Sizing of high lift devices using Parapy: output file

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Planform file name: example_cant_attain

The HLDs were sized using the CL_max of the clean wing provided by the user in the main file. No external analysis was carried out.

With the given inputs, no valid design could be created to attain the required CL_max

Input parameters:

| _ | |
|-----------------|--------------|
| wing_span | = 15.0 m |
| root_chord | = 6.0 m |
| taper_inner | = 0.7 |
| taper_outer | = 0.3 |
| kink_position | = 4.0 m |
| flap_gap | = 0.2 m |
| sweep_deg | = 25.0 deg |
| dihedral_deg | $= 5.0 \deg$ |
| front_spar | = 0.2 x/c |
| rear_spar | = 0.6 x/c |
| outer_flap_lim | = 0.6 y/b |
| fuselage_radius | = 1.5 m |
| clmax | = 2.9 |
| twist | $= 0.0 \deg$ |
| max_deflection | = 30.0 m/s |
| speed | = 100.0 |
| airfoil_name | = ex1 |
| flap_type | = Slotted |

Output parameters:

| Cl_max clan | = 1.6 |
|---------------------|---------------|
| Delta Cl_max | = 1.3 |
| Flap hinge location | = 0.6 x/c |
| Flap deflection | = 30.0 deg |
| Stall AoA | = Unknown deg |
| Flaps per wing | = 2 |
| Flapped wing area | = 0 m^2 |

Other parameters

| Mach number | = 0.2915 |
|-------------|----------|
| Kink chord | = 4.2 m |
| Tip chord | = 1.26 m |

Cross section of airfoil with the flap system:

