

EE650 Mini Project Presentation



Cantilever Oscillation Control

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System Description and deriving state space model

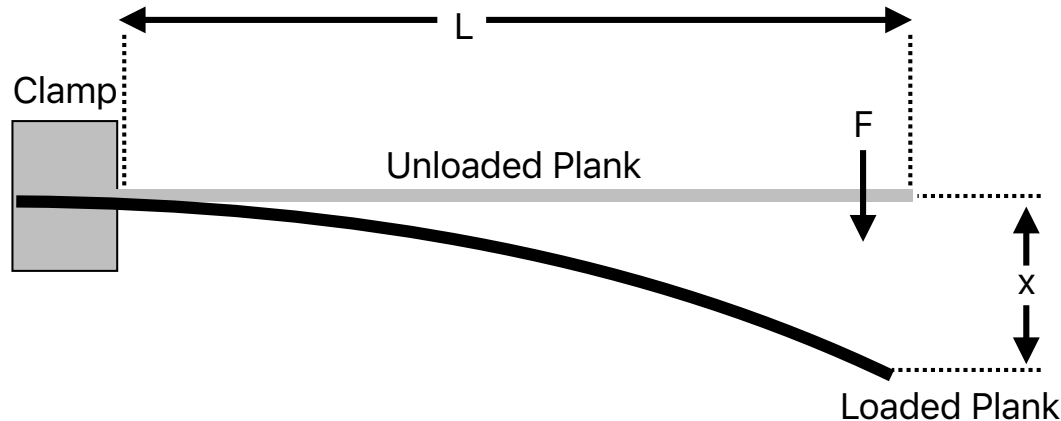
Let F is force applied and x is vertical displacement.

$$x_1 = x \quad \text{and} \quad x_2 = x'$$

Applying Newton's second Law:

$$mx'' + kx + cx' = F$$

Here F is input Force i.e. u (*control input*) = F



State Space Model (Cont...)

Using the values of x_1 and x_2 we get:

$$\dot{x}_2 = -kx_1/m - cx_2/m + u/m$$

$$\dot{x}_1 = x_2$$

The above equations are the state space equations.

As we want to control the vertical displacement i.e. x_1 ,

$$y = x_1 + 0 \cdot u = x_1 \rightarrow \text{Output Equation}$$

Using above equation we can find the A, B, C and D of state space model

State variables, Controllability and Observability matrices

$x_1 \rightarrow$ vertical deflection, $x_2 \rightarrow$ vertical velocity of the edge of the plank

$$(i) \quad \dot{x}_1 = x_2$$

$$(ii) \quad \dot{x}_2 = -\frac{k}{m} x_1 - \frac{c}{m} x_2 + \frac{u}{m}$$

$$\Rightarrow \dot{x} = \begin{bmatrix} 0 & 1 \\ -\frac{k}{m} & -\frac{c}{m} \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 0 \\ \frac{1}{m} \end{bmatrix} u$$

$$y = \begin{bmatrix} 1 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 0 \end{bmatrix} u$$

Controllability Matrix (Q_c)

$$Q_c = \begin{bmatrix} B & AB \end{bmatrix} \\ = \begin{bmatrix} 0 & 1/m \\ 1/m & -c/m^2 \end{bmatrix}$$

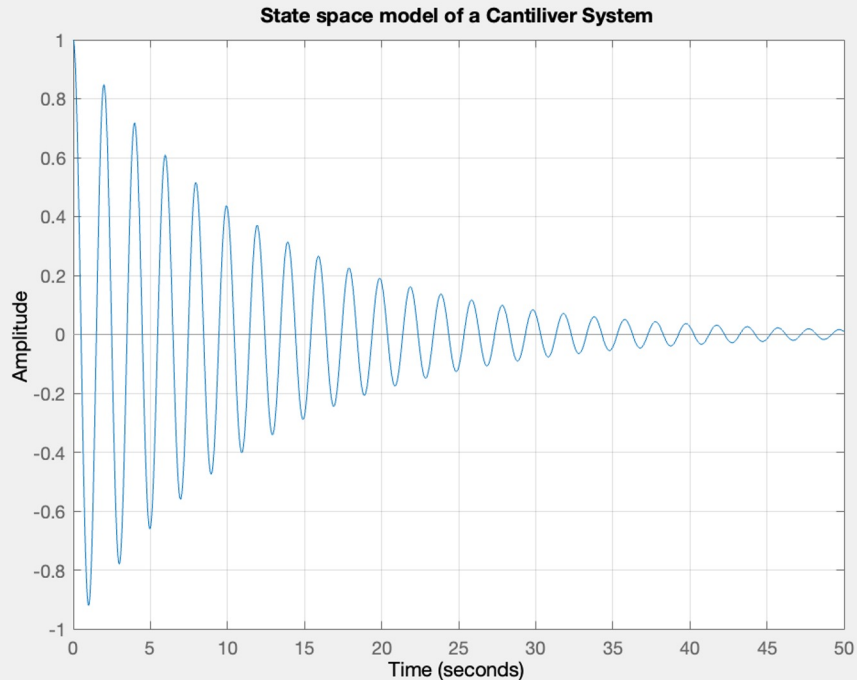
\rightarrow Rank = 2
Hence the system is controllable.

Observability Matrix (Q_o)

$$Q_o = \begin{bmatrix} C \\ CA \end{bmatrix} = \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix}$$

Rank = 2 \rightarrow Hence the system is observable.

System Response without controller:



```
>> stepinfo(sys)
```

ans =

struct with fields:

```
RiseTime: 0.3363
SettlingTime: 46.7473
SettlingMin: 2.5441e-04
SettlingMax: 0.0032
Overshoot: 92.0519
Undershoot: 0
Peak: 0.0032
PeakTime: 0.9935
```

Objective of the controller

1. To decrease the peak overshoot of the mass dashpot
2. To decrease the settling time of the mass dashpot

Benefits:

- Reduces the chance of an accident
- Increases the lifetime of swimming board
- Avoiding damage due to stress

Design of the state feedback controller

- To design a controller for the system we need to check the controllability of the system.
- The system is controllable if $C = [B \ AB]$ is a full rank matrix
- For our system, $C = [0 \ 0.0167; 0.0167 \ -0.0028]$, $rank(C) = 2$, hence the system is controllable
- At present the poles of the system are located at $-0.083 + 3.16i$, $-0.083 + 3.16i$ which can be found by $\det(sI - A)$
- In order to meet the desired objective of the controller we place the poles of the controlled system at $-10 - 10i$, $-10 + 10i$ using a state feedback controller

- Taking the constants:
 - $k = 600$,
 - $c = 10$,
 - $m = 60$,
- The state feedback controller is given by:
 - $u = -k * x$
 - $k = [k1 \ k2]$
- For finding $[k1 \ k2]$, we use MATLAB

```
Command Window
>> Co
Co =
      0      0.0167
  0.0167  -0.0028

>> rank(Co)
ans =
      2

>> k
k =
  1.0e+04 *
      1.1400      0.1190

>> |
```

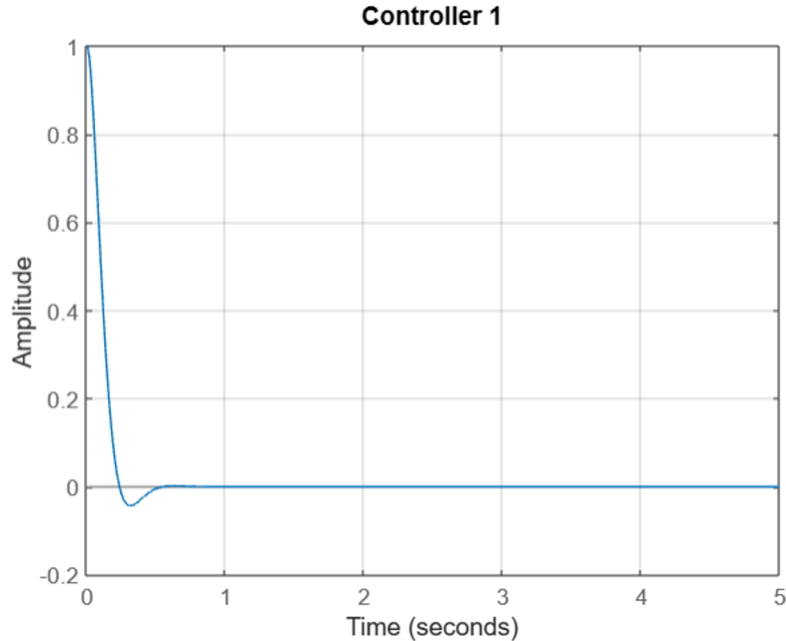

Code for controller

/MATLAB Drive/project.m

```
1  clc;clear,close all
2  %% State Space Model
3  k=600;
4  c=10;
5  m=60;
6  A = [0 1;-k/m -c/m];
7  B = [0; 1/m];
8  C = [1 0];
9  D = 0;
10 eigenVal = eig(A);
11 t=0:0.0001:5;
12 u=zeros(size(t));|
13 sys = ss(A,B,C,D);
14 x0=[1 0];
15 %
16 figure(1)
17 lsim(sys,u,t,x0)
18 title('State space model of a Cantiliver System')
19 grid on
```

```
20 %% Controller
21 Co=ctrb(A,B)
22 rank(Co)
23 k=place(A,B,[-10+10i, -10-10i]); %poles for 2 dimensional matrix
24 sys_cl=ss(A-B*k,B,C,0);
25 eig(sys_cl);
26 %
27 figure(2)
28 lsim(sys_cl,u,t,x0)
29 title('Controller 1')
30 grid on
31 %
32 k1=place(A,B,[-20+20i, -20-20i]); %poles for 2 dimensional matrix
33 sys_cl2=ss(A-B*k1,B,C,0);
34 eig(sys_cl2);
35 %
36 figure(3)
37 lsim(sys_cl2,u,t,x0)
38 title('Controller 2')
39 grid on
```

System Response after using the Controller 1



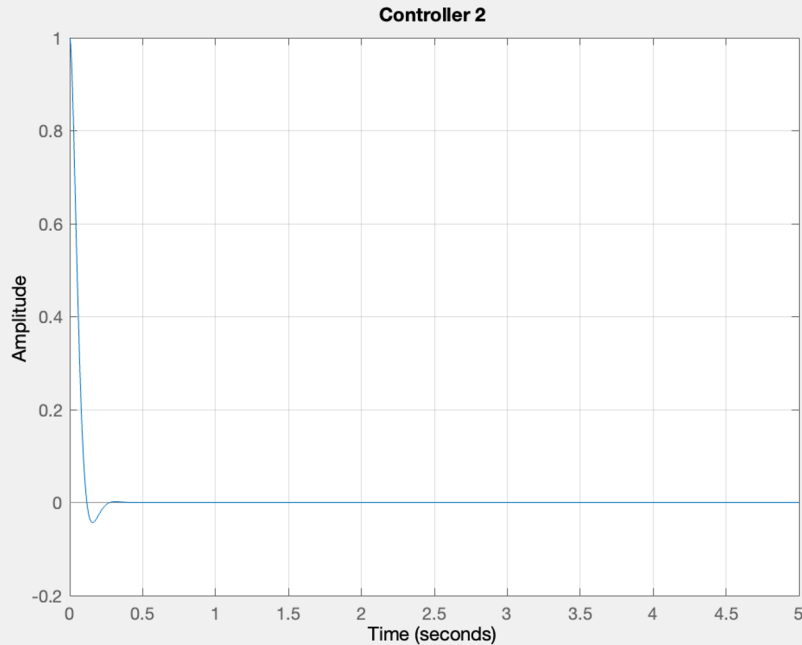
```
>> stepinfo(sys_cl)
```

```
ans =
```

struct with fields:

```
RiseTime: 0.1519  
SettlingTime: 0.4216  
SettlingMin: 7.5286e-05  
SettlingMax: 8.6934e-05  
Overshoot: 4.3210  
Undershoot: 0  
Peak: 8.6934e-05  
PeakTime: 0.3132
```

System Response after using the Controller 2



```
>> stepinfo(sys_cl2)
```

```
ans =
```

struct with fields:

```
    RiseTime: 0.0760
  SettlingTime: 0.2108
  SettlingMin: 1.8821e-05
  SettlingMax: 2.1734e-05
    Overshoot: 4.3210
    Undershoot: 0
         Peak: 2.1734e-05
    PeakTime: 0.1566
```

Justification for the Performance of the Controller

1. Decrease in the peak overshoot of the oscillations:

- Peak overshoot without controller = 92.05
- Peak overshoot with controller = 4.32
- **Decrease in peak overshoot = 95.3%**

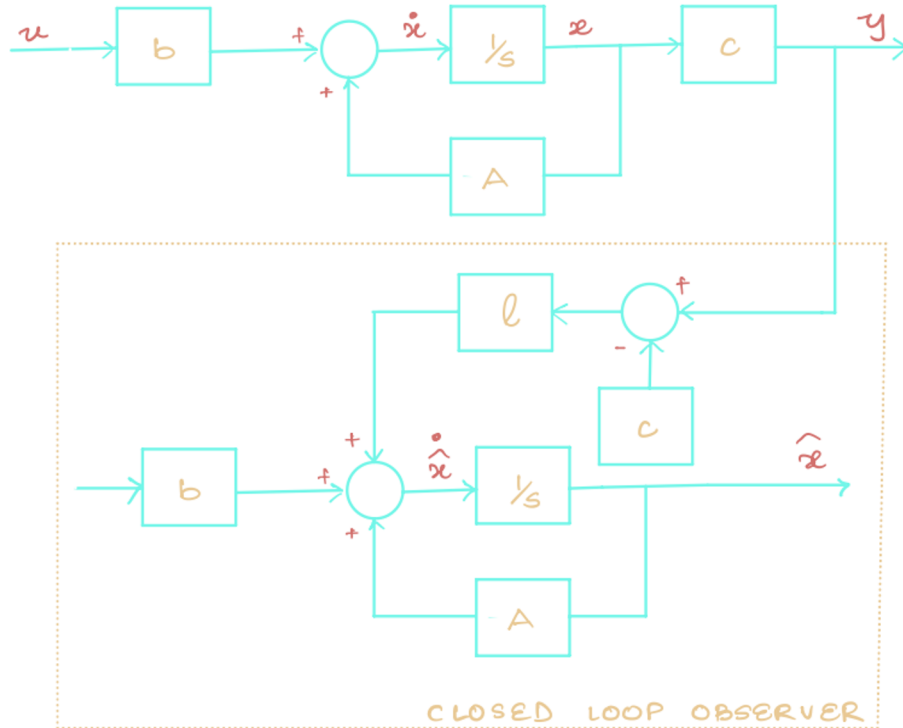
2. Decrease in the settling time of the oscillations:

- Settling time without controller = 46.7473
- Settling time with controller 2 = 0.2108
- **Decrease in Settling time = 99.5%**

Design of the Luenberger observer

- To design an observer we need to check the observability of the system
- The system is observable if the matrix $\mathbf{Q} = [\mathbf{C} ; \mathbf{CA}]$ is full rank
- $\mathbf{Q} = [\mathbf{1} \ \mathbf{0} ; \mathbf{0} \ \mathbf{1}]$, rank of \mathbf{Q} is 2 hence the system is observable
- In order to design the Luenberger observer we need to find the observer gain \mathbf{L}
- We need to design \mathbf{L} such that the eigenvalues of $\mathbf{A} - \mathbf{L} * \mathbf{C}$ lie in OLHP
- We consider the eigenvalues of $\mathbf{A} - \mathbf{L} * \mathbf{C}$ to be $-\mathbf{10}, -\mathbf{11}$
- Using MATLAB we find \mathbf{L}

Block Diagram



- $\dot{\hat{x}} = (A - LC)\hat{x} + Bu + Ly$
- $e(t) = x(t) - \hat{x}(t)$
- $\dot{e} = (A - LC)e$

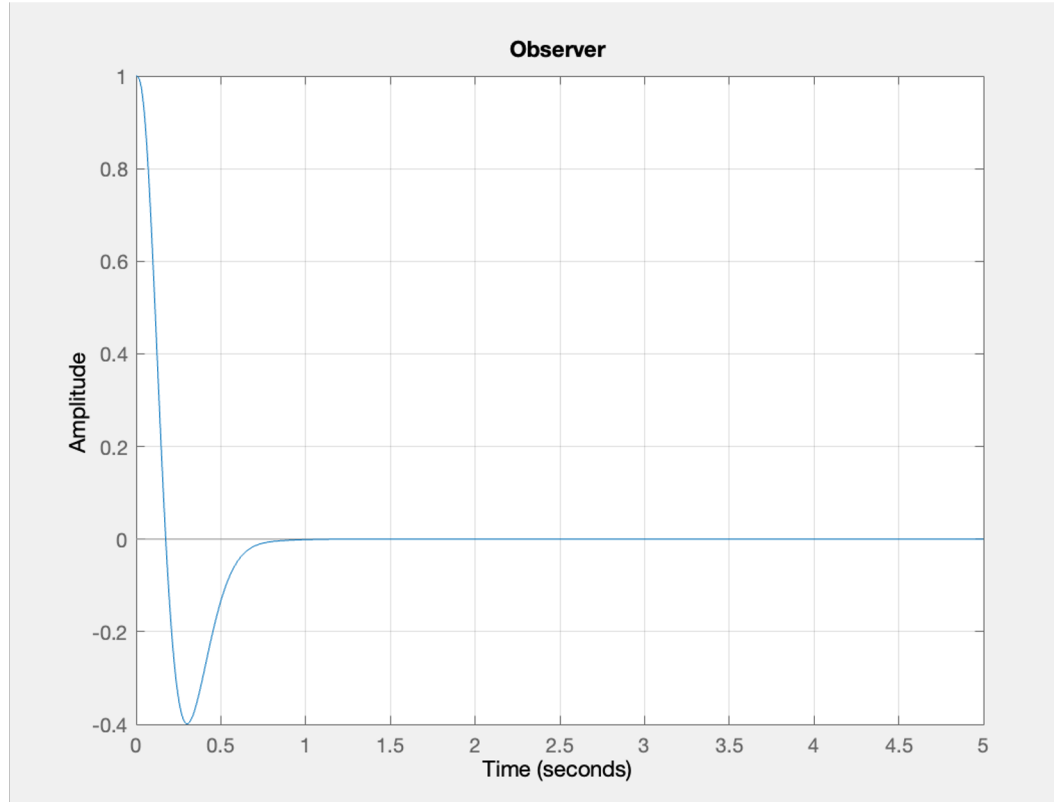
Finding Observer gain

```
>> ob  
  
ob =  
  
    1    0  
    0    1  
  
>> rank(ob)  
  
ans =  
  
    2  
  
>> l=place(A',C', [-10,-11])'  
  
l =  
  
    20.8333  
    96.5278
```

Code for observer with state estimator

```
%% Luenberger Observer
ob=obsv(A,C)
Ro=rank(ob)
l=place(A',C', [-10,-11])'; %k'
%
at=[A-B*k B*k;zeros(size(A)) A-l*C];
bt=[B; zeros(size(B))];
ct=[C zeros(size(C))];
sys_o=ss(at,bt,ct,0);
% x1=[0 0];
%
figure(4)
lsim(sys_o,u,t,[x0 x0])
title('Observer')
grid on
```

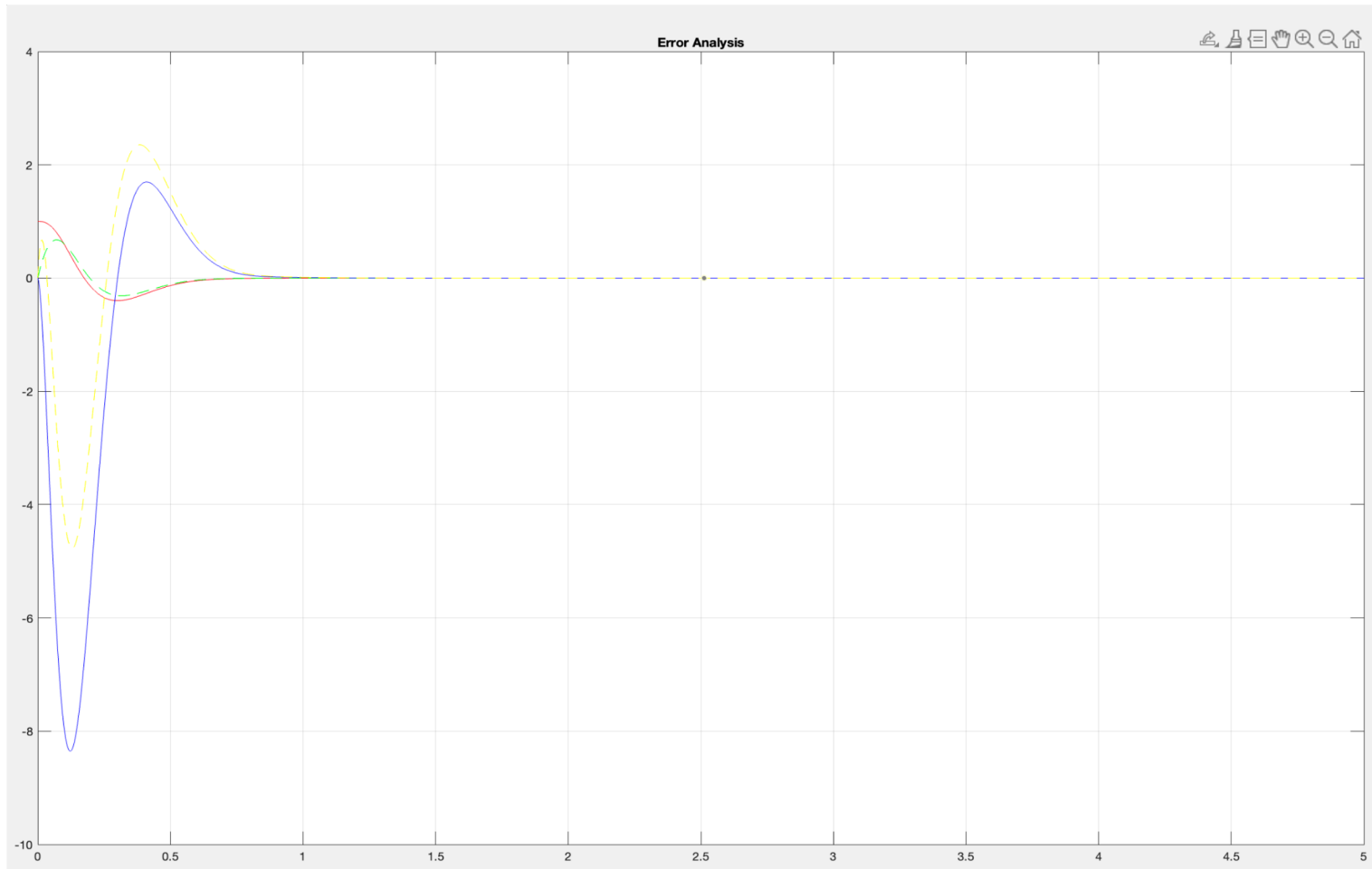
Observer Output



Error Analysis

```
%% Error analysis
n=2;
[y,t,x]=lsim(sys_o,u,t,[x0 x0]);
e=x(:,n+1:end);
x=x(:,1:n);
xe =x-e ; %obs
x1=x(:,1);x2=x(:,2); %plant states
xe1=xe(:,1);xe2=xe(:,2); %observer states
%
figure(5)
plot(t,x1,'-r',t,xe1,'--g',t,x2,'-b',t,xe2,'--y')
title('Error Analysis')
grid on
%
```

Plots



Justification for the Performance of the Observer

- The estimation error for both the states goes to zero
- The estimated states follow the system states
- The observer archives its performance

Thank You