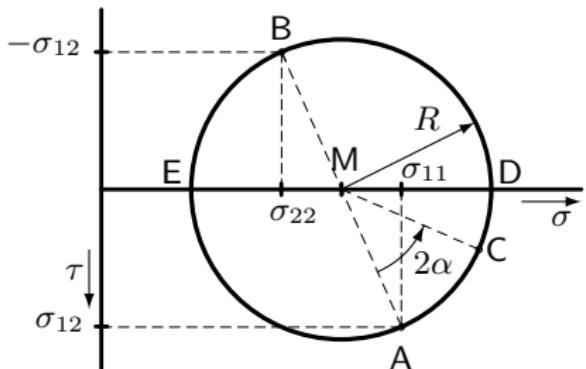
Eigenvectors in Mohr circle are separated by  $2|\alpha_1 - \alpha_2| = \pi$  (i.e.  $180^\circ$ )

$$\sigma_1 = \frac{1}{2}(\sigma_{11} + \sigma_{22}) + R = \frac{1}{2}(\sigma_{11} + \sigma_{22}) + \sqrt{\frac{1}{4}(\sigma_{11} - \sigma_{22})^2 + \sigma_{12}^2}$$

$$\sigma_2 = \frac{1}{2}(\sigma_{11} + \sigma_{22}) - R = \frac{1}{2}(\sigma_{11} + \sigma_{22}) - \sqrt{\frac{1}{4}(\sigma_{11} - \sigma_{22})^2 + \sigma_{12}^2}$$

Principal stresses are the same as found earlier!!

Memorize/practice construction procedure and the factor 2



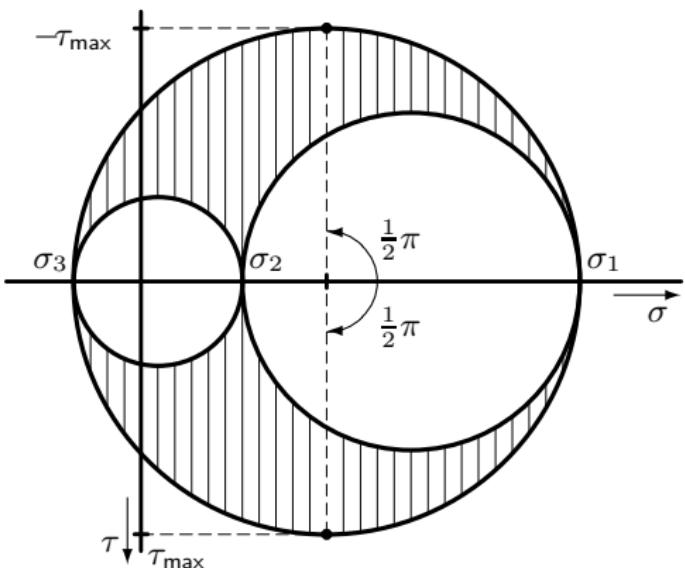
## Arbitrary, three dimensional stress state.

Note:  $\tau_{\max} = \frac{1}{2}(\sigma_1 - \sigma_3)$

Angle with first principal direction:

$$\frac{1}{2} (\pm \frac{1}{2}\pi) = \pm \frac{1}{4}\pi$$

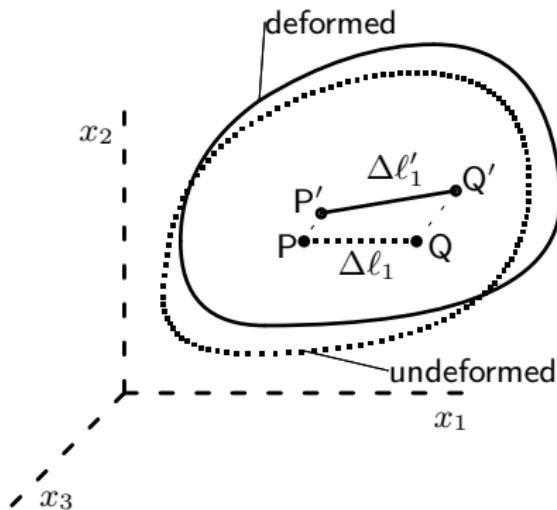
Points from an arbitrary chosen coordinate system will end up in the striped area!



## 4) Deformation

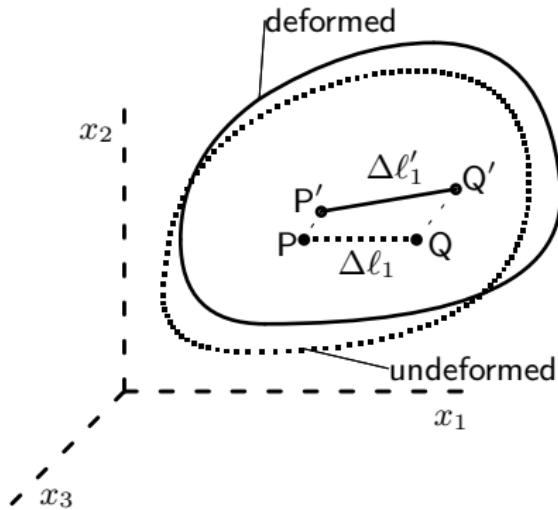
### 4.1) Strain

Consider a line segment PQ parallel with the  $x_1$ -axis.



## 4) Deformation

### 4.1) Strain



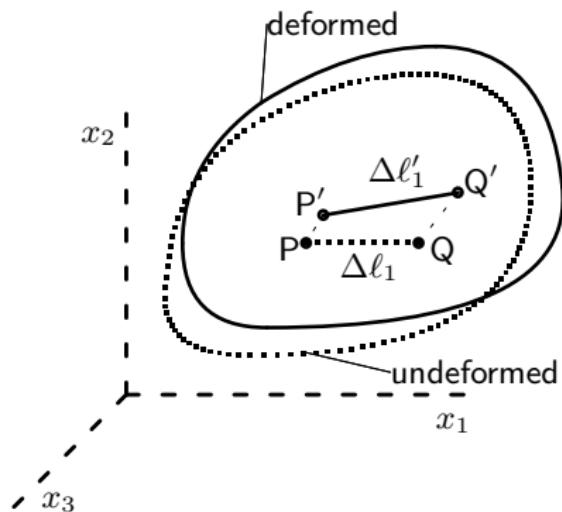
Consider a line segment  $PQ$  parallel with the  $x_1$ -axis.

The coordinates of  $P$  are  $x_1, x_2, x_3$   
Translation of  $P$  to  $P'$

is up with components  $u_i(x_1, x_2, x_3)$ .

## 4) Deformation

### 4.1) Strain



Translation of Q to  $Q'$  is  $\underline{u}_Q$  with components  $u_i(x_1 + \Delta\ell_1, x_2, x_3)$ .

$$\text{with: } u_i(x_1 + \Delta\ell_1, x_2, x_3) = u_i(x_1, x_2, x_3) + \Delta\ell_1 \frac{\partial u_i}{\partial x_1} + \text{H.O.T.}$$

Consider a line segment  $PQ$  parallel with the  $x_1$ -axis.

The

coordinates of  $P$  are  $x_1, x_2, x_3$

Translation of  $P$  to  $P'$  is  $\underline{u}_P$

with components  $u_i(x_1, x_2, x_3)$ .

The coordinates

of  $Q$  are  $x_1 + \Delta\ell_1, x_2, x_3$

Coordinates of the translated points:

$$P': \begin{pmatrix} x_1+u_1 \\ x_2+u_2 \\ x_3+u_3 \end{pmatrix} \quad Q': \begin{pmatrix} x_1+\Delta\ell_1+u_1+\frac{\partial u_1}{\partial x_1}\Delta\ell_1 \\ x_2+u_2+\frac{\partial u_2}{\partial x_1}\Delta\ell_1 \\ x_3+u_3+\frac{\partial u_3}{\partial x_1}\Delta\ell_1 \end{pmatrix}$$

Coordinates of the translated points:

$$P': \begin{pmatrix} x_1 + u_1 \\ x_2 + u_2 \\ x_3 + u_3 \end{pmatrix} \quad Q': \begin{pmatrix} x_1 + \Delta\ell_1 + u_1 + \frac{\partial u_1}{\partial x_1} \Delta\ell_1 \\ x_2 + u_2 + \frac{\partial u_2}{\partial x_1} \Delta\ell_1 \\ x_3 + u_3 + \frac{\partial u_3}{\partial x_1} \Delta\ell_1 \end{pmatrix}$$

Length of line segment PQ in the deformed shape:

$$\Delta\ell'_1 = \sqrt{\left(x_1^{(Q')} - x_1^{(P')}\right)^2 + \left(x_2^{(Q')} - x_2^{(P')}\right)^2 + \left(x_3^{(Q')} - x_3^{(P')}\right)^2} =$$

Coordinates of the translated points:

$$P': \begin{pmatrix} x_1 + u_1 \\ x_2 + u_2 \\ x_3 + u_3 \end{pmatrix} \quad Q': \begin{pmatrix} x_1 + \Delta\ell_1 + u_1 + \frac{\partial u_1}{\partial x_1} \Delta\ell_1 \\ x_2 + u_2 + \frac{\partial u_2}{\partial x_1} \Delta\ell_1 \\ x_3 + u_3 + \frac{\partial u_3}{\partial x_1} \Delta\ell_1 \end{pmatrix}$$

Length of line segment PQ in the deformed shape:

$$\begin{aligned}\Delta\ell'_1 &= \sqrt{\left(x_1^{(Q')} - x_1^{(P')}\right)^2 + \left(x_2^{(Q')} - x_2^{(P')}\right)^2 + \left(x_3^{(Q')} - x_3^{(P')}\right)^2} = \\ &= \sqrt{\left(\Delta\ell_1 + \frac{\partial u_1}{\partial x_1} \Delta\ell_1\right)^2 + \left(\frac{\partial u_2}{\partial x_1} \Delta\ell_1\right)^2 + \left(\frac{\partial u_3}{\partial x_1} \Delta\ell_1\right)^2} =\end{aligned}$$

Coordinates of the translated points:

$$P': \begin{pmatrix} x_1+u_1 \\ x_2+u_2 \\ x_3+u_3 \end{pmatrix} \quad Q': \begin{pmatrix} x_1+\Delta\ell_1+u_1+\frac{\partial u_1}{\partial x_1}\Delta\ell_1 \\ x_2+u_2+\frac{\partial u_2}{\partial x_1}\Delta\ell_1 \\ x_3+u_3+\frac{\partial u_3}{\partial x_1}\Delta\ell_1 \end{pmatrix}$$

Length of line segment PQ in the deformed shape:

$$\begin{aligned} \Delta\ell'_1 &= \sqrt{\left(x_1^{(Q')}-x_1^{(P')}\right)^2 + \left(x_2^{(Q')}-x_2^{(P')}\right)^2 + \left(x_3^{(Q')}-x_3^{(P')}\right)^2} = \\ &= \sqrt{\left(\Delta\ell_1+\frac{\partial u_1}{\partial x_1}\Delta\ell_1\right)^2 + \left(\frac{\partial u_2}{\partial x_1}\Delta\ell_1\right)^2 + \left(\frac{\partial u_3}{\partial x_1}\Delta\ell_1\right)^2} = \\ &= \Delta\ell_1 \sqrt{1 + 2\frac{\partial u_1}{\partial x_1} + \left(\frac{\partial u_1}{\partial x_1}\right)^2 + \left(\frac{\partial u_2}{\partial x_1}\right)^2 + \left(\frac{\partial u_3}{\partial x_1}\right)^2} \end{aligned}$$

Coordinates of the translated points:

$$P': \begin{pmatrix} x_1 + u_1 \\ x_2 + u_2 \\ x_3 + u_3 \end{pmatrix} \quad Q': \begin{pmatrix} x_1 + \Delta\ell_1 + u_1 + \frac{\partial u_1}{\partial x_1} \Delta\ell_1 \\ x_2 + u_2 + \frac{\partial u_2}{\partial x_1} \Delta\ell_1 \\ x_3 + u_3 + \frac{\partial u_3}{\partial x_1} \Delta\ell_1 \end{pmatrix}$$

Length of line segment PQ in the deformed shape:

$$\begin{aligned} \Delta\ell'_1 &= \sqrt{\left(x_1^{(Q')} - x_1^{(P')}\right)^2 + \left(x_2^{(Q')} - x_2^{(P')}\right)^2 + \left(x_3^{(Q')} - x_3^{(P')}\right)^2} = \\ &= \sqrt{\left(\Delta\ell_1 + \frac{\partial u_1}{\partial x_1} \Delta\ell_1\right)^2 + \left(\frac{\partial u_2}{\partial x_1} \Delta\ell_1\right)^2 + \left(\frac{\partial u_3}{\partial x_1} \Delta\ell_1\right)^2} = \\ &= \Delta\ell_1 \sqrt{1 + 2\frac{\partial u_1}{\partial x_1} + \left(\frac{\partial u_1}{\partial x_1}\right)^2 + \left(\frac{\partial u_2}{\partial x_1}\right)^2 + \left(\frac{\partial u_3}{\partial x_1}\right)^2} \\ &\approx \Delta\ell_1 \left(1 + \frac{\partial u_1}{\partial x_1}\right) \quad \text{if } \frac{\partial u_i}{\partial x_1} \ll 1 \end{aligned}$$

The strain in  $x_1$ -direction is:

$$\varepsilon_{11} = \frac{\Delta\ell'_1 - \Delta\ell_1}{\Delta\ell_1} \approx \frac{\Delta\ell_1 (1 + \frac{\partial u_1}{\partial x_1} - 1)}{\Delta\ell_1} = \frac{\partial u_1}{\partial x_1}$$

The strain in  $x_1$ -direction is:

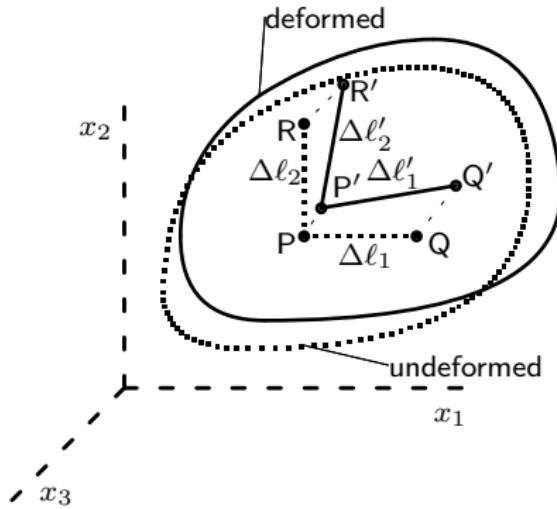
$$\varepsilon_{11} = \frac{\Delta\ell'_1 - \Delta\ell_1}{\Delta\ell_1} \approx \frac{\Delta\ell_1 (1 + \frac{\partial u_1}{\partial x_1} - 1)}{\Delta\ell_1} = \frac{\partial u_1}{\partial x_1}$$

Same for the other directions:

$$\varepsilon_{11} = \frac{\partial u_1}{\partial x_1} \quad \varepsilon_{22} = \frac{\partial u_2}{\partial x_2} \quad \varepsilon_{33} = \frac{\partial u_3}{\partial x_3}$$

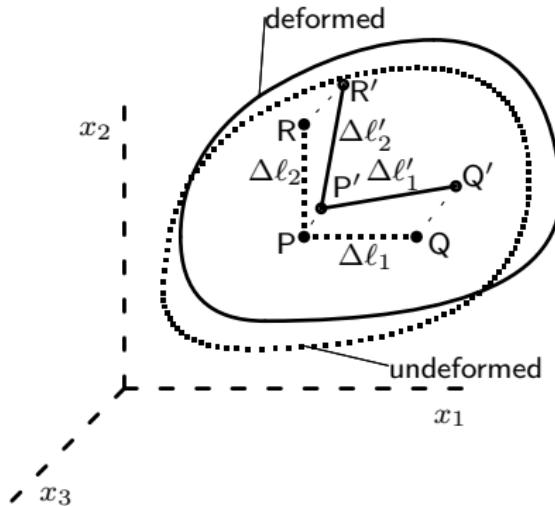
$$\text{if } \frac{\partial u_i}{\partial x_j} \ll 1 \text{ for } \begin{cases} i = 1, 2, 3 \\ j = 1, 2, 3 \end{cases}$$

## 4.2) Shear strain



Consider line segments PQ and PR in resp.  $x_1$ - and  $x_2$ -directions.

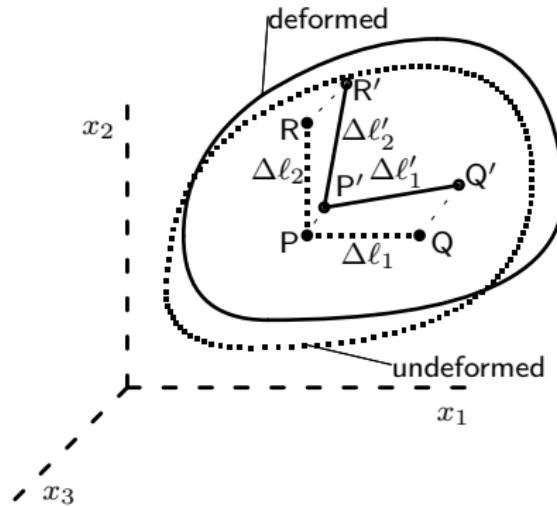
## 4.2) Shear strain



Consider line segments  $PQ$  and  $PR$  in resp.  $x_1$ - and  $x_2$ -directions.

The coordinates  
of point  $R$  are  $x_1, x_2 + \Delta\ell_2, x_3$

## 4.2) Shear strain



Consider line segments  $PQ$  and  $PR$  in resp.  $x_1$ - and  $x_2$ -directions.

The coordinates

of point  $R$  are  $x_1, x_2 + \Delta\ell_2, x_3$

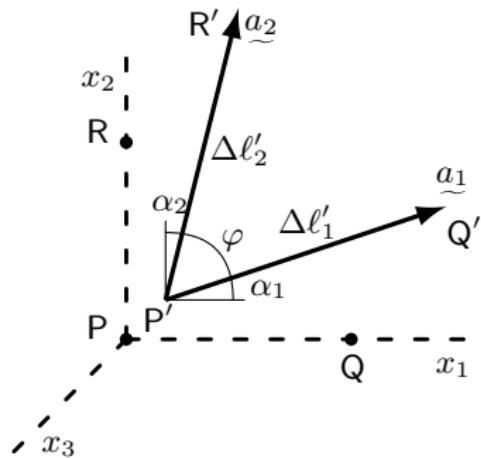
Besides

the translations of  $P \rightarrow P'$  and  $Q \rightarrow Q'$  we also have  $u_R$  from  $R \rightarrow R'$ ,

the components  $u_i(x_1, x_2 + \Delta\ell_2, x_3)$ .

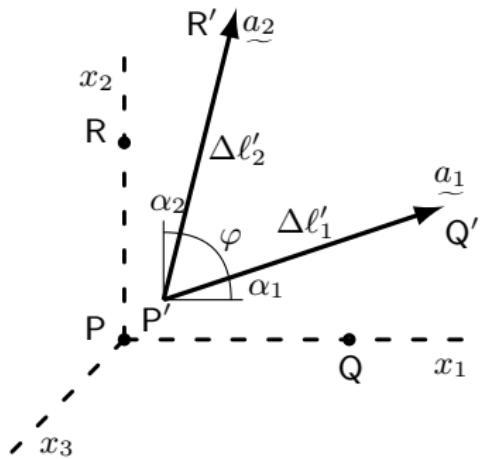
$$\text{with: } u_i(x_1, x_2 + \Delta\ell_2, x_3) = u_i(x_1, x_2, x_3) + \Delta\ell_2 \frac{\partial u_i}{\partial x_2} + \text{H.O.T.}$$

For the coordinates of the points  $P'$ ,  $Q'$  and  $R'$ , see reader.



The shear strain  $\gamma_{12}$  is the change in angle between  $PQ$  and  $PR$  compared to their original  $90^\circ$  angle.  
So:  $\gamma_{12} = \frac{1}{2}\pi - \varphi$

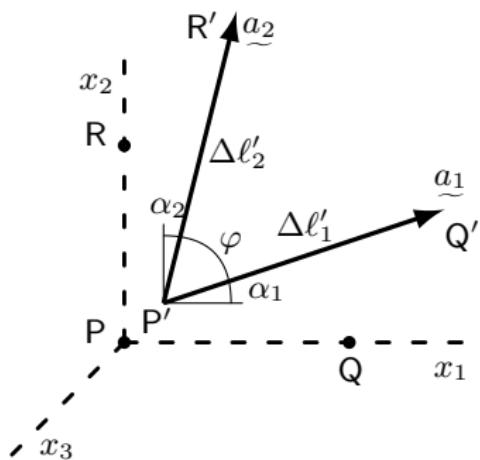
For the coordinates of the points  $P'$ ,  $Q'$  and  $R'$ , see reader.



The shear strain  $\gamma_{12}$  is the change in angle between  $PQ$  and  $PR$  compared to their original  $90^\circ$  angle.  
So:  $\gamma_{12} = \frac{1}{2}\pi - \varphi$

$$\begin{aligned}\underline{\underline{a}_1} \cdot \underline{\underline{a}_2} &= |\underline{\underline{a}_1}| |\underline{\underline{a}_2}| \cos(\varphi) \\ &= |\underline{\underline{a}_1}| |\underline{\underline{a}_2}| \cos\left(\frac{1}{2}\pi - \gamma_{12}\right) \\ &= |\underline{\underline{a}_1}| |\underline{\underline{a}_2}| \sin(\gamma_{12})\end{aligned}$$

For the coordinates of the points  $P'$ ,  $Q'$  and  $R'$ , see reader.



The shear strain  $\gamma_{12}$  is the change in angle between  $PQ$  and  $PR$  compared to their original  $90^\circ$  angle.  
So:  $\gamma_{12} = \frac{1}{2}\pi - \varphi$

$$\begin{aligned}\underline{\underline{a}}_1 \cdot \underline{\underline{a}}_2 &= |\underline{\underline{a}}_1| |\underline{\underline{a}}_2| \cos(\varphi) \\ &= |\underline{\underline{a}}_1| |\underline{\underline{a}}_2| \cos\left(\frac{1}{2}\pi - \gamma_{12}\right) \\ &= |\underline{\underline{a}}_1| |\underline{\underline{a}}_2| \sin(\gamma_{12})\end{aligned}$$

From which:

$$\gamma_{12} = \arcsin \left( \frac{\underline{\underline{a}}_1 \cdot \underline{\underline{a}}_2}{|\underline{\underline{a}}_1| |\underline{\underline{a}}_2|} \right)$$

After “some” calculations:

$$\boxed{\gamma_{12} = \frac{\partial u_1}{\partial x_2} + \frac{\partial u_2}{\partial x_1}} \quad \text{if } \frac{\partial u_i}{\partial x_j} \ll 1 \quad \text{for } \begin{cases} i = 1, 2, 3 \\ j = 1, 2, 3 \end{cases}$$

After "some" calculations:

$$\boxed{\gamma_{12} = \frac{\partial u_1}{\partial x_2} + \frac{\partial u_2}{\partial x_1}} \quad \text{if } \frac{\partial u_i}{\partial x_j} \ll 1 \quad \text{for } \begin{cases} i = 1, 2, 3 \\ j = 1, 2, 3 \end{cases}$$

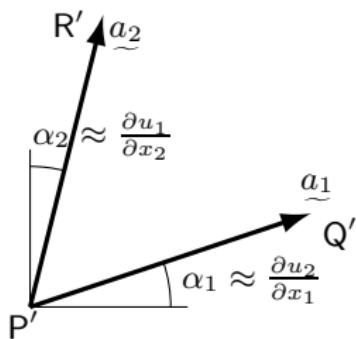
Shear strain  $\gamma_{12} = \alpha_1 + \alpha_2$

From the figure:  $\tan(\alpha_1) = \frac{\frac{\partial u_2}{\partial x_1} \Delta \ell_1}{\Delta \ell_1 + \frac{\partial u_1}{\partial x_1} \Delta \ell_1}$

As approximation:  $\alpha_1 \approx \frac{\partial u_2}{\partial x_1}$

And:  $\alpha_2 \approx \frac{\partial u_1}{\partial x_2}$

In general it holds that  $\alpha_1 \neq \alpha_2 !!!$



Results for 3 directions:

$$\gamma_{12} = \gamma_{21} = \frac{\partial u_1}{\partial x_2} + \frac{\partial u_2}{\partial x_1}$$

$$\gamma_{13} = \gamma_{31} = \frac{\partial u_1}{\partial x_3} + \frac{\partial u_3}{\partial x_1} \quad \text{if } \frac{\partial u_i}{\partial x_j} \ll 1 \quad \text{for } \begin{cases} i = 1, 2, 3 \\ j = 1, 2, 3 \end{cases}$$

$$\gamma_{23} = \gamma_{32} = \frac{\partial u_2}{\partial x_3} + \frac{\partial u_3}{\partial x_2}$$

Results for 3 directions:

$$\gamma_{12} = \gamma_{21} = \frac{\partial u_1}{\partial x_2} + \frac{\partial u_2}{\partial x_1}$$

$$\gamma_{13} = \gamma_{31} = \frac{\partial u_1}{\partial x_3} + \frac{\partial u_3}{\partial x_1} \quad \text{if } \frac{\partial u_i}{\partial x_j} \ll 1 \quad \text{for } \begin{cases} i = 1, 2, 3 \\ j = 1, 2, 3 \end{cases}$$

$$\gamma_{23} = \gamma_{32} = \frac{\partial u_2}{\partial x_3} + \frac{\partial u_3}{\partial x_2}$$

Summarised in index notation:

$$\boxed{\gamma_{ij} = \gamma_{ji} = \frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i}} \quad \text{with } i \neq j$$

## 4.3) Strain tensor

Strains and shear strains from the theory of *small deformations*:

$$\begin{aligned}\varepsilon_{11} &= \frac{\partial u_1}{\partial x_1} & \gamma_{12} = \gamma_{21} &= \frac{\partial u_1}{\partial x_2} + \frac{\partial u_2}{\partial x_1} \\ \varepsilon_{22} &= \frac{\partial u_2}{\partial x_2} & \gamma_{13} = \gamma_{31} &= \frac{\partial u_1}{\partial x_3} + \frac{\partial u_3}{\partial x_1} \\ \varepsilon_{33} &= \frac{\partial u_3}{\partial x_3} & \gamma_{23} = \gamma_{32} &= \frac{\partial u_2}{\partial x_3} + \frac{\partial u_3}{\partial x_2}\end{aligned}$$

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If  $i \neq j$  is used for  $\varepsilon_{ij} = \frac{1}{2} \gamma_{ij}$  then:

$$\varepsilon_{ij} = \frac{1}{2} \left( \frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right) = \frac{1}{2} (u_{i,j} + u_{j,i})$$

$$\text{for } \begin{cases} i = 1, 2, 3 \\ j = 1, 2, 3 \end{cases}$$

**Notation**  $\left( \varepsilon_{ij} = \frac{1}{2} \left( \frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right) = \frac{1}{2} (u_{i,j} + u_{j,i}) \right)$ :

$$\underline{\varepsilon} = \varepsilon_{ij} \hat{e}_i \hat{e}_j \quad [\varepsilon] = \begin{bmatrix} \varepsilon_{11} & \varepsilon_{12} & \varepsilon_{13} \\ \varepsilon_{21} & \varepsilon_{22} & \varepsilon_{23} \\ \varepsilon_{31} & \varepsilon_{32} & \varepsilon_{33} \end{bmatrix} = \begin{bmatrix} \varepsilon_{11} & \frac{1}{2}\gamma_{12} & \frac{1}{2}\gamma_{13} \\ \frac{1}{2}\gamma_{21} & \varepsilon_{22} & \frac{1}{2}\gamma_{23} \\ \frac{1}{2}\gamma_{31} & \frac{1}{2}\gamma_{32} & \varepsilon_{33} \end{bmatrix}$$

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### Outdated notation

$(x_1 \rightarrow x, \ x_2 \rightarrow y \text{ and } x_3 \rightarrow z, \ u_1 \rightarrow u, \ u_2 \rightarrow v \text{ and } u_3 \rightarrow w)$ :

$$\varepsilon_{xx} = \frac{\partial u}{\partial x} \ (= \varepsilon_x) \quad \varepsilon_{yy} = \frac{\partial v}{\partial y} \ (= \varepsilon_y) \quad \varepsilon_{zz} = \frac{\partial w}{\partial z} \ (= \varepsilon_z)$$

$$\varepsilon_{xy} = \frac{1}{2}\gamma_{xy} = \frac{1}{2} \left( \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \right) = \varepsilon_{yx} = \frac{1}{2}\gamma_{yx}$$

$$\varepsilon_{xz} = \frac{1}{2}\gamma_{xz} = \frac{1}{2} \left( \frac{\partial u}{\partial z} + \frac{\partial w}{\partial x} \right) = \varepsilon_{zx} = \frac{1}{2}\gamma_{zx}$$

$$\varepsilon_{yz} = \frac{1}{2}\gamma_{yz} = \frac{1}{2} \left( \frac{\partial v}{\partial z} + \frac{\partial w}{\partial y} \right) = \varepsilon_{zy} = \frac{1}{2}\gamma_{zy}$$

*Question:* Are  $\varepsilon_{ij}$  the components of a second order tensor?

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Apply rotation of coordinate axes,  $x_i \rightarrow x'_p$ :

$$\begin{aligned}x'_p &= x_1 \cos(x'_p, x_1) + x_2 \cos(x'_p, x_2) + x_3 \cos(x'_p, x_3) = \\&= x_1 R_{p1} + x_2 R_{p2} + x_3 R_{p3} = R_{pi} x_i\end{aligned}$$

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For the deformation components holds:

$$\begin{aligned}\varepsilon_{ij} &= \frac{1}{2} \left( \frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right) = \frac{1}{2} \left( \frac{\partial u_i}{\partial x'_q} \frac{\partial x'_q}{\partial x_j} + \frac{\partial u_j}{\partial x'_p} \frac{\partial x'_p}{\partial x_i} \right) = \\&= \frac{1}{2} \left( \frac{\partial u_i}{\partial x'_q} R_{qj} + \frac{\partial u_j}{\partial x'_p} R_{pi} \right)\end{aligned}$$

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Transformation components of displacement vector:  $u_i = R_{pi} u'_p \implies$

$$\begin{aligned}\varepsilon_{ij} &= \frac{1}{2} \left( R_{pi} \frac{\partial u'_p}{\partial x'_q} R_{qj} + R_{qj} \frac{\partial u'_q}{\partial x'_p} R_{pi} \right) = R_{pi} R_{qj} \frac{1}{2} \left( \frac{\partial u'_p}{\partial x'_q} + \frac{\partial u'_q}{\partial x'_p} \right) = \\&= R_{pi} R_{qj} \varepsilon'_{pq}\end{aligned}$$

Summarized: (from “new” to “old” and vice versa):

$$\varepsilon_{ij} = R_{pi} R_{qj} \varepsilon'_{pq}$$

and

$$\varepsilon'_{pq} = R_{pi} R_{qj} \varepsilon_{ij}$$

In matrix-vector-notation:

$$[\varepsilon] = [R]^T [\varepsilon'] [R]$$

and

$$[\varepsilon'] = [R] [\varepsilon] [R]^T$$

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$$\varepsilon_{ij} = R_{pi} R_{qj} \varepsilon'_{pq}$$

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In matrix-vector-notation:

$$[\varepsilon] = [R]^T [\varepsilon'] [R]$$

and

$$[\varepsilon'] = [R] [\varepsilon] [R]^T$$

So  $\varepsilon_{ij}$  are indeed components of a second order tensor. As well as the stress tensor, this tensor is symmetric; for the deformation matrix holds:  $[\varepsilon]^T = [\varepsilon]$ .

Also see the reader regarding the arbitrary oriented line segment PQ.

## 4.4) Principal strains and principal directions

Second order tensor, so principal directions follow from:

$$\det \begin{pmatrix} \varepsilon_{11} - \varepsilon & \varepsilon_{12} & \varepsilon_{13} \\ \varepsilon_{21} & \varepsilon_{22} - \varepsilon & \varepsilon_{23} \\ \varepsilon_{31} & \varepsilon_{32} & \varepsilon_{33} - \varepsilon \end{pmatrix} = \det([\varepsilon] - \varepsilon[I]) = 0$$

Resulting in the characteristic equation:  $\varepsilon^3 - E_1\varepsilon^2 + E_2\varepsilon - E_3 = 0$

## 4.4) Principal strains and principal directions

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Resulting in the characteristic equation:  $\varepsilon^3 - E_1\varepsilon^2 + E_2\varepsilon - E_3 = 0$

The *deformation invariants* are:

$$E_1 = \varepsilon_{11} + \varepsilon_{22} + \varepsilon_{33} = \varepsilon_{ii}$$

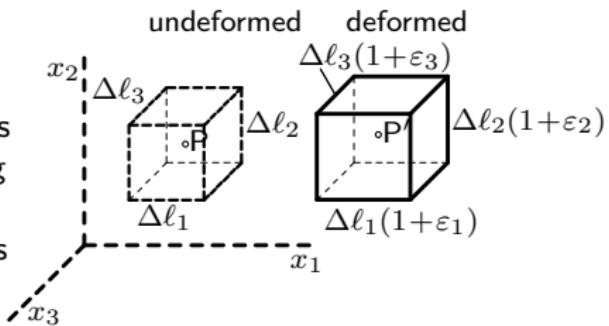
$$E_2 = \varepsilon_{11}\varepsilon_{22} - \varepsilon_{12}\varepsilon_{21} + \textcircled{O} = \frac{1}{2}(\varepsilon_{ii}\varepsilon_{jj} - \varepsilon_{ij}\varepsilon_{ji})$$

$$E_3 = \det([\varepsilon])$$

After solving we obtain the principal strains  $\varepsilon_1, \varepsilon_2, \varepsilon_3$  and the 3 orthogonal principal directions. The invariants can be expressed in terms of principal strains.

## 4.5) Volumetric strain

Consider a  $x_i$ -coordinate system oriented in such a way that it coincides with the principal directions of the deformed state. A corresponding infinitesimal volume element does not have any shear strain. The surfaces remain perpendicular to each other during deformation.



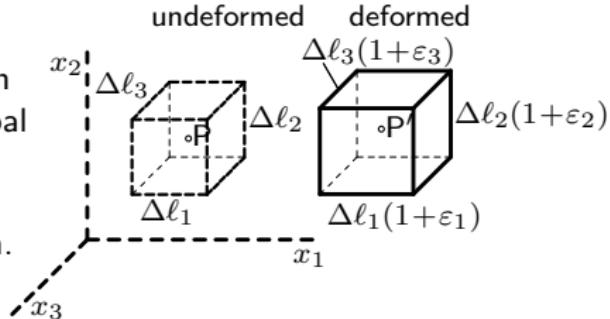
## 4.5) Volumetric strain

Consider

a  $x_i$ -coordinate system oriented in such a way that it coincides with the principal directions of the deformed state.

A corresponding infinitesimal volume element does not have any shear strain.

The surfaces remain perpendicular to each other during deformation.



$$\text{Volume undeformed: } V = \Delta\ell_1 \Delta\ell_2 \Delta\ell_3$$

$$\text{Volume deformed: } V' = \Delta\ell_1(1 + \varepsilon_1)\Delta\ell_2(1 + \varepsilon_2)\Delta\ell_3(1 + \varepsilon_3)$$

Volumetric strain:

$$\varepsilon_V = \frac{V' - V}{V} = (1 + \varepsilon_1)(1 + \varepsilon_2)(1 + \varepsilon_3) - 1 = \varepsilon_1 + \varepsilon_2 + \varepsilon_3 + \text{H.O.T.}$$

For small strains we ignore H.O.T. (also including the third invariant  $E_3$ ).

$$\Rightarrow \varepsilon_V = E_1 = \varepsilon_1 + \varepsilon_2 + \varepsilon_3 = \varepsilon_{11} + \varepsilon_{22} + \varepsilon_{33} \quad \text{or} \quad \boxed{\varepsilon_V = \varepsilon_{kk} = E_1}$$

## 4.6) Engineering strain/stress vs. true strain/stress

Note that:

engineering strain  $\varepsilon = \frac{\Delta L}{L_0}$  is defined relative to the original length,  $L_0$ ,  
while stress  $\sigma = \frac{F}{A_0}$  is defined relative to the original cross-section,  $A_0$ .

Neither can be assumed constant during deformation,  
so that a true strain increment becomes  $\delta\varepsilon_t = \frac{\delta L}{L(t)}$ ,  
defined relative to the momentary (at time  $t$ ) size of the sample,  $L(t)$ ,  
while the true stress becomes  $\sigma_t = \frac{F(t)}{A(t)}$ , with momentary cross-section  $A(t)$ .

Integrating the true strain increment  $\delta\varepsilon_t$  from 0 to  $\varepsilon_t$ ,  
and the right hand side from  $L_0$  to  $L(t)$ ,  
using  $\ln L(t) - \ln L(0) = \ln(L(t)/L_0)$ , with  $L(t) = L_0 + \Delta L$ ,  
yields the true strain  $\varepsilon_t = \ln(1 + \varepsilon)$ .

## 5) Material behavior

### 5.1) Hooke's law

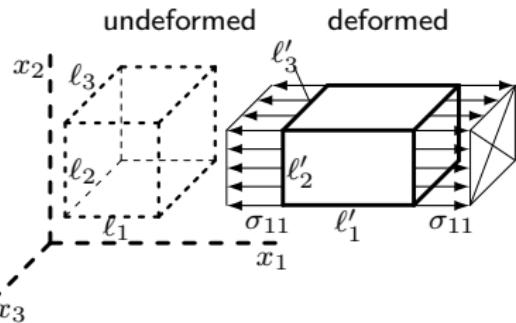
Linear elastic, isotropic material behavior.

$$\text{Strain: } \varepsilon_{11} = \frac{\ell'_1 - \ell_1}{\ell_1}$$

$$\text{Relation stress and strain: } \varepsilon_{11} = \frac{\sigma_{11}}{E}$$

E is the

Young's modulus (modulus of elasticity)



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### 5.1) Hooke's law

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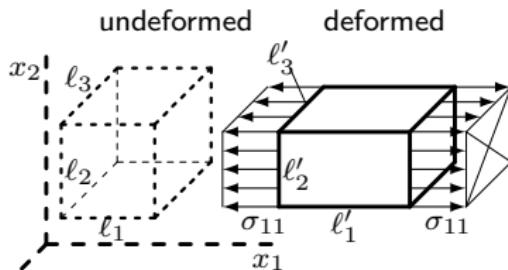
E is the

Young's modulus (modulus of elasticity)

$$\text{Lateral strain: } \varepsilon_{22} = \frac{\ell'_2 - \ell_2}{\ell_2} \leq 0 \quad \text{en} \quad \varepsilon_{33} = \frac{\ell'_3 - \ell_3}{\ell_3} \leq 0$$

$$\text{Relation lateral strain and stresses: } \varepsilon_{22} = \varepsilon_{33} = -\nu \frac{\sigma_{11}}{E}$$

$\nu$  is the Poisson's ratio



## Isotropic material behavior!!!

So, with a single normal stress  $\sigma_{11}$  (other stress components are zero!):

$$\varepsilon_{11} = \frac{\sigma_{11}}{E} \quad \text{and} \quad \varepsilon_{22} = \varepsilon_{33} = -\nu \frac{\sigma_{11}}{E}$$

The shear strain seems to remain equal to zero.

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The shear strain seems to remain equal to zero.

Also, when we apply a single normal stress  $\sigma_{22}$ , again with the other stress components equal to zero:

$$\varepsilon_{22} = \frac{\sigma_{22}}{E} \quad \text{and} \quad \varepsilon_{11} = \varepsilon_{33} = -\nu \frac{\sigma_{22}}{E}$$

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Likewise for normal stress  $\sigma_{33}$ :

$$\varepsilon_{33} = \frac{\sigma_{33}}{E} \quad \text{and} \quad \varepsilon_{11} = \varepsilon_{22} = -\nu \frac{\sigma_{33}}{E}$$

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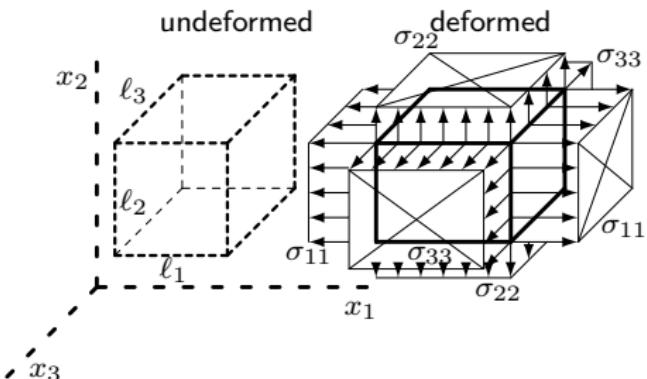
Also a combined stress state is possible. Superposition principle.

## Result:

$$\varepsilon_{11} = \frac{\sigma_{11}}{E} - \nu \frac{\sigma_{22}}{E} - \nu \frac{\sigma_{33}}{E}$$

$$\varepsilon_{22} = -\nu \frac{\sigma_{11}}{E} + \frac{\sigma_{22}}{E} - \nu \frac{\sigma_{33}}{E}$$

$$\varepsilon_{33} = -\nu \frac{\sigma_{11}}{E} - \nu \frac{\sigma_{22}}{E} + \frac{\sigma_{33}}{E}$$



This is Hooke's law in the case that only normal stresses are present!

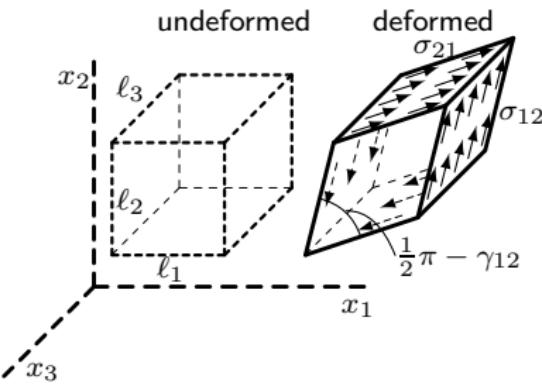
Shear, shear stresses and shear strain:

$$2\varepsilon_{12} = \gamma_{12} = \frac{\sigma_{12}}{G} = \frac{\sigma_{21}}{G}$$

$$2\varepsilon_{13} = \gamma_{13} = \frac{\sigma_{13}}{G} = \frac{\sigma_{31}}{G}$$

$$2\varepsilon_{23} = \gamma_{23} = \frac{\sigma_{23}}{G} = \frac{\sigma_{32}}{G}$$

The material constant  $G$  is the *shear modulus*.



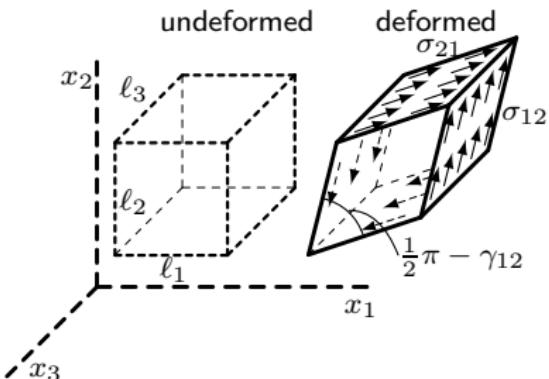
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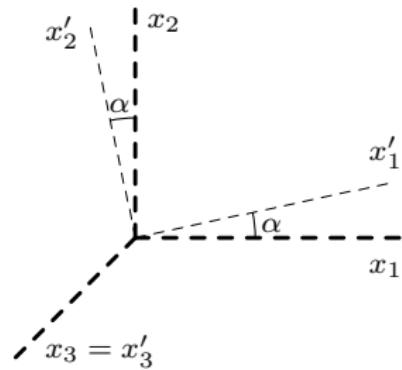
**NOTE:** With isotropic material behavior  $E$ ,  $G$  and  $\nu$  are correlated with each other. This correlation can be found via a coordinate system rotation.

## Linear stress state

$$[\sigma]_{x_i} = \begin{bmatrix} \sigma_{11} & 0 & 0 \\ 0 & 0 & 0 \\ 0 & 0 & 0 \end{bmatrix}$$

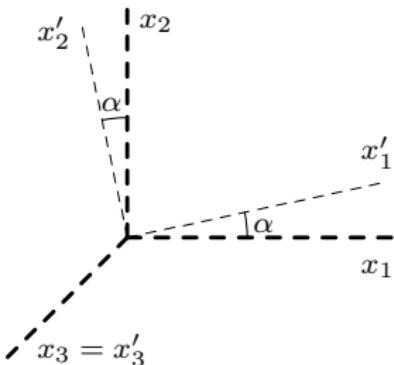
From Hooke's law follows:

$$\varepsilon_{11} = \frac{\sigma_{11}}{E} \quad ; \quad \varepsilon_{22} = \varepsilon_{33} = -\nu \frac{\sigma_{11}}{E}$$



## Linear stress state

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From Hooke's law follows:

$$\varepsilon_{11} = \frac{\sigma_{11}}{E} \quad ; \quad \varepsilon_{22} = \varepsilon_{33} = -\nu \frac{\sigma_{11}}{E}$$

so:

$$[\varepsilon] = \frac{1}{E} \begin{bmatrix} \sigma_{11} & 0 & 0 \\ 0 & -\nu\sigma_{11} & 0 \\ 0 & 0 & -\nu\sigma_{11} \end{bmatrix} \quad \text{and} \quad [R] = \begin{bmatrix} \cos \alpha & \sin \alpha & 0 \\ -\sin \alpha & \cos \alpha & 0 \\ 0 & 0 & 1 \end{bmatrix}$$

Solve for  $\sigma'_{12}$  with  $\sigma'_{12} = R_{1p} R_{q2} \sigma_{pq}$  or with  $[\sigma'] = [R][\sigma][R]^T$ :

$$\begin{bmatrix} \diamond & \sigma'_{12} & \diamond \\ \diamond & \diamond & \diamond \\ \diamond & \diamond & \diamond \end{bmatrix} = \begin{bmatrix} \cos \alpha & \sin \alpha & 0 \\ \diamond & \diamond & \diamond \\ \diamond & \diamond & \diamond \end{bmatrix} \begin{bmatrix} \sigma_{11} & 0 & 0 \\ 0 & 0 & 0 \\ 0 & 0 & 0 \end{bmatrix} \begin{bmatrix} \diamond & -\sin \alpha & \diamond \\ \diamond & \cos \alpha & \diamond \\ \diamond & 0 & \diamond \end{bmatrix}$$

All non-relevant values for the matrix multiplication are shown as a  $\diamond$ .

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All non-relevant values for the matrix multiplication are shown as a  $\diamond$ . Or shorter:

$$\sigma'_{12} = [\cos \alpha \quad \sin \alpha \quad 0] \begin{bmatrix} \sigma_{11} & 0 & 0 \\ 0 & 0 & 0 \\ 0 & 0 & 0 \end{bmatrix} \begin{bmatrix} -\sin \alpha \\ \cos \alpha \\ 0 \end{bmatrix} = -\sigma_{11} \cos \alpha \sin \alpha$$

The same holds for shear strain  $\varepsilon'_{12}$ . From  $[\varepsilon'] = [R][\varepsilon][R]^T$  follows:

$$\begin{aligned}\varepsilon'_{12} &= \begin{bmatrix} \cos \alpha & \sin \alpha & 0 \end{bmatrix} \frac{1}{E} \begin{bmatrix} \sigma_{11} & 0 & 0 \\ 0 & -\nu \sigma_{11} & 0 \\ 0 & 0 & -\nu \sigma_{11} \end{bmatrix} \begin{bmatrix} -\sin \alpha \\ \cos \alpha \\ 0 \end{bmatrix} = \\ &= -\frac{(1+\nu)}{E} \sigma_{11} \cos \alpha \sin \alpha\end{aligned}$$

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Due to isotropic behavior it must hold that:

$$\varepsilon'_{12} = \frac{1}{2} \gamma'_{12} = \frac{1}{2} \frac{\sigma'_{12}}{G} = -\frac{1}{2G} \sigma_{11} \cos \alpha \sin \alpha \quad (\sigma'_{12} = -\sigma_{11} \cos \alpha \sin \alpha)$$

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When we compare both expressions, the following important equation is found:

$$G = \frac{E}{2(1+\nu)}$$

**NOTE:** This equation only holds for *isotropic* material behavior!

**Summary of Hooke's law:**

$$\varepsilon_{11} = \frac{1}{E}(\sigma_{11} - \nu\sigma_{22} - \nu\sigma_{33}) ; \quad \gamma_{12} = 2\varepsilon_{12} = \frac{1}{G}\sigma_{12} \quad (\text{also } 12 \rightarrow 21)$$

$$\varepsilon_{22} = \frac{1}{E}(\sigma_{22} - \nu\sigma_{33} - \nu\sigma_{11}) ; \quad \gamma_{13} = 2\varepsilon_{13} = \frac{1}{G}\sigma_{13} \quad (\text{also } 13 \rightarrow 31)$$

$$\varepsilon_{33} = \frac{1}{E}(\sigma_{33} - \nu\sigma_{11} - \nu\sigma_{22}) ; \quad \gamma_{23} = 2\varepsilon_{23} = \frac{1}{G}\sigma_{23} \quad (\text{also } 23 \rightarrow 32)$$

Hooke's law in index notation. For strains:

$$\begin{aligned}\varepsilon_{11} &= \frac{\sigma_{11}}{E} - \nu \frac{\sigma_{22}}{E} - \nu \frac{\sigma_{33}}{E} = \frac{(1+\nu)}{E} \sigma_{11} - \frac{\nu}{E} (\sigma_{11} + \sigma_{22} + \sigma_{33}) = \\ &= \frac{(1+\nu)}{E} \left( \sigma_{11} - \frac{\nu}{(1+\nu)} \sigma_{kk} \right)\end{aligned}$$

Hooke's law in index notation. For strains:

$$\begin{aligned}\varepsilon_{11} &= \frac{\sigma_{11}}{E} - \nu \frac{\sigma_{22}}{E} - \nu \frac{\sigma_{33}}{E} = \frac{(1+\nu)}{E} \sigma_{11} - \frac{\nu}{E} (\sigma_{11} + \sigma_{22} + \sigma_{33}) = \\ &= \frac{(1+\nu)}{E} \left( \sigma_{11} - \frac{\nu}{(1+\nu)} \sigma_{kk} \right)\end{aligned}$$

$$\begin{aligned}\varepsilon_{22} &= \frac{\sigma_{22}}{E} - \nu \frac{\sigma_{33}}{E} - \nu \frac{\sigma_{11}}{E} = \frac{(1+\nu)}{E} \left( \sigma_{22} - \frac{\nu}{(1+\nu)} \sigma_{kk} \right) \\ \varepsilon_{33} &= \frac{\sigma_{33}}{E} - \nu \frac{\sigma_{11}}{E} - \nu \frac{\sigma_{22}}{E} = \frac{(1+\nu)}{E} \left( \sigma_{33} - \frac{\nu}{(1+\nu)} \sigma_{kk} \right)\end{aligned}$$

Hooke's law in index notation. For normal strains:

$$\begin{aligned}\varepsilon_{11} &= \frac{\sigma_{11}}{E} - \nu \frac{\sigma_{22}}{E} - \nu \frac{\sigma_{33}}{E} = \frac{(1+\nu)}{E} \sigma_{11} - \frac{\nu}{E} (\sigma_{11} + \sigma_{22} + \sigma_{33}) = \\ &= \frac{(1+\nu)}{E} \left( \sigma_{11} - \frac{\nu}{(1+\nu)} \sigma_{kk} \right)\end{aligned}$$

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So:

$$\varepsilon_{ij} = \frac{(1+\nu)}{E} \left( \sigma_{ij} - \frac{\nu}{(1+\nu)} \sigma_{kk} \right) \quad \text{if } i = j$$

For shear strains:

$$\begin{aligned}\varepsilon_{12} &= \frac{\sigma_{12}}{2G} = \frac{(1+\nu)}{E} \sigma_{12} & ; \quad \varepsilon_{13} &= \frac{\sigma_{13}}{2G} = \frac{(1+\nu)}{E} \sigma_{13} \\ \varepsilon_{23} &= \frac{\sigma_{23}}{2G} = \frac{(1+\nu)}{E} \sigma_{23}\end{aligned}$$

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So:

$$\varepsilon_{ij} = \frac{(1+\nu)}{E} \sigma_{ij} \text{ if } i \neq j \text{ and } \varepsilon_{ij} = \frac{(1+\nu)}{E} \left( \sigma_{ij} - \frac{\nu}{(1+\nu)} \sigma_{kk} \right) \text{ if } i = j$$

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Combined they result in Hooke's law in index notation (  $\sigma \rightarrow \varepsilon$  ) :

$$\boxed{\varepsilon_{ij} = \frac{(1+\nu)}{E} \left( \sigma_{ij} - \frac{\nu}{(1+\nu)} \sigma_{kk} \delta_{ij} \right) = \frac{1}{2G} \left( \sigma_{ij} - \frac{\nu}{(1+\nu)} \sigma_{kk} \delta_{ij} \right)}$$

for  $i = 1, 2, 3$  ;  $j = 1, 2, 3$ .

The *inverse*, stresses expressed in deformations.

We found:  $\varepsilon_{ij} = \frac{(1+\nu)}{E} \left( \sigma_{ij} - \frac{\nu}{(1+\nu)} \sigma_{kk} \delta_{ij} \right)$

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A summation of normal strains gives:

$$\begin{aligned}\varepsilon_{kk} &= \varepsilon_{11} + \varepsilon_{22} + \varepsilon_{33} = \frac{(1+\nu)}{E} \left( \sigma_{11} + \sigma_{22} + \sigma_{33} - 3 \frac{\nu}{(1+\nu)} \sigma_{kk} \right) = \\ &= \frac{(1+\nu)}{E} \left( \sigma_{kk} - 3 \frac{\nu}{(1+\nu)} \sigma_{kk} \right) = \frac{(1-2\nu)}{E} \sigma_{kk}\end{aligned}$$

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Rewriting Hooke's law:

$$\sigma_{ij} = \frac{E}{(1+\nu)} \varepsilon_{ij} + \frac{\nu}{(1+\nu)} \sigma_{kk} \delta_{ij} = \frac{E}{(1+\nu)} \varepsilon_{ij} + \frac{\nu}{(1+\nu)} \frac{E}{(1-2\nu)} \varepsilon_{kk} \delta_{ij}$$

Resulting in: Hooke's law      ( $\varepsilon \rightarrow \sigma$ ) :

$$\sigma_{ij} = \frac{E}{(1+\nu)} \left( \varepsilon_{ij} + \frac{\nu}{(1-2\nu)} \varepsilon_{kk} \delta_{ij} \right) = 2G \left( \varepsilon_{ij} + \frac{\nu}{(1-2\nu)} \varepsilon_{kk} \delta_{ij} \right)$$

for  $i = 1, 2, 3$  ;  $j = 1, 2, 3$ .

Resulting in; Hooke's law      ( $\varepsilon \rightarrow \sigma$ ) :

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for  $i = 1, 2, 3$  ;  $j = 1, 2, 3$ .

Notation according to Lamé:

$$\sigma_{ij} = \lambda \varepsilon_{kk} \delta_{ij} + 2\mu \varepsilon_{ij}$$

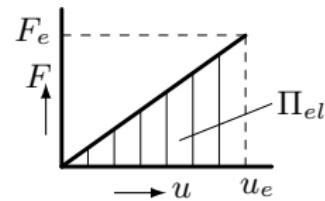
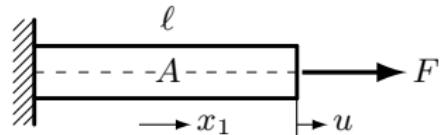
With the so called Lamé's (material) constants:

$$\lambda = \frac{\nu E}{(1+\nu)(1-2\nu)} \quad \text{en} \quad \mu = G = \frac{E}{2(1+\nu)}$$

## 5.2) Elastic energy

Work:

$$\Pi_{el} = \int_0^e F \, du = \frac{1}{2} F_e u_e$$



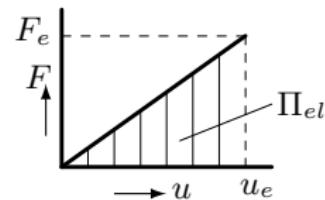
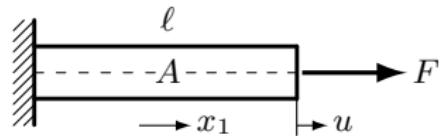
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At the end:

$$\sigma_{11} = \frac{F_e}{A} \quad \text{en} \quad \varepsilon_{11} = \frac{u_e}{\ell}$$



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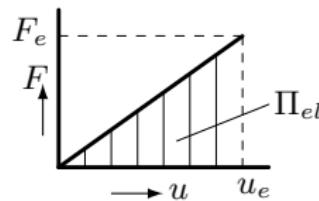
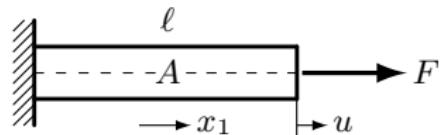
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At the end:

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Such that:

$$\Pi_{el} = \frac{1}{2} A \sigma_{11} \ell \varepsilon_{11} = \frac{1}{2} \sigma_{11} \varepsilon_{11} V$$



## 5.2) Elastic energy

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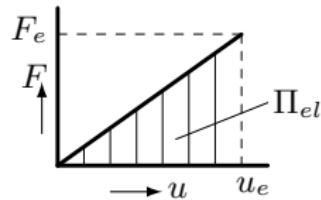
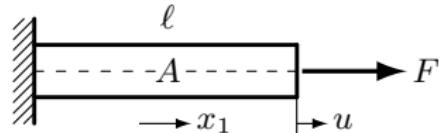
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*Specific internal elastic energy:*  $\pi_{el} = \frac{1}{2} \sigma_{11} \varepsilon_{11}$



## 5.2) Elastic energy

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At the end:

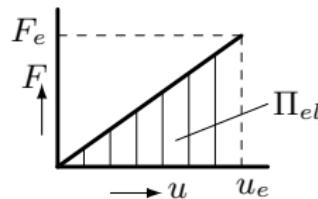
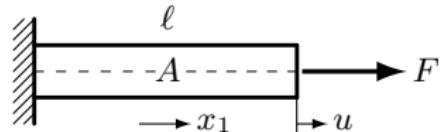
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Contribution to  $\pi_{el}$  during shearing:  $\frac{1}{2} \sigma_{12} \gamma_{12}$



An arbitrary three dimensional stress state:

$$\pi_{el} = \frac{1}{2}\sigma_{11}\varepsilon_{11} + \frac{1}{2}\sigma_{22}\varepsilon_{22} + \frac{1}{2}\sigma_{33}\varepsilon_{33} + \frac{1}{2}\sigma_{12}\gamma_{12} + \frac{1}{2}\sigma_{13}\gamma_{13} + \frac{1}{2}\sigma_{23}\gamma_{23}$$

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Written differently (with  $\gamma_{ij} = 2\varepsilon_{ij} = 2\varepsilon_{ji}$ ):

$$\begin{aligned}\pi_{el} &= \frac{1}{2}\sigma_{11}\varepsilon_{11} + \frac{1}{2}\sigma_{12}\varepsilon_{12} + \frac{1}{2}\sigma_{13}\varepsilon_{13} + \\ &+ \frac{1}{2}\sigma_{21}\varepsilon_{21} + \frac{1}{2}\sigma_{22}\varepsilon_{22} + \frac{1}{2}\sigma_{23}\varepsilon_{23} + \\ &+ \frac{1}{2}\sigma_{31}\varepsilon_{31} + \frac{1}{2}\sigma_{32}\varepsilon_{32} + \frac{1}{2}\sigma_{33}\varepsilon_{33}\end{aligned}$$

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So, for the *specific elastic energy* holds (summation convention!!):

$$\boxed{\pi_{el} = \frac{1}{2}\sigma_{ij}\varepsilon_{ij}}$$

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So, for the *specific elastic energy* holds (summation convention!!):

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Elastic energy of a body with volume  $V$ :

$$\Pi_{el} = \int_V \pi_{el} \, dV = \int_V \frac{1}{2}\sigma_{ij}\varepsilon_{ij} \, dV$$

By filling in Hooke's law in the equation of the specific elastic energy  $\pi_{el} = \frac{1}{2}\sigma_{ij}\varepsilon_{ij}$ , either the stresses or the strains can be eliminated.

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The *specific elastic volume-change energy* is:

$$\boxed{\pi_{el,vol} = \frac{1}{2}\sigma_m\varepsilon_V} \quad \text{with}$$

$$\sigma_m = \frac{1}{3}(\sigma_{11} + \sigma_{22} + \sigma_{33}) = \frac{1}{3}\sigma_{kk} \quad \text{and} \quad \varepsilon_V = \varepsilon_{kk}$$

With Hooke's law we have seen that:

$$\varepsilon_V = \varepsilon_{kk} = \frac{(1-2\nu)}{E} \sigma_{kk} = \frac{3(1-2\nu)}{E} \sigma_m = \frac{\sigma_m}{C} \quad (\sigma_m = C \varepsilon_V)$$

Where  $C$  is the bulk modulus.

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Where  $C$  is the bulk modulus. Filling in  $\pi_{el_{vol}}$ :

$$\pi_{el_{vol}} = \frac{1}{2} \sigma_m \varepsilon_V = \frac{1}{2} \sigma_m \frac{3(1-2\nu)}{E} \sigma_m = \frac{1}{2} \frac{(1-2\nu)}{3E} \sigma_{hh} \sigma_{kk}$$

Turns out to be invariant for coordinate system transformations.

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Turns out to be invariant for coordinate system transformations.

The *specific elastic shape-change (or distortional) energy*:

$$\begin{aligned}\pi_{el_{shape}} &= \pi_{el} - \pi_{el_{vol}} = \dots = \\ &= \frac{1}{6} \frac{(1+\nu)}{E} \{(\sigma_1 - \sigma_2)^2 + (\sigma_1 - \sigma_3)^2 + (\sigma_2 - \sigma_3)^2\}\end{aligned}$$

### 5.3) Deviatoric stresses and strains

Deviatoric stresses:  $\hat{\sigma}_{ij} = \sigma_{ij} - \sigma_m \delta_{ij} = \sigma_{ij} - \frac{1}{3} \sigma_{kk} \delta_{ij}$

Deviatoric strains:  $\hat{\varepsilon}_{ij} = \varepsilon_{ij} - \frac{1}{3} \varepsilon_V \delta_{ij} = \varepsilon_{ij} - \frac{1}{3} \varepsilon_{kk} \delta_{ij}$

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Deviatoric strains:  $\hat{\varepsilon}_{ij} = \varepsilon_{ij} - \frac{1}{3} \varepsilon_V \delta_{ij} = \varepsilon_{ij} - \frac{1}{3} \varepsilon_{kk} \delta_{ij}$

Also the deviatoric stresses have principal stresses and principal directions. The corresponding coefficients in the characteristic equation

$$\hat{\sigma}^3 - J_2 \hat{\sigma} - J_3 = 0$$

i.e., the deviatoric invariants are  $J_1 = 0$ ,  $J_2 > 0$  and  $J_3$ .

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It turns out that: **Law of Hooke:**  $\hat{\sigma}_{ij} = 2G \hat{\varepsilon}_{ij}$

$$\pi_{el\_shape} = \frac{1}{6} \frac{(1+\nu)}{E} \{(\sigma_1 - \sigma_2)^2 + (\sigma_1 - \sigma_3)^2 + (\sigma_2 - \sigma_3)^2\} = \frac{(1+\nu)}{E} J_2$$

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And:  $(\pi_{el} = \frac{1}{2} \sigma_{ij} \varepsilon_{ij} ; \quad \pi_{el\_vol} = \frac{1}{2} \sigma_m \varepsilon_V)$

$$\boxed{\pi_{el\_shape} = \frac{1}{2} \frac{(1+\nu)}{E} \hat{\sigma}_{ij} \hat{\sigma}_{ij} = \frac{1}{2} \frac{1}{2G} \hat{\sigma}_{ij} \hat{\sigma}_{ij} = \frac{1}{2} \hat{\sigma}_{ij} \hat{\varepsilon}_{ij}}$$

## 5.4) Failure criteria

1) Tresca, bases on maximum shear stress.

$$\sigma_{\text{eq}} = 2 \tau_{\max} = \max\{|\sigma_1 - \sigma_2|, |\sigma_1 - \sigma_3|, |\sigma_2 - \sigma_3|\}$$

Or, if  $\sigma_1 \geq \sigma_2 \geq \sigma_3$ :

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Or, if  $\sigma_1 \geq \sigma_2 \geq \sigma_3$ :

$$\sigma_{\text{eq}} = 2 \tau_{\max} = (\sigma_1 - \sigma_3)$$

2) Von Mises, based on the maximum specific elastic distortional energy.

$$\begin{aligned}\sigma_{\text{eq}} &= \sqrt{3 \frac{E}{1+\nu} \pi_{el\_shape}} = \sqrt{3 J_2} = \\ &= \sqrt{\frac{1}{2} \{(\sigma_1 - \sigma_2)^2 + (\sigma_1 - \sigma_3)^2 + (\sigma_2 - \sigma_3)^2\}}\end{aligned}$$

Shear stress criteria

Tresca, Coulomb, Quest, Mohr

Maximum shear stress is leading.

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Comparison 3-dim stress state with 1-dim stress state in a tensile test.  $(\sigma_1 = \sigma_{\text{eq}} ; \sigma_2 = \sigma_3 = 0)$

$$\tau_{\max} = \frac{1}{2}(\sigma_1 - \sigma_3) \quad (3\text{-dim})$$

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$$\sigma_{\text{eq}} \leq \bar{\sigma} \quad (\text{allowable stress})$$

Elastic distortional energy criteria  
von Mises, Huber, Hencky  
**Elastic distortional energy is leading.**

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$$\pi_{el\_shape} = \frac{1}{6} \frac{(1+\nu)}{E} \{ (\sigma_1 - \sigma_2)^2 + (\sigma_1 - \sigma_3)^2 + (\sigma_2 - \sigma_3)^2 \} \quad (3\text{-dim})$$

$$\pi_{el\_shape} = \frac{1}{3} \frac{(1+\nu)}{E} \sigma_{\text{eq}}^2 \quad (1\text{-dim})$$

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Elastic distortional energy is leading.

Comparison 3-dim stress state with 1-dim stress state in a tensile test.  $(\sigma_1 = \sigma_{\text{eq}} ; \sigma_2 = \sigma_3 = 0)$

$$\pi_{el\_shape} = \frac{1}{6} \frac{(1+\nu)}{E} \{ (\sigma_1 - \sigma_2)^2 + (\sigma_1 - \sigma_3)^2 + (\sigma_2 - \sigma_3)^2 \} \quad (3\text{-dim})$$

$$\pi_{el\_shape} = \frac{1}{3} \frac{(1+\nu)}{E} \sigma_{\text{eq}}^2 \quad (1\text{-dim})$$

$$\rightarrow \sigma_{\text{eq}} = \sqrt{\frac{1}{2} \{ (\sigma_1 - \sigma_2)^2 + (\sigma_1 - \sigma_3)^2 + (\sigma_2 - \sigma_3)^2 \}}$$

$$\sigma_{\text{eq}} \leq \bar{\sigma} \quad (\text{allowable stress})$$

## Special case

Determine the equivalent stresses of a beam with circular cross section. A bending and torsional moment are acting on the beam (Hibbeler pg.195 (5.7)  $T\rho/J$ ):

$$\sigma_{11} = \frac{M_b R}{I_b} \quad \text{with} \quad I_b = \frac{\pi}{4} R^4$$

$$\sigma_{12} = \frac{M_w R}{I_p} \quad \text{with} \quad I_p = \frac{\pi}{2} R^4 = 2 I_b$$

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The complete stress state becomes:

$$[\sigma] = \begin{bmatrix} \sigma_{11} & \sigma_{12} & 0 \\ \sigma_{12} & 0 & 0 \\ 0 & 0 & 0 \end{bmatrix} \quad \text{Principal stresses?}$$

$$\det \begin{pmatrix} \sigma_{11} - \sigma & \sigma_{12} & 0 \\ \sigma_{12} & -\sigma & 0 \\ 0 & 0 & -\sigma \end{pmatrix} = 0 \quad \rightarrow$$

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Solving determinant:  $(-\sigma)\{(\sigma_{11} - \sigma)(-\sigma) - \sigma_{12}^2\} = 0 \rightarrow$

$$\sigma_1 = 0 ; \quad \sigma^2 - \sigma_{11}\sigma - \sigma_{12}^2 = 0 \rightarrow \sigma_{2,3} = \frac{1}{2}\sigma_{11} \pm \frac{1}{2}\sqrt{\sigma_{11}^2 + 4\sigma_{12}^2}$$

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The solutions in the right order:

$$\sigma_1 = \frac{1}{2}\left(\sigma_{11} + \sqrt{\sigma_{11}^2 + 4\sigma_{12}^2}\right); \sigma_2 = 0; \sigma_3 = \frac{1}{2}\left(\sigma_{11} - \sqrt{\sigma_{11}^2 + 4\sigma_{12}^2}\right)$$

We just found:

$$\sigma_1 = \frac{1}{2} \left( \sigma_{11} + \sqrt{\sigma_{11}^2 + 4\sigma_{12}^2} \right); \sigma_2 = 0; \sigma_3 = \frac{1}{2} \left( \sigma_{11} - \sqrt{\sigma_{11}^2 + 4\sigma_{12}^2} \right)$$

Tresca:  $\sigma_{\text{eq}} = \sigma_1 - \sigma_3 = \sqrt{\sigma_{11}^2 + 4\sigma_{12}^2}$

von Mises:  $\sigma_{\text{eq}} = \sqrt{\frac{1}{2} \{ (\sigma_1 - \sigma_2)^2 + (\sigma_1 - \sigma_3)^2 + (\sigma_2 - \sigma_3)^2 \}} =$   
 $= \sqrt{\sigma_{11}^2 + 3\sigma_{12}^2}$

Note the difference!!

We just found:

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Note the difference!! Expressed in the bending and torsional moment:

Tresca:  $\sigma_{\text{eq}} = \frac{R}{I_b} \sqrt{M_b^2 + M_w^2}$

von Mises:  $\sigma_{\text{eq}} = \frac{R}{I_b} \sqrt{M_b^2 + \frac{3}{4}M_w^2}$

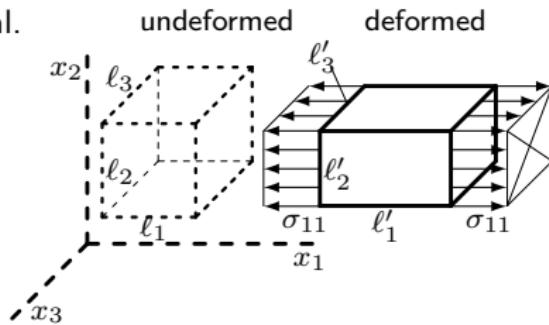
## 6) General material behavior

### 6.1) Hooke's law $\Rightarrow$ viscosity

Linear elastic, isotropic viscous material.

In the special case of 1D:

$$\text{Strain: } \varepsilon_{11} = \frac{\ell'_1 - \ell_1}{\ell_1}$$



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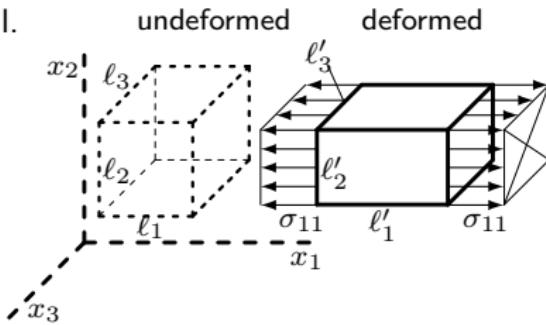
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$$\text{Strain: } \varepsilon_{11} = \frac{\ell'_1 - \ell_1}{\ell_1}$$

$$\text{Strain-rate: } \dot{\varepsilon}_{11} = \frac{\ell'_1 - \ell_1}{\ell_1 \Delta t}$$

Relation between stress and strain:

$$\sigma_{11} = E\varepsilon_{11} + \eta\dot{\varepsilon}_{11}$$



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$$\sigma_{11} = E\varepsilon_{11} + \dots + \eta\dot{\varepsilon}_{11}$$

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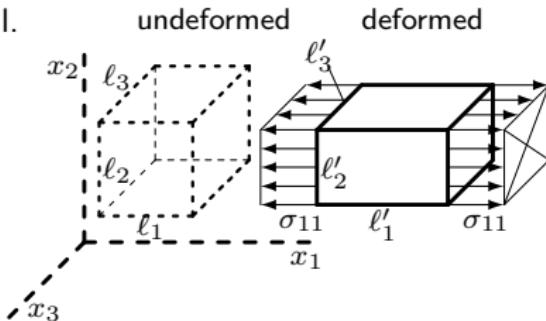
etc. ...

$E$  is the Young's modulus (modulus of elasticity) and  $\eta$  is the viscosity.

Also a combined stress state is possible. Superposition principle.

Resulting in; Hooke's law + viscous stress

( $\varepsilon \rightarrow \sigma$ ) :



$$\boxed{\sigma_{ij} = \frac{E}{(1+\nu)} \left( \varepsilon_{ij} + \frac{\nu}{(1-2\nu)} \varepsilon_{kk} \delta_{ij} \right) + \eta \dot{\varepsilon}_{ij}}$$

for  $i = 1, 2, 3 ; j = 1, 2, 3$ .

## 6.2) Elastic energy

Work:

$$\Pi_{el} = \int_0^e F \, du = \frac{1}{2} F_e u_e$$

At the end:

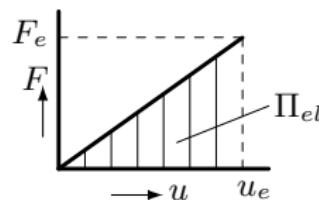
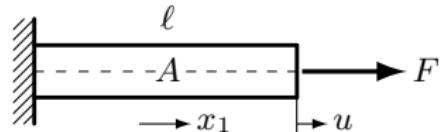
$$\sigma_{11} = \frac{F_e}{A} \quad \text{en} \quad \varepsilon_{11} = \frac{u_e}{\ell}$$

Such that:

$$\Pi_{el} = \frac{1}{2} A \sigma_{11} \ell \varepsilon_{11} = \frac{1}{2} \sigma_{11} \varepsilon_{11} V$$

*Specific internal elastic energy:*  $\pi_{el} = \frac{1}{2} \sigma_{11} \varepsilon_{11}$

Contribution to  $\pi_{el}$  during shearing:  $\frac{1}{2} \sigma_{12} \gamma_{12}$



An arbitrary three dimensional stress state:

$$\pi_{el} = \frac{1}{2}\sigma_{11}\varepsilon_{11} + \frac{1}{2}\sigma_{22}\varepsilon_{22} + \frac{1}{2}\sigma_{33}\varepsilon_{33} + \frac{1}{2}\sigma_{12}\gamma_{12} + \frac{1}{2}\sigma_{13}\gamma_{13} + \frac{1}{2}\sigma_{23}\gamma_{23}$$

Written differently (with  $\gamma_{ij} = 2\varepsilon_{ij} = 2\varepsilon_{ji}$ ):

$$\begin{aligned}\pi_{el} &= \frac{1}{2}\sigma_{11}\varepsilon_{11} + \frac{1}{2}\sigma_{12}\varepsilon_{12} + \frac{1}{2}\sigma_{13}\varepsilon_{13} + \\ &+ \frac{1}{2}\sigma_{21}\varepsilon_{21} + \frac{1}{2}\sigma_{22}\varepsilon_{22} + \frac{1}{2}\sigma_{23}\varepsilon_{23} + \\ &+ \frac{1}{2}\sigma_{31}\varepsilon_{31} + \frac{1}{2}\sigma_{32}\varepsilon_{32} + \frac{1}{2}\sigma_{33}\varepsilon_{33}\end{aligned}$$

So, for the *specific elastic energy* holds (summation convention!!):

$$\boxed{\pi_{el} = \frac{1}{2}\sigma_{ij}\varepsilon_{ij}}$$

Elastic energy of a body with volume  $V$ :

$$\Pi_{el} = \int_V \pi_{el} \, dV = \int_V \frac{1}{2}\sigma_{ij}\varepsilon_{ij} \, dV$$

## Visco-elastic energy:

Hooke's and (simplest) Newtonian viscous law in the equation of the specific elastic energy  $\pi_{el} = \frac{1}{2}\sigma_{ij}\varepsilon_{ij}$ , and  
 $\pi_{visc} = \int \sigma_{ij}^{visc} d\varepsilon_{ij} = \eta\dot{\varepsilon}_{ij} \int d\varepsilon_{ij}$ , for constant  $\eta$  and  $\varepsilon_{ij}$ .

Elastic and viscous energy of a body with volume  $V$ :

$$\Pi_{el} = \int_V \pi_{el} dV = \int_V \frac{1}{2}\sigma_{ij}\varepsilon_{ij} dV \quad \text{and} \quad \Pi_{visc} = \int_V \pi_{visc} dV$$

1D only: Elastic and viscous energy of a body with volume  $V$ :

$$\begin{aligned}\Pi &= \Pi_{el} + \Pi_{visc} = \int_V (\pi_{el} + \pi_{visc}) dV \\ &= \int_V \left( \frac{1}{2}E\varepsilon_{11}^2 + \eta\dot{\varepsilon}_{11}\varepsilon_{11} \right) dV\end{aligned}$$