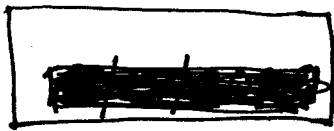
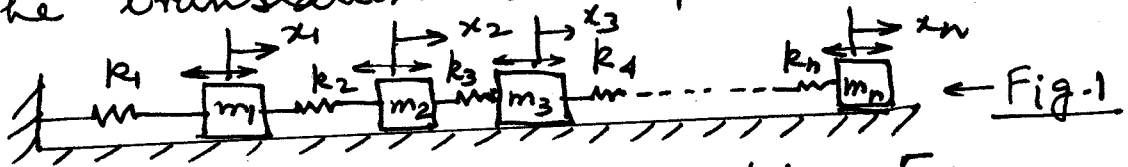


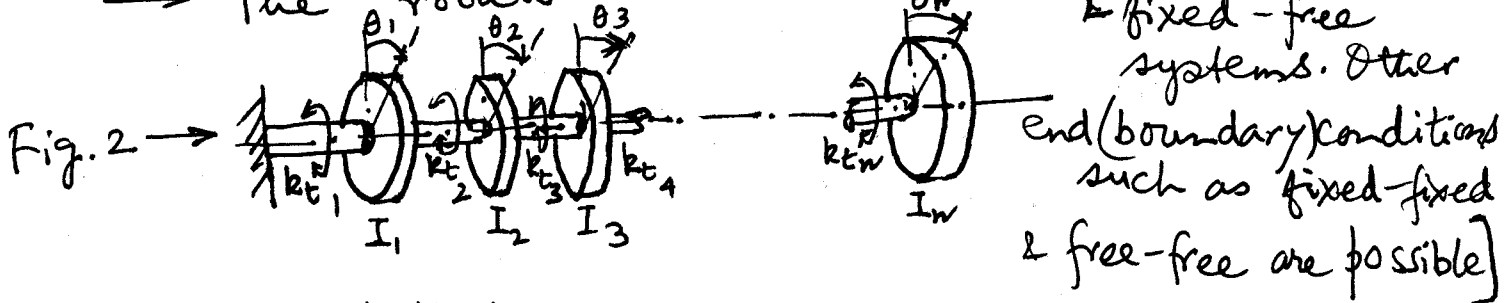
VA-6, Part 1 discrete (Page 1)
Vibration of systems having
(no. of) DOF > 2



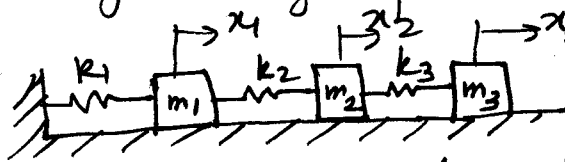
→ The translational undamped model (Free Vibration):



→ The rotational counterpart:

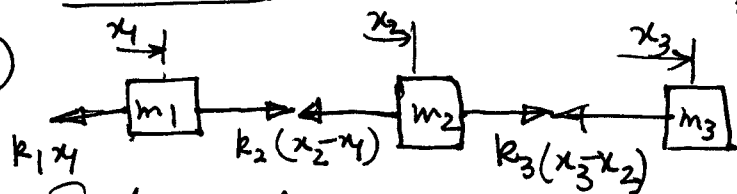


→ We shall first consider 3-DOF systems & then generalize for an n-DOF system.



For the translational system, x_1, x_2, x_3 are a set of generalized coordinates with static equilibrium positions as reference.

At time t , the relevant FBDs are:— (Friction neglected)
 (let $x_3 > x_2 > x_1$)



By Newton's 2nd law of motion,

$$m_1 \ddot{x}_1 = k_2(x_2 - x_1) - k_1 x_1 \quad \text{or,} \quad m_1 \ddot{x}_1 + (k_1 + k_2)x_1 - k_2 x_2 = 0 \quad \text{--- (1)}$$

$$m_2 \ddot{x}_2 = k_3(x_3 - x_2) - k_2(x_2 - x_1) \quad \text{or,} \quad m_2 \ddot{x}_2 - k_2 x_1 + (k_2 + k_3)x_2 - k_3 x_3 = 0 \quad \text{--- (2)}$$

$$m_3 \ddot{x}_3 = -k_3(x_3 - x_2) \quad \text{or,} \quad m_3 \ddot{x}_3 - k_3 x_2 + k_3 x_3 = 0 \quad \text{--- (3)}$$

①, ② & ③ are the required DEOM for undamped free vibrations, obtained by Newton's Method.

In matrix form, the DEOM can be written as:

$$[m] \{\ddot{x}\} + [k] \{x\} = \{0\} \quad \text{--- (4). Here } \{x\} = \begin{Bmatrix} x_1 \\ x_2 \\ x_3 \end{Bmatrix}, \{\ddot{x}\} = \begin{Bmatrix} \ddot{x}_1 \\ \ddot{x}_2 \\ \ddot{x}_3 \end{Bmatrix} \quad \text{(Continued)}$$

$$[m] = \begin{bmatrix} m_1 & 0 & 0 \\ 0 & m_2 & 0 \\ 0 & 0 & m_3 \end{bmatrix}; \quad [k] = \begin{bmatrix} (k_1+k_2) & -k_2 & 0 \\ -k_2 & (k_2+k_3) & -k_3 \\ 0 & -k_3 & k_3 \end{bmatrix} \quad \& \quad \{0\} = \begin{Bmatrix} 0 \\ 0 \\ 0 \end{Bmatrix}$$

(mass or inertia matrix) (stiffness matrix)

For the corresponding n-DOF system, the DEOM will have the same form

$$[m]\{\ddot{x}\} + [k]\{x\} = \{0\}, \quad \text{where}$$

$$\{x\} = \begin{Bmatrix} x_1 & x_2 & \dots & x_n \end{Bmatrix}^T; \quad \{\ddot{x}\} = \begin{Bmatrix} \ddot{x}_1 & \ddot{x}_2 & \dots & \ddot{x}_n \end{Bmatrix}^T$$

(displacement vector) (Acceleration vector)

$$[m] = \begin{bmatrix} m_1 & 0 & \dots & 0 \\ 0 & m_2 & \dots & 0 \\ \vdots & \vdots & \ddots & \vdots \\ 0 & 0 & \dots & m_n \end{bmatrix}; \quad [k] = \begin{bmatrix} k_{11} & k_{12} & \dots & k_{1n} \\ k_{21} & k_{22} & \dots & k_{2n} \\ \vdots & \vdots & \ddots & \vdots \\ k_{n1} & k_{n2} & \dots & k_{nn} \end{bmatrix} \quad \& \quad \{0\} = \begin{Bmatrix} 0 \\ 0 \\ \vdots \\ 0 \end{Bmatrix}$$

(n x n) (n x n)

→ For the 3-DOF system, we now derive the DEOM using Lagrange's equations.

There are 3 such equations & these are:

$$\frac{d}{dt} \left(\frac{\partial T}{\partial \dot{x}_i} \right) - \frac{\partial T}{\partial x_i} + \frac{\partial V}{\partial x_i} = 0; \quad i=1,2,3.$$

$$\text{Here } T = \frac{1}{2} m_1 \dot{x}_1^2 + \frac{1}{2} m_2 \dot{x}_2^2 + \frac{1}{2} m_3 \dot{x}_3^2$$

$$V = \frac{1}{2} k_1 x_1^2 + \frac{1}{2} k_2 (x_2 - x_1)^2 + \frac{1}{2} k_3 (x_3 - x_2)^2$$

Hence,

$$\frac{d}{dt} \left(\frac{\partial T}{\partial \dot{x}_1} \right) = m_1 \ddot{x}_1; \quad \frac{\partial T}{\partial x_1} = 0$$

$$\frac{\partial V}{\partial x_1} = k_1 x_1 + k_2 (x_2 - x_1)(-1) = (k_1 + k_2)x_1 - k_2 x_2$$

Thus, the first DEOM is:

$$m_1 \ddot{x}_1 + (k_1 + k_2)x_1 - k_2 x_2 = 0$$

These DEOM are the same as the ones obtained using Newton's method

$$\frac{d}{dt} \left(\frac{\partial T}{\partial \dot{x}_2} \right) = m_2 \ddot{x}_2; \quad \frac{\partial T}{\partial x_2} = 0$$

$$\frac{\partial V}{\partial x_2} = k_2 (x_2 - x_1) - k_3 (x_3 - x_2) = -k_2 x_1 + (k_2 + k_3)x_2 - k_3 x_3$$

& hence, the 2nd DEOM is:

$$m_2 \ddot{x}_2 - k_2 x_1 + (k_2 + k_3)x_2 - k_3 x_3 = 0$$

$$\frac{d}{dt} \left(\frac{\partial T}{\partial \dot{x}_3} \right) = m_3 \ddot{x}_3; \quad \frac{\partial T}{\partial x_3} = 0$$

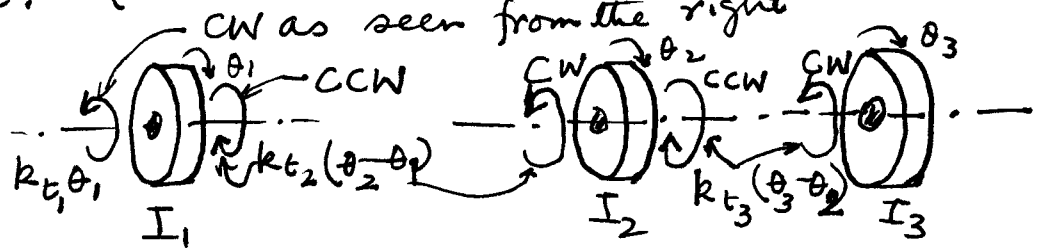
$$\frac{\partial V}{\partial x_3} = k_3 (x_3 - x_2)$$

& so, the 3rd DEOM is:

$$m_3 \ddot{x}_3 - k_3 x_2 + k_3 x_3 = 0$$

cont'd →

→ For fig. 2, if Newton-Euler (moment of momentum) ~~equation~~ method is used, draw a relevant FBD for each disk by assuming, say, $\theta_3 > \theta_2 > \theta_1$.
Then: ($\theta_1, \theta_2, \theta_3$ +ive CCW as seen from rightside)



$$\left. \begin{aligned} \text{Then, } I_1 \ddot{\theta}_1 &= k_{t2}(\theta_2 - \theta_1) - k_{t1}\theta_1 \\ I_2 \ddot{\theta}_2 &= k_{t3}(\theta_3 - \theta_2) - k_{t2}(\theta_2 - \theta_1) \\ \& \quad I_3 \ddot{\theta}_3 &= -k_{t3}(\theta_3 - \theta_2) \end{aligned} \right\}$$

$$\Rightarrow \left. \begin{aligned} I_1 \ddot{\theta}_1 + (k_{t1} + k_{t2})\theta_1 - k_{t2}\theta_2 &= 0 \\ I_2 \ddot{\theta}_2 - k_{t2}\theta_1 + (k_{t2} + k_{t3})\theta_2 - k_{t3}\theta_3 &= 0 \\ \& \quad I_3 \ddot{\theta}_3 - k_{t3}\theta_2 + k_{t3}\theta_3 &= 0 \end{aligned} \right\} \begin{array}{l} \text{The reqd.} \\ \text{DEOM} \end{array}$$

OR

$$\begin{bmatrix} I_1 & 0 & 0 \\ 0 & I_2 & 0 \\ 0 & 0 & I_3 \end{bmatrix} \begin{Bmatrix} \ddot{\theta}_1 \\ \ddot{\theta}_2 \\ \ddot{\theta}_3 \end{Bmatrix} + \begin{bmatrix} (k_{t1} + k_{t2}) & -k_{t2} & 0 \\ -k_{t2} & (k_{t2} + k_{t3}) & -k_{t3} \\ 0 & -k_{t3} & k_{t3} \end{bmatrix} \begin{Bmatrix} \theta_1 \\ \theta_2 \\ \theta_3 \end{Bmatrix} = \begin{Bmatrix} 0 \\ 0 \\ 0 \end{Bmatrix}$$

OR

$$[I] \{\ddot{\theta}\} + [K] \{\theta\} = \{0\}$$

where $[I] = \begin{bmatrix} I_1 & 0 & 0 \\ 0 & I_2 & 0 \\ 0 & 0 & I_3 \end{bmatrix}; [K] = \begin{bmatrix} (k_{t1} + k_{t2}) & -k_{t2} & 0 \\ -k_{t2} & (k_{t2} + k_{t3}) & -k_{t3} \\ 0 & -k_{t3} & k_{t3} \end{bmatrix};$

$$\{\ddot{\theta}\} = \begin{Bmatrix} \ddot{\theta}_1 \\ \ddot{\theta}_2 \\ \ddot{\theta}_3 \end{Bmatrix} \& \quad \{\theta\} = \begin{Bmatrix} \theta_1 \\ \theta_2 \\ \theta_3 \end{Bmatrix}$$

(P.T.O)

→ The Lagrange's Equns in this case are:

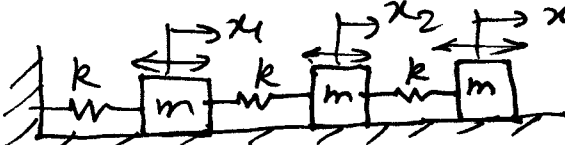
$$\frac{d}{dt} \left(\frac{\partial T}{\partial \dot{\theta}_i} \right) - \frac{\partial T}{\partial \theta_i} + \frac{\partial V}{\partial \theta_i} = 0; \quad i=1, 2, 3$$

$$T = \frac{1}{2} I_1 \dot{\theta}_1^2 + \frac{1}{2} I_2 \dot{\theta}_2^2 + \frac{1}{2} I_3 \dot{\theta}_3^2$$

$$V = \frac{1}{2} k_{t1} \theta_1^2 + \frac{1}{2} k_{t2} (\theta_2 - \theta_1)^2 + \frac{1}{2} k_{t3} (\theta_3 - \theta_2)^2$$

→ HW → Obtain the DEOM & Compare with the ones already obtained by using ^(moment-balance) Newton-Euler method.

→ We next illustrate the method of setting up the frequency equation, obtaining the natural frequencies ω_1, ω_2 & ω_3 & also obtain the modal vectors, the general free-vibration response and the conditions for principal modes of vibration, by following what was done for a 2-DOF system in a similar situation.

→ Example:-  (Neglect friction)

Obtain the natural frequencies and modal vectors. Obtain the general free-vibration response & conditions for the principal modes. Obtain a normalized modal matrix

Solution:- (PTO)

→ You can get the DEOM by either method.
 Using $k_1 = k_2 = k_3 = k$ & $m_1 = m_2 = m_3 = m$
 in the system in fig. 1 (page 1), we have:

(I) ... $\begin{cases} m\ddot{x}_1 + 2kx_1 - kx_2 = 0 \\ m\ddot{x}_2 - kx_1 + 2kx_2 - kx_3 = 0 \\ m\ddot{x}_3 - kx_2 + kx_3 = 0 \end{cases}$ The DEOM
 $[m]\{\ddot{x}\} + [k]\{x\} = \{0\}$

Here $[m] = \begin{bmatrix} m & 0 & 0 \\ 0 & m & 0 \\ 0 & 0 & m \end{bmatrix}$; $[k] = \begin{bmatrix} 2k & -k & 0 \\ -k & 2k & -k \\ 0 & -k & k \end{bmatrix}$,

$\{\ddot{x}\} = \begin{Bmatrix} \ddot{x}_1 \\ \ddot{x}_2 \\ \ddot{x}_3 \end{Bmatrix}$; $\{x\} = \begin{Bmatrix} x_1 \\ x_2 \\ x_3 \end{Bmatrix}$; $\{0\} = \begin{Bmatrix} 0 \\ 0 \\ 0 \end{Bmatrix}$.

Compare
 with the
 2-DOF
 Case
 all along

We assume: $\begin{cases} x_1 = X_1 \sin(\omega t + \phi) \\ x_2 = X_2 \sin(\omega t + \phi) \\ x_3 = X_3 \sin(\omega t + \phi) \end{cases}$ — (II)

Substitution of (II) in (I) leads to

Equations for obtaining X_1, X_2, X_3 $\begin{cases} (2k - m\omega^2)X_1 - kX_2 = 0 \text{ --- (1)} \\ -kX_1 + (2k - m\omega^2)X_2 - kX_3 = 0 \text{ --- (2)} \\ -kX_2 + (k - m\omega^2)X_3 = 0 \text{ --- (3)} \end{cases}$

We have no
 unique
 solution
 for X_1, X_2, X_3

For non-trivial X_1, X_2 & X_3 , we must have

(4) --- $\begin{vmatrix} (2k - m\omega^2) & -k & 0 \\ -k & (2k - m\omega^2) & -k \\ 0 & -k & (k - m\omega^2) \end{vmatrix} = 0$ (Reason explained earlier)

Relation (4) gives the frequency equation.

Also, from (1), $\frac{X_2}{X_1} = \frac{2k - m\omega^2}{k}$ — (5)

& in (2) & (3), $\frac{X_3}{X_2} = \frac{k}{k - m\omega^2}$ — (6) [We ignore (2) since it is a little complicated] (40)

Then $\frac{x_3}{x_1} = \frac{x_3}{x_2} \cdot \frac{x_2}{x_1} = \frac{2K - m\omega^2}{K - m\omega^2}$ (Using (5) & (6))
 --- (7)

Relations (5) & (7) will be used for

Computing $\frac{x_{21}}{x_{11}}, \frac{x_{31}}{x_{11}}$ etc. for all 3 modes.

x_{31} = amplitude of m_3 for 1st Principal mode etc.

We now expand relation (4) to get the frequency equation:

$$m^3 \omega^6 - 5Km^2 \omega^4 + 6K^2 m \omega^2 - K^3 = 0$$

(check this) OR

$$\left[\frac{\omega^2}{\left(\frac{K}{m}\right)} \right]^3 - 5 \left[\frac{\omega^2}{\left(\frac{K}{m}\right)} \right]^2 + 6 \left[\frac{\omega^2}{\left(\frac{K}{m}\right)} \right] - 1 = 0 \quad \text{--- (8)}$$

→ We now use a calculator to solve

(8) for $\left[\frac{\omega^2}{\left(\frac{K}{m}\right)} \right]$. This leads to
 { Be careful here. Remember that $\omega_1 < \omega_2 < \omega_3$ }

Your calculator may not give solutions in proper order

Using these in (5) & (7), we get

$$\left. \begin{aligned} \omega_1^2 &= 0.1981 \frac{K}{m} \\ \omega_2^2 &= 1.5549 \frac{K}{m} \\ \omega_3^2 &= 3.2470 \frac{K}{m} \end{aligned} \right\} \Rightarrow \begin{cases} \omega_1 = 0.4379 \sqrt{\frac{K}{m}} \\ \omega_2 = 1.2469 \sqrt{\frac{K}{m}} \\ \omega_3 = 1.8019 \sqrt{\frac{K}{m}} \end{cases}$$

↑ The reqd. natural frequencies

$$\frac{x_{21}}{x_{11}} = 1.8019; \frac{x_{31}}{x_{11}} = 2.2470$$

Thus, $\{X\}_1 = \begin{Bmatrix} x_{11} \\ x_{21} \\ x_{31} \end{Bmatrix} = \begin{Bmatrix} x_{11} \\ 1.8019x_{11} \\ 2.2470x_{11} \end{Bmatrix}$, the first modal vector (eigenvector) corresponding to $\omega = \omega_1$

Similarly $\{X\}_2 = \begin{Bmatrix} x_{12} \\ x_{22} \\ x_{32} \end{Bmatrix} = \begin{Bmatrix} x_{12} \\ 0.4451x_{12} \\ -0.8021x_{12} \end{Bmatrix}$ (PTO)
 x_{32} → amplitude of m_3 for 2nd pr. mode etc.
 one sign change → 2nd modal vector
 +, +, -

$$4 \quad \{X\}_3 = \begin{Bmatrix} X_{13} \\ X_{23} \\ X_{33} \end{Bmatrix} = \begin{Bmatrix} X_{13} \\ -1.2470 X_{13} \\ 0.5550 X_{13} \end{Bmatrix} \leftarrow \text{The 3rd modal vector corr. to } \omega = \omega_3$$

Two sign changes, $\uparrow +, -, +$ (plus to minus, minus to plus)

Where X_{11}, X_{12} & X_{13} are arbitrary, non-zero.
 → The other amplitudes are not arbitrary, though.
 → Remember that

X_{11} = amplitude of left mass (corr. to x_1)
 or, mass 1 for first principal (normal) mode of vibration etc., i.e.,

X_{ij} = amplitude of the i^{th} mass for j^{th} mode of vibration

with $\begin{cases} i = 1, 2, 3 \\ j = 1, 2, 3 \end{cases}$ where the i^{th} mass is associated with the i^{th} generalized coordinate x_i .

for example,
 Thus, X_{23} = amplitude of the 2nd mass for 3rd principal mode of vibration.

Now, ~~these~~ ^{three} principal modes of vibration are given as follows:-

First Principal mode $\left[\begin{array}{l} x_1(t) = X_{11} \sin(\omega_1 t + \phi_1) \\ x_2(t) = X_{21} \sin(\omega_1 t + \phi_1) \\ x_3(t) = X_{31} \sin(\omega_1 t + \phi_1) \end{array} \right\} \text{ OR } \left\{ \begin{array}{l} x_1 = X_{11} \sin(\omega_1 t + \phi_1) \\ x_2 = \mu_{21} X_{11} \sin(\omega_1 t + \phi_1) \\ x_3 = \mu_{31} X_{11} \sin(\omega_1 t + \phi_1) \end{array} \right.$

[Note:- We adopt the following notations:- [See Page 9 for clarification on

$\mu_{21} = \frac{X_{21}}{X_{11}}, \mu_{31} = \frac{X_{31}}{X_{11}}$

Similarly, $\mu_{22} = \frac{X_{22}}{X_{12}}, \mu_{32} = \frac{X_{32}}{X_{12}};$

$\mu_{23} = \frac{X_{23}}{X_{13}}, \mu_{33} = \frac{X_{33}}{X_{13}}$. This leads to the (PTO)

Read the clarification first & then come back...

following modal matrix:

$$[M] = \begin{bmatrix} X_{11} & X_{12} & X_{13} \\ X_{21} & X_{22} & X_{23} \\ X_{31} & X_{32} & X_{33} \end{bmatrix} = \begin{bmatrix} X_{11} & X_{12} & X_{13} \\ \mu_{21} X_{11} & \mu_{22} X_{12} & \mu_{23} X_{13} \\ \mu_{31} X_{11} & \mu_{32} X_{12} & \mu_{33} X_{13} \end{bmatrix}$$

with X_{11} , X_{12} & X_{13} arbitrary (nonzero)

& ~~we~~ a normalized modal matrix
could now be written as

$$[M] = \begin{bmatrix} 1 & 1 & 1 \\ \mu_{21} & \mu_{22} & \mu_{23} \\ \mu_{31} & \mu_{32} & \mu_{33} \end{bmatrix}$$

→ For a 2-DOF system, however, we
had adopted a simpler notation scheme
so that $[M] = \begin{bmatrix} \mu_1 & \mu_2 \end{bmatrix}$ instead

of $[M] = \begin{bmatrix} \mu_{21} & \mu_{22} \end{bmatrix}$
You may use either

2nd Principal mode $\left\{ \begin{array}{l} x_1(t) = X_{12} \sin(\omega_2 t + \phi_2) \\ x_2(t) = X_{22} \sin(\omega_2 t + \phi_2) \\ x_3(t) = X_{32} \sin(\omega_2 t + \phi_2) \end{array} \right\}$ OR $\left\{ \begin{array}{l} x_1(t) = X_{12} \sin(\omega_2 t + \phi_2) \\ x_2(t) = \mu_{22} X_{12} \sin(\omega_2 t + \phi_2) \\ x_3(t) = \mu_{32} X_{12} \sin(\omega_2 t + \phi_2) \end{array} \right\}$

3rd pr. mode $\left\{ \begin{array}{l} x_1(t) = X_{13} \sin(\omega_3 t + \phi_3) \\ x_2(t) = X_{23} \sin(\omega_3 t + \phi_3) \\ x_3(t) = X_{33} \sin(\omega_3 t + \phi_3) \end{array} \right\}$ OR $\left\{ \begin{array}{l} x_1(t) = X_{13} \sin(\omega_3 t + \phi_3) \\ x_2(t) = \mu_{23} X_{13} \sin(\omega_3 t + \phi_3) \\ x_3(t) = \mu_{33} X_{13} \sin(\omega_3 t + \phi_3) \end{array} \right\}$

→ For our problem here,

$$\mu_{21} = 1.8019, \mu_{31} = 2.2470$$

$$\mu_{22} = 0.4451, \mu_{32} = -0.8021$$

$$\mu_{23} = -1.2470, \mu_{33} = 0.5550$$

(PTD)

Hence, the required general free-vibration response is:-

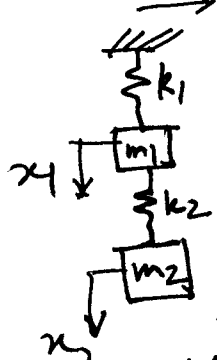
$$x_1(t) = X_{11} \sin(\omega_1 t + \phi_1) + X_{12} \sin(\omega_2 t + \phi_2) + X_{13} \sin(\omega_3 t + \phi_3)$$

$$x_2(t) = \mu_{21} X_{11} \sin(\omega_1 t + \phi_1) + \mu_{22} X_{12} \sin(\omega_2 t + \phi_2) + \mu_{23} X_{13} \sin(\omega_3 t + \phi_3)$$

$$x_3(t) = \mu_{31} X_{11} \sin(\omega_1 t + \phi_1) + \mu_{32} X_{12} \sin(\omega_2 t + \phi_2) + \mu_{33} X_{13} \sin(\omega_3 t + \phi_3)$$

where X_{11}, X_{12}, X_{13} are arbitrary (non-zero) & μ_{21} , etc. have the values mentioned on pg. 8.

→ ~~SOME~~ CLARIFICATIONS regarding notation for amplitude ratios:-



Many books use these notations

→ This may be reqd. for some students

← for this 2-DOF system, we used

$$\mu_1 = \frac{X_{21}}{X_{11}} = \frac{\text{Amplitude of 2nd mass for 1st pr. mode}}{\text{" " 1st " " " " " " " " " "}}$$

$$\mu_2 = \frac{X_{22}}{X_{12}} = \frac{\text{Ampl. of 2nd mass for 2nd pr. mode}}{\text{" " 1st " " " " " " " " " "}}$$

& the modal matrix was:

$$[M] = \begin{bmatrix} X_{11} & X_{12} \\ \mu_1 X_{11} & \mu_2 X_{12} \end{bmatrix} \text{ \& normalized}$$

$$[M] = \begin{bmatrix} 1 & 1 \\ \mu_1 & \mu_2 \end{bmatrix}. \text{ This was done for}$$

simplicity. $\mu_1 = \text{Amplitude ratio for 1st pr. mode}$
 & $\mu_2 = \text{" " " 2nd " " "}$

It worked nicely with a 2-DOF system. However, for an n -DOF system ($n > 2$), (Pro)

this scheme should be modified. For a 2-DOF system, we should replace μ_1 by μ_{21} & μ_2 by μ_{22}

so that, $[M] = \begin{bmatrix} 1 & 1 \\ \mu_{21} & \mu_{22} \end{bmatrix}$ & μ_{21} is the

element at (2,1) position (2nd row, first column) & μ_{22} is the element at (2,2) position of $[M]$ normalized.

Extending it to a 3-DOF system,

we get $[M] = \begin{bmatrix} 1 & 1 & 1 \\ \mu_{21} & \mu_{22} & \mu_{23} \\ \mu_{31} & \mu_{32} & \mu_{33} \end{bmatrix}$ (normalized)

Here

$$\mu_{21} = \frac{X_{21}}{X_{11}}$$

$$\mu_{31} = \frac{X_{31}}{X_{11}}$$

$$\mu_{22} = \frac{X_{22}}{X_{12}}$$

etc.

OR

$$[M] = \begin{bmatrix} X_{11} & X_{12} & X_{13} \\ \mu_{21} X_{11} & \mu_{22} X_{12} & \mu_{23} X_{13} \\ \mu_{31} X_{11} & \mu_{32} X_{12} & \mu_{33} X_{13} \end{bmatrix}$$

with $\begin{Bmatrix} X_{11} \\ X_{12} \\ X_{13} \end{Bmatrix}$ arbitrary (non-zero)

For a general n-DOF system,

$$[M] = \begin{bmatrix} 1 & 1 & \dots & 1 \\ \mu_{21} & \mu_{22} & \dots & \mu_{2n} \\ \vdots & \vdots & \ddots & \vdots \\ \mu_{n1} & \mu_{n2} & \dots & \mu_{nn} \end{bmatrix} \text{ (normalized)}$$

OR

$$[M] = \begin{bmatrix} X_{11} & X_{12} & \dots & X_{1n} \\ \mu_{21} X_{11} & \mu_{22} X_{12} & \dots & \mu_{2n} X_{1n} \\ \vdots & \vdots & \ddots & \vdots \\ \mu_{n1} X_{11} & \mu_{n2} X_{12} & \dots & \mu_{nn} X_{1n} \end{bmatrix}$$

($X_{11}, X_{12}, \dots, X_{1n} \rightarrow$ arbitrary (non-zero)) (P.T.O)

where $\mu_{21} = \frac{x_{21}}{x_{11}}, \mu_{n1} = \frac{x_{n1}}{x_{11}}$

$\mu_{22} = \frac{x_{22}}{x_{12}}, \mu_{n2} = \frac{x_{n2}}{x_{12}}$

& so on \rightarrow *back to GOV of page now*

\rightarrow We get back to our example problem (pg. 4)

\rightarrow It can be shown that for the n^{th} principal mode of vibration, there are specific relations to be satisfied among the ^{given} initial displacements $x_1(0), \dots, x_n(0)$ and the initial velocities $\dot{x}_1(0), \dots, \dot{x}_n(0)$.

• For a 2-DOF system, these were:

\rightarrow for first pr. mode, $x_2(0) = \mu_{21} x_1(0) \text{ or } \mu_{21} x_1(0)$
 & $\dot{x}_2(0) = \mu_{21} \dot{x}_1(0) \text{ or } \mu_{21} \dot{x}_1(0)$

\rightarrow for 2nd pr. mode, $x_2(0) = \mu_{22} x_1(0) \text{ or } \mu_{22} x_1(0)$
 & $\dot{x}_2(0) = \mu_{22} \dot{x}_1(0) \text{ or } \mu_{22} \dot{x}_1(0)$

• for a 3-DOF system, these would be:

\rightarrow For first mode, $\left. \begin{matrix} x_2(0) = \mu_{21} x_1(0) \\ x_3(0) = \mu_{31} x_1(0) \end{matrix} \right\} \& \left\{ \begin{matrix} \dot{x}_2(0) = \mu_{21} \dot{x}_1(0) \\ \dot{x}_3(0) = \mu_{31} \dot{x}_1(0) \end{matrix} \right.$

\rightarrow for 2nd mode, $\left. \begin{matrix} x_2(0) = \mu_{22} x_1(0) \\ x_3(0) = \mu_{32} x_1(0) \end{matrix} \right\} \& \left\{ \begin{matrix} \dot{x}_2(0) = \mu_{22} \dot{x}_1(0) \\ \dot{x}_3(0) = \mu_{32} \dot{x}_1(0) \end{matrix} \right.$

\rightarrow for 3rd mode, $\left. \begin{matrix} x_2(0) = \mu_{23} x_1(0) \\ x_3(0) = \mu_{33} x_1(0) \end{matrix} \right\} \& \left\{ \begin{matrix} \dot{x}_2(0) = \mu_{23} \dot{x}_1(0) \\ \dot{x}_3(0) = \mu_{33} \dot{x}_1(0) \end{matrix} \right.$ (PTD)

Required Conditions

to excite the various principal modes of vibration

→ (Ex. problem continued) → See pg. 4

A normalized modal matrix $[M] : \sim$

(See pg. 8) $[M] = \begin{bmatrix} 1 & 1 & 1 \\ \mu_{21} & \mu_{22} & \mu_{23} \\ \mu_{31} & \mu_{32} & \mu_{33} \end{bmatrix}$

(This completes the solution) $= \begin{bmatrix} 1 & 1 & 1 \\ 1.8019 & 0.4451 & -1.2470 \\ 2.2470 & -0.8021 & 0.5550 \end{bmatrix}$

\downarrow \downarrow \downarrow
 $\{x\}_1$ $\{x\}_2$ $\{x\}_3$

(Note:-

The first modal vector $\{x\}_1$, does not have any sign change as its elements are explored from top to bottom.

$\{x\}_2$ involves one sign change ($\begin{pmatrix} + \\ - \\ + \end{pmatrix}$)

$\{x\}_3$ " two " changes ($\begin{pmatrix} + \\ - \\ + \end{pmatrix}$)

plus to minus
minus to plus

