Question 2:

Suppose in Question #1 that a pitot tube can measure the speed of the aircraft at a rate of 20Hz. These measurements are corrupted by noise, modeled as white and WSS with a variance of 8 ft/sec. The measurements are uncorrelated with the disturbances at all times.

Assume that a Kalman filter is implemented to estimate the states based on these measurements. Letting $\mathbf{x}(t)$ be the states of the complete physical model, assume that $E[\mathbf{x}(0)] = 0$ and $E[\mathbf{x}(0)\mathbf{x}(0)^T] = \frac{1}{2}\mathbf{I}$. Plot the evolution in RMS uncertainty of the true values of the two outputs (i.e. $E[\tilde{y}_1(t)^2]$ and $E[\tilde{y}_2(t)^2]$). What is the steady-state RMS uncertainty in each output?

Hint: Note that the measurements (airspeed) used by the filter in this problem are different from the outputs we are interested in estimating! This poses no conceptual difficulties; just use a \boldsymbol{C} matrix in the design of the estimator that reflects the actual measurement, then use the original \boldsymbol{C} matrix to assess the output estimation error projects.

Given:-
$$\forall k = Hk \ Kk + \emptyset$$
 $R_k = 8 \ ft | Se = Var [\emptyset]$
 $E[\forall_k \forall_k] = 0$: only on measurement

$$E\left[\underline{x}(0)\right] = 0 \qquad P_0 = E\left[\underline{x}(0)\underline{x}(0)^{T}\right] = \underline{\bot}\underline{T}$$

$$\frac{\hat{x}}{2} = 0$$

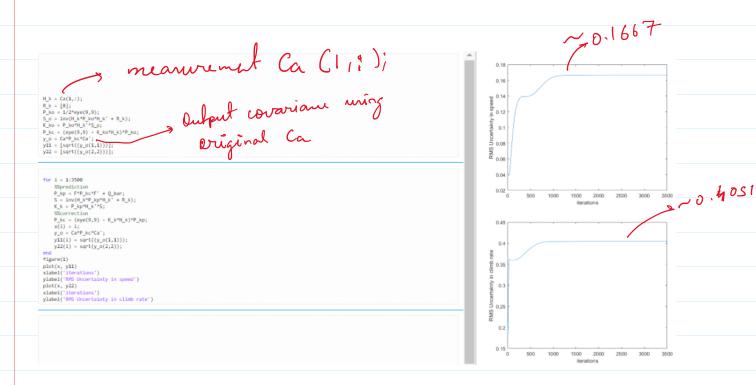
ung me Kalman Filter algorithm,

```
Hk = 1st evan of Ca of augmentel
materin. (: one meanirement
available)
```

Coding me algorismm in matlate

```
H_k = Ca(1,:);
R_k = [8];
P_ko = 1/2*eye(9,9);
S_o = inv(H_k*P_ko*H_k' + R_k);
K_ko = P_ko*H_k'*$_0;
P_kc = (eye(9,9) - K_ko*H_k)*P_ko;
y_o = Ca*P_kc*Ca';
y11 = [sqrt((y_o(1,1)))];
y22 = [sqrt((y_o(2,2)))];
```

```
for i = 1:3500
    %%prediction
    P_{kp} = F^*P_{kc}^*F' + Q_{bar}^*;
    S = inv(H_k*P_kp*H_k' + R_k);
    K_k = P_{kp}*H_k'*S;
    %%correction
    P_kc = (eye(9,9) - K_k*H_k)*P_kp;
x(i) = i;
    y_0 = Ca*P_kc*Ca';
    y11(i) = sqrt((y_o(1,1)));
    y22(i) = sqrt(y_0(2,2));
figure(1)
plot(x, y11)
xlabel('iterations')
ylabel('RMS Uncertainty in speed')
plot(x, y22)
xlabel('iterations')
ylabel('RMS Uncertainty in climb rate')
```



ung "dare" funtion to chuk me steady state

ung "dare funtion to chak the steady state

RMS unentainty in each output,

Po = dare (F', H', O, R)

uning Discrete algebraic

Ricalli earatori.

Ks = Po x M_K x (H_K x Roo x M_K) + R_K)

Po = (I - Ko M_K) Po -

 $E \left[\frac{\dot{y}}{3}, (t)^{2} \right] = \int 0.0278 \approx 0.16673$ Cspeed at steady state) $E \left[\frac{\dot{y}}{3}, (t)^{2} \right] = \int 0.1651 \approx 0.40509$ Cdimb rate at steady state