Master's Thesis

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Contents

List of Figures

List of Tables

Abstract

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1 Introduction

2 Motivation

3 1D Simulation

Bamboo is a natural composite which is composed of fibers embedded in a matrix of parenchyma cells. Bamboo fibers are mainly composed of cellulose, hemicellulose and lignin [L.Y.Mwaikambo et al.]. Fibers are spread out across the cross-section in a graded manner with higher density towards the periphery. Also, the size of parenchyma cells decreases along the radially outward direction as the air content reduces. This results in axisymmetric areal density variation in the radial direction. [Plot showing the distribution of fibers.]

This can be modeled as the distribution of two materials, and air (captured in parenchyma cells). First material being the denser and stiffer fibers and second material being the parenchyma cellular material excluding air. The properties of fibers and parenchyma are taken from [Mannan et al., L.Y.Mwaikambo et al.].

Bamboo, like any other living organism, is a result of an evolutionary process. Survival of the fittest means that bamboo species is nature's best solution for some natural condition lead to the existence of bamboo. Bamboo grows tall up to 20m to rise above the other competing plantation. With such a slender structure, bending load due to high-speed tropical winds are a significant constraint which the evolution had overcome. Bamboo has a very high specific strength, which is also the desired attribute in industrial applications. It is desired to develop composites with high stiffness and lower weight.

3.1 Formulation

Therefore, we frame a constrained problem, optimizing the radial distribution of two material, with properties corresponding to that of fibers and parenchyma, in an annular cross-section composites. The objective function to optimize is specific strength subjected to the constraints of maximum stress and maximum bending moment. The dimensions of the composites are kept the same as in [Mannan et al.]. Also, the limits of max bending moment and stress are taken from [Mannan et al.]. The general problem formulation is written as

$$
\begin{aligned}
&\underset{\chi}{\text{maximize}} && strength(\chi) \\
&\text{subject to} && \sigma(r) \leq \sigma_{max} && \forall r \in [r_i, r_o] \\
& && M \leq M_{max}
\end{aligned}
$$

where, χ is the distribution of the two material in the domain whose inner radius is r_i and outer radius is r_o. σ is the stress in the longitudinal direction and M is the bending moment.

3.2 Implementation

Now, flexural rigidity is used as measure of strength which can be written as

$$
\begin{aligned}
EI(r) = \iint_R E(r) y^2 \, dA &= \int_{r_1}^{r_2} \int_0^{2\pi} E(r)(r\sin\theta)^2 (r dr d\theta) \\
&= \pi \int_{r_1}^{r_2} E(r) r^3 dr \qquad \text{Integrating over } \theta
\end{aligned}
\qquad (1)
$$
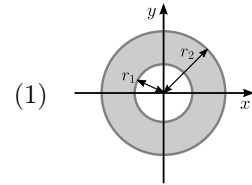
Therefore, specific flexural rigidity will be written as

$$
\begin{aligned}
strength(\chi) &= \frac{EI}{\rho} \\
&= \frac{\pi \sum_{r_i}^{r_o} E(r) r^3 \Delta r}{2\pi r \sum_{r_i}^{r_o} \rho(r) \Delta r} \\
&= \frac{\sum_{r_i}^{r_o} E(r) r^3}{2 \sum_{r_i}^{r_o} \rho(r) r}
\end{aligned}
\tag{2}
$$

where

$$
\begin{aligned}
E(r) &= \chi_1(r) E_1 + \chi_2(r) E_2 \\
\rho(r) &= \chi_1(r) \rho_1 + \chi_2(r) \rho_2
\end{aligned}
\tag{3}
$$

Here $\{E_1, \rho_1\}$ and $\{E_2, \rho_2\}$ are the material properties of the fibers and parenchyma respectively. χ_1 and χ_2 are the proportion of first and second material corresponding to fiber and parenchyma respectively.

Assuming small deformation, from Euler Bernoulli Beam Theory, we get strain ε_{xx} as
Put Schematics for Beam Theory

$$
\begin{aligned}
\varepsilon_{xx} &= \frac{\Delta x' - \Delta x}{\Delta x} \\
&= \frac{(R+y)\Delta\theta - R\Delta\theta}{R\Delta\theta} \\
&= \frac{y}{R}
\end{aligned}
\tag{4}
$$

Therefore, using constitutive relation, we get σ_{xx} as

$$
\sigma_{xx} = E\varepsilon_{xx} = \frac{E}{R}y
\tag{5}
$$

$$
\sigma = \frac{E}{R} r \sin\theta \quad (\because y = r\sin\theta)
\tag{6}
$$

Now, moment can be written as

$$
\begin{aligned}
M(x) &= \int \int y \cdot \sigma(x,y) \cdot dy dz \\
M &= \int_{r_i}^{r_o} \int_0^{2\pi} \sigma(r) r^2 sin\theta \, dr \, d\theta \qquad \text{For a particular } x \\
&= \int_{r_i}^{r_o} \int_0^{2\pi} \frac{E(r) r sin\theta}{R} r^2 sin\theta \, dr \, d\theta \qquad \text{(From eqn (6))} \\
&= \pi \int_{r_i}^{r_o} \frac{r^3 E(r)}{R} \, dr
\end{aligned}
\tag{7}
$$

3.3 Optimization Problem

Using (2), (6), (7), the optimization problem becomes

$$
\max_{\chi} \quad \frac{\sum_{r_i}^{r_o} E(r)r^3}{\sum_{r_i}^{r_o} \rho(r)r}
$$

$$
s.t. \quad \frac{rE(r)}{R} \leq \sigma_{max} \quad \forall r \in [r_i, r_o] \tag{8}
$$

$$
\sum_{r_i}^{r_o} r^3 E(r) \leq \frac{M_{max}R}{\pi \Delta r}
$$

3.4 Cell Size

For calculating parenchyma cell size, it is assumed that the cell thickness remains constant as we move from inner radius to the outer radius. The cell thickness has been determined from the dimensions given in Figure 4(a) in [?].
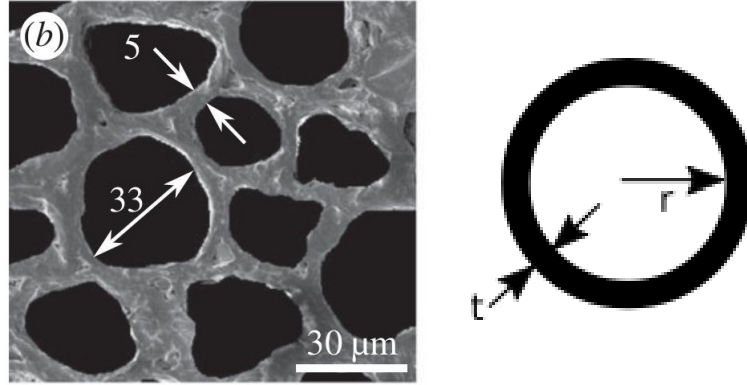
Figure 1: Cross-sectional micrograph showing the dimensions of typical parenchyma cell [?]

Parenchyma cells consists of cellular material and air packets. Assuming cell to be spherical, let v_a be the volume of air and v_p be the volume of cellular material. From fig. 1

$$v_a = \frac{4}{3}\pi r^3 \qquad\qquad\qquad v_p = \frac{4}{3}\pi$$

3.5 Results

3.5.1 Normal
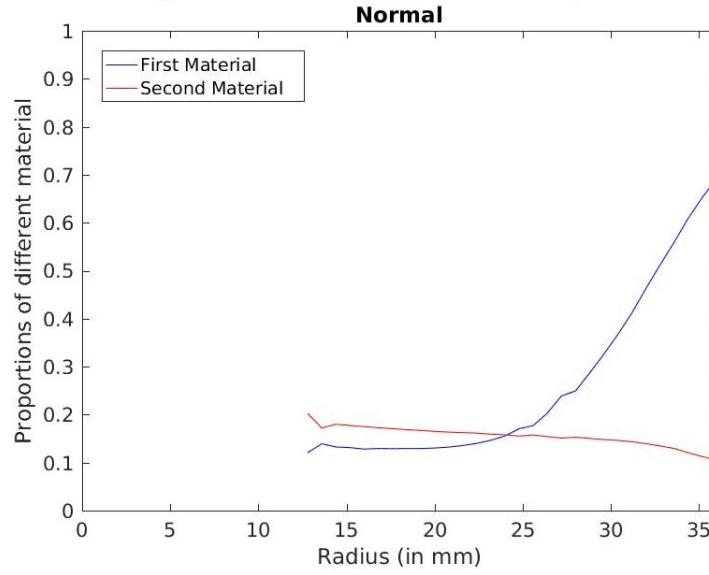
Figure 2: Axisymmetric radial distribution of fibres and parenchyma obtained for an annular rod with inner radius 12mm and outer radius 36mm with $\sigma_{max} = 28$MPa and $M_{max} = 580$ N-m
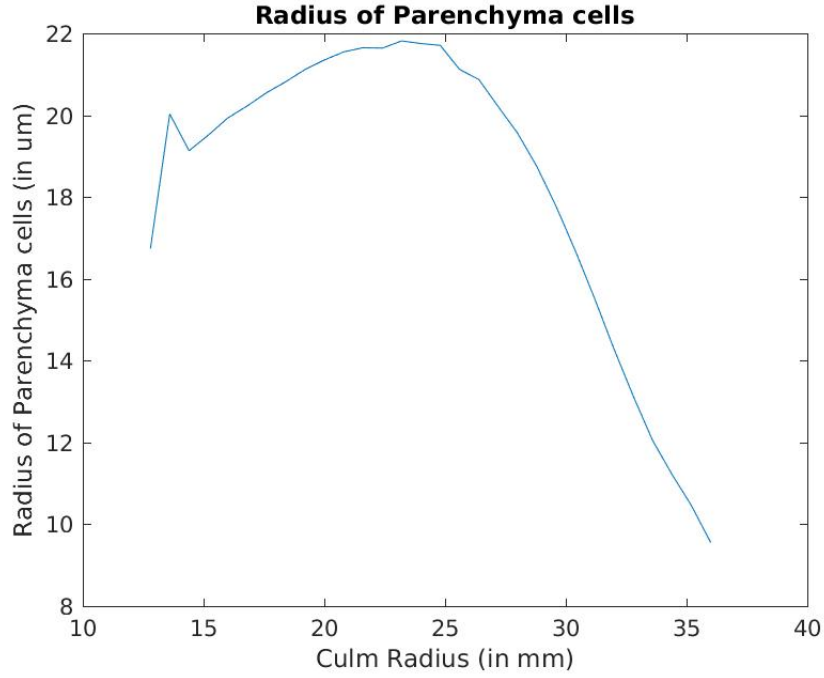
Figure 3: Radius of parenchyma cell size obtained for an annular rod with inner radius 12mm and outer radius 36mm with $\sigma_{max} = 28$MPa and $M_{max} = 580$ N-m

3.5.2 Extreme

We have also obtained results for different types of bamboos, like bamboos with large cross-sectional diameter but thin annular cross-section and bamboos with small diameter but solid cross section with no hollow center.

It should be noted that we didn't have the experimental data for the set of constraint for these cases of bamboos. To obtain these constraints we first provide a rough estimate using the Euler Bernoulli beam equation. Then if the obtained variation of parenchyma cell size goes beyond what is generally observed in nature that set of constraints are rejected and the optimization problem is solved for a new set of sequentially generated constraints. This process is repeated untill a variation of cell size consistent with nature is observed.
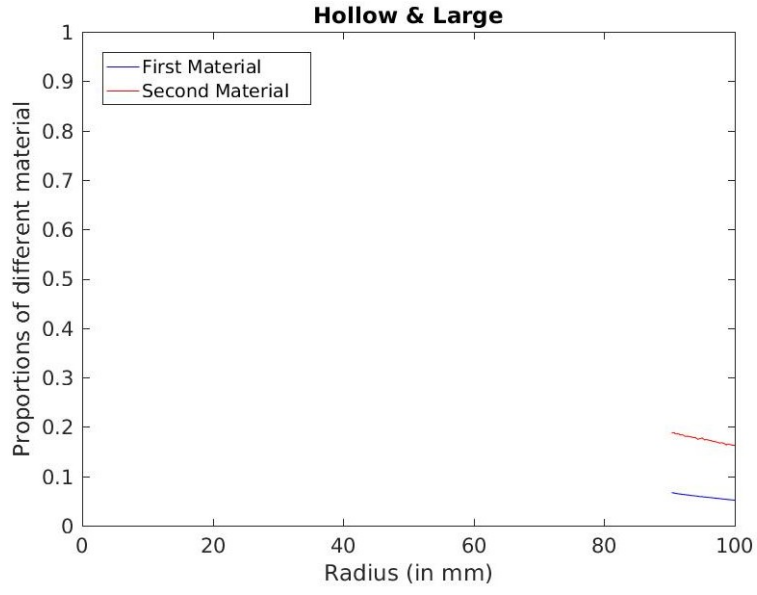
(a)
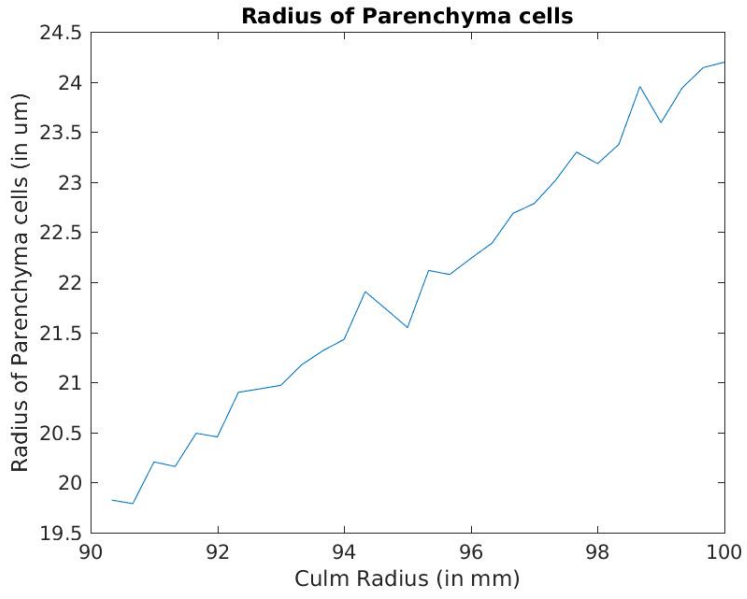
(b)

Figure 4: (a)Axisymmetric radial distribution of fibres and parenchyma, (b)Radius of parenchyma cell size obtained for an annular rod with inner radius 90mm and outer radius 100mm with $\sigma_{max} = 150$MPa and $M_{max} = 630$ N-m
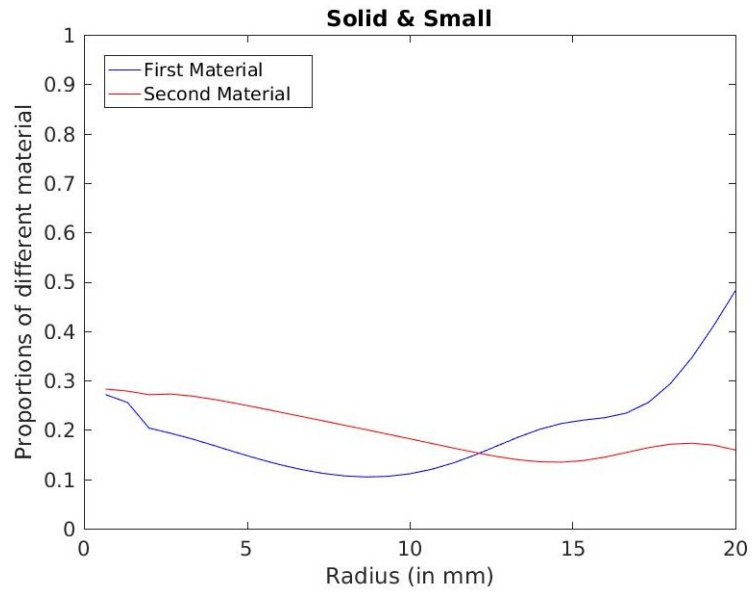
(a)
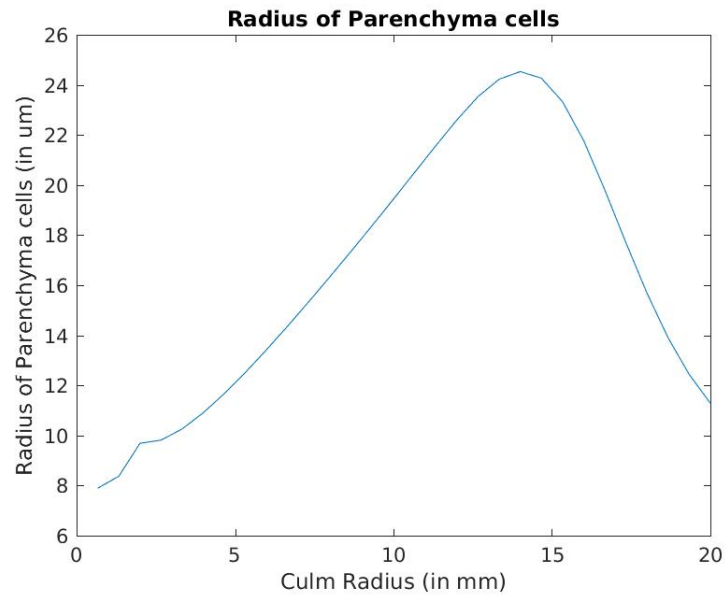
(b)

Figure 5: (a)Axisymmetric radial distribution of fibres and parenchyma, (b)Radius of parenchyma cell size obtained for an solid rod with radius 20mm with $\sigma_{max} = 24$MPa and $M_{max} = 420$ N-m

3.5.3 Height
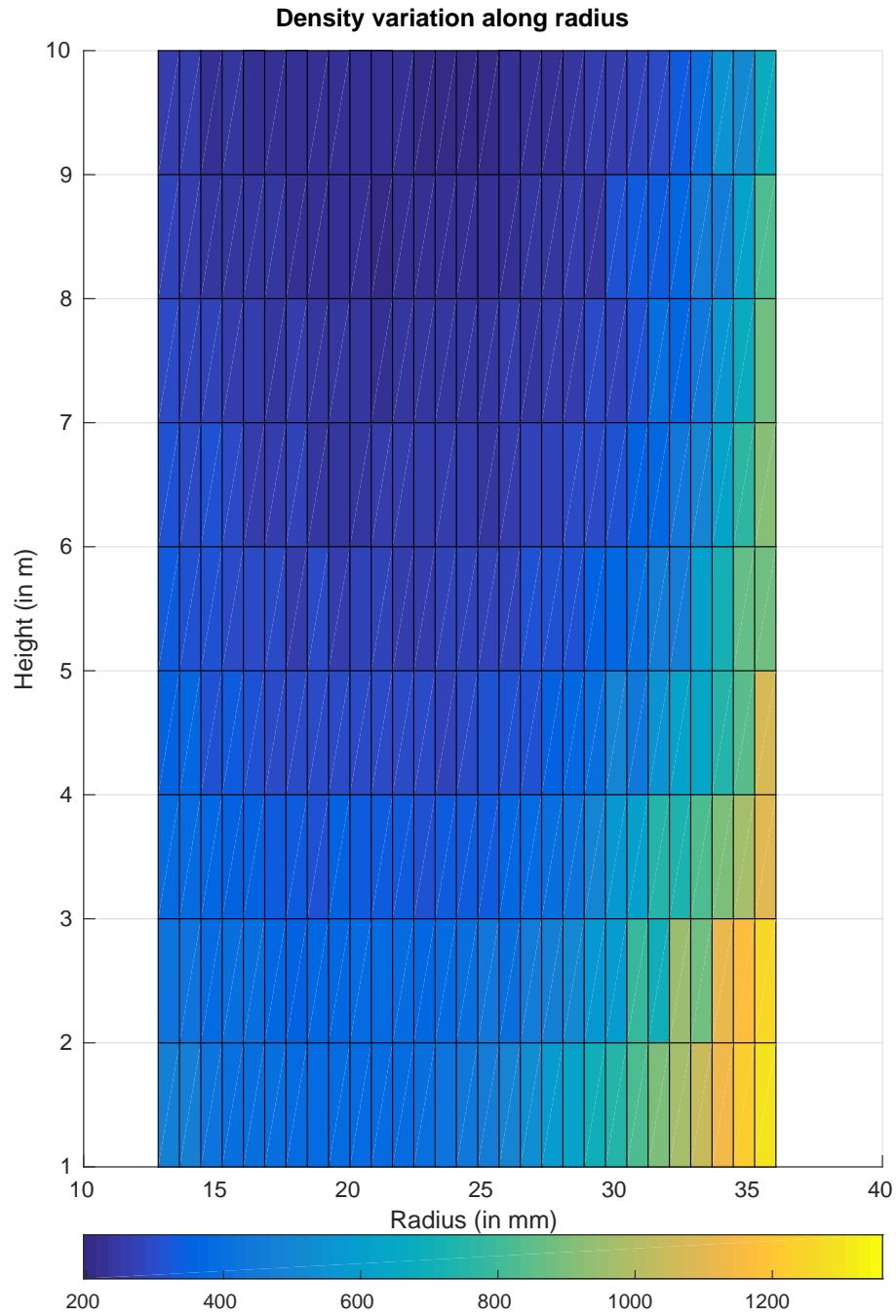
Figure 6: Density variation with height and radius obtained for an annular rod with inner radius 12mm and outer radius 36mm

3.6 Conclusion

Assuming axisymmetry, the radial distribution of constituent materials of bamboo is optimal for the constraints of maximum stress and bending moment that are applicable to bamboo under the loading conditions.

4 2D Simulation

In the previous chapter, we discussed the optimality of the one-dimensional distribution of the given material of the bamboo for some constraints. To discuss the optimality of the structure originating from this one-dimensional distribution, we will first develop a process for generating optimal structures using topology optimization and finite element methods. Once, we have a sense of what an optimal structure for the given material and loading constraint, we can compare them with the structures found in natural bamboo and be able to comment on its optimality. So, in this chapter, we will discuss the topology optimization of the microstructure for the given constraints on macro-structure.

4.1 Overview

First, an optimization problem is defined for minimizing the overall compliance of the structure with the constraints on volume. For calculating compliance ($\frac{1}{2}F^T U$), at every iteration, we find the displacement U for given force F and other boundary conditions on the macro domain. Displacements are determined by finite element analysis where the stiffness matrix for the macro domain elements is obtained by homogenization method. Each macro domain element is a called base cell and is further meshed to obtain the microstructure. To obtain the homogenized properties, another finite element analysis is done on this microdomain by applying periodic boundary conditions. It is the topology of this structure that we want to optimize. To achieve this, we link the optimality of the objective function to the topology of the microstructure through sensitivity analysis. For each element in the microdomain, a number is determined based on how critical is that element is for minimizing the objective function. After each iteration, the structure is updated based on the sensitivity numbers. This process is repeated until the volume constraint is achieved and the sensitivity numbers converge.

4.2 Optimization Problem

$$
\begin{aligned}
&\underset{\chi}{\text{minimize}}: \quad C = \frac{1}{2}\mathbf{F^T U} \\
&\text{subject to:} \quad \sum_{j=1}^{N} V_j x_j - V_f = 0 \qquad \text{where} \quad x_j = \{0,1\}
\end{aligned}
\tag{9}
$$

Here, $\mathbf{F}$ is the applied load, $\mathbf{U}$ is the displacement of the macrostructure, V_f is the specified volume fraction, V_j is the volume of jth element of the base cell which has a total of N elements. x_j are the design variables represent the relative density. Element is made of material 1 for $x_j = 1$ and material 2 for $x_j = 0$.

¡Schematic of the cantilever beam¿

¡Schematic of the microstructure¿

4.3 Finite Element Analysis

The optimization problem defined in (9) requires the finite element analysis to be done on two scales, one for macrostructure and another for the base cell. For the microstructure, since the design variables can be either 0 or 1, Solid Isotropic Material with Penalization model[**?**] is used for obtaining the material properties. Thus, the elasticity matrix can be written as

$$
\mathbf{D} = x_j^p \mathbf{D}^1 + (1 - x_j^p)\mathbf{D}^2
\tag{10}
$$

where $\mathbf{D}^1$ and $\mathbf{D}^2$ are the elasticity matrices for material 1 and 2 respectively, p is the penalization constant. For the macrodomain, we have the following FE equation

$$\mathbf{KU} = \mathbf{F} \tag{11}$$

where $\mathbf{K}$ is the assembled stiffness matrix for the macrostructure whose each component is calculated using

$$\mathbf{K}_i = \int_{V_i} \mathbf{B}_i^T \mathbf{D}_i^H \mathbf{B}_i dV_i \tag{12}$$

where $\mathbf{B}_i$ is the elemental strain-displacement matrix and $\mathbf{D}_i^H$ is the homogenized elasticity matrix calculated from the finite analysis of the base cell. The homogenization process[?] is discussed in the subsequent chapters. In this study, all base cells are assumed to be identical and therefore, equation (12) can be simplified as

$$\mathbf{K}_i = \int_{V_i} \mathbf{B}^T \mathbf{D}^H \mathbf{B} dV_i \tag{13}$$

4.4 Homogenization

4.4.1 1D Elasticity

$$\sigma^\varepsilon = E^\varepsilon \frac{\partial u^\varepsilon}{\partial x} \tag{14}$$

$$\frac{\partial \sigma^\varepsilon}{\partial x} + \gamma^\varepsilon = 0 \quad E^\varepsilon \, \gamma^\varepsilon \rightarrow macroscopically uniform \tag{15}$$

Inside each cell,

$$E^\varepsilon(x, \frac{x}{\varepsilon}) \quad = E(y) \tag{16}$$

$$\gamma^\varepsilon(x, \frac{x}{\varepsilon}) \quad = \gamma(y) \tag{17}$$

Let

$$u^\varepsilon(x) = u^0 x, y + \varepsilon u^1(x,y) + \varepsilon^2 u^2(x,y) + ... \tag{18}$$

$$\sigma^\varepsilon(x) = \sigma^0 x, y + \varepsilon \sigma^1(x,y) + \varepsilon^2 \sigma^2(x,y) + ... \tag{19}$$

4.4.2 Optimal Design of Elastic structures

$$\mathbf{b} \rightarrow \text{body forces}$$
$$\mathbf{t} \rightarrow \text{surface tractions}$$

Optimal choice of $\mathbb{C}_{ijkl} \in U_{ad} \leftarrow$ admissible set of elasticity
$\mathbb{C}_{ijkl}(\mathbf{x}) \forall \mathbf{x} \in \Omega$ has 21 independent components
$a_E(\mathbf{u}, \mathbf{v}) = \int_\Omega \mathbb{C}_{ijkl} \, \varepsilon_{kl}(\mathbf{u}) \, \varepsilon_{kl}(\mathbf{v}) d\mathbf{v} \rightarrow$ energy bilinear form
$L(\mathbf{v}) = \int_\Omega \mathbf{v} \, d\mathbf{x} + \int_{\partial\Omega_t} \mathbf{t} \cdot \mathbf{v} ds \rightarrow$ load linear form.

Minimum compliance problem:

$$minimize \quad\quad L(\mathbf{v}), \tag{20}$$

$$subject\,to \quad\quad \mathbb{C}_{ijkl} \in \mathbb{U}_{ad} \tag{21}$$

$$a_E(\mathbf{u}, \mathbf{v}) = L(\mathbf{v}) \quad \forall \mathbf{v} \in \mathbb{U} \tag{22}$$

16

where $\mathbb{U} \to$ kinematically admissible displacements.

For optimal shape design:

$$\mathbb{C}_{ijkl}(\mathbf{x}) = \chi(\mathbf{x})\overline{\mathbb{C}}_{ijkl}, \quad \text{where } \overline{\mathbb{C}}_{ijkl} \to \text{stiffness matrix of the material} \tag{23}$$

$$\chi(\mathbf{x}) = \begin{cases} 1 & \text{if } \mathbf{x} \in \Omega^m, \\ 0 & \text{if } \mathbf{x} \in \Omega \backslash \Omega^m \end{cases} \tag{24}$$

where $\Omega^m \to$ part of the domain occupied by the material.

For sizing problem:

$$\mathbb{C}_{ijkl}(\mathbf{x}) = h(\mathbf{x})\overline{\mathbb{C}}_{ijkl} \tag{25}$$

$$\int_\Omega \chi(\mathbf{x})d\mathbf{x} = V_f \tag{26}$$

$$\& \quad \int_\Omega h(\mathbf{x})d\mathbf{x} = V_f. \tag{27}$$

where $h(x)$ is a sizing function.

Traditionally shape design problems are initiated in the following manner:

$$Ref\ doamin : \Omega_0 \in \mathbb{R}^3 \tag{28}$$

$$\underline{\phi} : \Omega_0 \to \phi(\Omega_0) \text{is a diffeomorphism.} \tag{29}$$

$$L(\mathbf{v}) = \int_{\Omega_0} \mathbf{f} \cdot \mathbf{v} |det(D\underline{\phi}^{-1})| d\mathbf{x} + \int_{\partial\Omega_t} \mathbf{t} \cdot \mathbf{v} |det(D\underline{\phi}^{-1})| ds \tag{30}$$

$$a_E = \int_\Omega \mathbb{C}_{ijkl}(\mathbf{x}\varepsilon_{kl}(\mathbf{v})\varepsilon_{ij}(\mathbf{v})d\mathbf{x}$$
$$= \int_{\Omega_0} \mathbb{C}_{ijkl}\varepsilon_{kl}(\mathbf{v})\varepsilon_{ij}(\mathbf{v})|det(D\underline{\phi}^{-1})|d\mathbf{x} \tag{31}$$

Now,

$$\mathbb{C}_{ijkl}\varepsilon_{kl} = \mathbb{C}_{ijkl}\frac{1}{2}(u_{k,l} + u_{l,k})$$
$$= \frac{1}{2}\mathbb{C}_{ijkl}u_{k,l} + \frac{1}{2}\mathbb{C}_{ijlk}u_{l,k} \tag{32}$$
$$= \mathbb{C}_{ijkl}u_{k,l}$$

$$a_E = \int_{\Omega_0} \mathbb{C}_{ijkl}u_{k,l}(\mathbf{u})u_{i,j}(\mathbf{v})|det(D\underline{\phi}^{-1}|d\mathbf{x}$$
$$= \int_{\Omega_0} \mathbb{C}_{ijkl}\frac{\partial u_k}{\partial \mathbf{x}_m}(D\underline{\phi}^{-1})_{ml}\frac{\partial u_i}{\partial \mathbf{x}_p}(D\phi^{-1})_{pj}|det(D\underline{\phi}^{-1})|d\mathbf{x} \tag{33}$$

$$\Rightarrow \mathbb{C}_{ijkl}(D\underline{\phi}^{-1})_{ml}(D\underline{\phi}^{-1})_{pj}|det(D\underline{\phi}^{-1})| = \bar{\mathbb{C}}_{ipkm} \tag{34}$$

$$\bar{\mathbb{C}}_{ijkl} = \mathbb{C}_{ipkm}(D\underline{\phi}^{-1})_{lm}(D\underline{\phi}^{-1})_{jp}|det(D\underline{\phi}^{-1})| \tag{35}$$

Treating $\underline{\phi}$ as a design variable is tedious.

4.4.3 Homogenization method

$$E_{ijkl}^{\varepsilon}(\mathbf{x}) = E_{ijkl}(\mathbf{x}, \mathbf{y}), \qquad \mathbf{y} = \frac{\mathbf{x}}{\varepsilon} \tag{36}$$

The tensor E_{ijkl}^{ε} is a material constant which satisfies the symmetry condition and is assumed to satisfy strong ellipticity condition for every $\mathbf{x}$.

$$\Rightarrow E_{ijkl}^{\varepsilon} = E_{jikl}^{\varepsilon} = E_{ijlk}^{\varepsilon} = E_{klij}^{\varepsilon} \tag{37}$$

$$E_{ijkl}^{\varepsilon}(\mathbf{x})\mathbf{X}_{ij}\mathbf{X}_{kl} \geq m\mathbf{X}_{ij}\mathbf{X}_{ij} \qquad \text{for some } m > 0 \,\&\, \forall \quad \mathbf{X}_{ij} = \mathbf{X}_{ji} \tag{38}$$

Let the domain Ω has a boundary Γ. Let $\mathbf{f}$ be the body force acting on Ω and $\mathbf{t}$ be the traction acting on Γ_t part of the boundary Γ. Also, let Γ_D be the part of boundary on which displacement is defined. Then the displacement $\mathbf{u}^{\varepsilon}$ can be obtained as the solution to the following minimization problem

$$\min_{\mathbf{v}^{\varepsilon} \in U} \quad F^{\varepsilon}(\mathbf{v}^{\varepsilon}), \tag{39}$$

where F^{ε} is total potential energy given as

$$F^{\varepsilon}(\mathbf{v}^{\varepsilon}) = \frac{1}{2}\int_{\Omega} E_{ijkl}^{\varepsilon}\varepsilon_{kl}(\mathbf{v}^{\varepsilon})\varepsilon_{ij}(\mathbf{v}^{\varepsilon})dx - \int_{\Omega} \mathbf{f}\cdot\mathbf{v}^{\varepsilon}dx - \int_{\Gamma_t}\mathbf{t}\cdot\mathbf{v}^{\varepsilon}ds \tag{40}$$

and $\mathcal{U}$ is the set of admissible displacements defined such that

$$\mathcal{U} = \{\mathbf{v} = v_i\mathbf{e}_i : v_i \in H^1(\Omega) \text{ and } \mathbf{v} \in \mathcal{G} \text{ on } \Gamma_D\} \tag{41}$$

where $\mathcal{G}$ is set of displacement defined along the boundary Γ_D.
Let

$$\mathbf{v}^{\varepsilon}(\mathbf{x}) = \mathbf{v}_0(\mathbf{x}) + \varepsilon\mathbf{v}_1(\mathbf{x}, \mathbf{y}), \qquad \mathbf{y} = \frac{\mathbf{x}}{\varepsilon}. \tag{42}$$

Using chain rule for functions in two variables

$$
\begin{aligned}
\frac{\partial f(\mathbf{x}, \mathbf{y}(\mathbf{x}))}{\partial \mathbf{x}} &= \frac{\partial f}{\partial \mathbf{x}} + \frac{\partial f}{\partial \mathbf{y}}\frac{\partial \mathbf{y}}{\partial \mathbf{x}} \\
&= \frac{\partial f}{\partial \mathbf{x}} + \frac{1}{\varepsilon}\frac{\partial f}{\partial \mathbf{y}}
\end{aligned}
\tag{43}
$$

Using above two equations, we can write the linerized strain as

$$
\begin{aligned}
\epsilon_{ij}(\mathbf{v}^{\varepsilon}(\mathbf{x})) &= \frac{\partial(v_{0i}(\mathbf{x}) + \varepsilon v_{1i}(\mathbf{x}, \mathbf{y}))}{\partial x_j} \\
&= \frac{\partial v_{0i}}{\partial x_j} + \varepsilon\left\{\frac{\partial v_{1i}}{\partial x_j} + \frac{1}{\varepsilon}\frac{\partial v_{1i}}{\partial y_j}\right\} \\
&\approx \frac{\partial v_{0i}}{\partial x_j} + \frac{\partial v_{1i}}{\partial y_j} \qquad \leftarrow \{\varepsilon << 1\}
\end{aligned}
\tag{44}
$$

Therefore, equation (40) can be written as

$$
\begin{aligned}
F^{\varepsilon}(\mathbf{v}^{\varepsilon}) = \frac{1}{2}\int_{\Omega} E_{ijkl}^{\varepsilon}\left(\frac{\partial v_{0k}}{\partial x_l} + \frac{\partial v_{1k}}{\partial y_l}\right)\left(\frac{\partial v_{0i}}{\partial x_j} + \frac{\partial v_{1i}}{\partial y_j}\right)dx \\
- \int_{\Omega}\mathbf{f}\cdot\mathbf{v}_0 dx - \int_{\Gamma_t}\mathbf{t}\cdot\mathbf{v}_0 ds + \varepsilon R^{\varepsilon}(\mathbf{v}_0, \mathbf{v}_1)
\end{aligned}
\tag{45}
$$

Here, R^ε is the contribution of $\varepsilon\mathbf{v}_1$ in the calculation of energy from body force and traction. Using

$$\lim_{\varepsilon\to 0}\int_\Omega \Phi(x,x/\varepsilon)dx = \frac{1}{|Y|}\int_\Omega\int_Y \Phi(x,y)dy dx, \tag{46}$$

we get,

$$\lim_{\varepsilon\to 0} F^\varepsilon(\mathbf{v}^\varepsilon) = F(\mathbf{v}_0,\mathbf{v}_1)$$
$$= \frac{1}{2|Y|}\int_\Omega\int_Y E_{ijkl}(x,y)\left(\frac{\partial v_{0k}}{\partial x_l} + \frac{\partial v_{1k}}{\partial y_l}\right)\left(\frac{\partial v_{0i}}{\partial x_j} + \frac{\partial v_{1i}}{\partial y_j}\right)dy\, dx \tag{47}$$
$$- \int_\Omega \mathbf{f}\cdot\mathbf{v}_0 dx - \int_{\Gamma_t}\mathbf{t}\cdot\mathbf{v}_0 ds$$

A minimizer $\{\mathbf{u}_0,\mathbf{u}_1\}$ of the functional F, follow the following equations:

$$\frac{1}{|Y|}\int_\Omega\int_Y E_{ijkl}(x,y)\left(\frac{\partial u_{0k}}{\partial x_l} + \frac{\partial u_{1k}}{\partial y_l}\right)\left(\frac{\partial v_{0i}}{\partial x_j}\right)dy\, dx$$
$$= \int_\Omega \mathbf{f}\cdot\mathbf{v}_0 dx + \int_{\Gamma_t}\mathbf{t}\cdot\mathbf{v}_0 ds \qquad \text{for every } \mathbf{v}_0 \tag{48}$$

$$\frac{1}{|Y|}\int_\Omega\int_Y E_{ijkl}(x,y)\left(\frac{\partial u_{0k}}{\partial x_l} + \frac{\partial u_{1k}}{\partial y_l}\right)\left(\frac{\partial v_i}{\partial x_j}\right)dy\, dx = 0, \qquad \text{for every } \mathbf{v}_1 \tag{49}$$

Now, from localizing u_{1k}

$$u_{1k}(x,y) = -\chi_k^{pq}(y)\frac{\partial u_{0p}}{\partial x_q}(x), \tag{50}$$

$$\Rightarrow \int_\Omega\int_Y E_{ijkl}(x,y)\left(\frac{\partial u_{0k}}{\partial x_l} - \frac{\partial \chi_k^{pq}}{\partial y_l}\frac{\partial u_{0p}}{\partial x_q}\right)\frac{\partial v_i}{\partial x_j}dy\, dx = 0$$

$$\int_\Omega\int_Y \left(E_{ijkl}\frac{\partial u_{0k}}{\partial x_l} - E_{ijkl}\frac{\partial \chi_k^{pq}}{\partial y_l}\frac{\partial u_{0p}}{\partial x_q}\right)\frac{\partial v_i}{\partial x_j}dy\, dx = 0$$

$$\int_\Omega\int_Y \left(E_{ijkl}\frac{\partial u_{0k}}{\partial x_l} - E_{ijpq}\frac{\partial \chi_p^{kl}}{\partial y_q}\frac{\partial u_{0k}}{\partial x_l}\right)\frac{\partial v_i}{\partial x_j}dy\, dx = 0$$

$$\int_\Omega\int_Y \frac{\partial u_{0k}}{\partial x_l}\left(E_{ijkl} - E_{ijpq}\frac{\partial \chi_p^{kl}}{\partial y_q}\right)\frac{\partial v_i}{\partial x_j}dy\, dx = 0$$

$$\int_\Omega \frac{\partial u_{0k}}{\partial x_l}dx\cdot\int_Y \left(E_{ijkl} - E_{ijpq}\frac{\partial \chi_p^{kl}}{\partial y_q}\right)\frac{\partial v_i}{\partial x_j}dy = 0$$

$$\Rightarrow \int_Y \left(E_{ijkl} - E_{ijpq}\frac{\partial \chi_p^{kl}}{\partial y_q}\right)\frac{\partial v_i}{\partial x_j}dy = 0 \qquad \text{for k, l} = 1 \text{ and } 2, \tag{51}$$

Similarly, substituting equation (50) in (48) gives the homogenized equation.

$$\text{LHS} = \frac{1}{|Y|} \int_\Omega \int_Y E_{ijkl}(x,y) \left(\frac{\partial u_{0k}}{\partial x_l} + \frac{\partial u_{1k}}{\partial y_l} \right) \left(\frac{\partial v_{0i}}{\partial x_j} \right) dy\, dx$$

$$= \frac{1}{|Y|} \int_\Omega \int_Y \left(E_{ijkl} \frac{\partial u_{0k}}{\partial x_l} - E_{ijpq} \frac{\partial \chi_p^{kl}}{\partial y_q} \frac{\partial u_{0k}}{\partial x_l} \right) \frac{\partial v_{0i}}{\partial x_j} dy\, dx$$

$$= \frac{1}{|Y|} \int_\Omega \left\{ \int_Y \left(E_{ijkl} - E_{ijpq} \frac{\partial \chi_p^{kl}}{\partial y_q} \right) dy \right\} \frac{\partial u_{0k}}{\partial x_l} \frac{\partial v_{0i}}{\partial x_j} dx$$

$$= \int_\Omega E_{ijkl}^H(x) \frac{\partial u_{0k}}{\partial x_l} \frac{\partial v_{0i}}{\partial x_j} dx$$

Homogenized equation

$$\int_\Omega E_{ijkl}^H(x) \frac{\partial u_{0k}}{\partial x_l} \frac{\partial v_{0i}}{\partial x_j} dx = \int_\Omega \mathbf{f} \cdot \mathbf{v}_0 dx + \int_{\Gamma_t} \mathbf{t} \cdot \mathbf{v}_0 ds \qquad \text{for every } \mathbf{v}_0 \tag{52}$$

where $E_{ijkl}^H(x)$ is

$$\boxed{E_{ijkl}^H = \frac{1}{|Y|} \int_Y \left(E_{ijkl} - E_{ijpq} \frac{\partial \chi_p^{kl}}{\partial y_q} \right) dy} \tag{53}$$

Now, Define

$$a_H(\mathbf{u}, \mathbf{v}) = \int_\Omega E_{ijkl}^H(\mathbf{x}) \frac{\partial u_k}{\partial x_l} \frac{\partial v_i}{\partial x_j} dx, \tag{54}$$

$$a_Y(\chi^{kl}, \mathbf{v}) = \int_Y E_{ijpq}(\mathbf{x}, \mathbf{y}) \frac{\partial \chi_p^{kl}}{\partial y_q} \frac{\partial v_i}{\partial y_j} dy, \tag{55}$$

$$L_Y^{kl}(\mathbf{v}) = \int_Y E_{ijkl} \frac{\partial v_i}{\partial y_j} dy \tag{56}$$

At microscopic level, we have

$$a_Y(\chi^{kl}, \mathbf{v}) = L_Y^{kl}(\mathbf{v}) \qquad \forall \mathbf{v} \in \mathcal{U}_Y, \tag{57}$$

At macroscopic level, we have

$$a_H(\mathbf{u}, \mathbf{v}) = L(\mathbf{v}) \qquad \forall \mathbf{v} \in \mathcal{U}_0 \tag{58}$$

where $\mathcal{U}_0$ is homogeneous case of $\mathcal{U}$, i.e., $\mathbf{g} = 0$.

4.4.4 Implementation 2D Homogenization

Basic homogenization equation,

$$u_{1i}(\mathbf{x}, \mathbf{y}) = -\chi_i^{pq} \frac{\partial u_{0p}(\mathbf{x})}{\partial x_q} \tag{59}$$

Solve χ_p^{kl} from:

$$\int_Y \left(E_{ijkl} - E_{ijpq} \frac{\partial \chi_p^{kl}}{\partial y_q} \right) \frac{\partial v_{1i}}{\partial y_j} dy = 0 \tag{60}$$

Compute:

$$E_{ijkl}^H = \frac{1}{|Y|} \int_Y \left(E_{ijkl} - E_{ijpq} \frac{\partial \chi_p^{kl}}{\partial y_q} \right) dy \tag{61}$$

20

4.4.5 Examples

Consider: k=1, l=1

$$\int_Y E_{ijkl} \frac{\partial v_i}{\partial y_j}\, dy = \int_Y E_{ij11} \frac{\partial v_i}{\partial y_j}\, dy$$
$$= \int_Y \left(E_{1111} \frac{\partial v_1}{\partial y_1} + E_{2211} \frac{\partial v_2}{\partial y_2} \right) dy \tag{62}$$

$$\int_Y E_{ijpq} \frac{\partial \chi_p^{kl}}{\partial y_q} \frac{\partial v_i}{\partial y_j}\, dy = \int_Y E_{ijpq} \frac{\partial \chi_p^{11}}{\partial y_q} \frac{\partial v_i}{\partial y_j}\, dy$$

$$= \int_Y \Bigg\{ E_{11pq} \frac{\partial \chi_p^{11}}{\partial y_q} \frac{\partial v_1}{\partial y_1} + E_{12pq} \frac{\partial \chi_p^{11}}{\partial y_q} \frac{\partial v_1}{\partial y_2}$$
$$+ E_{21pq} \frac{\partial \chi_p^{11}}{\partial y_q} \frac{\partial v_2}{\partial y_1} + E_{22pq} \frac{\partial \chi_p^{11}}{\partial y_q} \frac{\partial v_2}{\partial y_2} \Bigg\} dy$$

$$= \int_Y \Bigg\{ \left(E_{1111} \frac{\partial \chi_1^{11}}{\partial y_1} + E_{1112} \frac{\partial \chi_1^{11}}{\partial y_2} + E_{1121} \frac{\partial \chi_2^{11}}{\partial y_1} + E_{1122} \frac{\partial \chi_2^{11}}{\partial y_2} \right) \frac{\partial v_1}{\partial y_1}$$
$$+ \left(E_{1211} \frac{\partial \chi_1^{11}}{\partial y_1} + E_{1212} \frac{\partial \chi_1^{11}}{\partial y_2} + E_{1221} \frac{\partial \chi_2^{11}}{\partial y_1} + E_{1222} \frac{\partial \chi_2^{11}}{\partial y_2} \right) \frac{\partial v_1}{\partial y_2}$$
$$+ \left(E_{2111} \frac{\partial \chi_1^{11}}{\partial y_1} + E_{2112} \frac{\partial \chi_1^{11}}{\partial y_2} + E_{2121} \frac{\partial \chi_2^{11}}{\partial y_1} + E_{2122} \frac{\partial \chi_2^{11}}{\partial y_2} \right) \frac{\partial v_2}{\partial y_1}$$
$$+ \left(E_{2211} \frac{\partial \chi_1^{11}}{\partial y_1} + E_{2212} \frac{\partial \chi_1^{11}}{\partial y_2} + E_{2221} \frac{\partial \chi_2^{11}}{\partial y_1} + E_{2222} \frac{\partial \chi_2^{11}}{\partial y_2} \right) \frac{\partial v_2}{\partial y_2} \Bigg\} dy$$

$$= \int_Y \Bigg\{ \left(E_{1111} \frac{\partial \chi_1^{11}}{\partial y_1} + E_{1112} \cancelto{0}{\frac{\partial \chi_1^{11}}{\partial y_2}} + E_{1121} \cancelto{0}{\frac{\partial \chi_2^{11}}{\partial y_1}} + E_{1122} \frac{\partial \chi_2^{11}}{\partial y_2} \right) \frac{\partial v_1}{\partial y_1}$$
$$+ \left(E_{1211} \cancelto{0}{\frac{\partial \chi_1^{11}}{\partial y_1}} + E_{1212} \frac{\partial \chi_1^{11}}{\partial y_2} + E_{1221} \frac{\partial \chi_2^{11}}{\partial y_1} + E_{1222} \cancelto{0}{\frac{\partial \chi_2^{11}}{\partial y_2}} \right) \left(\frac{\partial v_1}{\partial y_2} + \frac{\partial v_2}{\partial y_1} \right)$$
$$+ \left(E_{2211} \frac{\partial \chi_1^{11}}{\partial y_1} + E_{2212} \cancelto{0}{\frac{\partial \chi_1^{11}}{\partial y_2}} + E_{2221} \cancelto{0}{\frac{\partial \chi_2^{11}}{\partial y_1}} + E_{2222} \frac{\partial \chi_2^{11}}{\partial y_2} \right) \frac{\partial v_2}{\partial y_2} \Bigg\} dy$$

$$= \int_Y \Bigg\{ \left(E_{1111} \frac{\partial \chi_1^{11}}{\partial y_1} + E_{1122} \frac{\partial \chi_2^{11}}{\partial y_2} \right) \frac{\partial v_1}{\partial y_1}$$
$$+ E_{1212} \left(\frac{\partial \chi_1^{11}}{\partial y_2} + \frac{\partial \chi_2^{11}}{\partial y_1} \right) \left(\frac{\partial v_1}{\partial y_2} + \frac{\partial v_2}{\partial y_1} \right)$$
$$+ \left(E_{2211} \frac{\partial \chi_1^{11}}{\partial y_1} + E_{2222} \frac{\partial \chi_2^{11}}{\partial y_2} \right) \frac{\partial v_2}{\partial y_2} \Bigg\} dy \tag{63}$$

Therefore, using equations (57) , (63) and (62) for k=1, l=1 we have:

$$
\int_Y \left\{ \left(E_{1111} \frac{\partial \chi_1^{11}}{\partial y_1} + E_{1122} \frac{\partial \chi_2^{11}}{\partial y_2} \right) \frac{\partial v_1}{\partial y_1} \right.
$$
$$
+ E_{1212} \left(\frac{\partial \chi_1^{11}}{\partial y_2} + \frac{\partial \chi_2^{11}}{\partial y_1} \right) \left(\frac{\partial v_1}{\partial y_2} + \frac{\partial v_2}{\partial y_1} \right)
$$
$$
\left. + \left(E_{2211} \frac{\partial \chi_1^{11}}{\partial y_1} + E_{2222} \frac{\partial \chi_2^{11}}{\partial y_2} \right) \frac{\partial v_2}{\partial y_2} \right\} dy =
$$
$$
\int_Y \left(E_{1111} \frac{\partial v_1}{\partial y_1} + E_{2211} \frac{\partial v_2}{\partial y_2} \right) dy
$$

(64)

From equation (61), we can write

$$
E_{1111}^H = \quad \frac{1}{|Y|} \int_Y \left(E_{1111} - E_{1111} \frac{\partial \chi_1^{11}}{\partial y_1} - E_{1122} \frac{\partial \chi_2^{11}}{\partial y_2} \right) dy \tag{65}
$$

$$
E_{2211}^H = \quad \frac{1}{|Y|} \int_Y \left(E_{2211} - E_{2211} \frac{\partial \chi_1^{11}}{\partial y_1} - E_{2222} \frac{\partial \chi_2^{11}}{\partial y_2} \right) dy \tag{66}
$$

$$
E_{1211}^H = \quad -\frac{1}{|Y|} \int_Y \left(E_{1212} \frac{\partial \chi_1^{11}}{\partial y_2} + E_{1221} \frac{\partial \chi_2^{11}}{\partial y_1} \right) dy \tag{67}
$$

Let $\chi_1^{11} = \Phi_1, \chi_2^{11} = \Phi_2$ and $E_{1111} = D_{11}, E_{2222} = D_{22}, E_{1212} = D_{66}, E_{1122} = E_{2211} = D_{12}$

$$
\int_Y \left\{ \left(D_{11} \frac{\partial \Phi_1^{11}}{\partial y_1} + D_{12} \frac{\partial \Phi_2^{11}}{\partial y_2} \right) \frac{\partial v_1}{\partial y_1} \right.
$$
$$
+ D_{66} \left(\frac{\partial \Phi_1^{11}}{\partial y_2} + \frac{\partial \Phi_2^{11}}{\partial y_1} \right) \left(\frac{\partial v_1}{\partial y_2} + \frac{\partial v_2}{\partial y_1} \right)
$$
$$
\left. + \left(D_{12} \frac{\partial \Phi_1^{11}}{\partial y_1} + D_{22} \frac{\partial \Phi_2^{11}}{\partial y_2} \right) \frac{\partial v_2}{\partial y_2} \right\} dy =
$$
$$
\int_Y \left(D_{11} \frac{\partial v_1}{\partial y_1} + D_{12} \frac{\partial v_2}{\partial y_2} \right) dy
$$

(68)

Also,

$$
D_{11}^H = \frac{1}{|Y|} \int_Y \left(D_{11} - D11 \frac{\partial \Phi_1}{\partial y_1} - D_{12} \frac{\partial \Phi_2}{\partial y_2} \right) dy \tag{69}
$$

Rearranging Eq. (68)

$$
\int_Y \left\{ \frac{\partial v_1}{\partial y_1} \quad \frac{\partial v_2}{\partial y_2} \quad \frac{\partial v_1}{\partial y_2} + \frac{\partial v_2}{\partial y_1} \right\} \begin{bmatrix} D_{11} & D_{12} & 0 \\ D_{12} & D_{22} & 0 \\ 0 & 0 & D66 \end{bmatrix}
$$
$$
\times \begin{bmatrix} \frac{\partial \Phi_1}{\partial y_1} \\ \frac{\partial \Phi_2}{\partial y_2} \\ \frac{\partial \Phi_1}{\partial y_2} + \frac{\partial \Phi_2}{\partial y_1} \end{bmatrix} dY
$$
$$
= \int_Y \left\{ \frac{\partial v_1}{\partial y_1} \quad \frac{\partial v_2}{\partial y_2} \quad \frac{\partial v_1}{\partial y_2} + \frac{\partial v_2}{\partial y_1} \right\} \begin{bmatrix} D_{11} \\ D_{12} \\ 0 \end{bmatrix} dY
$$

(70)

22

Let us define

$$
\mathbf{b} =
\begin{bmatrix}
\frac{\partial}{\partial y_1} & 0 \\
0 & \frac{\partial}{\partial y_2} \\
\frac{\partial}{\partial y_1} & \frac{\partial}{\partial y_2}
\end{bmatrix}
\tag{71}
$$

and

$$
\mathbf{D} = \begin{bmatrix} \mathbf{d}_1 & \mathbf{d}_2 & \mathbf{d}_3 \end{bmatrix}
\tag{72}
$$

Then Eq (68), can be written as

$$
\int_Y \mathbf{v}^T \mathbf{b}^T \mathbf{D} \mathbf{b} \Phi \, dY = \int_Y \mathbf{v}^T \mathbf{b}^T \mathbf{d}_1 \qquad \forall \mathbf{v} \in \mathbf{V}_Y
\tag{73}
$$

and eq. (69) becomes:

$$
\boxed{D_{11}^H = \frac{1}{|Y|} \int_Y \left(D_{11} - \mathbf{d}_1^T \mathbf{b} \Phi \right) dy}
\tag{74}
$$

4.5   Periodic Boundary Conditions

The process of homogenization requires the displacements corresponding to uniform strain fields, i.e., $[\epsilon_{11}, \epsilon_{22}, 2\epsilon_{12}] = [1, 0, 0]^T, [0, 1, 0]^T, [0, 0, 1]^T$. Periodic boundary conditions are needed to solve the finite element problem for uniform strain fields.

If u and v are the components of displacements in p_1 and p_2 directions respectively, then from periodicity, we have

$$
\begin{aligned}
u(p_1, p_2) &= u(p_1 + P_1, p_2) = u(p_1, p_2 + P_2) = u(p_1 + P_1, p_2 + P_2), \\
v(p_1, p_2) &= v(p_1 + P_1, p_2) = v(p_1, p_2 + P_2) = v(p_1 + P_1, p_2 + P_2).
\end{aligned}
\tag{75}
$$

For cases $[1, 0, 0]^T$ and $[0, 1, 0]^T$, from symmetry of loading and geometry w.r.t. q_1, we have

$$
u(p_1, p_2^0) = u(p_1, p_2^0 + P_2),
\tag{76}
$$

$$
v(p_1, p_2^0) = -v(p_1, p_2^0 + P_2).
\tag{77}
$$

Similarly, using the symmetry about q_2, we have

$$
u(p_1^0, p_2) = -u(p_1^0 + P_1, p_2),
\tag{78}
$$

$$
v(p_1^0, p_2) = v(p_1^0 + P_1, p_2).
\tag{79}
$$

From eqs (83), (84), and (75), we get

$$
u(p_1^0, p_2) = u(p_1^0 + P_1, p_2) = 0,
\tag{80}
$$

$$
v(p_1, p_2^0) = v(p_1, p_2^0 + P_2) = 0.
\tag{81}
$$

Eqs (82), (85), (87) and (86) are the required periodic boundary conditions for strain fields $[1, 0, 0]^T$ and $[0, 1, 0]^T$.

Now, For the case of $[0, 0, 1]^T$, loading is anti-symmetric and geometry is symmetric w.r.t. q_1 and q_2, as a result, we have

$$
u(p_1, p_2^0) = -u(p_1, p_2^0 + P_2),
\tag{82}
$$

$$
v(p_1, p_2^0) = v(p_1, p_2^0 + P_2).
\tag{83}
$$

$$
u(p_1^0, p_2) = u(p_1^0 + P_1, p_2),
\tag{84}
$$

$$
v(p_1^0, p_2) = -v(p_1^0 + P_1, p_2).
\tag{85}
$$

From eqs (82), (85), and (75), we get

$$u(p_1, p_2^0) = u(p_1, p_2^0 + P_2) = 0, \tag{86}$$

$$v(p_1^0, p_2) = v(p_1^0 + P_1, p_2) = 0. \tag{87}$$

Eqs (83), (84), (87) and (86) are the required periodic boundary conditions for strain field $[0, 0, 1]^T$.

4.5.1 Implmentation

5 Results

[1] [2]

6 References

References

[1] John Doe. Title. *Journal*, 2017.

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