

PROB 3-20 FS | USING THE PREVIOUS RESULTS FOR 3-20 AND A FACTOR OF SAFETY OF 2. DETERMINE A MATERIAL IN TABLE A-22 THAT CAN BE SAFELY USED.

GIVEN:

1. PRINCIPAL STRESSES $\sigma_1 = 21.04 \text{ ksi}$, $\sigma_2 = 5.27 \text{ ksi}$, $\sigma_3 = -27.71 \text{ ksi}$
2. FACTOR OF SAFETY

ASSUMPTIONS:

1. LINEAR ELASTIC RESPONSE
2. MATERIAL BEYOND YIELD

FIND:

1. AN APPROPRIATE MATERIAL FROM A-22

SOLUTION:

$$\sigma_{vm} = \sqrt{\frac{1}{2}[(\sigma_1 - \sigma_2)^2 + (\sigma_1 - \sigma_3)^2 + (\sigma_2 - \sigma_3)^2]} = \sqrt{\frac{1}{2}[(21.04 \text{ ksi} - 5.27 \text{ ksi})^2 + (21.04 \text{ ksi} + 27.71 \text{ ksi})^2 + (5.27 \text{ ksi} + 27.71 \text{ ksi})^2]} \\ = 43.09 \text{ ksi}$$

$$\sigma_{vm} = \frac{S_y}{n} \Rightarrow S_{y,min} = n \cdot \sigma_{vm} = 2 \cdot 43.09 \text{ ksi} = 86.17 \text{ ksi}$$

THE ONLY TWO MATERIALS THAT CAN BE LOADED AS STATED ARE STEEL 1045 & 4142

Table A-22

Results of Tensile Tests of Some Metals* Source: J. Datsko, "Solid Materials," chap. 32 in Joseph E. Shigley, Charles R. Mischke, and Thomas H. Brown, Jr. (eds.-in-chief), *Standard Handbook of Machine Design*, 3rd ed., McGraw-Hill, New York, 2004, pp. 32.49-32.52.

Number	Material	Condition	Strength (Tensile)					Fracture Strain ϵ_f
			Yield S_y MPa (kpsi)	Ultimate S_u MPa (kpsi)	Fracture, σ_f MPa (kpsi)	Coefficient $\sigma_{0.2}$ MPa (kpsi)	Strain Strength, Exponent m	
1018	Steel	Annealed	220 (32.0)	341 (49.5)	628 (91.1) [†]	620 (90.0)	0.25	1.05
1144	Steel	Annealed	358 (52.0)	646 (93.7)	898 (130) [†]	992 (144)	0.14	0.49
1212	Steel	HR	193 (28.0)	424 (61.5)	729 (106) [†]	758 (110)	0.24	0.85
1045	Steel	Q&T 600°F	1520 (220)	1580 (230)	2380 (345)	1880 (273) [†]	0.041	0.81
4142	Steel	Q&T 600°F	1720 (250)	1930 (280)	2340 (340)	1760 (255) [†]	0.048	0.43
303	Stainless steel	Annealed	241 (35.0)	601 (87.3)	1520 (221) [†]	1410 (205)	0.51	1.16
304	Stainless steel	Annealed	276 (40.0)	568 (82.4)	1600 (233) [†]	1270 (185)	0.45	1.67
2011	Aluminum alloy	T6	169 (24.5)	324 (47.0)	325 (47.2) [†]	620 (90)	0.28	0.10
2024	Aluminum alloy	T4	296 (43.0)	446 (64.8)	533 (77.3) [†]	689 (100)	0.15	0.18
7075	Aluminum alloy	T6	542 (78.6)	593 (86.0)	706 (102) [†]	882 (128)	0.13	0.18

*Values from one or two heats and believed to be attainable using proper purchase specifications. The fracture strain may vary as much as 100 percent.

[†]Derived value.