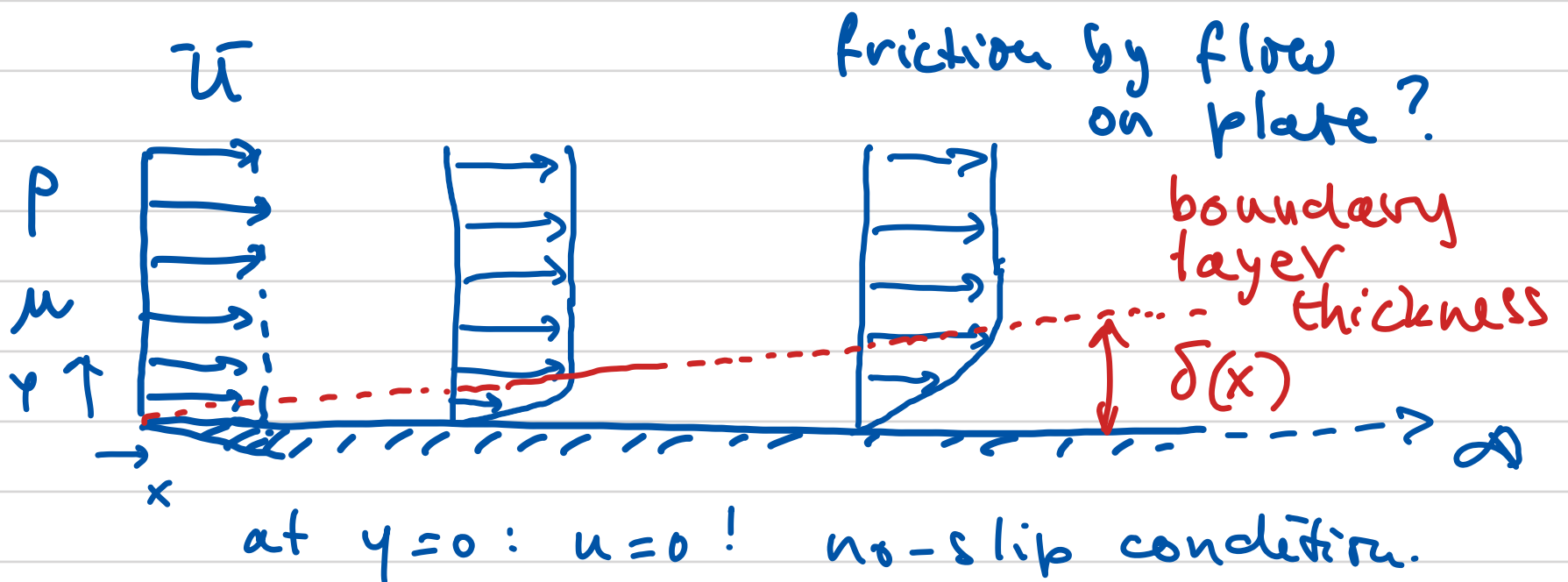


# Fluid Mechanics 1

## Lect #7 Flat Plate Boundary layer



Question:  $\delta(x)$ ?  $\rightarrow \tau(x)$ ?

Assumption: incompressible  $\rho = \text{const.}$   
 gravity neglected: " $g = 0$ "  
 steady  
 $\mu = \text{constant.}$

Mass conservation:  $\frac{\partial \rho}{\partial t} + \frac{\partial}{\partial x_j} (\rho u_j) = 0$  (with  $\rho = \text{const.}$ )

$$\rho \frac{\partial u_j}{\partial x_j} = 0 \quad \rho \neq 0 \Rightarrow \frac{\partial u_j}{\partial x_j} = 0$$

$$\Rightarrow \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} + \frac{\partial w}{\partial z} = 0 \quad \text{2D}$$

$$\Rightarrow \boxed{\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0}$$

Momentum conservation:

$$\cancel{\frac{\partial \rho u_i}{\partial t}} + \frac{\partial}{\partial x_j} (\rho u_i u_j - \tau_{ji}) = \cancel{g}^0$$

$$\Rightarrow \rho \frac{\partial}{\partial x_j} (u_i u_j) - \frac{\partial}{\partial x_j} \left\{ -p \delta_{ji} + \mu \left( \frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right) + 0 \right\} = 0$$

$$\frac{\partial}{\partial x_j} (u_i u_j) = \frac{\partial u_i}{\partial x_j} u_j + u_i \frac{\partial u_j}{\partial x_j}$$

$\rho = \text{const}$

$$\frac{\partial}{\partial x_j} (p \delta_{ij}) = \frac{\partial p}{\partial x_i}$$

$$\frac{\partial}{\partial x_j} \left( \mu \frac{\partial u_i}{\partial x_j} \right) = \mu \frac{\partial^2 u_i}{\partial x_j \partial x_j} \leftarrow \text{sum!}$$

$$= \mu \left( \frac{\partial^2 u_i}{\partial x^2} + \frac{\partial^2 u_i}{\partial y^2} \right)$$

$$\frac{\partial}{\partial x_j} \left( \mu \frac{\partial u_j}{\partial x_i} \right) = \mu \frac{\partial}{\partial x_j} \left( \frac{\partial u_j}{\partial x_i} \right) = \mu \frac{\partial}{\partial x_i} \left( \frac{\partial u_j}{\partial x_j} \right) = 0$$

mass:

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0$$

$$i=1: \quad u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} = -\frac{1}{\rho} \frac{\partial p}{\partial x} + \frac{\mu}{\rho} \left( \frac{\partial^2 u}{\partial x^2} + \frac{\partial^2 u}{\partial y^2} \right)$$

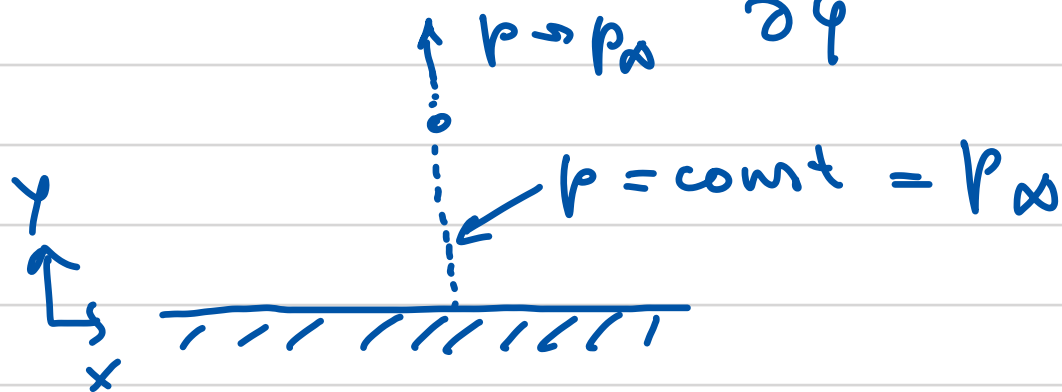
$$i=2: \quad 0 \approx -\frac{1}{\rho} \frac{\partial p}{\partial y} + 0$$



streamlines ~ straight.

3 eq's, 3 unknowns:  $u(x,y), v(x,y), p(x,y)$ .

Let's start with  $\frac{\partial p}{\partial y} = 0$



$p$  is uniform:

$$\boxed{p(x,y) = p_\infty}$$

$$\Rightarrow \frac{\partial p}{\partial x} = 0$$

Final assumption  
by Prandtl

$$\left| \frac{\partial^2 u}{\partial x^2} \right| \ll \left| \frac{\partial^2 u}{\partial y^2} \right|$$

Final step: make non-dimensional.

parameters:  $\rho, \mu, U$  dimensions: kg, m, s  
 $\Rightarrow$  zero non-dimensional number.

$$\tilde{x} \equiv \frac{\rho \bar{u} x}{\mu}$$

$$\tilde{y} \equiv \frac{\rho \bar{u} y}{\mu}$$

independent var.

$$\tilde{u} \equiv u / \bar{u}$$

$$\tilde{v} \equiv v / \bar{u}$$

dependent var.

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0$$

$$\frac{\partial u}{\partial x} \frac{\partial \tilde{x}}{\partial x} + \frac{\partial v}{\partial y} \frac{\partial \tilde{y}}{\partial y} = 0$$

$$\Rightarrow \frac{\partial \tilde{u} \bar{u}}{\partial \tilde{x}} \left( \frac{\partial \tilde{x}}{\partial x} \right) + \dots = 0$$

$$\Rightarrow \frac{\rho \bar{u}^2}{\mu} \left( \frac{\partial \tilde{u}}{\partial \tilde{x}} + \frac{\partial \tilde{v}}{\partial \tilde{y}} \right) = 0$$

$$\Rightarrow \left\{ \begin{array}{l} \frac{\partial \tilde{u}}{\partial \tilde{x}} + \frac{\partial \tilde{v}}{\partial \tilde{y}} = 0 \\ \tilde{u} \frac{\partial \tilde{u}}{\partial \tilde{x}} + \tilde{v} \frac{\partial \tilde{u}}{\partial \tilde{y}} = \frac{\partial^2 \tilde{u}}{\partial \tilde{y}^2} \\ \tilde{u}(\tilde{x}, 0) = 0 \quad \tilde{v}(\tilde{x}, 0) = 0 \quad \tilde{u}(\tilde{x}, \infty) = 1 \end{array} \right.$$

Boundary layer Equations

Prandtl 1904

Compass: 1903 Wright brothers  
1st motorized flight  
1905 Einstein:  
Special relativity.

1<sup>st</sup> step: define  $\tilde{\psi}(x, y)$  ("psi") such that:

$$\tilde{u} = \frac{\partial \tilde{\psi}}{\partial y} \quad - \tilde{v} = \frac{\partial \tilde{\psi}}{\partial x}$$

$$\Rightarrow \frac{\partial \tilde{u}}{\partial x} + \frac{\partial \tilde{v}}{\partial y} = \frac{\partial}{\partial x} \left( \frac{\partial \tilde{\psi}}{\partial y} \right) + \frac{\partial}{\partial y} \left( - \frac{\partial \tilde{\psi}}{\partial x} \right)$$

$$= \frac{\partial^2 \tilde{\psi}}{\partial x^2 \partial y} - \frac{\partial^2 \tilde{\psi}}{\partial y^2 \partial x} = 0 \quad \text{mass conservation automatically satisfied.}$$

instead of unknowns  $\tilde{u}$  and  $\tilde{v}$   
we only have unknown  $\tilde{\psi}$

Stream function.

momentum eq:

$$\tilde{\psi}_{xx} \equiv \frac{\partial^2 \tilde{\psi}}{\partial x^2}$$

$$\tilde{\psi}_{yy} \tilde{\psi}_{xx} - \tilde{\psi}_{xy} \tilde{\psi}_{xy} = \tilde{\psi}_{yyy}$$

Partial Diff. Eq

3<sup>rd</sup> order, non-linear

Blasius (PhD-student of Prandtl).

$$\text{try } \tilde{\psi}(x, y) = \sqrt{x} f\left(\frac{y}{\sqrt{x}}\right)$$

$$\zeta \equiv \eta$$

$\Rightarrow$  momentum equation becomes:

$$f f'' + 2 f''' = 0$$

Ordinary Diff. Eq

non-linear 3<sup>rd</sup> order

$$f(\eta) \quad \eta = \frac{y}{\sqrt{x}}$$

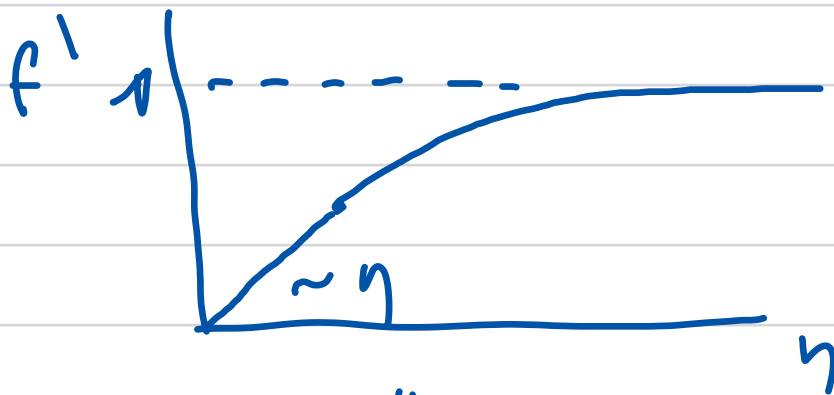
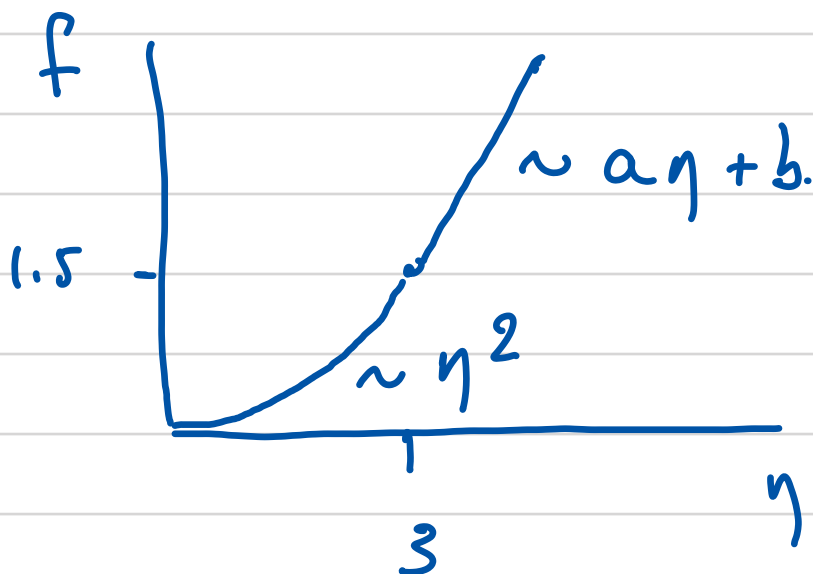
$$f'(\eta) \equiv \frac{df}{d\eta} \text{ etc.}$$

Blasius Equation

Solution: by computer.

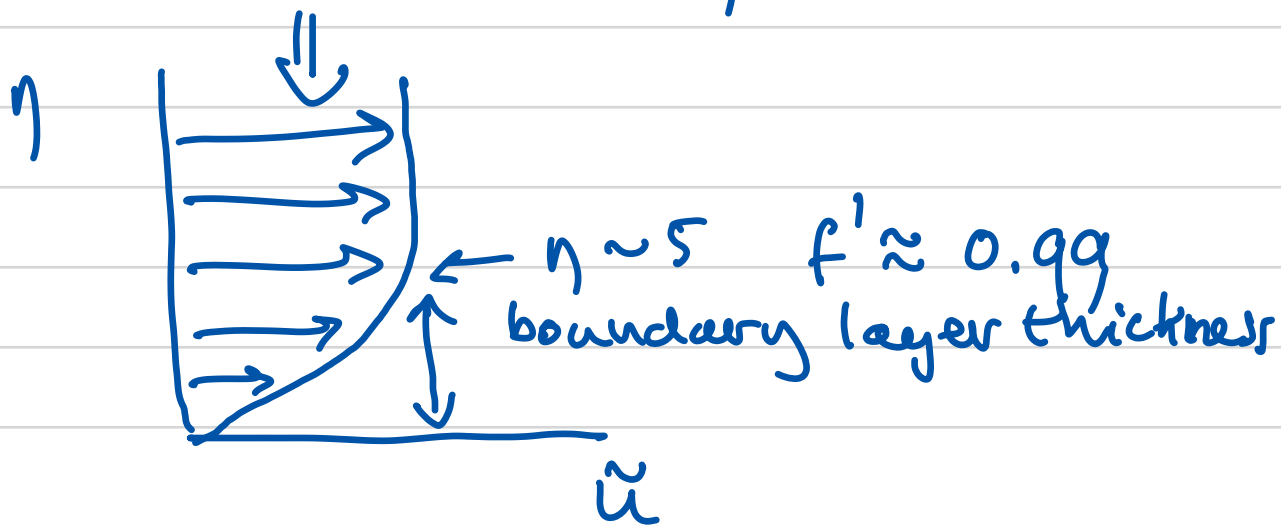
3 b.c.'s:  $f'(0) = 0$   $f(0) = 0$   $f'(\infty) = 1$

Solution:



note:

$$\underline{\underline{\tilde{u} = f'}}$$



$$\eta = 5 \Rightarrow \frac{\eta}{\sqrt{x}} = 5$$

$$\Rightarrow \eta = 5 \cdot \sqrt{x}$$

$$\Rightarrow \delta(x) = 5 \cdot \sqrt{x}$$

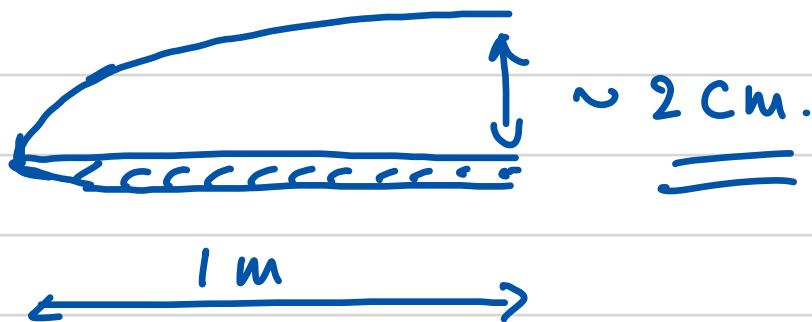
$$\Rightarrow \boxed{\delta(x) = 5 \cdot \sqrt{\frac{x \nu}{U}}}$$



$$\nu \equiv \mu / \rho \quad \text{kinematic viscosity}$$

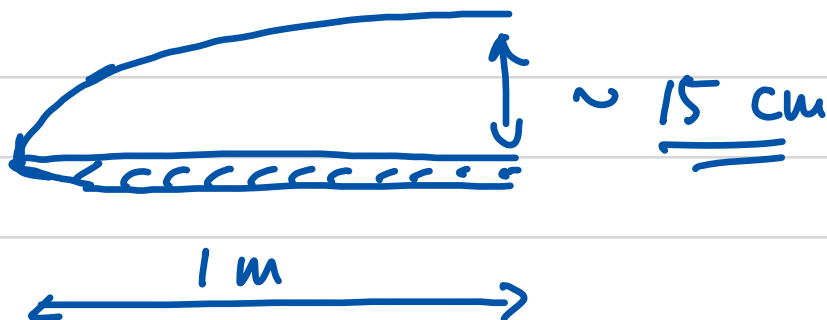
Example 1: air,  $U = 1 \text{ m/s}$

$$\nu = 18 \cdot 10^{-6} \text{ m}^2/\text{s}$$

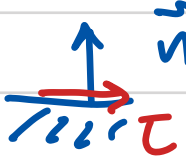


Example 2: water,  $U = 1 \text{ m/s}$

$$\nu = 9 \cdot 10^{-4} \text{ m}^2/\text{s}$$



Finally: shear stress on the wall.

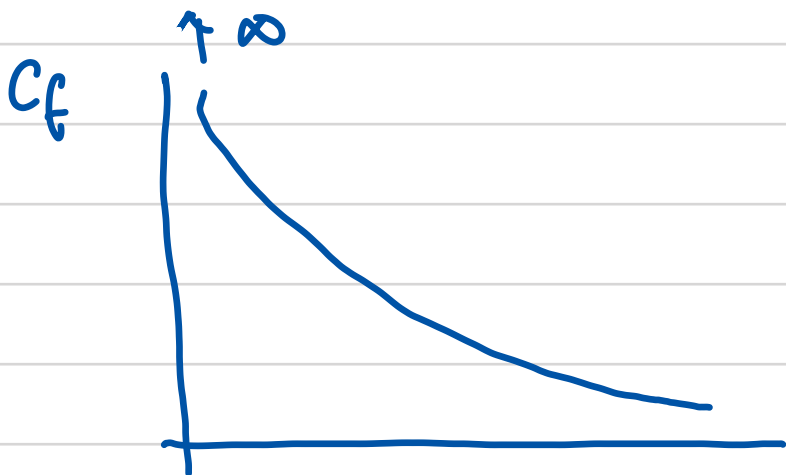
  $\vec{n} = \begin{pmatrix} 0 \\ 1 \end{pmatrix} \Rightarrow \tau = \mu \frac{\partial u}{\partial y}$   
by fluid  
on wall

Blasius solution gives  $f''(0) = \underline{0.332}$

$$\Rightarrow \tau_w = \frac{0.332}{\sqrt{\frac{U_\infty x}{\nu}}} \cdot \frac{1}{2} \rho U_\infty^2$$

Friction coefficient:

$$c_f \equiv \frac{\tau_w}{\frac{1}{2} \rho U_\infty^2} = \frac{0.332}{\sqrt{\frac{U_\infty x}{\nu}}} \sim \frac{1}{\sqrt{x}}$$



local infinite shear stress  $x$  but  $\int_0^L \frac{1}{\sqrt{x}} dx = 2\sqrt{x} \Big|_0^L = \text{finite.}$