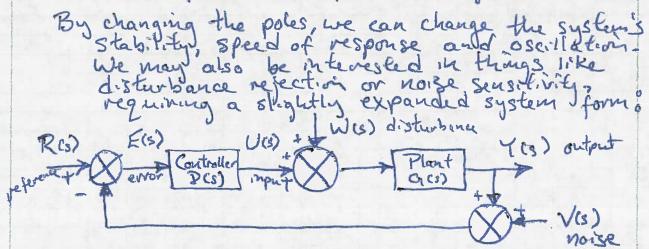
Closed-Loop System Kesponse From our study of haplace transforms, it is clear that the poles of a transfer function are a key characteristic of the system. The poles tell us whether or not the system is stable and provide the basic "building brocks" that combine to give the time response. to give the time response. This leads us to the key idea of £105... Closing a feed back loop around a system Changes the location of the poles! Let's say we start with a system G(s) = B(s) U(s) G(s) The poles satisfy A(s) = 0 (these are "open loop" poles) Now let's close a feedback loop around Gis) together with a controller, D(s)... Res) + (S) Pes) Pes) (Ges) (Ges) $\frac{Y(s)}{R(s)} = \frac{D(s)G(s)}{1 + D(s)G(s)}$ The poles now satisfy: 1+D(s)G(s)=0 -or- A(s)+D(s)B(s)=0

These are the "closed-loop" poles of our system with feed back (we have closed the feedback loop and changed the behavior)



We now have three signals coming on from onto de our system (the reference, a distrurbence and noise).

To determine the output resulting from these signals we just work our way through the diagram: Y(s) = G(s) [W(s) + U(s)] = G(s) [W(s) + D(s) E(s)] = G(s) [W(s) + D(s) (R(s) - Y(s) - V(s))] Y(s) + D(s) G(s) Y(s) = D(s) G(s) R(s) + G(s) W(s) - D(s) G(s) V(s) => Y(s) = DG R(s) + G W(s) - DG V(s) Similarly, for our error and input signals we have Ecs) = THOG R(S) - G W(S) + DG V(S) U(s) = P R(s) - DG W(s) - D V(s) We have none transfer functions we can use to determine the response of output, error or input to reference, disturbance or noise. All of these transfer functions share a characteristic... The denominator of each transfer function is 1+DG => They all have the same poles! Solutions to 1+DG) G(s) = 0 So poles are a characteristic of the system. Taking our definition of Stability to be having all of the poles in the left half plane, stability is the same for any of these responses. Let's work through concepts of closed-loop system vesponse using the example of car following: Se spacing W(s) road grade following) Controller (S) Car

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We can model the vehicle simply as: $m \overset{\circ}{x} = u - b\overset{\circ}{x}$ so $m s^2 \times (s) = U(s) - bs \times (s)$ $= \frac{\chi_f(s)}{U(s)} = \frac{1}{ms^2 + bs} = G(s)$

The open loop system poles are at 5=0 and 5=-bly.
Does this make sense? What does the pole of \$=0 imply?

Once we specify a controller, we can look at the system poles and characteristics such as stability.

Let's start with the simplest possible controller ...

Proportional Control - DC8) = K a constant

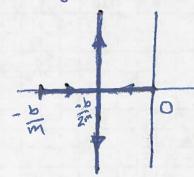
1+ Des) Ges = 1+ K = 0

=> ms2+bs+ K=0

open loop poles are of seo, se-bling

Closed loop poles are of $3 = \frac{b}{2m} + \frac{Jb^2 - 4mk}{2m}$

As Kincreases, the closed loop pole locations more away from the open loop locations.



At low values of K, the poles are real.

As Kincreases, the polise become complex but with a constant real part at - 1/2m.

(what does this mean for the system response?)

So we have some ability to locate the poles of the closed-loop system. What is a good location? One approach to answering this is to look at the step response of the Isystem.

Performance Specifications

Usually when putting performance specifications on the systems we look at the transfer function from the reference to the output:

The most common input used to specify performance is the unit step. It is much easier in practice to produce something resembling a step change in the reference than it is to produce an impulse in most cases.

For the car example, we have:

$$\frac{X_{f}}{R} = \frac{K}{1 + \frac{K}{ms^2 + bs}} = \frac{K}{ms^2 + bs + K}$$

Does a step change in r(t) make any sense? This would be a step change in the position of the lead car which is not physically possible. However,

$$\frac{X_f(s)}{R(s)} = \frac{SX_f}{SR} = \frac{X_f}{R} = \frac{R}{ms^2 + bs + K}$$

So we can think of the step response as also describing the change in speed of the following vehicle if I the lead vehicle speed suddenly changes.

This transfer function has a particular form that makes it easy to specify the step response. More complicated systems with a dominant set of closed-loop poles (a single pair close to the imaginary axis) will often resemble this response even if they don't fit it exactly.

We can consider the response in terms of a natural frequency, war, and a damping ratio, of.

$$\frac{X_{f}(S)}{R(S)} = \frac{K}{mS^{2} + bS + K} = \frac{\omega n^{2}}{S^{2} + 29\omega_{n} s \omega n^{2}}$$

$$\theta = \frac{5}{m} \frac{19}{9} + \frac{1}{\omega d} = \omega n \sqrt{1 - 9^{2}} \qquad \omega n = \frac{1}{m} \frac{1}{m}$$

$$\theta = \frac{1}{2} \sqrt{m} \frac{1}{m} \frac$$

Step response: Yes) = wn 1 => y(+)= 1-e ot [coswat + sinwat] This looks something like: See Section 3.3 for more examples with different damping ratios. We can specify a number of things: Rise time to 2 108 (This is based on observation) At the peak, g(t) = oe (coswat + and sinwat) - e - ot (-wasinwat + or cos wat) = e-ort (or sinwet + every wat) = 0 => sinuxt = 0 at the peak Peak time tp = to (This is exact) y(tp)= 1-e-on/wd (cost + 5 sy/1) = 1 + e -900 x/600 JI-92 = 1 + e - T9/51-92 Mp Peak overshoot Mp = e - 79/1-92 9=0.5, Mp=0.16 9=0.7, Mp=0.05 y(t) = 1-e (coswat + o sinwat)

decay oscillate When e-ot = 0.01, the response must be within 1% of the final value of 1 => orts = 4.6 Settling time (1%) ts = 4.6 (other settling times can

be defined similarly)

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