

Katalyst Take Home Problem

Congratulations, you've made it to the next round of the interview process. For this round, we ask candidates to solve a product and technical focused problem. The deliverables are a solution to a technical problem prompt provided below. **Note: Feel free to ask any clarifying questions if you are stuck or unsure about a part of the problem**

Deliverable Format: Please provide all relevant work (Includes code, handwritten work, assumptions made, and relevant outputs)

Please turn in all parts of the problem within 5 business days from receiving the problem set. If selected for the next stage, you will be asked to walkthrough your presentation and technical problem.

Technical Problem Prompt:

You are developing navigation software for a satellite. The satellite needs to maintain accurate knowledge of its position and velocity over time.

Solve the following technical problem as part of this take home:

You receive measurements of the satellite via two ground sensors and a GPS receiver on the satellite. The sensors provide the following information:

1. Two Ground Electro-Optical Sensors: Each ground station provides angular position measurements in the form of Right Ascension (RA) and Declination (Dec) and angular rate measurements, \dot{RA} and \dot{Dec} in the form shown below:

$$z_{GS} = \begin{bmatrix} RA \\ Dec \\ \dot{RA} \\ \dot{Dec} \end{bmatrix}$$

The measurements are subject to noise, characterized by constant noise covariance:

$$R_{GS1} = \begin{bmatrix} 1 & & & \\ & 1 & & \\ & & 0.01 & \\ & & & 0.01 \end{bmatrix}, R_{GS2} = \begin{bmatrix} 0.01 & & & \\ & 0.01 & & \\ & & 0.0001 & \\ & & & 0.0001 \end{bmatrix}$$

in units of square arcseconds and square arcseconds per square seconds.

2. A GPS Receiver on the Satellite: Provides periodic state measurements (position and velocity in Cartesian coordinates) when available with constant noise covariance shown below.

$$z_{GPS} = \begin{bmatrix} x \\ y \\ z \\ \dot{x} \\ \dot{y} \\ \dot{z} \end{bmatrix}, R_{GPS} = \begin{bmatrix} 5000^2 & & & & & \\ & 5000^2 & & & & \\ & & 5000^2 & & & \\ & & & 0.5^2 & & \\ & & & & 0.5^2 & \\ & & & & & 0.5^2 \end{bmatrix},$$

in units of m and m/s.

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The two ground sensors are located at the following geodetic coordinates:

$$G1 = -111.536^\circ, +35.097^\circ, 2.206 \text{ km}$$

$$G2 = -70.692^\circ, -29.016^\circ, 2.380 \text{ km}$$

Provided in the attached measurements.csv file are ~3.3 hours of measurements from the two ground stations and the GPS. The initial time t_0 is '2024-11-24T05:04:30.000'.

Using the provided data, compute the following (assume Keplerian dynamics):

1. Predict the satellite's position $t_f + 2$ hours into the future.
2. Estimate the satellite's state at $t_f + 5$ hours.
3. For each step, briefly describe your approach and any key assumptions made.

As part of the solution, answer the following questions:

1. What are the steps to accomplish this problem?
2. Critical Thinking Exercise
 - a. What are the physical mechanics that can explain the behavior seen in the dataset?
 - b. What could the orbit explain about the mission of this spacecraft?
 - c. What would a spacecraft need to operate like this?
 - d. Create a high-level block diagram for the GNC system of this spacecraft.

Instructions for completion:

Make sure all relevant work is included with the technical problem. This should include assumptions made, intermediate steps, and code.