



AIRLINI	AIRLINE CARD NO		TITLE FLIGHT CONTROL CABLES - LEFT MAIN GEAR			CARD NO. 6-00-01	
DATE	TASK INSPECTION - DETAILED		WELL		RELATED CARD W-32-440-00-02		
TAIL NUMBER	WORK AREA L MAIN W/W	VERSION 1.1	THRESHOLD 6600 FC	REPEAT 6600 FC	APPLIC AIRPLANE	ABILITY ENGINE	
STATION	SKILL AIRPL	1.2 NOTE	36 MO 36 MO		ALL	ALL	
		ACCESS			ZONE 133		

Perform a detail visual inspection of the control cables within the left main landing gear wheel well for broken wires. Check associated pulleys, brackets, and mechanisms for condition and security of installation. The following cables are located in the left MLG wheel well:

- A. Aileron control cables
- B. Spoiler control cables
- C. Speed brake control cables

Note: The control cables must be displaced full travel in each direction for complete inspection at seals, pulleys, and fairlead areas.

INTERVAL NOTE: Whichever occurs first.

A. References

Reference	Title
AMM 12-26-00-600-801	Control Cable Lubrication (P/B 301)

B. Consumable Materials

Reference	Description	Specification
G00034	Cotton Wiper - Process Cleaning Absorbent Wiper	AMS3819 Class 1 Grade A or B
	(Cheesecloth, Gauze)	Form 1 (Supersede BMS15-5 CL
		A)

EFFECTIVITY ILF ALL	SOURCE MRB	FLIGHT CONTROL CABLES - LEFT MAIN GEAR	WELL
		D633A109-ILF 27-226-00-01	Page 1 of 7 Jun 15/2023





DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				27-226-00-01

TASK 20-20-31-200-801

MECH INSP

1. Control Cable Wire Rope - Inspection

(Figure 1)

A. Prepare for the Inspection

SUBTASK 20-20-31-100-001



DO NOT APPLY SOLVENTS, GREASE, OR OIL TO STAINLESS STEEL CONTROL CABLES. THESE MATERIALS CAN COLLECT CONTAMINATION THAT CAN CAUSE DAMAGE TO THE INTERNAL SURFACES OF THE CRES CABLE STRANDS. THIS CAN DECREASE THE SERVICE LIFE OF THE CABLE.



DO NOT USE SOLVENT OR HEAT TO THIN GREASE. DO NOT USE SOLVENT TO CLEAN CABLES, SINCE SOLVENT DILUTES AND REMOVES GREASE FROM INSIDE CABLE STRANDS. DO NOT APPLY OR SPRAY BMS 3-23 ON CONTROL CABLES.

- (1) If it is necessary, clean the control cables with lint-free cotton wiper, G00034, that is clean and dry.
 - (a) Remove the old grease and dirt from the surface of the control cable.
 - (b) Clean the control cable for the full length of the cable for the full length of travel through fairleads, air pressure seals, over pulleys, quadrants, and drums.

SUBTASK 20-20-31-200-003

(2) Perform a detailed visual inspection to make sure that the cable does not contact parts other than pulleys, quadrants, cable seals, or grommets installed to control cable routing.

NOTE: The minimum cable clearance from other parts is 0.2 in. (5.1 mm), except 0.1 in. (2.5 mm) within 10 in. (254 mm) of a pulley or quadrant.

(a) Look for evidence of contact with other parts. Correct the condition if evidence of contact is found.

B. Control Cable Wire Rope Inspection

SUBTASK 20-20-31-200-015

- (1) Ignore this step if it is not applicable to your work:
 - (a) Make sure that the flight control cables are displaced full travel in each direction for the complete inspection at seals, pulleys, and fairlead areas.

SUBTASK 20-20-31-200-013

- (2) Perform a detailed visual inspection of the cable runs for incorrect routing, kinks in the wire rope, or other damage.
 - (a) Replace the cable assembly if:
 - 1) A wear pattern exists where the individual wires in a strand appear to blend together (outer wires worn by more than 40 percent) (Figure 1).
 - 2) A kink is found.
 - Corrosion is found.

EFFECTIVITY ILF ALL	SOURCE MRB	FLIGHT CONTROL CABLES - LEFT MAIN GEAR WEL	.L	
			Page 2	





DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 27-226		
SIIDTASK	20-20-31-2	00-014					MECH	INSP
		a detailed visual ins	pection of	the cable.				
` '			•		r using a color tracer fil	ament or		
_		thread per MIL-8342	20H. The c	condition of the o	colored nonmetallic three or strength of the cable	eads within a		
(8	a) Ru	b a cloth along the c	able to fin	d the broken wi	res.			
	NC	OTE: The cloth will c	atch on br	oken wires.				
(k	o) Re	place the 7X7 cable	assembly	if:				
	1)	There are two or	more brok	en wires in 12 ir	n. (305 mm) of cable.			
	2)	There are three o	r more bro	oken wires anyw	here in the total cable	assembly.		
(0	c) Re	place the 7X19 cabl	e assemb	ly if:				
	1)	There are four or	more brok	cen wires in 12 i	n. (305 mm) of cable.			
	2)				ere in the total cable as	sembly.		
(0	,	spect the carbon stee		-		•		
,	1)	•			on the control cable.			
	2)		s not suffic		k: Control Cable Lubric	cation, AMM		
		NOTE: Do not ap	oply the gr	ease or oil to the	e stainless steel (CRES	6) control		
			END OF	TASK ———				
	EFFECTIVI		SOURCE MRB	FLIGHT CONTR	ROL CABLES - LEFT MA	IN GEAR WEI	 _L	
				D633A109-ILF 27-226-00-01			Page 3 un 15/	





	D	ATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 27-226		
	TAS	K 20-	20-31-	-200-802					MECH	INSI
2.	Insp	ectio	n of th	ne Control Cable Fitti	ngs					
	A.	Prod	edure	•						
		SUBTA	SK 20-20-	31-200-007						
		(1)		orm a detailed visual inst t (wire locking, cotter pi			the means of locking t	he joints are		
			(a)	Install any missing par	ts.					
		SUBTA	SK 20-20-	31-200-008						
		(2)		orm a detailed inspections or corrosion.	on of the sv	vaged portions o	of swaged end fittings f	or surface		
			(a)	Replace the cable ass	embly if cr	acks or corrosio	n are found.			
		SUBTA	SK 20-20-	31-200-009						
		(3)	Perfo	rm a detailed visual ins	spection of	the unswaged p	portion of the end fitting	g.		
				Replace the cable ass fitting is bent more tha			f corrosion is present, of	or if the end		
		SUBTA	SK 20-20-	31-200-010						
		(4)	Perfo	orm a detailed visual ins	spection of	the turnbuckle.				
			(a)	Replace the turnbuckle	e if a crack	is visible or if co	orrosion is present.			
					- END OF	TASK ———				





	С	ATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING 0 27-226		
	TAS	K 20-	-20-31-2	200-805					MECH	INSP
3.			n of Pu							
	Α.		cedure							
	Α.		ASK 20-20-3	1-200-011						
		(1)			pection to	make sure that	pulleys are free to rota	ite.		
		` '		Replace pulleys which						
		SUBTA	ASK 20-20-3	1-200-012						
		(2)	Perfor	m a detailed visual ins	pection of	f the pulleys for	conditions shown in (Fig	gure 2).		
			(a) F	Replace pulleys which	are not in	a normal condit	tion.			
					END OF	TASK ———				
			EFFEC1	ΓΙVΙΤΥ	SOURCE	FLIGHT CONTR	ROL CABLES - LEFT MA	IN GEAR WF	LL LL	
			ILF A	ALL	MRB					
						D633A109-ILF			Page 5	of 7
						27-226-00-01		F	eb 15/	2016

ILF



DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CARD NO. 27-226-00-01
					21-220-00-01
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			y	·/////////////////////////////////////	
WIRE					
		CABLE STR EXAMPLE)		·/////////////////////////////////////	
\	OOO OOO OOO		(/ /////20)		
	000				
			EACH O	UTER WIRE WORN LESS TI ORN AREAS NOT BLENDE	HAN 40%
	7x7 CABLE		(**	OIN AILAONOT BEENDE	.5)
	000	3		2	
000	000000000000000000000000000000000000000		y	<u> </u>	
		<u>/ </u>			
	0000000000			<u> </u>	
	00000000000000000000000000000000000000	_			
	80000	4		HOUTER WIRE WORN 40-5	
CABLE STRAND	000		(WC	ORN AREAS ARE BLENDED))
(EXAMPLE) (19 WIRES)					
(10 1111120)	7x19 CABLE			1	
					_
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			EACH W	VIRE IS WORN MORE THAN	150%
00					
TO COLOR					
	00000		3 THE O	OUTER WIRE WEAR AREA C	N CABLE
	000		STRAI	ND. A VISIBLE SPACE BETV	VEEN
			WIRES SURF	OR A FULLY BLEND ACE. 2	Eυ
EXAM	PLE OF INTERNAL WEA	R			
			4 CABLE	WEAR MAY OCCUR ON O	NE SIDE
1 VISIBLE S	PACE BETWEEN WIRES	3.	CABLE	OR ON FULL CIRCUMFERE E WEAR CAN EXTEND ALOI	NG THE
	NDITION RESULTING IN		CABLE	FOR A DISTANCE EQUAL TRAVEL.	
	SURFACES BETWEEN		CABLE		5914 S0006562076_V3
			Wear Patterns		
			Figure 1		
EFFEC"		SOURCE	FLIGHT CONTI	ROL CABLES - LEFT MAII	N GEAR WELL
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			D633A109-ILF		Page 6 of 7
			27-226-00-01		Jun 15/2015
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27-226-00	
CABLE TENSION TOO HIGH PULLEY NOT ALIGNED CORRECTLY	
PULLEY GROOVE WITH CABLE NOT ALIGNED CORRECTLY	
EXCESSIVE WEAR	
PULLEY WILL NOT TURN NORMAL CONDITION	
NORMAL CONDITION	
F25724 S0006562077_\	V3
Pulley Wear Patterns Figure 2	
	
EFFECTIVITY ILF ALL SOURCE MRB FLIGHT CONTROL CABLES - LEFT MAIN GEAR WELL	
D633A109-ILF Pa 27-226-00-01 Jun	ige 7 of 7 n 15/2015





AIRLINE	CARD NO	LEFT AND RIGHT MAIN LANDING GEAR MANUAL			BOEING CARD NO. 32-220-00-01		
DATE	TASK FUNCTIONAL		EXTENSION SYSTE	M	RELATED CARD		
TAIL NUMBER	WORK AREA CREW CABIN	VERSION 1.1	THRESHOLD 36 MO	REPEAT 36 MO	APPLICA AIRPLANE	ABILITY	
STATION	SKILL AIRPL				ALL	ALL	
		ACCESS S2122			ZONE 212 734 744		

Perform a functional check of the left and right main landing gear manual extension system and alternate extension bypass valve.

A. References

Reference	Title
AMM 07-11-01-580-815	Lift the Airplane with the Jacks (P/B 201)
AMM 07-11-01-580-816	Lower the Airplane Off the Jacks (P/B 201)
AMM 24-22-00-860-812	Remove Electrical Power (P/B 201)
AMM 29-11-00-860-801	Hydraulic System A or B Pressurization (P/B 201)
AMM 29-11-00-860-805	Hydraulic System A or B Power Removal (P/B 201)
AMM 32-00-01-080-801	Landing Gear Downlock Pins Removal (P/B 201)
AMM 32-00-01-480-801	Landing Gear Downlock Pins Installation (P/B 201)
AMM 32-09-10-740-801	Proximity Switch Electronics Unit (PSEU) BITE Test - Ground Test (P/B 501)

B. Tools/Equipment

NOTE: When more than one tool part number is listed under the same "Reference" number, the tools shown are alternates to each other within the same airplane series. Tool part numbers that are replaced or non-procurable are preceded by "Opt:", which stands for Optional.

Reference	Description
STD-1184	Scale - Spring, 0-100 Lbs, Tension

EFFECTIVITY ILF ALL	SOURCE MRB	LEFT AND RIGHT MAIN LANDING GEAR MANUA SYSTEM	L EXTENSION
		D633A109-ILF 32-220-00-01	Page 1 of 10 Feb 15/2024





DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				32-220-00-01

TASK 32-34-00-730-801

MECH INSP

1. Main Gear Manual Extension System Test - Airplane on Jacks

(Figure 1)

A. General

(1) The normal extension/retraction system will not operate with the cover for the alternate extend T handle open; make sure that the cover is closed when you will extend or retract the landing gear with the landing gear control handle.

B. Prepare for the Test

SUBTASK 32-34-00-480-002



MAKE SURE THE DOWNLOCK PINS ARE INSTALLED ON ALL THE LANDING GEAR. WITHOUT THE DOWNLOCK PINS, THE LANDING GEAR CAN RETRACT AND CAUSE INJURIES TO PERSONS AND DAMAGE TO EQUIPMENT.

(1) Make sure that the downlock pins are installed in the nose and main landing gear, do this task: Landing Gear Downlock Pins Installation, AMM TASK 32-00-01-480-801.

SUBTASK 32-34-00-580-004

(2) Do this task: Lift the Airplane with the Jacks, AMM TASK 07-11-01-580-815.

SUBTASK 32-34-00-860-005

(3) For hydraulic system A, do this task: Hydraulic System A or B Pressurization, AMM TASK 29-11-00-860-801.

SUBTASK 32-34-00-860-006

(4) Make sure that these circuit breakers are closed:

CAPT Electrical System Panel, P18-2

Row	<u>Col</u>	<u>Number</u>	<u>Name</u>
В	3	C01312	ENGINE 1 RUN/PWR

F/O Electrical System Panel, P6-3

<u>Row</u>	<u>Col</u>	<u>Number</u>	<u>Name</u>
В	17	C00129	LANDING GEAR LATCH & PRESS WARN
С	15	C01355	LANDING GEAR AIR/GND SYS 2
С	16	C01356	LANDING GEAR AIR/GND SYS 1
D	1	C01399	PSEU PRI
D	2	C01400	PSEU ALTN
D	16	C01432	LANDING GEAR ALTN EXTEND SOL
Е	12	C00314	INDICATOR MASTER DIM SECT 2
F	11	C00317	INDICATOR MASTER DIM SECT 5
F	13	C01179	INDICATOR MASTER DIM SECT 7

EFFECTIVITY ILF ALL	SOURCE MRB	LEFT AND RIGHT MAIN LANDING GEAR MANUAL EXTENSION SYSTEM			
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DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				32-220-00-01

SUBTASK 32-34-00-860-008

MECH INSP

(5) Open these circuit breakers and install safety tags:

F/O Electrical System Panel, P6-1

Row	Col	<u>Number</u>	<u>Name</u>
D	13	C00120	WEATHER RADAR RT

F/O Electrical System Panel, P6-2

Row	<u>Col</u>	<u>Number</u>	<u>Name</u>
С	15	C00799	HYD SYS LDG GR SYS XFR VALVE SEC
С	16	C00781	HYD SYS LDG GR SYS XFR VALVE PRI

F/O Electrical System Panel, P6-3

Row	Col	<u>Number</u>	<u>Name</u>
D	18	C00451	LANDING GEAR AURAL WARN

SUBTASK 32-34-00-860-010

(6) Make sure that the control lever for the landing gear is in the DN position.

SUBTASK 32-34-00-860-025



OBEY THE INSTRUCTIONS IN THE PROCEDURE TO PUT THE SPEEDBRAKE HANDLE TO THE DOWN POSITION. IF YOU DO NOT OBEY THE INSTRUCTIONS, INJURIES TO PERSONNEL, AND DAMAGE TO EQUIPMENT CAN OCCUR.

(7) Put the speedbrake handle to the down position.

SUBTASK 32-34-00-860-011

(8) Move the No. 1 and No. 2 throttle levers to the full forward position.

C. Main Gear Manual Extension System Test

SUBTASK 32-34-00-860-012

(1) Put the control lever in the OFF position.

SUBTASK 32-34-00-700-002

- (2) Pull the manual extension handle for the left main gear.
 - (a) Make sure that the full travel of the T handle is at least 18.5 in. (469.9 mm).

SUBTASK 32-34-00-700-003

- (3) Release the manual extension handle for the left main gear.
 - (a) Make sure that the handle returns to the stowed position without hesitation or binding.

SUBTASK 32-34-00-480-003

(4) Attach a spring scale (0-100 Lbs), STD-1184, to the manual extension T handle for the left main gear.

SUBTASK 32-34-00-730-001

(5) Pull the manual extension handle for the left main gear.

EFFECTIVITY ILF ALL	SOURCE MRB	LEFT AND RIGHT MAIN LANDING GEAR MANUAL EX	(TENS	SION
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DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C. 32-220			
	(a)	Make sure that the loa (13.3 N).	ad to start th	ne T handle fror	m the seat has a minim	um of 3.0 lbf	MECH	INSI	
	(b)	Make sure that the loa	ad during th	e full travel of the	he T handle is not more	e than			
		18.0 lbf (80.1 N).	must be no	illad upward an	d forward when you me	acuro tho			
			simulate a	person seated	in the First Officer's sea				
	(c)	Make sure that the ma	anual exten	sion mechanisn	n operates freely.				
SUBTA	ASK 32-34	4-00-730-011							
(6)	Pull	the manual extension h	nandle for th	ne right main ge	ear.				
	(a)	Make sure that the ful	I travel of th	ne T handle is a	t least 18.5 in. (469.9 n	nm).			
SUBTA	ASK 32-34	4-00-730-012							
(7)	Rele	ease the manual extens	ion handle	for the right ma	in gear.				
	(a)	Make sure that the habinding.	ndle return	s to the stowed	position without hesita	tion or			
SUBTA	ASK 32-34	4-00-860-038							
(8)	Rem	nove the spring scale from	om the mar	nual extension T	handle for the left mai	n gear.			
SUBTA	ASK 32-34	4-00-480-012							
(9)	Atta	ch a spring scale to the	manual ex	tension T handl	le for the right main gea	ar.			
SUBTA	ASK 32-3	4-00-730-013							
(10)	Pull	the manual extension h	nandle for th	ne right main ge	ear.				
	(a)	Make sure that the load to start the T handle from the seat has a minimum of 3.0 lbf (13.3 N).							
	(b)	Make sure that the load 18.0 lbf (80.1 N).	ake sure that the load during the full travel of the T handle is not more than .0 lbf (80.1 N).						
			simulate a	person seated	d forward when you me in the First Officer's se				
	(c)	Make sure that the ma	anual exten	sion mechanisn	n operates freely.				
SUBTA	ASK 32-34	4-00-080-004							
(11)		nove the downlock pin f rnlock Pins Removal, A		•	•	nding Gear			
SUBTA	ASK 32-3	4-00-860-048							
(12)	Rem	nove the spring scale from	om the mar	nual extension T	「handle for the right ma	ain gear.			
SUBTA	ASK 32-3	4-00-860-019							
(13)	Mak	e sure that the cover fo	r the T han	dle is closed.					
	NOT	E: The extension/retra extend T handle op		m will not opera	ate with the cover for th	e alternate			
SUBTA	ASK 32-3	4-00-860-017							
(14)	Mov	e the control lever for the	ne landing (gear to the DN p	position.				
		ECTIVITY F ALL	SOURCE MRB	LEFT AND RIGI	HT MAIN LANDING GEA	R MANUAL E	KTEN:	SIC	
				D633A109-ILF 32-220-00-01			age 4 Oct 15/		





DATE TAIL NUMBER STATION AIRLINE CARD NO. BOEING CARD NO. 32-220-00-01

SUBTASK 32-34-00-730-003

MECH INSP



MAKE SURE THAT ALL PERSONS AND EQUIPMENT ARE CLEAR OF THE MAIN LANDING GEAR. FAST MOVEMENT OF THE MAIN LANDING GEAR CAN CAUSE INJURIES TO PERSONS AND DAMAGE TO EQUIPMENT.

(15) Move the control lever for the landing gear to UP and wait while the main gear retracts.

SUBTASK 32-34-00-860-026

(16) Open the access door to the manual extension T handles.

SUBTASK 32-34-00-730-007

(17) Pull the manual extension handle of the left main gear.

NOTE: The T handle must be pulled upward and forward. This will simulate a person seated in the First Officer's seat. Do not pull the handle straight up.

SUBTASK 32-34-00-700-011

- (18) Make sure that the left main gear free falls to the down and locked position.
 - (a) Make sure that the green light for the left main gear is on.
 - (b) Make sure that the red light for the left main gear is on.

NOTE: The red light for the main gear will stay on to show that the landing gear control lever and landing gear positions disagree.

SUBTASK 32-34-00-700-012

(19) Release the manual extension handle for the left main gear.

SUBTASK 32-34-00-730-014

(20) Pull the manual extension handle of the right main gear.

NOTE: The T handle must be pulled upward and forward. This will simulate a person seated in the First Officer's seat. Do not pull the handle straight up.

SUBTASK 32-34-00-700-033

- (21) Make sure that the right main gear free falls to the down and locked position.
 - (a) Make sure that the green light for the right main gear is on.
 - (b) Make sure that the red light for the right main gear is on.

<u>NOTE</u>: The red light for the main gear will stay on to show that the landing gear control lever and landing gear positions disagree.

SUBTASK 32-34-00-730-015

(22) Release the manual extension handle for the right main gear.

SUBTASK 32-34-00-860-029

- (23) Move the control lever for the landing gear to the DOWN position.
 - (a) Make sure that the red light for the main gear is off.

SUBTASK 32-34-00-860-037

(24) Make sure that the cover for the T-handle is closed.

SUBTASK 32-34-00-860-027

(25) Move the control lever for the landing gear to the OFF position.

EFFECTIVITY ILF ALL	SOURCE MRB	LEFT AND RIGHT MAIN LANDING GEAR MANUA SYSTEM	AL EXTENSION
		D633A109-ILF 32-220-00-01	Page 5 of 10 Mar 15/2023





DATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CA 32-220- 0				
SUBTA	ASK 32-3	4-00-730-009	•				MECH	INSI		
(26)			ne landing	gear to UP and	wait while the main gear	retracts.				
SUBTA		4-00-860-024			· ·					
(27)	Put t	the landing gear control	l lever in th	e OFF position.						
SUBTA	ASK 32-3	4-00-860-030								
(28)	Ope	n the access door to the	e manual e	xtension T hand	dles.					
SUBTA	ASK 32-3	4-00-700-015								
(29) Attach a spring scale (0-100 Lbs), STD-1184, to the manual extension T handle for the left main gear.										
SUBTA		4-00-730-004								
(30)	Pull	the manual extension h		•						
	(a)	Make sure that the ma 50.0 lbf (222.4 N).	aximum loa	d when you pull	I the T handle is not mor	e than				
			simulate a	person seated	d forward when you mea in the First Officer's seat					
SUBTA	ASK 32-3	4-00-700-006								
(31)	Mak	e sure that the left mair	n gear free	falls to the dow	n and locked position.					
	(a)	Make sure that the green light for the left main gear is on.								
(b) Make sure that the red light for the left main gear is on.										
				n gear will stay o gear positions	on to show that the landi disagree.	ng gear				
SUBTA	ASK 32-3	4-00-080-005								
(32)	Rem	nove the spring scale fro	om the mar	nual extension T	handles for the left mai	n gear.				
SUBTA		4-00-730-017								
(33)	Atta	ch a spring scale to the	manual ex	tension T handl	le for the right main gear	-				
		4-00-860-050								
(34)	, ,	the manual extension h								
	(a)	Make sure that the maximum load when you pull the T handle is not more than 50.0 lbf (222.4 N).								
			simulate a	person seated	d forward when you mea in the First Officer's seat					
SUBTA	ASK 32-3	4-00-860-051								
(35)	Mak	e sure that the right ma	in gear free	e falls to the dov	wn and locked position.					
	(a)	Make sure that the gre	een light fo	r the right main	gear is on.					
	(b)	(b) Make sure that the red light for the right main gear is on.								
				n gear will stay o gear positions	on to show that the landi disagree.	ng gear				
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		ECTIVITY F ALL	SOURCE MRB	LEFT AND RIGI SYSTEM	HT MAIN LANDING GEAR	MANUAL EX	TENS	SIO		





	DATE		1	TAIL NUMBER		STATION	AIRLINE CARD NO.	32-220		
	SUBT	ASK 32-34-00)-860-052		<u> </u>	'		-	MECH	INS
	(36)			rol lever for t	the landing	gear to the DOV	VN position.			
		(a) N	/lake su	re that the re	ed lights for	the right and left	t main gear are off.			
	SUBT	ASK 32-34-00)-860-053							
	(37)	(37) Remove the spring scale from the manual extension T handles for the right main gear.								
		JBTASK 32-34-00-860-054								
	(38) Make sure that the cover for the T-handles is closed.									
D.	D. Put the Airplane Back to its Usual Condition									
		SUBTASK 32-34-00-860-015 (1) For Hydraulic System A. do this task: Hydraulic System A. or B. Power Removal, AMM								
	(1) For Hydraulic System A, do this task: Hydraulic System A or B Power Removal, AMM TASK 29-11-00-860-805.									
	SUBT	ASK 32-34-00	0-480-005							
	(2)					ear that you test K 32-00-01-480-	ed, do this task: Landi 801.	ng Gear		
	SUBT	ASK 32-34-00			,					
	(3)	Close	these ci	rcuit breaker	rs:					
		F/O EI	ectrical	System Pa	nel, P6-2					
		Row	Col	Number	<u>Name</u>					
C 15 C00799 HYD SYS LDG GR SYS XFR VALVE SEC										
C 16 C00781 HYD SYS LDG GR SYS XFR VALVE PRI										
	F/O Electrical System Panel, P6-3									
		Row Col Number Name								
		D	18	C00451	LANDING	GEAR AURAL	WARN			
		ASK 32-34-00		4 N - 0 -	41	. 4 . 4	II '4' (fII - f4)			
	(4)			1 and No. 2	throttle leve	ers back to the id	lle position (full aft).			
	(5)	ASK 32-34-00 Clase 1		uit breaker:						
	(0)			System Pa	nol D6 1					
		Row		Number	Name					
		D	13	C00120		R RADAR RT				
	SUBT	ASK 32-34-00)-860-022							
	(6)			wer is not ne 0-860-812.	ecessary, do	this task: Remo	ove Electrical Power, A	MM		
	SUBT	ASK 32-34-00)-580-003							
	(7)	Do this	s task: L	ower the Air	plane Off th	e Jacks, AMM T	ASK 07-11-01-580-816	3.		
	(,)	20	, taoni E		piano on an	5 5001.0,7 mm				
_										
		ILF A			SOURCE MRB	LEFT AND RIGH SYSTEM	HT MAIN LANDING GEA	AR MANUAL E	XTENS	SI
						D633A109-ILF		Р	age 7	of
						32-220-00-01			/lar 15/	





DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C		
					32-220	-00-01 MECH	INSP
SUBTASK 32-34-0		⊏l4:-			1644 -1-		INOF
this ta	sk: Proximity Switch sk: Proximity Switch E 32-09-10-740-801.	Electronics	Unit (PSEU) for Unit (PSEU) BIT	stored faults using the FE Test - Ground Test,	Self test, do AMM		
		END OF	TASK ——				
EFFECT ILF A	TIVITY ALL	SOURCE MRB	LEFT AND RIGH SYSTEM	IT MAIN LANDING GEA	R MANUAL E	XTENS	SION
			D633A109-ILF		P	age 8 (of 10



TAIL NUMBER

DATE



737-600/700/800/900 **TASK CARDS**

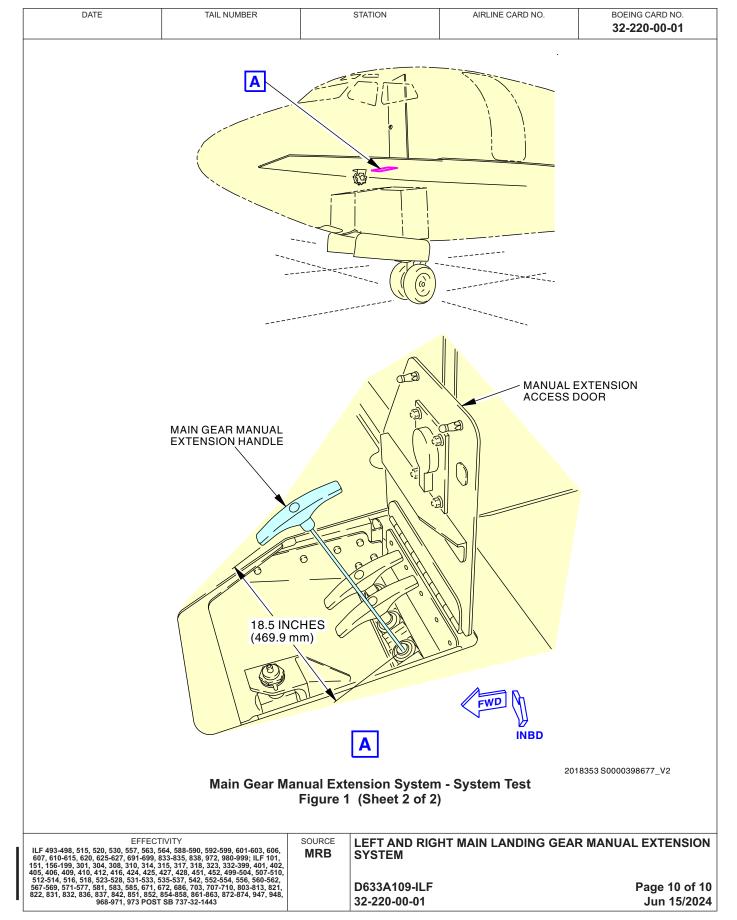
STATION

BOEING CARD NO.

AIRLINE CARD NO. 32-220-00-01 MANUAL EXTENSION ACCESS DOOR MAIN GEAR MANUAL EXTENSION HANDLES \ 18.5 INCHES **INBD** G27422 S0006575350_V2 Main Gear Manual Extension System - System Test Figure 1 (Sheet 1 of 2) EFFECTIVITY
ILF 101, 151, 156-199, 301, 304, 308, 310, 314, 315, 317, 318, 323, 332-399, 401, 402, 405, 406, 409, 410, 412, 416, 424, 425, 427, 428, 451, 452, 499-504, 507-510, 512-514, 516, 518, 523-528, 531-533, 535-537, 542, 552-554, 556, 560-562, 567-569, 571-577, 581, 583, 585, 671, 672, 686, 703, 707-710, 803-813, 821, 822, 831, 832, 836, 837, 842, 851, 852, 854-858, 861-863, 872-874, 947, 948, 968-971, 973 PRE 5B 737-32-1443 SOURCE LEFT AND RIGHT MAIN LANDING GEAR MANUAL EXTENSION **MRB SYSTEM** D633A109-ILF Page 9 of 10 Jun 15/2024 32-220-00-01











AIRLINE	AIRLINE CARD NO		LEFT MAIN LANDING GEAR AND LANDING GEAR			CARD NO. 6-01-01
DATE	ZONAL (GV)		DOORS		RELATED CARD	
TAIL NUMBER	WORK AREA L MAIN GEAR	VERSION 1.1	THRESHOLD 5500 FC	REPEAT 5500 FC	APPLIC AIRPLANE	ABILITY ENGINE
STATION	SKILL AIRPL	1.2 NOTE	30 MO 30 MO		ALL	ALL
		ACCESS			730	

Perform an external zonal inspection (GV) of the left main landing gear and landing gear doors. (EZAP)

INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 5500 FC/30 MO is satisfied by this zonal inspection.

A. References

Reference	Title
AMM 05-00-00-910-804	Enhanced Zonal Inspection Program (EZAP) Precautions (P/B 201)
AMM 20-60-07-913-801	Protection of the EWIS During Maintenance (P/B 201)

-	EFFECTIVITY ILF ALL	SOURCE MRB	LEFT MAIN LANDING GEAR AND LANDIN	IG GEAR DOORS
			D633A109-ILF 32-806-01-01	Page 1 of 4 Feb 15/2022

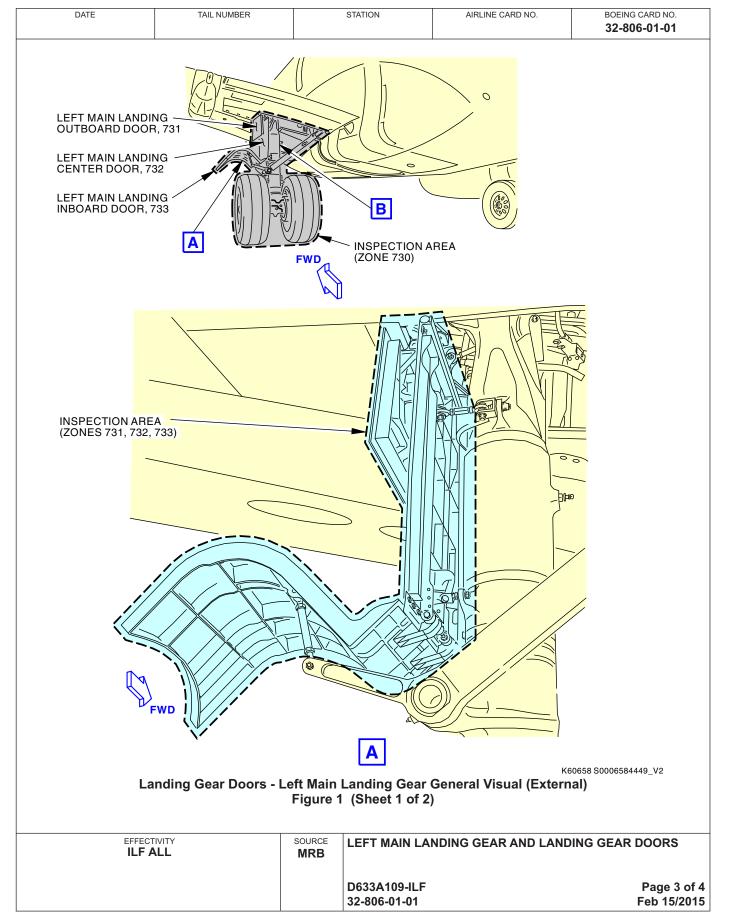




	С	ATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 32-806		
	EWI		41-07-2	210-803		,			MECH	INSP
1.	EXT		AL - ZO	NAL (GV): Left Main L	anding (Gear and Land	ing Gear Doors			
	A.	Gen (1)		onal inspection procedu	ure satisf	ies the required	Enhanced Zonal Analy	sis		
	D	Zon		dure (EZAP) - derived Z	Zonal ins _l	pection requirem	nent for this zone.			
	B.	S. Zonal Inspection SUBTASK 05-41-07-210-003								
	(1) Do the zonal inspection following the procedures in Enhanced Zonal Inspection Program (EZAP) Precautions, AMM TASK 05-00-00-910-804.									
	 SUBTASK 05-41-07-210-007 (2) Perform an external zonal inspection (GV) of the left main landing gear and landing gear doors (EZAP). SUBTASK 05-41-07-910-002 (3) Refer to Protection of the EWIS During Maintenance, AMM TASK 20-60-07-913-801 for the protection and caution information that will minimize contamination and accidental damage to Electrical Wiring Interconnect Systems (EWIS) during maintenance. 									
				-		TASK —	,			
			EFFECT		SOURCE MRB	LEFT MAIN LAN	NDING GEAR AND LAND	ING GEAR D	OORS	
						D633A109-ILF 32-806-01-01			Page 2 eb 15/	











STATION DATE TAIL NUMBER AIRLINE CARD NO. BOEING CARD NO. 32-806-01-01 INSPECTION AREA (ZONE 734) INBD K62597 S0006584450_V2 Landing Gear Doors - Left Main Landing Gear General Visual (External) Figure 1 (Sheet 2 of 2) EFFECTIVITY ILF ALL SOURCE LEFT MAIN LANDING GEAR AND LANDING GEAR DOORS **MRB** D633A109-ILF Page 4 of 4

32-806-01-01

Feb 15/2015





AIRLINE	AIRLINE CARD NO		RIGHT MAIN LANDING GEAR AND LANDING			CARD NO.)-02-01
DATE	TASK ZONAL (GV)		GEAR DOORS	RELATED CARD		
TAIL NUMBER	WORK AREA R MAIN GEAR	VERSION 1.1	THRESHOLD 5500 FC	REPEAT 5500 FC	APPLIC AIRPLANE	CABILITY
STATION	SKILL AIRPL	1.2 NOTE	30 MO	30 MO	ALL	ALL
		ACCESS			ZONE 740	

Perform an external zonal inspection (GV) of the right main landing gear and landing gear doors. (EZAP)

INTERVAL NOTE: Whichever comes first. The EZAP inspection requirement with interval 5500 FC/30 MO is satisfied by this zonal inspection.

A. References

Reference	Title
AMM 05-00-00-910-804	Enhanced Zonal Inspection Program (EZAP) Precautions (P/B 201)
AMM 20-60-07-913-801	Protection of the EWIS During Maintenance (P/B 201)

EFFECTIVITY ILF ALL	SOURCE MRB	RIGHT MAIN LANDING GEAR AND LANDING	GEAR DOORS
		D633A109-ILF 32-810-02-01	Page 1 of 4 Feb 15/2022

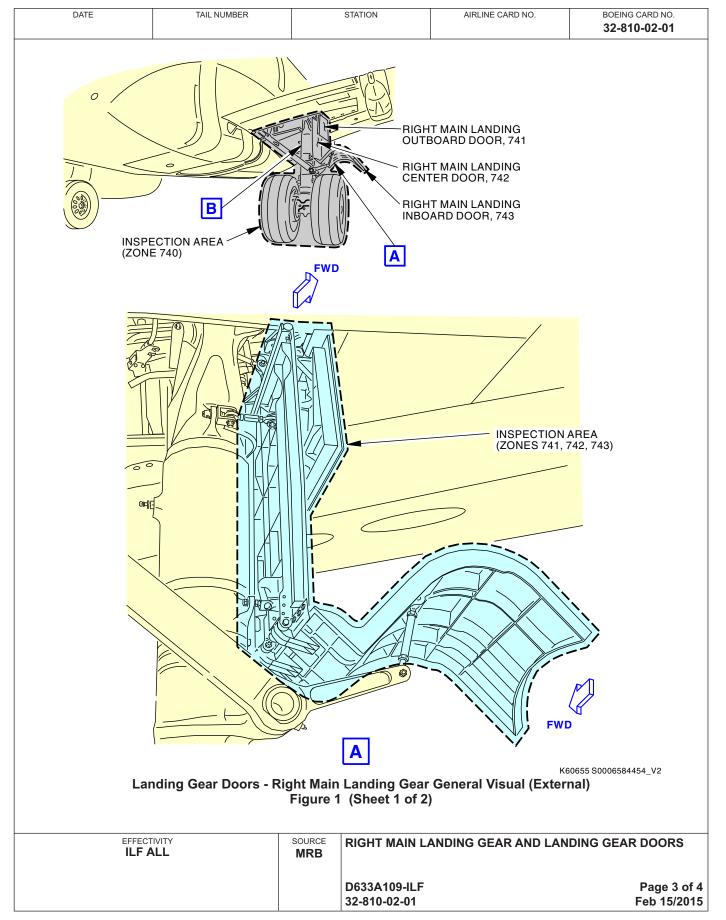




	С	ATE		TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING C 32-810			
	EWI								MECH	INSP	
1.				210-805 NAL (GV): Right Main	Landino	n Gear and Lan	ding Gear Doors				
		EXTERNAL - ZONAL (GV): Right Main Landing Gear and Landing Gear Doors (Figure 1)									
	A.	Gen	eral								
		(1)		onal inspection procedudure (EZAP) - derived Z			Enhanced Zonal Analynent for this zone.	sis			
	B. Zonal Inspection										
		(1) Do the zonal inspection following the procedures in Enhanced Zonal Inspection Program (EZAP) Precautions, AMM TASK 05-00-00-910-804.									
		SUBTA	sk 05-41-0 Perfor	7-210-008			main landing gear and	landing gear			
		suвта (3)	SK 05-41-0 Refer the pro	r-910-003 to Protection of the EW otection and caution info	ormation	that will minimiz	AMM TASK 20-60-07-9 ze contamination and ac VIS) during maintenanc	ccidental			
						TASK —	,				
			EFFECT ILF A		SOURCE MRB	RIGHT MAIN LA	ANDING GEAR AND LAN	DING GEAR	DOOR	S	
						D633A109-ILF 32-810-02-01			Page 2 eb 15/		











STATION DATE TAIL NUMBER AIRLINE CARD NO. BOEING CARD NO. 32-810-02-01 INSPECTION AREA (ZONE 744) K62599 S0006584455_V2 Landing Gear Doors - Right Main Landing Gear General Visual (External) Figure 1 (Sheet 2 of 2) EFFECTIVITY ILF ALL SOURCE RIGHT MAIN LANDING GEAR AND LANDING GEAR DOORS **MRB** D633A109-ILF Page 4 of 4 Feb 15/2015 32-810-02-01





AIRLIN	E CARD NO	E/E ACC	TITLE CESS DOOR LUBR		BOEING CARD NO. 52-120-00-01		
DATE	TASK LUBRICATE				RELATE	D CARD	
TAIL NUMBER	WORK AREA E/E COMPARTMENT	VERSION 1.1	THRESHOLD 2 YR	REPEAT 2 YR	APPLIC/	ABILITY ENGINE	
STATION	SKILL AIRPL	-			ALL	ALL	
		ACCESS 117A			ZONE 117 118		

Lubricate the E/E access door handle latching mechanism (rack and pinion gear and the lock pins).

A. Consumable Materials

Reference	Description	Specification
D00633	Grease - Aircraft General Purpose	BMS3-33

EFFECTIVITY ILF ALL	SOURCE MRB	E/E ACCESS DOOR LUBRICATION		
		D633A109-ILF 52-120-00-01	Page 1 of 7 Jun 15/2022	





DATE	TAIL NUMBER	STATION	AIRLINE CARD NO.	BOEING CARD NO.
				52-120-00-01

TASK 12-25-41-640-801

MECH INSP

1. Electronic Equipment Access Door Servicing

(Figure 1, Table 1)

A. Prepare for Servicing

SUBTASK 12-25-41-010-005

(1) Turn the latch handle to the closed position.

SUBTASK 12-25-41-010-007

(2) Open this access panel:

Number Name/Location

117A Electronic Equipment Access Door

SUBTASK 12-25-41-010-002

(3) Do the following to open this access panel:

Number Name/Location

117AW Equipment Access Door Cover

(Figure 1)

- (a) Remove the bolt [105], washer [106], and nut [107] that attach the collar [104] to the latch mechanism [2].
- (b) Remove the collar [104] and the washer [108].
- (c) Remove the screws [101] and the screws [102] that attach the cover [103] to the door.
- (d) Remove the cover [103].

SUBTASK 12-25-41-010-006

- (4) Remove the support plate [110] from the latch mechanism [2] as follows (Figure 1):
 - (a) Remove the screws [109] that attach the support plate [110] to the latch mechanism [2].

NOTE: After you remove the screws [109], the bearings [113] and the spacers [112, 114, 115, 116 and 117] are not held in position.

(b) Remove the support plate [110] and the washer [111].

B. Procedure

SUBTASK 12-25-41-640-001

- (1) Lubricate the components with the applicable material shown in (Table 1, Figure 1):
 - (a) grease, D00633

Table 1 Electronic Equipment Access Door Lubrication (Fig. 301)

Item No.	Nomenclature	Lubricant	Method of Application	Number of Points
1	Latch Pins (4)	grease, D00633	Hand	4
2	Latch Mechanism (1)	grease, D00633	Hand	1

EFFECTIVITY ILF ALL	SOURCE MRB	E/E ACCESS DOOR LUBRICATION	
		D633A109-ILF 52-120-00-01	Page 2 of 7 Jun 15/2022

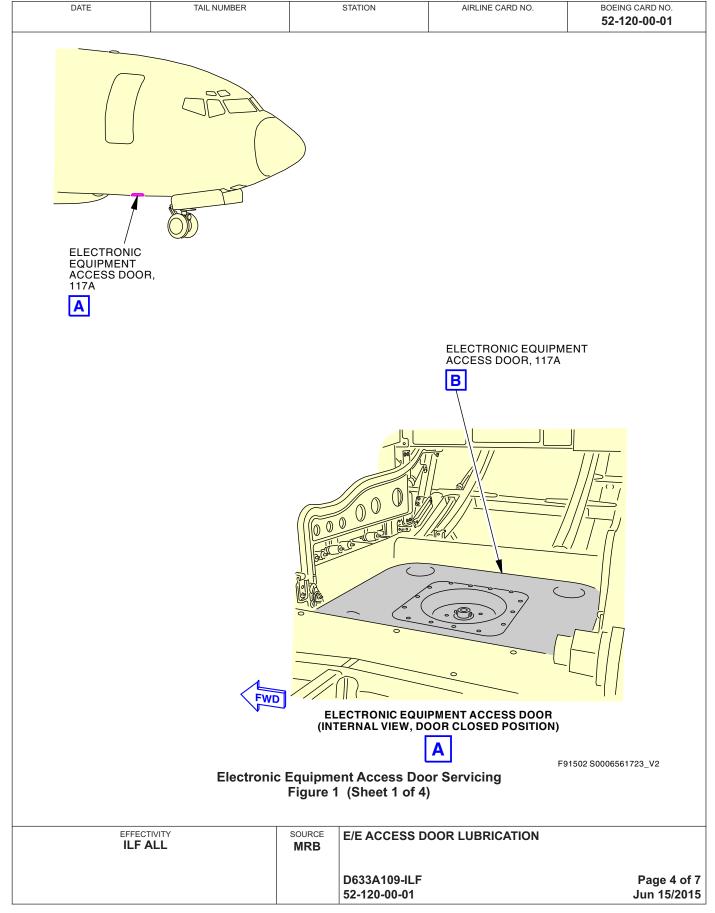




	DATE		TAIL NUMBER STATION AIRLINE CARD NO.		BOEING CARD NO. 52-120-00-01					
C.	Put	the Airplane Back to Its Usual Condition						MECH	INS	
	SUBTASK 12-25-41-410-003									
	(1)	Insta	II the su	upport plate [110]	on the lat	ch mechanism	[2] as follows (Figure 1):		
	(a) Install the washer [111] on the latch mechanism [2].									
		(b)		support plate [11 d 117] and the be			over the spacers [112,	114, 115,		
	(c) Install the screws [109].									
	SUBTA	SK 12-25	-41-010-003							
	(2)	Do th	ne follov	ving to close this	access pa	anel:				
		Num	<u>ber</u>	Name/Location	1					
		117A	W	Equipment Acce	ss Door C	Cover				
		(Figu	re 1):							
		(a)	Put the	cover [103] in its	s correct p	osition over the	e latch mechanism [2].			
		(b)	Install	the washer [108]	and the c	ollar [104] on th	ne latch mechanism [2].			
		(c)	Install				to attach the collar [104	4] to the		
		(d)			and screw	vs [102] to attac	ch the cover [103] to the	door.		
	SUBTA	` '	-41-410-004			1				
	(3)			ccess panel:						
	(-)	Num		Name/Location						
		117A								
						TASK ———				
					LIND OI	IASK ———				
			CTIVITY ALL		SOURCE MRB	E/E ACCESS D	OOR LUBRICATION			

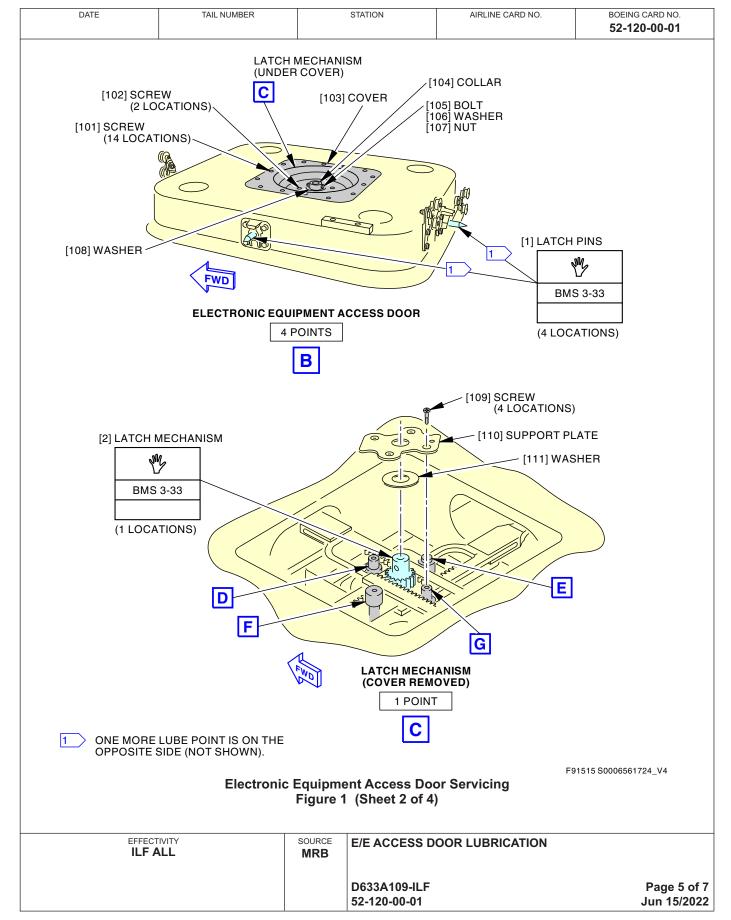
ILF















DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CARD NO. 52-120-00-01
	[114] S [113] B [115] S	JPPORTIN ATE PACER EARING SPACER		or Servicing	RTING
EFFEC ILF	TIVITY	SOURCE MRB	E/E ACCESS D	OOR LUBRICATION	
			D633A109-ILF 52-120-00-01		Page 6 of 7 Jun 15/2015





DATE	TAIL NUMBER		STATION	AIRLINE CARD NO.	BOEING CARD NO. 52-120-00-01				
	[110] SUP PLA	SCREW PORTING FE SPACER BEARING SPACER		[110] SUPPORTING PLATE	52-120-00-01				
	Electronic Equipment Access Door Servicing Figure 1 (Sheet 4 of 4)								
EFF IL	ECTIVITY F ALL	SOURCE MRB	E/E ACCESS D	OOR LUBRICATION					
			D633A109-ILF 52-120-00-01		Page 7 of 7 Jun 15/2015				