VA-6, PART3 (8) Modal Analysis for an n-Dof system; Let the Deom be $[m]\{\dot{x}\}+[\kappa]\{x\}=\{F(t)\}-\{1\}$ Where $[m]=[m_1, m_2, m_2, m_2]$, $[\kappa]=[k_1, k_1, k_2, k_2, k_2]$ $[m_1]$ $[m_2]$ $[m_2$ our aim is to Obtain the forced response may of the system (1) by uncoupling the DEOM. full) To this end, we first obtain the modal Ameningle vectors and the associated natural frequencies. For a system having a large number of DOF, these can be obtained by a numerical technique such as the matrix iteration method. ($\omega_1 \omega_2 - - \omega_n$) Let [r] = [r21 r22 -- r2n] be a normalized modal matsix. To uncouple 1), we introduce a new set of generalized coordinates $\{p\} = \{p_n(t)\} \}$ such that $\{x(t)\} = \{y\} \{p(t)\} - \{y\} \}$ Then, {xi}=[M]{p} (\$i}={p, pi---pi}) & 1 transforms into: [m][n] {p}+[k][M] {p} = {F(t)} Fremultiplying both sides by SMT, we get [M] [M] [M] [P] + [M] [K] [M] {P] = [M] {F(t)} - -3 But by virtue of the orthogonality principle (to be established soon), the off-diagonal terms

of [M] [M] = [M] would be all zeros. Similarly, SMT(N)SM=[K]also would be diagonal. Also, let SMTSF3= Saj= Saj(t)?
Then, 3 can be written as: [M]{b}+[K]{b}= {Q(t)}--Where $[M] = \begin{bmatrix} M_{11} & 0 & --- & -0 \\ 0 & M_{22} & --- & 0 \\ \bar{0} & \bar{0} & \bar{-} & \bar{-} & M_{nn} \end{bmatrix}, [K] = \begin{bmatrix} K_{11} & 0 & --- & 0 \\ 0 & K_{22} & --- & 0 \\ \bar{0} & \bar{0} & \bar{-} & \bar{-} & \bar{-} & \bar{-} \\ \bar{0} & \bar{0} & \bar{-} & \bar{-} & \bar{-} & \bar{-} \end{bmatrix}.$ -> M11, M22 etc. are called generalized masses & K1, K22 etc. are called generalized stiffnesses. Now Dean be explicitly written as: (5)— $\begin{pmatrix}
M_{11} p_1 + K_{11} p_1 = Q_1(t) \\
M_{22} p_2 + K_{22} p_2 = Q_2(t)
\end{pmatrix}$ These are a set of the suppose the suppose of the suppose o The Coordinates p, (t), p2(t), ---, Px(t) are called principal coordinates, some authors call them normal coordinates or natural coordinates.) In passing, note that the number of sets of principal coordinates is theoretically infinite since the transformation {x(t)}= [1] {p(t)} could be invoked with

 $[H] = \begin{bmatrix} \chi_{11} & \chi_{12} & --- & \chi_{1n} \\ \mu_{21} \chi_{11} & \mu_{22} \chi_{12} & --\mu_{2n} \chi_{1n} \\ \mu_{n1} \chi_{11} & \mu_{n2} \chi_{12} & --\mu_{nn} \chi_{1n} \end{bmatrix} \text{ where } \chi_{1,1} \chi_{12} , \dots, \chi_{1n} \chi_{1n}$ are arbitrary. Hence, Sp(t) = [M] {xtt)} will tome different for different X1, X12, -- , XIN] The now obtain p(t), p2(t), --, p(t) from (5) by using Duhamal's integral. For instance, M, P, + KIIP, = Q, (t) has the forced Presponse $P_i(t) = \int_{\Lambda}^{C} Q_i(t)g_i(t-t)dt$ where g,(t)= Linu,t. [This git) is obtained by comparison with $m \dot{x} + k x = Q_{i}(t)$ for which reforced = $x(t) = \int Q_{1}(t) g(t-t) dt$ with g(t)= finan sinat.] Note that, $\omega_1 = \sqrt{\frac{K_{11}}{M_{11}}}, \omega_2 = \sqrt{\frac{K_{22}}{M_{22}}}$ etc. (So, after you somin the [M] & [K] matrices, check whether \(\frac{K_{ii}}{M_{ii}}\) & does indeed give ω , etc. If not, you made a mistake somewhere.) So, in general, $P_r(t) = \int_0^t Q_r(\tau)g_r(t-\tau)d\tau$

in general, $P_r(t) = \int_0^t Q_r(\tau) g_r(t-\tau) d\tau$ Where $P_r(t) = \frac{1}{M_{rr} w_r} Sin w_r t$; r = 1, 2, ..., n. After obtaining p(t), --, Ph(t) this way,
the required forced responses in terms
of x(t), -, xult can be obtained from {x(t)} = [M] {p(t)}, that is, \(\text{xi(t)} \) = \[\frac{\mu_2}{\mu_1} \\ \frac{\mu_2}{\mu_2} - \frac{\mu_2}{\mu_n} \\ \frac{\mu_2(t)}{\mu_n(t)} \\ \frac{\mu_n(t)}{\mu_n(t)} \] $\gamma(t) = \beta_i(t) + \beta_i(t) + - - + \beta_n(t)$ 22(t) = 1/2, P,(t) + M2 P2(t) + --+ M2 Pn(t) The rayd forced response. $x_{n}(t) = \mu_{n} p_{1}(t) + \mu_{n} p_{2}(t) + - + \mu_{n} p_{n}(t)$ Note - If of you are asked to obtain a set of principal soordinates for a given n-Dor system, you should use the relation {p(t)}=[M]{x(t)} & Main Pr(t) as a linear combination of 2(t), 2(t), ---, 2(t) for r=1,2,-,n. (8) We shall digress for a while before we come back to modal analysis. - The now prove the orthogonality principle (Irelations) {A}, [m] {A} =0 { {A}, [K] {A} =0 for rts (wr + ws) when [m]=[m] + [K] = [K] ([[m], [K] both symmetric). The proof starts with the relations

(1) & (2) are Brained Wr [m] SA] = [K] SA], from [m] {x}+[4{x}={0}] W2 [m] {A} = (K) {A} - (2) with {2} = {A}Sin(wtrg) /x)=-62/43/sin(wtop) Υ=1,2,-,n; Δ=1,2,--, ~ but r ≠ s $\int_{0}^{\infty} -\omega^{2} [m] \{A\} + [\kappa] \{A\} = 0$ Premultiplying both sides of (1) by
[A], we set since sin (Wtop) to at all times. So, w2[m]{A]={K}{A} 02 { 4} [[] { A] = [A] [[[] [A]] , For w=wr, wchove {A}={A}-{A}-" W=Ws, EAJ=1A]s. Premultiplying @ by {A}, weger Thus, as Town [A] = [4] [A] or Ws2 {A}_T[m] {A}_s = {A}_T[w] {A}_s - @. Taxing transpose of 3), we have W, 2 [[A] [[M] [A]] = [[A] [K] [A]] T =) 22 {43, T (m) T (\$43) T) = A], T (x) T (43, T) T ([A] [B][c])T = [c] [a] [A] T = Wr2 {A}_ [m] {A]_ = {A}_ [k] [A] ; (J) (4 (A)T)T assuming [m] & [K) are both symmetric so that [w] = (m) & [w] = [v] Subtracting 3 from 4 we get $(6s^2 - \omega r^2) \{43r^7 [m] \{43s = 0\}$ Here it water, then {A} [m] [A]=0 & This is the mass ofthogonality relation. Then from @ or O, {AZT [W]AZ=0 4 this is the stiffness orthogonality relation. Remember these. Note that

the symmetry Em7 & Ek) are hece many for the orthogonality relations to be true. In certain formulations, [m] for [x] may not be symmetric. So, be a little careful before you presume orthogonality relations are valid for your problem. Another important but not so obvious point is that EAS, I EAS, must correspond to two different natural frequencies, ie. Wy + Ws. There are systems with repeated natural frequencies, for instance, double rost of the frequency equation giving rise to, say, wy = ws. The funny thing is, we can obtain independent vectors {A} of EASs corresponding to these this repeated natural frequency. These may not satisfy the arthogonality relations. (b) We now go back to modal analysis. Presently, we shall consider a damped system.

With linear viscous damping, the DEOM will be of the form:

[m] {xi}+[c]{x}+[k]{x}={F(t)}-(t) [c] = [cq q2 - qn] For example, for the 3-por system shown, $\begin{bmatrix}
c \\
c
\end{bmatrix} = \begin{bmatrix}
c_1 + c_2 & -c_2 & 0 \\
-c_2 & c_2 + c_3 & -c_3 \\
0 & -c_3 & c_3
\end{bmatrix}$ 4344 mi K2 \$ 4)C2 (Check Itis) It can be shown that
the coordinate transformation
the coordinate transformation
\$23= [4] {b}, want uncouple [m2] K37 0163 m3] the DEOM (), where [M] is a modal matrix for the corresponding undamped system [m] {xi}+[k]{x}={0}. This is because after using above transformation and premultiplication by [MT would result in the following " + [M] [M] [M] = [M] [A] + Although [m] (m) [m] = [M] & [m] (k) [m] = [K] would be diagonal, ENTECTEN won't be so. Thus, we don't get unconfled D TOM. - In practice, honever often the off-diagonal teams of [M'Ec](M) are

small & can be neglected. Then the uncoupled DEOM would be: M, P, + C, P, + E, P, = Q, (W $M_{21} \stackrel{b}{>} + C_{22} \stackrel{b}{>} + K_{22} \stackrel{p}{>}_2 = Q_2(E)$ Seach of the above differential equations can be sorred for Steady state oration using Suhamel's integral. This will involve the damped impulse response Shemember $g(t) = \frac{1}{M_{rr}} \frac{-g_r \omega_r t}{\omega_{dr}}$ Shemember $g(t) = \frac{1}{m \omega_d} \frac{-g_r \omega_r t}{e}$ sinust for $\frac{m\ddot{n} + c\ddot{n} + kx = f(b)}{2}$ Here $W_r = W_r \sqrt{1-S_r^2}$, a damped ratural frequency corresponding to Mrr Pr + Crr Pr + Krr Fr = Qr (t); r=12,-02, Also, $S_r = \frac{C_{rr}}{2\sqrt{M_{rr} K_{rr}}}$. (5) Proportional Damping (Rayleigh Damping)a scase of proportional damping. Lepare constants
[M][C][M]= X[M]M -> Diagonal ->

4 [M][M]= X[M][K][M] is also diagonal. The DFOM thus can be unconflect and then solved for steady state response using Duhanel's integral. A special type of proportional downpring is Rayleigh Damping. In this case, it is a soumed that [c] can be expressed as follows: [c]= x[m]+p[K].-4) HW problem Starting with the DEOM [m] {xi}+[c]{xi}+[k]{x}={frm show that when (c) is of the form 1), the FOM Can be uncoupled. END OF VA-6, Part3