## Universidade Federal de Alagoas - UFAL Centro de Tecnologia - CTEC Curso de Engenharia Civil

Mecânica dos Sólidos 3 - ECIV051D (2020.2)

### **Exercícios:**

Deslocamentos em vigas isostáticas usando a equação diferencial da elástica

Encontro Assíncrono

(Versão atualizada 23/07/2021)

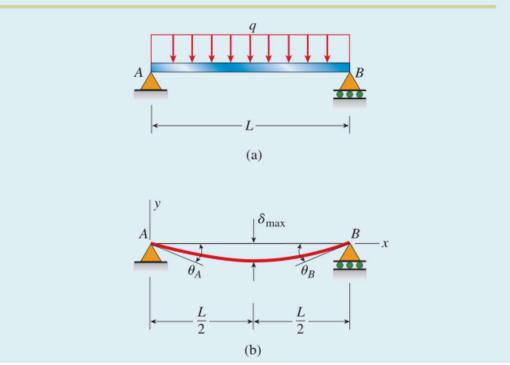
Monitores: Hugo Vinícius F. Azevedo Milton Mateus G. Santos Ricardo A. Fernandes

Professor/Supervisor: Adeildo S. Ramos Jr.

Referência: Gere & Goodno (2013). Mechanics of Materials, 8th Edition. Cengage Learning.

Determine the equation of the deflection curve for a simple beam AB supporting a uniform load of intensity q acting throughout the span of the beam (Fig. 9-8a).

Also, determine the maximum deflection  $\delta_{\rm max}$  at the midpoint of the beam and the angles of rotation  $\theta_{\rm A}$  and  $\theta_{\rm B}$  at the supports (Fig. 9-8b). (*Note:* The beam has length L and constant flexural rigidity EL.)



restart:

### Reações de apoio

$$\begin{aligned} & assign\Big(solve\Big(\left\{R_A+R_B-q\cdot L=0,R_B\cdot L-(q\cdot L)\cdot \frac{L}{2}=0\right\},\left\{R_A,R_B\right\}\Big)\Big): \\ & R_A,R_B=\frac{q\;L}{2}\;,\;\frac{q\;L}{2} \end{aligned}$$

#### **Momento fletor**

$$\begin{split} M &:= \mathit{unapply}\Big(\mathit{solve}\Big(M - R_A \cdot x + (q \cdot x) \cdot \frac{x}{2} = 0, M\Big), x\Big) : \\ M(x) &= \frac{1}{2} \ q \, L \, x - \frac{1}{2} \ q \, x^2 \end{split}$$

$$\begin{aligned} & \textit{diff}\left(v(x), x\$2\right) = \frac{\mathsf{d}^2}{\mathsf{d}x^2} \ v(x) \\ & v := \textit{unapply}\bigg(\textit{rhs}\bigg(\textit{dsolve}\bigg(\Big\{\textit{diff}\left(v(x), x\$2\right) = \frac{M(x)}{EI}, v(0) = 0, v(L) = 0\Big\}, v(x)\bigg)\bigg), x\bigg) : \end{aligned}$$

$$v(x) = \frac{q L x^3}{12 EI} - \frac{q x^4}{24 EI} - \frac{q L^3 x}{24 EI}$$

### Deflexão máxima

$$solve(D[1](v)(x_{\max}) = 0, x_{\max}) = \frac{L}{2}, \left(\frac{1}{2} + \frac{\sqrt{3}}{2}\right)L, \left(\frac{1}{2} - \frac{\sqrt{3}}{2}\right)L$$

$$evalf(\%) = 0.50000000000000 L, 1.366025404 L, -0.3660254040 L$$

$$\delta_{\max} := -v\left(\frac{L}{2}\right) = \frac{5}{384} \frac{qL^4}{EI}$$

## Ângulo de rotação nas extremidades

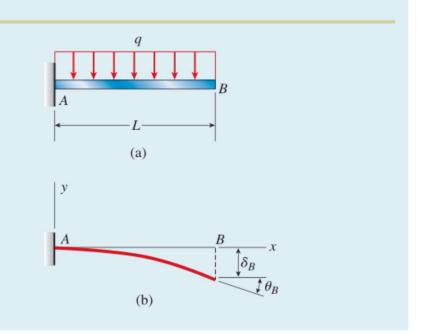
$$\theta_A := -D[1](v)(0) = \frac{1}{24} \frac{qL^3}{EI}$$

$$\theta_B := D[1](v)(L) = \frac{1}{24} \frac{qL^3}{EI}$$

Referência: Gere & Goodno (2013). Mechanics of Materials, 8th Edition. Cengage Learning.

Determine the equation of the deflection curve for a cantilever beam AB subjected to a uniform load of intensity q (Fig. 9-10a).

Also, determine the angle of rotation  $\theta_B$  and the deflection  $\delta_B$  at the free end (Fig. 9-10b). (*Note:* The beam has length L and constant flexural rigidity EI.)



restart:

### Reações de apoio

$$\begin{aligned} & assign \bigg( solve \bigg( \left\{ R_A - q \cdot L = 0, \, M_A - \left( q \cdot L \right) \cdot \frac{L}{2} = 0 \right\}, \, \left\{ R_A, \, M_A \right\} \bigg) \bigg) : \\ & R_A, \, M_A = q \, L, \, \frac{q \, L^2}{2} \end{aligned}$$

#### Momento fletor

$$\begin{split} M &:= \mathit{unapply}\Big(\mathit{solve}\Big(M + M_A - R_A \cdot x + (q \cdot x) \cdot \frac{x}{2} = 0, M\Big), x\Big) : \\ M(x) &= -\frac{1}{2} \ q \ L^2 + q \ L \ x - \frac{1}{2} \ q \ x^2 \end{split}$$

$$v := unapply \left( rhs \left( dsolve \left( \left\{ diff(v(x), x\$2) = \frac{M(x)}{EI}, v(0) = 0, D[1](v)(0) = 0 \right\}, v(x) \right) \right), x \right) :$$

$$v(x) = -\frac{q(L-x)^4}{24EI} - \frac{qL^3x}{6EI} + \frac{qL^4}{24EI}$$

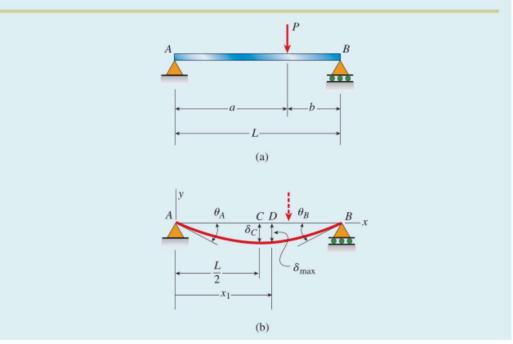
Ângulo de rotação e deflexão na extermidade livre 
$$\theta_B := -\mathrm{D}[1](v)(L) = \frac{1}{6} \ \frac{q \ L^3}{EI}$$
 
$$\delta_B := -v(L) = \frac{1}{8} \ \frac{q \ L^4}{EI}$$

$$\delta_B := -v(L) = \frac{1}{8} \frac{q L^4}{EI}$$

Referência: Gere & Goodno (2013). Mechanics of Materials, 8th Edition. Cengage Learning.

A simple beam AB supports a concentrated load P acting at distances a and b from the left-hand and right-hand supports, respectively (Fig. 9-12a).

Determine the equations of the deflection curve, the angles of rotation  $\theta_A$  and  $\theta_B$  at the supports, the maximum deflection  $\delta_{\max}$ , and the deflection  $\delta_C$  at the midpoint C of the beam (Fig. 9-12b). (Note: The beam has length L and constant flexural rigidity EI.)



restart : L := a + b :

$$assign(solve(\{R_A + R_B - P = 0, -P \cdot a + R_B \cdot L = 0\}, \{R_A, R_B\})):$$

$$R_A, R_B = \frac{Pb}{a+b}, \frac{Pa}{a+b}$$

#### Momento fletor

$$\begin{split} &M_{AP} := unapply \big( solve \big( M - R_A \cdot x = 0, M \big), x \big) : \\ &M_{PB} := unapply \big( solve \big( -M + R_B \cdot (L - x) = 0, M \big), x \big) : \\ &M_{AP}(x), M_{PB}(x) = \frac{P \ b \ x}{a + b}, \frac{P \ a \ (a + b - x)}{a + b} \end{split}$$

$$\begin{aligned} & bc \coloneqq v_{AP}(0) = 0, v_{PB}(L) = 0, v_{AP}(a) = v_{PB}(a), \text{D[1]}(v_{AP})(a) = \text{D[1]}(v_{PB})(a) : \\ & res \coloneqq dsolve \bigg( \bigg\{ diff \big(v_{AP}(x), x\$2 \big) = \frac{M_{AP}(x)}{EI}, diff \big(v_{PB}(x), x\$2 \big) = \frac{M_{PB}(x)}{EI}, bc \bigg\}, \big\{ v_{AP}(x), v_{PB}(x) \big\} \bigg) : \\ & v_{AP} \coloneqq unapply(simplify(rhs(res[1])), x) : \\ & v_{PB} \coloneqq unapply(simplify(rhs(res[2])), x) : \end{aligned}$$

$$\begin{split} v_{AP}(x) &= -\frac{P\,b\,x\,\left(a^2 + 2\,a\,b - x^2\right)}{6\,\left(a + b\right)\,EI} \\ v_{PB}(x) &= \frac{\left(a^2 - 2\,a\,x - 2\,x\,b + x^2\right)\,\left(a + b - x\right)\,P\,a}{6\,\left(a + b\right)\,EI} \\ simplify\!\left(v_{AP}(x) - \left(-\frac{P\cdot b\cdot x}{6\cdot L\cdot EI}\cdot\left(L^2 - b^2 - x^2\right)\right)\right) &= 0 \\ simplify\!\left(v_{PB}(x) - \left(-\frac{P\cdot b\cdot x}{6\cdot L\cdot EI}\cdot\left(L^2 - b^2 - x^2\right) - \frac{P\cdot \left(x - a\right)^3}{6\cdot EI}\right)\right) &= 0 \end{split}$$

### Ângulos de rotação nos suportes

$$\begin{split} & \theta_{A} \coloneqq -\mathrm{D}[1] \left(v_{AP}\right)(0) = \frac{1}{6} \ \frac{P \, b \, \left(a^{2} + 2 \, a \, b\right)}{\left(a + b\right) \, EI} \\ & \theta_{B} \coloneqq simplify \left(\mathrm{D}[1] \left(v_{PB}\right)(L)\right) = \frac{1}{6} \ \frac{b \, \left(2 \, a + b\right) \, P \, a}{\left(a + b\right) \, EI} \\ & simplify \left(\theta_{A} - \left(\frac{P \cdot a \cdot b \cdot (L + b)}{6 \cdot L \cdot EI}\right)\right) = 0 \\ & simplify \left(\theta_{B} - \left(\frac{P \cdot a \cdot b \cdot (L + a)}{6 \cdot L \cdot EI}\right)\right) = 0 \end{split}$$

#### Deflexão máxima

Considerando  $a \ge b$ 

$$solve(D[1](v_{AP})(x_{\max}) = 0, x_{\max}) = \frac{\sqrt{3 a^2 + 6 a b}}{3}, -\frac{\sqrt{3 a^2 + 6 a b}}{3}$$

$$x_{\max} := \%[1] = \frac{1}{3} \sqrt{3 a^2 + 6 a b}$$

$$\delta_{\max} := simplify(-v_{AP}(x_{\max})) = \frac{1}{27} \frac{P b \sqrt{3} \sqrt{a (a + 2 b)} a (a + 2 b)}{(a + b) EI}$$

$$simplify\left(\delta_{\max} - \left(\frac{P \cdot b \cdot (L^2 - b^2)}{9 \cdot \sqrt{3} \cdot L \cdot EI}\right)\right) = 0$$

#### Deflexão do ponto médio da viga

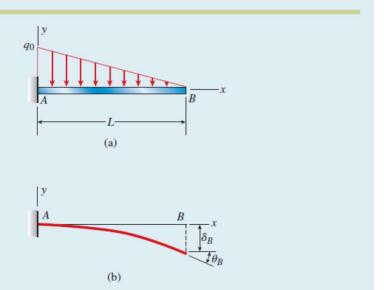
Considerando  $a \ge b$ 

$$\begin{split} \delta_{C} &:= simplify \bigg( - v_{AP} \bigg( \frac{L}{2} \bigg) \bigg) = \frac{1}{48} \frac{ \left( 3 \ a^2 + 6 \ a \ b - b^2 \right) \ P \ b}{EI} \\ simplify \bigg( \delta_{C} - \bigg( \frac{P \cdot b \cdot \left( 3 \cdot L^2 - 4 \cdot b^2 \right)}{48 \cdot EI} \bigg) \bigg) = 0 \end{split}$$

Referência: Gere & Goodno (2013). Mechanics of Materials, 8th Edition. Cengage Learning.

Determine the equation of the deflection curve for a cantilever beam AB supporting a triangularly distributed load of maximum intensity  $q_0$  (Fig. 9-14a).

Also, determine the deflection  $\delta_B$  and angle of rotation  $\theta_B$  at the free end (Fig. 9-14b). Use the fourth-order differential equation of the deflection curve (the load equation). (Note: The beam has length L and constant flexural rigidity EL.)



restart:

$$q := \mathit{unapply}\Big(q_0 - q_0 \cdot \frac{x}{L}, x\Big) :$$

### Reaçoes de apoio

$$\begin{split} & assign\bigg(solve\bigg(\left\{R_{A}^{}-\frac{q_{0}\cdot L}{2}=0,\,M_{A}^{}-\left(\frac{q_{0}\cdot L}{2}\right)\cdot\frac{L}{3}=0\right\},\,\left\{R_{A}^{},\,M_{A}^{}\right\}\bigg)\bigg):\\ & R_{A}^{},\,M_{A}^{}=\frac{q_{0}^{}\,L}{2}\,,\,\frac{q_{0}^{}\,L^{2}}{6} \end{split}$$

#### Momento fletor

$$\begin{split} M &:= \mathit{unapply}\Big(\mathit{solve}\Big(M + M_A - R_A \cdot x + (q(x) \cdot x) \cdot \frac{x}{2} + \Big(\big(q_0 - q(x)\big) \cdot \frac{x}{2}\,\Big) \cdot \frac{2 \cdot x}{3} = 0, M\Big), x\Big) : \\ M(x) &= -\frac{q_0\left(L^3 - 3\,L^2\,x + 3\,x^2\,L - x^3\right)}{6\,L} \end{split}$$

$$v := unapply \left( rhs \left( dsolve \left( \left\{ diff(v(x), x\$2) = \frac{M(x)}{EI}, v(0) = 0, D[1](v)(0) = 0 \right\}, v(x) \right) \right), x \right) :$$

$$v(x) = -\frac{q_0 (L - x)^5}{120 L EI} - \frac{q_0 L^3 x}{24 EI} + \frac{q_0 L^4}{120 EI}$$

$$simplify \left( v(x) - \left( -\frac{q_0 \cdot x^2}{120 \cdot L \cdot EI} \cdot \left( 10 \cdot L^3 - 10 \cdot L^2 \cdot x + 5 \cdot L \cdot x^2 - x^3 \right) \right) \right) = \mathbf{0}$$

Deflexão e ângulo de rotação na extermidade livre

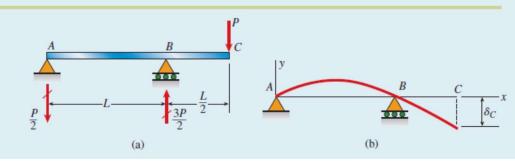
$$\delta_B := -v(L) = \frac{1}{30} \frac{q_0 L^4}{EI}$$

$$\theta_B := -D[1](v)(L) = \frac{1}{24} \frac{q_0 L^3}{EI}$$

Referência: Gere & Goodno (2013). Mechanics of Materials, 8th Edition. Cengage Learning.

A simple beam AB with an overhang BC supports a concentrated load P at the end of the overhang (Fig. 9-15a). The main span of the beam has length L and the overhang has length L/2.

Determine the equations of the deflection curve and the deflection  $\delta_{c}$  at the end of the overhang (Fig. 9-15b). Use the third-order differential equation of the deflection curve (the shear force equation). (Note: The beam has constant flexural rigidity EI.)



restart:

### Reações de apoio

$$assign \left( solve \left( \left\{ R_A + R_B - P = 0, R_B \cdot L - P \cdot \frac{3 \cdot L}{2} = 0 \right\}, \left\{ R_A, R_B \right\} \right) \right) :$$

$$R_A, R_B = -\frac{P}{2}, \frac{3 P}{2}$$

#### Momento fletor

$$\begin{split} &M_{AB} \coloneqq unapply \big( solve \big( M - R_A \cdot x = 0, M \big), x \big) : \\ &M_{BC} \coloneqq unapply \Big( solve \Big( -M - P \cdot \Big( \frac{3 \cdot L}{2} - x \Big) = 0, M \Big), x \Big) : \\ &M_{AB}(x) = -\frac{P \cdot x}{2} \\ &M_{BC}(x) = -\frac{P \cdot (3 \cdot L - 2 \cdot x)}{2} \\ &simplify \big( M_{AB}(L) - M_{BC}(L) \big) = 0 \end{split}$$

### Equação da curva de deflexão

$$bc := v_{AB}(0) = 0, v_{AB}(L) = 0, v_{BC}(L) = 0, D[1](v_{AB})(L) = D[1](v_{BC})(L) :$$

$$res := dsolve\left\{diff\left(v_{AB}(x), x\$2\right) = \frac{M_{AB}(x)}{EI}, diff\left(v_{BC}(x), x\$2\right) = \frac{M_{BC}(x)}{EI}, bc\right\}, \left\{v_{AB}(x), v_{BC}(x)\right\}\right\}$$

$$res := \left\{v_{AB}(x) = -\frac{Px^3}{12EI} + \frac{PL^2x}{12EI}, v_{BC}(x) = \frac{-3L^3P + 10L^2Px - 9PLx^2 + 2Px^3}{12EI}\right\}$$

$$(1)$$

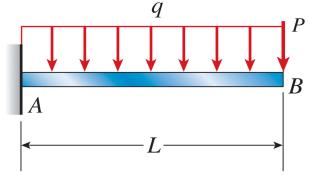
 $v_{AB} := unapply(simplify(rhs(res[1])), x) :$  $v_{BC} := unapply(simplify(rhs(res[2])), x) :$ 

$$\begin{split} v_{AB}(x) &= \frac{P \, x \, \left(L^2 - x^2\right)}{12 \, EI} \\ v_{BC}(x) &= -\frac{\left(3 \, L - x\right) \, P \, \left(L - 2 \, x\right) \, \left(L - x\right)}{12 \, EI} \\ simplify \left(v_{BC}(x) - \left(-\frac{P}{12 \cdot EI} \cdot \left(3 \cdot L^3 - 10 \cdot L^2 \cdot x + 9 \cdot L \cdot x^2 - 2 \cdot x^3\right)\right)\right) = 0 \end{split}$$

# Deflexão na extremidade em balanço

$$\delta_C := -v_{BC} \left( \frac{3 \cdot L}{2} \right) = \frac{1}{8} \frac{PL^3}{EI}$$

### Problema adicional #1: recalques de apoio



restart:

### Reacões de apoio

$$\begin{aligned} & assign \bigg( solve \bigg( \left\{ R_A - q \cdot L - P = 0, \, M_A - \left( q \cdot L \right) \cdot \frac{L}{2} \, - P \cdot L = 0 \right\}, \, \left\{ R_A, \, M_A \right\} \bigg) \bigg) : \\ & R_A, \, M_A = q \, L + P, \, \frac{1}{2} \, q \, L^2 + P \, L \end{aligned}$$

#### Momento fletor

$$\begin{split} M &:= \mathit{unapply}\Big(\mathit{solve}\Big(M + M_A - R_A \cdot x + (q \cdot x) \cdot \frac{x}{2} = 0, M\Big), x\Big) : \\ M(x) &= -\frac{1}{2} \ q \ L^2 - P \ L + L \ q \ x + P \ x - \frac{1}{2} \ q \ x^2 \end{split}$$

### Equação da curva de deflexão e rotação

$$\begin{aligned} bc &:= v(0) = v_0, \, \text{D[1]}(v)(0) = \theta_0: \\ v &:= unapply\Big(expand\Big(rhs\Big(dsolve\Big(\Big\{diff(v(x), x\$2) = \frac{M(x)}{EI}, bc\Big\}, v(x)\Big)\Big)\Big), x\Big): \\ \theta &:= unapply(expand(diff(v(x), x)), x): \\ v(x) &= -\frac{q\,x^4}{24\,EI} + \frac{L\,x^3\,q}{6\,EI} + \frac{x^3\,P}{6\,EI} - \frac{L^2\,x^2\,q}{4\,EI} - \frac{L\,x^2\,P}{2\,EI} + \theta_0\,x + v_0 \\ \theta(x) &= -\frac{q\,x^3}{6\,EI} + \frac{L\,x^2\,q}{2\,EI} + \frac{x^2\,P}{2\,EI} - \frac{L^2\,x\,q}{2\,EI} - \frac{L\,x\,P}{EI} + \theta_0 \end{aligned}$$

### Avaliação de rotação e deflexão em x = L

$$expand(\theta(L)) = -\frac{qL^3}{6EI} - \frac{L^2P}{2EI} + \theta_0$$

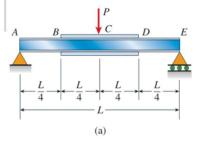
$$expand(v(L)) = -\frac{qL^4}{8EI} - \frac{L^3P}{3EI} + \theta_0L + v_0$$

### Problema adicional #2: Exemplo 9-13 (viga não prismática)

Referência: Gere & Goodno (2013). Mechanics of Materials, 8th Edition. Cengage Learning.

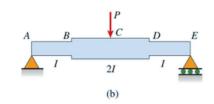
### • Example 9-13

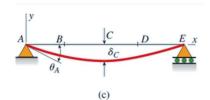
Example 9-13: Simple beam with two different moments of inertia



A beam *ABCDE* on simple supports is constructed from a wide-flange beam by welding cover plates over the middle half of the beam (Fig. 9-28a). The effect of the cover plates is to double the moment of inertia (Fig. 9-28b). A concentrated load *P* acts at the midpoint *C* of the beam.

Determine the equations of the deflection curve, the angle of rotation  $\theta_{\rm A}$  at the left-hand support, and the deflection  $\delta_{\rm C}$  at the midpoint (Fig. 9-28c).





restart:

### Reações de apoio

$$assign\left(solve\left(\left\{R_{A}+R_{E}-P=0,\ -P\cdot\frac{L}{2}\right.\right.+R_{E}\cdot L=0\right),\ \left\{R_{A},R_{E}\right\}\right)\right):$$
 
$$R_{A},R_{E}=\frac{P}{2},\ \frac{P}{2}$$

### Momento fletor

$$M_{AC} := solve(M - R_A \cdot x = 0, M) = \frac{1}{2} Px$$

### Equação da curva de deflexão

$$eqn := diff (v_{AB}(x), x\$2) = \frac{M_{AC}}{EI}, diff (v_{BC}(x), x\$2) = \frac{M_{AC}}{2 \cdot EI}:$$

$$bc := v_{AB}(0) = 0, D[1](v_{BC}) \left(\frac{L}{2}\right) = 0, v_{AB} \left(\frac{L}{4}\right) = v_{BC} \left(\frac{L}{4}\right), D[1](v_{AB}) \left(\frac{L}{4}\right) = D[1](v_{BC}) \left(\frac{L}{4}\right):$$

$$res := dsolve(\{eqn, bc\}, \{v_{AB}(x), v_{BC}(x)\}):$$

$$v_{AB} := unapply(simplify(rhs(res[1])), x):$$

$$v_{BC} := unapply(simplify(rhs(res[2])), x):$$

$$v_{AB}(x) = -\frac{Px(15L^2 - 32x^2)}{384EI}$$

$$v_{BC}(x) = -\frac{P(L^3 + 24L^2x - 32x^3)}{768EI}$$

# Ângulo de rotação em A e deflexão em C

$$\theta_{A} := -D[1](v_{AB})(0) = \frac{5}{128} \frac{PL^{2}}{EI}$$

$$\delta_{C} := -v_{BC}(\frac{L}{2}) = \frac{3}{256} \frac{PL^{3}}{EI}$$