CEE432/CEE532/MAE541 Developing Software for Engineering Applications

Lecture 16: More on Truss Analysis and Variational Technique

Space Truss Analysis

Comparison with Planar Truss

- Differences
 - 3 nodal coordinates
 - 3 degrees-of-freedom per node
- Similarities
 - All other behavior characteristics are the same

Space Truss Element

$$\mathbf{k}_{2\times2}^{'}\mathbf{d}_{2\times1}^{'}=\mathbf{f}_{2\times1}^{'}$$

$$\mathbf{d}_{2\times 1}' = \mathbf{T}_{2\times 6}\mathbf{d}_{6\times 1} \quad \Longrightarrow$$

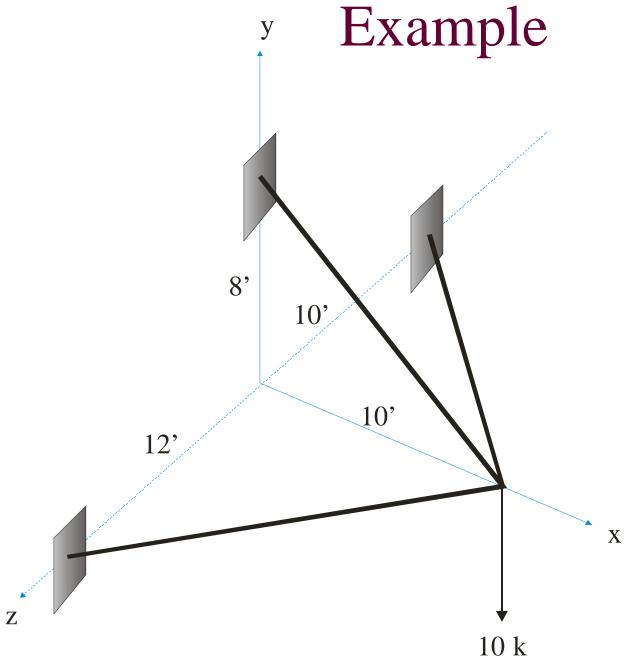
$$\mathbf{f}_{6\times 1} = \mathbf{T}_{6\times 2}^T \mathbf{f}_{2\times 1}'$$

Element Equations in Global Coordinate System

$$\mathbf{k}_{6\times 6}\mathbf{d}_{6\times 1}=\mathbf{f}_{6\times 1}$$

where

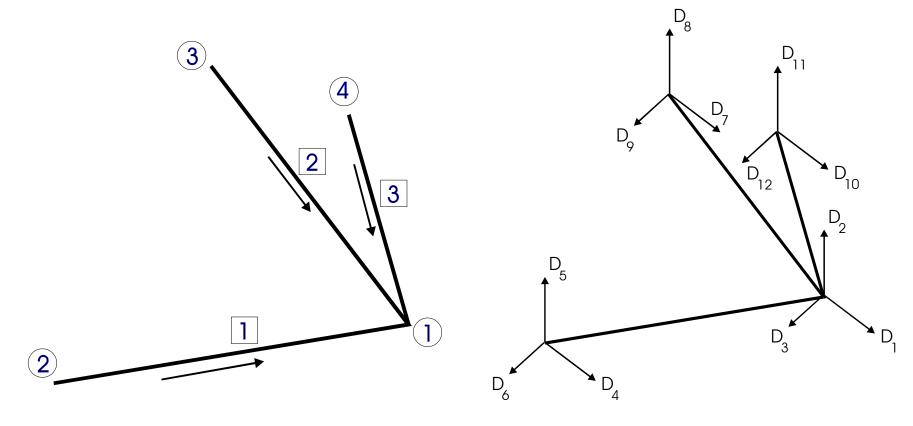
$$\mathbf{k}_{6\times6} = \mathbf{T}_{6\times2}^{\mathbf{T}} \mathbf{k}_{2\times2}' \mathbf{T}_{2\times6}$$



E = 29000 ksi

 $A = 2.5 \text{ in}^2$

FE Model



	1.5852	0	-1.9022	-1.5852	0	1.9022	D_4	$\left(f_1^1\right)$	
10 ⁵		0	0	0	0	0	D_5	$\left f_2^1\right $	
			2.2826	1.9022	0	-2.2826	$\int D_6 \left(\ \ \right)$	$\int f_3^1 \left(\right.$	
				1.5852	0	-1.9022	D_1	$-\int f_4^1$	•
					0	0	D_2	$\left f_5^1\right $	
	_ Sym					2.2826	$\lfloor D_3 \rfloor$	$\left[f_6^1\right]$	

10 ⁵	2.8767	-2.3013	0	-2.8767	2.3013	0	$\bigcap \left[D_{7} \right]$	$\left(f_1^2\right)$
		1.8411	0	2.3013	-1.8411	0	$ D_8 $	$ f_2^2 $
			0	0	0	0	$\left \int D_9 \right $	$\int f_3^2 \left[\right]$
				2.8767	-2.3013	0	$ D_1 ^{-s}$	$\int f_4^2 \int$
					1.8411	0	$ D_2 $	$ f_5^2 $
	_ Sym					0	$\left\lfloor \left\lfloor D_{3} \right floor ight ceil$	$\left[f_6^2\right]$

	2.1361	0	2.1361	-2.1361	0	-2.1361	$\left[D_{10} ight]$	$\left(f_1^3\right)$
10 ⁵		0	0	0	0	0	D_{11}	$\left f_2^3\right $
			2.1361	-2.1361	0	-2.1361	$\int D_{12} \Big[$	$\int f_3^3$
				2.1361	0	2.1361	D_1	$\int f_4^3$
					0	0	D_2	$\left f_5^3\right $
	_ Sym					2.1361	$\left[D_{3} \right]$	$\left[f_6^3\right]$

Steps 3 and 4: System Equations after BC

$$\begin{bmatrix} 6.5979 & -2.3013 & 2.3386 \\ -2.3013 & 1.8411 & 0 \\ 2.3386 & 0 & 4.4187 \end{bmatrix} \begin{bmatrix} D_1 \\ D_2 \\ D_3 \end{bmatrix} = \begin{bmatrix} 0 \\ -10^4 \\ 0 \end{bmatrix}$$

Step 5: Nodal Displacements

$$D_1 = -3.3703(10^{-2}) in$$

 $D_2 = -9.6445(10^{-2}) in$
 $D_3 = 1.7838(10^{-3}) in$

Element 1

$$f' = 386778[0.64]$$

$$-0.768$$

$$f' = 386778 \begin{bmatrix} 0.64 & 0 & -0.768 & -0.64 & 0 & 0.768 \end{bmatrix} \begin{cases} 0 \\ 0 \\ -0.0337 \\ -0.09644 \\ 0.00178 \end{bmatrix} = 8875 lb(C)$$
Element 2

$$f' = 471775[0.781 -0.625 0 -0.781 0.625$$

$$-0.625$$

$$\begin{bmatrix}
0 \\
0 \\
-0.0337 \\
-0.09644 \\
0.00178
\end{bmatrix} = -16008 lb (T)$$

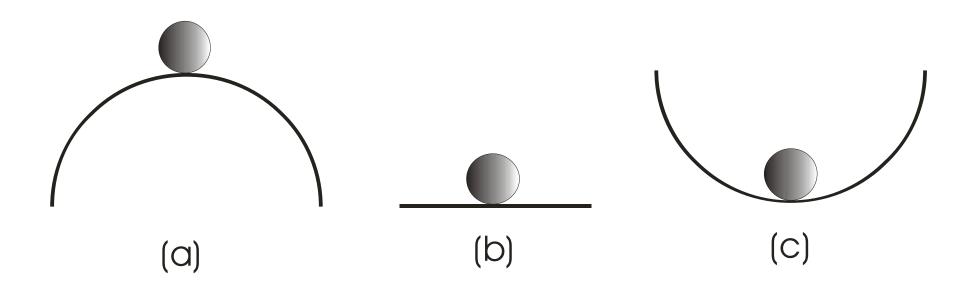
$$f' = 427210 \begin{bmatrix} 0.707 & 0 & -0.707 & -0.707 & 0 & 0.707 \end{bmatrix} \begin{cases} 0 \\ 0 \\ -0.0337 \\ -0.09644 \\ 0.00178 \end{cases} = 9642 lb(C)$$

Theorem of Minimum Potential Energy

• The Theorem of Minimum Potential Energy states that for a <u>conservative</u> system, amongst all <u>admissible configurations</u> those that satisfy the equations of <u>equilibrium</u> make the potential energy stationary with respect to small variations of displacement. If the stationary condition is a <u>minimum</u>, the equilibrium state is <u>stable</u>.

$$\Pi = \Pi(D_1, D_2, \dots, D_n) = \Pi(\mathbf{D})$$

$$\frac{\partial \Pi}{\partial \mathbf{D}} = 0 \quad i = 1, \quad 2, \dots \quad n$$



Total Potential Energy

 Π = strain energy + work potential

$$\Pi = \int_{V} U_{0} dV - \int_{V} \mathbf{f}^{T} \mathbf{F} dV - \int_{S} \mathbf{f}^{T} \Phi dS - \mathbf{D}^{T} \mathbf{P}$$

Strain Energy Density

$$U_0 = \frac{1}{2} \{ \boldsymbol{\varepsilon} \}^T \mathbf{E} \{ \boldsymbol{\varepsilon} \} - \{ \boldsymbol{\varepsilon} \}^T \mathbf{E} \{ \boldsymbol{\varepsilon}_0 \} + \{ \boldsymbol{\varepsilon} \}^T \{ \boldsymbol{\sigma}_0 \}$$

Stress-strain Relationship

$$\{\boldsymbol{\sigma}\} = \mathbf{E}\{\boldsymbol{\varepsilon}\} - \mathbf{E}\{\boldsymbol{\varepsilon}_0\} + \{\boldsymbol{\sigma}_0\}$$

Strain-displacement Relationship

$$\varepsilon_{x} = \frac{\partial u}{\partial x} \qquad \gamma_{xy} = \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x}$$

$$\varepsilon_{y} = \frac{\partial v}{\partial y}$$

Problem Statement

Consider a bar of constant cross-section, length and modulus of elasticity subjected to a constant axial force at the right tip and fixed at the left end. Compute the tip displacement and the state of stress in the bar.



Solution

Assumed solution
$$u(x) = a_0 + a_1 x$$

EBC
$$u(x=0)=0$$

Assumed

Displacement

$$u(x) = a_0 + a_1 x$$

$$u(x = 0) = 0$$

$$u(x=0) = 0 = a_0$$

$$u(x) = a_1 x$$

Strain-Disp.

$$\varepsilon_{x} = \frac{du}{dx} = a_{1}$$

Stress-Strain

$$\sigma_{x} = E\varepsilon_{x}$$

Total Potential Energy

$$\Pi(a_1) = \int_{V}^{L} U_0 \, dV - PD$$

$$\Pi(a_1) = \int_{0}^{L} \frac{1}{2} (a_1) (E) (a_1) A dx - P(a_1 L)$$

$$\Pi(a_1) = \frac{1}{2} a_1^2 E A L - P a_1 L$$

Minimization

$$\frac{d\Pi}{da_1} = 0 = a_1 EAL - PL \Rightarrow a_1 = \frac{P}{AE}$$

Final Solution

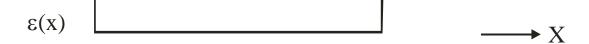
$$u(x) = \frac{Px}{AE}$$

$$\varepsilon_{x} = \frac{du}{dx} = \frac{P}{AE}$$

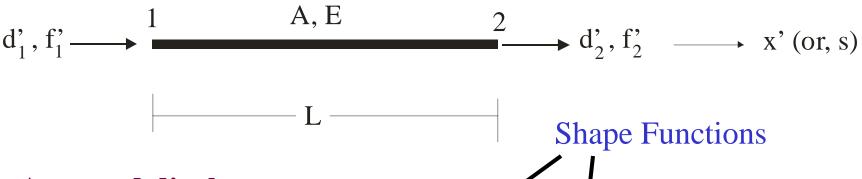
$$\sigma = E\varepsilon_{x} = \frac{P}{A}$$







$$\sigma(x)$$



Assumed displacement

$$u(s) = \phi_1(s)d_1' + \phi_2(s)d_2' = \frac{L-s}{L}d_1' + \frac{s}{L}d_2'$$

Strain-Disp.

$$\varepsilon = \frac{du}{ds} = \frac{d}{ds} \begin{bmatrix} \phi_1 & \phi_2 \end{bmatrix} \begin{Bmatrix} d_1 \\ d_2 \end{Bmatrix} = \begin{bmatrix} -\frac{1}{L} & \frac{1}{L} \end{bmatrix} \begin{Bmatrix} d_1 \\ d_2 \end{Bmatrix} = \mathbf{B}_{1 \times 2} \mathbf{d}_{2 \times 1}$$

Stress-Strain

$$\sigma_{x} = E\varepsilon_{x}$$

Total Potential Energy

$$U = \int_{V} U_{0} dV = \int_{0}^{L} \frac{1}{2} \varepsilon_{x} \sigma_{x} A ds = \int_{0}^{L} \frac{1}{2} \left[\mathbf{B}_{1 \times 2} \mathbf{d}_{2 \times 1}^{T} \right]^{T} E \left[\mathbf{B}_{1 \times 2} \mathbf{d}_{2 \times 1}^{T} \right] A ds$$

$$U = \left[\mathbf{d}^{T} \right]_{1 \times 2}^{T} \left[\int_{0}^{L} \mathbf{B}_{2 \times 1}^{T} (EA)_{1 \times 1} \mathbf{B}_{1 \times 2} ds \right] \left[\mathbf{d}^{T} \right]_{2 \times 1}^{T} = \left[\mathbf{d}^{T} \right]_{1 \times 2}^{T} \left[\mathbf{k}^{T} \right]_{2 \times 2} \left[\mathbf{d}^{T} \right]_{2 \times 1}^{T}$$

$$\left[\mathbf{k}^{T} \right]_{2 \times 2}^{T} = \int_{0}^{L} \mathbf{B}_{2 \times 1}^{T} (EA)_{1 \times 1} \mathbf{B}_{1 \times 2} ds = \frac{AE}{L} \left[\frac{1}{-1} \right]_{1 \times 2}^{T} \right]$$

Minimization

$$W = -\left[\mathbf{d}'\right]_{2\times 1}^{T} \left[\mathbf{f}'\right]_{2\times 1}$$

$$\Pi(\mathbf{d}') = U + W = \frac{1}{2} \left[\mathbf{d}'\right]_{1\times 2}^{T} \left[\mathbf{k}'\right]_{2\times 2} \left[\mathbf{d}'\right]_{2\times 1}^{T} - \left[\mathbf{d}'\right]_{2\times 1}^{T} \left[\mathbf{f}'\right]_{2\times 1}$$

$$\frac{\partial \Pi}{\partial \mathbf{d}'} = 0 \Rightarrow \frac{AE}{L} \left[\frac{1}{-1} \frac{|-1|}{1}\right] \left\{\frac{d_{1}'}{d_{2}'}\right\} = \left\{\frac{f_{1}'}{f_{2}'}\right\}$$

$$or, \left[\mathbf{k}'\right]_{2\times 2} \left[\mathbf{d}'\right]_{2\times 1} = \left[\mathbf{f}'\right]_{2\times 1}$$

Step 6: Secondary unknowns

$$\varepsilon = \frac{du}{ds} = \frac{d}{ds} \left(\phi_1 d_1' + \phi_2 d_2' \right) = \frac{d_2' - d_1'}{L}$$

$$\sigma = E\varepsilon$$

$$N = \sigma A$$

Thermal Loading

Theory

$$\varepsilon_{0} = \alpha \Delta T$$

$$\left(\mathbf{f}_{t}^{'}\right)_{2\times 1} = EA\varepsilon_{0} \begin{cases} -1\\1 \end{cases}$$

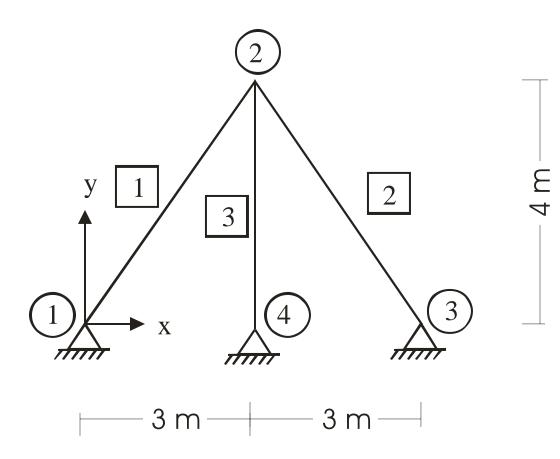
$$\mathbf{f}_{t} = \mathbf{T}^{T}\mathbf{f}_{t}^{'}$$

$$\sigma = E(\varepsilon - \varepsilon_{0})$$

$$or, \sigma = E\left[\frac{d_{2}^{'} - d_{1}^{'}}{L} - \alpha(\Delta T)\right]$$

Implementation

- 1. Compute thermal load vector for each element with change in temperature. Add to **F**.
- 2. For each element with temperature change, subtract the initial strain.



$$A = 0.01 \text{ m}^2$$

$$E = 200 \text{ GPa}$$

$$\alpha$$
= 1.2(10⁻⁵) m/m- 0 C

$$\Delta T_1 = 50^{0}C$$

Units: N, m

Solution without Element 3

$$4(10^8)\begin{bmatrix} 0.72 & 0 \\ 0 & 1.28 \end{bmatrix} \begin{Bmatrix} D_3 \\ D_4 \end{Bmatrix} = \begin{Bmatrix} F_3 \\ F_4 \end{Bmatrix}$$

Thermal Load Vector

$$\left(\mathbf{f}_{t}^{'}\right)_{2\times 1} = EA\varepsilon_{0} \begin{Bmatrix} -1\\1 \end{Bmatrix}$$

$$(\mathbf{f}_{t})_{2\times 1} = (2\times 10^{11})(0.01)(1.2\times 10^{-5})(50) \begin{Bmatrix} -1\\1 \end{Bmatrix}$$

$$\left(\mathbf{f}_{t}^{'}\right)_{2\times1} = \begin{Bmatrix} -1200000\\1200000 \end{Bmatrix} N$$

$$\Rightarrow \mathbf{f}_{t} = \mathbf{T}^{T} \mathbf{f}_{t}^{'} = \begin{cases} lf_{1}^{'} \\ mf_{1}^{'} \\ lf_{2}^{'} \\ mf_{2}^{'} \end{cases} = \begin{cases} -720000 \\ -960000 \\ 720000 \\ 960000 \end{cases}$$

Nodal Displacements

$$10^{8} \begin{bmatrix} 2.88 & 0 \\ 0 & 5.12 \end{bmatrix} \begin{Bmatrix} D_{3} \\ D_{4} \end{Bmatrix} = \begin{Bmatrix} 720000 \\ 960000 \end{Bmatrix}$$
$$\begin{Bmatrix} D_{3} \\ D_{4} \end{Bmatrix} = 10^{-4} \begin{Bmatrix} 25 \\ 18.75 \end{Bmatrix} m$$

$$\mathbf{d}_{2\times 1}' = \mathbf{T}_{2\times 4}\mathbf{d}_{4\times 1}$$

$$\mathbf{d}_{2\times 1}' = \begin{bmatrix} 0.6 & 0.8 & 0 & 0 \\ 0 & 0 & 0.6 & 0.8 \end{bmatrix} \begin{cases} 0 \\ 25(10^{-4}) \\ 18.75(10^{-4}) \end{cases} = \begin{cases} 0 \\ 30(10^{-4}) \end{cases} m$$

$$\varepsilon = \frac{d_2' - d_1'}{L} = \frac{30 \times 10^{-4}}{5} = 6 \times 10^{-4}$$

$$\sigma = E\left(\varepsilon - \varepsilon_0\right) = 2\left(10^{11}\right)\left(6 \times 10^{-4} - 6 \times 10^{-4}\right) = 0$$

$$\mathbf{d}_{2\times 1}^{'} = \mathbf{T}_{2\times 4}\mathbf{d}_{4\times 1}$$

$$\mathbf{d}_{2\times 1}' = \begin{bmatrix} 0.6 & -0.8 & 0 & 0 \\ 0 & 0 & 0.6 & -0.8 \end{bmatrix} \begin{cases} 25(10^{-4}) \\ 18.75(10^{-4}) \\ 0 \\ 0 \end{cases} = \begin{cases} 0 \\ 0 \end{cases} m$$

$$\varepsilon = \frac{d_2' - d_1'}{L} = 0$$

$$\sigma = E(\varepsilon - \varepsilon_0) = 0$$