



AIRLINI	E CARD NO	FLIGHT CONT	TITLE ROL CABLES - LE	FT MAIN GEAR	BOEING 0 27-226	
DATE	TASK INSPECTION - DETAILED		WELL		RELATE W-32-44	
TAIL NUMBER	WORK AREA L MAIN W/W	VERSION 1.1	THRESHOLD 6600 FC	REPEAT 6600 FC	APPLIC.	ABILITY ENGINE
STATION	SKILL AIRPL	1.2 NOTE	36 MO	36 MO	ALL	ALL
		ACCESS			ZONE 133	

Perform a detail visual inspection of the control cables within the left main landing gear wheel well for broken wires. Check associated pulleys, brackets, and mechanisms for condition and security of installation. The following cables are located in the left MLG wheel well:

- A. Aileron control cables
- B. Spoiler control cables
- C. Speed brake control cables

Note: The control cables must be displaced full travel in each direction for complete inspection at seals, pulleys, and fairlead areas.

INTERVAL NOTE: Whichever occurs first.

A. References

Reference	Title
AMM 12-26-00-600-801	Control Cable Lubrication (P/B 301)

B. Consumable Materials

Reference	Description	Specification
G00034	Cotton Wiper - Process Cleaning Absorbent Wiper	AMS3819 Class 1 Grade A or B
	(Cheesecloth, Gauze)	Form 1 (Supersede BMS15-5 CL
		A)

EFFECTIVITY HAZ ALL	SOURCE MRB	FLIGHT CONTROL CABLES - LEFT MAIN GEAR	R WELL
		D633A109-HAZ 27-226-00-01	Page 1 of 7 Jun 15/2023



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				27-226-00-01

TASK 20-20-31-200-801

MECH INSP

1. Control Cable Wire Rope - Inspection

(Figure 1)

A. Prepare for the Inspection

SUBTASK 20-20-31-100-001



DO NOT APPLY SOLVENTS, GREASE, OR OIL TO STAINLESS STEEL CONTROL CABLES. THESE MATERIALS CAN COLLECT CONTAMINATION THAT CAN CAUSE DAMAGE TO THE INTERNAL SURFACES OF THE CRES CABLE STRANDS. THIS CAN DECREASE THE SERVICE LIFE OF THE CABLE.



DO NOT USE SOLVENT OR HEAT TO THIN GREASE. DO NOT USE SOLVENT TO CLEAN CABLES, SINCE SOLVENT DILUTES AND REMOVES GREASE FROM INSIDE CABLE STRANDS. DO NOT APPLY OR SPRAY BMS 3-23 ON CONTROL CABLES.

- (1) If it is necessary, clean the control cables with lint-free cotton wiper, G00034, that is clean and dry.
 - (a) Remove the old grease and dirt from the surface of the control cable.
 - (b) Clean the control cable for the full length of the cable for the full length of travel through fairleads, air pressure seals, over pulleys, quadrants, and drums.

SUBTASK 20-20-31-200-003

- (2) Perform a detailed visual inspection to make sure that the cable does not contact parts other than pulleys, quadrants, cable seals, or grommets installed to control cable routing.
 - NOTE: The minimum cable clearance from other parts is 0.2 in. (5.1 mm), except 0.1 in. (2.5 mm) within 10 in. (254 mm) of a pulley or quadrant.
 - Look for evidence of contact with other parts. Correct the condition if evidence of contact is found.

B. Control Cable Wire Rope Inspection

SUBTASK 20-20-31-200-015

- (1) Ignore this step if it is not applicable to your work:
 - (a) Make sure that the flight control cables are displaced full travel in each direction for the complete inspection at seals, pulleys, and fairlead areas.

SUBTASK 20-20-31-200-013

- (2) Perform a detailed visual inspection of the cable runs for incorrect routing, kinks in the wire rope, or other damage.
 - (a) Replace the cable assembly if:
 - 1) A wear pattern exists where the individual wires in a strand appear to blend together (outer wires worn by more than 40 percent) (Figure 1).
 - 2) A kink is found.
 - Corrosion is found.

EFFECTIVITY HAZ ALL	SOURCE MRB	FLIGHT CONTROL CABLES - LEFT MAIN GEAR WEL	.L	
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DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C. 27-226		
SUBTASK 20-20-3	1-200-014					MECH	INSP
	m a detailed visual insp	ection of	the cable.				
,	: Most cables are ident thread per MIL-83420	ified by t H. The c	he manufacture condition of the c	r using a color tracer file colored nonmetallic thre or strength of the cable	ads within a		
(a) F	Rub a cloth along the ca	ble to fin	d the broken wi	res.			
• • •	NOTE: The cloth will ca						
(b) F	——— Replace the 7X7 cable a	ssembly	r if:				
. ,	There are two or m	ore brok	en wires in 12 ir	n. (305 mm) of cable.			
	,			here in the total cable a	assembly.		
	Replace the 7X19 cable		•		,		
	•		-	n. (305 mm) of cable.			
	,			ere in the total cable ass	sembly.		
	nspect the carbon steel		-				
` '	·			on the control cable.			
	,	not suffic		k: Control Cable Lubric	ation, AMM		
	NOTE: Do not app cables.	ly the gr	ease or oil to the	e stainless steel (CRES) control		
	——- Е	END OF	TASK ———				
EFFECT HAZ .		SOURCE MRB	FLIGHT CONTR	ROL CABLES - LEFT MAI	N GEAR WEL	_L	
			D633A109-HAZ 27-226-00-01			Page 3 un 15/	



	DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 27-226-00-01									
								21-220	1	
				200-802					MECH	INSP
2.	Insp	ectio	n of th	e Control Cable Fitting	gs					
	A.	Pro	cedure							
		SUBTA	NSK 20-20-3	31-200-007						
		(1)		rm a detailed visual insp (wire locking, cotter pin			the means of locking the	ne joints are		
			(a) I	Install any missing parts	S.					
		SUBTA	NSK 20-20-3	31-200-008						
		(2)		rm a detailed inspection s or corrosion.	of the sv	waged portions	of swaged end fittings fo	or surface		
			(a) I	Replace the cable asse	mbly if cr	acks or corrosic	n are found.			
		SUBTA	NSK 20-20-3	31-200-009						
		(3)	Perfor	rm a detailed visual insp	ection of	f the unswaged	portion of the end fitting	J.		
				Replace the cable asse fitting is bent more than			f corrosion is present, c	or if the end		
		SUBTA	NSK 20-20-3	31-200-010						
		(4)	Perfor	rm a detailed visual insp	ection of	f the turnbuckle.				
			(a) I	Replace the turnbuckle	if a crack	c is visible or if c	orrosion is present.			
					END OF	TASK ———				
				<u>.</u>						
			HAZ		SOURCE MRB	FLIGHT CONTE	ROL CABLES - LEFT MA	IN GEAR WEI	LL	
						D633A109-HAZ			Page 4	



	С	DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C		
								27-226	_	
				200-805					MECH	INSP
3.	Insp	ectio	n of Pu	<u>ılleys</u>						
	Α.	Pro	cedure							
		SUBTA	ASK 20-20-3	1-200-011						
		(1)	Perfor	m a detailed visual ins	pection to	make sure that	pulleys are free to rota	ıte.		
			(a) F	Replace pulleys which	are not fro	ee to rotate.				
		SUBTA	ASK 20-20-3	1-200-012						
		(2)	Perfor	m a detailed visual ins	pection o	f the pulleys for	conditions shown in (Fig	gure 2).		
			(a) F	Replace pulleys which	are not in	a normal condi	tion.			
					END OF	TASK ———				
					LIID OI	7.01.				
			EFFECT HAZ	rivity ALL	SOURCE MRB	FLIGHT CONTR	ROL CABLES - LEFT MA	IN GEAR WE	LL	
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						D633A109-HAZ			Page 5	of 7
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DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CARD NO. 27-226-00-01
WIRE		CABLE STR. (EXAMPLE)		·*************************************	
	7x7 CABLE		EACH OU (W	JTER WIRE WORN LESS TI ORN AREAS NOT BLENDE	HAN 40% D)
CABLE STRAND (EXAMPLE)		3	EAC! (WC	H OUTER WIRE WORN 40-5 ORN AREAS ARE BLENDED	0% 0))
(19 WIRES)	7x19 CABLE			1	
			EACHW	VIRE IS WORN MORE THAN	1 50%
FXAMP	PLE OF INTERNAL WEA	⊾R	STRAM WIRES	OUTER WIRE WEAR AREA C ND. A VISIBLE SPACE BETV S 1 OR A FULLY BLEND ACE. 2	VEEN
1 VISIBLE SF 2 WEAR COI	PACE BETWEEN WIRES NDITION RESULTING IN SURFACES BETWEEN	S. WIRES. Cable	ONLY CABLE CABLE	E WEAR MAY OCCUR ON O OR ON FULL CIRCUMFERE E WEAR CAN EXTEND ALOI E FOR A DISTANCE EQUAL E TRAVEL.	NCE. NG THE
EFFECTI HAZ A		SOURCE MRB	FLIGHT CONTI	ROL CABLES - LEFT MAII	N GEAR WELL
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CABLE TENSION TOO HIGH PULLEY NOT ALIGNED CORRECTLY PULLEY GROOVE WITH EXCESSIVE WEAR PULLEY WILL NOT TURN NORMAL CONDITION PULLEY WILL NOT TURN NORMAL CONDITION F22-724 500000662077_N2 PUBLEY WEAR PATTERNS FIGURE 2 EXERCITARITY HAZ ALL BOARCE MRB FLIGHT CONTROL CABLES - LEFT MAIN GEAR WELL D633A109-HAZ Page 7 of 7	DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CARD NO. 27-226-00-01
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HAZ ALL MRB D633A109-HAZ Page 7 of 7			F	Figure 2		
	EFFE HA	ECTIVITY Z ALL		FLIGHT CONTI	ROL CABLES - LEFT MAI	N GEAR WELL
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AIRLINE	AIRLINE CARD NO		LEFT AND RIGHT MAIN LANDING GEAR MANUAL			ARD NO. -00-01
DATE	TASK FUNCTIONAL		EXTENSION SYSTE	RELATEI	D CARD	
TAIL NUMBER	WORK AREA CREW CABIN	VERSION 1.1	THRESHOLD 36 MO	REPEAT 36 MO	APPLICA AIRPLANE	ABILITY
STATION	SKILL AIRPL				ALL	ALL
		ACCESS S2122			ZONE 212 734 744	

Perform a functional check of the left and right main landing gear manual extension system and alternate extension bypass valve.

A. References

Reference	Title
AMM 07-11-01-580-815	Lift the Airplane with the Jacks (P/B 201)
AMM 07-11-01-580-816	Lower the Airplane Off the Jacks (P/B 201)
AMM 24-22-00-860-812	Remove Electrical Power (P/B 201)
AMM 29-11-00-860-801	Hydraulic System A or B Pressurization (P/B 201)
AMM 29-11-00-860-805	Hydraulic System A or B Power Removal (P/B 201)
AMM 32-00-01-080-801	Landing Gear Downlock Pins Removal (P/B 201)
AMM 32-00-01-480-801	Landing Gear Downlock Pins Installation (P/B 201)
AMM 32-09-10-740-801	Proximity Switch Electronics Unit (PSEU) BITE Test - Ground Test (P/B 501)

B. Tools/Equipment

NOTE: When more than one tool part number is listed under the same "Reference" number, the tools shown are alternates to each other within the same airplane series. Tool part numbers that are replaced or non-procurable are preceded by "Opt:", which stands for Optional.

Reference	Description
STD-1184	Scale - Spring, 0-100 Lbs, Tension

EFFECTIVITY HAZ ALL	SOURCE MRB	LEFT AND RIGHT MAIN LANDING GEAR SYSTEM	MANUAL EXTENSION
		D633A109-HAZ 32-220-00-01	Page 1 of 9 Feb 15/2024



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				32-220-00-01

TASK 32-34-00-730-801

MECH INSP

1. Main Gear Manual Extension System Test - Airplane on Jacks

(Figure 1)

A. General

(1) The normal extension/retraction system will not operate with the cover for the alternate extend T handle open; make sure that the cover is closed when you will extend or retract the landing gear with the landing gear control handle.

B. Prepare for the Test

SUBTASK 32-34-00-480-002



MAKE SURE THE DOWNLOCK PINS ARE INSTALLED ON ALL THE LANDING GEAR. WITHOUT THE DOWNLOCK PINS, THE LANDING GEAR CAN RETRACT AND CAUSE INJURIES TO PERSONS AND DAMAGE TO EQUIPMENT.

(1) Make sure that the downlock pins are installed in the nose and main landing gear, do this task: Landing Gear Downlock Pins Installation, AMM TASK 32-00-01-480-801.

SUBTASK 32-34-00-580-004

(2) Do this task: Lift the Airplane with the Jacks, AMM TASK 07-11-01-580-815.

SUBTASK 32-34-00-860-005

(3) For hydraulic system A, do this task: Hydraulic System A or B Pressurization, AMM TASK 29-11-00-860-801.

SUBTASK 32-34-00-860-006

(4) Make sure that these circuit breakers are closed:

CAPT Electrical System Panel, P18-2

Row	<u>Col</u>	<u>Number</u>	<u>Name</u>
В	3	C01312	ENGINE 1 RUN/PWR

F/O Electrical System Panel, P6-3

<u>Row</u>	<u>Col</u>	<u>Number</u>	<u>Name</u>
В	17	C00129	LANDING GEAR LATCH & PRESS WARN
С	15	C01355	LANDING GEAR AIR/GND SYS 2
С	16	C01356	LANDING GEAR AIR/GND SYS 1
D	1	C01399	PSEU PRI
D	2	C01400	PSEU ALTN
D	16	C01432	LANDING GEAR ALTN EXTEND SOL
Е	12	C00314	INDICATOR MASTER DIM SECT 2
F	11	C00317	INDICATOR MASTER DIM SECT 5
F	13	C01179	INDICATOR MASTER DIM SECT 7

EFFECTIVITY HAZ ALL	SOURCE MRB	LEFT AND RIGHT MAIN LANDING GEAR MANUAL EX	KTENS	ION
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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				32-220-00-01

SUBTASK 32-34-00-860-008

MECH INSP

(5) Open these circuit breakers and install safety tags:

F/O Electrical System Panel, P6-1

Row	Col	<u>Number</u>	<u>Name</u>
D	13	C00120	WEATHER RADAR RT

F/O Electrical System Panel, P6-2

Row	<u>Col</u>	<u>Number</u>	<u>Name</u>
С	15	C00799	HYD SYS LDG GR SYS XFR VALVE SEC
С	16	C00781	HYD SYS LDG GR SYS XFR VALVE PRI

F/O Electrical System Panel, P6-3

<u>Row</u>	Col	<u>Number</u>	<u>Name</u>
D	18	C00451	LANDING GEAR AURAL WARN

SUBTASK 32-34-00-860-010

(6) Make sure that the control lever for the landing gear is in the DN position.

SUBTASK 32-34-00-860-025



OBEY THE INSTRUCTIONS IN THE PROCEDURE TO PUT THE SPEEDBRAKE HANDLE TO THE DOWN POSITION. IF YOU DO NOT OBEY THE INSTRUCTIONS, INJURIES TO PERSONNEL, AND DAMAGE TO EQUIPMENT CAN OCCUR.

(7) Put the speedbrake handle to the down position.

SUBTASK 32-34-00-860-011

(8) Move the No. 1 and No. 2 throttle levers to the full forward position.

C. Main Gear Manual Extension System Test

SUBTASK 32-34-00-860-012

(1) Put the control lever in the OFF position.

SUBTASK 32-34-00-700-002

- (2) Pull the manual extension handle for the left main gear.
 - (a) Make sure that the full travel of the T handle is at least 18.5 in. (469.9 mm).

SUBTASK 32-34-00-700-003

- (3) Release the manual extension handle for the left main gear.
 - (a) Make sure that the handle returns to the stowed position without hesitation or binding.

SUBTASK 32-34-00-480-003

(4) Attach a spring scale (0-100 Lbs), STD-1184, to the manual extension T handle for the left main gear.

SUBTASK 32-34-00-730-001

(5) Pull the manual extension handle for the left main gear.

EFFECTIVITY HAZ ALL	SOURCE MRB	LEFT AND RIGHT MAIN LANDING GEAR MANUAL EX	(TENS	ION
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737-600/700/800/900 **TASK CARDS**

DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 32-220	
	(a)	 Make sure that the loa	 ıd to start t	he T handle fro	⊥ m the seat has a minim	um of 3.0 lbf	MECH IN
	(-)	(13.3 N).					
	(b)	Make sure that the load 18.0 lbf (80.1 N).	d during th	ne full travel of t	he T handle is not more	e than	
			simulate a	person seated	d forward when you me in the First Officer's se		
	(c)	Make sure that the ma	anual exten	sion mechanisr	m operates freely.		
SUBTA	NSK 32-3	4-00-730-011					
(6)	Pull	the manual extension h	andle for t	he right main ge	ear.		
	(a)	Make sure that the full	travel of the	he T handle is a	at least 18.5 in. (469.9 n	nm).	
SUBTA	NSK 32-34	4-00-730-012					
(7)	Rele	ease the manual extens	ion handle	for the right ma	ain gear.		
	(a)	Make sure that the habinding.	ndle return	s to the stowed	position without hesita	tion or	
SUBTA	SK 32-3	4-00-860-038					
(8)	Rem	nove the spring scale from	om the mai	nual extension ⁻	T handle for the left mai	n gear.	
SUBTA	SUBTASK 32-34-00-480-012						
(9)	Atta	ch a spring scale to the	manual ex	tension T hand	le for the right main gea	ar.	
SUBTA		4-00-730-013					
(10)	Pull	the manual extension h	andle for t	he right main ge	ear.		
	(a)	Make sure that the loa (13.3 N).	id to start t	he T handle froi	m the seat has a minim	um of 3.0 lbf	
	(b)	Make sure that the load 18.0 lbf (80.1 N).	ıd during th	ne full travel of t	he T handle is not more	e than	
			simulate a	person seated	d forward when you me in the First Officer's sea		
	(c)	Make sure that the ma	anual exten	sion mechanisr	m operates freely.		
SUBTA	NSK 32-34	4-00-080-004					
(11)		nove the downlock pin for Inlock Pins Removal, Al			gears, do this task: La 01.	nding Gear	
SUBTA	SK 32-3	4-00-860-048					
(12)	Rem	nove the spring scale fro	om the mar	nual extension ⁻	T handle for the right ma	ain gear.	
SUBTA	SK 32-3	4-00-860-019					
(13)	Mak	e sure that the cover fo	r the T han	dle is closed.			
	NOT	E: The extension/retra extend T handle op	-	em will not opera	ate with the cover for th	e alternate	
SUBTA	SK 32-3	4-00-860-017					
(14)	Mov	e the control lever for th	ne landing	gear to the DN	position.		
		ECTIVITY Z ALL	source MRB	LEFT AND RIG SYSTEM	HT MAIN LANDING GEA	R MANUAL E	XTENSIC
				D633A109-HAZ 32-220-00-01	2		Page 4 o



737-600/700/800/900 TASK CARDS

DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 32-220-00-01

SUBTASK 32-34-00-730-003

MECH INSP



MAKE SURE THAT ALL PERSONS AND EQUIPMENT ARE CLEAR OF THE MAIN LANDING GEAR. FAST MOVEMENT OF THE MAIN LANDING GEAR CAN CAUSE INJURIES TO PERSONS AND DAMAGE TO EQUIPMENT.

(15) Move the control lever for the landing gear to UP and wait while the main gear retracts.

SUBTASK 32-34-00-860-026

(16) Open the access door to the manual extension T handles.

SUBTASK 32-34-00-730-007

(17) Pull the manual extension handle of the left main gear.

NOTE: The T handle must be pulled upward and forward. This will simulate a person seated in the First Officer's seat. Do not pull the handle straight up.

SUBTASK 32-34-00-700-011

- (18) Make sure that the left main gear free falls to the down and locked position.
 - (a) Make sure that the green light for the left main gear is on.
 - (b) Make sure that the red light for the left main gear is on.

NOTE: The red light for the main gear will stay on to show that the landing gear control lever and landing gear positions disagree.

SUBTASK 32-34-00-700-012

(19) Release the manual extension handle for the left main gear.

SUBTASK 32-34-00-730-014

(20) Pull the manual extension handle of the right main gear.

NOTE: The T handle must be pulled upward and forward. This will simulate a person seated in the First Officer's seat. Do not pull the handle straight up.

SUBTASK 32-34-00-700-033

- (21) Make sure that the right main gear free falls to the down and locked position.
 - (a) Make sure that the green light for the right main gear is on.
 - (b) Make sure that the red light for the right main gear is on.

<u>NOTE</u>: The red light for the main gear will stay on to show that the landing gear control lever and landing gear positions disagree.

SUBTASK 32-34-00-730-015

(22) Release the manual extension handle for the right main gear.

SUBTASK 32-34-00-860-029

- (23) Move the control lever for the landing gear to the DOWN position.
 - (a) Make sure that the red light for the main gear is off.

SUBTASK 32-34-00-860-037

(24) Make sure that the cover for the T-handle is closed.

SUBTASK 32-34-00-860-027

(25) Move the control lever for the landing gear to the OFF position.

EFFECTIVITY HAZ ALL	SOURCE MRB	LEFT AND RIGHT MAIN LANDING GEAR MANUSYSTEM	JAL EXTENSION
		D633A109-HAZ 32-220-00-01	Page 5 of 9 Jun 15/2023



737-600/700/800/900 **TASK CARDS**

DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C. 32-220-		
							MECH	INSF
(26)		4-00-730-009 to the control lever for the	he landing (near to LIP and	wait while the main gea	r retracts		
` ,		4-00-860-024	no landing ;	gear to or and	wait wille the main gee	ii rotidoto.		
(27)		the landing gear contro	l lever in th	e OFF position				
` ,		4-00-860-030			-			
(28)		n the access door to th	e manual e	xtension T han	dles.			
SUBTA	ASK 32-3	4-00-700-015						
(29)	(29) Attach a spring scale (0-100 Lbs), STD-1184, to the manual extension T handle for the left main gear.							
SUBTA	ASK 32-3	4-00-730-004						
(30)	Pull	the manual extension h	nandle for t	ne left main gea	ar.			
(a) Make sure that the maximum load when you pull the T handle is not more than 50.0 lbf (222.4 N).								
	NOTE: The T handle must be pulled upward and forward when you measure the load. This will simulate a person seated in the First Officer's seat. Do not pull the handle straight up.							
SUBTA	ASK 32-3	4-00-700-006						
(31)	Mak	e sure that the left mair	n gear free	falls to the dow	n and locked position.			
(a) Make sure that the green light for the left main gear is on.								
(b) Make sure that the red light for the left main gear is on.								
				n gear will stay gear positions	on to show that the land disagree.	ling gear		
SUBTA	ASK 32-3	4-00-080-005						
(32)	Rem	nove the spring scale from	om the mar	nual extension	T handles for the left ma	in gear.		
SUBTA	ASK 32-3	4-00-730-017						
(33)	Atta	ch a spring scale to the	manual ex	tension T hand	lle for the right main gea	ır.		
SUBTA		4-00-860-050						
(34)	Pull	the manual extension h						
	(a)	Make sure that the ma 50.0 lbf (222.4 N).	aximum loa	d when you pu	ll the T handle is not mo	re than		
			simulate a	person seated	nd forward when you me in the First Officer's sea			
SUBTA	ASK 32-3	4-00-860-051						
(35)	Mak	e sure that the right ma	in gear free	e falls to the do	wn and locked position.			
	(a)	Make sure that the gre	een light fo	the right main	gear is on.			
	(b)	Make sure that the red	d light for th	ne right main ge	ear is on.			
				n gear will stay gear positions	on to show that the land disagree.	ling gear		
		ECTIVITY Z ALL	SOURCE MRB	LEFT AND RIG SYSTEM	HT MAIN LANDING GEA	R MANUAL EX	KTENS	SIOI
				D633A109-HAZ	Z		Page 6 un 15/	



737-600/700/800/900 **TASK CARDS**

	DATE		1	AIL NUMBER		STATION	AIRLINE CARD NO. BOEING 32-220			
	SUBTA	SK 32-34-0	0-860-052		'			-	MECH	INS
	(36)	Move	the cont	rol lever for t	the landing	gear to the DOV	VN position.			
		(a) N	/lake su	re that the re	d lights for	the right and lef	t main gear are off.			
	SUBTA	SK 32-34-0	0-860-053							
	(37)	Remo	ve the s	pring scale fr	om the ma	nual extension T	handles for the right r	main gear.		
		SK 32-34-0								
	(38)					idles is closed.				
D. Put the Airplane Back to its Usual Cond				ack to its Us	sual Condi	tion				
	suвтаsк 32-34-00-860-015 (1) For Hydraulic System A, do this task: Hydraulic System A or B Power Removal, AMM			/al ΔMM						
	(1)			0-860-805.	J tilis task.	Tydraulic Gystei	II A OI D I OWEI NEIIIO	vai, Aiviivi		
	SUBTA	SK 32-34-0	0-480-005							
	(2)						ed, do this task: Landi	ng Gear		
		Downl	ock Pins	s Installation,	, AMM TASI	< 32-00-01-480-	801.			
		SK 32-34-0		rouit brooker						
	(3)			rcuit breaker						
		F/O E Row	lectrical Col	System Pa Number	nel, P6-2 <u>Name</u>					
C 15 C00799 HYD SYS LDG GR SYS XFR VALVE SEC										
C 16 C00781 HYD SYS LDG GR SYS XFR VALVE PRI										
		E/O E	lootriool	System Da	nol D6 2					
		Row	ectricai Col	System Pa Number	Name					
		D	18	C00451		GEAR AURAL	WARN			
	SUBTA	SK 32-34-0	0-860-021							
	(4)	Move	the No.	1 and No. 2	throttle leve	rs back to the id	lle position (full aft).			
	SUBTA	SK 32-34-0	0-840-004							
	(5)	Close	this circ	uit breaker:						
				System Pa						
		Row	<u>Col</u>	'	Name					
		D	13	C00120	WEATHE	R RADAR RT				
	(6)	SK 32-34-0		wer is not no	ocesany de	this task: Pomo	ove Electrical Power, A	N / N / /		
	(0)			wei is not ne 0-860-812.	cessary, uc	tilis task. Keilic	ove Electrical Fower, A	IIVIIVI		
	SUBTA	SK 32-34-0	0-580-003							
	(7)	Do thi	s task: L	ower the Air	plane Off th	e Jacks, AMM T	ASK 07-11-01-580-816	6.		
		HAZ /			SOURCE MRB	LEFT AND RIGI SYSTEM	HT MAIN LANDING GEA	AR MANUAL E	XTENS	SIC
						D633A109-HAZ			Page 7	7 ^
						32-220-00-01			un 15/	

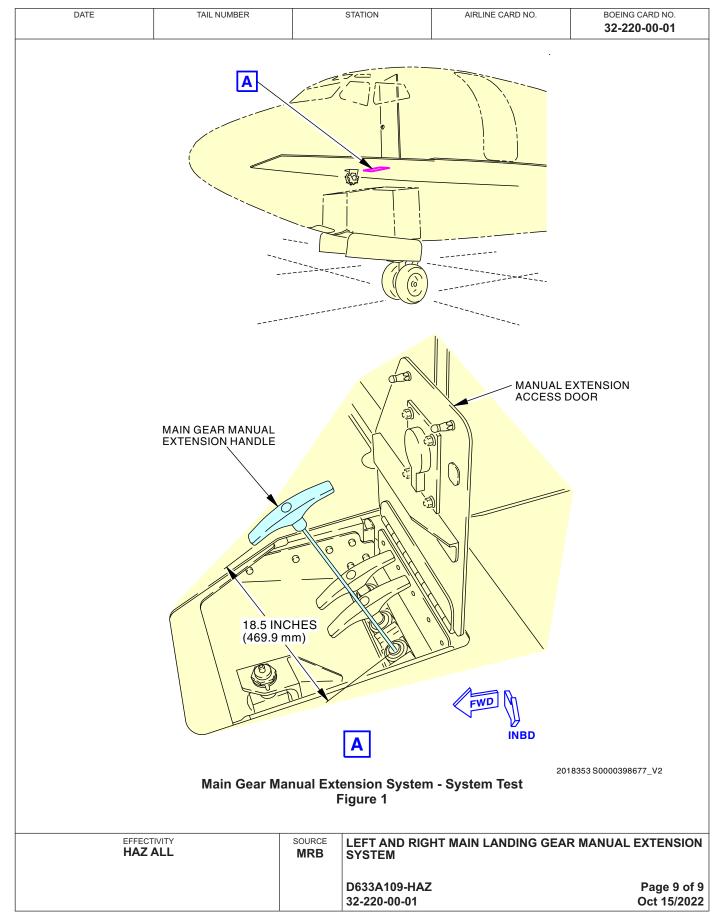




DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING 0 32-220		
SUBTASK 32-34-0	0-200-002		I			MECH	INSP
(8) Check this tas	the Proximity Switch	Electronic lectronics	Unit (PSEU) for Unit (PSEU) BI	stored faults using the TE Test - Ground Test, a	self test, do AMM		
		END OF	TASK ———				
EFFECT HAZ	IVITY ALL	SOURCE MRB	LEFT AND RIGH SYSTEM	HT MAIN LANDING GEA	R MANUAL E	XTENS	SION
			D633A109-HAZ 32-220-00-01		J	Page 8 un 15/	of 9 2023











AIRLINE	AIRLINE CARD NO		LEFT MAIN LANDING GEAR AND LANDING GEAR			BOEING CARD NO. 32-806-01-01	
DATE	TASK ZONAL (GV)		DOORS		RELATED CARD		
TAIL NUMBER	WORK AREA L MAIN GEAR	VERSION 1.1			APPLICABILITY AIRPLANE ENGINE		
STATION	SKILL AIRPL				ALL	ALL	
		ACCESS			ZONE 730		

Perform an external zonal inspection (GV) of the left main landing gear and landing gear doors. (EZAP)

INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 5500 FC/30 MO is satisfied by this zonal inspection.

A. References

Reference	Title
AMM 05-00-00-910-804	Enhanced Zonal Inspection Program (EZAP) Precautions (P/B 201)
AMM 20-60-07-913-801	Protection of the EWIS During Maintenance (P/B 201)

EFFECTIVITY HAZ ALL	SOURCE MRB	LEFT MAIN LANDING GEAR AND LAN	IDING GEAR DOORS
		D633A109-HAZ 32-806-01-01	Page 1 of 4 Feb 15/2022

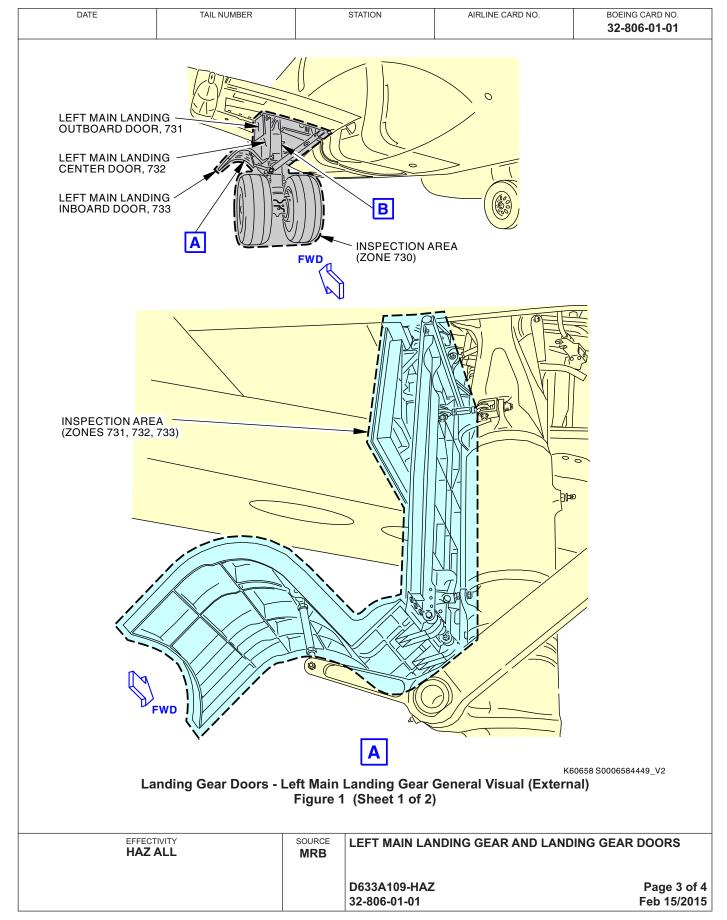




	Г	ATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 32-806		
1.	EXT	K 05- ERN/	AL - ZO	210-803 NAL (GV): Left Main l	Landing	Gear and Land	ing Gear Doors		MECH	INSF
		ure 1)								
	A.	Gen (1)	This Z	onal inspection proced dure (EZAP) - derived i			Enhanced Zonal Analy nent for this zone.	sis		
	B. Zonal Inspection									
	(1) Do the zonal inspection following the procedures in Enhanced Zonal Inspection Program (EZAP) Precautions, AMM TASK 05-00-00-910-804. SUBTASK 05-41-07-210-007 (2) Perform an external zonal inspection (GV) of the left main landing gear and landing gear doors (EZAP). SUBTASK 05-41-07-910-002 (3) Refer to Protection of the EWIS During Maintenance, AMM TASK 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to Electrical Wiring Interconnect Systems (EWIS) during maintenance.									
					END OF	TASK —	, -			
			EFFECT HAZ		SOURCE MRB	LEFT MAIN LAI	NDING GEAR AND LAND	DING GEAR D	OORS	
						D633A109-HAZ 32-806-01-01			Page 2 eb 15/	











STATION DATE TAIL NUMBER AIRLINE CARD NO. BOEING CARD NO. 32-806-01-01 INSPECTION AREA (ZONE 734) INBD K62597 S0006584450_V2 Landing Gear Doors - Left Main Landing Gear General Visual (External) Figure 1 (Sheet 2 of 2) EFFECTIVITY SOURCE LEFT MAIN LANDING GEAR AND LANDING GEAR DOORS HAZ ALL **MRB** D633A109-HAZ Page 4 of 4

32-806-01-01

Feb 15/2015





AIRLINE	CARD NO	RIGHT MAIN LANDING GEAR AND LANDING			BOEING CARD NO. 32-810-02-01	
DATE	TASK ZONAL (GV)		GEAR DOORS	RELATED CARD		
TAIL NUMBER		VERSION 1.1	THRESHOLD 5500 FC	REPEAT 5500 FC	APPLICABILITY AIRPLANE ENG	CABILITY
STATION	SKILL AIRPL	1.2 NOTE	30 MO	30 MO	ALL	ALL
		ACCESS			ZONE 740	

Perform an external zonal inspection (GV) of the right main landing gear and landing gear doors. (EZAP)

INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 5500 FC/30 MO is satisfied by this zonal inspection.

A. References

Reference	Title
AMM 05-00-00-910-804	Enhanced Zonal Inspection Program (EZAP) Precautions (P/B 201)
AMM 20-60-07-913-801	Protection of the EWIS During Maintenance (P/B 201)

EFFECTIVITY	SOURCE	RIGHT MAIN LANDING GEAR AND LANDING	GEAR DOORS
HAZ ALL	MRB		
		D633A109-HAZ 32-810-02-01	Page 1 of 4 Feb 15/2022

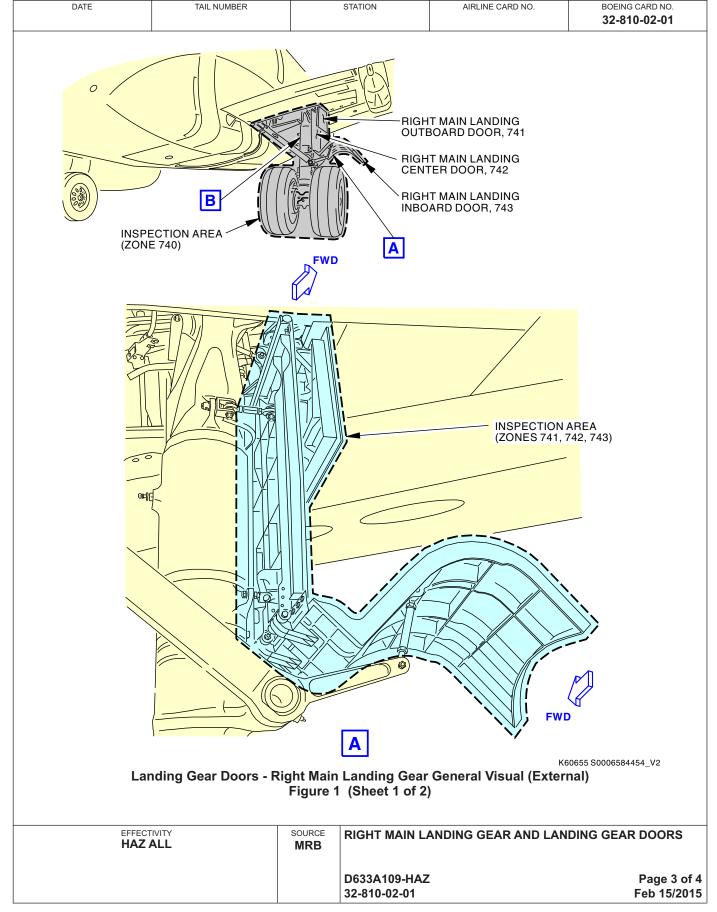




	DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 32-810-02-01													
	EWI	S			_				MECH	INSP				
	TAS	K 05-	41-07-2	210-805										
1.	EXT	ERN	AL - ZO	NAL (GV): Right Mair	Landing	g Gear and Lan	ding Gear Doors							
	(Fig	ure 1)												
	A.	Gen	eral											
		(1)		onal inspection proced dure (EZAP) - derived			Enhanced Zonal Analy nent for this zone.	sis						
	В.	Zon	al Insp	ection										
		SUBTA	JBTASK 05-41-07-210-005											
	(1) Do the zonal inspection following the procedures in Enhanced Zonal Inspection Program (EZAP) Precautions, AMM TASK 05-00-00-910-804.													
		SUBTA	NSK 05-41-0	7-210-008										
	(2) Perform an external zonal inspection (GV) of the right main landing gear and landing gear doors (EZAP).						r							
		SUBTA	NSK 05-41-0	7-910-003										
		(3)	the pro	otection and caution in	formation	that will minimiz	AMM TASK 20-60-07-9 ze contamination and a VIS) during maintenand	ccidental						
	——— END OF TASK ———													
END OF TACK														
			EFFECT	TIVITY	SOURCE	RIGHT MAIN I	ANDING GEAR AND LAN	IDING GEAD	DOOP	<u> </u>				
			HAZ		MRB	NIGHT WAIN LA	ANDING GEAR AND LAN	DING GEAR	DOOK	.5				
						D633A109-HAZ 32-810-02-01	:	ŀ	Page 2 Feb 15/					











STATION DATE TAIL NUMBER AIRLINE CARD NO. BOEING CARD NO. 32-810-02-01 INSPECTION AREA (ZONE 744) K62599 S0006584455_V2 Landing Gear Doors - Right Main Landing Gear General Visual (External) Figure 1 (Sheet 2 of 2) EFFECTIVITY HAZ ALL SOURCE RIGHT MAIN LANDING GEAR AND LANDING GEAR DOORS **MRB** D633A109-HAZ Page 4 of 4 Feb 15/2015 32-810-02-01





AIRLIN	E CARD NO	E/E AC	TITLE CESS DOOR LUBR	ICATION	BOEING CARD NO. 52-120-00-01	
DATE	TASK LUBRICATE				RELATE	D CARD
TAIL NUMBER	WORK AREA E/E COMPARTMENT	VERSION 1.1	THRESHOLD 2 YR	REPEAT 2 YR	APPLICA AIRPLANE	ABILITY ENGINE
STATION	SKILL AIRPL				ALL	ALL
		ACCESS 117A			ZONE 117 118	

Lubricate the E/E access door handle latching mechanism (rack and pinion gear and the lock pins).

A. Consumable Materials

Reference	Description	Specification
D00633	Grease - Aircraft General Purpose	BMS3-33

EFFECTIVITY HAZ ALL	SOURCE MRB	E/E ACCESS DOOR LUBRICATION	
		D633A109-HAZ 52-120-00-01	Page 1 of 7 Jun 15/2022



737-600/700/800/900 TASK CARDS

DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				52-120-00-01

TASK 12-25-41-640-801

MECH INSP

1. Electronic Equipment Access Door Servicing

(Figure 1, Table 1)

A. Prepare for Servicing

SUBTASK 12-25-41-010-005

(1) Turn the latch handle to the closed position.

SUBTASK 12-25-41-010-007

(2) Open this access panel:

Number Name/Location

117A Electronic Equipment Access Door

SUBTASK 12-25-41-010-002

(3) Do the following to open this access panel:

Number Name/Location

117AW Equipment Access Door Cover

(Figure 1)

- (a) Remove the bolt [105], washer [106], and nut [107] that attach the collar [104] to the latch mechanism [2].
- (b) Remove the collar [104] and the washer [108].
- (c) Remove the screws [101] and the screws [102] that attach the cover [103] to the door.
- (d) Remove the cover [103].

SUBTASK 12-25-41-010-006

- (4) Remove the support plate [110] from the latch mechanism [2] as follows (Figure 1):
 - (a) Remove the screws [109] that attach the support plate [110] to the latch mechanism [2].

NOTE: After you remove the screws [109], the bearings [113] and the spacers [112, 114, 115, 116 and 117] are not held in position.

(b) Remove the support plate [110] and the washer [111].

B. Procedure

SUBTASK 12-25-41-640-001

- (1) Lubricate the components with the applicable material shown in (Table 1, Figure 1):
 - (a) grease, D00633

Table 1 Electronic Equipment Access Door Lubrication (Fig. 301)

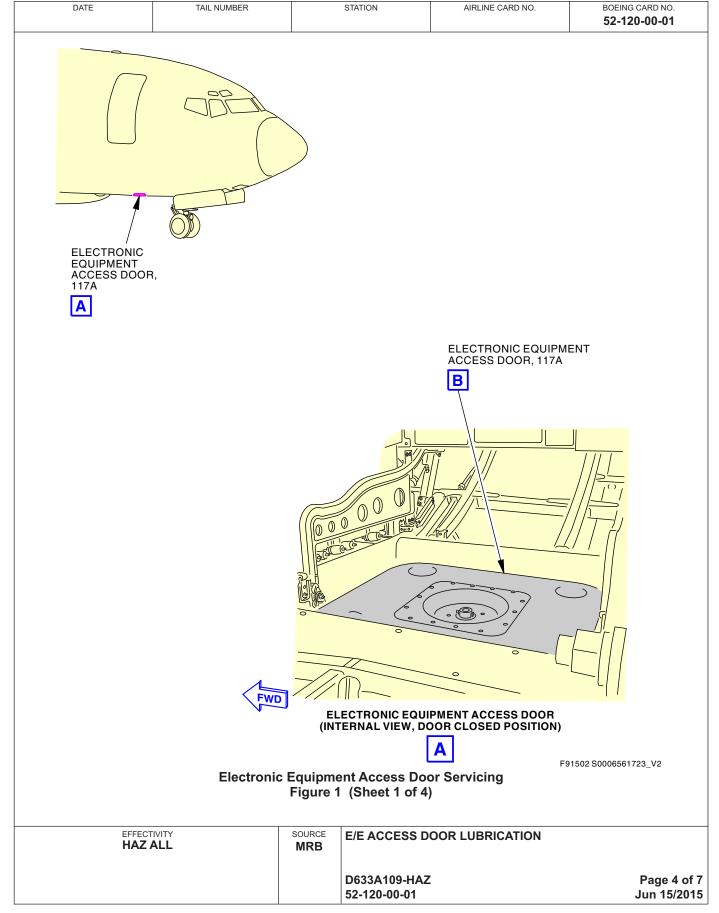
Item No.	Nomenclature	Lubricant	Method of Application	Number of Points
1	Latch Pins (4)	grease, D00633	Hand	4
2	Latch Mechanism (1)	grease, D00633	Hand	1

EFFECTIVITY HAZ ALL	SOURCE MRB	E/E ACCESS DOOR LUBRICATION		
		D633A109-HAZ 52-120-00-01	Page 2 Jun 15/	



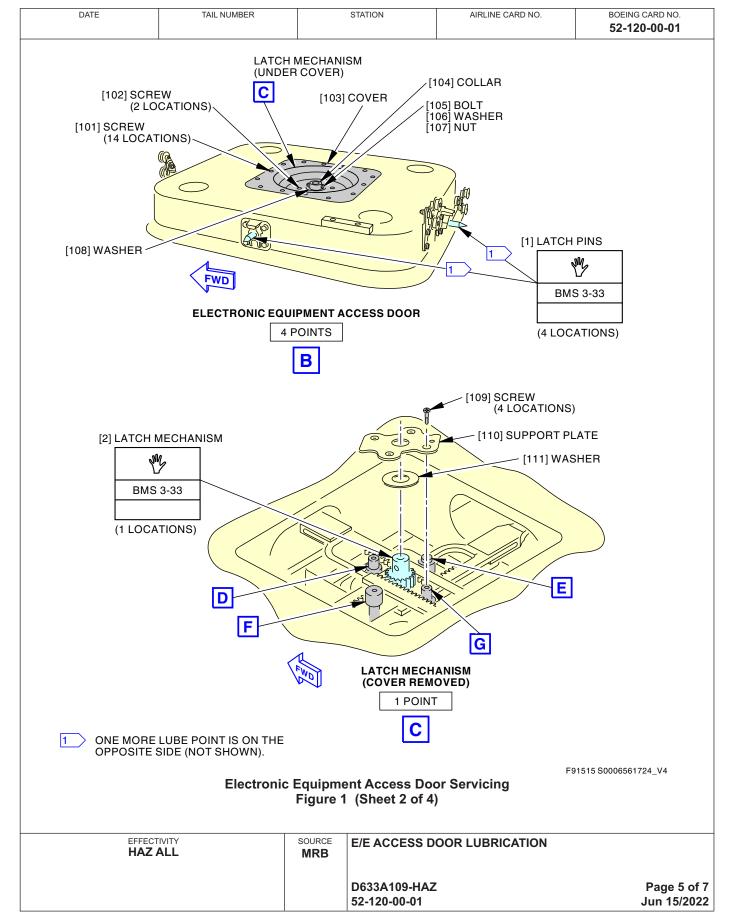
	DATE			TAIL NUMBER STATION AIRLINE CARD NO. BOEING C. 52-120-			CARD NO. 0-00-01				
C.	Put	Put the Airplane Back to Its Usual Condition						1	MECH I		
	SUBTASK 12-25-41-410-003										
	(1)	Insta	II the su	upport plate [110]	on the la	tch mechanism	[2] as follows (Figure 1):			
		(a)	Install	the washer [111]	on the lat	tch mechanism	[2].				
		(b)		e support plate [1 nd 117] and the be			over the spacers [112,	114, 115,			
		(c)		the screws [109].		•					
	SUBTA	` ,	i-41-010-003								
	(2)			wing to close this	access p	anel:					
	` '	Num		Name/Location							
		117A	W	Equipment Acce	ess Door	Cover					
	(Figure 1): (a) Put the cover [103] in its correct position over the latch mechanism [2].										
		(b)				•	ne latch mechanism [2].				
		(c)	Install				to attach the collar [104	4] to the			
		(d)			and scre	ws [102] to attac	ch the cover [103] to the	door.			
	SUBTA	` ,									
	SUBTASK 12-25-41-410-004 (3) Close this access panel:										
	Number Name/Location										
		117A	١	Electronic Equip	oment Ac	cess Door					
	——— END OF TASK ———										
			CTIVITY		SOURCE	E/E ACCESS D	OOR LUBRICATION				
			CTIVITY Z ALL		SOURCE MRB	E/E ACCESS DO			Page 3		















DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 52-120-00-01 [109] SCREW [109] SCREW [110] SUPPORTING PLATE [110] SUPPORTING PLATE \Box (0)[114] SPACER [112] SPACER [113] BEARING - [113] BEARING \odot - [115] SPACER 000 2325885 S0000527595_V2 **Electronic Equipment Access Door Servicing** Figure 1 (Sheet 3 of 4) EFFECTIVITY SOURCE **E/E ACCESS DOOR LUBRICATION HAZ ALL MRB** D633A109-HAZ Page 6 of 7 Jun 15/2015 52-120-00-01





DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CARD NO. 52-120-00-01
	[110] SUP PLA	SCREW PORTING TE SPACER BEARING SPACER		[110] SUPPORTING PLATE [113] BEARING [112] SPACER	52-120-00-01
	Electronic	c Equipme Figure ′	ent Access Doc 1 (Sheet 4 of 4)	or Servicing	
EFF HA	ECTIVITY AZ ALL	SOURCE MRB	E/E ACCESS DO	OOR LUBRICATION	
			D633A109-HAZ 52-120-00-01		Page 7 of 7 Jun 15/2015