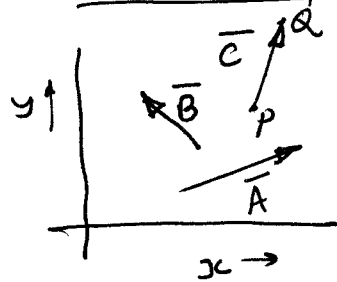
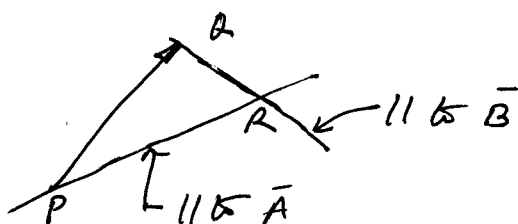


(ii) The expansion theorem :- Suppose we have two vectors  $\vec{A}$  &  $\vec{B}$  such that  $\vec{A} \neq \alpha \vec{B}$ ,  $\alpha$  being a constant. (see fig.) Then we say that  $\vec{A}$  &  $\vec{B}$  are linearly independent. So,



any third vector  $\vec{C}$  (we are considering 2-D vectors for the moment) can be expressed as a linear combination of  $\vec{A}$  &  $\vec{B}$ , that is, we can find constants  $\alpha$  &  $\beta$  such that  $\vec{C} = \alpha \vec{A} + \beta \vec{B}$ . This can be seen as follows :- let  $\vec{C} = \vec{PQ}$ .

Through P, draw a line parallel to  $\vec{A}$  & through Q, draw a line parallel to  $\vec{B}$ . Let these intersect at R.



$$\begin{aligned} \text{Then, } \vec{C} = \vec{PQ} &= \vec{PR} + \vec{RQ} \\ &= \alpha \vec{A} + \beta \vec{B} \end{aligned}$$

Now, we can represent  $\vec{A}$  also as  $\begin{Bmatrix} A_x \\ A_y \end{Bmatrix}$ ,  $\vec{B}$  as  $\begin{Bmatrix} B_x \\ B_y \end{Bmatrix}$  &  $\vec{C}$  as  $\begin{Bmatrix} C_x \\ C_y \end{Bmatrix}$  where  $A_x$  is the x-component of  $\vec{A}$  etc. Then,

$$\begin{Bmatrix} C_x \\ C_y \end{Bmatrix} = \alpha \begin{Bmatrix} A_x \\ A_y \end{Bmatrix} + \beta \begin{Bmatrix} B_x \\ B_y \end{Bmatrix}$$

→ So, if we have the modal vectors  $\{x\}_1$  &  $\{x\}_2$  linearly independent, then, any 2-D vector  $\{u\} = \begin{Bmatrix} u_1 \\ u_2 \end{Bmatrix}$  can be expressed as :

$$\{u\} = c_1 \{x\}_1 + c_2 \{x\}_2$$

& this is called the expansion theorem. This theorem has far-reaching consequences in

our studies. Actually if  $\omega_1 \neq \omega_2$ , then ②  
 $\{x\}_1$  &  $\{x\}_2$  will be linearly independent. What  
happens when  $\omega_1 = \omega_2$  will be taken up  
later. So, in our example problem,  
 $\omega_1 \neq \omega_2$  and  $\{x\}_1 = \begin{Bmatrix} 1 \\ 1.618 \end{Bmatrix}$  &  $\{x\}_2 = \begin{Bmatrix} 1 \\ -0.618 \end{Bmatrix}$   
are linearly independent. If this were  
not so,  ~~$\{x\}_1$  &  $\{x\}_2$~~  then  $\{x\}_2 = \alpha \{x\}_1$  for  
some constant  $\alpha$  & so,  $\{x\}_2$  &  $\{x\}_1$  would  
basically represent the same modal  
vector. This is so because if  $\{x\}_1$  is  
a modal vector,  ~~$\{x\}_2$~~  corresponding to  
 $\omega = \omega_1$ , then  $\alpha \{x\}_1$  is also a modal vector  
for  $\omega = \omega_1$ , since the elements of a modal  
vector can be multiplied by an arbitrary  
non-zero number & it still remains a  
modal vector. [Remember we said that  
in  $\{x\}_1 = \begin{Bmatrix} x_{11} \\ \mu_1 x_{11} \end{Bmatrix}$ ,  $x_{11}$  is arbitrary? So,  
 $x_{11}$  could be replaced by  $\alpha x_{11}$  without problem]

→ Now, we said any arbitrary  $\{u\} = \begin{Bmatrix} u_1 \\ u_2 \end{Bmatrix}$  can  
be expanded as:  $\{u\} = c_1 \{x\}_1 + c_2 \{x\}_2$  -- (i)  
How do we find  $c_1$  &  $c_2$ ? We shall apply  
a trick to bring into picture the orthogonality  
relation & this is what we do:  
From (i), premultiplications will give

$$\{x_1\}^T [m] \{u\} = c_1 \{x_1\}^T [m] \{x\}_1 + c_2 \{x_1\}^T [m] \{x\}_2$$

But by mass orthogonality of the modal vectors, we have

$$\{x\}_1^T [m] \{x\}_2 = 0. \text{ Hence, } c_1 = \frac{\{x\}_1^T [m] \{u\}}{\{x\}_1^T [m] \{x\}_1}.$$

$$\text{Similarly, } c_2 = \frac{\{x\}_2^T [m] \{u\}}{\{x\}_2^T [m] \{x\}_2}, \text{ check.}$$

→ You could, of course, premultiply both sides of (i) by  $\{x\}_1^T [K]$  instead of  $\{x\}_1^T [m]$  & then obtain  $c_1$  using stiffness orthogonality relation  $\{x\}_1^T [K] \{x\}_2 = 0$ .

But normally  $[m]$  is simpler than  $[K]$ , so we used  $[m]$  instead of  $[K]$ .

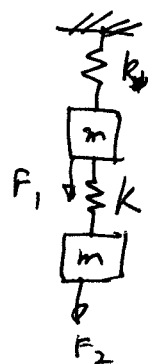
→ Now try to see this: Suppose ~~the~~ our example system is executing vibratory motion  $\{x_1(t)\}$  under a set of complex forcing functions.

Then, at an instant  $t = t_1$ , the vector  $\begin{Bmatrix} x_1(t_1) \\ x_2(t_1) \end{Bmatrix}$  can be expressed as:-

$$\begin{Bmatrix} x_1(t_1) \\ x_2(t_1) \end{Bmatrix} = c_1 \{x\}_1 + c_2 \{x\}_2, \text{ by the expansion theorem. At } t = t_1 + dt, \begin{Bmatrix} x_1(t_1 + dt) \\ x_2(t_1 + dt) \end{Bmatrix} = c'_1 \{x\}_1 + c'_2 \{x\}_2$$

Where  $c'_1$  is very little different from  $c_1$  &  $c'_2$  is also very little different from  $c_2$ . So, may be we could write  $c_1 = c_1(t)$  &  $c_2 = c_2(t)$ , so that  $\begin{Bmatrix} x_1(t) \\ x_2(t) \end{Bmatrix} = c_1(t) \{x\}_1 + c_2(t) \{x\}_2$ ?

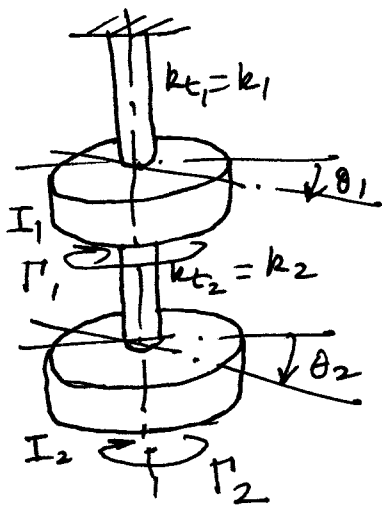
~~Actually~~ So, it seems we could obtain the forced response if we could find  $c_1(t)$  &  $c_2(t)$ ? Actually, when you will study



④

vibration of continuous systems, you will see that several approximate methods for finding forced response depend on an expansion theorem like the above with the modal vectors replaced by the so called eigenfunctions.

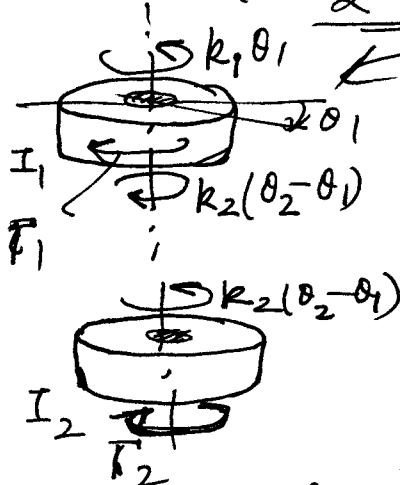
### ⑤ The rotational 2-DOF system:-



$T_1 = T_1(t)$   
&  $T_2 = T_2(t)$  are  
the applied  
external torques

The standard 2-DOF system for torsional vibration is shown here, damping neglected. To derive the DEOM, we might use the ~~the~~ moment balance method as follows:-

Let  $\theta_2 > \theta_1$ . The FBDs are:-



∴ The DEOM are:-

$$I_1 \ddot{\theta}_1 = -k_1 \theta_1 + k_2 (\theta_2 - \theta_1) + T_1$$

$$I_2 \ddot{\theta}_2 = -k_2 (\theta_2 - \theta_1) + T_2$$

OR

$$I_1 \ddot{\theta}_1 + (k_1 + k_2) \theta_1 - k_2 \theta_2 = T_1$$

$$I_2 \ddot{\theta}_2 - k_2 \theta_1 + k_2 \theta_2 = T_2$$

Hence, in matrix form, the DEOM are:

$$[I] \{\ddot{\theta}\} + [k] \{\theta\} = \{T\} \quad \text{--- (1) where}$$

$[I] = \begin{bmatrix} I_1 & 0 \\ 0 & I_2 \end{bmatrix}$  is the inertia matrix,  $[k] = \begin{bmatrix} k_1 + k_2 & -k_2 \\ -k_2 & k_2 \end{bmatrix}$  is the stiffness matrix,  $\{\theta\} = \begin{Bmatrix} \theta_1(t) \\ \theta_2(t) \end{Bmatrix}$  is the (angular) displacement vector,  $\{\ddot{\theta}\} = \begin{Bmatrix} \ddot{\theta}_1(t) \\ \ddot{\theta}_2(t) \end{Bmatrix}$  is the (angular) acceleration vector &  $\{T\} = \begin{Bmatrix} T_1(t) \\ T_2(t) \end{Bmatrix}$  is the torque vector.  $\rightarrow$

For free vibration, the DEOM are:

(5)

$$I_1 \ddot{\theta}_1 + (k_1 + k_2) \theta_1 - k_2 \theta_2 = 0 \quad \text{--- (2)}$$

$$\& I_2 \ddot{\theta}_2 - k_2 \theta_1 + k_2 \theta_2 = 0 \quad \text{--- (3)}$$

$$\text{Assume } \left. \begin{aligned} \theta_1 &= (H)_1 \sin(\omega t + \phi) \\ \theta_2 &= (H)_2 \sin(\omega t + \phi) \end{aligned} \right\} \Rightarrow \left. \begin{aligned} \ddot{\theta}_1 &= -\omega^2 (H)_1 \sin(\omega t + \phi) \\ \ddot{\theta}_2 &= -\omega^2 (H)_2 \sin(\omega t + \phi) \end{aligned} \right\}$$

Substitution in (2) & (3) results in

$$\left. \begin{aligned} & [(k_1 + k_2) - I_1 \omega^2] (H)_1 - k_2 (H)_2 = 0 \\ & -k_2 (H)_1 + (k_2 - I_2 \omega^2) (H)_2 = 0 \end{aligned} \right\} \text{The amplitude equations}$$

For non-trivial  $(H)_1$  &  $(H)_2$ ,

$$\begin{vmatrix} (k_1 + k_2) - I_1 \omega^2 & -k_2 \\ -k_2 & k_2 - I_2 \omega^2 \end{vmatrix} = 0, \text{ giving the frequency equation.}$$

Its solution gives  $\omega_1$  &  $\omega_2$  as before.

The modal vectors are:-

$$\{(H)\}_1 = \left\{ \begin{matrix} (H)_{11} \\ \mu_1 (H)_{11} \end{matrix} \right\} \& \{(H)\}_2 = \left\{ \begin{matrix} (H)_{12} \\ \mu_2 (H)_{12} \end{matrix} \right\}$$

& the normalized modal vectors are:

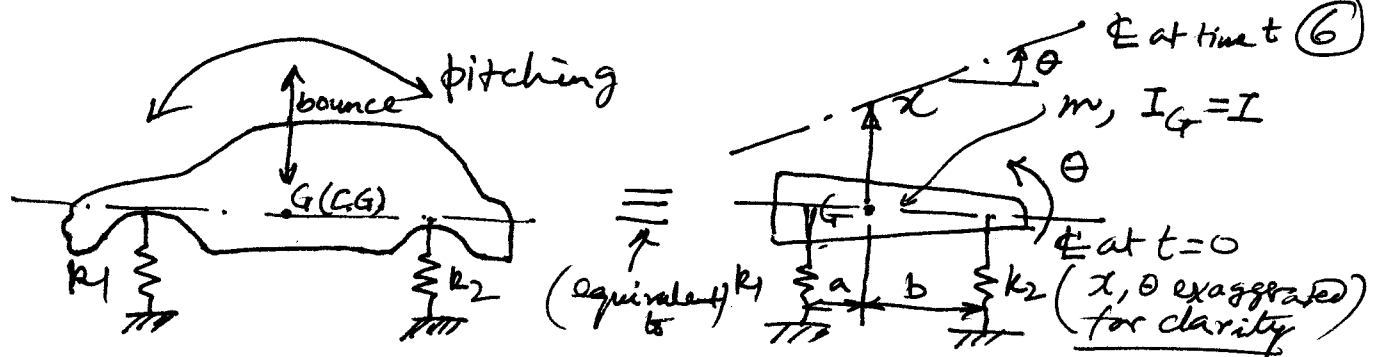
$$\{(H)\}_1 = \left\{ \begin{matrix} 1 \\ \mu_1 \end{matrix} \right\} \& \{(H)\}_2 = \left\{ \begin{matrix} 1 \\ \mu_2 \end{matrix} \right\} \text{ where}$$

$\mu_1$  &  $\mu_2$  are defined as in a translational system.

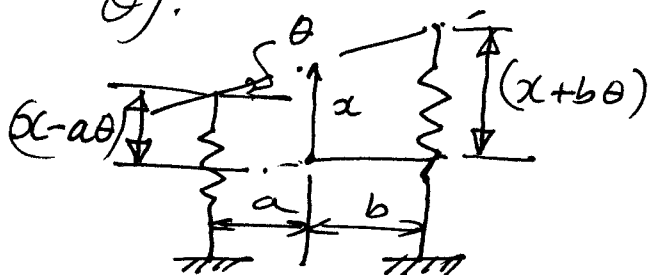
An important example:- In this example of a 2-DOF system, both translational and rotational vibrations are involved.

Here a 2-DOF model of a car is considered to study its bounce (up & down) & pitching <sup>motion</sup> (angular motion about a horizontal axis).  
~~also~~ We consider free-vibration only.

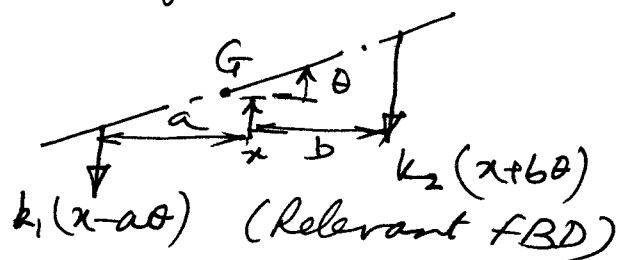
→



Here the assumptions are that the C.G. of the car moves up & down (denoted by generalized coordinate  $x$ ) and the carbody rotates about a horizontal axis through  $G$  (rotation denoted by generalized coordinate  $\theta$ ). ( $x, \theta$  are both actually small)



(Dimensions arbitrary for illustration)



Let  $m = \text{mass of car}$   
 $I = \text{Moment of inertia about an axis through G.}$

$$\text{So, } m\ddot{x} = -k_1(x - a\theta) - k_2(x + b\theta)$$

$$\& I\ddot{\theta} = +k_1(x - a\theta)a - k_2(x + b\theta)b$$

$$\Rightarrow m\ddot{x} + (k_1 + k_2)x - (k_1a - k_2b)\theta = 0 \quad (1)$$

$$\& I\ddot{\theta} - (k_1a - k_2b)x + (k_1a^2 + k_2b^2)\theta = 0 \quad (2)$$

① & ② are the required DEOM.

In matrix form, these can be written as:

$$\begin{bmatrix} m & 0 \\ 0 & I \end{bmatrix} \begin{Bmatrix} \ddot{x} \\ \ddot{\theta} \end{Bmatrix} + \begin{bmatrix} (k_1 + k_2) & -(k_1a - k_2b) \\ -(k_1a - k_2b) & k_1a^2 + k_2b^2 \end{bmatrix} \begin{Bmatrix} x \\ \theta \end{Bmatrix} = \begin{Bmatrix} 0 \\ 0 \end{Bmatrix}$$

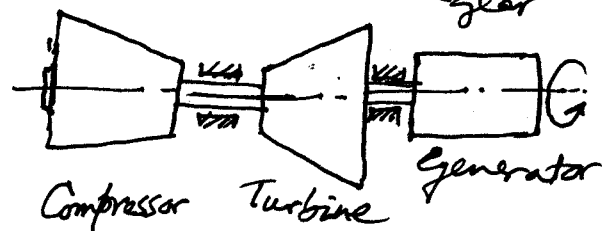
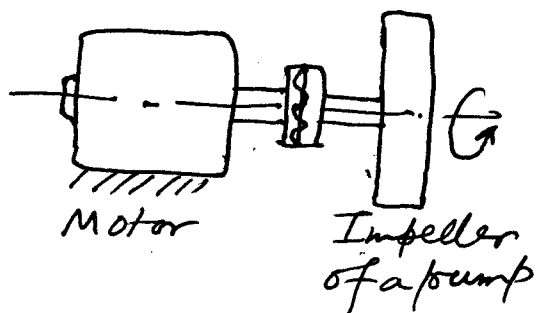
Check  $\rightarrow$  For finding  $x(t)$  &  $\theta(t)$ , we assume

$$\left. \begin{aligned} x &= X \sin(\omega t + \phi) \\ \& \theta &= \Theta \sin(\omega t + \phi) \end{aligned} \right\}$$

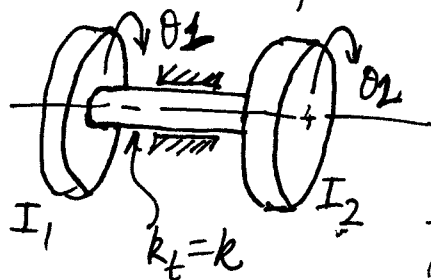
& proceed as in the earlier cases.

Example (from S.S. Rao, Mechanical Vibrations, 6th Ed, p. 534)  
Study & do example 5.7:

⑤ Semidefinite (free-free) systems:- These are also known as ~~semidefinite~~ free-free systems and are very important in mechanical engineering. 3 ~~free~~ examples are shown below & here. →



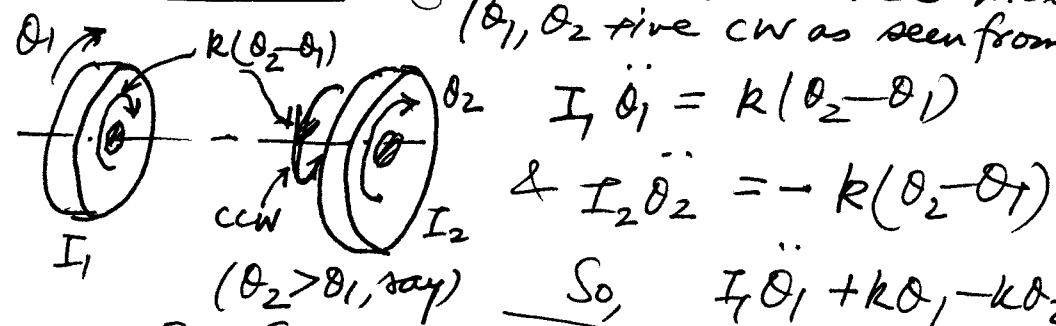
→ We consider the following simplified model for free-vibration studies:



$\theta_1(t)$  &  $\theta_2(t)$  are the generalized coordinates which have zero values at static equilibrium.

To visualize free vibration of this system, you may assume that  $I_1$  is held fixed,  $I_2$  is given an initial twist  $\theta_2(0)$  & then both discs are released. The system would execute torsional free vibration. We, as usual, assume small amplitude oscillations. However, here there is an interesting feature of this problem: it has a zero natural frequency as you will see soon. →

→ The DEOM:- (Moment balance method)  
 ( $\theta_1, \theta_2$  are CW as seen from the right)



So,  $I_1 \ddot{\theta}_1 + k\theta_1 - k\theta_2 = 0$  — (1)  
 $I_2 \ddot{\theta}_2 - k\theta_1 + k\theta_2 = 0$  — (2)

(1) & (2) are the reqd DEOM.

→ It is interesting to note that by adding (1) & (2), we get  $I_1 \ddot{\theta}_1 + I_2 \ddot{\theta}_2 = 0$  & integrating this equation, we get  $I_1 \dot{\theta}_1 + I_2 \dot{\theta}_2 = \text{Constant}$ , which is the conservation of angular momentum relation for our free-free system.

→ To get  $\theta_1(t)$  &  $\theta_2(t)$ , we assume  $\theta_1 = \hat{\theta}_1 \sin(\omega t + \phi)$ ,  $\theta_2 = \hat{\theta}_2 \sin(\omega t + \phi)$ .  
 Then substitution in (1) & (2) gives:

$(k - I_1 \omega^2) \hat{\theta}_1 - k \hat{\theta}_2 = 0$  — (3)

&  $-k \hat{\theta}_1 + (k - I_2 \omega^2) \hat{\theta}_2 = 0$  — (4)

So, frequency eqn. is:

$$\begin{vmatrix} k - I_1 \omega^2 & -k \\ -k & k - I_2 \omega^2 \end{vmatrix} = 0$$

$\Rightarrow k^2 - k(I_1 + I_2)\omega^2 + I_1 I_2 \omega^4 - k^2 = 0$

$\Rightarrow \omega^2 [I_1 I_2 \omega^2 - k(I_1 + I_2)] = 0$

So,  $\omega_1 = 0$ ,  $\omega_2 = \sqrt{\frac{k(I_1 + I_2)}{I_1 I_2}}$ .

This zero natural frequency actually means



rigid body motion is possible, that is, our disc-shaft system can rotate as a single rigid body under proper initial conditions. This setup can also execute torsional oscillations at  $\omega_2$ .

→ To get  $\{H\}_1$  &  $\{H\}_2$  <sup>for first principal mode,</sup> we substitute  $\omega_1 = 0$  in (3) or (4) & get  $\mu_1 = \frac{(H)_{21}}{(H)_{11}} = \frac{k}{k} = 1$

Hence,  $\{H\}_1 =$  modal vector for first pr. mode  
 $= \begin{Bmatrix} (H)_{11} \\ (H)_{11} \end{Bmatrix}$ . The normalized <sub>arbitrary</sub>

$$\{H\}_1 = \begin{Bmatrix} 1 \\ 1 \end{Bmatrix} \text{ (Remember).}$$

for  $\omega = \omega_2$ , (3) gives  $\frac{(H)_{22}}{(H)_{12}} = \frac{k - I_1 \omega_2^2}{k}$

or,  $\mu_2 = \frac{(H)_{22}}{(H)_{12}} = \frac{k - \frac{k(I_1 + I_2)}{I_2}}{k} = -\frac{I_1}{I_2}$

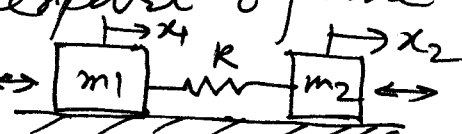
Hence,  $\{H\}_2 = \begin{Bmatrix} (H)_{12} \\ -\frac{I_1}{I_2} (H)_{12} \end{Bmatrix} = \begin{Bmatrix} (H)_{12} \\ \mu_2 (H)_{12} \end{Bmatrix}$ ,  $(H)_{12}$  arbitrary.

$\left( \{H\}_2 \right)$  <sub>normalized</sub>  $= \begin{Bmatrix} 1 \\ \mu_2 \end{Bmatrix}$ .

→ One interesting thing to note about the stiffness matrix  ~~$\begin{bmatrix} k_1 + k_2 & k_2 \\ k_2 & k_1 \end{bmatrix}$~~   $\begin{bmatrix} k & -k \\ -k & k \end{bmatrix} = [k]$  is that  $\det [k] = k^2 - k^2 = 0$ .

This shows that ~~the~~  $[k]$  is a semidefinite matrix & so our system

is semidefinite.

→ The translational counterpart of the semidefinite system is: 

⑧ To derive the DEOM for all the systems taken up so far by using Lagrange's eqns

(Free vibration only)  
① There are 2 Lagrange eqns for a 2-DOF system.  
Here  $x_1$  &  $x_2$  are the generalized coordinates.  
The Lagrange eqns are:

$$\frac{d}{dt} \left( \frac{\partial T}{\partial \dot{x}_1} \right) - \frac{\partial T}{\partial x_1} + \frac{\partial V}{\partial x_1} = 0 \quad \text{--- (1)}$$

$$\& \frac{d}{dt} \left( \frac{\partial T}{\partial \dot{x}_2} \right) - \frac{\partial T}{\partial x_2} + \frac{\partial V}{\partial x_2} = 0 \quad \text{--- (2)}$$

Now,  $T = \frac{1}{2} m_1 \dot{x}_1^2 + \frac{1}{2} m_2 \dot{x}_2^2$

$$V = \frac{1}{2} k_1 x_1^2 + \frac{1}{2} k_2 (x_2 - x_1)^2$$

So,  $\frac{\partial T}{\partial \dot{x}_1} = m_1 \dot{x}_1$ ,  $\frac{d}{dt} \left( \frac{\partial T}{\partial \dot{x}_1} \right) = m_1 \ddot{x}_1$ ,  $\frac{\partial T}{\partial x_1} = 0$ ,

$$\frac{\partial V}{\partial x_1} = k_1 x_1 - k_2 (x_2 - x_1) = (k_1 + k_2) x_1 - k_2 x_2$$

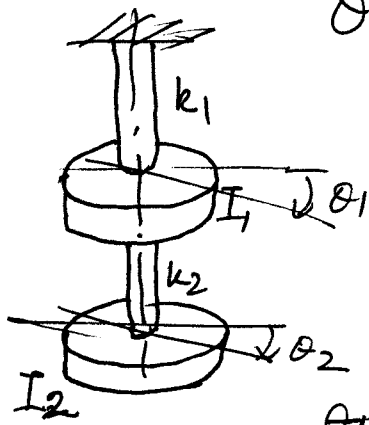
So, from (1),  $m_1 \ddot{x}_1 + (k_1 + k_2) x_1 - k_2 x_2 = 0$  (1<sup>st</sup> DEOM)

Also,  $\frac{d}{dt} \left( \frac{\partial T}{\partial \dot{x}_2} \right) = m_2 \ddot{x}_2$ ,  $\frac{\partial T}{\partial x_2} = 0$ ,  $\frac{\partial V}{\partial x_2} = k_2 (x_2 - x_1)$

Hence, from (2), we get  $m_2 \ddot{x}_2 - k_2 x_1 + k_2 x_2 = 0$ , which is the 2<sup>nd</sup> DEOM.

→

②



$\theta_1, \theta_2 \rightarrow$  the generalized coordinates (11)

$$T = \frac{1}{2} I_1 \dot{\theta}_1^2 + \frac{1}{2} I_2 \dot{\theta}_2^2$$

$$V = \frac{1}{2} k_1 \theta_1^2 + \frac{1}{2} k_2 (\theta_2 - \theta_1)^2$$

The Lagrange equations in this case are (for free vibration):

$$\frac{d}{dt} \left( \frac{\partial T}{\partial \dot{\theta}_1} \right) - \frac{\partial T}{\partial \theta_1} + \frac{\partial V}{\partial \theta_1} = 0 \quad \text{--- (1)}$$

$$\& \quad \frac{d}{dt} \left( \frac{\partial T}{\partial \dot{\theta}_2} \right) - \frac{\partial T}{\partial \theta_2} + \frac{\partial V}{\partial \theta_2} = 0 \quad \text{--- (2)}$$

$$\text{So, } \frac{d}{dt} \left( \frac{\partial T}{\partial \dot{\theta}_1} \right) = I_1 \ddot{\theta}_1, \quad \frac{\partial T}{\partial \theta_1} = 0, \quad \frac{\partial V}{\partial \theta_1} = k_1 \theta_1 - k_2 (\theta_2 - \theta_1)$$

$$\text{From (1), } I_1 \ddot{\theta}_1 + (k_1 + k_2) \theta_1 - k_2 \theta_2 = 0 \quad \text{--- (3)}$$

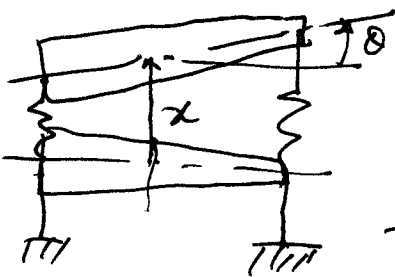
$$\text{Also, } \frac{d}{dt} \left( \frac{\partial T}{\partial \dot{\theta}_2} \right) = I_2 \ddot{\theta}_2, \quad \frac{\partial T}{\partial \theta_2} = 0, \quad \frac{\partial V}{\partial \theta_2} = k_2 (\theta_2 - \theta_1)$$

$$\text{From (2), } I_2 \ddot{\theta}_2 - k_2 \theta_1 + k_2 \theta_2 = 0 \quad \text{--- (4)}$$

③ & ④ are the required DEOM.

③

(see figs. for earlier method)



$x, \theta$  are the generalized coordinates.

The Lagrange eqns for free vibration are:

$$\frac{d}{dt} \left( \frac{\partial T}{\partial \dot{x}} \right) - \frac{\partial T}{\partial x} + \frac{\partial V}{\partial x} = 0 \quad \text{--- (1)}$$

$$\& \quad \frac{d}{dt} \left( \frac{\partial T}{\partial \dot{\theta}} \right) - \frac{\partial T}{\partial \theta} + \frac{\partial V}{\partial \theta} = 0 \quad \text{--- (2)}$$

$$\text{Here } T = \frac{1}{2} m \dot{x}^2 + \frac{1}{2} I \dot{\theta}^2$$

$$V = \frac{1}{2} k_1 (x - a\theta)^2 + \frac{1}{2} k_2 (x + b\theta)^2$$

Hence,  $\frac{d}{dt}\left(\frac{\partial T}{\partial \dot{x}}\right) = m\ddot{x}$ ,  $\frac{\partial T}{\partial x} = 0$ ,  $\frac{\partial V}{\partial x} = k_1(x-a) + k_2(x+b)$  (12)  
 $= (k_1 + k_2)x - (k_1a - k_2b)$

Thus, from ①, we get

$$m\ddot{x} + (k_1 + k_2)x - (k_1a - k_2b) = 0 \quad \text{--- (3)}$$

Again,  $\frac{d}{dt}\left(\frac{\partial T}{\partial \dot{\theta}}\right) = I\ddot{\theta}$ ,  $\frac{\partial T}{\partial \theta} = 0$ ,  $\frac{\partial V}{\partial \theta} = -k_1a(x-a) + k_2b(x+b)$

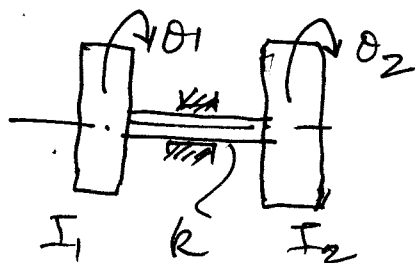
$$\text{or, } \frac{\partial V}{\partial \theta} = -(k_1a - k_2b)x + (k_1a^2 + k_2b^2)\theta$$

So, from ②, we get

$$I\ddot{\theta} - (k_1a - k_2b)x + (k_1a^2 + k_2b^2)\theta = 0 \quad \text{--- (4)}$$

③ & ④ are the req'd DEOM.

④



$\theta_1, \theta_2$  are the generalized coordinates.

The ~~the~~ Lagrange equations are:

$$\frac{d}{dt}\left(\frac{\partial T}{\partial \dot{\theta}_1}\right) - \frac{\partial T}{\partial \theta_1} + \frac{\partial V}{\partial \theta_1} = 0 \quad \text{--- (1)}$$

$$\frac{d}{dt}\left(\frac{\partial T}{\partial \dot{\theta}_2}\right) - \frac{\partial T}{\partial \theta_2} + \frac{\partial V}{\partial \theta_2} = 0 \quad \text{--- (2)}$$

$$T = \frac{1}{2}I_1\dot{\theta}_1^2 + \frac{1}{2}I_2\dot{\theta}_2^2, \quad V = \frac{1}{2}k(\theta_2 - \theta_1)^2$$

Differentiation will lead to the same DEOM as before.

Complete the derivation.

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VA-4,  
End of Part 3

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