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# The Euler-Lagrange Equation

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# Airlines



Figure 1: Mudd Air (adapted from [Air19])



# Airline



Figure 2: Flight 1 (UPS9859), Saturday 03/23/2019

# Airline, cont.



Figure 3: Flight 2 (FDX50252), Friday 03/22/2019

# The difference:



Figure 4: Wind patterns at 70hPa during Flight 1



## The difference:



Figure 5: Wind patterns at 70hPa during Flight 2



## Question:

How do airlines calculate the best trajectory?



# Key Features:

- ▶ Object of interest: a path  $\mathbf{q}(t) = (x(t), y(t), z(t))$ .
- ▶ Givens:
  - A region  $X \subset \mathbb{R}^3$  we can fly through (Pacific Ocean airspace)
  - Start/end points  $\mathbf{x}_0, \mathbf{x}_1$  (Honolulu/Anchorage)
  - Drag:  $\mathbf{F}_D(t, \mathbf{q}(t), \dot{\mathbf{q}}(t))$
- ▶ Goal: find  $\mathbf{q}(t)$  minimizing *total* travel time.



Big idea: find best path

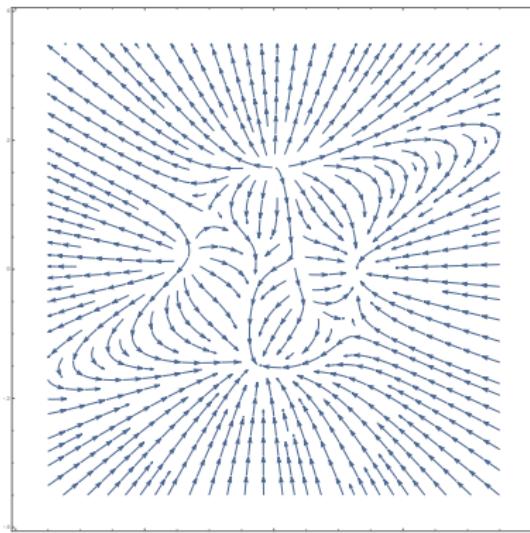


Figure 6: Drag field



# Defining “Cost”

- ▶ Again: drag is given by  $\mathbf{F}_D(t, \mathbf{q}(t), \dot{\mathbf{q}}(t))$
- ▶ Define “instantaneous cost” function:

$$\mathcal{L}(t, \mathbf{q}(t), \dot{\mathbf{q}}(t))$$

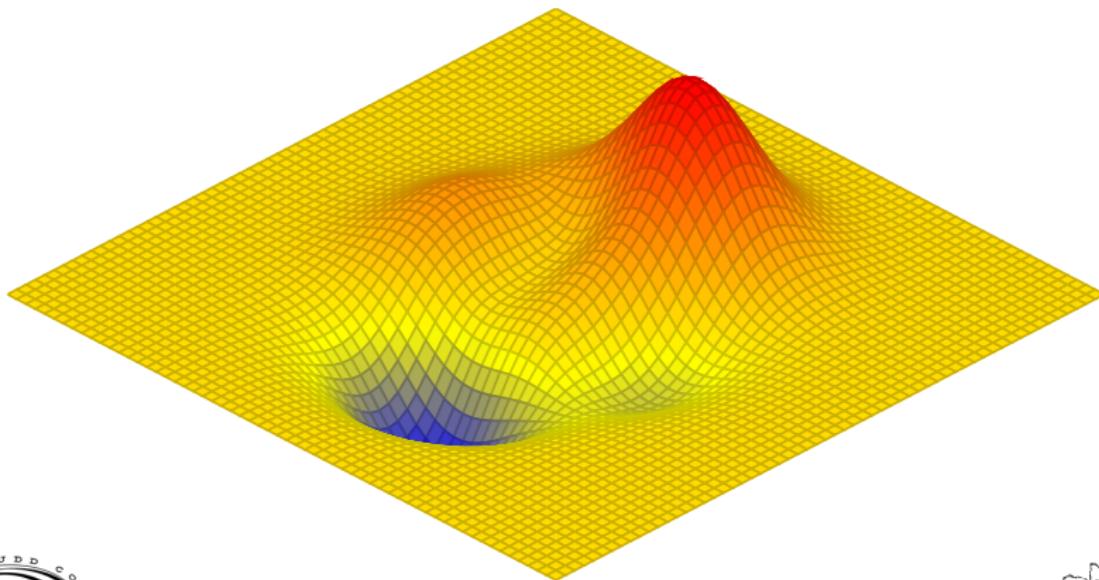
- ▶ Then define the total “cost” of trip:

$$C[\mathbf{q}] = \int_{t_0}^{t_1} \mathcal{L}(t, \mathbf{q}(t), \dot{\mathbf{q}}(t)) \, dt$$

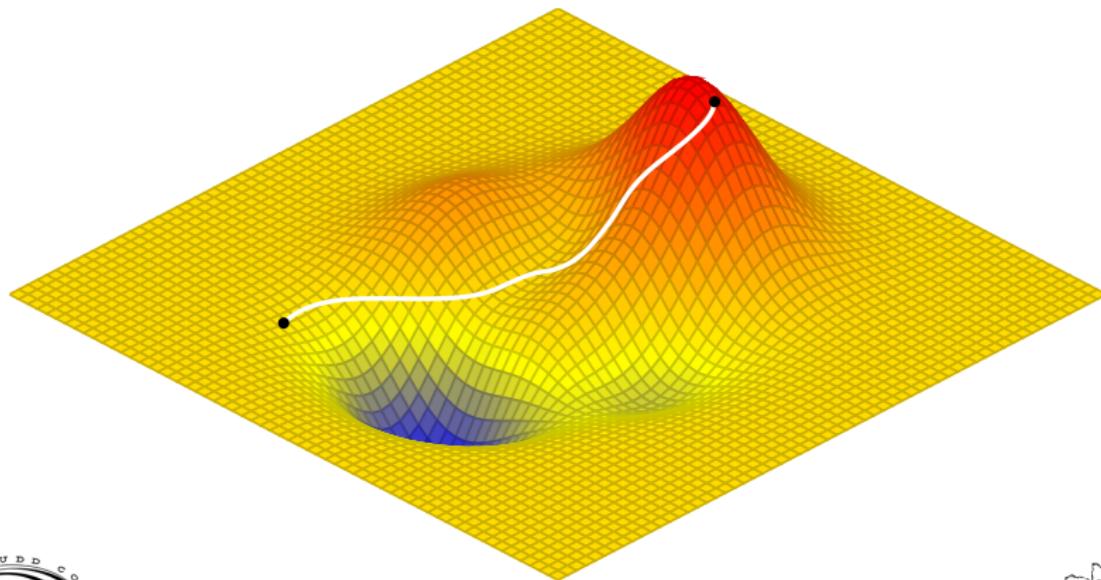
- ▶ What we want: an analogue of the first derivative test



# Big idea, cont:



# Big idea, cont:



# Solution: Euler-Lagrange

## Theorem (Euler-Lagrange)

Let  $X$  be our space of interest, and let  $\mathbf{x}_0, \mathbf{x}_1 \in X$ . Let  $\mathbf{q}(t)$  be a path from  $\mathbf{x}_0$  to  $\mathbf{x}_1$ . Then if  $\mathbf{q}(t)$  minimizes the “total cost function”

$$C[\mathbf{q}(t)] = \int_{t_0}^{t_1} \mathcal{L}(t, \mathbf{q}(t), \dot{\mathbf{q}}(t)) \, dt,$$

$\mathbf{q}(t)$  is a solution to the differential equation

$$\frac{\partial \mathcal{L}}{\partial \mathbf{q}} - \frac{d}{dt} \left( \frac{\partial \mathcal{L}}{\partial \dot{\mathbf{q}}} \right) = 0.$$



# Overview:

1. Suppose an optimal  $\mathbf{q}(t)$  exists.
2. Add some perturbation  $\alpha \cdot \boldsymbol{\eta}(t)$  to get a non-optimal  $\mathbf{q}'(t)$ .
3.  $C[\mathbf{q}'(t)]$  is minimized when  $\alpha = 0$ .
4. Hence, differentiating in  $\alpha$ , we can show

$$\frac{dC}{d\alpha} \Big|_{\alpha=0} = \int_{t_0}^{t_1} \left[ \frac{\partial \mathcal{L}}{\partial \mathbf{q}} - \frac{d}{dt} \left( \frac{\partial \mathcal{L}}{\partial \dot{\mathbf{q}}} \right) \right] \boldsymbol{\eta}(t) \, dt = 0$$

5. It follows that

$$\boxed{\frac{\partial \mathcal{L}}{\partial \mathbf{q}} - \frac{d}{dt} \left( \frac{\partial \mathcal{L}}{\partial \dot{\mathbf{q}}} \right) = 0}$$



Suppose an optimal  $\mathbf{q}(t)$  exists:

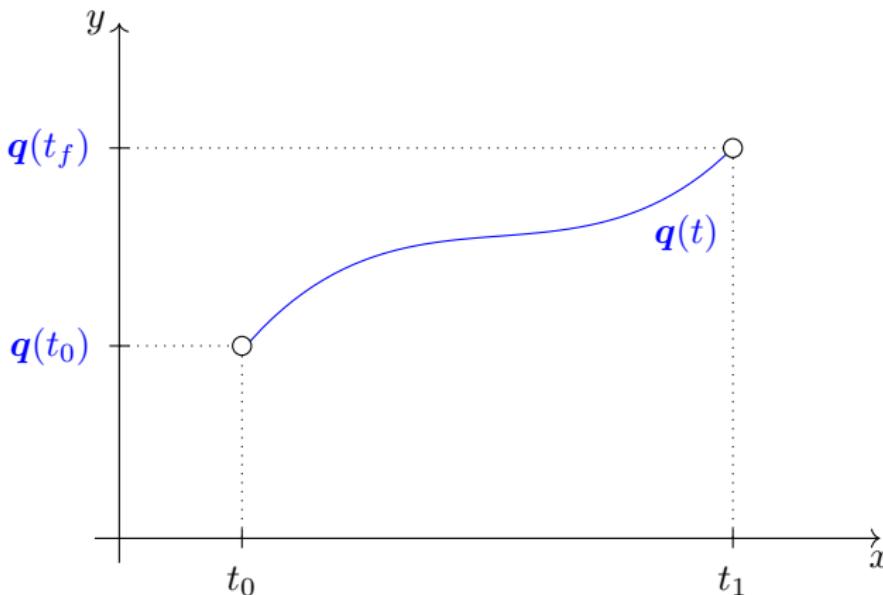


Figure 7: Optimal  $\mathbf{q}(t)$



For some  $\eta(t)$ , add  $\alpha \cdot \eta(t)$ :

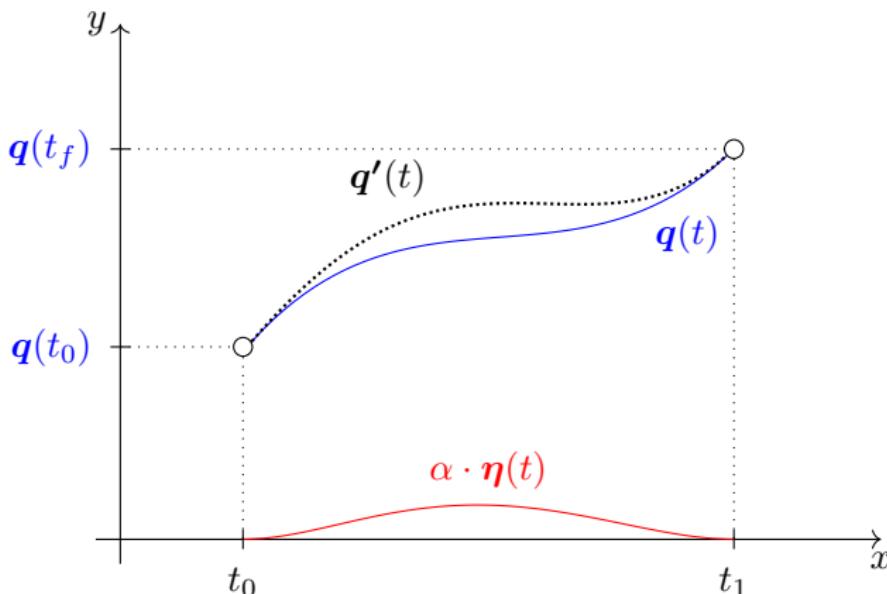


Figure 8: Add a small perturbation  $\alpha \cdot \eta(t)$



For smaller  $\alpha$ ,  $q'(t)$  closer to optimal:

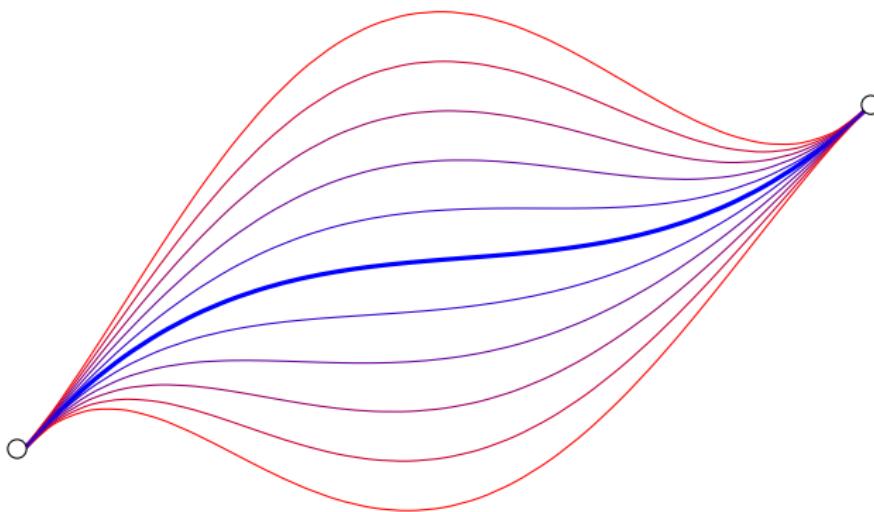


Figure 9:  $q'(t)$  for various values of  $\alpha$



Turn  $C$  into optimizing a 1-D function:

- ▶ Consider  $C[\mathbf{q}'(t)]$  as a function of just  $\alpha$ :

$$C'(\alpha) = C[\mathbf{q}(t) + \alpha \cdot \boldsymbol{\eta}(t)].$$

This is just a map from  $\mathbb{R} \rightarrow \mathbb{R}$ !

- ▶ Note that  $C'(0) = C[\mathbf{q}(t)]$ , so

$$0 = \left. \frac{dC'(\alpha)}{d\alpha} \right|_{\alpha=0}$$

- ▶ Repeated application of the chain rule and integration by parts yields the desired result!

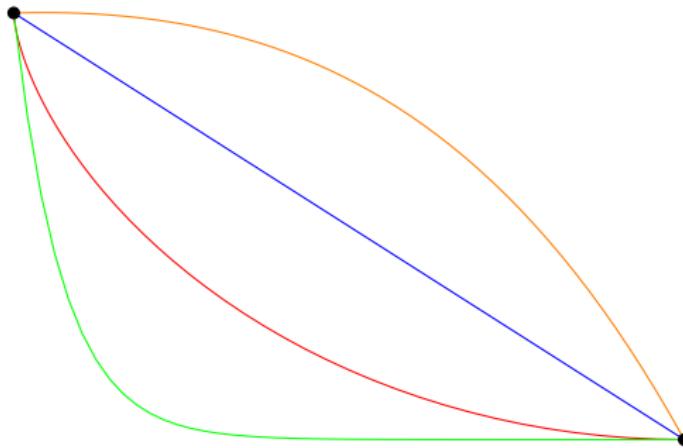


# Examples

- ▶ Optimal control
- ▶ Geodesics & Minimal surfaces of revolution
- ▶ Lagrangian Mechanics:  $\mathcal{L} = T - V$
- ▶ Brachistochrone problem



# Brachistochrone



# References

-  Airplane Movie Poster 24"x36", Mar 2019.  
[Online; accessed 31. Mar. 2019].

