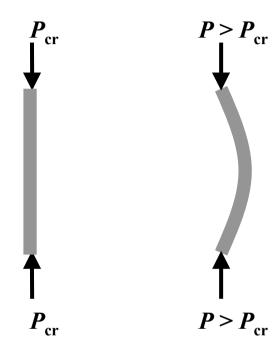
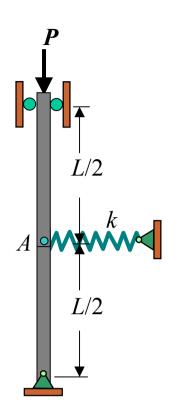
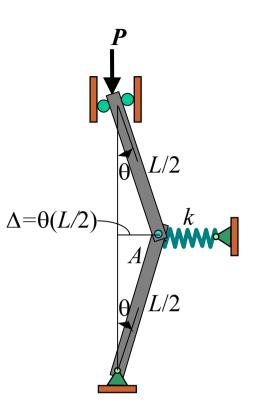
BUCKLING OF COLUMNS

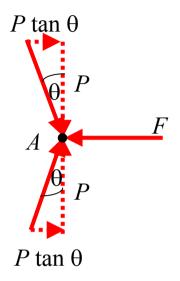
- Critical Load
- Ideal Column with Pin Supports
- Columns Having Various Supports

Critical Load









$$\stackrel{+}{\longrightarrow} \Sigma F_x = 0$$
:

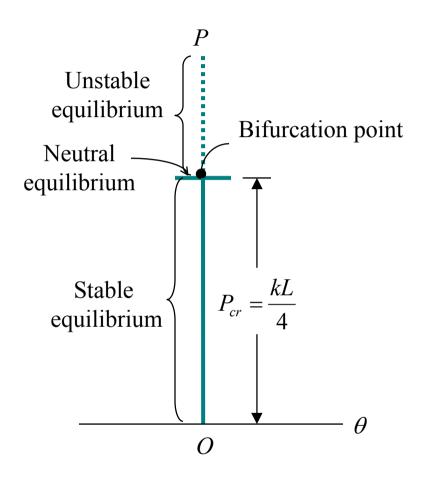
$$2P \tan \theta = F$$

$$2P \tan \theta = k\Delta$$

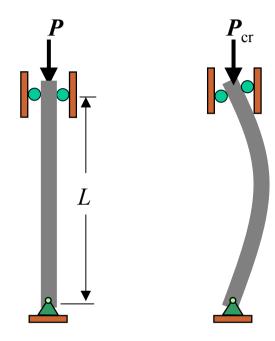
For small θ ,

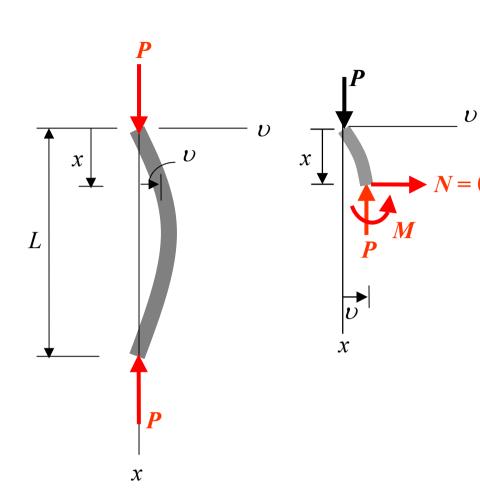
$$2P\theta = k(\theta \frac{L}{2})$$

$$P_{cr} = \frac{kL}{4}$$



Ideal Column with Pin Supports





$$\Sigma M_{x} = 0;$$

$$P \upsilon + M = 0$$

$$M = -P \upsilon$$

• Moment-curvature

$$M = EI \frac{d^2 v}{dx^2} = -Pv$$

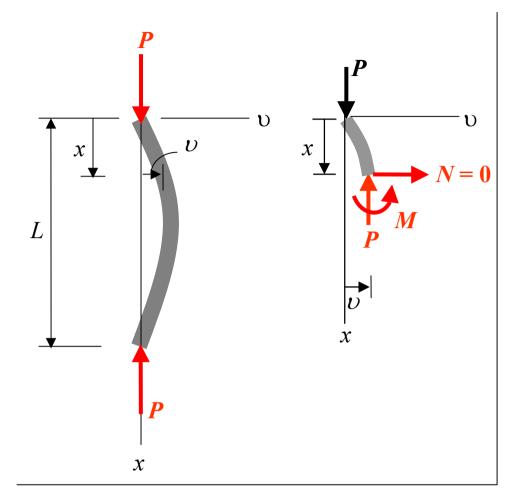
$$EI \frac{d^2 v}{dx^2} + Pv = 0$$

$$\frac{d^2 v}{dx^2} + (\frac{P}{EI})v = 0$$

$$\frac{d^2 v}{dx^2} + (\sqrt{\frac{P}{EI}})^2 v = 0 \quad ---*$$

$$v'' + c^2 v = 0$$

$$v = C_1 \sin(\sqrt{\frac{P}{EI}}x) + C_2 \cos(\sqrt{\frac{P}{EI}}x)$$



$$\upsilon = C_1 \sin(\sqrt{\frac{P}{EI}}x) + C_2 \cos(\sqrt{\frac{P}{EI}}x)$$

Boundary condition

$$\Rightarrow x = 0 \quad , \quad v = 0$$

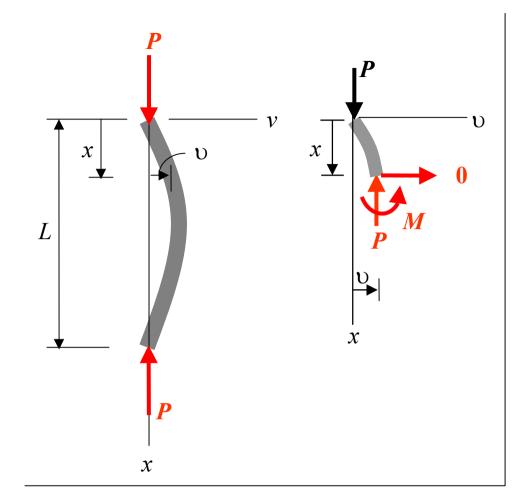
$$0 = C_1(0) + C_2(1), \quad C_2 = 0$$

$$\Rightarrow x = L \quad , \quad v = 0$$

$$C_1 \sin(\sqrt{\frac{P}{EI}}L) = 0 \quad , \quad C_1 \neq 0$$

$$\sin(\sqrt{\frac{P}{EI}}L) = 0 = \sin(n\pi)$$

$$\sqrt{\frac{P}{EI}}L = n\pi \quad ; \quad n = 1, 2, 3, ...$$



$$\sqrt{\frac{P}{EI}}L = n\pi \quad : \quad n = 1, 2, 3, \dots$$

• Critical Load P_{cr}

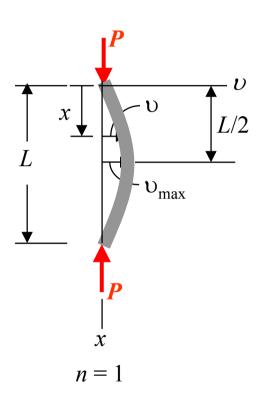
$$\sqrt{\frac{P}{EI}}L = n\pi$$

$$\frac{P}{EI}L^2 = n^2\pi^2$$

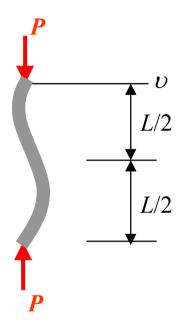
$$P = \frac{n^2 \pi^2 EI}{L^2} \quad ----*$$

$$P_{cr} = \frac{\pi^2 EI}{L^2} \quad ----*$$

$$P_{cr} = \frac{n^2 \pi^2 EI}{L^2}$$
 , $n = 1, 2, 3, ...$

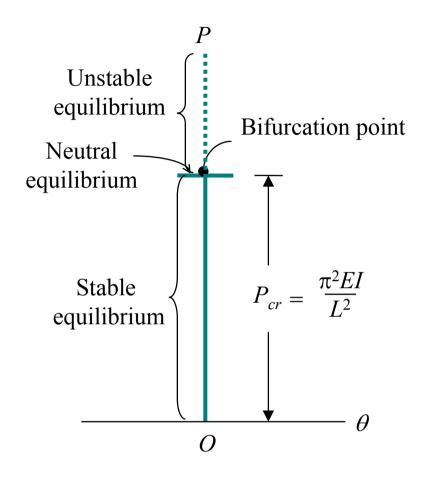


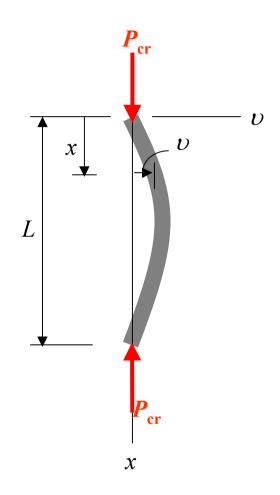
$$P_{cr} = \frac{1^2 \pi^2 EI}{L^2}$$



$$n = 2$$

$$P_{cr} = \frac{2^2 \pi^2 EI}{L^2}$$





Critical Stress

$$P_{cr} = \frac{\pi^2 EI}{(KL)^2}$$
$$= \frac{\pi^2 E(Ar^2)}{(KL)^2}$$
$$\frac{P_{cr}}{A} = \frac{\pi^2 E}{(K - L)^2}$$

$$\sigma_{cr} = \frac{\pi^2 E}{\left(\frac{KL}{r}\right)^2}$$

r = radius of gyration

K = effective-length factor, for pin-pin column K = 1 $K \frac{L}{r} = effective slenderness \ ratio$

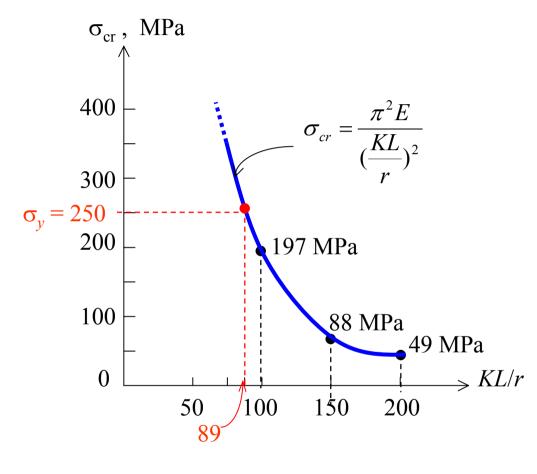
Structural steel A 36

$$E = 200 \text{ GPa}$$

$$\sigma_v = 250 \text{ MPa}$$

$$\sigma_{cr} = \frac{\pi^2 E}{\left(\frac{KL}{r}\right)^2} = \frac{\pi^2 (200 \times 10^3 MPa)}{\left(\frac{KL}{r}\right)^2}$$

KL/r	$\sigma_{\rm cr}({\rm MPa})$
89	250
100	197
125	126
150	88
175	64
200	49
225	39



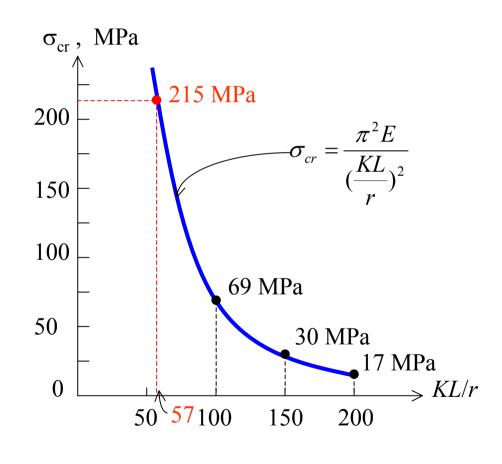
Structural steel
A 36

Aluminum

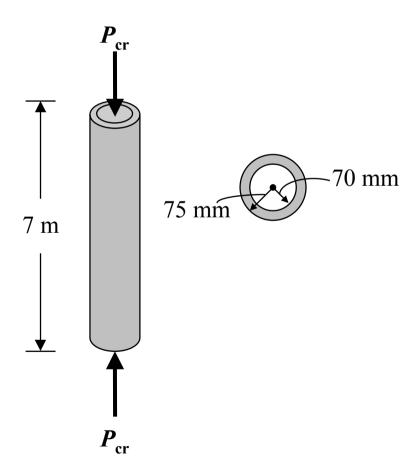
$$E = 70 \text{ GPa}$$
 $\sigma_y = 215 \text{ MPa}$

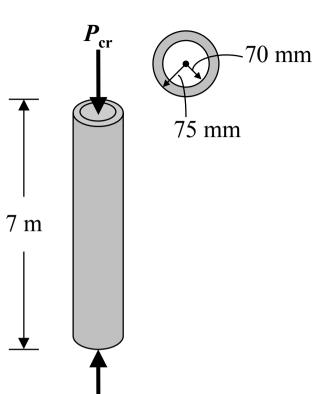
$$\sigma_{cr} = \frac{\pi^2 E}{\left(\frac{KL}{r}\right)^2} = \frac{\pi^2 (70 \times 10^3 MPa)}{\left(\frac{KL}{r}\right)^2}$$

KL/r	$\sigma_{\rm cr}({ m MPa})$
57	215
75	122.8
100	69.1
125	44.2
150	30.7
175	22.6
200	17.3



A 7 m long A-36 steel tube having the cross section shown is to be used as a pin-ended column. Determine the maximum allowable axial load the column can support so that it does not buckle or yield. Take the yield stress of 250 MPa





Using Eq. 5 to obtain the critical load with $E_{st} = 200$ GPa,

$$P_{cr} = \frac{\pi^2 EI}{L^2}$$

$$=\frac{\pi^2[(200\times10^6)(\frac{\pi}{4}0.075^4-\frac{\pi}{4}0.07^4)}{7^2}$$

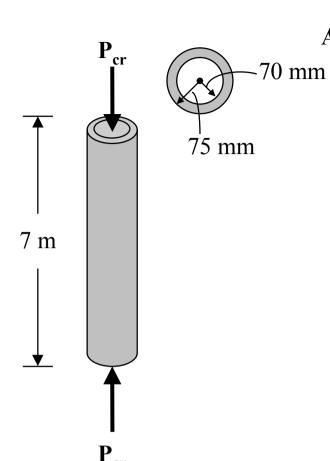
$$= 241.4 \text{ kN}$$

This force creates an average compressive stress in the column of

$$\sigma_{cr} = \frac{P_{cr}}{A} = \frac{241.43}{\pi (0.075)^2 - \pi (0.070)^2}$$
$$= 106 \text{ MPa} < \sigma_{Y} = 250 \text{ MPa} \quad \text{O.K}$$

The maximum allowable axial load the column can support is 241.73 kN

Alternate method:



Thermate method:

$$\sigma_{cr} = \frac{\pi^2 E}{(\frac{KL}{r})^2}$$

$$r^2 = \frac{I}{A} = \frac{\pi (.075^4 - .070^4)/4}{\pi (.075^2 - .070^2)} = .002631 \, m^2$$

$$r = 79.5 \, mm$$

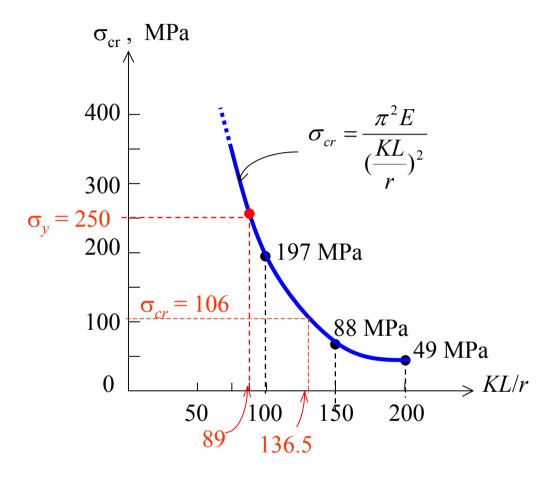
$$\frac{KL}{r} = \frac{(1)(7)}{(0.00795)} = 136.5$$

$$\sigma_{cr} = \frac{\pi^2 E}{(\frac{KL}{r})^2} = \frac{\pi^2 (200 \times 10^3 \, MPa)}{(136.5)^2}$$

$$= 106 \, \text{MPa} < \sigma_{\text{Y}} = 250 \, \text{MPa} \quad \text{O.K}$$

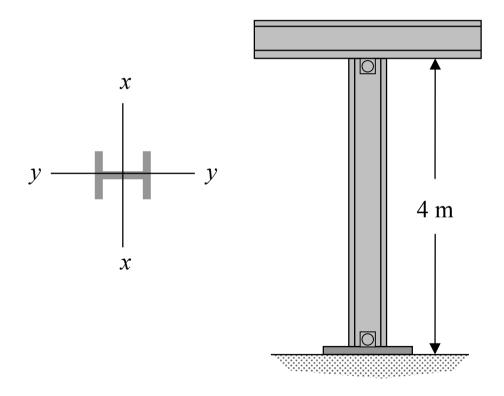
$$P_{\text{cr}} = \sigma_{\text{cr}} A = (106 \times 106) \, \pi \, (.075^2 - .070^2)$$

$$= 241.7 \, \text{kN}$$

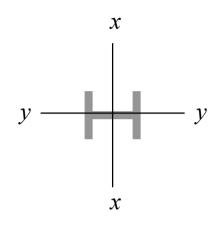


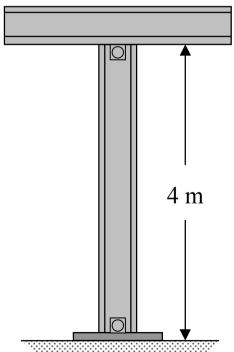
Structural steel
A 36
R 40 (4000 kg/cm²)

The A-36 steel *W*200x46 member show is to be used as a pin-connected column. Determine the largest axial load it can support before it either begins to buckle or the steel yields.



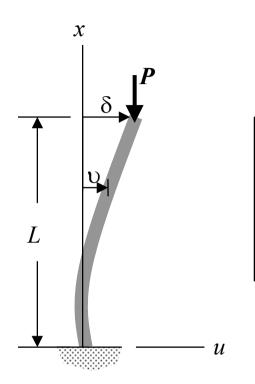
• Pinned - Pinned Column





A-36 steel W200x46 $A = 5890 \text{ mm}^2$, $I_x = 45.5 \text{x} 10^6 \text{ mm}^4$, and $I_y = 15.3 \text{x} 10^6 \text{ mm}^4$ $P_{cr} = \frac{\pi^2 EI}{L^2}$ $=\frac{\pi^2(200\times10^6)(15.3\times10^{-6})}{\Delta^2}$ = 1887.6 kN $\sigma_{cr} = \frac{P_{cr}}{A}$ $=\frac{1887.56}{5890\times10^{-6}}$ $= 320.5 \text{ MPa} > \sigma_{Y} = 250 \text{ MPa}$ $P_{allow} = \sigma \bullet A = \sigma_y \bullet A$ $= (250 \times 10^6 \ Pa)(5890 \times 10^{-6} \ m^2)$ $=1472 \ kN$

Columns Having Various Type of Supports



$$EI\frac{d^2v}{dx^2} = P(\delta - v)$$

$$\frac{d^2v}{dx^2} + \frac{P}{EI}v = \frac{P}{EI}\delta$$
 ----(7)

This equation is non-homogeneous because of the nonzero term on the right side. The solution consists of both a complementary and particular solution, namely,

$$\upsilon = C_1 \sin(\sqrt{\frac{P}{EI}}x) + C_2 \cos(\sqrt{\frac{P}{EI}}x) + \delta$$

The constants are determined from the boundary conditions. At x = 0, v = 0, so that $C_2 = -\delta$. Also,

$$\frac{dv}{dx} = C_1 \sqrt{\frac{P}{EI}} \cos(\sqrt{\frac{P}{EI}}x) - C_2 \sqrt{\frac{P}{EI}} \sin(\sqrt{\frac{P}{EI}}x)$$

At x = 0, dv/dx = 0, so that $C_1 = 0$. The deflection curve is therefore

$$\upsilon = \delta[1 - \cos(\sqrt{\frac{P}{EI}x})] \qquad ----(8)$$

Since the deflection at the top of the column is δ , that is, at x = L, $v = \delta$, we require

$$\delta = \delta[1 - \cos(\sqrt{\frac{P}{EI}x})]$$
 --> $\delta\cos(\sqrt{\frac{P}{EI}L}) = 0$

Since $\delta \neq 0$,

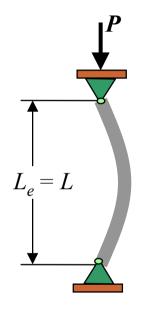
$$\cos(\sqrt{\frac{P}{EI}}L) = 0$$
 or $\cos(\sqrt{\frac{P}{EI}}L) = \cos(\frac{n\pi}{2})$

The smallest critical load occurs when n = 1, so that

$$P_{cr} = \frac{\pi^2 EI}{4L^2} \qquad ----(9)$$

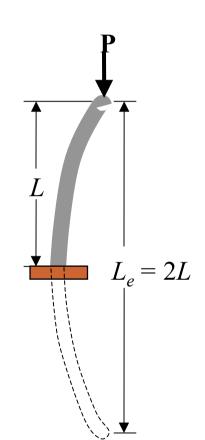
• Effective Length (L_e)

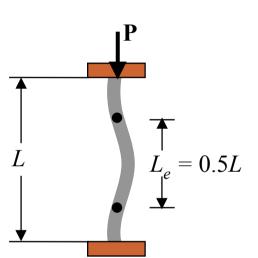
$$P_{cr} = \frac{\pi^2 EI}{(KL)^2} \quad or \quad \sigma_{cr} = \frac{\pi^2 E}{(KL/r)^2}$$



Pinned -pinned ends

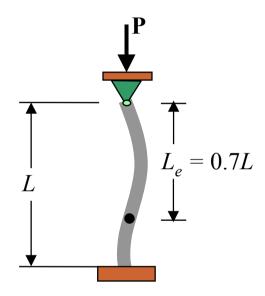
$$K = 1$$





Fixed fixed ends

$$K = 0.5$$



Fixed - Pinned ends

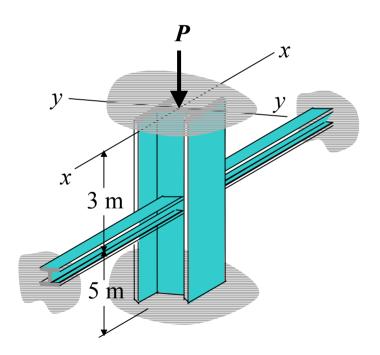
$$K = 0.7$$

Fixed - free ends

$$K=2$$

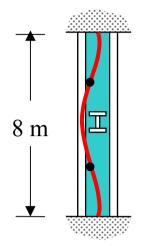
Note: K = effective-length factor

A W 150x24 (A=3060 mm², I_x = 13.4x106 mm⁴, I_y = 1.83x106 mm⁴) steel column is 8 m long and is fixed at its ends as shown. Its load-carrying capacity is increased by bracing it about the y-y (weak) axis using struts that are assumed to be pin-connected to its mid-height. Determine the load it can support so that the column does not buckle nor the material exceed the yield stress. Take E = 200 GPa and σ_v = 250 MPa



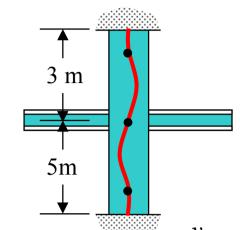
$$E = 200 \text{ GPa}$$
, $\sigma_{v} = 414 \text{ MPa}$

$$W 150x24 A = 3060 mm^2$$



$$I_x = 13.4 \times 10^6 \text{ mm}^4$$
Fixed (top)
Fixed (bottom)
$$K_x = 0.5$$

$$r_x = 66.2 \text{ mm}$$



 $I_y = 1.83 \times 10^6 \text{ mm}^4$ Pinned (top) Fixed (bottom)

$$K_{y} = 0.7$$

 $r_{y} = 24.5 \text{ mm}$

y-y axis buckling

 $\boldsymbol{\chi}$

3 m

x-x axis buckling

• Yield Stress (σ_v)

$$P_Y = \sigma_V A = (250 \times 10^6 Pa)(3060 \times 10^{-6} m^2) = 765 \text{ kN}$$

• Bucking x-x axis

$$(P_{cr})_x = \frac{\pi^2 E I_x}{(KL)_x^2} = \frac{\pi^2 (200 \times 10^6 \, kPa)(13.4 \times 10^{-6} \, m^4)}{(0.5 \times 8 \, m)^2} = 1653 \, kN$$

• Bucking y-y axis

$$(P_{cr})_y = \frac{\pi^2 E I_y}{(KL)_y^2} = \frac{\pi^2 (200 \times 10^6 \, kPa)(1.83 \times 10^{-6} \, m^4)}{(0.7 \times 5 \, m)^2} = 294.9 \, kN$$

NOTE

Structural steel A 36

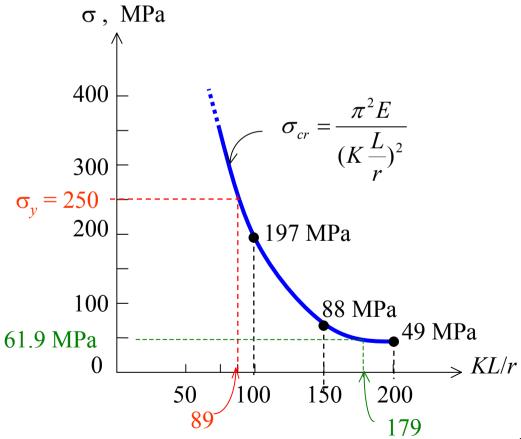
$$E = 200 \text{ GPa}$$

$$\sigma_{cr} = \frac{\pi^2 E}{(\frac{KL}{r})^2} = \frac{\pi^2 (200 \times 10^3 MPa)}{(\frac{KL}{r})^2}$$

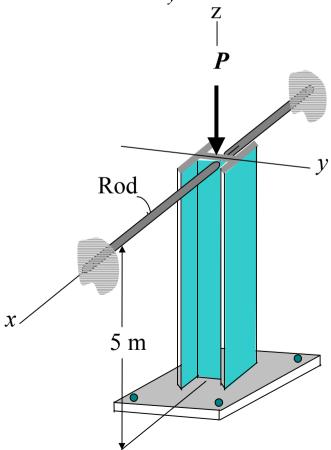
KL/r	$\sigma_{\rm cr}({\rm MPa})$
89	250
100	197
125	126
150	88
175	64
200	49
225	39

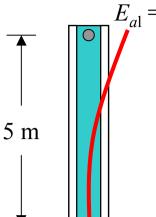
$$\left(\frac{KL}{r}\right)_x = \frac{(0.5)(8\times10^3)}{66.2} = 60.42$$

$$\left(\frac{KL}{r}\right)_y = \frac{(0.7)(5\times10^3)}{24.5} = 178.6 < ---$$
 buckling occurs



The aluminum column is fixed at its bottom and is braced at its top by two rods so as to prevent movement at the top along the x axis, If it is assumed to be fixed at its base, determine the largest allowable load P that can be applied. Use a factor of safety for buckling of F.S. = 3.0. Take $E_{al} = 70$ GPa, $\sigma_y = 215$ MPa, $A = 7.5(10^{-3})$ m², $I_x = 61.3(10^{-6})$ m⁴, $I_y = 23.2(10^{-6})$ m⁴.





 $E_{al} = 70 \text{ GPa}, \ \sigma_{y} = 215 \text{ MPa}, \ A = 7.5(10^{-3}) \text{ m}^{2}$

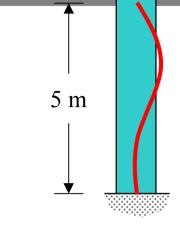
$$I_x = 61.3(10^{-6}) \text{ m}^4$$

Free (top)

Fixed (bottom)

$$K_x = 2$$

$$r_{x} = 90 \text{ mm}$$



 $I_y = 23.2(10^{-6}) \text{ m}^4$

Pinned (top)

Fixed (bottom)

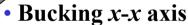
$$K_{\rm v} = 0.7$$

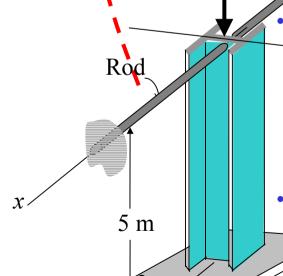
$$r_{v} = 50 \text{ mm}$$

x-x axis buckling

• Yield Stress (σ_v)

$$P_Y = \sigma_y A = (215 \times 10^6 Pa)(7.5 \times 10^{-3} m^2) = 1612 \text{ kN}$$





$$(P_{cr})_x = \frac{\pi^2 E I_x}{(KL)_x^2} = \frac{\pi^2 (70 \times 10^6 \, kPa)(61.3 \times 10^{-6} \, m^4)}{(2 \times 5 \, m)^2} = 425 \, kN$$

$$P_{allow} = \frac{P_{cr}}{F.S} = \frac{425 \, kN}{3} = 141 \, kN$$

• Bucking y-y axis

$$(P_{cr})_y = \frac{\pi^2 E I_y}{(KL)_y^2} = \frac{\pi^2 (70 \times 10^6 \, kPa)(23.2 \times 10^{-6} \, m^4)}{(0.7 \times 5 \, m)^2} = 1308 \, kN$$

$$P_{allow} = \frac{P_{cr}}{F.S} = \frac{1308 \, kN}{3} = 436 \, kN$$

NOTE

Aluminum

$$E = 70 \text{ Gpa}$$
 $\sigma_y = 215 \text{ MPa}$

$$E = 70 \text{ Gpa}$$
 $\sigma_{y} = 215 \text{ MPa}$

$$\sigma_{cr} = \frac{\pi^{2} E}{(\frac{KL}{r})^{2}} = \frac{\pi^{2} (70 \times 10^{3} MPa)}{(\frac{KL}{r})^{2}} \qquad \sigma_{cr}, N$$

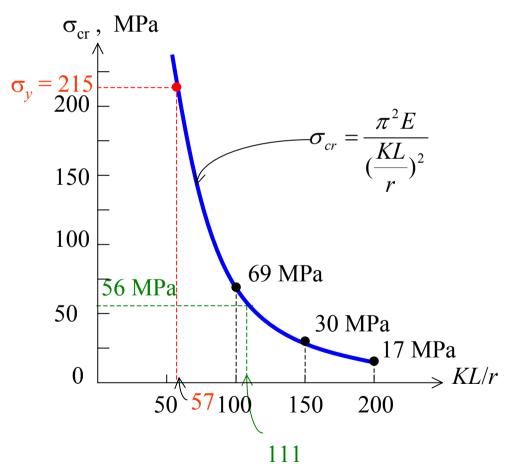
$$\sigma_{y} = 215$$

$$\sigma_{y} = 215$$

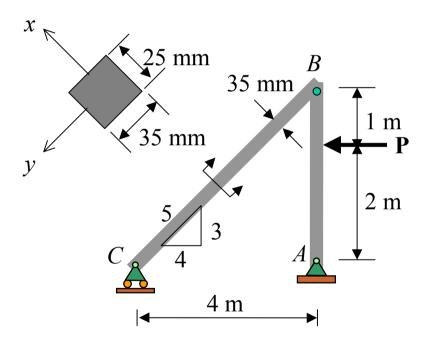
KL/r	$\sigma_{\rm cr}({\rm MPa})$
57	215
75	122.8
100	69.1
125	44.2
150	30.7
175	22.6
200	17.3

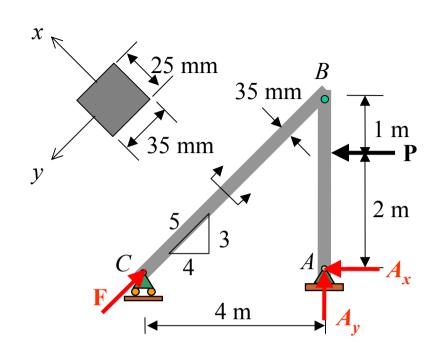
$$\left(\frac{KL}{r}\right)_y = \frac{0.7 \times 5 \times 10^3 \ mm}{50 \ mm} = 70$$

$$\left(\frac{KL}{r}\right)_x = \frac{2 \times 5 \times 10^3 \ mm}{90 \ mm} = 111.1 < --- occur buckling$$



Determine the maximum load P the column can support before it either begins to buckle or the steel yields. Assume that member BC is pinned at its end for the x-x axis and fixed for y-y axis buckling. Take E = 200 GPa, $\sigma_y = 250$ MPa.





$$F = \frac{5}{6}P - ---*$$

$$F_{Y} = \frac{5}{6}P_{Y} = \sigma_{Y}A = (250 \times 10^{3})(.025 \times .035) = 218.8 \text{ kN}$$

$$F_Y = \frac{5}{6}P_Y = \sigma_Y A = (250 \times 10^3)(.025 \times .035) = 218.8 \, kN$$

$$P_{Y} = 262.5 \, kN$$

$$\frac{A_x}{A_y} \qquad \frac{(KL)_x}{\left[\frac{1}{12} \frac{(0.025)(0.035)^3}{(0.025)(0.035)}\right]^{1/2}} = 495$$

•
$$\left(\frac{KL}{r}\right)_y = \frac{0.5(5)}{\left[\frac{1}{12} \frac{(0.035)(0.025)^3}{(0.035)(0.025)}\right]^{1/2}} = 346$$

•
$$F = \sigma_{cr} A = \frac{\pi^2 E}{\left(\frac{KL}{r}\right)^2} A$$

$$\frac{5}{6}P = \frac{\pi^2 (200 \times 10^9)}{(495)^2} (0.025)(0.035)$$

$$P = 8.46 \text{ kN} < 262.5 \text{ kN}$$