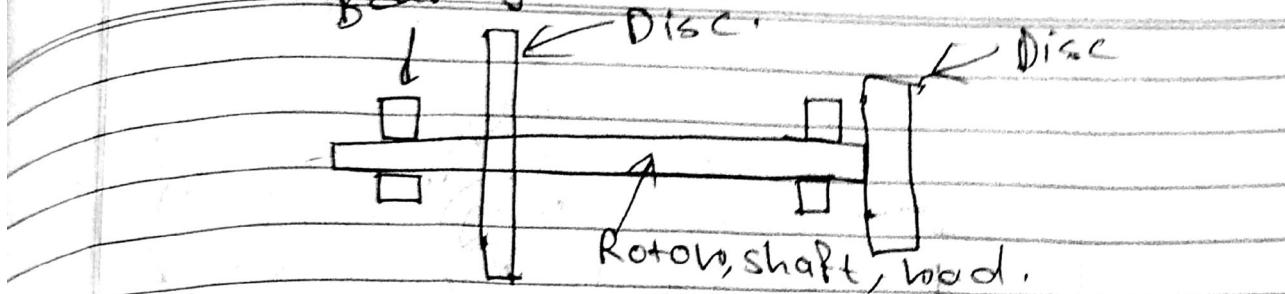
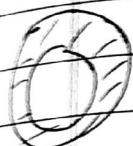
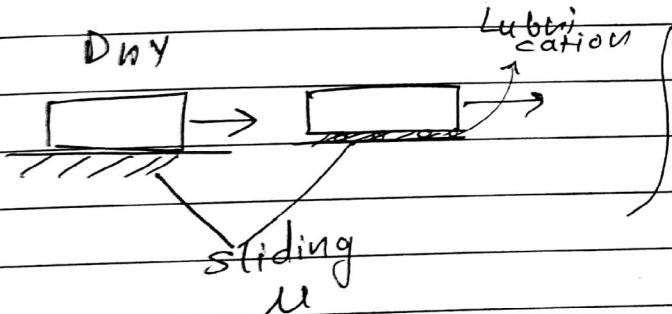
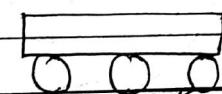


bearing.impeller vs propellerRolling Element Bearing (REB)→ Non conformal (only line contact)Journal Bearing→ Conformal (face contact)Much less pressure than REB as surface area taking the load is large.→ Load = $P = \frac{w}{D \times L}$ = Projected area
(D × L)DryLubricationRolling.Rolling resistanceAnti-friction.Journal Bearing → starting friction ishigh but afterwards itbecomes veryless asJournal loses contact withbearing.

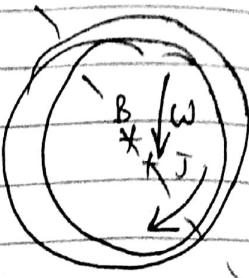
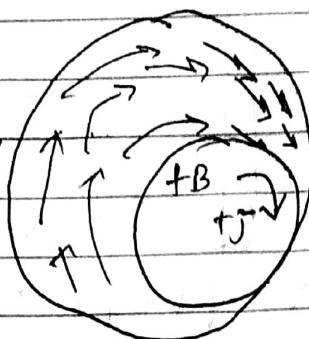
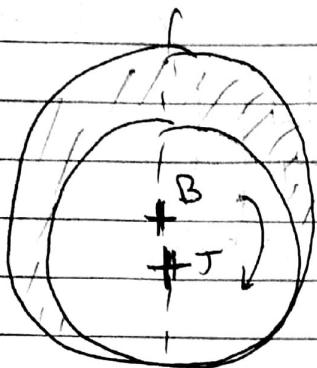
Hydrodynamic \rightarrow pressure is developed automatically by rotating motion

classmate

Date _____

Page _____

Hydrostatic \rightarrow pressure is artificially provided.



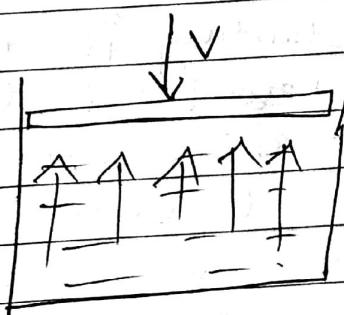
Intermediate

Lubricant flow from diverging to converging zone and high pressure will be developed in converging zone

At operating speed stable configuration



pressure distribution



Squeeze-film

classmate
Date _____
Page _____

Hybrid bearing (Based on loading on type of lubrication)

① Hydrostatic + Hydrodynamic

② Axial + Thrust

Tilting pad makes more stable bearing.

Herringbone grooved journal bearing.

Textured thrust pad.

For gas/air lubrication:

requires very high RPM to ~~develop~~ develop a pressure.

Surface finish needs to be very high σ in nanometers (nm)

Surface finish on lathe is ~~5~~ 5 micron.



Stable



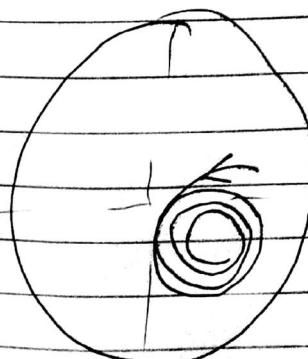
whirling.

shaft rotates

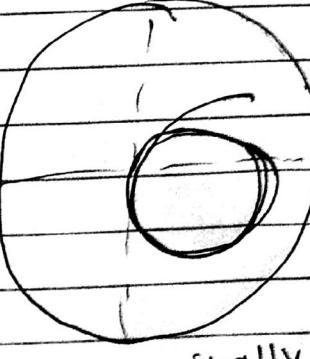
about a line other than its

centreline.

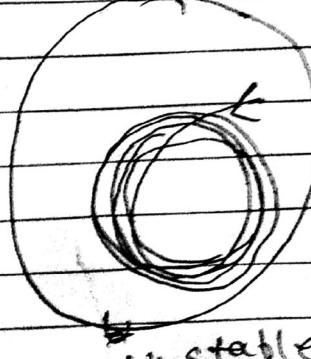
Journal will spin about its centre and whirl about bearing centre.



Stable



Marginal stability



Unstable.

Tunnel bearing - Megapascal pressure
Rolling \rightarrow Mega pascal pressure

classmate

Date _____

Page _____

Deep groove Ball

- (1) Contact angle up to 5°

- (2) No axial load

Angular contact ball

- (1) contact angle up to 30°

- (2) Can bear axial load.

Salome-Meca

Assignment

What is natural frequency

resonating frequency

resonance

critical speed

modal frequency.

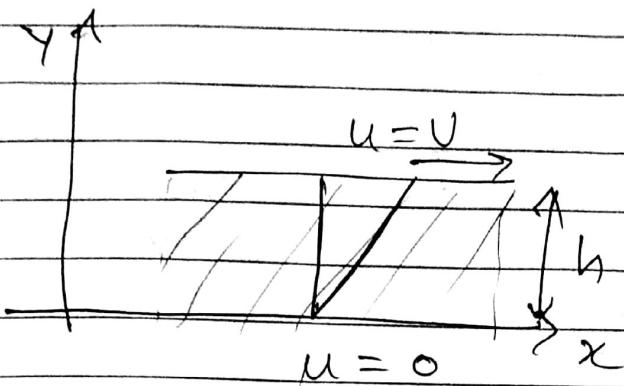
Properties of lubricant

Viscosity	Specific heat
Density	Thermal Conductivity Conductivity
	Acidity and Alkalinity
	Oxidation stability
	Flash Point \rightarrow catches fire temp at which

Foaming

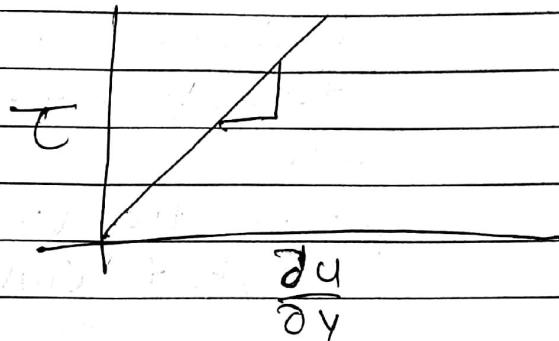
Pour point

Demulsibility

Viscosity →For a newtonian fluid

$$\tau \propto \frac{\partial u}{\partial y}$$

Shear stress Shear strain rate



$$\tau = n \frac{\partial u}{\partial y}$$

↓
Coefficient of viscosity,
absolute
(dynamic)

$$\tau = n \frac{\partial u}{\partial y} = n \frac{U}{h}$$

S.I C.G.S

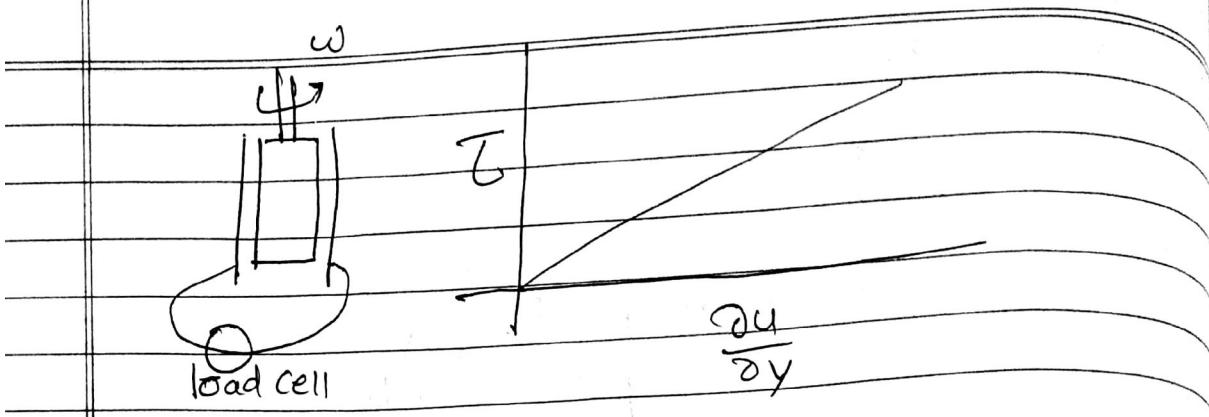
$$n = \frac{\tau h}{U} = \left[\frac{\text{Pa.s}}{\text{m/s}} \right] = \left[\text{Pa.s} \right] \frac{\text{dynes}}{\text{cm}^2}$$

C P (CentiPoise) → Poise.
↳ Lubricant

Kinematic Viscosity

$$\nu = \frac{n}{\rho} = \frac{\text{m}^2}{\text{s}} \Rightarrow \frac{\text{cm}^2}{\text{s}} = \boxed{\text{stoke}}$$

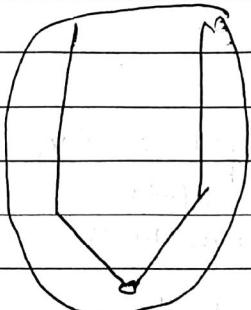
↓
(Centi Stoke) CS



Innen rotates
outer which
restricted by a
load cell and
measures torque
to overcome shear
stress.

Kinematic Viscosity measurement

Saybolt Universal Seconds



$$U = \left(0.22 t - \frac{180}{t} \right) \text{ CS}$$

time

Required to Fluid
come out

Density measurement

Measure weight, Measure gravity
(using pendulum)

then volume.

Effect of temp on Viscosity

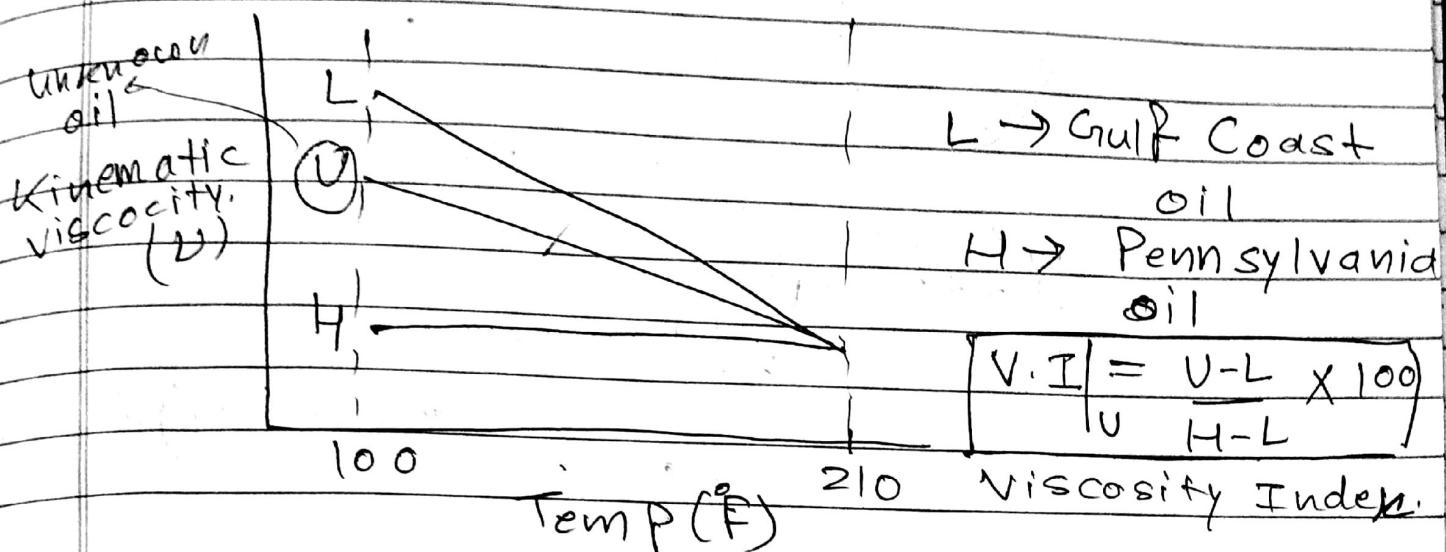
Temp ↑ Oil ↓ Gas ↑

Viscosity
is because
of inter
molecular
interaction.

Viscosity
because of
random movement
of molecules.

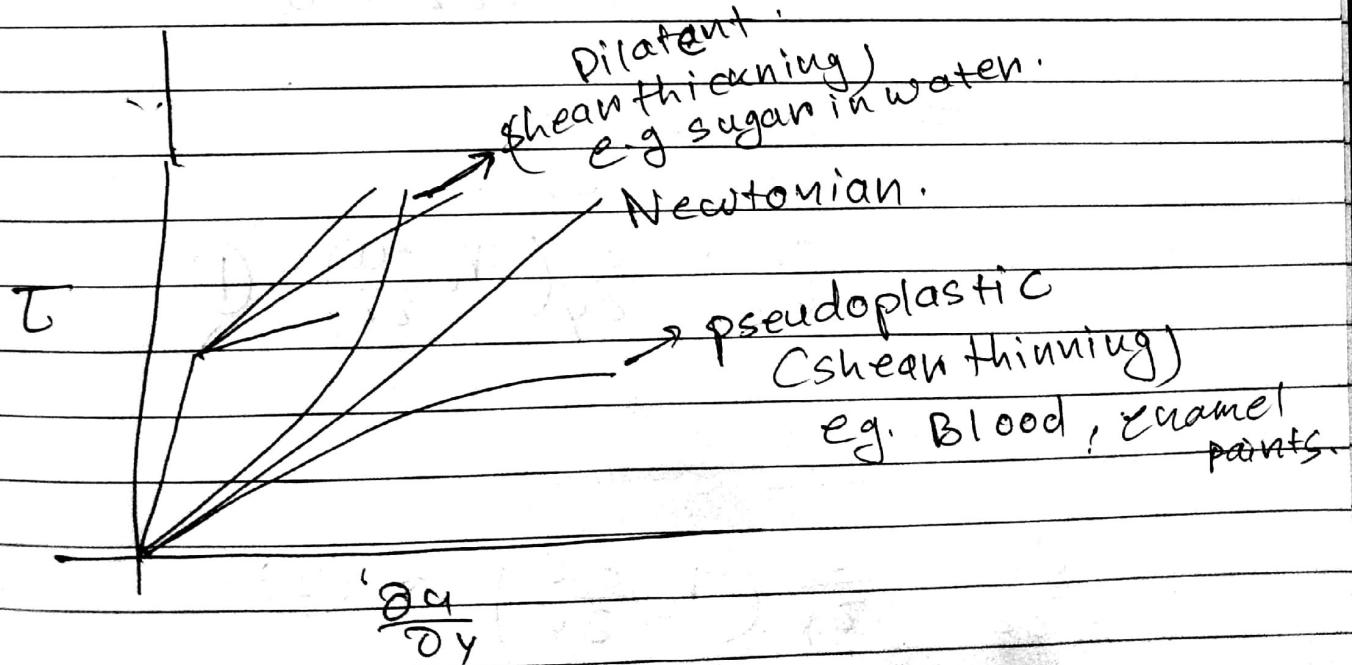
ASTM

$$\log_{10} \log_{10} (V + 0.8) = n \log_{10} T + c$$

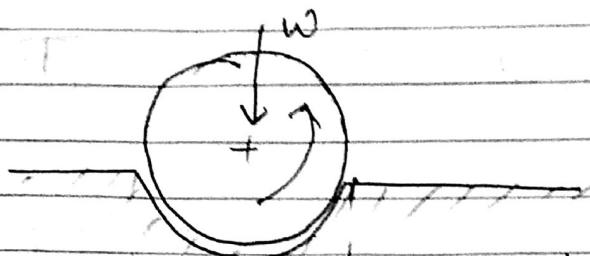
Effect of pressure on Viscosity

Pressure \uparrow oil \uparrow as \uparrow

Banus Relationship $\eta = \eta_0 e^{kP}$ $\xrightarrow{\text{pressure}} \downarrow$
 Viscosity at ambient temp $\xrightarrow{\text{viscosity coefficient}}$



Lubrication Technology



Experiment of Tower
(1883)

Reynolds eqn (1886)

$$\text{actual } P = \frac{w}{LD} \text{ assumed}$$

$$(T + \frac{\partial T}{\partial y} dy) dx dz$$

$$P dy dz$$

$$(P + \frac{\partial P}{\partial x} dx) dy dz$$

$$T dx dz$$

x-direction

$$P dy dz + (T \frac{\partial T}{\partial y} dy) dx dz - (P + \frac{\partial P}{\partial x} dx) dy dz$$

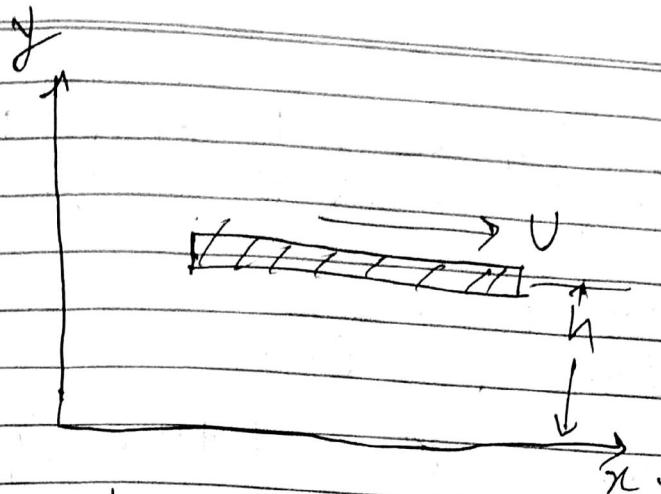
$$- T dx dz = 0$$

$$\Rightarrow \frac{\partial T}{\partial y} = \frac{\partial P}{\partial x}$$

$$\frac{\partial P}{\partial x} = \frac{\partial}{\partial y} \left(n \frac{\partial u}{\partial y} \right) - ①$$

z-direction

$$\frac{\partial P}{\partial z} = \frac{\partial}{\partial y} \left(n \frac{\partial \omega}{\partial y} \right) - ②$$



Boundary conditions

at $y=0, u=0, w=0$

at $y=h, u=U, w=0$

Integrating eqn ① and ②

$$u = \frac{1}{2n} \frac{\partial P}{\partial x} y(y-h) + \frac{U}{h} y$$

$$w = \frac{1}{2n} \frac{\partial P}{\partial z} y(y-h)$$

Flow in x and z directions per unit width.

$$q_x = \int_0^h u dy, q_z = \int_0^h w dy$$

$$q_x = -\frac{h^3}{12n} \frac{\partial P}{\partial x} + Uh$$

$$q_z = -\frac{h^3}{12n} \frac{\partial P}{\partial z}$$

For steady flow

$$\frac{\partial q_x}{\partial x} + \frac{\partial q_z}{\partial z} = 0$$

Navier S

$$\frac{\partial}{\partial x} \left(\frac{h^3}{n} \frac{\partial P}{\partial x} \right) + \frac{\partial}{\partial z} \left(\frac{h^3}{n} \frac{\partial P}{\partial z} \right) \quad \begin{array}{l} \text{Pressure} \\ \text{induced flow} \\ \text{Poiseuille} \\ \text{term} \end{array}$$

$$= 6U \frac{\partial h}{\partial x} + ch \frac{\partial U}{\partial x}. \quad \begin{array}{l} \text{Velocity} \\ \text{induced} \\ \text{flow} \\ (\text{Couette term}) \end{array}$$

↓

Stretch term. (most of
the time
is zero
as U is cons
term)

$$\boxed{\frac{\partial}{\partial x} \left(\frac{h^3}{n} \frac{\partial P}{\partial x} \right) + \frac{\partial}{\partial z} \left(\frac{h^3}{n} \frac{\partial P}{\partial z} \right) = 6U \frac{\partial h}{\partial x}}$$

↳ steady
state
Reynolds eqn.

Navier-Stokes equation:

$$\textcircled{1} \quad \rho \frac{D u}{D t} = \rho g - \frac{\partial P}{\partial x} + \frac{\partial}{\partial x} \left[n \left[2 \frac{\partial u}{\partial x} - 2 \frac{\partial v}{\partial x} + \frac{\partial v}{\partial y} + \frac{\partial w}{\partial z} \right] \right]$$

$$\textcircled{2} \quad \rho \frac{D v}{D t} = - \dots \quad \textcircled{3} \quad \rho \frac{D w}{D t} = - \dots$$

$$+ \frac{\partial}{\partial y} \left[n \left(\frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \right) \right] + \frac{\partial}{\partial z} \left[n \left(\frac{\partial w}{\partial x} + \frac{\partial u}{\partial z} \right) \right]$$

$$\frac{D u}{D t} = u \frac{\partial u}{\partial x} + v \frac{\partial v}{\partial y} + w \frac{\partial w}{\partial z} + \frac{\partial u}{\partial t}$$

$$\textcircled{4} \quad \frac{\partial \rho}{\partial t} + \frac{\partial}{\partial x} (\rho u) + \frac{\partial}{\partial y} (\rho v) + \frac{\partial}{\partial z} (\rho w) = 0$$

continuity eqn

Navier Stokes \rightarrow Reynolds Assumptions
Assumptions of thin film lubrication.

- ① Inertia and body forces ~~are~~ are negligible as compared to pressure and viscous terms
- ② The film is thin, so there is no variation of pressure across the fluid film.
- ③ There is no slip across the fluid boundaries.
- ④ No external forces act on the film
- ⑤ The flow is viscous and laminar.
- ⑥ The derivatives of u and w w.r.t y are much larger than other derivatives of velocity component.
- ⑦ height of the fluid film should be very small compared to the length of contact (l_c) $\Rightarrow \leq 10^{-3}$

$$\left. \frac{\partial w}{\partial z} \right\} \quad 0 = -\frac{\partial P}{\partial x} + \frac{\partial}{\partial y} \left(\eta \frac{\partial u}{\partial y} \right) \quad \left. \right\}$$

$$0 = -\frac{\partial P}{\partial y} \quad \text{momentum}$$

$$0 = -\frac{\partial P}{\partial z} + \frac{\partial}{\partial y} \left(\eta \frac{\partial w}{\partial y} \right)$$

$$\frac{\partial S}{\partial t} + \frac{\partial}{\partial x} (su) + \frac{\partial}{\partial y} (sv) + \frac{\partial}{\partial z} (sw) = 0$$

\nearrow
continuity.

$$u = \frac{1}{2n} \frac{\partial P}{\partial x} y^2 + c_1 y + c_3$$

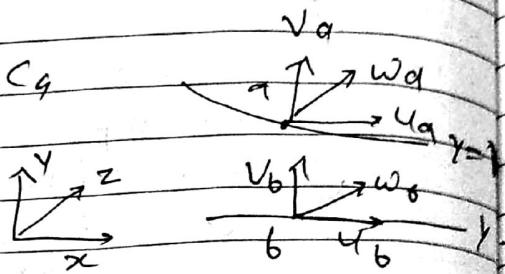
$$w = \frac{1}{2n} \frac{\partial P}{\partial z} y^2 + c_2 y + c_4$$

at $y = 0$

$$u = u_b, w = w_b$$

at $y = h$

$$u = u_a, w = w_a$$



$$u = \frac{1}{2n} \frac{\partial P}{\partial x} y(y-h) + \left(\frac{h-y}{h}\right) u_b + \frac{y}{h} u_a$$

$$w = \frac{1}{2n} \frac{\partial P}{\partial z} y(y-h) + \left(\frac{h-y}{h}\right) w_b + \frac{y}{h} w_a$$

$$\begin{aligned} \frac{\partial}{\partial y} (\epsilon v) &= -\frac{1}{2} \left\{ \frac{\partial}{\partial x} \left[\frac{\epsilon}{n} \frac{\partial P}{\partial x} y(y-h) \right] \right. \\ &\quad \left. + \frac{\partial}{\partial z} \left[\frac{\epsilon}{n} \frac{\partial P}{\partial z} y(y-h) \right] \right\} \\ &\quad - \frac{\partial}{\partial x} \left[\frac{\epsilon}{n} \left\{ \left(\frac{h-y}{h} \right) u_b + \frac{y}{h} u_a \right\} \right] \\ &\quad - \frac{\partial}{\partial z} \left[\frac{\epsilon}{n} \left\{ \left(\frac{h-y}{h} \right) w_b + \frac{y}{h} w_a \right\} \right] \\ &\quad - \frac{\partial \epsilon}{\partial t} \end{aligned}$$

$$\begin{aligned} \epsilon(v_a - v_b) &= -\frac{1}{2} \left\{ \int_0^h \frac{\partial}{\partial x} \left[\frac{\epsilon}{n} \frac{\partial P}{\partial x} y(y-h) \right] dy \right. \\ &\quad \left. + \int_0^h \frac{\partial}{\partial z} \left[\frac{\epsilon}{n} \frac{\partial P}{\partial z} y(y-h) \right] dy \right\} \\ &\quad - \int_0^h \frac{\partial}{\partial x} \left[\frac{\epsilon}{n} \left\{ \left(\frac{h-y}{h} \right) u_b + \frac{y}{h} u_a \right\} \right] dy \\ &\quad - \int_0^h \frac{\partial}{\partial z} \left[\frac{\epsilon}{n} \left\{ \left(\frac{h-y}{h} \right) w_b + \frac{y}{h} w_a \right\} \right] dy \end{aligned}$$

$$-\frac{h \partial s}{\partial t}$$

Leibnitz rule.

$$\int_0^h \frac{\partial f(x, y, z)}{\partial x} dz = \frac{\partial}{\partial x} \int_0^h f(x, y, z) dz - F(x, y, h) \frac{\partial h}{\partial x}$$

Generalized Reynolds Equation

$$\frac{\partial}{\partial x} \left(\frac{sh^3}{12\eta} \frac{\partial P}{\partial x} \right) + \frac{\partial}{\partial z} \left(\frac{sh^3}{12\eta} \frac{\partial P}{\partial z} \right) = \left(\frac{u_a + u_b}{2} \right) \frac{\partial (sh)}{\partial x}$$

Poiseuille
Term

(pressure
induced)

$$+ \left(\frac{w_a + w_b}{2} \right) \frac{\partial}{\partial z} (sh) + \frac{\partial (sh)}{\partial t}$$

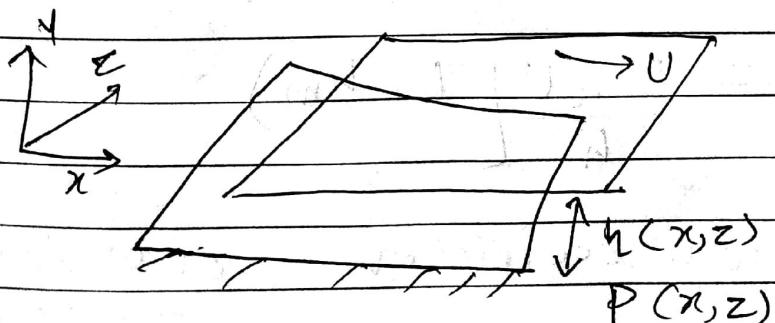
Couette term
(velocity induced)

Squeeze
term

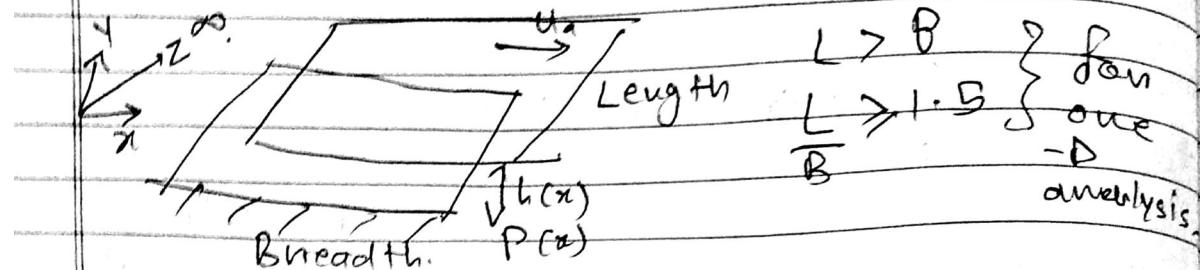
IF density variation can be neglected
during bearing operation and assuming
boundary velocities in only one direction
the above equation reduces to :
(and steady state is assumed)

$$\frac{\partial}{\partial x} \left(\frac{h^3}{12\eta} \frac{\partial P}{\partial x} \right) + \frac{\partial}{\partial z} \left(\frac{h^3}{12\eta} \frac{\partial P}{\partial z} \right) = \frac{u_a + u_b}{2} \frac{\partial h}{\partial x}$$

$\left(\frac{u_a + u_b}{2} \right)$



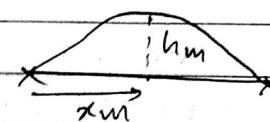
~~No~~ Assuming no variation is z-direction
i.e. in z direction is assumed to be infinite length



$$\frac{\partial}{\partial x} \left(\frac{h^3}{n} \frac{\partial P}{\partial x} \right) = 12 U \frac{\partial h}{\partial x}.$$

$$\frac{\partial P}{\partial x} = \frac{12 n U}{h^2} + \frac{n}{h^3} C_1$$

At both open ends, pressure will be atmospheric thus gauge pressure at both the ends will be zero.



This pressure will reach a maximum value

In between the two ends.

$$\therefore \frac{\partial P}{\partial x} = 0, x = x_m, h = h_m$$

$$0 = \frac{12 n U}{h_m^2} + \frac{n}{h_m^3} C_1$$

$$C_1 = -12 U h_m$$

$$\frac{\partial P}{\partial x} = \frac{12 n U}{h^3} [h - h_m]$$

Friction coefficient is of order 10^{-2}
 $\rightarrow (0.02)$

Force (load)
friction
flow

Flow of oil between plates along their length will carry heat generated because of friction.

Depending on flow output flow input needs to be controlled.

$$u = \frac{1}{2n} \frac{\partial P}{\partial x} y(y-h) + \frac{(h-y)}{n} u_b + \frac{y}{h} u_a$$

$$w = \frac{1}{2n} \frac{\partial P}{\partial x} y(y-h) + \frac{1}{2n} \frac{\partial P}{\partial z} y(y-h)$$

$(w_a = w_b = 0)$

$$q_x = \int_0^h u dy, q_z = \int_0^h w dy.$$

$$q_x \cdot q_z = - \frac{h^3}{12n} \frac{\partial P}{\partial x} + \frac{h}{2} (u_a + u_b)$$

$$q_z = - \frac{h^3}{12n} \frac{\partial P}{\partial z}$$

$$Q_z = \int_0^B q_z dx.$$

Shear stress

$$\tau_n = \eta \frac{\partial u}{\partial y}, \tau_z = \eta \frac{\partial w}{\partial y}$$

$$\tau_n = \frac{1}{2} \frac{dP}{dx} (2y-h) + \frac{\eta}{h} (u_a - u_b)$$

$$\tau_z = \frac{1}{2} \frac{\partial P}{\partial z} (2y-h)$$

Force

z -direction friction is not does not come into picture as the upper surface is moving only in x -direction.

Friction Force

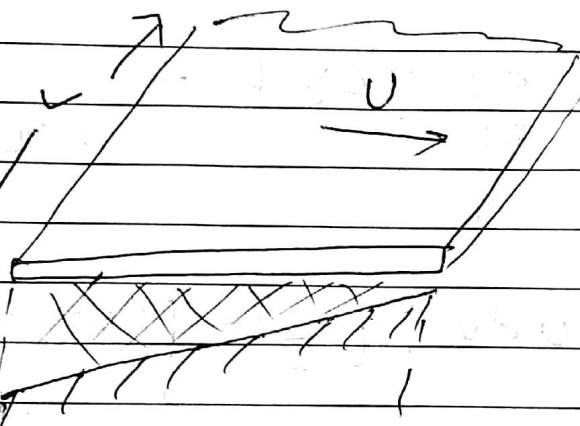
$$F = \int \int T_x dA (dx dz)$$

$$T_{x1}|_{y=0} \text{ and } T_{x2}|_{y=h}$$

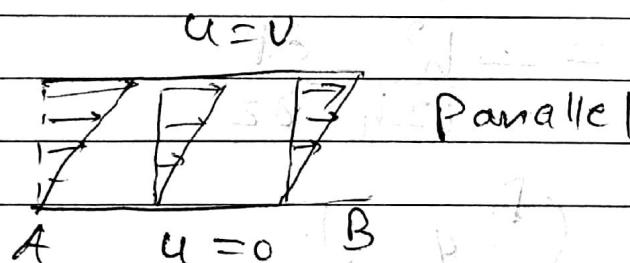
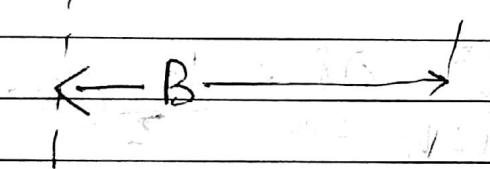
$$F_x|_{y=0,h} = \int_0^z \int_0^x \left[\mp \frac{h}{2} \frac{\partial P}{\partial x} + \mu (u_a - u_b) \right] dx dz$$

Slider

Bearing



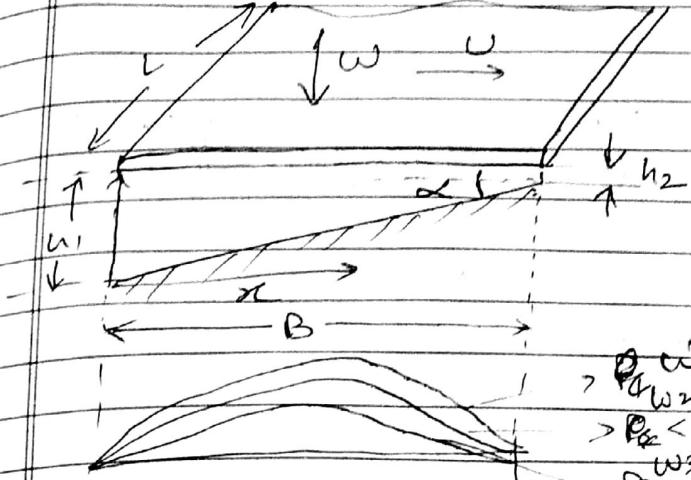
Converging, thus pressure is developed.



Profile
Pressure will be linear in-between two ends where

$$\frac{dP}{dx} = 0$$

Slider Bearing



$L \rightarrow \infty$
 \therefore One-dimensional
 Applied Load ^{Analy}
 sis.
 Controls the
 pressure distri-
 bution and gap
 between the
 plates.

$w_1 > p_w_1$
 $w_2 < p_w_1$
 $w_3 < p_w_2$ doing reverse
 analysis where we assume ~~some~~ certain
 geometry (gap) and find out the
 load carrying capacity.

$$\frac{\partial P}{\partial x} = \frac{6nUV}{h^3} (h - hm)$$

$$\tan \alpha = \frac{h_1 - h_2}{B} \quad \alpha = \frac{h_1 - h_2}{B} \quad (\alpha \text{ is small})$$

$$h = h_1 - \alpha x$$

$$\int dP = \int \frac{6nUV}{(h_1 - \alpha x)^3} (h_1 - \alpha x - hm) dx$$

$$P = 6nUV \left[\frac{1}{\alpha(h_1 - \alpha x)} - \frac{hm}{2\alpha(h_1 - \alpha x)^2} \right] + C_1$$

$$P = 0 \text{ at } x=0 \text{ and } B$$

$$0 = 6nUV \left[\frac{1}{\alpha h_1} - \frac{hm}{2\alpha h_1} \right] + C_1$$

$$0 = 6nUV \left[\frac{1}{\alpha(h_1 - \alpha B)} - \frac{hm}{2\alpha(h_1 - \alpha B)} \right] + C_1$$

$$\Rightarrow \frac{hm}{n+1} = C_1 = -\frac{6nUV}{\alpha h_2} \left(\frac{1}{h_1} - \frac{1}{h_1 + n} \right)$$

$$n = h_1/h_2 \text{ (attitude)}$$

\therefore Pressure distribution is.

$$P = \frac{6nUB}{h_2^2} \left(\frac{1}{n^2 - 1} \right) \left[\left(\frac{h_1}{h_1 - \alpha x} - 1 \right) \left(1 - \frac{h_2}{h_1 - \alpha x} \right) \right]$$

Load carrying capacity.

$$W = L \int_0^B P dx$$

$$= \frac{6nUB^2L}{h_2^2} \frac{1}{(n-1)^2} \left(\ln(n) - \frac{2(n-1)}{n+1} \right)$$

Shear stress

$$\tau = \frac{h}{2} \frac{\partial P}{\partial x} + \frac{\mu U}{h}$$

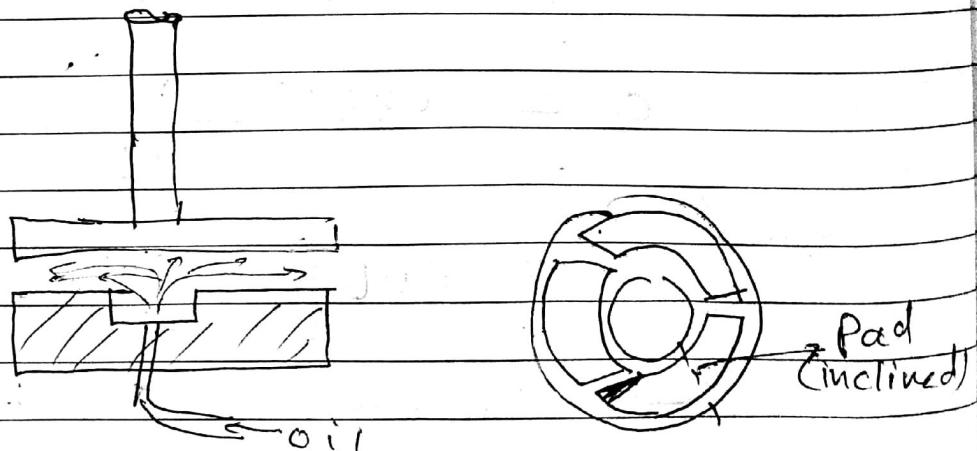
Friction Force

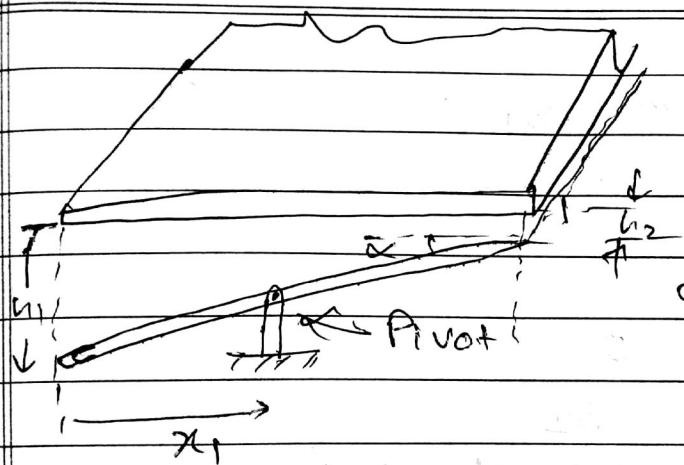
$$F = L \int_0^B \tau dx$$

$$F = \frac{nUBL}{h_2} \left(\frac{1}{n-1} \right) \left[4 \ln(n) - \frac{6(n-1)}{n+1} \right]$$

Coefficient of friction

$$\mu = \frac{F}{W} = \frac{h_2}{B} \left[\frac{2(n^2-1) \ln(n) - 3(n-1)^2}{3(n+1) \ln(n) - 6(n-1)} \right]$$





α can change if the bottom plate is pivoted.

Centre of pressure falls on pivot so as to balance the bottom plate about it.

Position of Pivot (x_1)

$$x_1 = \frac{L}{W} \int_0^B p dx$$

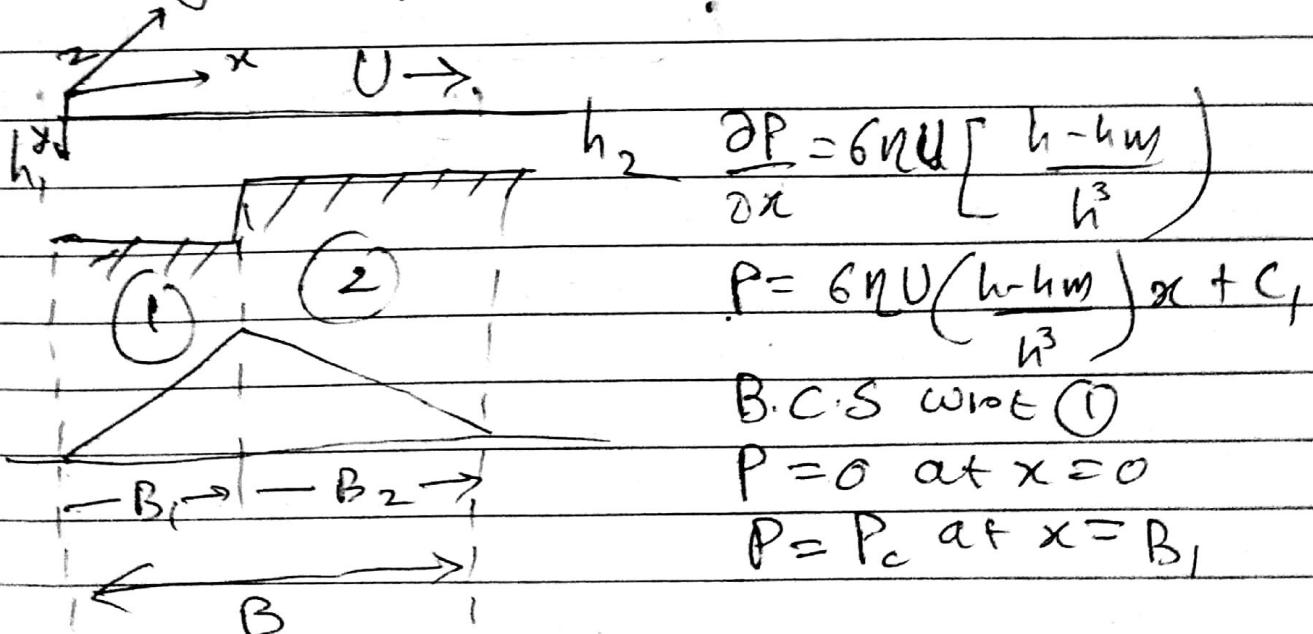
$$\frac{x_1}{B} = \frac{n(2+n) \ln(n) - (n-1)[2 \cdot 5(n-1) + 3]}{(n^2-1)(\ln(n)) - 2(n-1)^2}$$

Max Load

$$\frac{dw}{dn} = 0 \quad n = 2.18$$

$$\left[\frac{x_1}{B} = \frac{5}{9} \right] \leftarrow \text{For optimum load,}$$

Rayleigh Step Bearing



$$C=0, P_c = 6nU \left[\frac{h_1 - h_m}{h_1^3} \right] B_1 \quad \text{--- (1)}$$

B.C.S for (2)

$$P = 0 \text{ at } x = 0$$

$$P = P_c \text{ at } x = B_2$$

$$C_1 = 0, P_c = 6nU \left(\frac{h_m - h_2}{h_2^3} \right) B_2 \quad \text{--- (2)}$$

equating (1) and (2)

$$\boxed{h_m = \frac{h_1 h_2 (B_1 h_2^2 + B_2 h_1^2)}{B_1 h_2^3 + B_2 h_1^3}} \quad \text{--- (4+1)}$$

$$P|_{(1)} = \frac{6nU}{h_1^2} \left[1 - \frac{h_2 (B_1 h_2^2 + B_2 h_1^2)}{B_1 h_2^3 + B_2 h_1^3} \right] x \quad \text{from left}$$

$$P|_{(2)} = \frac{6nU}{h_2^2} \left[\frac{h_1 (B_1 h_2^2 + B_2 h_1^2)}{B_1 h_2^3 + B_2 h_1^3} - 1 \right] x \quad \text{from right}$$

Load carrying capacity

$$w = L \int_0^B P_1 dx + L \int_0^{B_2} P_2 dx.$$

$$w = \frac{3nUL B B_2 (B - B_2) (n-1)}{(B_2 n^3 + B - B_2) h_2^2}$$

$$h = \frac{h_1}{h_2}$$

Optimum Load carrying capacity

Variables: B_2, n (should be adjusted)

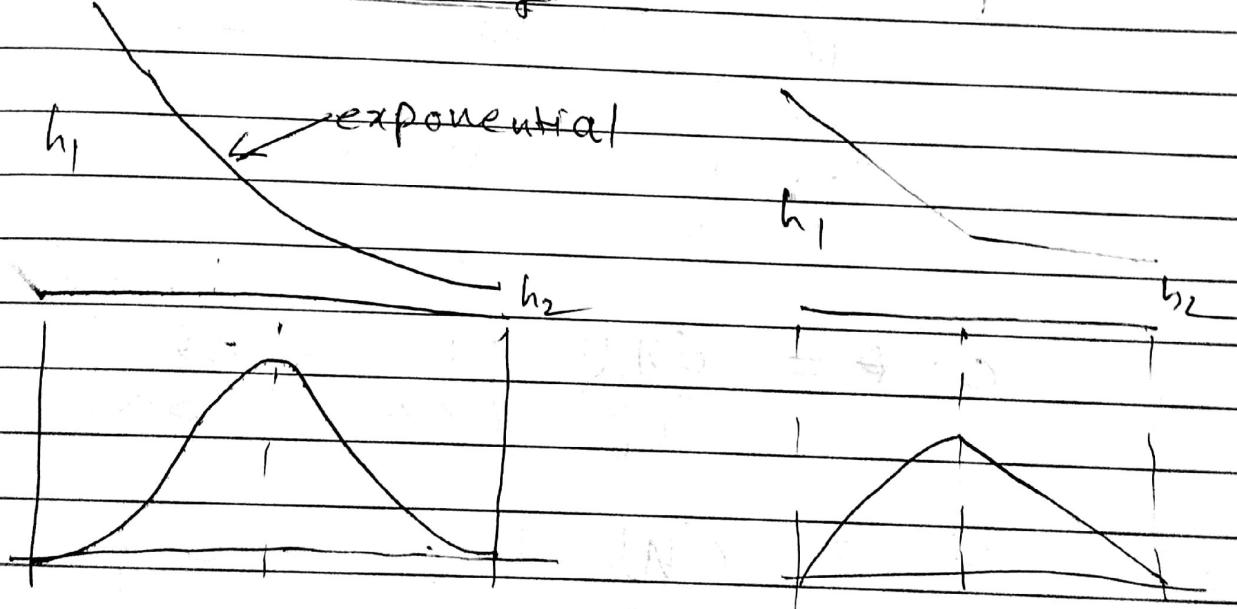
$$\frac{\partial W}{\partial B_2} = 0 \text{ and } \frac{\partial W}{\partial n} = 0.$$

$$B_2 = B \quad [n = 1866]$$

$$\left[\frac{B_1}{B_2} = 2.549 \right]$$

(HW) Find friction
and coefficient
of friction.
Centre of pressure

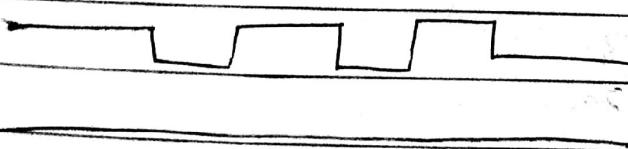
Composite Bearings



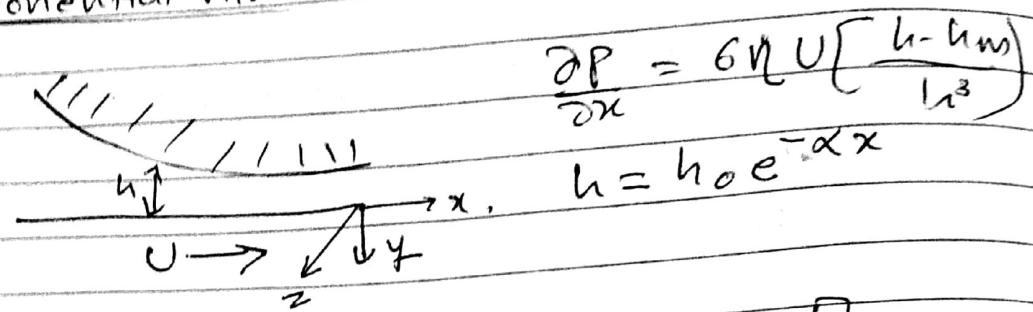
h_1, h_2 are in order

of microns

(30, 40, 50 etc.)



Exponential film



$$\frac{\partial P}{\partial x} = 6 \eta U \left[\frac{h - h_m}{h^3} \right]$$

$$\frac{\partial P}{\partial x} = 6 \eta U \left(\frac{h_0 e^{-\alpha x} - h_m}{h_0^3 e^{-3\alpha x}} \right)$$

$$P = 6 \eta U \left[\frac{h_0 e^{2\alpha x}}{2\alpha h_0^3} - \frac{h_m e^{3\alpha x}}{h_0^3 3\alpha} \right] + C_1$$

$$P = 6 \eta U \left[\frac{h_0 e^{2\alpha x}}{2\alpha h_0^2} - \frac{h_0 e^{-\alpha x_m} e^{3\alpha x}}{h_0^3 3\alpha} \right]$$

$$P = \frac{6 \eta U}{h_0^2} \left[\frac{e^{2\alpha x}}{2\alpha} - \frac{e^{-\alpha x_m} e^{3\alpha x}}{3\alpha} \right] + C_1$$

$$\text{at } x=0 \quad P=0$$

$$\text{at } x=-\infty \quad P=0$$

$$0 = \frac{6 \eta U}{h_0^2} \left[\frac{1}{2\alpha} - \frac{e^{-\alpha x_m}}{3\alpha} \right] + C_1$$

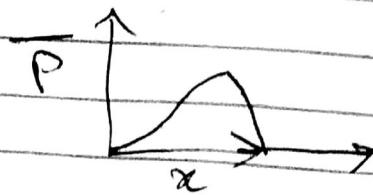
$$0 = \frac{6 \eta U}{h_0^2} [0 - 0] + C_1$$

$$C_1 = 0$$

$$e^{-\alpha x_m} = \frac{3}{2}$$

$$P = \frac{3 \eta U}{\alpha h_0^2} (e^{2\alpha x} - e^{3\alpha x})$$

$$P = \frac{P_0 h_0^2}{3 \eta U} = e^{2\alpha x} - e^{3\alpha x} \leftarrow \text{Non-dimensional pressure.}$$



Non-dimensional solution independent of viscosity, plate velocity, initial film thickness (h_0)

Load carrying capacity of exponential film:

$$W = L \int_{-\infty}^0 P dx.$$

$$W = \eta U L$$

$$2\alpha h_0^2$$

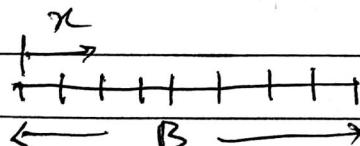
$$\boxed{\bar{W} = \frac{W h_0^2 \alpha}{\eta U L} = \frac{1}{2}} \quad \text{Non-dimensional weight carrying capacity.}$$

Numerical solution of slider bearings, (FDM)



$$\frac{\partial}{\partial x} \left[h^3 \frac{\partial P}{\partial x} \right] = 6 \eta U \frac{\partial h}{\partial x}.$$

$$\frac{\partial^2 P}{\partial x^2} = \left(\frac{6 \eta U}{h^3} - \frac{3}{h} \frac{\partial P}{\partial x} \right) \frac{\partial h}{\partial x}.$$



No. of nodes $N_x = 9$

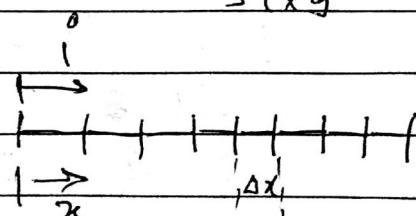
$$x = [$$

$$]_{1 \times g} \quad h = h_i - \alpha x.$$

$$h = [$$

$$]_{1 \times g}$$

(CDM)



$$\frac{\partial P}{\partial x} \Big|_i = \frac{P_{i+1} - P_{i-1}}{2 \Delta x}.$$

$$\frac{\partial^2 P}{\partial x^2} \Big|_i = \frac{P_{i+1} - P_i - \frac{\partial P}{\partial x} \Big|_{i-1/2}}{\Delta x}$$

$$= \frac{P_{i+1} - P_i}{\Delta x} - \frac{P_i - P_{i-1}}{\Delta x}$$

$$= \frac{P_{i+1} + P_{i-1} - 2P_i}{(\Delta x)^2}$$

$$\frac{P_{i+1} - 2P_i + P_{i-1}}{\Delta x^2} = \left(\frac{6n_U}{h_i^3} - \frac{3}{h_i} \frac{(P_{i+1} - P_{i-1})}{2\Delta x} \right) \cdot \frac{(h_{i+1} - h_{i-1})}{2\Delta x}$$

$$\frac{P_{i+1} - 2P_i + P_{i-1}}{\Delta x^2}$$

$$= \frac{6n_U}{h_i^3} \left(\frac{h_{i+1} - h_{i-1}}{2\Delta x} \right) - \frac{3}{h_i} \left[\frac{h_{i+1} - h_{i-1}}{2\Delta x} \right] (P_{i+1} - P_{i-1})$$

$$\frac{2P_i}{\Delta x^2} = P_{i+1} \left[\frac{1}{\Delta x^2} + \frac{3}{h_i} \left[\frac{h_{i+1} - h_{i-1}}{2\Delta x} \right] \right]$$

$$+ P_{i-1} \left[\frac{1}{\Delta x^2} - \frac{3}{h_i} \left[\frac{h_{i+1} - h_{i-1}}{2\Delta x} \right] \right]$$

$$- \frac{6n_U}{h_i^3} \left[\frac{h_{i+1} - h_{i-1}}{2\Delta x} \right]$$

while $\text{Conv} > 10^{-5}$

For $i = 2 \dots n_x - 1$

$$E = \text{RHS} - P(i) \rightarrow P(i) = P(i) + 1.25 E$$

end

$$\text{conv} = \sum P - \sum P_{old}$$

$$\sum P_{old}$$

$$P_{old} = P$$

end

$$0.8 - 1.3$$

$< 1 \oplus$ under relaxation
 $> 1 \oplus$ over

Homework.

Take, $(B, L, U, \eta, \alpha, h_1, h_2)$ values

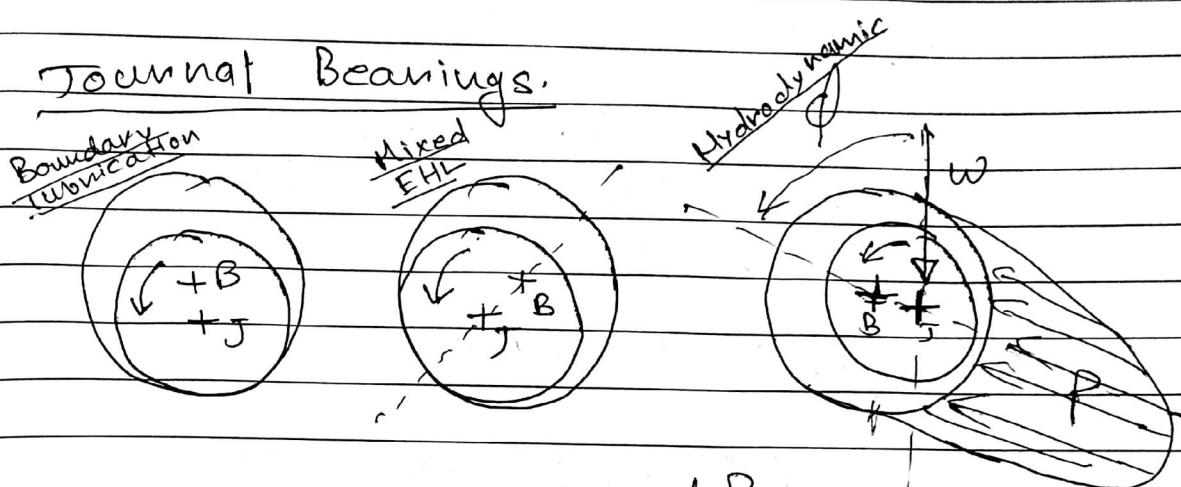
Solve using MATLAB

Plot Pressure VS B ,

Calculate $w = L \int_0^B p dx$.

Plot w vs $n (h_1/h_2)$
(1 to 2)

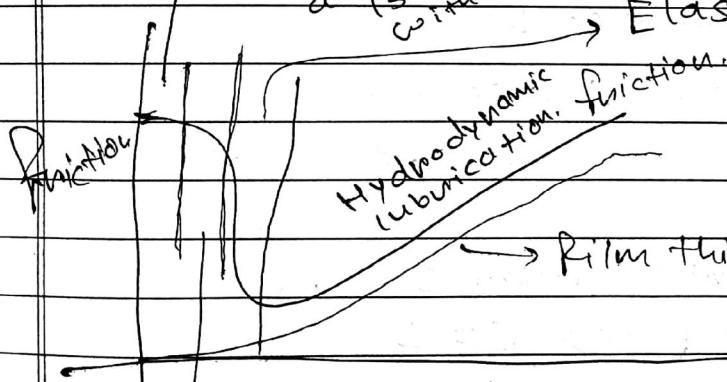
Journal Bearings.



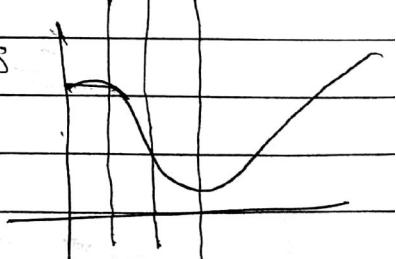
Ward P are not along same line still equilibrium is achieved because

Boundary lubrication gap is less than a nanometer or friction acting on the surface of bearing acts in contact with bearing.

Elastohydrodynamic lubrication.

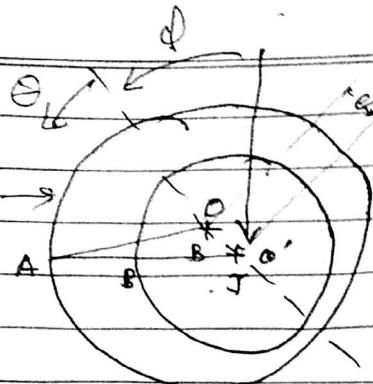


Swibick Curve.



Mixed lubrication

Contact will be there as well as off fluid film
in microns)



Ininitely long

$$\frac{\partial}{\partial x} \left(h^3 \frac{\partial P}{\partial x} \right) = 6 \eta U \frac{dh}{dx}$$

$$x = R\theta \quad dx = R d\theta$$

$$\frac{\partial}{\partial \theta} \left(h^3 \frac{\partial P}{\partial \theta} \right) = 6 \eta U R \frac{dh}{d\theta}$$

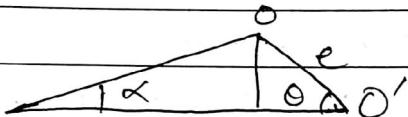
$$U = \omega R$$

$$h(\theta) = AB = A\theta' - R$$

$$A\theta' = A\theta \cos \alpha + e \cos \theta$$

$$= R + c + e \cos \theta$$

\downarrow bearing clearance.



$$h = c + e \cos \theta$$

$$= C(1 + e \cos \theta)$$

\downarrow eccentricity ratio.

$C \rightarrow$ varies from 0 to 1

when $e = 1 \Rightarrow C = e$

$C \rightarrow$ eccentricity when bearing

Journal is in contact

with bearing

at $h_m \frac{\partial P}{\partial \theta} = 0$

$$\frac{\partial P}{\partial \theta} = 6 \eta U R \frac{h - h_m}{h^3}$$

$$P = \frac{6\eta UR}{c^2} \left[\int_{(1+\epsilon \cos \alpha)}^{\infty} \frac{d\alpha}{\sqrt{1 + \epsilon^2 \cos^2 \alpha}} - h_m \int_{I_2}^{I_3} \frac{d\alpha}{\sqrt{(1+\epsilon \cos \alpha)^2}} \right]$$

I_2 I_3

$$1 + \epsilon^2 \cos^2 \alpha + 2\epsilon \cos \alpha.$$

Sommerfeld Substitution for I_2, I_3

$$\cos \gamma = \frac{\epsilon + \cos \alpha}{1 + \epsilon \cos \alpha}$$

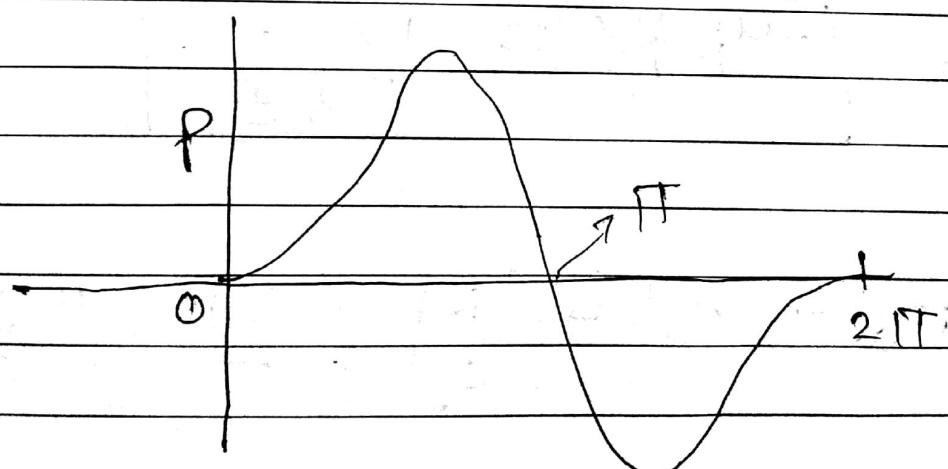
$$\sin \alpha = \frac{(1 - \epsilon^2)^{1/2} \sin \gamma}{1 + \epsilon \cos \alpha}$$

$$\cos \alpha = \frac{\cos \gamma - \epsilon}{1 - \epsilon \cos \gamma} \quad d\alpha = \frac{(1 - \epsilon^2)^{1/2}}{1 - \epsilon \cos \gamma} d\gamma$$

$$P = 0 \text{ at } \alpha = 0 \quad C_1 = 0$$

$$P = 0 \text{ at } \alpha = 2\pi \quad h_m = \frac{2C(1-\epsilon^2)}{2+\epsilon^2}$$

$$P = \frac{6\eta U R \epsilon}{c^2} \frac{(2 + \epsilon \cos \alpha) \sin \alpha}{(2 + \epsilon^2)(1 + \epsilon \cos \alpha)}$$



$\phi \rightarrow$ Attitude Angle.

CLASSMATE

Date _____
Page _____

$$w_n = w \cos \phi = -L \int_0^{2\pi} P \cos \theta R d\theta,$$

$$\omega_\phi = \omega \sin \phi = L \int_0^{2\pi} P \sin \theta R d\theta,$$

$$\omega_r = w \cos \phi = 0 \Rightarrow \phi = \frac{\pi}{2}.$$

~~w_r~~

$$\omega_\phi = \frac{12n\pi UL C (R/c)^2}{(2+\epsilon^2)(1-\epsilon^2)^{1/2}} = \omega$$

shear stress

$$\tau_j = \frac{nU}{h} + \frac{h}{2R} \frac{dP}{d\theta}.$$

$$F_j = \int_0^{2\pi} \tau_j L R d\theta = \cancel{2\pi U L R} \cancel{\frac{h}{2R}}$$

~~1/2~~ $\int_0^{2\pi} \tau_j^2 d\theta$

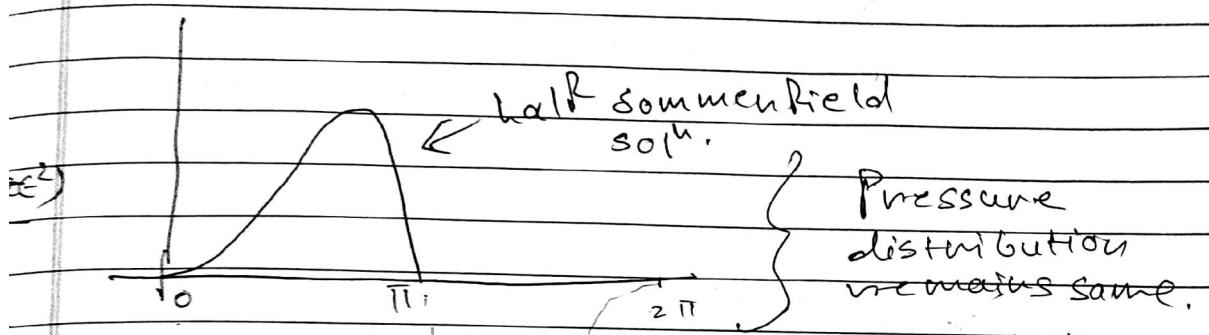
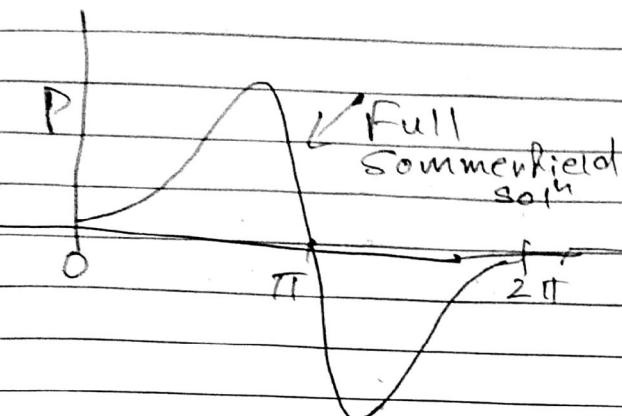
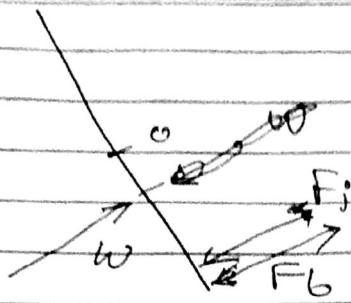
$$F_j = n U L \left(\frac{R}{c}\right) \frac{4\pi(1+2\epsilon^2)}{(2+\epsilon^2)(1-\epsilon^2)^{1/2}}$$

$$\mu_j = \frac{F_j}{\omega} = \frac{C}{R} \left(\frac{1+2\epsilon^2}{3\epsilon} \right)$$

$$\boxed{\mu_j \left(\frac{R}{c}\right) = \frac{1+2\epsilon^2}{3\epsilon}}$$

$$F_b = n U L \left(\frac{R}{c}\right) \left(\frac{4\pi(1-\epsilon^2)}{(2+\epsilon^2)} \right)$$

$$RF_j = RF_b + w_e$$



$$\omega_r - \omega \cos \phi = \int_0^{\pi} P \cos \theta d\theta$$

$$\omega \phi = \omega \sin \phi - \int_0^{\pi} P \sin \theta d\theta$$

$$\omega_r = 12 \eta U L \left(\frac{R}{c} \right)^2 \frac{c^2}{(2+c^2)(1-\epsilon^2)}$$

$$\omega \phi = 6 \eta U L \left(\frac{R}{c} \right) \frac{\pi \epsilon}{(2+\epsilon^2)(1-\epsilon^2)^{1/2}}$$

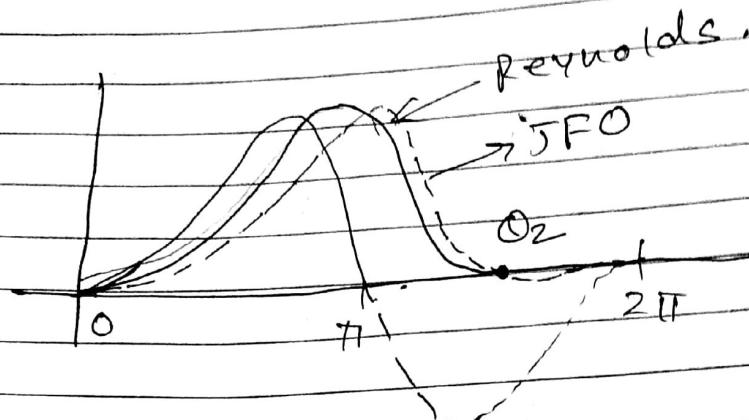
$$\omega = \sqrt{\omega_r^2 + \omega \phi^2}$$

$$= 6 \eta U L \left(\frac{R}{c} \right)^2 \frac{\epsilon (c^2 - \epsilon^2 (\pi^2 - 4))^{1/2}}{(2+\epsilon^2)(1-\epsilon^2)}$$

$$\phi = \tan^{-1} \left(\frac{\omega \phi}{\omega_r} \right) = \tan^{-1} \left(\frac{\pi}{2 \epsilon} \sqrt{1-\epsilon^2} \right)$$

$$\text{Sommerfeld No. } S = \frac{n N}{P} \left(\frac{R}{c} \right)^2 \quad | \quad \begin{aligned} N &= n p s \\ P &= \frac{w}{2 \pi \epsilon} \end{aligned}$$

$$S = \frac{(2 + \epsilon^2) (1 - C^2)^{1/2}}{2 \pi^2 \epsilon}$$



$C_1, h_m, O_2 \leftarrow$ constants to find,

$$P = 0 \text{ at } \theta = 0$$

$$\begin{aligned} P = 0 \text{ at } \theta = \theta_2 & \quad | \text{ Reynolds} \\ \frac{\partial P}{\partial \theta} = 0 \text{ at } \theta = \theta_2 & \quad | \text{ cavitation,} \end{aligned}$$

Perrinoff Solution (lightly load)

$$\epsilon \ll \frac{h}{c}$$

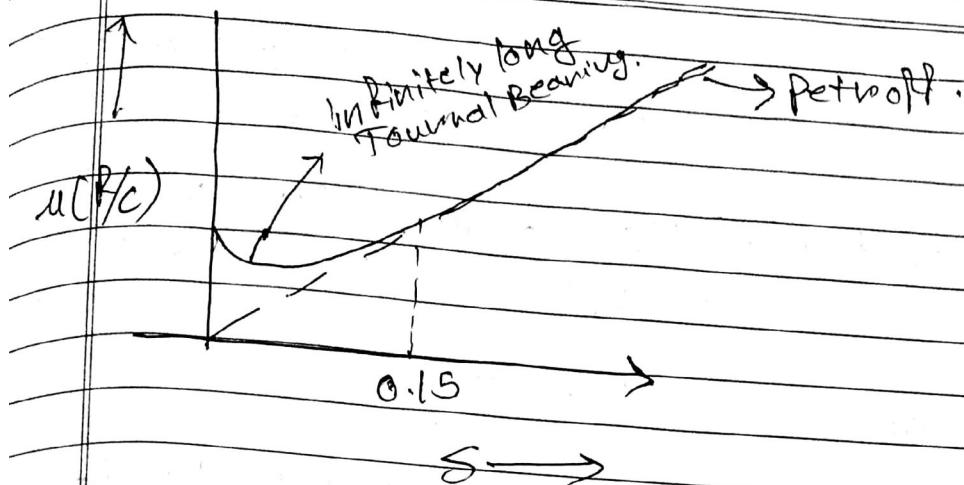
$$\bar{C} = \frac{nu}{h} \quad F = \bar{C} A.$$

$$U = 2\pi R N, \quad A = 2\pi R L, \quad N_{\text{min}}, \quad \text{PPS}$$

$$F = \frac{4\pi^2 \rho R^2 N L}{c}$$

$$u = \frac{F}{\omega} = 2\pi^2 \frac{\rho R N}{P} \left(\frac{R}{c} \right) \quad P = \frac{\omega}{2\pi R}$$

$$u \left(\frac{R}{c} \right) = 2\pi^2 s$$

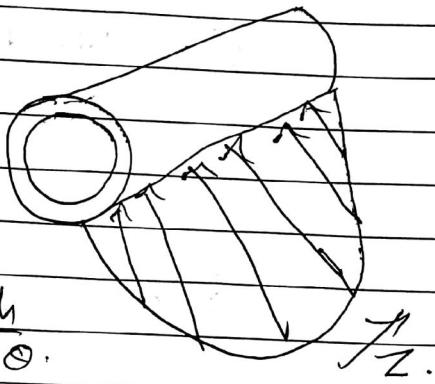


Ininitely short Journal Bearings
(Narrow Bearings)

$$\frac{\partial P}{\partial z} \left(h^3 \frac{\partial P}{\partial z} \right) = 6 \eta U \frac{\partial h}{\partial x}$$

~~Ansatz~~

$$x=R\theta \quad dx=Rd\theta$$



~~$$\frac{\partial}{\partial z} \left(h^3 \frac{\partial P}{\partial z} \right) = \frac{6 \eta U}{R} \frac{\partial h}{\partial \theta}$$~~

$$P = \frac{6 \eta U}{R h^3} \frac{\partial h}{\partial \theta} \frac{z^2}{2} + C_1 z + C_2$$

B.C.S $P=0$ at $z=\pm \frac{L}{2}$

$$P = \frac{3 \eta U}{R c^2} \left[\frac{L^2 - z^2}{4} \right] \frac{\sin \theta}{(1 + c \cos \theta)^3}$$

$$w_r = -2 \int_0^{2\pi} \int_{-\frac{L}{2}}^{\frac{L}{2}} P \cos \theta R d\theta dz$$

$$w_\phi = 2 \int_0^{2\pi} \int_{-\frac{L}{2}}^{\frac{L}{2}} P \sin \theta R d\theta dz$$

$$w_r = \frac{\eta U L^3}{C^2} \frac{c^2}{(1 - c^2)^2} \quad w_\phi = \frac{\eta U L^3 \pi c}{4 C^2 (1 - c^2)^{3/2}}$$

$$\omega = \sqrt{\omega_h^2 + \omega_p^2} = \frac{\pi V L^3 c}{4 C^2 (1 - \epsilon^2)^2} \left[\pi^2 (1 - \epsilon^2) + 16 \epsilon^2 \right]$$

$$\phi = \tan^{-1} \left(\frac{\omega_p}{\omega_h} \right) = \tan^{-1} \left(\frac{\pi (1 - \epsilon^2)}{4 C} \right)$$

Shear stress $\tau = \frac{\eta U}{h}$

Friction Force $F = \int_0^{2\pi} \eta \cdot \frac{U}{h} L R d\alpha$
 $= \eta U L R \left(\frac{2\pi}{(1 - \epsilon^2)^{1/2}} \right)$

Coefficient of friction $\mu = \frac{F}{w}$

$$\mu(R/c) = \frac{2\pi^2 S}{(1 - \epsilon^2)^{1/2}}$$

Flow $dq_z = \frac{h^3}{12n} \frac{\partial P}{\partial z}$

$$Q_z = -2 \int_0^{2\pi} \frac{R h^3}{12n} \frac{\partial P}{\partial z} \Big|_{z=4/2} d\alpha.$$

$$Q_z = \epsilon U L C$$

$L/D > 1 \rightarrow$ infinitely long bearing.
 $L/D < 1 \rightarrow$ narrow bearing

Flute Bearings (Numerical Soln, FDM)

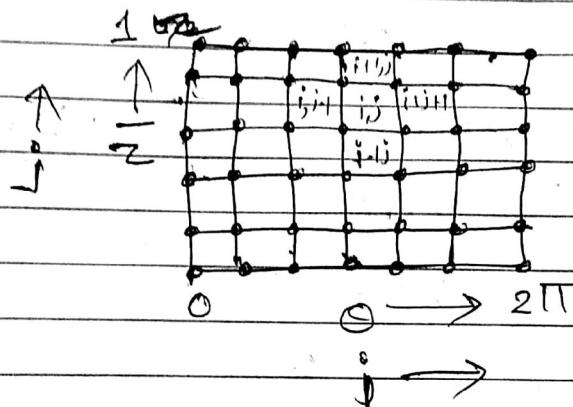
$$\frac{\partial}{\partial x} \left(\frac{h^3}{12n} \frac{\partial P}{\partial x} \right) + \frac{\partial}{\partial z} \left(\frac{h^3}{12n} \frac{\partial P}{\partial z} \right) = \frac{U_a + U_b}{z} \frac{\partial h}{\partial x}$$

$$\Theta = \frac{x}{R}, \bar{z} = \frac{2z}{L}, \bar{h} = \frac{h}{2}, U = U_a + U_b$$

$$\frac{\partial}{\partial \Theta} \left(\bar{h}^3 \frac{\partial \bar{P}}{\partial \Theta} \right) + \left(\frac{D}{L} \right)^2 \frac{\partial}{\partial z} \left(\bar{h}^3 \frac{\partial \bar{P}}{\partial z} \right) = \frac{\partial \bar{h}}{\partial \Theta}$$

$$\bar{P} = \frac{P_c^2}{6nUR} \quad D = 2R \quad \frac{L}{D} = ? \quad \epsilon = ?$$

$$\frac{\partial^2 \bar{P}}{\partial \Theta^2} + \left(\frac{D}{L} \right)^2 \frac{\partial^2 \bar{P}}{\partial z^2} + \frac{3}{h} \frac{\partial \bar{P}}{\partial \Theta} \frac{\partial \bar{h}}{\partial \Theta} = \frac{1}{\bar{h}^3} \frac{\partial \bar{h}}{\partial \Theta}$$



$$\frac{\partial P}{\partial \Theta}|_{ij} = \frac{P_{i+1,j} - P_{i-1,j}}{2 \Delta \Theta}$$

$$\frac{\partial^2 P}{\partial \Theta^2}|_{ij} = \frac{\partial P}{\partial \Theta}|_{i+\frac{1}{2},j} - \frac{\partial P}{\partial \Theta}|_{i-\frac{1}{2},j}$$

$$= \frac{P_{i+1,j} - P_{i-1,j} - P_{i,j} + P_{i-1,j}}{2 \Delta \Theta}$$

$$= \frac{P_{i+1,j} + P_{i-1,j} - 2P_{i,j}}{2 \Delta \Theta^2}$$

$$h = \epsilon [1 + \cos x]$$

$$\frac{\partial P}{\partial z^2} |_{(i,j)} = \frac{P_{i,j+1} - P_{i,j-1}}{2\Delta z^2}$$

$$\frac{\partial^2 P}{\partial z^2} |_{(i,j)} = \frac{P_{i,j+1} + P_{i,j-1} - 2P_{i,j}}{\Delta z^2}$$

$$\epsilon = ?, L/D = ? \\ n_n = 10, n_2 = 12, P = C \cdot f_{max, min}$$

$$P_{old} = P$$

Δz

$$\frac{P(i,j) - 2P(i,j) + P(i,j)}{\Delta z^2}$$

$$+ \left(\frac{D}{L}\right)^2 \frac{P(i,j+1) + P(i,j-1) - 2P(i,j)}{\Delta z^2}$$

$$+ \frac{3}{h_i} \frac{P(i+1,j) - P(i-1,j)}{\Delta z} \cdot \frac{(h_{i+1} - h_{i-1})}{\Delta z}$$

$$= \frac{1}{[h_i]^3} \left[\frac{h_{i+1} - h_{i-1}}{\Delta z} \right]$$

load carrying capacity

$$W = \iint_0^{2\pi} p d\theta dz$$

$$= 2 \int_0^{10\pi} \int_0^{C^2} \overline{P} \cdot h_{UR} R d\theta dz L$$

$$= \frac{6h_{UR} L}{C^2} \int_0^1 \int_0^{2\pi} \overline{P} d\theta dz$$

$$\bar{W} = \frac{WC^2}{\sigma n VR^2 L} = \int_0^1 \int_0^{2\pi} \bar{P} d\alpha dz$$

Friction

$$F = \int_0^L \int_0^{2\pi R} \bar{c}_x dx dz$$

$$= \int_0^L \int_0^{2\pi R} \left(\frac{h}{z} \frac{\partial P}{\partial x} + \frac{u}{h} u \right) dx dz$$

$$= \frac{RLNU}{C} \int_0^1 \int_0^{2\pi} \left[3 \frac{h}{z} \frac{\partial \bar{P}}{\partial \alpha} + \frac{L}{h} \right] d\alpha dz$$

$$\bar{F} = \frac{FC}{nVRL} = \int_0^1 \int_0^{2\pi} \left[3 \frac{h}{z} \frac{\partial \bar{P}}{\partial \alpha} + \frac{L}{h} \right] d\alpha dz$$

$$\mu = \frac{F}{w} = \frac{\bar{F} n V R L C^2}{C w g n V R^2 L}$$

$$\mu(R/c) = \frac{\bar{F}}{g \bar{w}}$$

End Flow

$$d\bar{Q}_z = -h \left. \frac{1}{12n} \frac{\partial P}{\partial z} \right|_{z=L} dz$$

$$\bar{Q}_z = \int_0^{2\pi R} \left. -\frac{h^3}{12n} \frac{\partial P}{\partial z} \right|_{z=L} dx$$

$$\bar{Q}_z = \frac{\bar{Q}_z L}{V C R^2} = \int_0^{2\pi} \left. -\frac{h^3}{12n} \frac{\partial \bar{P}}{\partial z} \right|_{z=L} d\alpha$$

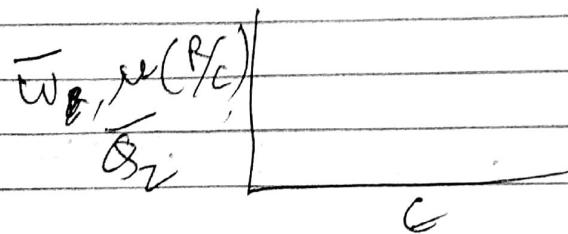
HW

~~Case~~

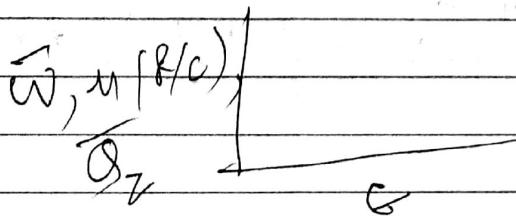
$$\epsilon/\theta = 1 \quad \epsilon = 0.4$$

Case 2: Solve it & plot P
Ans:

$$\epsilon = 0.2, 0.4, 0.6, 0.8$$

W[±]

Case II: $\epsilon = 0.4, \epsilon/\theta = 0.5, 1, 1.5, 2$



$$P(i,i) = \left[\frac{+2}{\Delta \theta^2} + \frac{2}{\Delta z^2} \left(\frac{\theta}{L} \right)^2 \right]$$

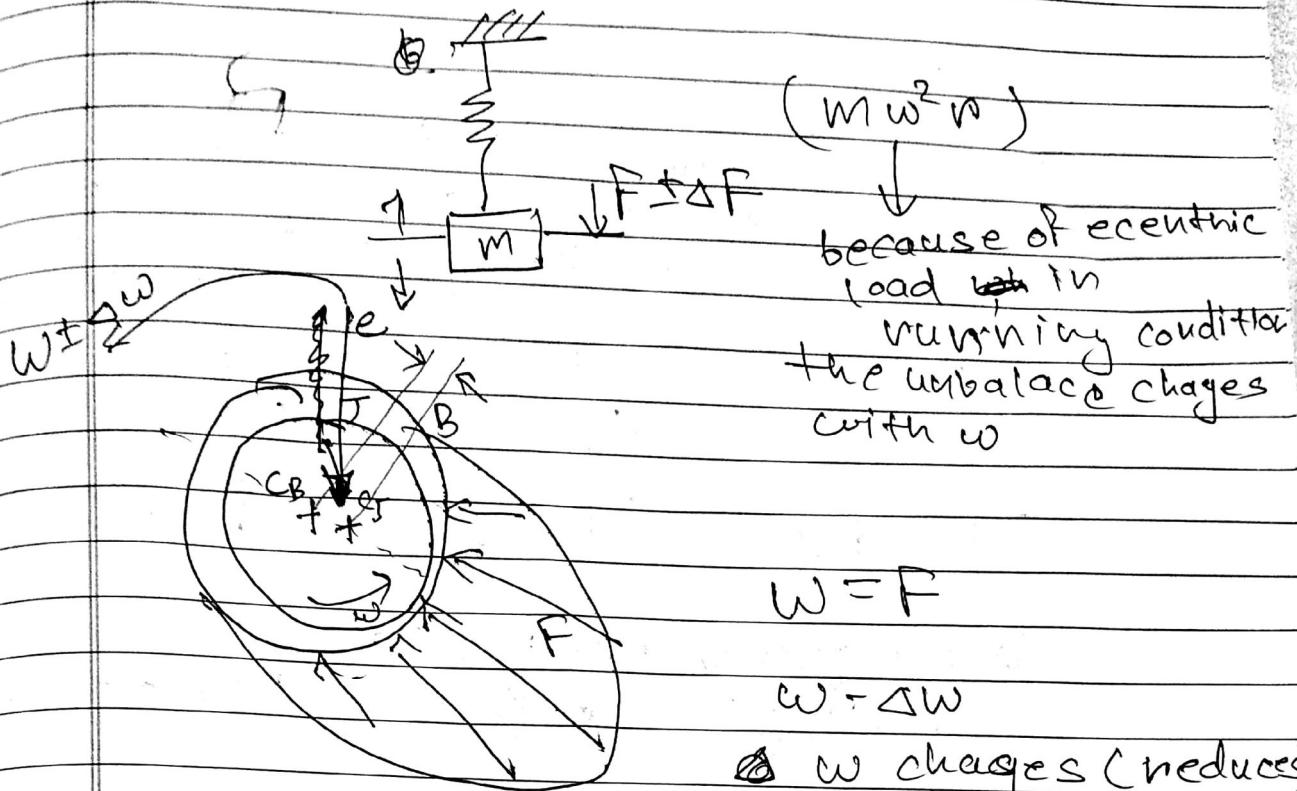
$$= + \left(\frac{P(i+1,j) + P(i-1,j)}{\Delta \theta^2} \right)$$

$$+ \left(\frac{\theta}{L} \right)^2 \left(\frac{P(i,j+1) + P(i,j-1)}{\Delta z^2} \right)$$

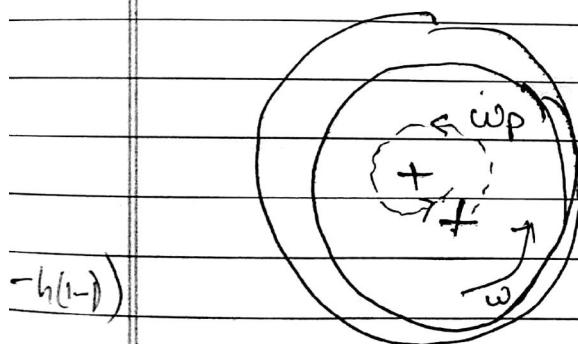
$$+ \frac{3}{h(i)} \left(\frac{P(i+1,j) - P(i-1,j)}{\Delta \theta^2} \right) (h(i+1) - h(i-1))$$

$$- \frac{1}{(h(i))^3} \left(\frac{h(i+1) - h(i-1)}{\Delta z} \right)$$

Hydrodynamic Instability



In mean time centre of journal revolves around bearing centre, because of this unbalance.



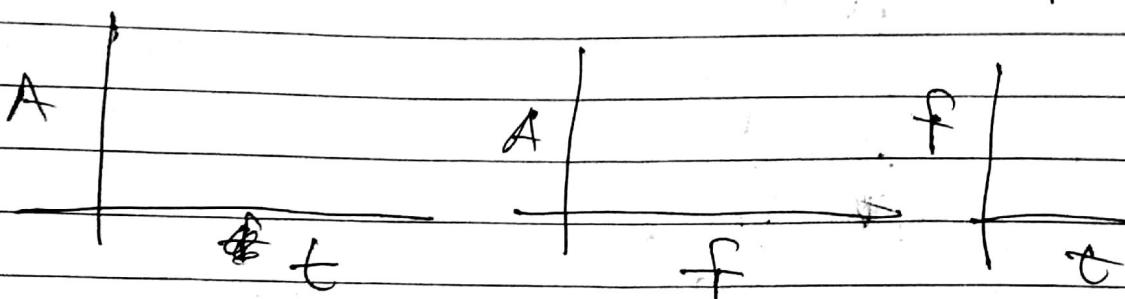
$wp \rightarrow$ whirling frequency
 $\omega \rightarrow$ spinning frequency

wp depends of ω , and thus it is inherently present in rotating system.

$$\frac{wp}{\omega} \approx \frac{1}{2}$$

$$\begin{array}{c} x(t) \\ \dot{x}(t) \\ \ddot{x}(t) \end{array}$$

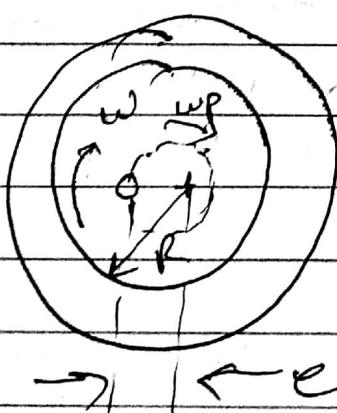
Measuring $\ddot{x}(t)$ involves less noise as to obtain other two ($\dot{x}(t), x(t)$) we need to integrate (less noise amplification).



• Frequency varies with time during run up and run down.

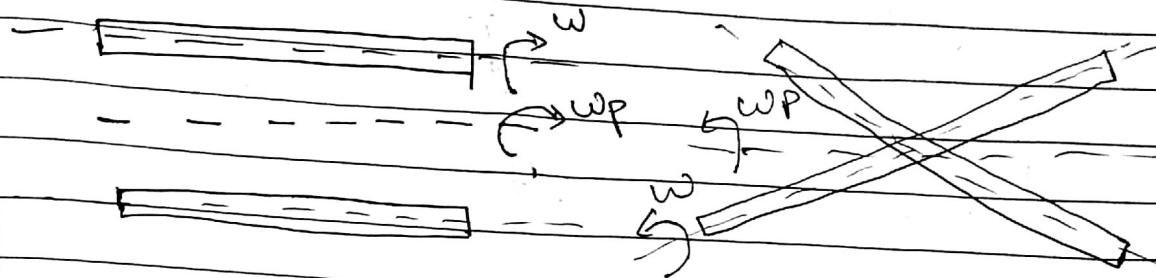
Cause of Instability

Theoretical prediction



$m w^2 e$

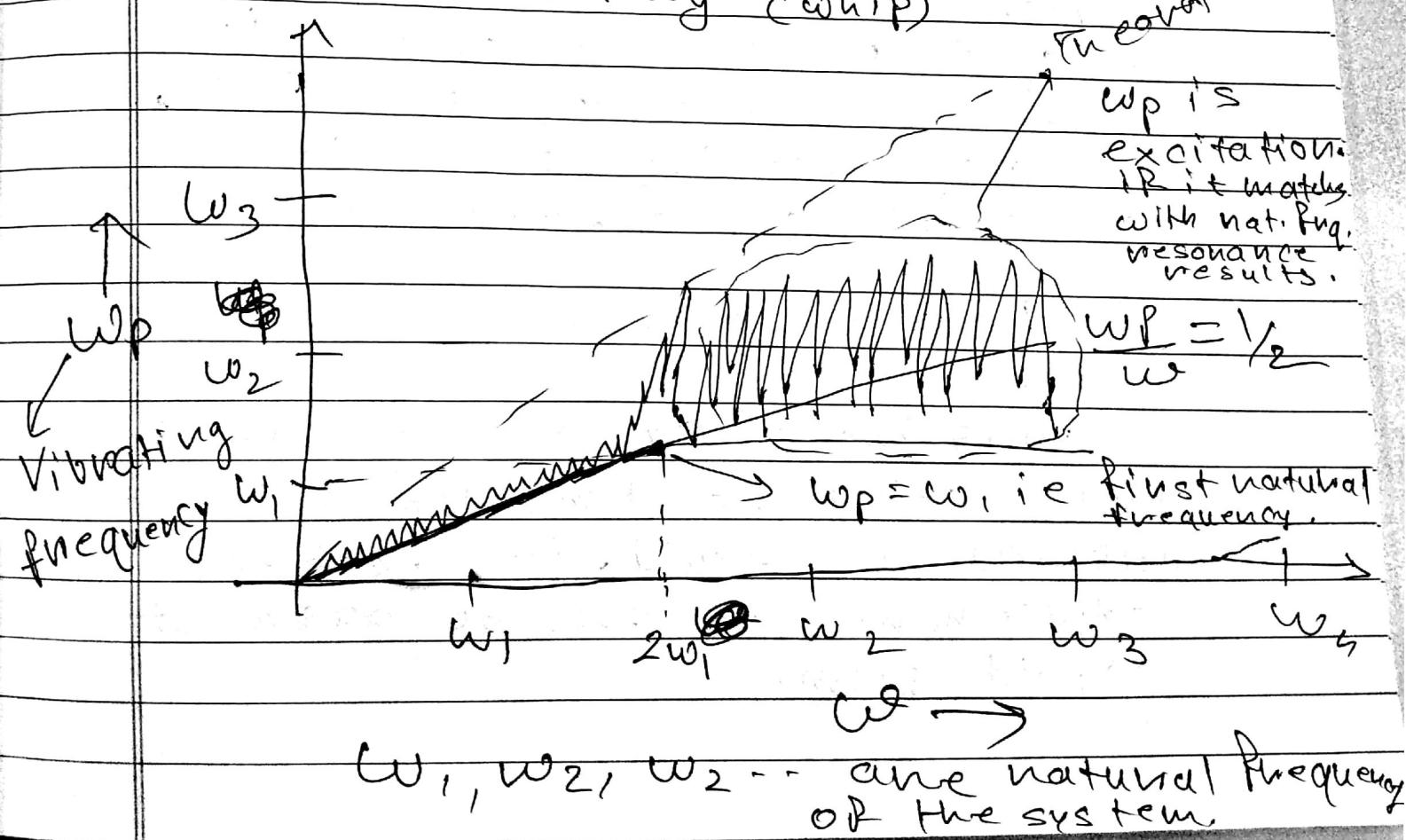
Type of Instability



Translatory whint
 γ_D is small

Conical whin
 γ_D is large.

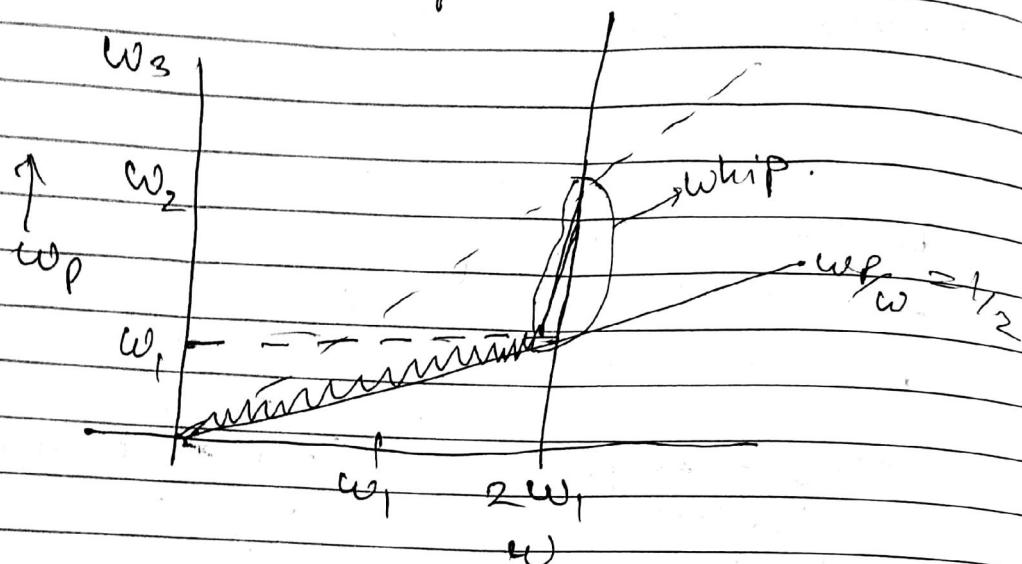
1. Subsynchronous (whint)
2. Resonating (whip)



Practically, after ω_p reaches $2\omega_1$, no further increase in ω is possible.

\therefore journal bearing can reach maximum up to $2\omega_1$ speed.

$2\omega_1$ is called as stability threshold speed.



$\rightarrow 0.83 \times (2\omega_1)$ design

To increase threshold (stability) speed for a given system i.e. ω_1, ω_2 ... nat freq. are fixed, slope of ω_p vs ω line should be decreased.

How, to do this?

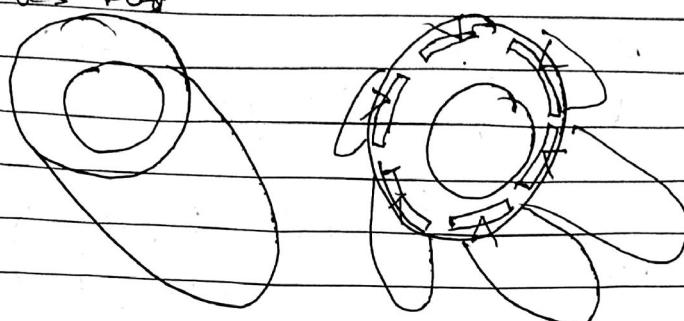
This instability occurs due to the settling time because of lubricating fluid. i.e. pressure adjustment is slow because of liquid film.

$$x = x e^{iwf} \quad \frac{x}{F} = \frac{1}{K} e^{iwf}$$

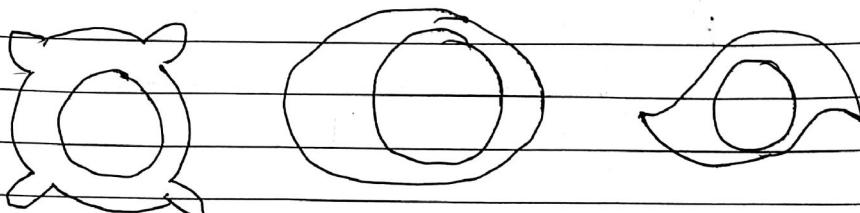
Date _____
Page _____

To tackle this, pivoted pads are introduced so that pressure gets distributed over number of small sections and thus pressure adjustment is quick in such system.

~~Plane Box~~

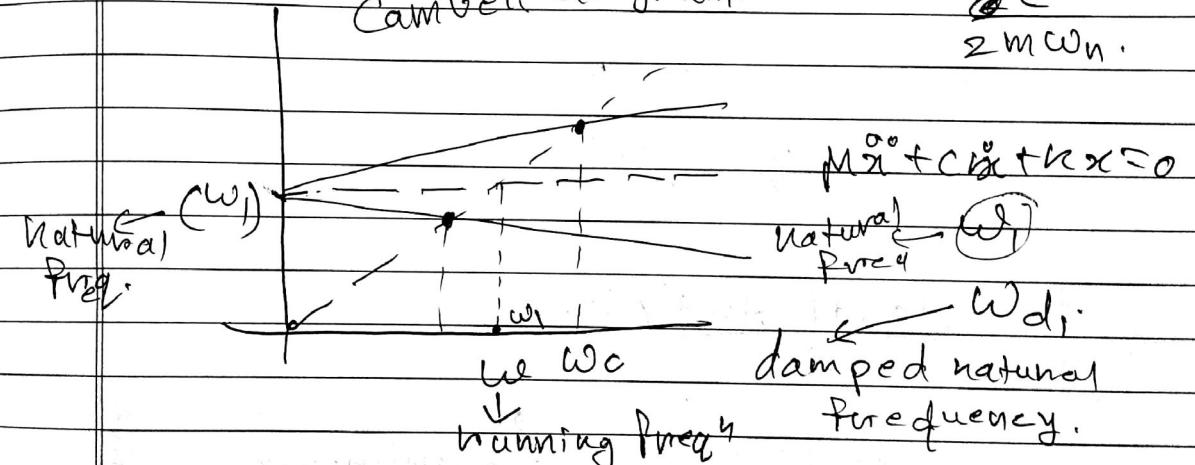


Other solutions

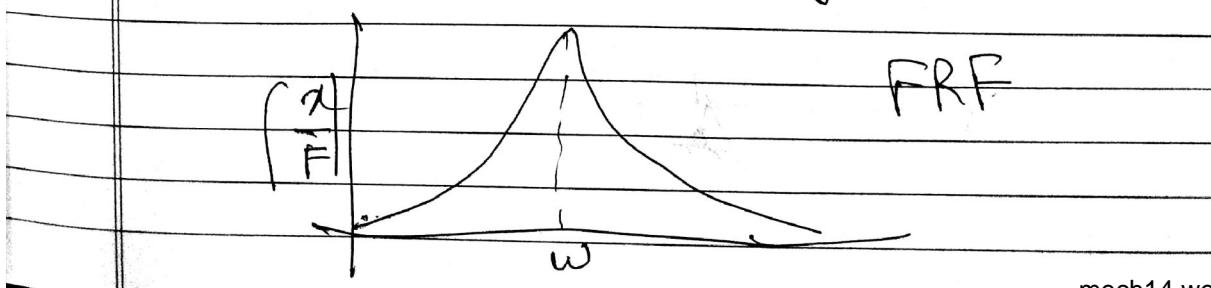


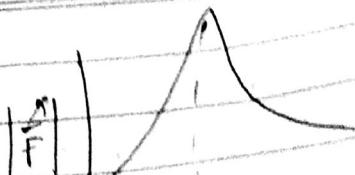
Campbell diagram.

~~at~~ ω_c
 $2m\omega_n$.



at ~~at~~ zero running speed $w_d = w_1$





Phase

$\omega \rightarrow$

$$\alpha = \frac{Y_K}{F}$$

$$\cos(\omega t + \beta) + \phi$$

$$\phi = \tan^{-1} \left(\frac{2Y_K}{1 - \nu^2} \right)$$

A

180

F

phase

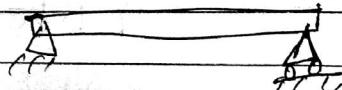
-180

Bode Plot.

Modal Analysis.

Theoretical

Experimental



$$M\ddot{x} + Kx = 0.$$

$$M\ddot{x} + Kx = 0$$

$$[M^{-1}K]$$

$$M\omega^2 + K = 0$$

$$M\omega^2 = -K.$$



eigenvalues, eigenvectors.

$$[R]_{10 \times 10}$$

$$[V]_{10 \times 10}$$

Operating deflection shape (shape for frequencies in between natural frequencies)

Operating deflection shapes are functions of all mode shapes.

$$v = R(v_1, v_2, v_3, v_4, \dots)$$

\swarrow
mode shape

corresponding

to v_1 , and so on.

$$v = [v_i] \times [w_i]$$

mode shapes ~~weightage~~