

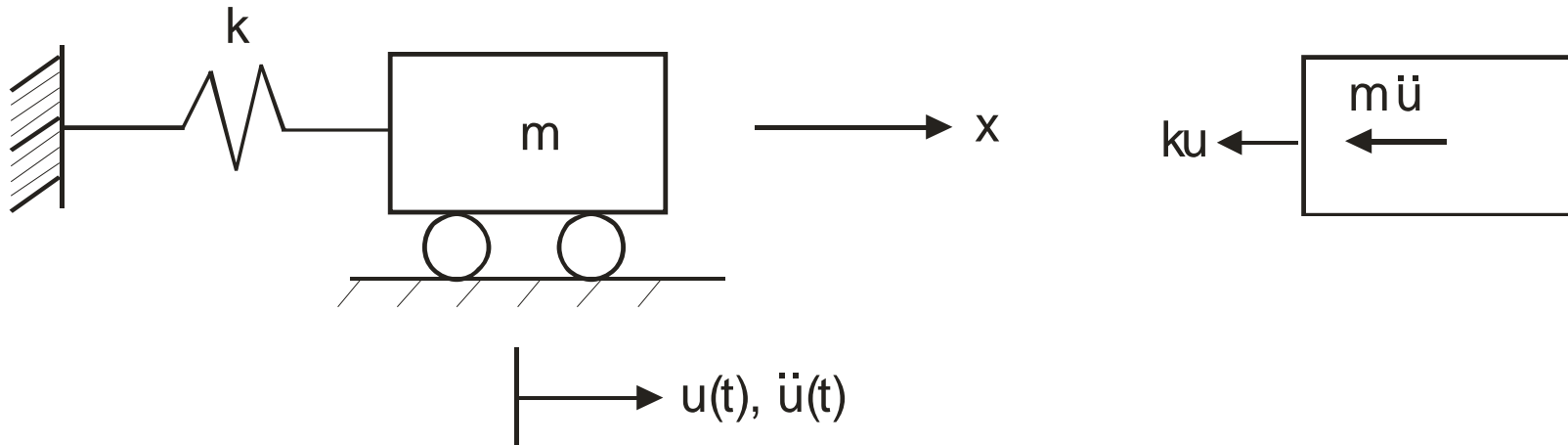
**CEE432/CEE532/MAE541**

Developing Software for  
Engineering Applications

**Lecture 12:  
Eigenproblems**

# Background

# Introduction: Free Vibration



**D'Alembert's Principle**  $m\ddot{u} + ku = 0$

**Let**  $\omega^2 = \frac{k}{m} \Rightarrow \ddot{u} + \omega^2 u = 0$

# Introduction: Free Vibration

**Angular Frequency**       $\omega = \sqrt{k/m} \quad rad/s$

**Natural Frequency**       $f = \omega/2\pi \quad Hz$

**Natural Period**       $T = 1/f \quad s$

# Introduction: Free Vibration

## Solution

$$u(t) = C_1 \cos \omega t + C_2 \sin \omega t$$

$$t = t_0 \quad u = u_0 \quad \text{Initial displacement}$$

$$\dot{u} = \dot{u}_0 \quad \text{Initial velocity}$$

## Solution to free, undamped vibration

$$u = A \cos(\omega t - \alpha)$$

$$\text{Amplitude} \quad A = \sqrt{u_0^2 + \left(\frac{\dot{u}_0}{\omega}\right)^2} \quad \text{Phase Angle} \quad \alpha = \tan^{-1} \frac{\dot{u}_0}{\omega u_0}$$

# Introduction: Forced Vibration

## Harmonic forcing function

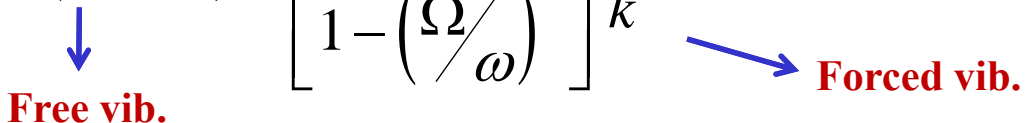
$$m\ddot{u} + ku = P \sin \Omega t$$

$$p_m = \frac{P}{m} \Rightarrow \ddot{u} + \omega^2 u = p_m \sin \Omega t$$

## Solution (particular solution + general solution)

$$u = C_1 \cos \omega t + C_2 \sin \omega t + C_3 \sin \Omega t$$

$$u = X \sin(\omega t + \phi) + \left[ \frac{1}{1 - \left(\frac{\Omega}{\omega}\right)^2} \right] \frac{P}{k} \sin \Omega t$$



**Free vib.** **Forced vib.**

# Introduction: Forced Vibration

$$u = \left[ \frac{1}{1 - \left(\frac{\Omega}{\omega}\right)^2} \right] \frac{P}{k} \sin \Omega t = \frac{1}{\beta} \frac{P}{k} \sin \Omega t$$

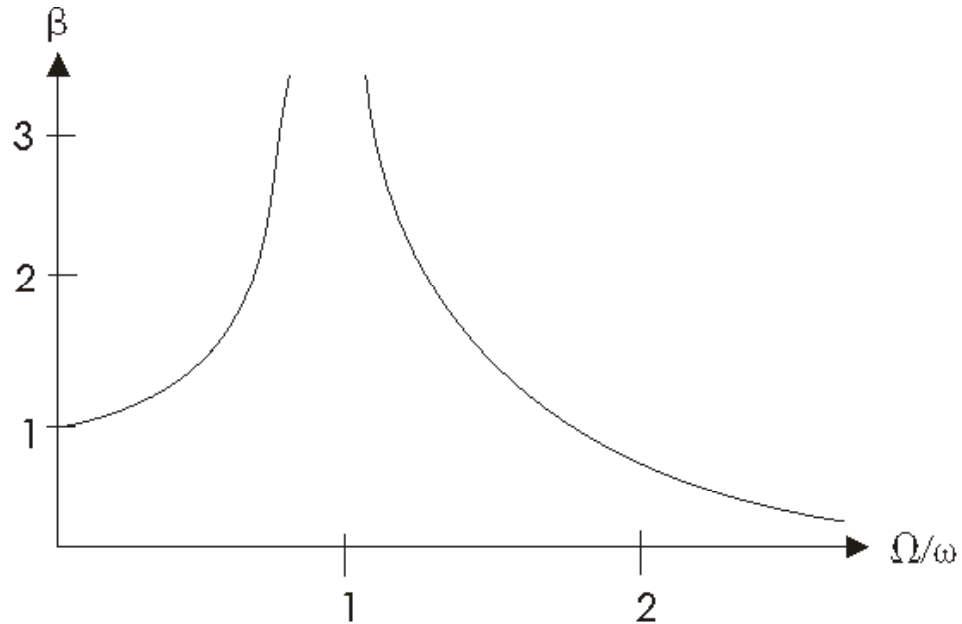
Frequency  
Ratio

Equivalent  
Static  
Deflection

$$r = \frac{\Omega}{\omega}; X_0 = \frac{P}{k}$$

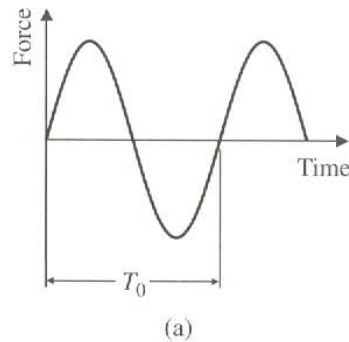
$$X_f = \frac{X_0}{1 - r^2} = (DMF) X_0$$

Dynamic  
Magnification  
Factor

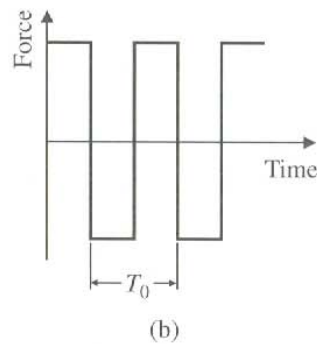


Resonance state

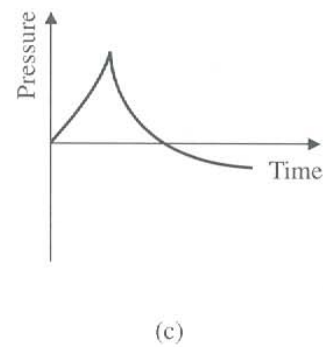
# Dynamic Loads



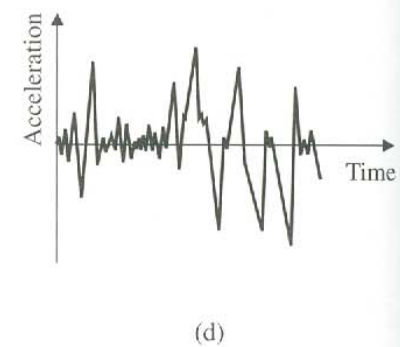
Simple  
Harmonic



Periodic,  
Non-  
Harmonic



Non-  
periodic,  
short  
duration



Non-  
periodic,  
long  
duration



# Examples

- Environmental
  - Wind (non-periodic, long duration)
  - Earthquake (Wave) (non-periodic, long duration)
- Machinery (periodic, harmonic and non-harmonic)
- Vehicular (varied)
- Blast (non-periodic, short duration)

# Numerical Techniques

# Eigenvalue Analysis

## Properties

$$\mathbf{K}_{n \times n} \mathbf{\Phi}_{n \times n} = \mathbf{\Lambda}_{n \times n} \mathbf{M}_{n \times n} \mathbf{\Phi}_{n \times n}$$

- $\mathbf{K}$  is symmetric and positive definite
- $\mathbf{M}$  is symmetric  $0 \leq \lambda_1 \leq \lambda_2 \leq \dots \leq \lambda_n$
- $n$  real eigenvalues

# Eigenvalue Analysis

## Properties

$$\mathbf{K}\boldsymbol{\varphi}_i = \lambda_i \mathbf{M}\boldsymbol{\varphi}_i$$

$$\boldsymbol{\varphi}_i^T \mathbf{K} \boldsymbol{\varphi}_j = 0$$

$$\boldsymbol{\varphi}_i^T \mathbf{M} \boldsymbol{\varphi}_j = 0$$

$$\boldsymbol{\varphi}_i^T \mathbf{K} \boldsymbol{\varphi}_j = \lambda_i$$

$$\boldsymbol{\varphi}_i^T \mathbf{M} \boldsymbol{\varphi}_i = 1$$

# Solution Techniques

## Characteristic Polynomial Technique

$$[\mathbf{K} - \lambda \mathbf{M}] \boldsymbol{\phi} = 0$$

$$\det[\mathbf{K} - \lambda \mathbf{M}] = 0$$

The above equation is a polynomial of order  $n$ . The roots of the polynomial are the eigenvalues.

# Example

$$\mathbf{K}_{3 \times 3} \boldsymbol{\Phi}_{3 \times 3} = \lambda \mathbf{M}_{3 \times 3} \boldsymbol{\Phi}_{3 \times 3}$$

$$\begin{bmatrix} 3 & 2 & 1 \\ 2 & 2 & 1 \\ 1 & 1 & 1 \end{bmatrix} \begin{Bmatrix} \varphi_1 \\ \varphi_2 \\ \varphi_3 \end{Bmatrix} - \lambda \begin{bmatrix} 1 & 0 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 1 \end{bmatrix} \begin{Bmatrix} \varphi_1 \\ \varphi_2 \\ \varphi_3 \end{Bmatrix} = \begin{Bmatrix} 0 \\ 0 \\ 0 \end{Bmatrix}$$

$$\mathbf{K}_{3 \times 3} - \lambda \mathbf{M}_{3 \times 3}$$

$$\begin{bmatrix} 3 - \lambda & 2 & 1 \\ 2 & 2 - \lambda & 1 \\ 1 & 1 & 1 - \lambda \end{bmatrix} \begin{Bmatrix} \varphi_1 \\ \varphi_2 \\ \varphi_3 \end{Bmatrix} = \begin{Bmatrix} 0 \\ 0 \\ 0 \end{Bmatrix}$$

# Example

$$\det(\mathbf{K}_{3 \times 3} - \lambda \mathbf{M}_{3 \times 3}) = 0$$

$$\lambda_1 = 0.308$$

$$\lambda^3 - 6\lambda^2 + 5\lambda - 1 = 0 \Rightarrow \lambda_2 = 0.643$$

$$\lambda_3 = 5.049$$

$$\lambda_1 = 0.308$$

$$\begin{bmatrix} 3 - 0.308 & 2 & 1 \\ 2 & 2 - 0.308 & 1 \\ 1 & 1 & 1 - 0.308 \end{bmatrix} \begin{Bmatrix} \varphi_1 \\ \varphi_2 \\ \varphi_3 \end{Bmatrix} = \begin{Bmatrix} 0 \\ 0 \\ 0 \end{Bmatrix}$$

# Example

Let  $\varphi_3 = 1$

$$\begin{bmatrix} 2.692 & 2 \\ 2 & 1.692 \end{bmatrix} \begin{Bmatrix} \varphi_1 \\ \varphi_2 \end{Bmatrix} = \begin{Bmatrix} -1 \\ -1 \end{Bmatrix} \Rightarrow \begin{Bmatrix} \varphi_1 \\ \varphi_2 \end{Bmatrix} = \begin{Bmatrix} 0.555 \\ -1.247 \end{Bmatrix}$$

Hence

$$\begin{Bmatrix} \varphi_1 \\ \varphi_2 \\ \varphi_3 \end{Bmatrix} = \begin{Bmatrix} 0.555 \\ -1.247 \\ 1 \end{Bmatrix} = \begin{Bmatrix} -0.445 \\ 1 \\ -0.802 \end{Bmatrix}$$



# Example

$$\mathbf{K}_{3 \times 3} \boldsymbol{\Phi}_{3 \times 3} = \boldsymbol{\Lambda}_{3 \times 3} \mathbf{M}_{3 \times 3} \boldsymbol{\Phi}_{3 \times 3}$$

$$\begin{bmatrix} 3 & 2 & 1 \\ 2 & 2 & 1 \\ 1 & 1 & 1 \end{bmatrix} \begin{bmatrix} 0.591 & 0.737 & 0.328 \\ -1.328 & -0.409 & 0.263 \\ 1.065 & -0.919 & 0.146 \end{bmatrix} =$$

$$\begin{bmatrix} 0.308 & 0 & 0 \\ 0 & 0.643 & 0 \\ 0 & 0 & 5.049 \end{bmatrix} \begin{bmatrix} 1 & 0 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 1 \end{bmatrix} \begin{bmatrix} 0.591 & 0.737 & 0.328 \\ -1.328 & -0.409 & 0.263 \\ 1.065 & -0.919 & 0.146 \end{bmatrix}$$

# Rayleigh-Ritz Analysis

**Consider**

$$\mathbf{K}\phi = \lambda\mathbf{M}\phi \quad \mathbf{K} \text{ and } \mathbf{M} \text{ are positive definite}$$

**Rayleigh Minimum Principle**

$$\lambda_1 = \min(\rho(\phi)) = \min\left(\frac{\phi^T \mathbf{K} \phi}{\phi^T \mathbf{M} \phi}\right) \quad 0 < \lambda_1 \leq \rho(\phi) \leq \lambda_n < \infty$$

# Solution Techniques

## Inverse Iteration Method

- Step 1: Assume  $\mathbf{u}^0$ . Set  $k=0$ .
- Step 2: Set  $k=k+1$ .
- Step 3: Compute  $\mathbf{v}^{k-1} = \mathbf{M}\mathbf{u}^{k-1}$
- Step 4: Solve  $\mathbf{K}\hat{\mathbf{u}}^k = \mathbf{v}^{k-1}$
- Step 5: Let  $\hat{\mathbf{v}}^k = \mathbf{M}\hat{\mathbf{u}}^k$
- Step 6: Estimate  $\lambda^k = \frac{\hat{\mathbf{u}}^{kT} \hat{\mathbf{v}}^{k-1}}{\hat{\mathbf{u}}^{kT} \hat{\mathbf{v}}^k}$

# Inverse Iteration

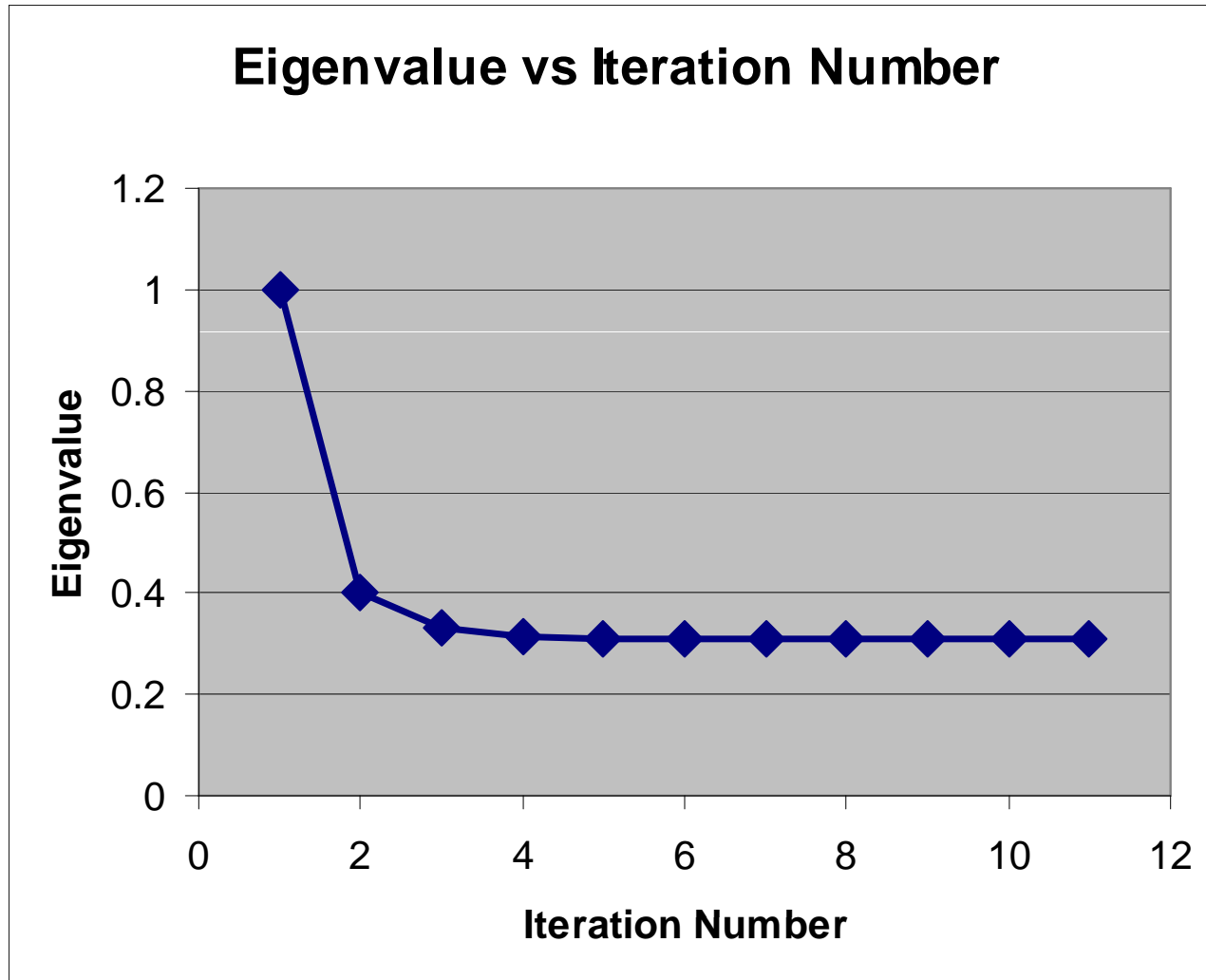
- Step 7: Normalize eigenvector  $\mathbf{u}^k = \frac{\hat{\mathbf{u}}^k}{\left( \hat{\mathbf{u}}^{k^T} \hat{\mathbf{v}}^k \right)^{1/2}}$
- Step 8: Convergence check  $\left| \frac{\lambda^k - \lambda^{k-1}}{\lambda^k} \right| \leq \textit{tolerance}$
- Step 9: If not converged, go to Step 2.

# Example

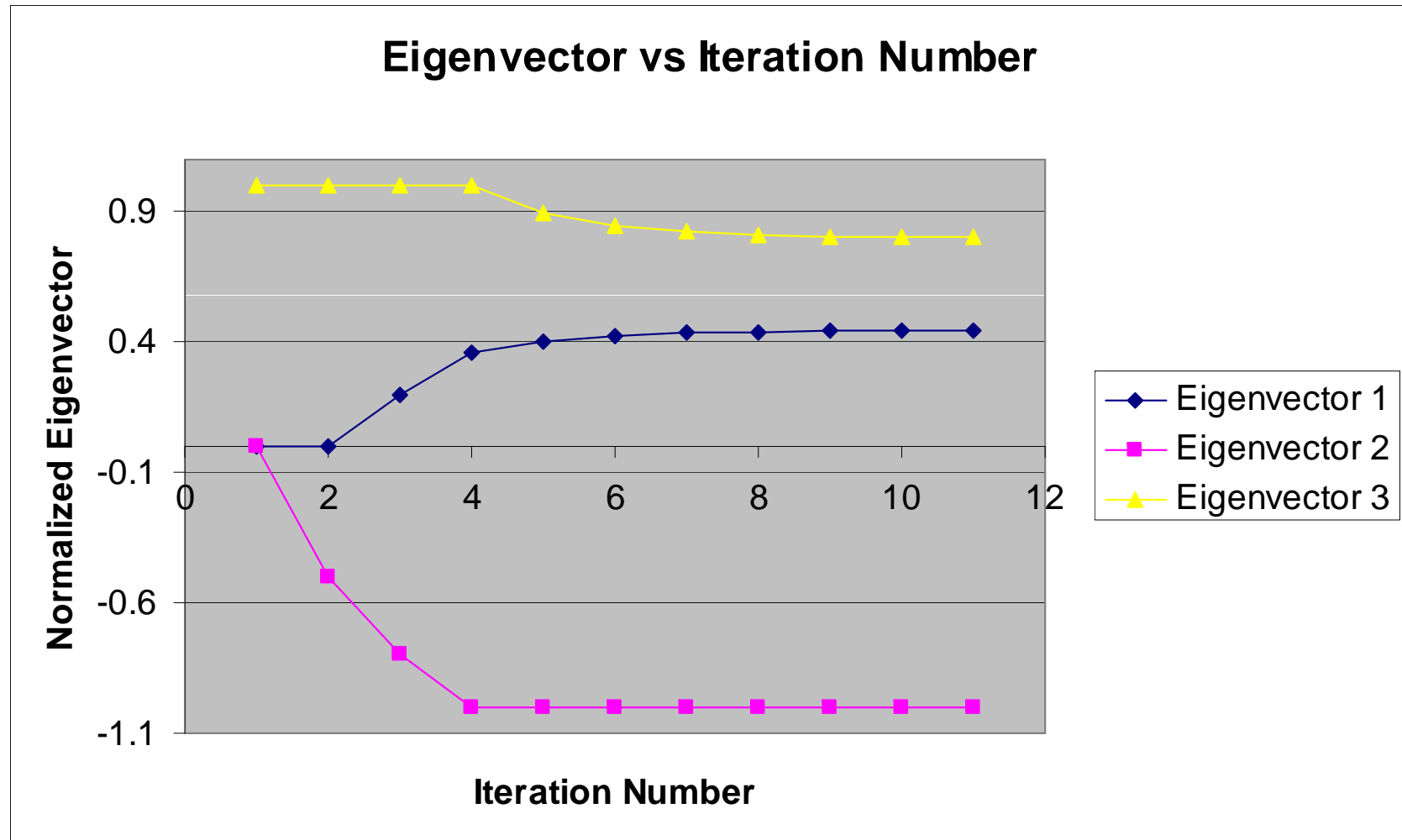
$$\mathbf{K}_{3 \times 3} \mathbf{\Phi}_{3 \times 3} = \mathbf{\Lambda}_{3 \times 3} \mathbf{M}_{3 \times 3} \mathbf{\Phi}_{3 \times 3}$$

$$\begin{bmatrix} 3 & 2 & 1 \\ 2 & 2 & 1 \\ 1 & 1 & 1 \end{bmatrix} \begin{Bmatrix} \varphi_1 \\ \varphi_2 \\ \varphi_3 \end{Bmatrix} = \mathbf{\Lambda} \begin{bmatrix} 1 & 0 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 1 \end{bmatrix} \begin{Bmatrix} \varphi_1 \\ \varphi_2 \\ \varphi_3 \end{Bmatrix}$$

# Example



# Example



# Other Techniques

- Jacobi Method
- Generalized Jacobi Method
- Subspace Iteration Method
- Lanczos Method



# Example Program 16.8.1

# Shifting

**Original problem**      $\mathbf{K}\Phi = \lambda\mathbf{M}\Phi$

**Shift  $\mathbf{K}$**       $\hat{\mathbf{K}} = \mathbf{K} - \rho\mathbf{M}$

**New problem**      $\hat{\mathbf{K}}\Psi = \mu\mathbf{M}\Psi$

It can be shown that

$$\lambda_i = \rho + \mu_i$$

$$\phi_i = \psi_i$$

# Example

## Original problem

$$\begin{bmatrix} 3 & -3 \\ -3 & 3 \end{bmatrix} \Phi = \lambda \begin{bmatrix} 2 & 1 \\ 1 & 2 \end{bmatrix} \Phi$$

$$\det(\mathbf{K} - \lambda \mathbf{M}) = 3\lambda^2 - 18\lambda = 0 \Rightarrow \begin{aligned} \lambda_1 &= 0 \\ \lambda_2 &= 6 \end{aligned}$$

## New problem $\rho = -2$

$$\begin{bmatrix} 7 & -1 \\ -1 & 7 \end{bmatrix} \Phi = \lambda \begin{bmatrix} 2 & 1 \\ 1 & 2 \end{bmatrix} \Phi$$

$$\det(\mathbf{K} - \lambda \mathbf{M}) = \lambda^2 - 10\lambda + 16 = 0 \Rightarrow \begin{aligned} \lambda_1 &= 2 \\ \lambda_2 &= 8 \end{aligned}$$

# Structural Analysis

# Lagrangian, L

$$L = T - \Pi$$



kinetic energy

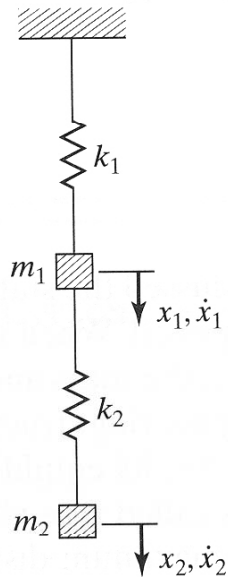
Hamilton's Principle  
(extremize the  
functional)

$$I = \int_{t_1}^{t_2} L(D_1, D_2, \dots, D_n) dt$$

Equations of motion

$$\frac{d}{dt} \left( \frac{\partial L}{\partial \dot{D}_i} \right) - \frac{\partial L}{\partial D_i} = 0 \quad i = 1, \dots, n$$

# Equations of Motion



$$\frac{d}{dt} \left( \frac{\partial L}{\partial \dot{x}_1} \right) - \frac{\partial L}{\partial x_1} = m_1 \ddot{x}_1 + k_1 x_1 - k_2 (x_2 - x_1) = 0$$

$$\frac{d}{dt} \left( \frac{\partial L}{\partial \dot{x}_2} \right) - \frac{\partial L}{\partial x_2} = m_2 \ddot{x}_2 - k_2 (x_2 - x_1) = 0$$



$$\begin{bmatrix} m_1 & 0 \\ 0 & m_2 \end{bmatrix} \begin{Bmatrix} \ddot{x}_1 \\ \ddot{x}_2 \end{Bmatrix} + \begin{bmatrix} (k_1 + k_2) & -k_2 \\ -k_2 & k_2 \end{bmatrix} \begin{Bmatrix} x_1 \\ x_2 \end{Bmatrix} = \begin{Bmatrix} 0 \\ 0 \end{Bmatrix}$$



$$\mathbf{M}\ddot{\mathbf{D}} + \mathbf{K}\mathbf{D} = \mathbf{0}$$

$$T = \frac{1}{2} m_1 \dot{x}_1^2 + \frac{1}{2} m_2 \dot{x}_2^2$$

$$\Pi = \frac{1}{2} k_1 x_1^2 + \frac{1}{2} k_2 (x_2 - x_1)^2$$

# Mass Matrix

$$T = \frac{1}{2} \int_V \dot{\mathbf{u}}^T \dot{\mathbf{u}} \rho dV$$

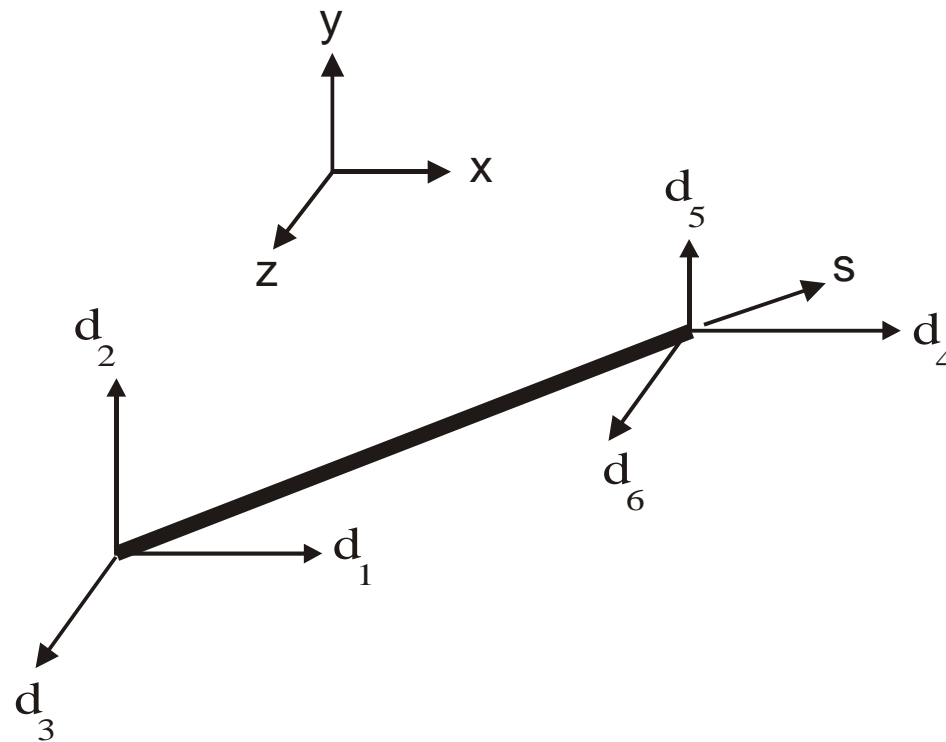
$$\dot{\mathbf{u}} = [\dot{u} \quad \dot{v} \quad \dot{w}]^T$$

$$\mathbf{u}_{3 \times 1} = \mathbf{N}_{3 \times n} \mathbf{D}_{n \times 1}$$

$$\dot{\mathbf{u}}_{3 \times 1} = \mathbf{N}_{3 \times n} \dot{\mathbf{D}}_{n \times 1}$$

$$T = \frac{1}{2} \int_V \dot{\mathbf{u}}^T \dot{\mathbf{u}} \rho dV = \frac{1}{2} \dot{\mathbf{D}}^T \left[ \int_e \rho \mathbf{N}^T \mathbf{N} dV \right] \dot{\mathbf{D}} = \frac{1}{2} \dot{\mathbf{D}}_{1 \times n}^T \mathbf{m}_{n \times n} \dot{\mathbf{D}}_{n \times 1}$$

# Space Truss Element





# Space Truss Element

## Consistent mass matrix

$$\begin{Bmatrix} u \\ v \\ w \end{Bmatrix} = \begin{bmatrix} 1-a & 0 & 0 & a & 0 & 0 \\ 0 & 1-a & 0 & 0 & a & 0 \\ 0 & 0 & 1-a & 0 & 0 & a \end{bmatrix} \begin{Bmatrix} d_1 \\ d_2 \\ d_3 \\ d_4 \\ d_5 \\ d_6 \end{Bmatrix} = \mathbf{N}_{3 \times 6} \mathbf{d}_{6 \times 1}$$

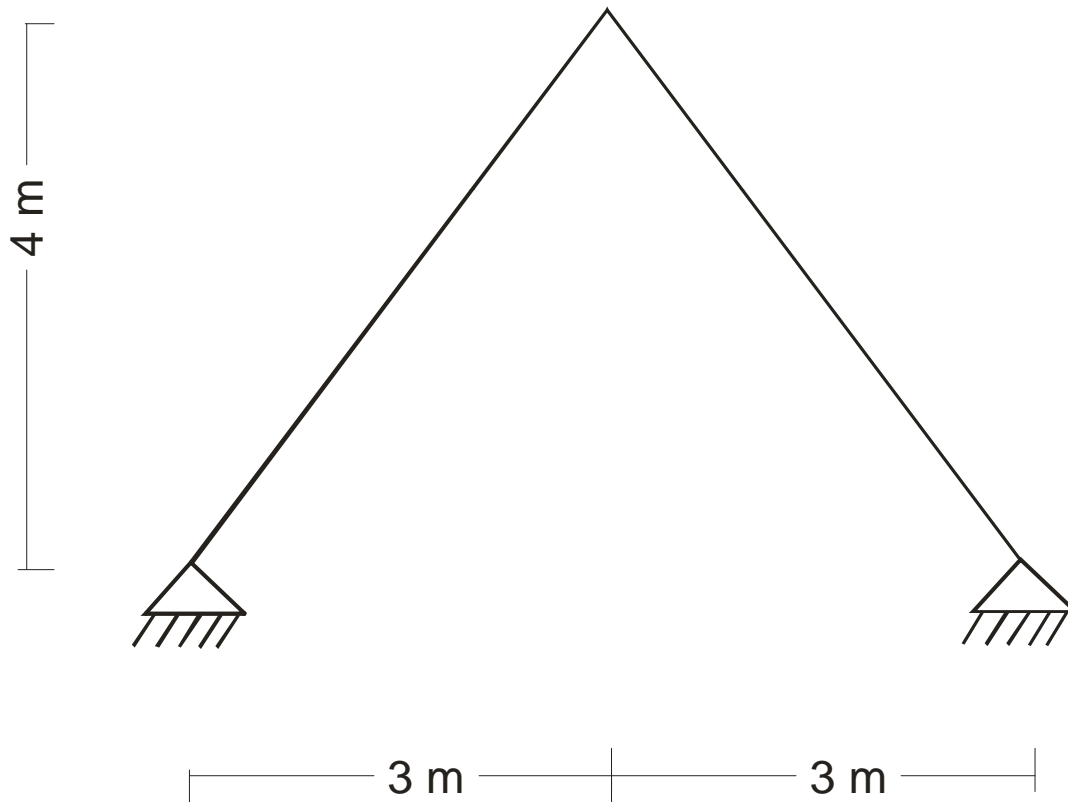
$$\mathbf{m}_{6 \times 6} = \int_{\Omega} \rho \mathbf{N}^T \mathbf{N} dV$$

# Space Truss Element

Consistent mass matrix (global coord. System)

$$\mathbf{m}_{6 \times 6} = \frac{\rho AL}{6} \begin{bmatrix} 2 & 0 & 0 & 1 & 0 & 0 \\ 0 & 2 & 0 & 0 & 1 & 0 \\ 0 & 0 & 2 & 0 & 0 & 1 \\ 1 & 0 & 0 & 2 & 0 & 0 \\ 0 & 1 & 0 & 0 & 2 & 0 \\ 0 & 0 & 1 & 0 & 0 & 2 \end{bmatrix}$$

# Example: Ex1



$$E = 200 \text{ GPa}$$

$$\rho = 7850 \text{ kg/m}^3$$

$$A = 0.01 \text{ m}^2$$

$$f_1 = 167 \text{ Hz}$$

$$f_2 = 223 \text{ Hz}$$

# Beam Element

## Planar beam element (2 dof/element)

$$\mathbf{m}'_{4 \times 4} = \frac{\rho AL}{420} \begin{bmatrix} 156 & & & \\ 22L & 4L^2 & & \\ 54 & 13L & 156 & \\ -13L & -3L^2 & -22L & 4L^2 \end{bmatrix} \begin{matrix} \\ \\ \\ \text{SYM} \end{matrix}$$

# Beam Element

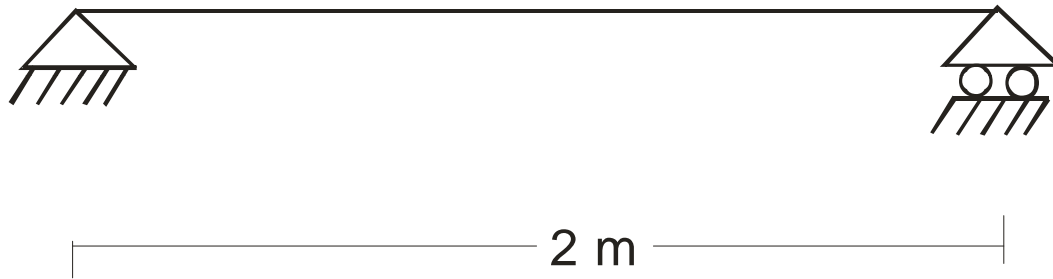
## Planar beam element with axial deformation

$$\mathbf{m}'_{6 \times 6} = \frac{\rho AL}{420} \begin{bmatrix} 140 & & & & & \\ & 156 & & & & \\ & 22L & 4L^2 & & & \\ 70 & 0 & 0 & 140 & & \\ & 54 & 13L & 0 & 156 & \\ 0 & -13L & -3L^2 & 0 & -22L & 4L^2 \end{bmatrix} \quad \text{SYM}$$

$$\mathbf{m}_{6 \times 6} = \mathbf{T}_{6 \times 6}^T \mathbf{m}'_{6 \times 6} \mathbf{T}_{6 \times 6}$$

Transformation from local-  
to-global

# Example



$$E = 10^{10} \text{ Pa}$$

$$\rho = 5000 \text{ kg/m}^3$$

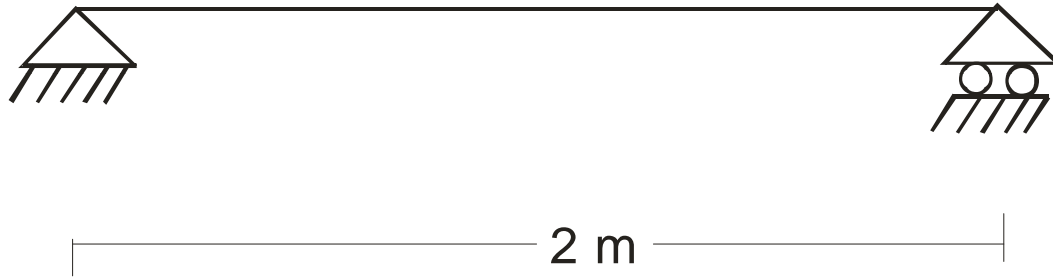
$$A = 0.001 \text{ m}^2$$

$$I = 0.0001 \text{ m}^4$$

## Analytical solution (bending modes)

$$\omega_n = \frac{(n\pi)^2}{L^2} \sqrt{\frac{EI}{\rho A}} \quad n = 1, 2, 3, \dots$$

## Example: Ex21 (1 Element)



$$E = 10^{10} \text{ Pa}$$

$$\rho = 5000 \text{ kg/m}^3$$

$$A = 0.001 \text{ m}^2$$

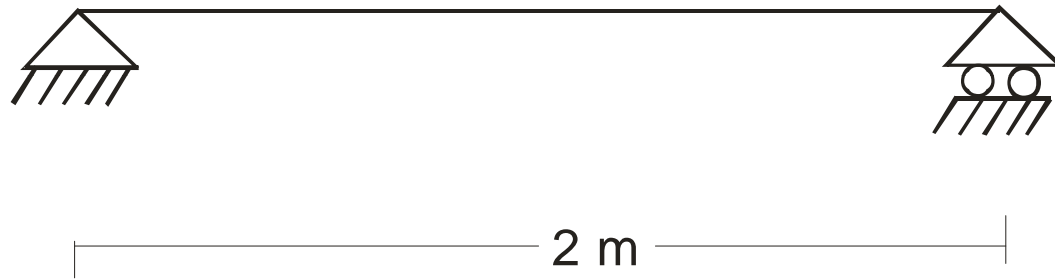
$$I = 0.0001 \text{ m}^4$$

$$f_1 = 195 \text{ Hz}$$

$$f_2 = 195 \text{ Hz}$$

$$f_3 = 893 \text{ Hz}$$

## Example: Ex22 (2 Element)



$$E = 10^{10} \text{ Pa}$$

$$\rho = 5000 \text{ kg/m}^3$$

$$A = 0.001 \text{ m}^2$$

$$I = 0.0001 \text{ m}^4$$

$$f_1 = 176 \text{ Hz}$$

$$f_4 = 780 \text{ Hz}$$

$$f_2 = 181 \text{ Hz}$$

$$f_5 = 1960 \text{ Hz}$$

$$f_3 = 633 \text{ Hz}$$



# Convergence Study

Mode\ Elements	Frequency (Hz)				
	1	2	4	8	<b>Exact</b>
1 (bending)	195	176	176	176	175.6
2	195	181	177	177	
3	893	633	561	538	
4 (bending)	-	780	705	703	702.5
5	-	1960	1020	920	

# Notes

- Convergence takes place with increasing number of elements
- Convergence is from above
- Convergence of higher modes takes place more slowly

# Lumped Mass Matrix

$$\mathbf{m}'_{6 \times 6} = \frac{\rho AL}{2} \begin{bmatrix} 1 & & & & & \\ & 1 & & & & \\ & 0 & 1 & & & \\ & 0 & 0 & 0 & & \\ & 0 & 0 & 0 & 1 & \\ & 0 & 0 & 0 & 0 & 1 \\ & 0 & 0 & 0 & 0 & 0 & 0 \end{bmatrix} \quad \text{SYM}$$



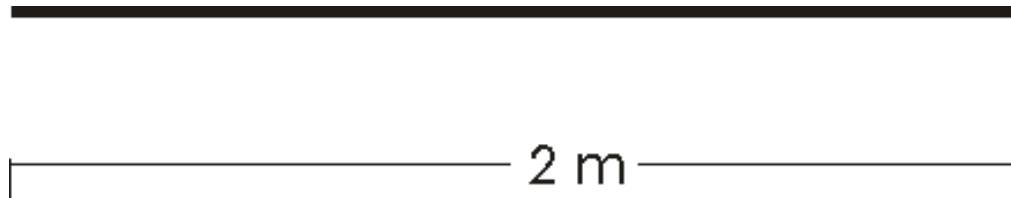
# Consistent versus Lumped Mass (8 Elements)

Mode	Frequency (Hz)	
	Consistent	Lumped
1	176	176
2	177	176
3	538	523
4	703	702
5	920	849

# Notes

- Both consistent and lumped mass matrices are valid formulations
- There are several forms of creating diagonal mass matrices

# Floating Beam



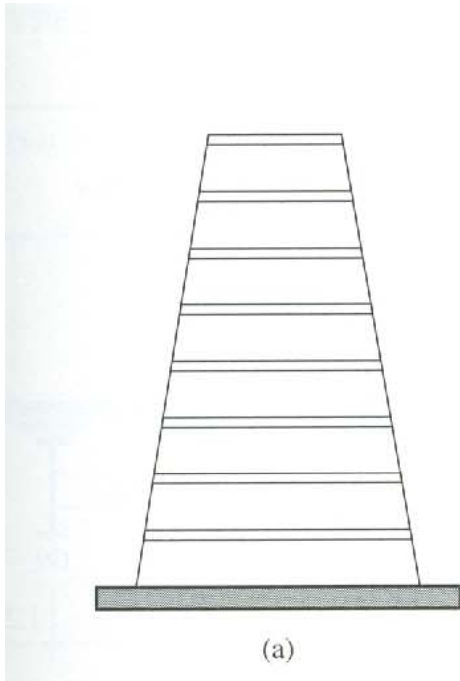
$$E = 10^{10} \text{ Pa}$$

$$\rho = 5000 \text{ kg/m}^3$$

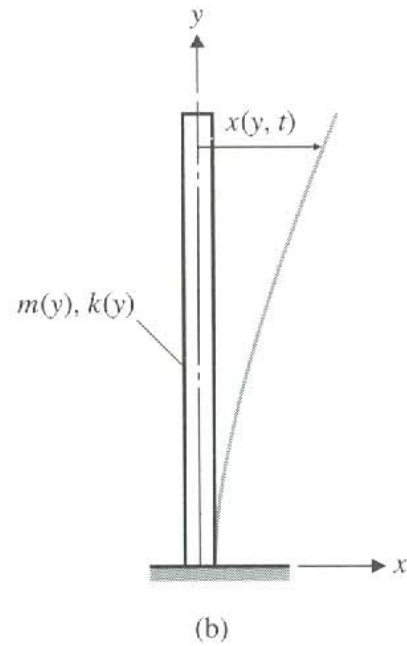
$$A = 0.001 \text{ m}^2$$

$$I = 0.0001 \text{ m}^4$$

# Frame Analysis



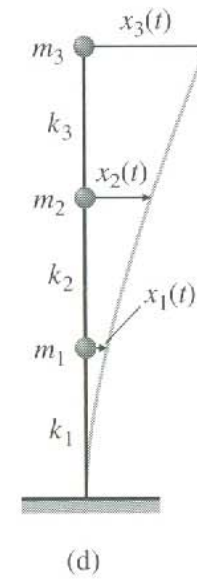
**Multi-storied Building**



**Continuous Model**

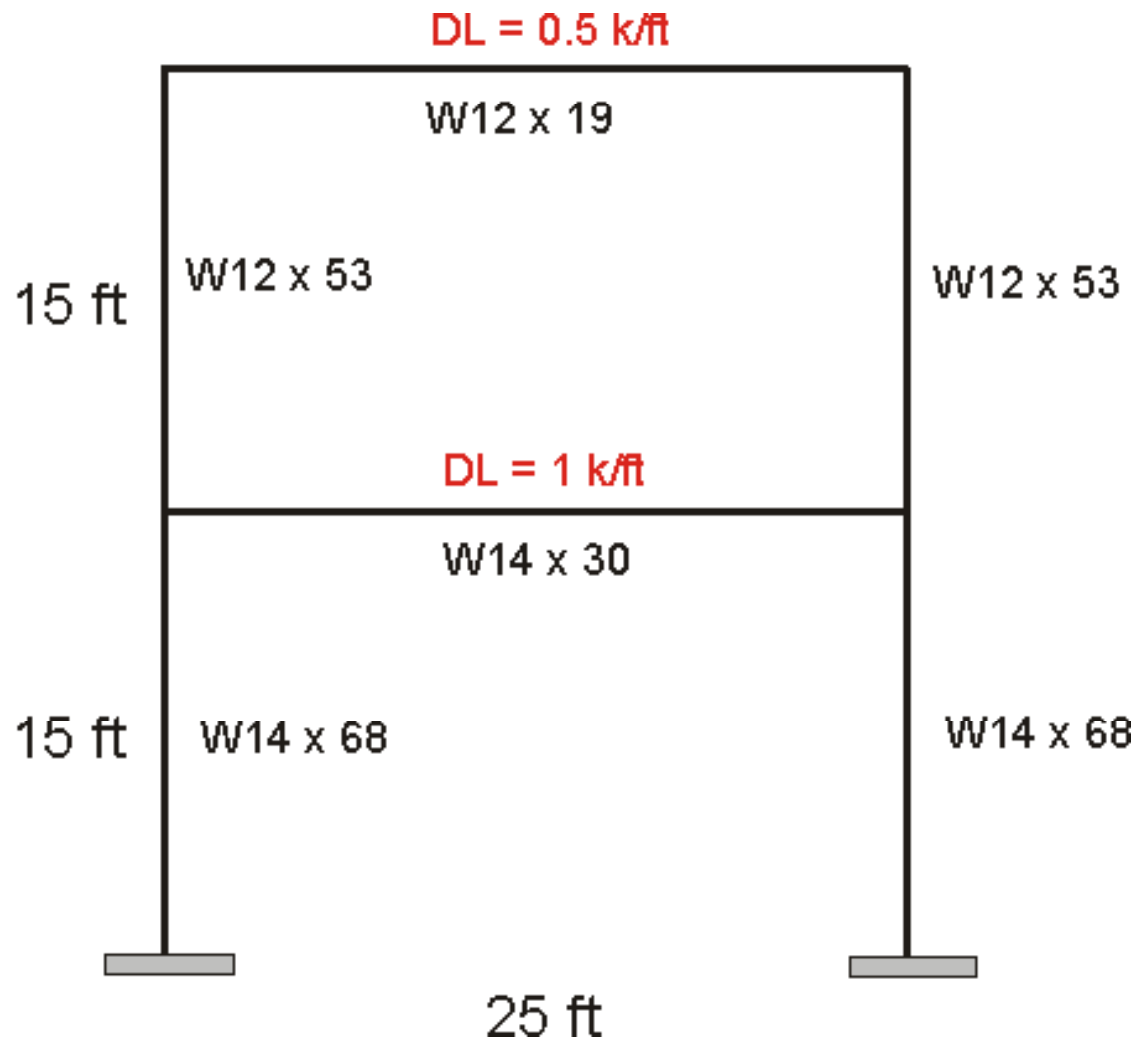


**SDOF Discrete Model**



**3-DOF Discrete Model**

# Example





# Results

Mode	Eigenvalue	Freq. (Hz)	Time Period (s)
1	2024	7.16	0.14
2	23808.7	24.6	0.04
3	50357.4	35.7	0.028
4	51838.9	36.2	0.028
10	870642	148.5	0.0067
11	943579	154.6	0.0065
12	1.46357(10 <sup>6</sup> )	192.5	0.0052