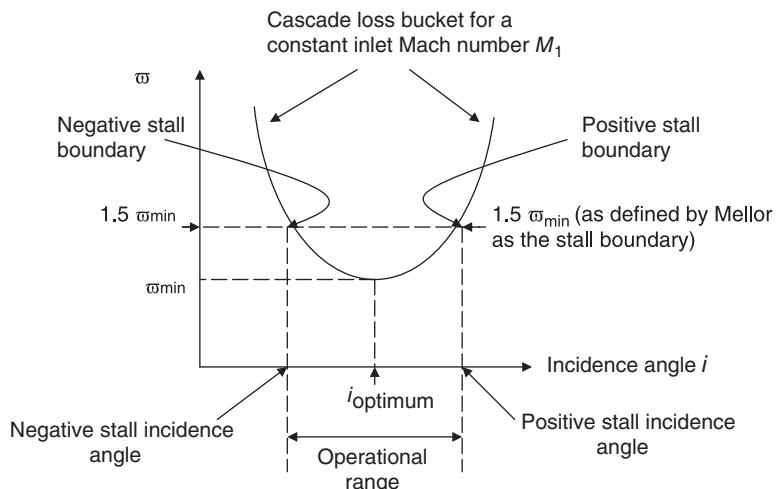


APPENDIX I

Cascade Data

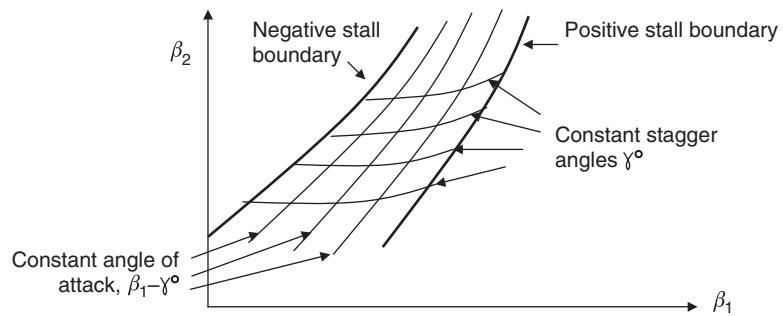
NACA-65 Series Cascade Data

Cascade total pressure loss data are plotted in the shape of a “bucket” in terms of incidence angle. A definition sketch is shown.

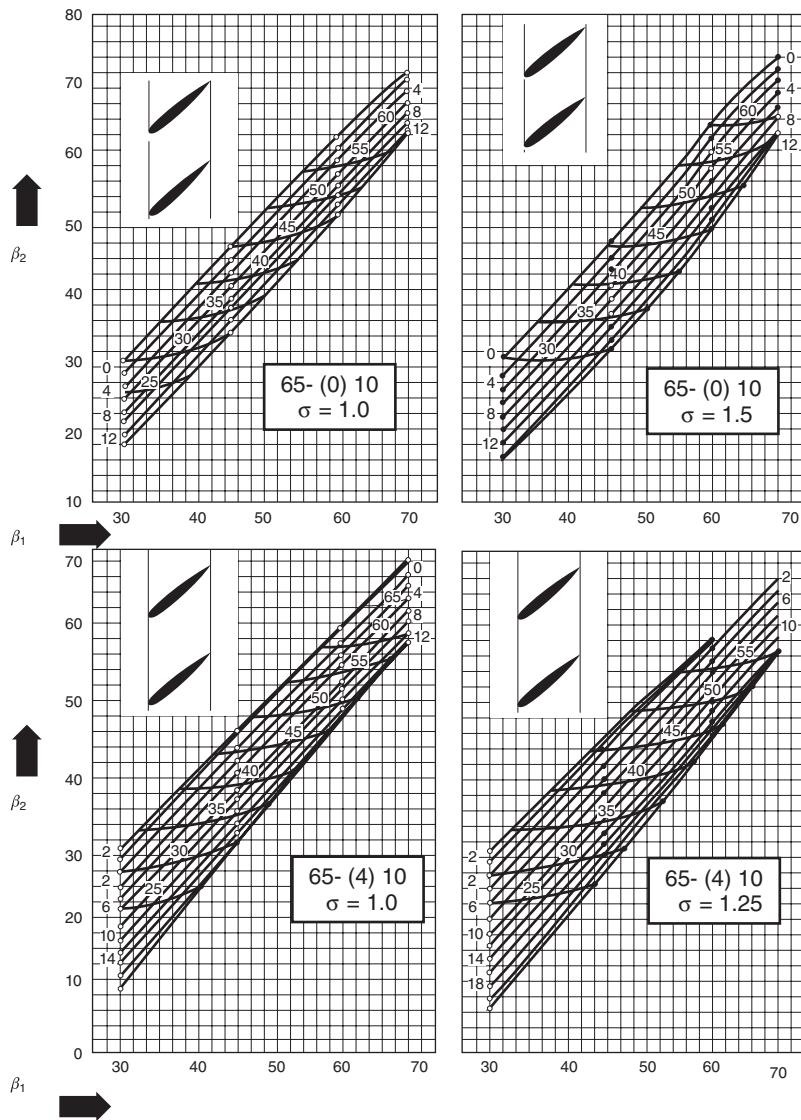


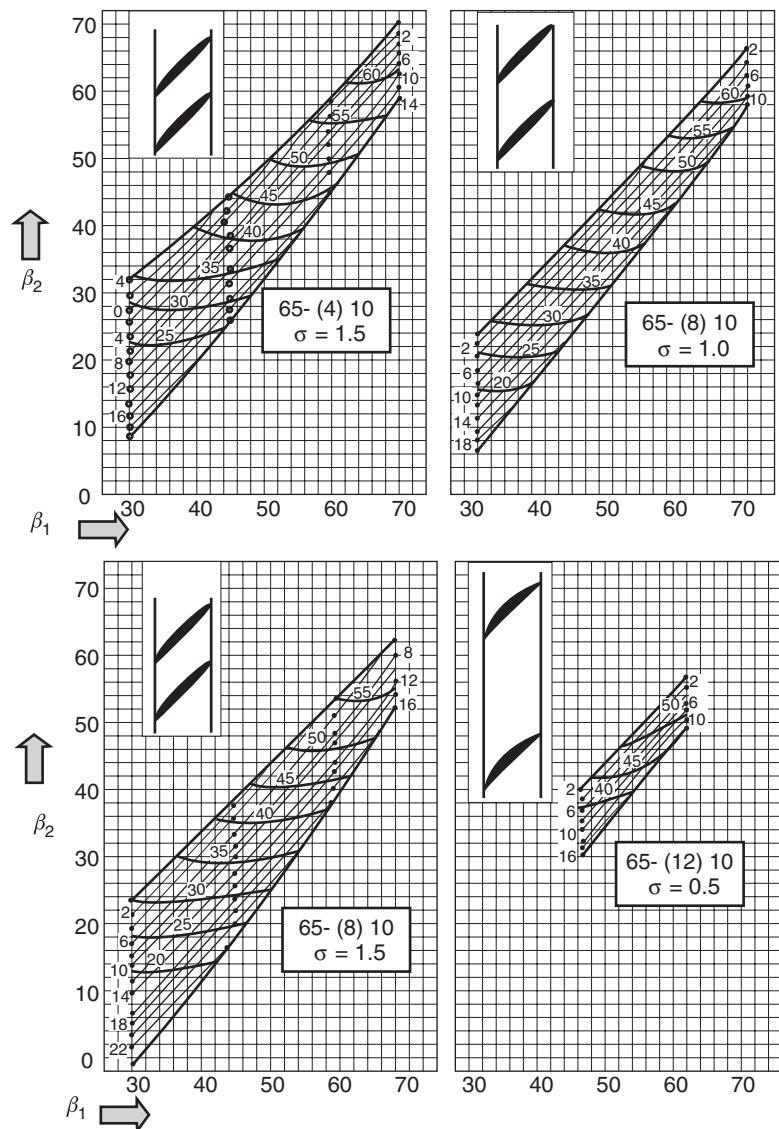
Definition sketch for the cascade stall boundaries, according to Mellor

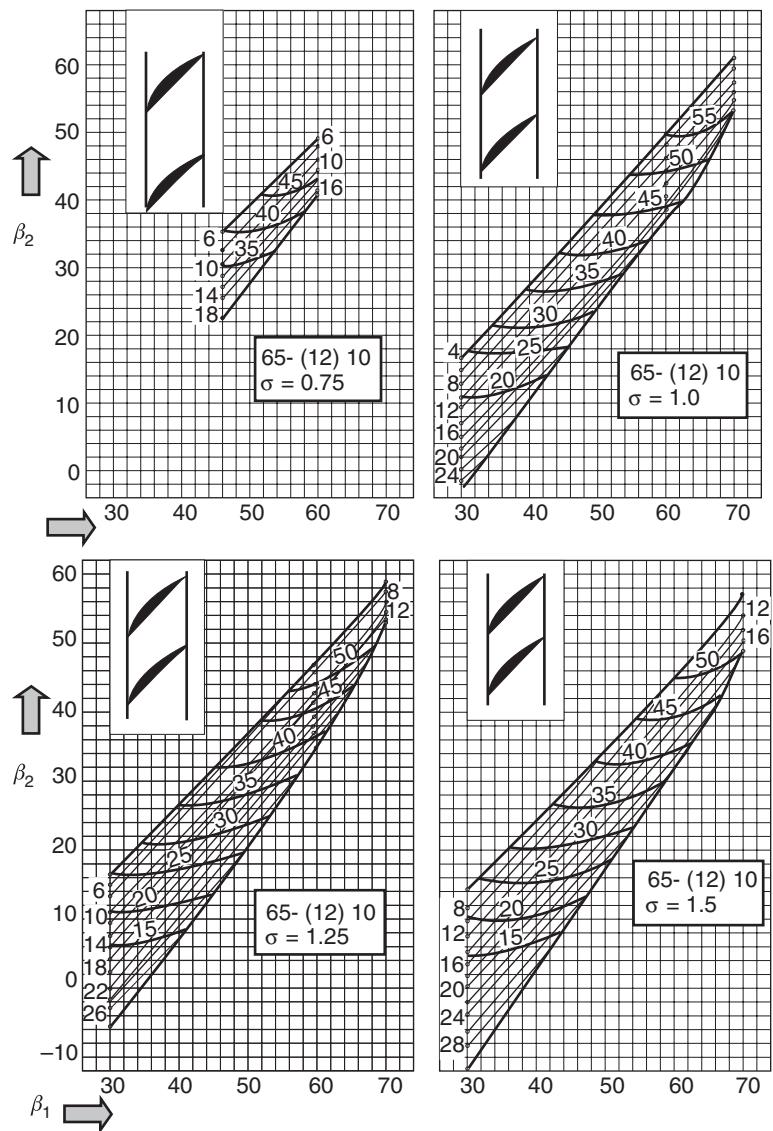
Mellor presents the operational range of each 65-series airfoils in various cascade arrangements that are very useful for preliminary design purposes. Mellor's unpublished graphical correlations (originated at MIT's Gas Turbine Laboratory) were published by Horlock (1973), Hill and Peterson (1992), among others. In this appendix, Mellor's cascade data presentation is graphed according to the following definition sketch.

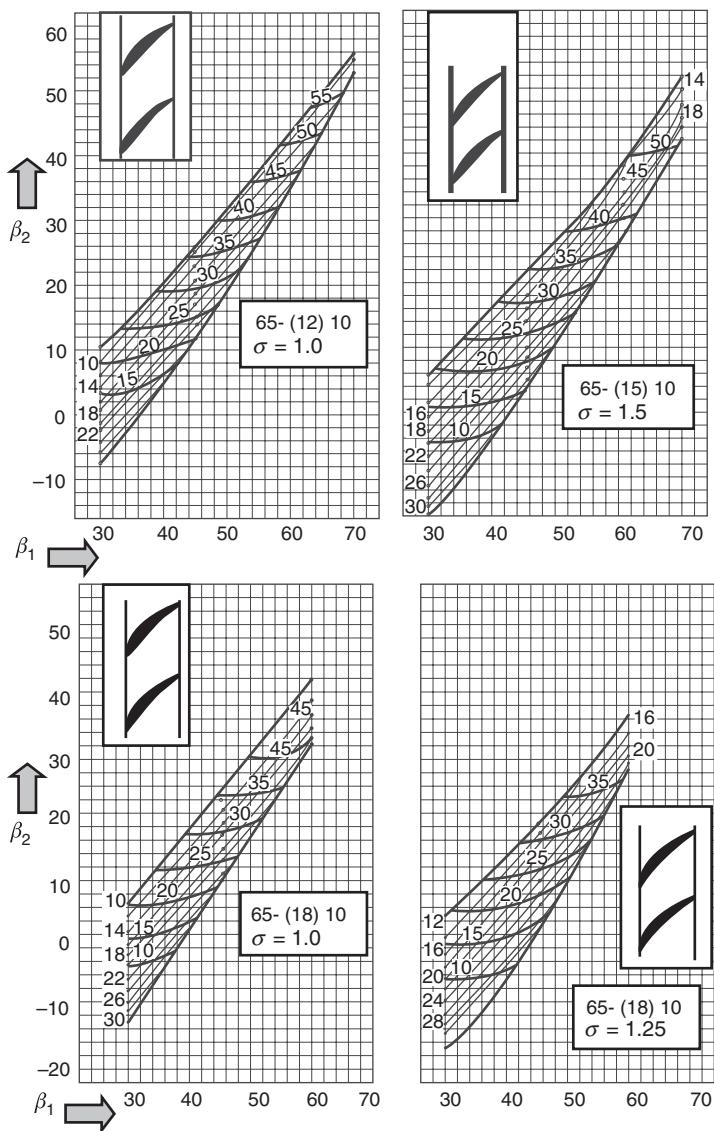


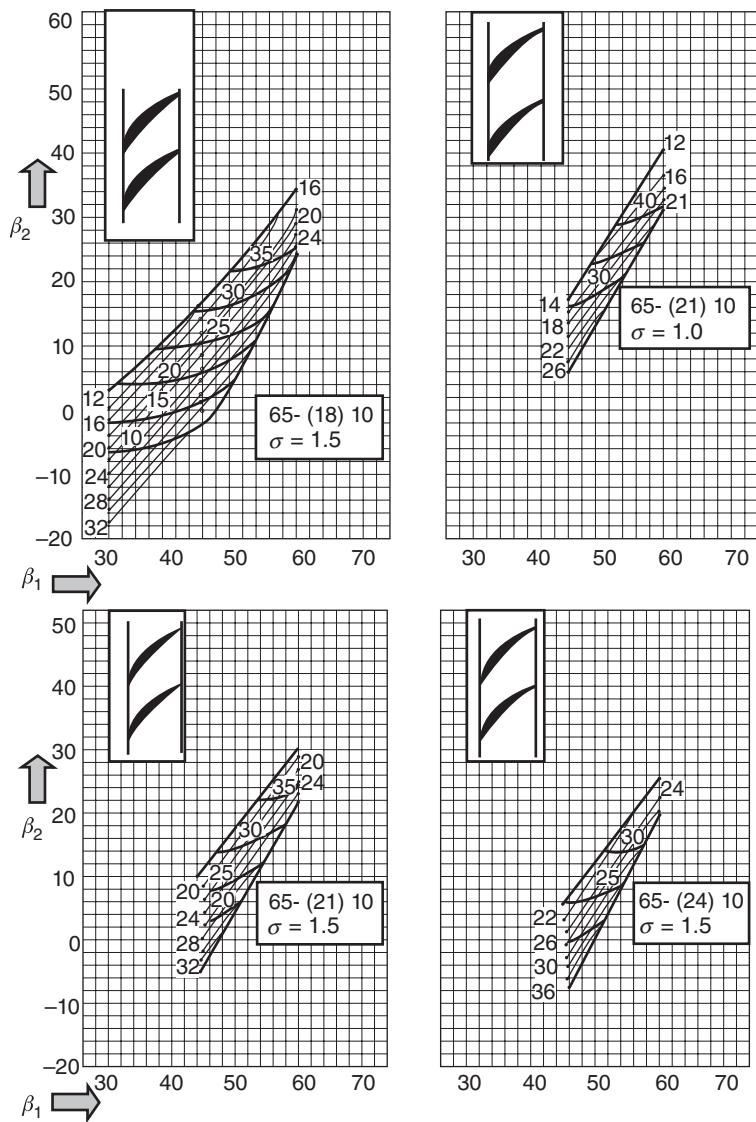
Cascade Data











APPENDIX J

Websites

The websites of NASA and major aircraft engine manufacturers are listed for reference (all websites were last accessed in December 2013).

www.hq.nasa.gov
www.grc.nasa.gov
www.pw.utc.com
www.pwc.ca
www.ge.com
www.mtu.de
www.rolls-royce.com
www.sneecma.com
www.enginealliance.com
www.cfm56.com
www.geae.com
www.V2500.com
www.williams-int.com