

UNIT-I

- *1. Natural C.S
- *2. Man made C.S
- *3. Combinational C.S
- *4. Linear & non-linear C.S
- *5. Lumped & Distributed C.S
- *6. Time Variant & invariant C.S
- *7. Continous & non-Continous C.S.
- *8.
- *9. Open-loop C.S
- *10. Closed-loop C.S.

* UNIT-I *

* System :-

* System is a combination of Elements or Components which are connected in a sequence to perform specific function is called a "System."

* Every physical object is a system.

* Ex:-

* A class room is a good example of physical system.

* A room along with the combination of benches, lights, fans, black boards etc. can be called a class room which acts as an elementary system.

* A lamp is made up of glass, filament is a physical system.

* Similarly, systems can be any type i.e., physical, biological, mechanical, electrical, etc.

* Control system :-

* Control means to regulate, to direct or to command.

* Control system is an arrangement of diff physical elements connected in such a manner.

So as to regulate direct or command itself
of some other system.

* In a system output quantity is controlled by varying its input quantity, the system is called C.S.

* If quantity is called Controlled Variable of response.

* If quantity is called Command Signal or Excitation

* Classification / Types of C.S. :-

* 1. Natural C.S. :-

* The biological system, Systems inside human being is called Natural C.S.

* Respiratory System inside human being, thy. system activates certain glands & regulates temperature of human body.

* The systems that are designed & manufactured by human beings. Such systems like Vehicles, Robots etc.

* Ex:- An automobile system with gears, accelerator, breaking system is a good Example

*3. Combinational C.S. \rightarrow

* It is one having Natural & Man-made C.S.

C.S.

* i.e., driver driving a vehicle

*4. Time varying & Time-Invariant C.S. \rightarrow

* If the system's o/p changes with time then

time variant C.S.

\rightarrow * In which parameters of the system are

Varying with time.

* Ex: Space vehicles whose mass decreases with time

as it leaves Earth mass is a parameter that decreases.

* Ex: Rocket uses dynamic surge damping, can change

with time bcoz their air density changes

with altitude.

\rightarrow The parameters of the system are not changing w.r.t time.

i.e., if its o/p's are function of time but

parameters are independent of time.

* Ex: All electrical networks consist R, L, C are time invariant.

Because, Values of R, L, C are Constant.

* 5 linear & non-linear systems

* If a system obeys Homogeneity principle of
Satisfies Super position principle of Homogeneity of adding
then it is linear system.

* Super position theorem states that the response
Produced by simultaneous application of two
different forcing functions is sum of individual systems.

* Ex: $y(t) = at^2$ → Non-linear.

* 6. Continuous time & Discrete time systems:

* All system variables are functions of continuous
variable time "t". At any time "t", depends on

* Ex: Speed control of a DC motor using Tachometer.

* One of more systems variables are known only at
certain discrete interval of time, they are not
continuously depend on time.

* Ex: MP of MC system.

*7. Deterministic or Random/ Stochastic System

- * A C.S. is deterministic when its response to I/P as well as behaviour to External disturbances is predictable.
- * If such response is unpredictable then it is said to be Random Systems.

*8. lumped parameters or Distributed parameter C.S.,

- * A C.S. can be described by Ordinary D.Eq, is called lumped parameter C.S.

*Ex:- Electrical Net with diff parameters R, L, C.

* A C.S. that can be described by Partial D.Eq,

*Ex:- Transmission line.

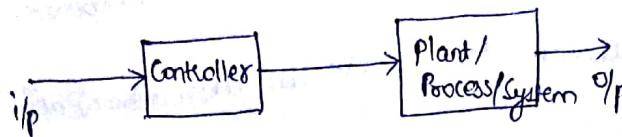
*9. SISO, & MIMO systems,

* A system having only single I/p & o/p is called SISO

* A system having multi I/p & o/p is called MIMO

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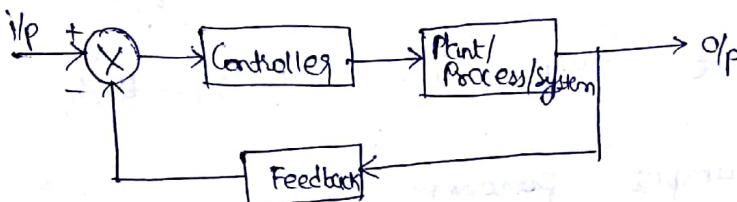
- * Open loop & closed loop C.S.:



* Controlling action is not dependant on i/o/p.

* o/p is Independent on i/p.

* Systems are more stable.



* Controlling action is dependant on i/p.

* o/p can be controlled by changing the i/p.

* Systems are not stable but more accurate & reliable.

* user can control the o/p i.e., desired o/p can be obtained.

* O.L.C.S.:

* A system in which o/p depends on i/p but controlling action is independent of i/p i.e., o/p quantity has no effect upon on i/p quantity. is called an O.L.C.S.

* o/p is not fed back to the i/p action

*Ex:-

1. Automatic washing machines

2. Fan with Blades.

- *3. TV without Remote.
- *4. Automatic toaster.
- *5. Automatic door system.
- *6. Traffic light controller without measuring density of traffic.
- *7. by splinker.
- *8. Stepper motor positioning system.

*Advantages :-

- *) O.L. Systems are simple & economical.
- *) Easier to Construct.
- *) more stable.

*Disadvantages :-

- *) 1. O.L. Systems are inaccurate & unreliable.
- *) 2. the changes in the o/p due to the external disturbances are not detected automatically.
- *) An physical system which doesn't automatically detect the variation in its o/p is called O.L. system.

*G.L.C.S :- / {Automatic CS}

* A system in which the o/p has an effect upon i/p quantity in such a manner to maintain desired o/p value is called G.L.C.S.

- *Ex:-
- *1. Fan with blades & regulator.
 - *2. manual washing machines.
 - *3. Automatic tuning Control.
 - *4. TV with remote.
 - *5. Stabilizer
 - *6. missile launching vehicle system.

* Advantages:-

- *) The closed loop systems are accurate.
- *) These systems are less affected by noise.
- *) Sensitivity of the system may be (greatly) made small to make the system more stable.

* Disadvantages:-

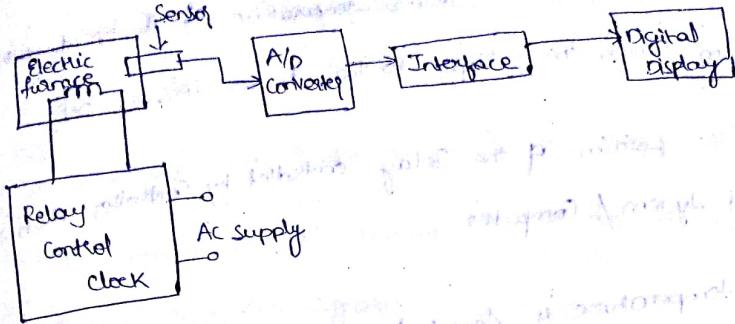
- *) less stable due to the F/B, the system may lead to oscillatory response.
- *) F/B Reduces the overall gain of the system.
- *) Expensive
- *) Complex to design.

* Examples of C.S:-

- *) 1. Temperature Control System:-
(T.C.S)
- *) Open loop system:-

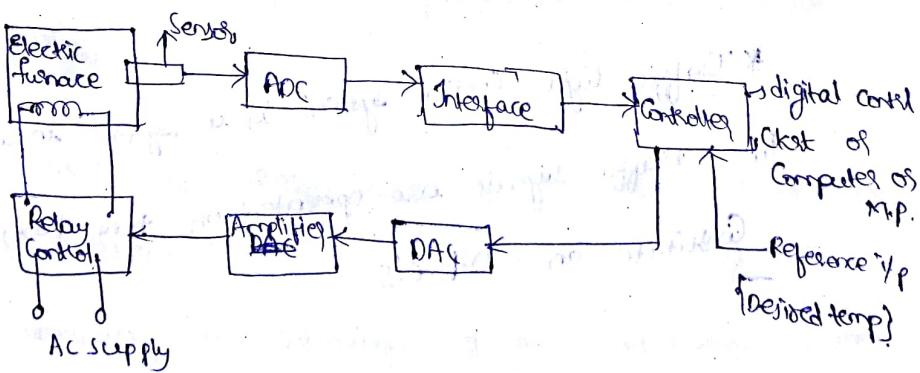
* Electric furnace is an open loop CS.
* The op in this system is desired temperature.

The temperature of the system is raised by heat generated by heating element (coil), the op of temperature depends on time during which the supply to the heater remains "on".



- * The ON & OFF supply is controlled by time setting of the relay
- * the temperature is measured by a sensor which gives an analog voltage corresponding to the temperature of the furnace.
- * This analog is converted to digital {using ADC} & is given to LCD or display system. which displays the temperature of furnace.
- * In this system if there is an change in the op temperature then the time setting of the relay is not altered automatically.

* Closed loop TC system:



- * The op. of the system is the desired temperature & it depends on the time during which the supply to the heater remains ON.
- * ON & OFF position of the relay controlled by Controller which is a digital system / Computer.
- * Actual temperature is sensed by sensor & connected into digital by ADC, the Computer or Controller reads the actual temperature & compares desired / required temperature if it finds any difference then it sends a signal to switch ON or OFF the Relay through DAC & amplifier.

* Hence, the system automatically acts any changes in the op. (required temperature).

- * Hence, it is a C.L.C.S.
- * TRAFFIC light CONTROL SYSTEM :-

*① open loop T.L.C.S.

* Traffic light control system is a system in which the traffic signals are operated on time bases constitute an O.P.C.S.

it depends on the
remaining ON.

Controller, which is
connected to

connected into
leads the
mixed temperature
a signal

out of DAC &
inverter

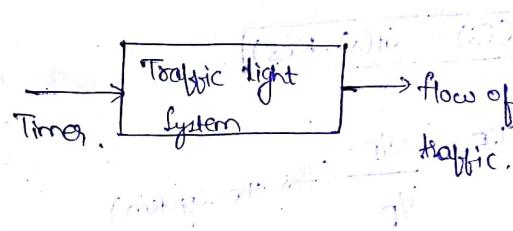
get in the op.

* The sequence of control signals are based on time slot given to each signal.

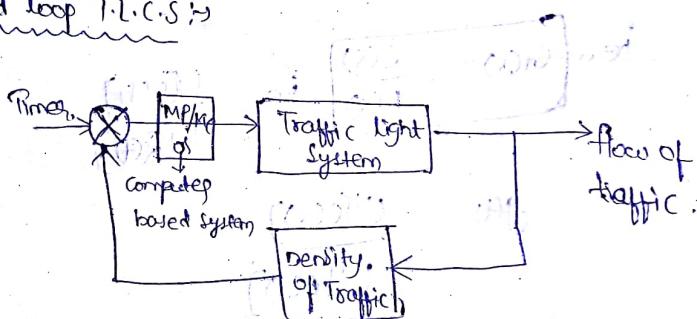
* The time slots are decided based on the traffic study.

The system will not measure the density of the traffic before giving the signals.

* Since, time slot doesn't change according to density, the system is O.L. System.



*2. Closed Loop T.L.C.S:



* If the time slot of the signals are decided based on density of traffic.

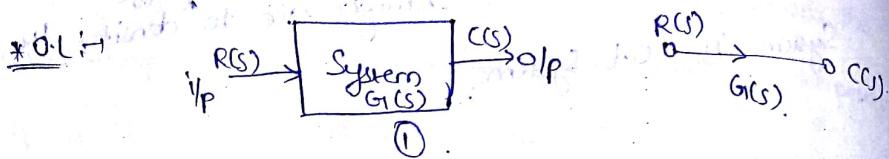
* In closed loop Traffic light C.S. The density of the traffic is measured on all sides & information is fed to a computer, the timings of the control signals are decided by the computer based on density of traffic.

* Since, the C.L.S dynamically changes the timmings, the flow of vehicles will be better than Q.L.S.

* Differences b/w OLCs & CLCs are in cam scanner

* F/B characteristics.

* Effect of feedback in CCS

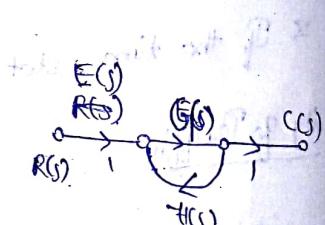
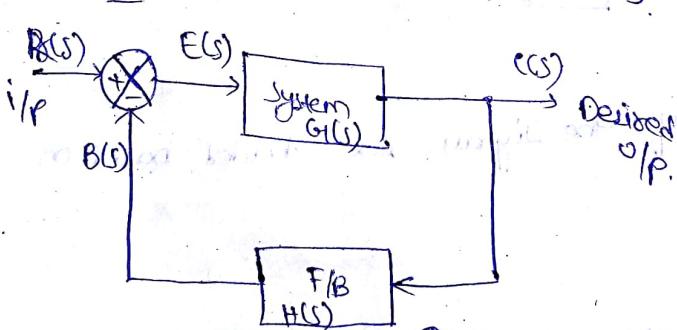


$$CCS = G(s) \cdot R(s)$$

$$TF = \frac{O/P}{I/P} \text{ (for the system)}$$

$$\text{i.e., } G(s) = \frac{CCS}{R(s)} \text{ i.e., } \frac{L[CCS]}{L[R(s)]}$$

$$J(s) = \frac{L[CCS]}{L[R(s)]}$$



$$\text{Ans: } \frac{G(s)}{R(s)} = \text{Desired Probability of having}$$

green light without experiencing any other

yellow phase or total phase at that time

* Feed back Systems play an important role in modern Engineering because they have the possibility for being adaptive to perform their assigned tasks automatically.

- * A Non feed back { O.L. System } is represented by block diagram & signal flow graph, is shown in figure ①.
- * There is no provision within the system for supervision of o/p & no mechanism is provided to check the system behaviour.

* F/B Control system or C.L.C.S or Automatic C.s is represented by Block & signal flow graphs are shown in ②. figure.

* It is having two signals, one is i/p signal & other is F/B signal derived from the o/p of the system.

* the F/B signal gives system capability to act as self correcting mechanism.

$$E(s) = R(s) - B(s)$$

$$B(s) = E(s) \cdot G(s)$$

$$E(s) = R(s) - B(s)$$

? Scanners

$$G(s)$$

$$\begin{array}{c} G(s) \\ \text{---} \\ \text{---} \\ H(s) \end{array}$$

at P

$$\Rightarrow C(s) = G(s) [R(s) - B(s)]$$

$$\Rightarrow C(s) = G(s).R(s) - G(s).B(s)$$

$$\Rightarrow C(s) = G(s).R(s) - G(s).C(s).H(s)$$

$$\Rightarrow C(s) = G(s) [R(s) - C(s).H(s)]$$

$$\Rightarrow C(s) [1 + G(s).H(s)] = G(s).R(s)$$

$$\Rightarrow \frac{C(s)}{R(s)} = \frac{G(s)}{1 + G(s).H(s)}$$

* gain can be controlled

* Sensitivity of the system can be reduced by $\frac{1}{1 + G(s).H(s)}$

* F/B Characteristics

* Effect of F/B in L(s)

* F/B means automatic regulation of control B.

→ ① Controlled variable accurately follows the desired O/p.

→ ② Effect on the controlled variable on the

External disturbance other than those

associated with the F/B sensor are greatly reduced.

Effect

③

on
dev

* There

E
P

R

o

* F/B

A G

* Effect

* 1.

* 2.

③ Effect of variation in controller & process parameters (Forward path)

on the system performance is reduced to acceptable level.

* These variations are caused due to Component age & Environmental change, temperature etc i.e..

F/B in Control loop allows accurate control of the plant even when process gain or controlled parameters are not known accurately.

* F/B in C.S. greatly improves the speed of its response compared to the speed of the response in O.L. system.

* Effect of F/B on C.S. :-

* 1. Effect of F/B on gain.

* 2. Consider an O.L. system with overall transfer function as $C(s) = \frac{G(s)}{1+G(s)H(s)}$.

* If the F/B $H(s)$ is introduced then its overall gain becomes $\frac{G(s)}{1+G(s)H^2(s)}$.

For a negative F/B C.S. the gain is reduced by a factor of $\frac{1}{1+G(s)H(s)}$.

* due to the negative F/B overall gain of the system is deduced.

* Reduction of parameter variation by -ve F/B is

* Parameters of the system may vary with age, with changing environment

The main purpose of F/B in CS is to reduce the sensitivity of the system to parameter variation.

* Sensitivity is a measure of effectiveness of F/B in reducing the influence of these variations on system performance.

* Let us define sensitivity on a quantitative basis.

* for open loop case

$$C(s) = R(s) \cdot G(s)$$

* Suppose due to parameter variation $G(s)$ is changed to $G(s) + \Delta G(s)$

$$\text{where } G(s) \gg \Delta G(s)$$

* The o/p of o.l system changes to

$$C(s) + \Delta C(s) = [G(s) + \Delta G(s)] R(s)$$

$$\Delta C(s) = \Delta G(s) \cdot R(s) \rightarrow ①$$

* For closed loop case the O/P is

$$C(s) = \frac{G(s)}{1+G(s) \cdot H(s)}$$

$$\text{But } G(s) = G(s) + AG(s); G(s) \gg AG(s)$$

$$\Rightarrow C(s) + \Delta C(s) = \frac{G(s) + AG(s)}{1 + [G(s) + AG(s)] \cdot H(s)} = \frac{G(s) + AG(s)}{1 + G(s) \cdot H(s)}$$

$$\Rightarrow \Delta C(s) = \frac{\Delta G(s)}{1 + G(s) \cdot H(s)} \rightarrow ②$$

* From eq. ① & ②, seen that in comparison to O.L & C.L

due to variation in $G(s) \cdot H(s)$ is reduced by a factor of $\frac{1}{1 + G(s) \cdot H(s)}$.

* Effect of F/B on Time Constant of CS :-

* Consider an O.L system with overall T.F of

$$G(s) = \frac{K}{1 + ST}$$

with this system is subjected to unit step if the

response can be obtained as

$$\frac{C(s)}{R(s)} = G(s); R(s) = 1, \text{ unit step.}$$

$$\Rightarrow C(s) = \left(\frac{K}{1+ST} \right) \cdot \frac{1}{s}$$

$$\Rightarrow C(s) = \frac{K}{s(1+ST)} = \frac{A}{s} + \frac{B}{1+ST}$$

$$C(t) = L^{-1}[C(s)]$$

$$\Rightarrow A(1+ST) + SB = K$$

* for $s=0 \Rightarrow A = K$

* for $s = -\frac{1}{T} \Rightarrow B = -KT$

$$\Rightarrow C(s) = \frac{K}{s} - \frac{KT}{1+ST} = K \left(\frac{1}{s} - \frac{T}{1+ST} \right)$$

$$\Rightarrow C(t) = K \left[1 - e^{-\frac{t}{T}} \right] \rightarrow ①$$

* C.L.C.S.:

$$C(s) = \frac{G(s) R(s)}{1+G(s)H(s)} = \frac{\frac{K}{1+ST}}{1+\left(\frac{K}{1+ST}\right)h} = \frac{\frac{K}{1+ST}}{1+\frac{Kh}{1+ST}} = \frac{K}{1+Kh}$$

$$C(s) = \frac{k}{(1+ST)+kh} \cdot \frac{1}{s}$$

$$= \frac{1}{s} \cdot \frac{\frac{K}{1+Kh}}{s + \left(\frac{1+Kh}{T} \right)} = \frac{A}{s} + \frac{B}{s + \left(\frac{1+Kh}{T} \right)}$$

$$A \left(s + \left(\frac{1+Kh}{T} \right) \right) + BS = \frac{K}{1+Kh}$$

$$\text{at } s=0 \Rightarrow B = \frac{K}{1+Kh}$$

$$\text{at } s = -\left(\frac{1+Kh}{T} \right) \Rightarrow B = -\frac{K}{1+Kh}$$

$$C(s) = \frac{K}{1+kh} \left[\frac{1}{s} - \frac{1}{s + \left(\frac{1+kh}{T} \right)} \right]$$

$$C(t) = \frac{K}{1+kh} \left[1 - e^{-\frac{t}{\left(\frac{1+kh}{T} \right)}} \right]$$

$$(1+kh)^{-1} = \frac{T}{1+kh}$$

* From the F/B the new time constant is $\frac{T}{1+kh}$ for the system.

Values of "h" $\epsilon [0, 1]$. If $K > 1$, the time constant is $< T$.

\therefore In O.L System the time constant is $\frac{1}{1+kh}$.

In C.L System the time constant is $\frac{1}{1+kh}$.

* So, less Time Constant in C.L System, if the time constant is less, the system gives faster response.

* Hence, F/B improves time response of a system.

* 4 Effect of F/B on Stability

* F/B reduces the Time Constant & makes the system respond more fast, hence, transient response decays more quickly.

* Consider the O.L TF of a system as $C(s) = \frac{K}{R(s) \cdot 1+ST}$.

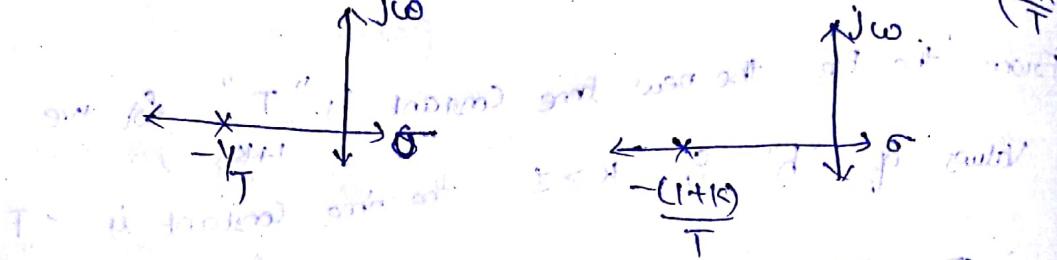
Open loop pole is located at $s = -\frac{1}{T}$.

* Consider unity negative F/B is introduced in the system.

then overall TF of C.L system becomes

$$* \frac{C(s)}{R(s)} = \frac{\frac{K}{1+ST}}{1 + \left(\frac{K}{1+ST}\right)(1)} = \frac{\frac{K}{1+ST}}{1+ST+\frac{K}{1+ST}} = \frac{\frac{K}{1+ST}}{\frac{1+ST+K}{1+ST}} = \frac{K}{1+ST+K}$$

* Then, the closed loop pole is located at $s = -\left(\frac{1+K}{T}\right)$.



* F/B Controls the Time response, Stability of a system depends on location of poles in S-plane. Thus, it can be concluded that the F/B effect on the stability of F/B may be improve the unstable system response.

* Effect of F/B on disturbance,

* Every C.S has some non-linearities present in it it effects on output of the system.

* Some External disturbance (signal or noise) also make the system output inaccurate.

* The effect of External disturbance one high frequency noise in Electronic application, thermal noise in Amplifier, C.R.O, etc.

$$\frac{k_T}{s + \left(\frac{1+k}{T}\right)}$$

$$= -\left(\frac{1+k}{T}\right)$$

* The disturbances may be present in forward paths as well as in F_B paths. [read these two from F/B].

* 07/11/17 *

* Mathematical modeling of a control system: →

* Any system can be described by math. Eq's → the math eq's

- * Types: →
- * 1. Mechanical Translational System. {M.T.S} → Diff Eq or P.D.E
 - * 2. Mechanical Rotational System. {M.R.S}

Def:

* The set of math eq's describing the dynamic characteristics of a system is called math. modeling of a system.

* These math eq's may be D.Eq's or P.D.Eq's.

* The response (pos) of the system can be studied by solving the diff eq for various I/P conditions.

* Two types mech. systems are in C.S. they are, ① & ②

* M.T.S: →

* Consider a Mech. System in which motion takes place along a straight line. Such systems are translational.

* These systems are characterised by

① linear displacement

② Velocity

③ Acceleration

* There are 3 basic elements in translational system, they are

(i) Mass (m), (ii) Dashpot & Friction (B), (iii) Spring (K).

* Newton's 2nd law :-

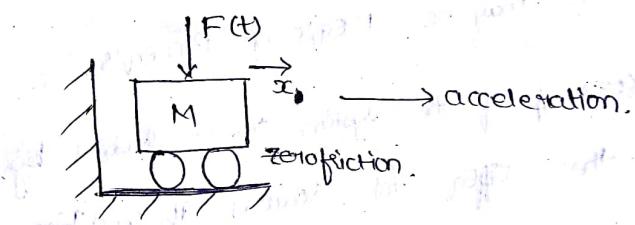
* For any body the sum of applying forces is equal to the force resulting the motion in any given direction.

is equal to zero.

(F)

* The sum of applying forces is equal to the sum of resulting forces.

(i) Mass (m):



$F(t)$ → applied force; x = displacement.

* Consider an ideal mass element which has negligible friction & elasticity. Let a force is applied on it, which produces a displacement "x(t)" in the direction of applied force "F(t)". The mass (m) will offer an opposing force

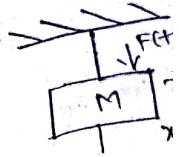
* Which is proportional to acceleration of the body.

$F_m \propto a$. of opposing force due to mass element first

$$F_m = Ma$$

$$F_m = M \cdot \frac{dx^2}{dt^2}$$

{ M = Co-efficient}



$$F_m = M \cdot \frac{dx^2}{dt^2}$$

(ii) Dashpot

Can be

* Dashpot

air -

* Friction

between

* Let f

force

Both the end are free but above one end is fixed

real system, they are

Forces &

given direction

cal to the

negligible

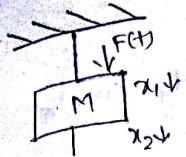
it, which

of applied

resisting force

body.

Element F_M .



$$F_M = M \cdot \frac{d^2(x_1 - x_2)}{dt^2}$$

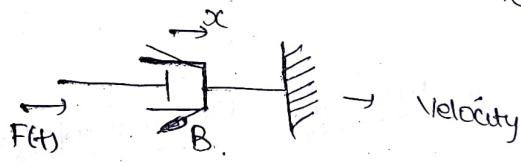
(ii) Dashpot or Friction (B):

* The Friction Existing in mechanical System can be represented by Dashpot

* Dashpot is a piston moving in a cylinder filled with air fluid

* Friction may be between moving element & fixed support or between two moving surfaces.

* Let force "F" applied on it, the dashpot will offer an opposing force which is proportional to velocity of the body.



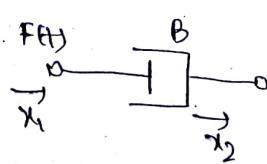
$$F_B \propto v.$$

$$F_B = B \cdot v.$$

F_B = opposing force due to friction of dashpot.

$$F_B = B \cdot \frac{dx}{dt}$$

{ Both the ends are free but above one end is fixed }

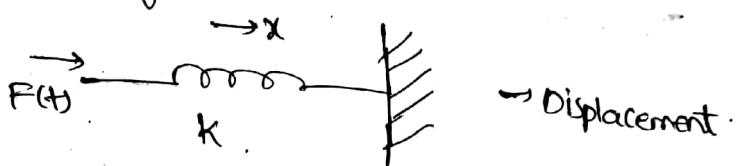


$$F_B = B \cdot \frac{d}{dt} (x_1 - x_2).$$

*(iii) Spring (k):

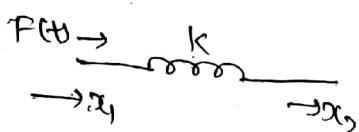
* The ~~electro~~ elastic deformation of the body can be represented by spring.

* Let a force is applied on it, the spring will offer an opposing force, which is proportional to displacement of the body.



$$F_k \propto x. \quad \{ F_k = \text{opposing force due to spring } (k) \}$$

$$F_k = kx$$



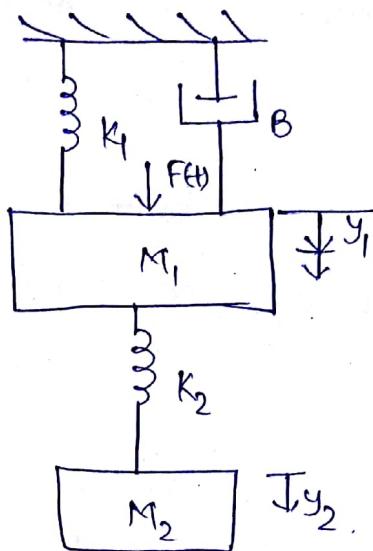
$$F_k = k(x_1 - x_2)$$

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* Determination of TF of mechanical Translational system:

* write the D.E. of mechanical system & T.F. $\frac{Y_1(s)}{F(s)}$

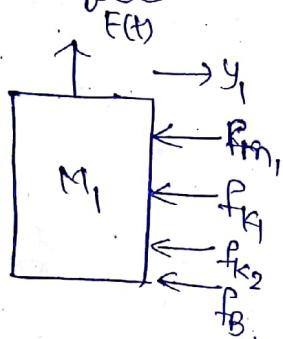
* ①



* so

Free body diagram ↗

* for M_1



$$F(t) = P_{f_m1} + P_{f_K1} + P_{f_K2} + P_{f_B}$$

$$F(t) = M_1 \cdot \frac{d^2y_1}{dt^2} + K_1 y_1 + B \cdot \frac{dy_1}{dt} + K_2 (y_1 - y_2)$$

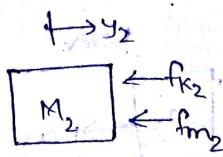
* Apply LT on B.S.

$$\Rightarrow F(s) = M_1 s^2 \cdot Y_1(s) + K_1 Y_1(s) + B \cdot S \cdot Y_1(s) + K_2 (Y_1(s) - Y_2(s))$$

$$\Rightarrow F(s) = Y_1(s) [M_1 s^2 + K_1 + SB + K_2] - Y_2(s) [K_2] \rightarrow ①$$

$$G \cdot T.F. \cdot \frac{Y_2(s)}{F(s)}$$

$\cancel{\text{from } M_2 \text{ is}}$



$$\rightarrow f_{k2} + f_{m2} = 0.$$

$$\Rightarrow k_2(Y_2 - Y_1) + M_2 \cdot \frac{d^2 y_2}{dt^2} = 0.$$

$$\Rightarrow k_2 Y_2(s) - k_2 Y_1(s) + M_2 \cdot s^2 \cdot Y_2(s) = 0.$$

$$\Rightarrow k_2 Y_1(s) = k_2 Y_2(s) + M_2 s^2 \cdot Y_2(s).$$

$$\Rightarrow Y_1(s) = Y_2(s) + \frac{M_2 s^2 \cdot Y_2(s)}{k_2}. \rightarrow ②$$

From ① & ② we get.

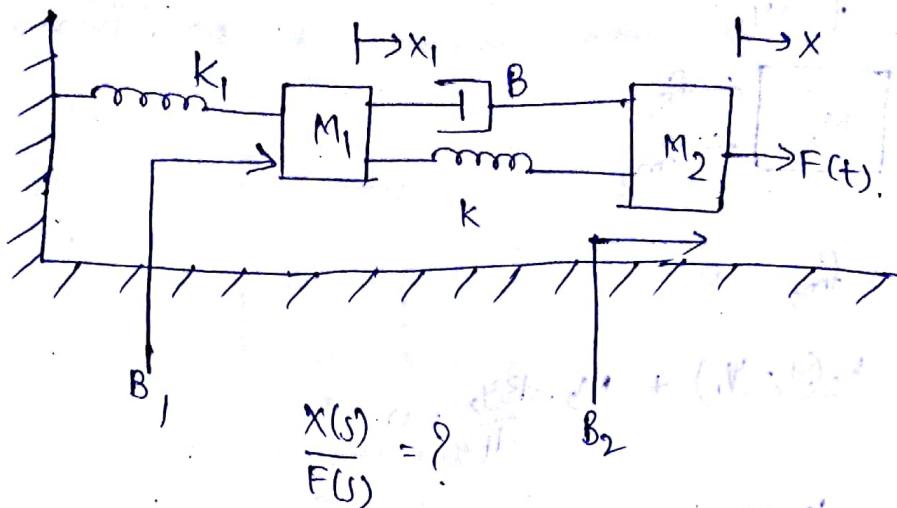
$$\Rightarrow F(s) = Y_2(s) \left[1 + \frac{M_2 s^2}{k_2} \right] \left[M_1 s^2 + k_1 + S_B + k_2 \right]$$

$$\Rightarrow \frac{Y_2(s)}{F(s)} = \frac{1}{(M_1 s^2 + k_1 + S_B + k_2) \left[1 + \frac{M_2 s^2}{k_2} \right] - k_2}$$

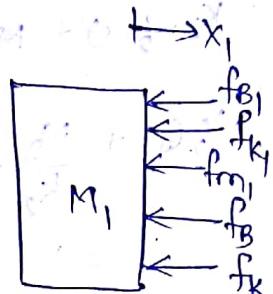
$$\Rightarrow \frac{Y_2(s)}{F(s)} = \frac{k_2}{M_1 s^2 k_2 + k_1 k_2 + S_B k_2 + k_2^2 + M_1 s^4 M_2 + M_2 s^2 k_1 + S_B M_2 - M_2 s^2 k_2 - k_2}$$

$$\Rightarrow \frac{Y_2(s)}{F(s)} = \frac{k_2}{(k_2 + M_2 s^2)(M_1 s^2 + S_B + k_1 + k_2) - k_2^2}$$

(2)



* Sol) * 2 Mass elements \rightarrow 2 D.Eq's.

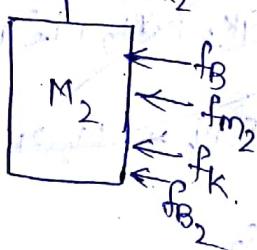


$$* f_k + f_B + f_k \neq f_{m_1} + f_{B_1} = 0$$

$$\Rightarrow B \cdot \frac{d(x - x_0)}{dt} + B_1 x_1 + K(x_1 - x_0) + m_1 \frac{d^2 x_1}{dt^2} + B_1 \frac{dx_1}{dt} = 0$$

$$\Rightarrow B s x_1(s) + K x_1(s) + K [x_1(s)] \neq K x_1(s) + m_1 s^2 x_1(s) + (B s x_1(s)) = 0$$

$$= [K x_1(s)] = x_1(s) [B + K_1 + K + m_1 s^2 + B_1]$$



$$\Rightarrow x_1(s) = \frac{(K + B s) x(s)}{B s + K_1 + K + m_1 s^2 + B_1 s} \rightarrow ①$$

$$f_B + f_{m_2} + f_k + f_{B_2} = F(t).$$

$$\Rightarrow B \cdot \frac{d(x_0 - x)}{dt} + m_2 \cdot \frac{d^2 x_2}{dt^2} + K(x_0 - x) + B_2 \cdot \frac{dx}{dt} = 0 \quad (t)$$

$$= BSx(s) - BSx_0(s) + m_2 \cdot s^2 \cdot X(s) + K \cdot X(s) - Kx_0(s)$$

$$+ B_2 s X(s) = F(s).$$

$$\Rightarrow X(s) [B + m_2 s^2 + K + B_2] - [K + B] X(s) = F(s)$$

$$[B + K_1 + K + m_2 s^2 + B_2]$$

$$- F(s).$$

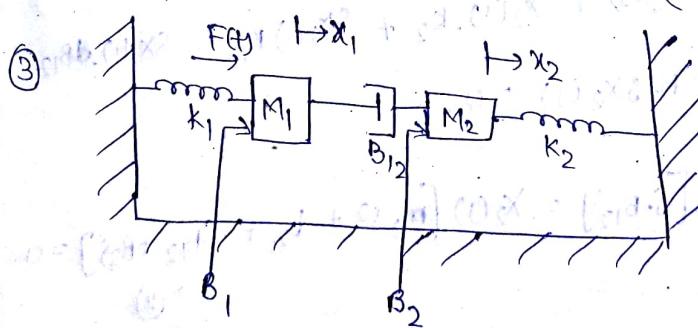
$$\Rightarrow -X(s) [K + B_2] + X(s) [BS + m_2 s^2 + K + B_2 s] = F(s)$$

$$\Rightarrow \frac{-(K + B_2) \cdot X(s)}{BS + K_1 + K + m_1 s^2 + B_1 s} + X(s) [BS + m_2 s^2 + K + B_2 s] = F(s)$$

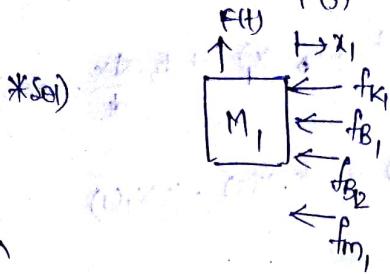
$$\Rightarrow X(s) \left[\frac{-(K + B_2)}{BS + K_1 + K + m_1 s^2 + B_1 s} + \frac{[BS + m_2 s^2 + K + B_2 s]}{BS + K_1 + K + m_1 s^2 + B_1 s} \right] = F(s)$$

$$\Rightarrow \frac{X(s)}{F(s)} = \frac{BS + K_1 + K + m_1 s^2 + B_1 s}{(BS + m_2 s^2 + K + B_2 s)(BS + K_1 + K + m_1 s^2 + B_1 s)}$$

$$- (K + B_2)^2$$



$$\frac{x_1(s)}{F(s)} = ?; \quad \frac{x_2(s)}{F(s)} = ?$$



From ① & ② we
⇒ $F(s) =$

$$\Rightarrow F(s) =$$

$\cancel{\times} \frac{x_1(s)}{F(s)}$

$$F(t) = k_1 x_1 + B_1 \cdot \frac{dx_1}{dt} + B_{12} \cdot \frac{d(x_1 - x_2)}{dt} + m_1 \cdot \frac{d^2 x_1}{dt^2}$$

$$F(s) = K_1 x_1(s) + B_1 s \cdot x_1(s) + B_{12} s \cdot x_1(s) + B_{12} s \cdot x_2(s)$$

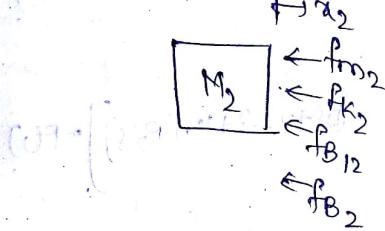
$$F(s) = e^{-st} x_1(s) [K_1 + B_1 s + B_{12} s + m_1 s^2] - (B_{12} s) x_2(s)$$

From ① &

$$\Rightarrow F(s)$$

①

$$\Rightarrow F(s)$$



$$\Rightarrow m_2 \cdot \frac{d^2 x_2}{dt^2} + k_2 x_2 + B_{12} \cdot \frac{d(x_2 - x_1)}{dt} + B_2 \frac{dx_2}{dt} = 0.$$

$\cancel{\times} \frac{x_2(s)}{F(s)}$

$$\Rightarrow m_2 \cdot s^2 \cdot x_2(s) + x_2(s) \cdot k_2 + s x_2(s) B_{12} - s x_1(s) B_{12} + B_2 s x_2(s) = 0.$$

$$\Rightarrow x_1(s) [s \cdot B_{12}] = x_2(s) [m_2 s^2 + k_2 + s B_{12} + B_2 s] = 0.$$

②

from ① & ② we get

$$\Rightarrow F(s) = X_1(s) [k_1 + B_1 s + B_{12} s + m_1 s^2] - B_{12} s \left[\frac{S B_{12} \cdot X_1(s)}{k_2 + S B_{12} + S B_2} \right] + m_2 s^2$$

$$\Rightarrow F(s) = X_1(s) \left[k_1 + B_1 s + B_{12} s + m_1 s^2 - \frac{s^2 B_{12}^2}{k_2 + S B_{12} + S B_2 + m_2 s^2} \right]$$

$$\Rightarrow \frac{X_1(s)}{F(s)} = \frac{k_2 + S B_{12} + S B_2 + m_2 s^2}{(k_1 + B_1 s + B_{12} s + m_1 s^2)(k_2 + S B_{12} + S B_2 + m_2 s^2) - S^2 B_{12}^2}$$

from ① & ② we get.

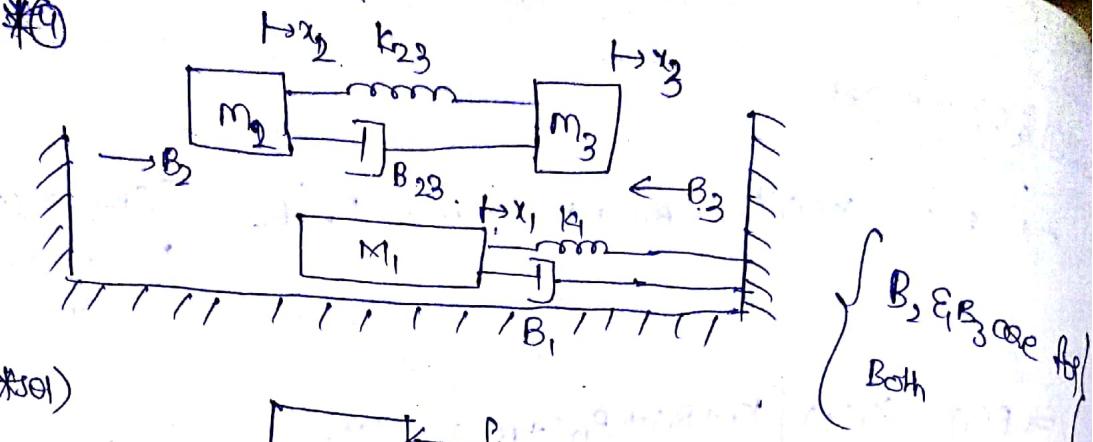
$$\Rightarrow F(s) = (k_1 + B_1 s + B_{12} s + m_1 s^2) \left[\frac{m_2 s^2 + k_2 + S B_{12} + S B_2}{S B_{12}} \right] \cdot X_2(s) - S B_{12} \cdot X_2(s)$$

$$\Rightarrow F(s) = X_2(s) \left[\frac{(k_1 + B_1 s + B_{12} s + m_1 s^2)(m_2 s^2 + k_2 + S B_{12} + S B_2) - S^2 B_{12}^2}{S B_{12}} \right]$$

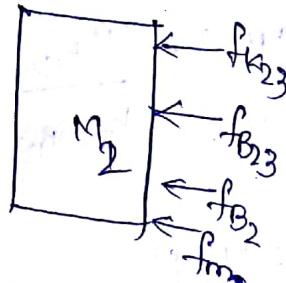
$$\Rightarrow \frac{X_2(s)}{F(s)} = \frac{S B_{12}}{(k_1 + B_1 s + B_{12} s + m_1 s^2)(m_2 s^2 + k_2 + S B_{12} + S B_2) - S^2 B_{12}^2}$$

$\therefore B_{12}$

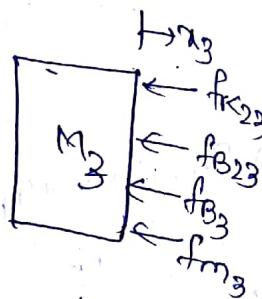
$) = 0$.



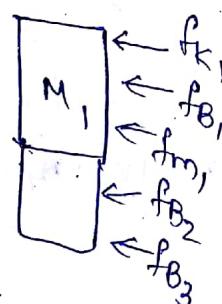
(b)



$$m_2 \cdot \frac{d^2x_2}{dt^2} + k_{23}(x_2 - x_3) + B_{23} \cdot \frac{d(x_2 - x_3)}{dt} + B_2 \cdot \frac{dx_2}{dt} = 0$$



$$m_3 \cdot \frac{d^2x_3}{dt^2} + k_{23}(x_3 - x_2) + B_{23} \cdot \frac{d(x_3 - x_2)}{dt} + B_3 \cdot \frac{dx_3}{dt} = 0$$



$$B_3 \cdot \frac{df_{B1}}{dt} + B_2 \cdot \frac{d(x_1 - x_2)}{dt} + m_1 \cdot \frac{d^2x_1}{dt^2} + k_1(x_1) + B_1 \cdot \frac{dx_1}{dt} = 0$$

* Mechanical Rotational System *

Motion about fixed axis in such systems force gets replaced by movement about fixed axis (i.e., force ~~is~~ distance about fixed axis) which is called Torque.

Sum of Torques applied on a body (or) system must be equal to sum of torques "consumed" by the different elements of the system in order to produce angular displacement (θ), angular velocity (ω), & angular acceleration (α).

* Weight of rotational mechanical system is represented by movement of inertia of the mass. Elastic deformation of the body can be represented by spring (Torsional spring).

* Friction Existing in mechanical rotational system can be represented by Damper / Dumper (B)

θ = Angular displacement \rightarrow Radian

ω = $\frac{d\theta}{dt}$ \rightarrow Velocity \rightarrow Radian/Second.

α = $\frac{d^2\theta}{dt^2}$ \rightarrow Angular acc \rightarrow Radian/Sec.

T = applied Torque \rightarrow Newton/metric.

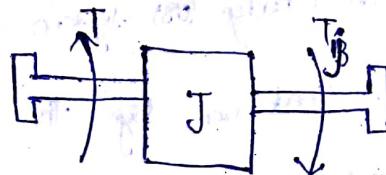
J = moment of inertia \rightarrow $\text{kg m}^2/\text{radian}$

B = Rotational friction co-efficient \rightarrow Newtonmeter
Radian.

K = Stiffness of the spring $\rightarrow \frac{N-m}{\theta}$

* Rotational system has three basic elements

① moment of inertia:



$$\text{and acceleration } I \cdot \frac{d^2\theta}{dt^2}$$

$$I \cdot \frac{d^2\theta}{dt^2} = J \cdot \text{Torque}$$

* Let Torque "T" is applied on it, the opposing torque due to moment of inertia is proportional to angular acceleration of the body.

② Dashpot / Damper:

* Let Torque "T" is applied on it, the opposing torque due to frictional Co-efficient is proportional to angular velocity of the body.



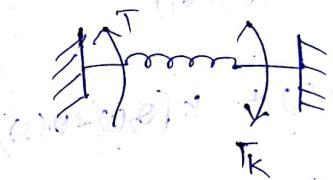
$$T_B \propto \frac{d\theta}{dt}$$

$$T_B = B \cdot \frac{d\theta}{dt}$$

* Torsional spring:

* Let torque "T" is applied on it; the opposing

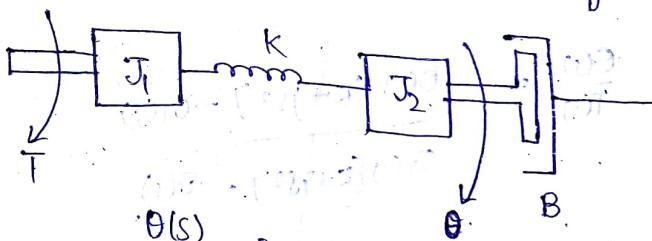
torque due to spring is proportional to angular displacement.



$$* T \propto \theta$$

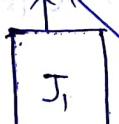
$$T = K\theta$$

* Find the TF of the mechanical rotational system.



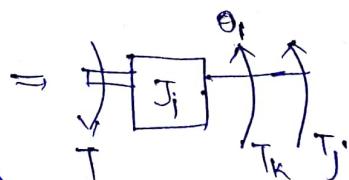
(*)

$$\frac{\theta(s)}{T(s)} = ?$$

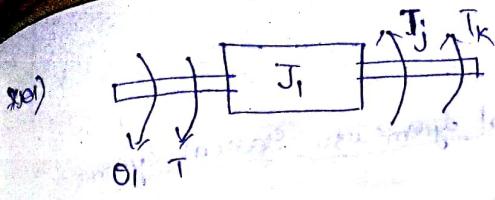


$$T = T_k + T_j$$

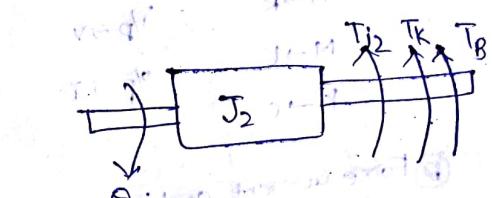
$$T = K\theta + J \cdot \frac{d^2\theta}{dt^2}$$



$$T = K(\theta_2 - \theta_1) + J \cdot \frac{d^2\theta_1}{dt^2}$$



$$T = T_j + T_k = J_1 \frac{d^2\theta_1}{dt^2} + K(\theta_1 - \theta)$$



$$J_2 + T_k + T_B = 0$$

$$J_2 \cdot \frac{d^2\theta}{dt^2} + B \cdot \frac{d\theta}{dt} + K(\theta - \theta_1) = 0$$

Solve we get $\frac{\theta(s)}{T(s)} = \frac{k}{(k + j_2 s^2 + Bs)(k + j_1 s^2) - k^2}$

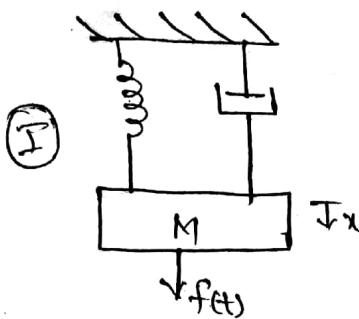
Take 9 pages xelox

$T(s)\theta(s)$

* 19/12/17 *

* Convert the given mechanical systems into Electrical systems

* ①



(1) Force voltage analogous system

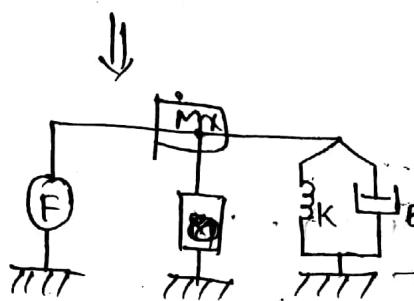
$$F \rightarrow V$$

$$B \rightarrow R$$

$$M \rightarrow L$$

$$K \rightarrow C$$

$$\frac{1}{C} \rightarrow I$$



(2) Force current analogous system

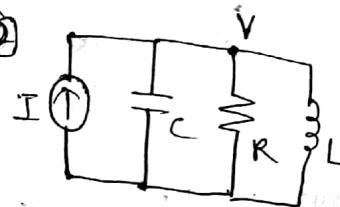
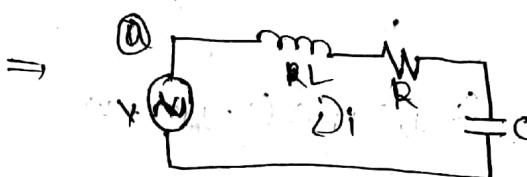
$$F \rightarrow I$$

$$B \rightarrow \frac{1}{R}$$

$$M \rightarrow C$$

$$K \rightarrow L$$

$$\frac{1}{L} \rightarrow I$$



* for ①

$$f(t) = M \cdot \frac{d^2x}{dt^2} + Kx + B \cdot \frac{dx}{dt}$$

$$f(t) @ \quad F(s) = X(s) [Ms^2 + Bs + K]$$

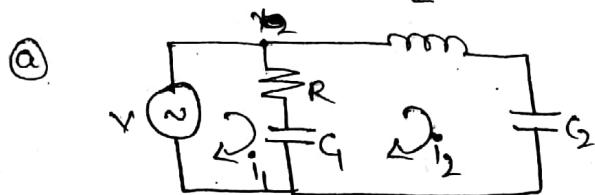
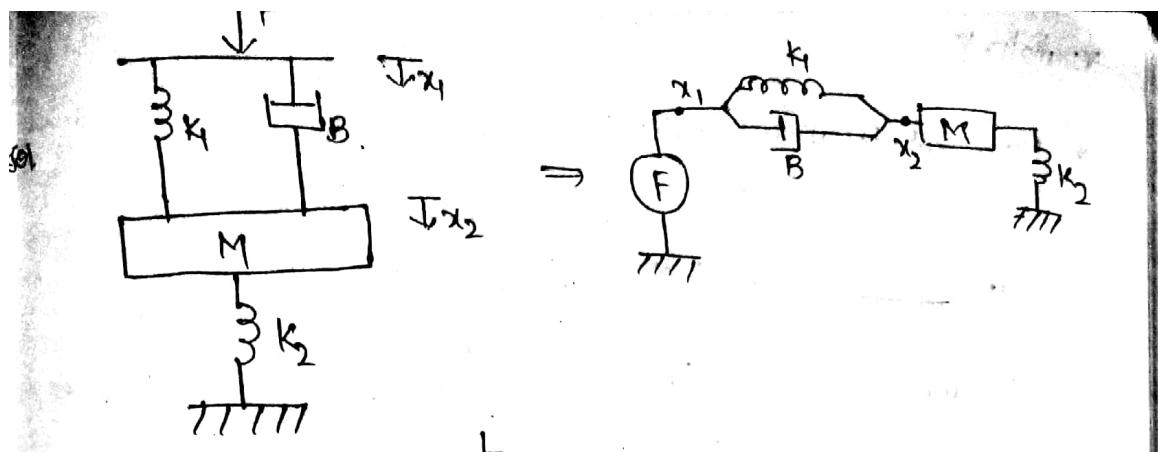
$$V = L \cdot \frac{di}{dt} + R \cdot i + \frac{1}{C} \int v \cdot dt$$

$$V(s) = L \cdot I(s) \cdot s + R \cdot I(s) + \frac{1}{C} I(s) \Rightarrow V(s) = I(s) [Ls + R + \frac{1}{C}]$$

f@②

$$I = C \cdot \frac{dv}{dt} + \frac{V}{R} + \frac{1}{L} \int v \cdot dt$$

$$I(s) = V(s) \left[Cs + \frac{1}{R} + \frac{1}{Ls} \right]$$



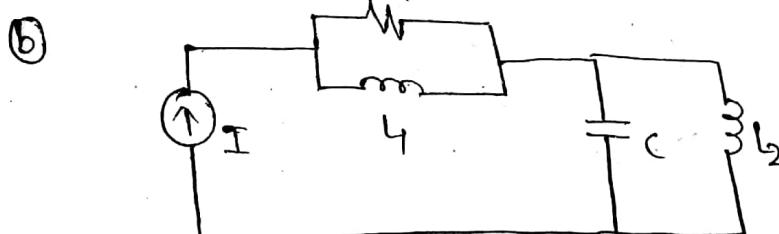
$$V = R(i_1 + i_2) + \frac{1}{C_1} \int (i_1 + i_2) dt$$

$$0 = L \frac{di_2}{dt} + \frac{1}{C_2} \int i_2 dt + \frac{1}{C_1} \int (i_2 - i_1) dt + R(i_2 - i_1)$$

(c)

$$V(s) = R \cdot [I_1(s) - I_2(s)] + \frac{1}{C_2 s} [I_2(s) - I_1(s)]$$

$$0 = L \cdot I_2(s) \cdot s + \frac{1}{C_2 s} I_2(s) + \frac{1}{C_1 s} [I_2(s) - I_1(s)] + R(I_2(s) - I_1(s))$$



like x_{200} from here & next 28/09/17 *

29/12/17

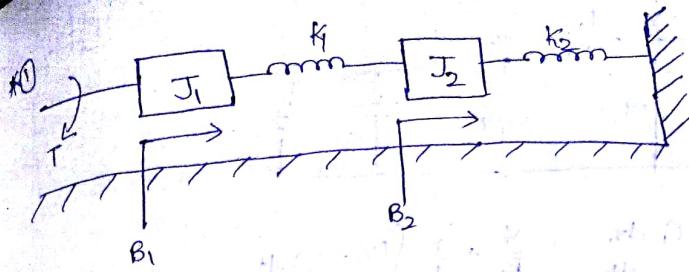
* Mechanical Rotational System →

Parameter	$T \rightarrow V$	$T \rightarrow I$
Torque (T)	V	I
Inertia (J)	L	C
Dashpot (B)	R	X _R
Toroidal Spring (K)	C	M
Junction	load	node
Newton's law.	KVL	KCL

(θ is the initial angle, $\dot{\theta}$ is the angular velocity, $\ddot{\theta}$ is the angular acceleration)

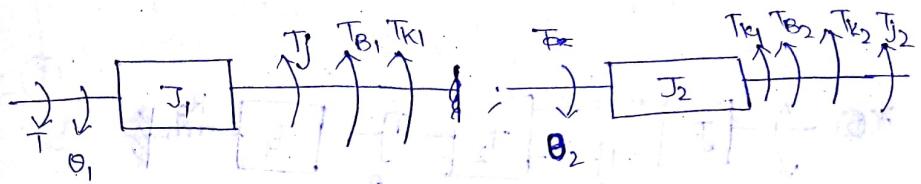
* Mechanical Translational System →

Parameter	$F \rightarrow V$	$F \rightarrow I$
Force (F)	Voltage (V)	Current (I)
mass (m)	Inductance (L)	Capacitance (C)
Dashpot (B)	Resistor (R)	Conductor (K)
Spring (k)	Capacitance (C)	Inductance (L)



* Write differential eqn's.

* Free body diagram is:

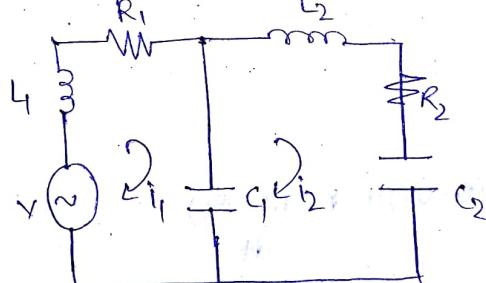


$$* T = T_{j_1} + T_{B_1} + T_{k_1}$$

$$\Rightarrow T = J_1 \cdot \frac{d^2\theta_1}{dt^2} + B_1 \cdot \frac{d\theta_1}{dt} + K_1(\theta_1 - \theta_2)$$

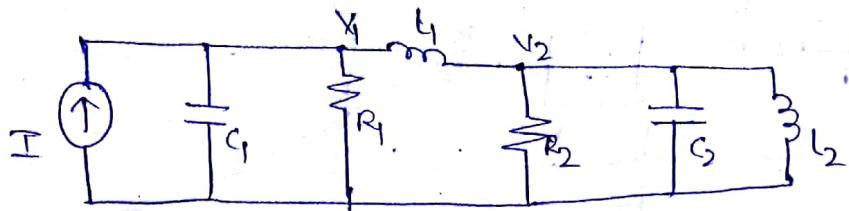
$$* T = T_{k_1} + T_{k_2} + T_{B_2} + T_{j_2}$$

$$\Rightarrow 0 = K_1(\theta_2 - \theta_1) + K_2 \cdot \theta_2 + B_2 \cdot \frac{d\theta_2}{dt} + J_2 \cdot \frac{d^2\theta_2}{dt^2}$$



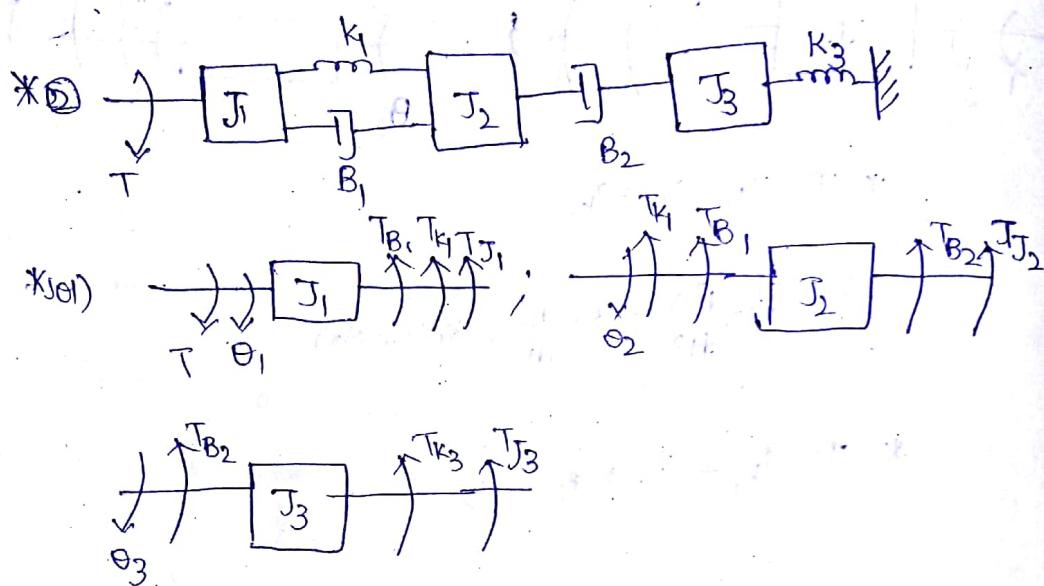
$$V = L_1 \cdot \frac{di_1}{dt} + i_1 R_1 + \frac{1}{C_1} \int (i_1 - i_2) dt$$

$$0 = -L_2 \frac{di_2}{dt} + \frac{1}{C_2} \int i_2 dt + \frac{1}{R_2} \int (i_2 - i_1) dt + i_2 R_2$$



$$I = C_1 \frac{dv_1}{dt} + \frac{V_2}{R_1} + \frac{1}{L_1} \int (v_1 - v_2) dt$$

$$0 = \frac{1}{L_1} \int (v_2 - v_1) dt + \frac{V_2}{R_2} + S_2 \cdot \frac{dv_2}{dt} + \frac{1}{L_2} \int v_2 dt$$



$$T = T_{J_1} + T_{K_1} + T_{B_1}$$

$\oplus J_1$

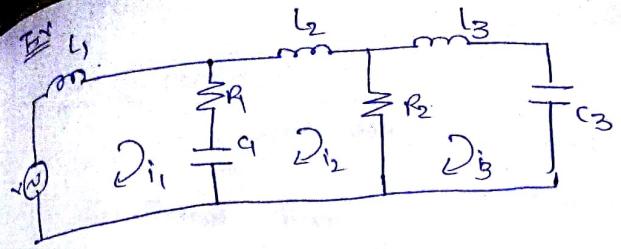
$$\Rightarrow T = J_1 \cdot \frac{d^2\theta_1}{dt^2} + K_1(\theta_2 - \theta_1) + B_1 \cdot \frac{d(\theta_1 - \theta_2)}{dt}$$

$\oplus J_2$

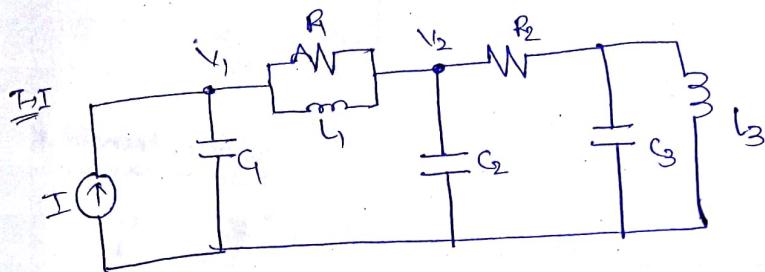
$$\Rightarrow 0 = J_2 \cdot \frac{d^2\theta_2}{dt^2} + K_1(\theta_2 - \theta_1) + B_2 \cdot \frac{d(\theta_2 - \theta_3)}{dt} + K_2(\theta_3 - \theta_2)$$

$\oplus J_3$

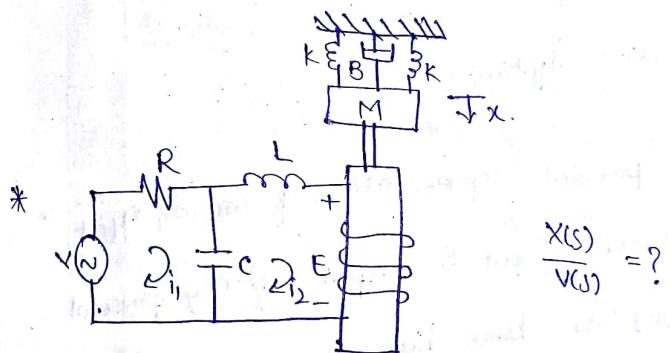
$$0 = J_3 \cdot \frac{d^2\theta_3}{dt^2} + K_3\theta_3 + B_2 \cdot \frac{d(\theta_3 - \theta_2)}{dt}$$



{write Equations for both}
Ckts



T_{B_2} T_{J_2}



* Take off

Some

* Focus

* F

* F

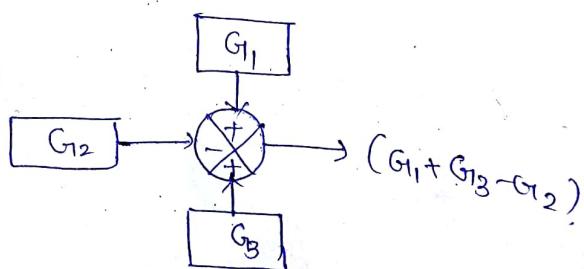
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i

* Block diagram deduction system:

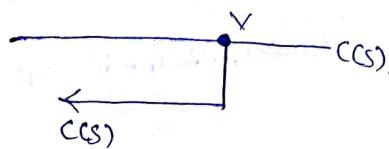
* A short hand pictorial representation of cause & effect relationship between input & output of any physical system is called as block diagram.

* A point at which two or more blocks are added or subtracted is known as summing point.



* Take off point?

* A point at which output signal is sampled & given to some other block.



* Forward path;

* The direction of signals from ip to op is known as F.p.

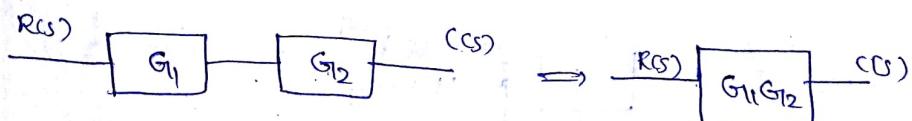
* F/B Path;

* The direction of signal flow from op to ip is known as F/B path.

* BDR Rule;

* 1. Blocks connected in series/cascade

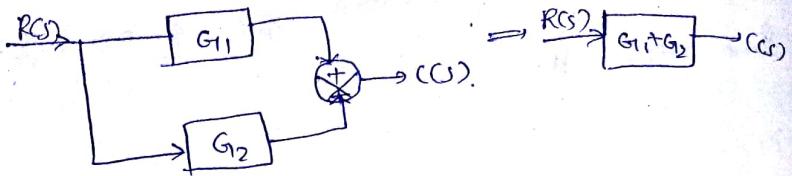
then the overall Transfer function is the product of individual TF's.



* If a no. of blocks are in cascade/series, it can be replaced by its product.

Blocks Connected in parallel :-

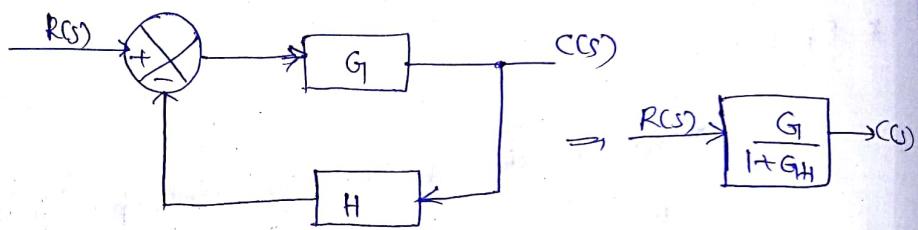
*5 moving



* If a no. of blocks are in parallel, it can be replaced by its summation.

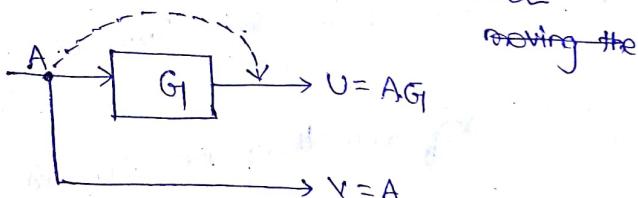
*3. Elimination of Single loop negative F/B :-

*6. ~~one~~

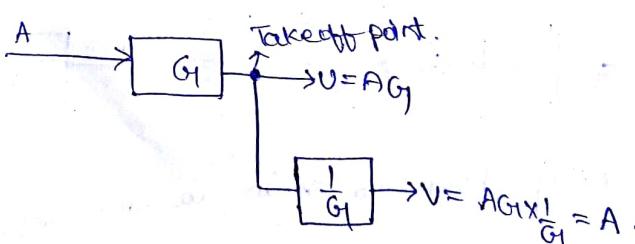


*4. moving the take off point after the block :-

*7.

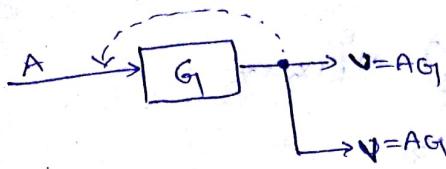


*After moving the taking point after the block it is given by

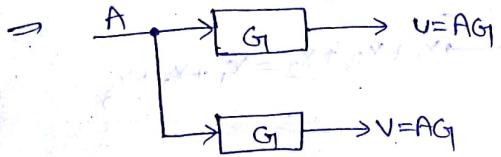


*5 moving the take-off point before the block:

$$1+G_2 \rightarrow C(s)$$

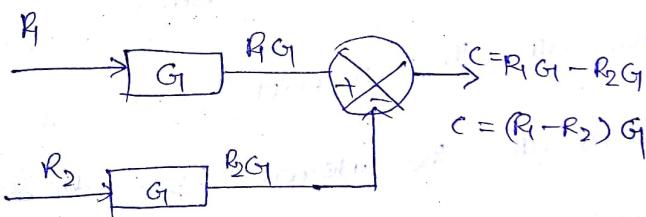
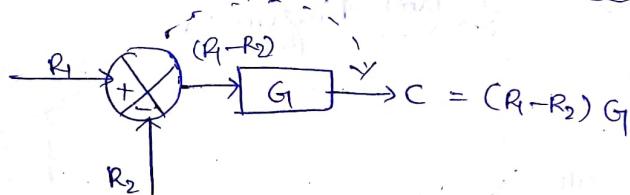


replaced by it

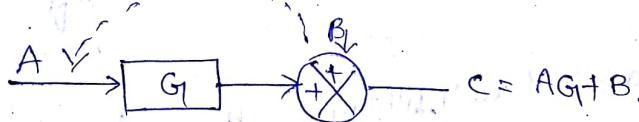


*6. moving the summing point after the block:

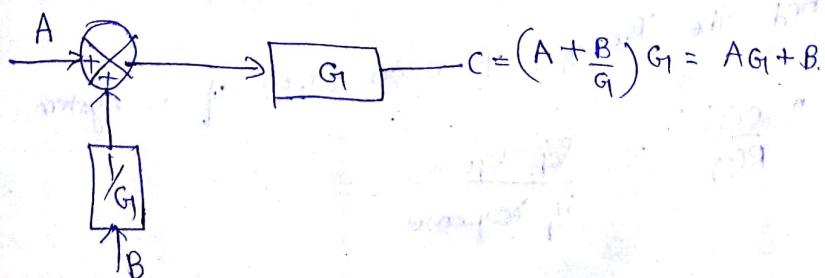
$$\frac{G}{1+GH} \rightarrow C(s)$$



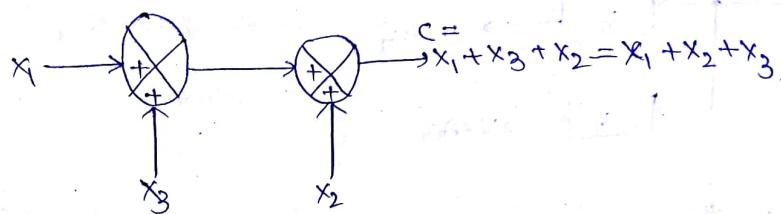
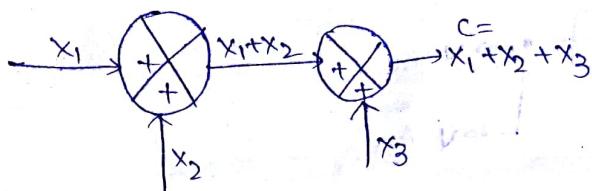
*7. moving the summing point before the block:



is given by



* 8. Interchanging of summing points ↗



* Procedure for reducing block diagram ↗

* Steps ↗

* 1. Reduce all the cascaded blocks

* 2. Reduce all the parallel blocks

* 3. Reduce all the internal F/B feed back loops.

* 4. always try to shift take off point towards right & summing points towards left.

* 5. Repeat 1 to 4, till simple block diagram is obtained

* 6. Find the Transfer function (T.F) of the system

$$\frac{C(s)}{R(s)} = \frac{\text{O/p response}}{\text{I/p response}}$$



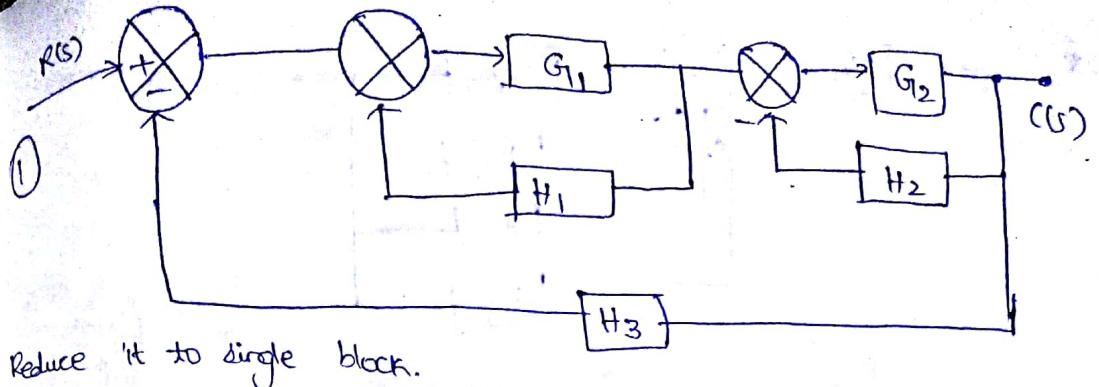
Reduce it

* (a)



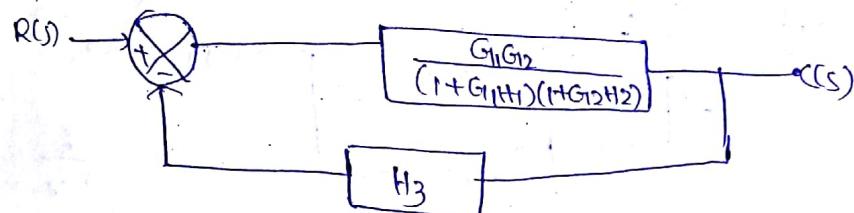
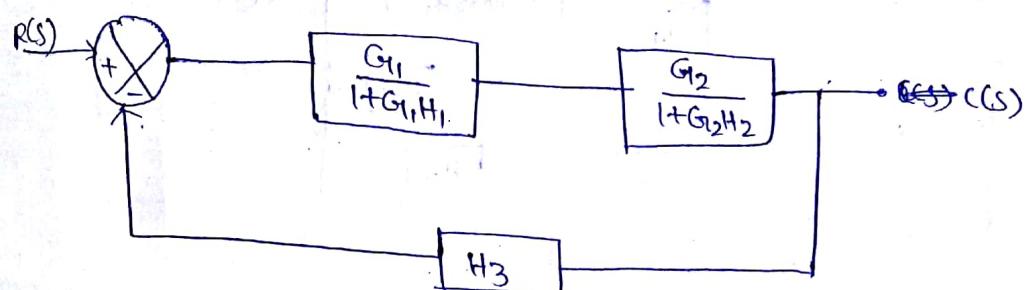
R(s) —

(2)



Reduce it to single block.

* $s(s)$

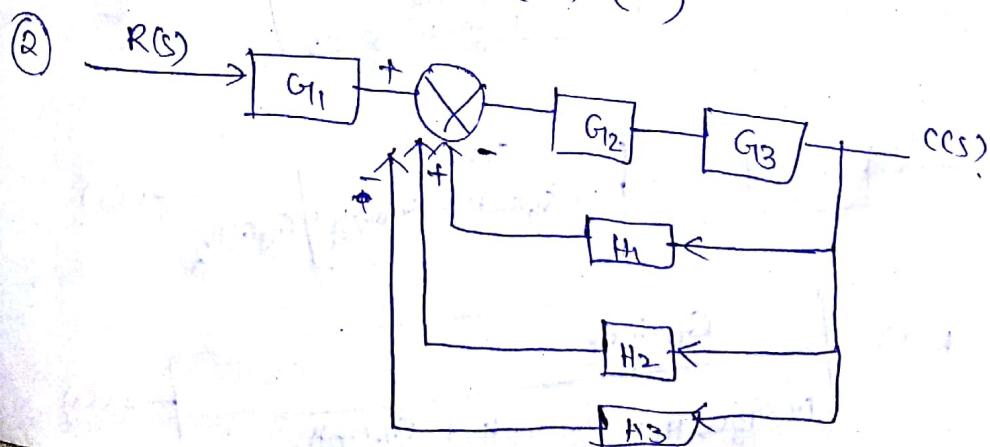


$$\frac{G_1 G_2}{(1+G_1 H_1)(1+G_2 H_2)} \cdot \frac{1 + \frac{G_1 G_2}{(1+G_1 H_1)(1+G_2 H_2)} \cdot H_3}{C(s)}$$

loops.

towards

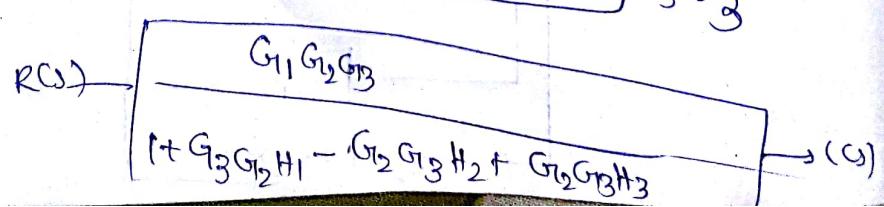
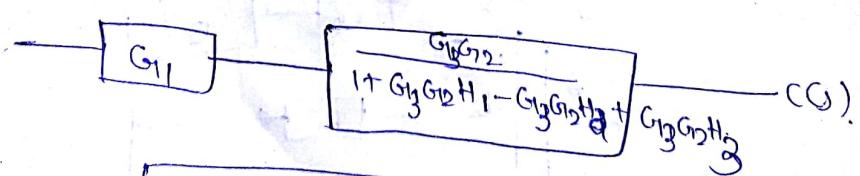
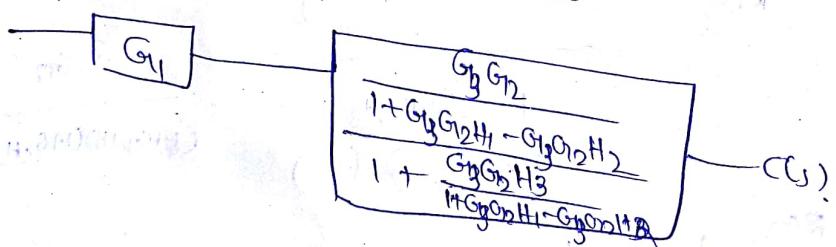
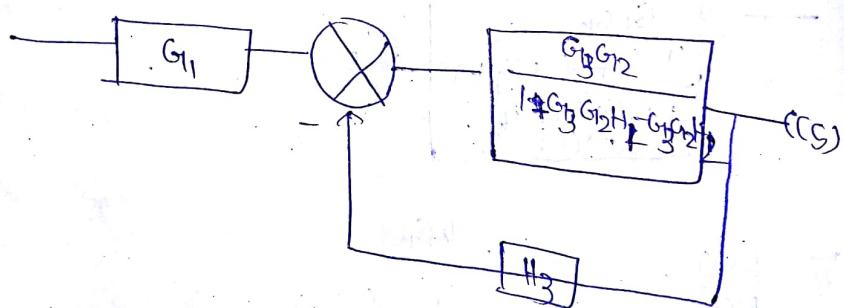
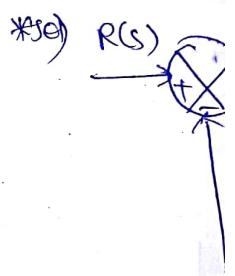
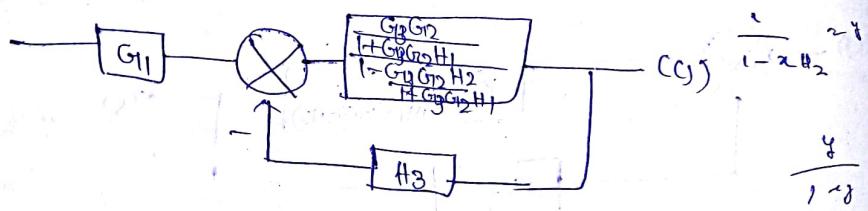
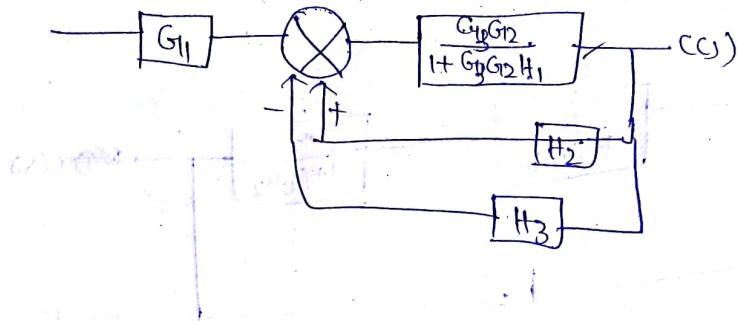
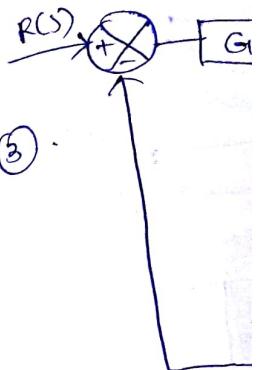
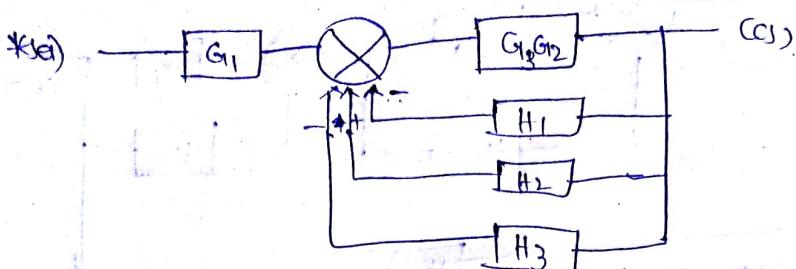
$$\Rightarrow \frac{C(s)}{R(s)} = \frac{\frac{G_1 G_2}{(1+G_1 H_1)(1+G_2 H_2)}}{1 + \frac{G_1 G_2 H_3}{(1+G_1 H_1)(1+G_2 H_2)}} = \frac{G_1 G_2}{(1+G_1 H_1)(1+G_2 H_2) + G_1 G_2 H_3}$$



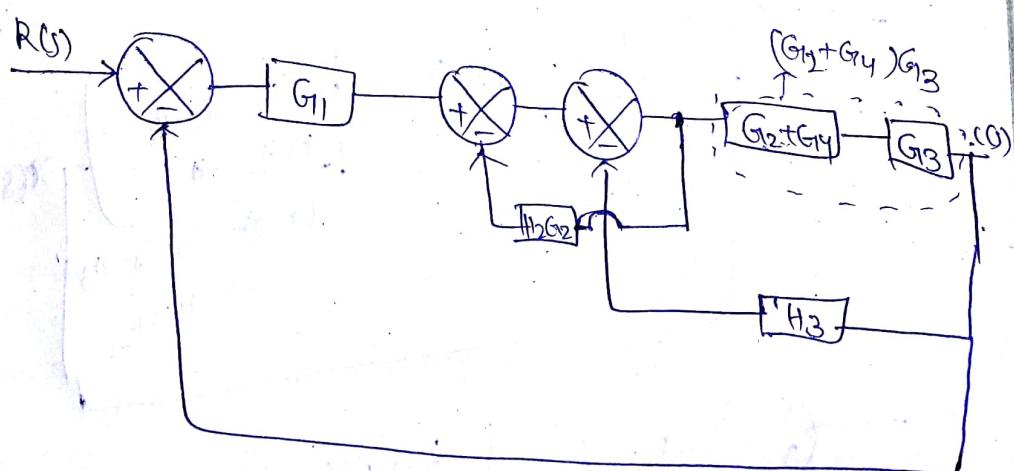
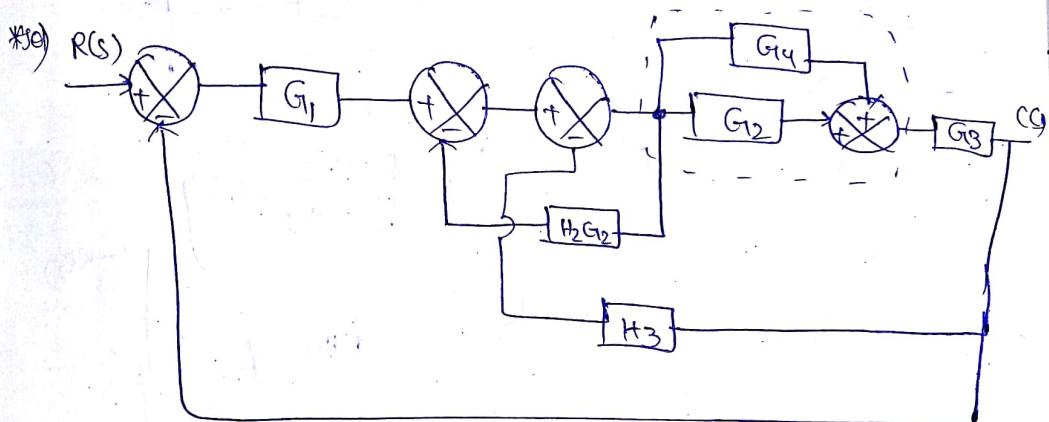
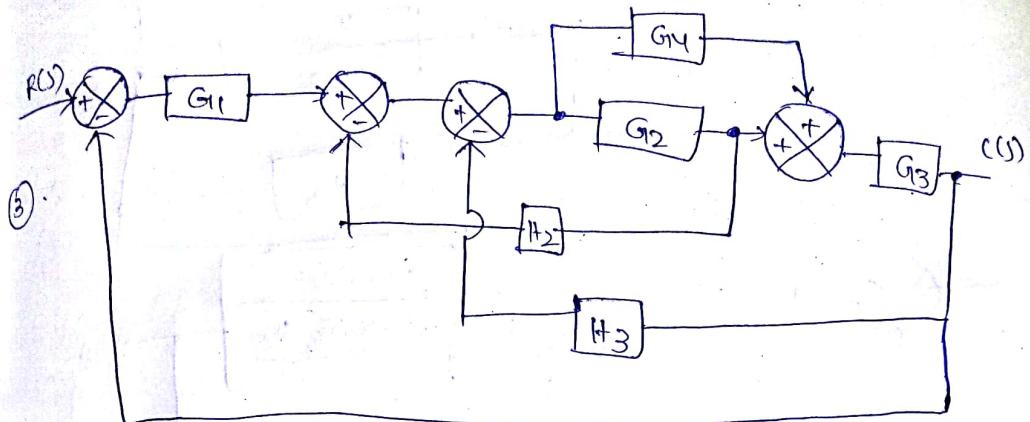
can be obtained

the system

$$\Rightarrow \frac{C(s)}{R(s)} = 1$$

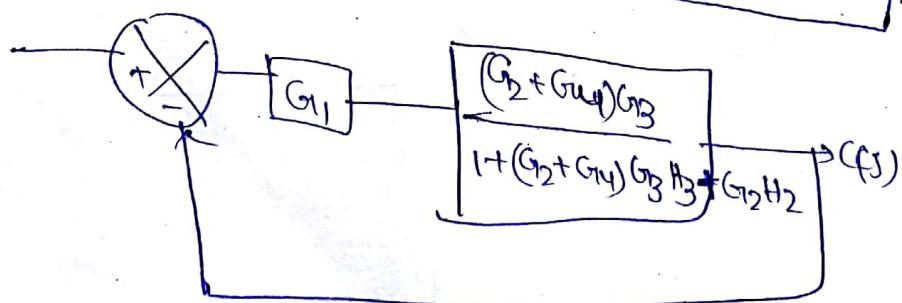
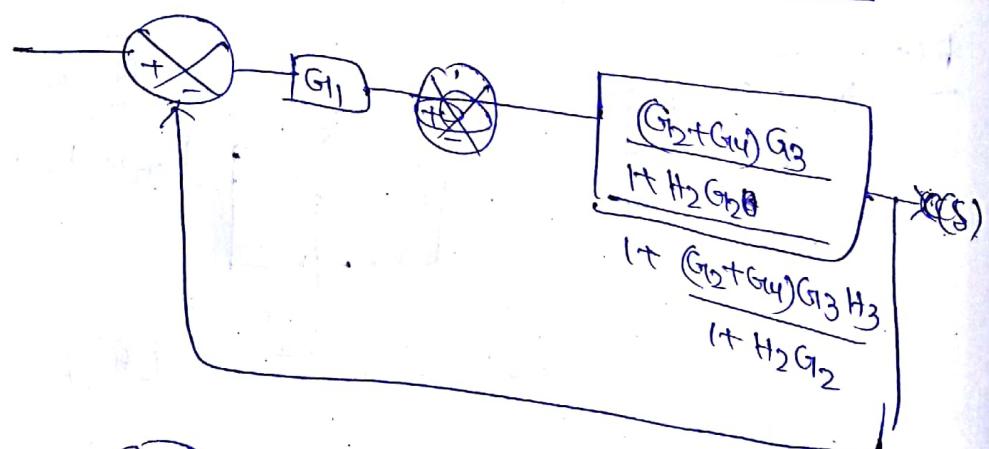
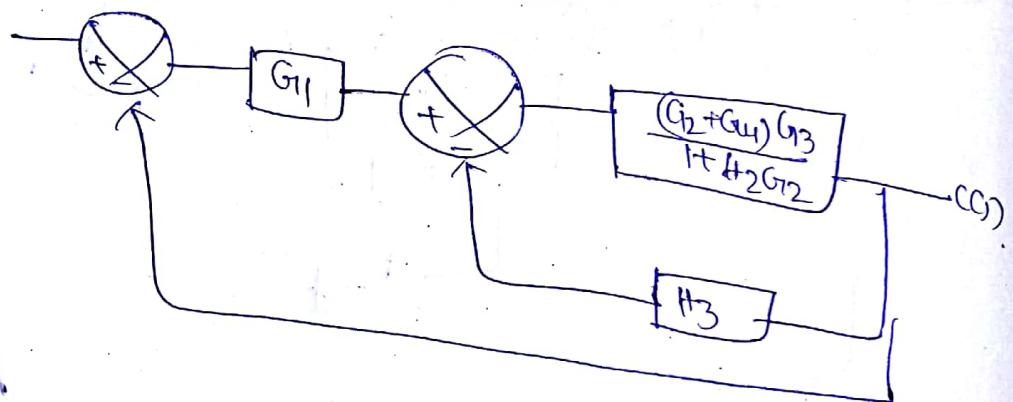
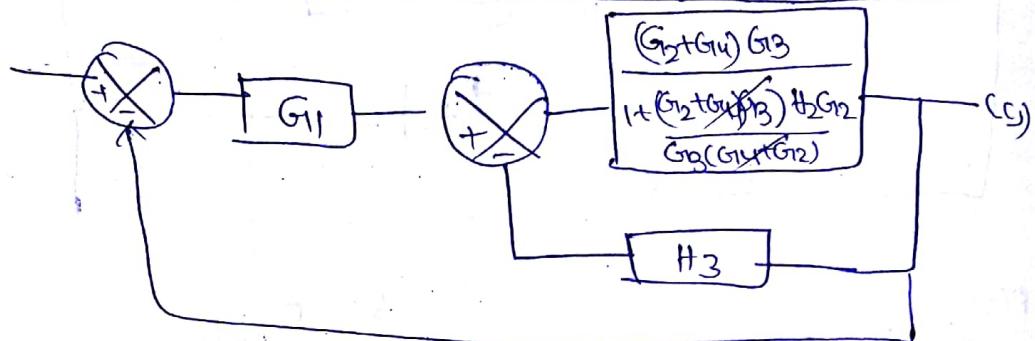
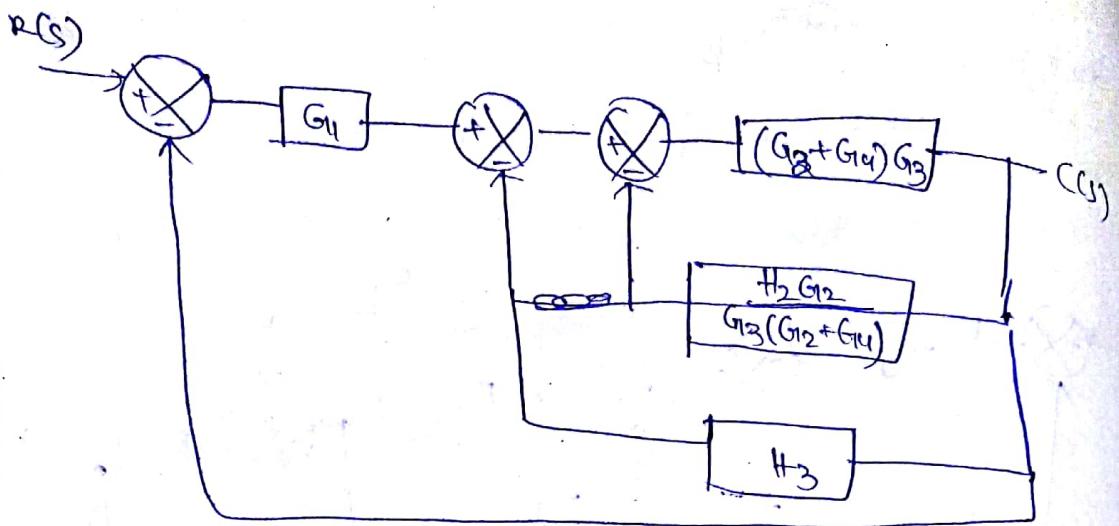


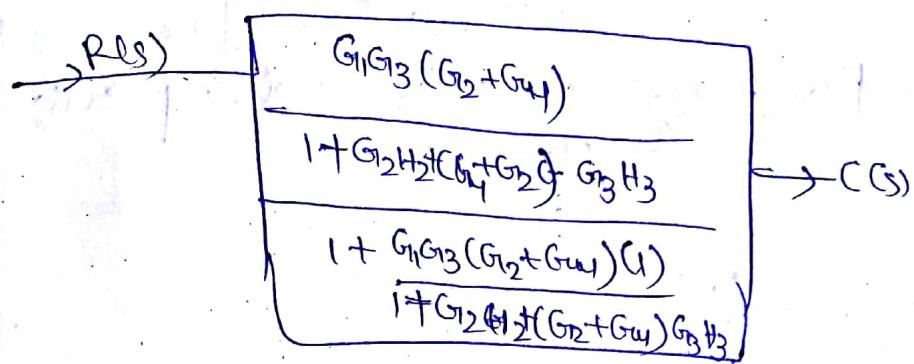
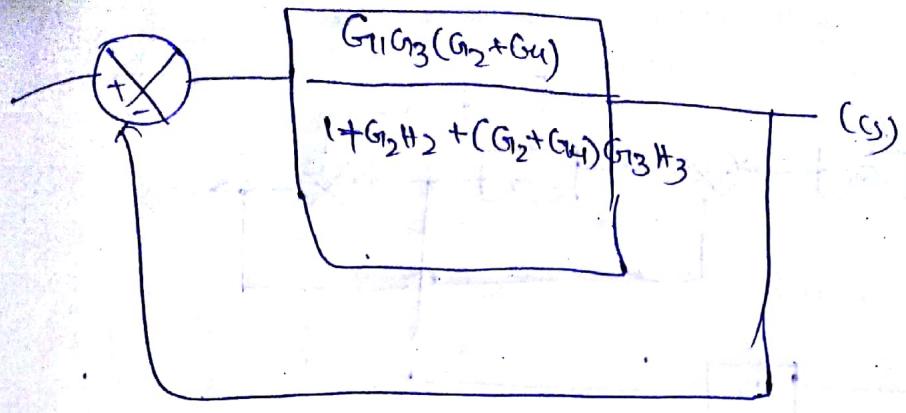
$$\Rightarrow \frac{C(s)}{R(s)} = \frac{G_1 G_2 G_3}{1 + G_1 G_3 H_1 - G_1 G_3 H_2 + G_1 G_3 H_3}$$



6).

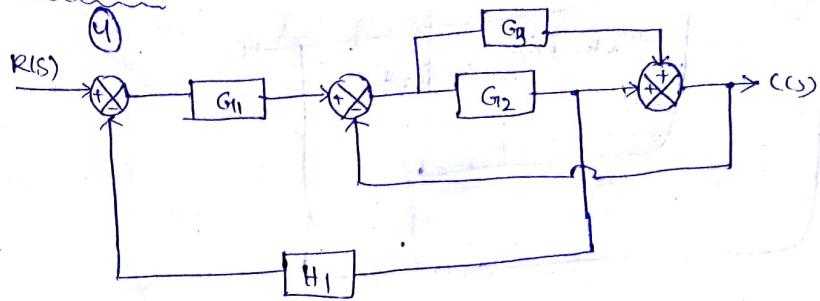
(c)



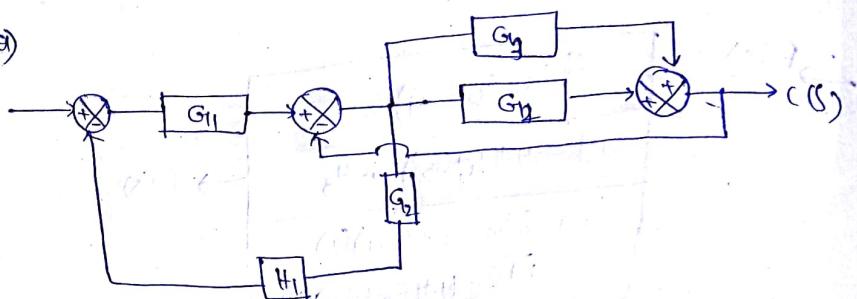


$$\frac{C(s)}{R(s)} = \frac{G_1 G_3 (G_2 + G_4)}{1 + G_2 H_2 + G_2 G_3 H_3 + G_4 G_3 H_3 + G_1 G_3 G_2 + G_1 G_3 G_4}$$

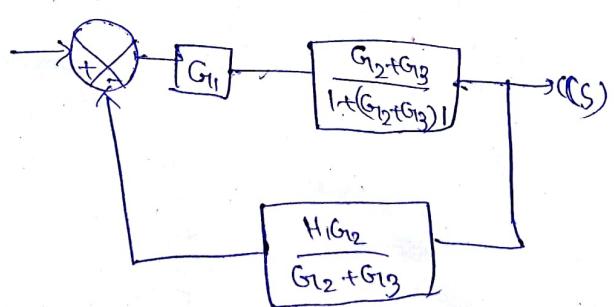
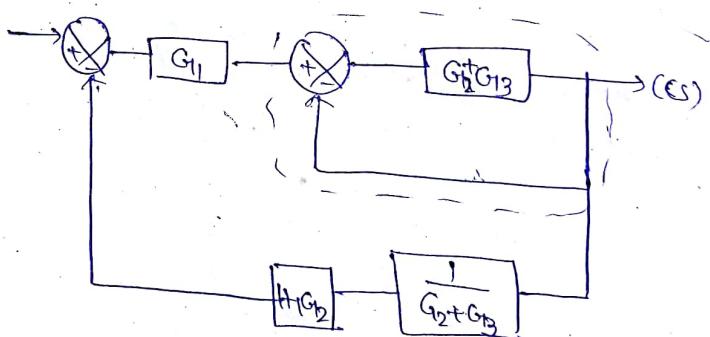
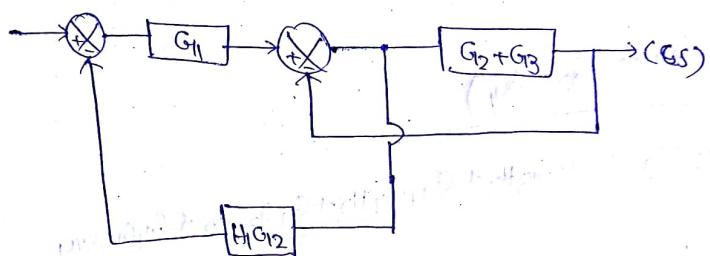
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(5a)

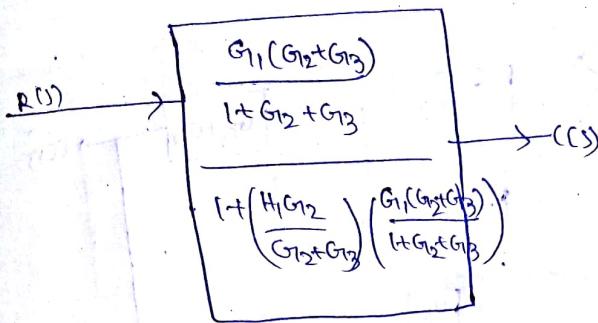
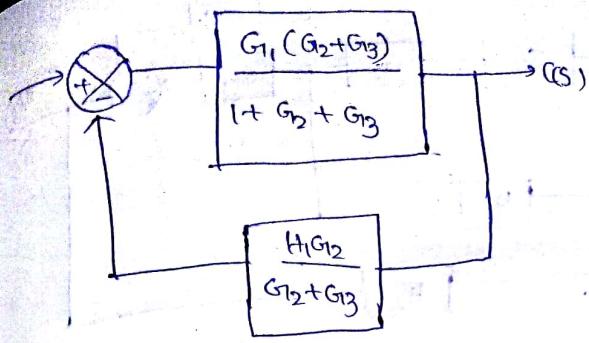


R(s)



$\frac{CCS}{RCS}$

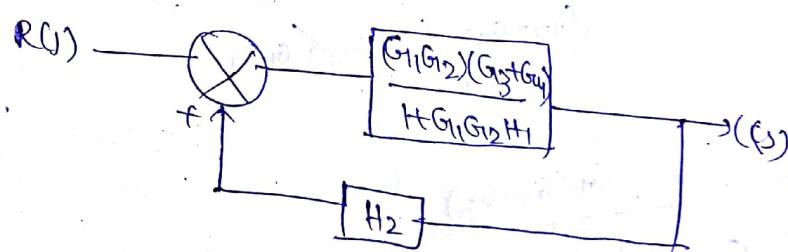
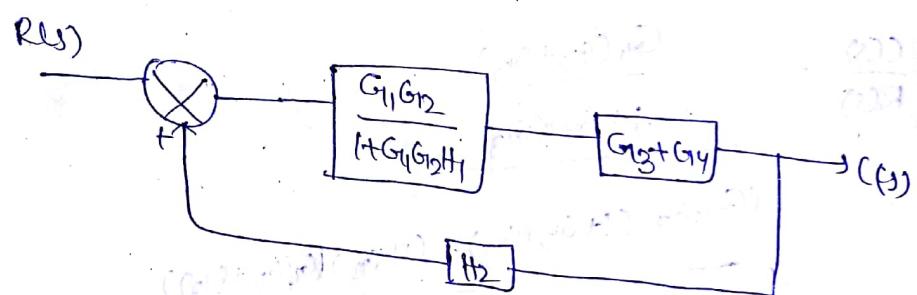
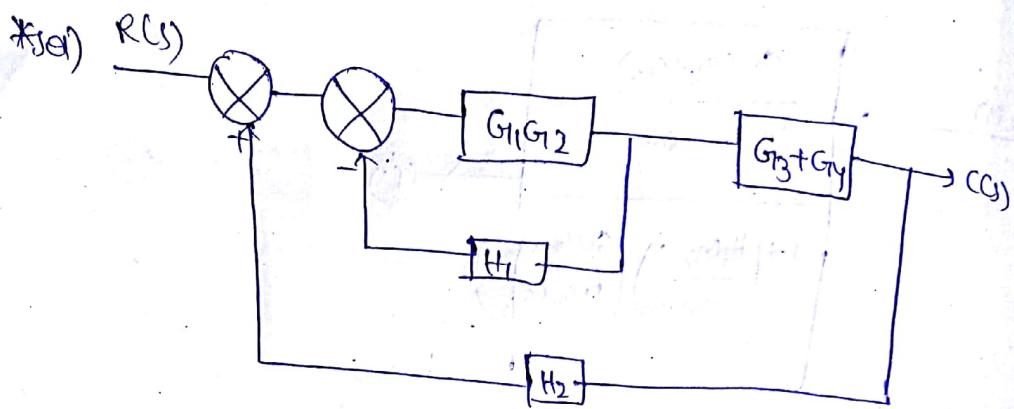
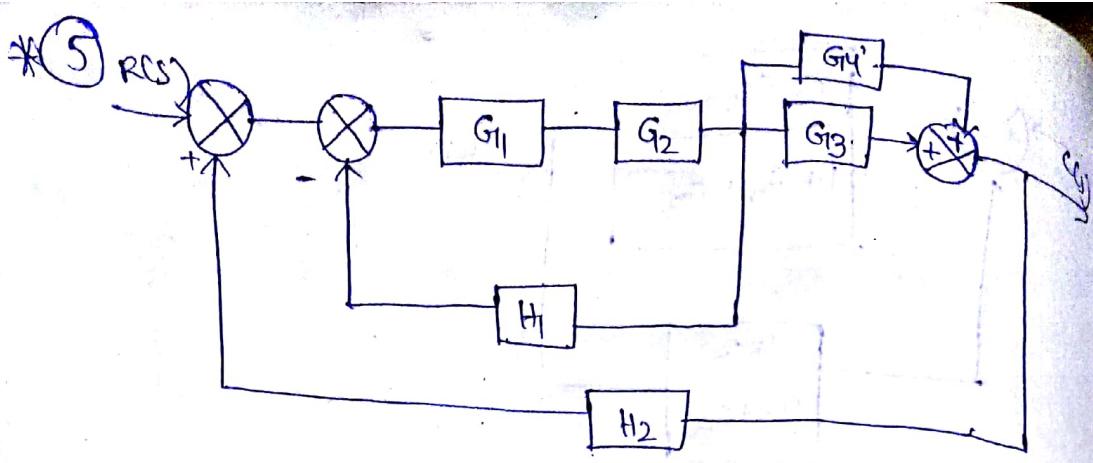
C
R



$$\frac{CCS}{RCS} = \frac{\frac{G_{11}(G_{12}+G_{13})}{1+G_{12}+G_{13}}}{\frac{(G_{12}+G_{13})(1+G_{12}+G_{13}) + (H_1G_{12})(G_{11}(G_{2}+G_{3}))}{(G_{12}+G_{13})(1+G_{12}+G_{13})}}$$

$$= \frac{G_{11}(G_{12}+G_{13})}{G_{12}+G_{13} \left[(1+G_{12}+G_{13}) + H_1G_{12}G_{11} \right]}$$

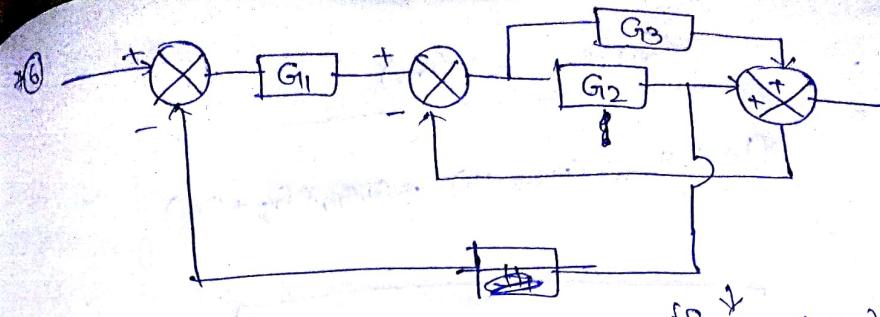
$$\frac{CCS}{RCS} \Rightarrow \frac{G_{11}(G_{12}+G_{13})}{1+G_{12}+G_{13} + G_{11}G_{12}H_1}$$



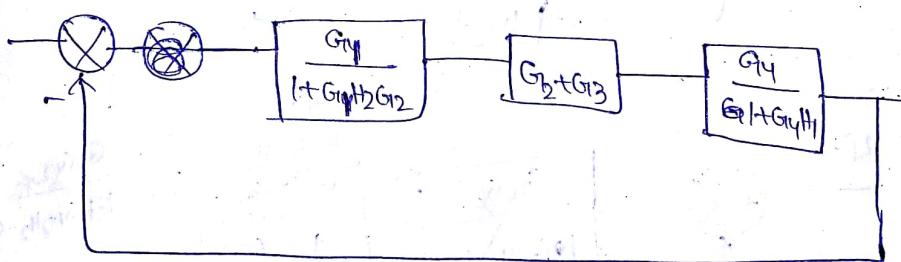
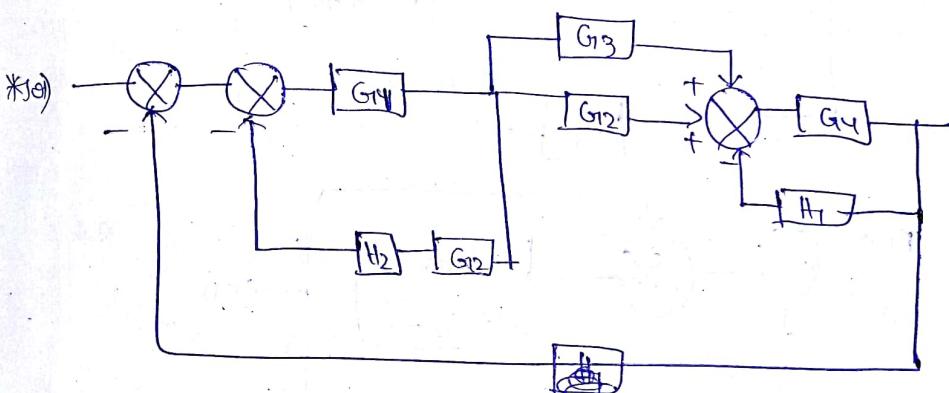
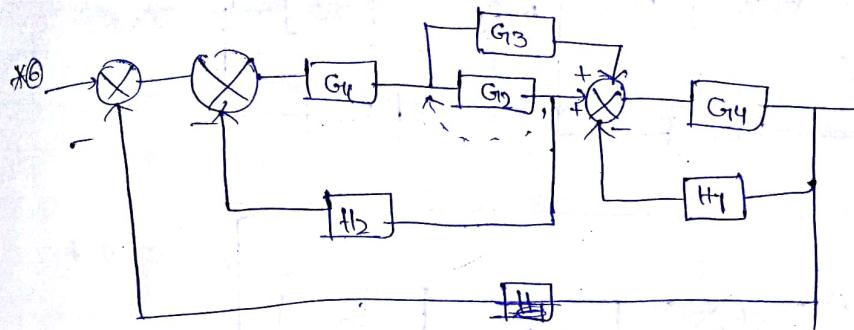
$$\frac{C(s)}{R(s)} = \frac{G_1 G_2 (G_3 + G_4)}{1 + G_1 G_2 H_1}$$

$$= \frac{(G_1 G_2) (G_3 + G_4)}{1 + G_1 G_2 H_1} \cdot H_2$$

$$\frac{C(s)}{R(s)} = \frac{G_1 G_2 (G_3 + G_4)}{1 + G_1 G_2 H_1 + G_1 G_2 (G_3 + G_4) H_2}$$

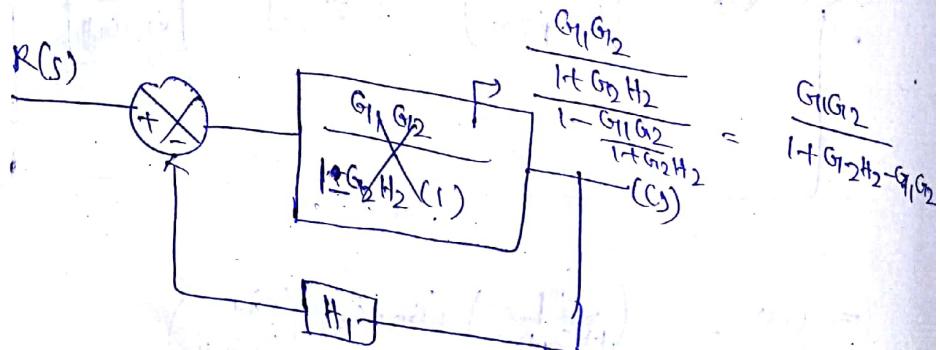
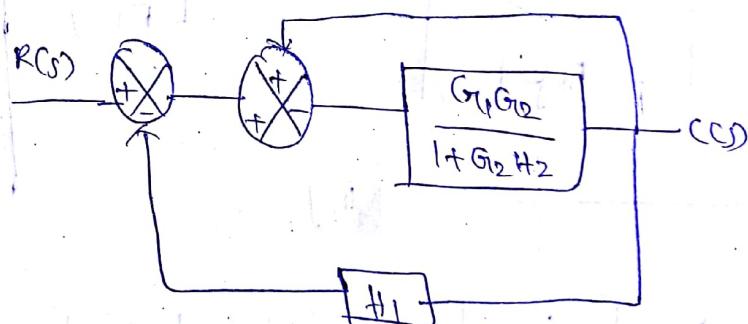
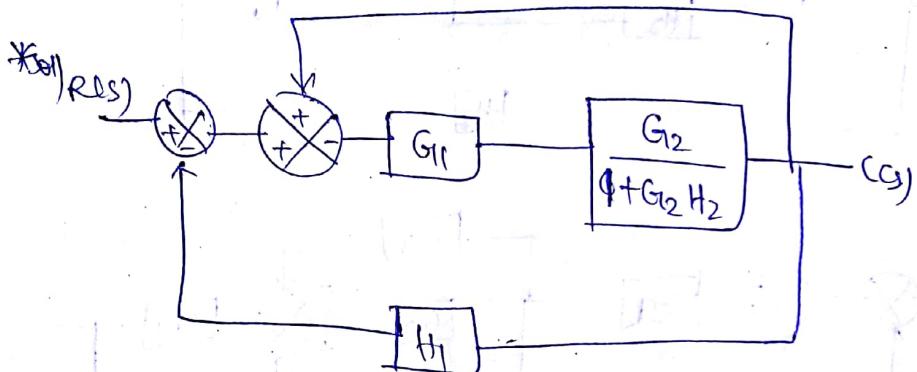
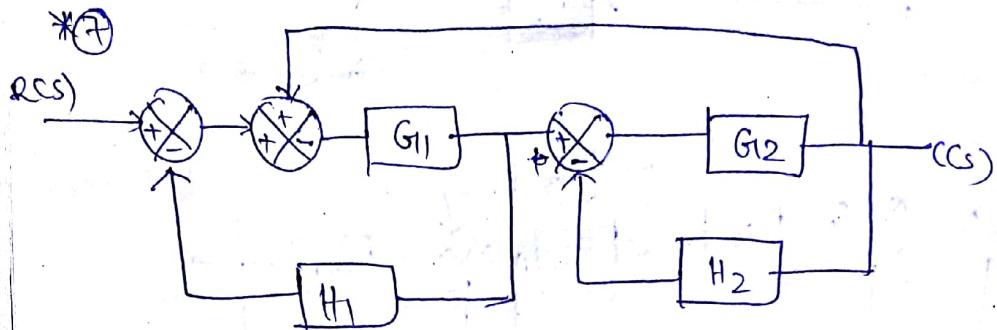


{Same prob 4}



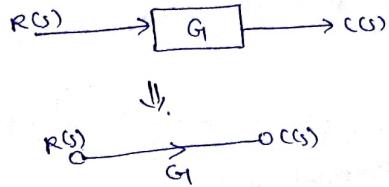
$$\frac{C(s)}{R(s)} = \frac{\left(\frac{G_4}{1+G_4H_2G_2} \right) (G_2 + G_3) \cdot \left(\frac{G_4}{1+G_4H_1} \right)}{1 + \frac{(G_1)(G_2 + G_3)(G_4)}{(1+G_1H_2G_2)(1+G_4H_1)}} \quad (1)$$

$$\frac{C(s)}{R(s)} = \frac{(G_1)(G_2 + G_3)(G_4)}{(1 + G_1 H_2 G_2)(1 + G_4 H_1) + G_1 G_4 (G_2 + G_3)}$$



$$\frac{C(s)}{R(s)} = \frac{\frac{G_1 G_2}{1 + G_2 H_2 - G_1 G_2}}{\frac{1 + G_1 G_2 H_2}{1 + G_2 H_2 - G_1 G_2}} = \frac{G_1 G_2}{1 + G_2 H_2 + G_1 G_2 H_1 - G_1 G_2}$$

* Signal flow graph \rightarrow



* It is nothing but the graphical representation of relationship between the variables of a set of linear algebraic eq written in the form of cause & effect relationship.

* It consists of a network in which nodes represent each of the system variables are connected by directed branches, the directed branches have associated branch gains.

* A signal can be transmitted through a branch only in the direction of arrow.

* Nodes \rightarrow

* A node (v) Junction point is a point represents system variable (v) signal.

* Branch \rightarrow

* A Branch is a direct line segment joining two nodes, the arrow on the branch indicates the direction of signal flow & the gain of the branch is transmittance.

* Transmittance / gain \rightarrow

* The gain acquired by the signal when

G_1/G_2

$$\frac{G_1/G_2}{1+G_1H_2-G_1G_2}$$

It flows from one node to another is called Branchance or gain.

* Self loop :-
* A !

* It can be either real or complex value.

* Signal
values

* Path :-

* A path is a traversal of connected branches in the direction of the branch arrows.

* using rules

* The path should not cross a node more than once.

Equation ad

* Forward path :- F_p

* It is a path from an input node to ~~an~~ output node that doesn't cross any node more than once.

* To find
a formula
evid. ti

* F.P gain :-

* The product of the branch gains describe in traversing a F.P is called F.P gain.

* $P_i = F$
 $P_i = I$

* Loop :-

* A loop is a path which originates & terminates at the same node.

* $\Delta =$

* Non-touching loops :-

* Loops which doesn't have common node then the loops are non-touching loops.

$\Delta_i =$

mittance of gain

* Self loop

* A loop consisting of single branch is called self loop.

* Signal flow graph {S.F.G} Reductions

* S.F.G of a system can be reduced fitting using rules of signal flow graph algebra i.e., by writing equation at every node & then rearranging these eq's to get the ratio of o/p to i/p.

* To find the TF of a system using SFG, SJ major has developed a formula called by this name Mason's gain formula which is used to find T.F as given below.

$$T.F = \frac{o/p}{i/p} = \frac{1}{\Delta} \sum_{i=1}^n P_i A_i$$

* P_i = F.P gain of i th F.P.

P_1 = 1st F.P gain ; P_2 = 2nd F.P gain ;

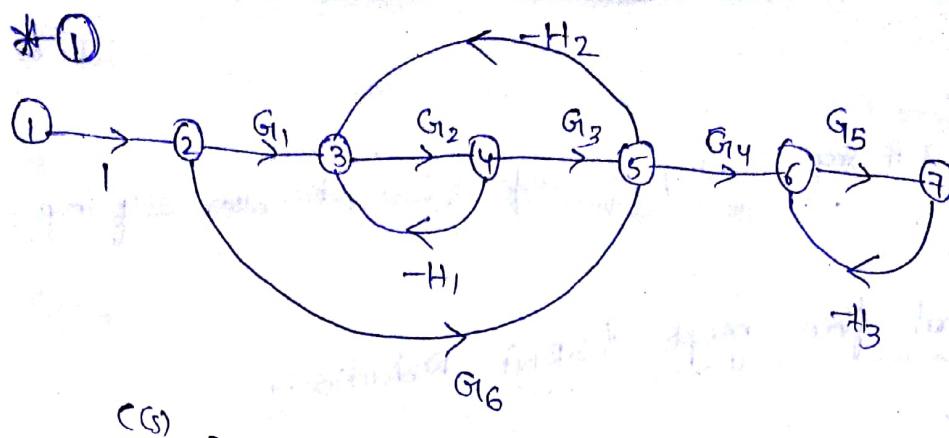
* $\Delta = 1 - [$ sum of gain product of individual loop gains $]$

+ [sum " " " Two non-touching " "]

- [" " " Three " " " "]

+ [" " " Four " " " "]

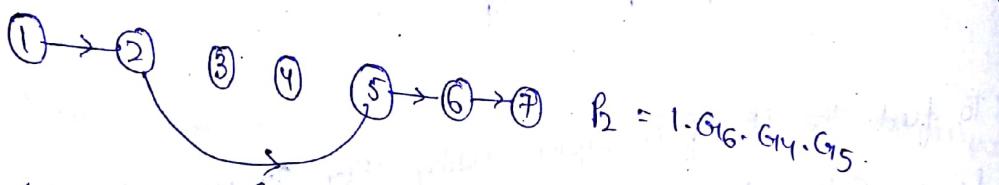
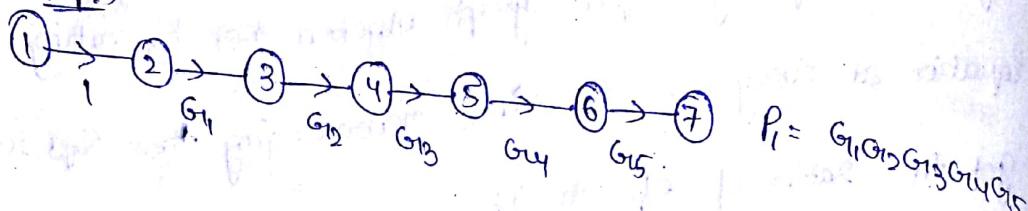
$A_i = \Delta$ for that part of the graph which is not touching i th F.P.



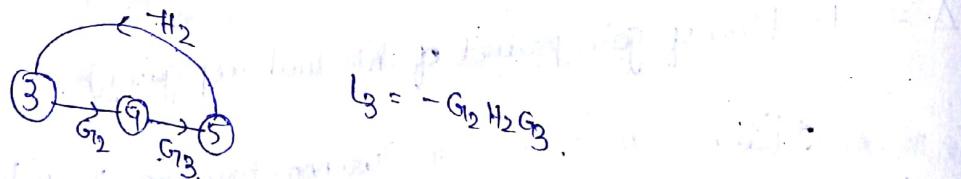
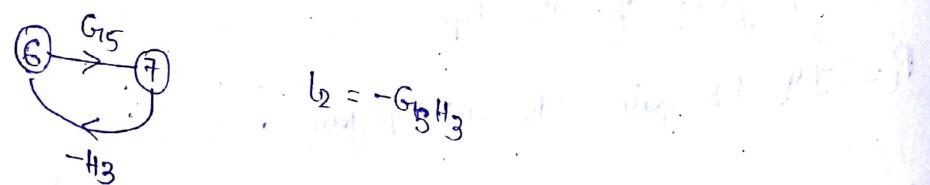
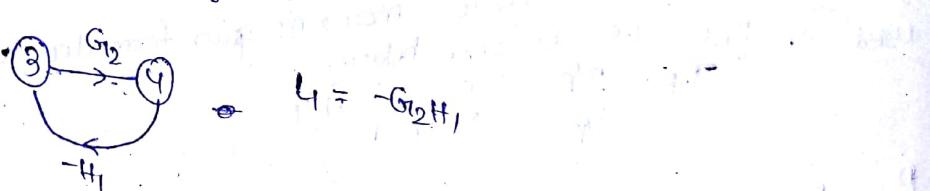
$$\frac{C(S)}{R(S)} = ?$$

* ②

* Ep. 1



* loops



* Non-touching loops

$L_1 \& L_3$ are non touching if no nodes in common
 $L_2 \& L_3$ " " "

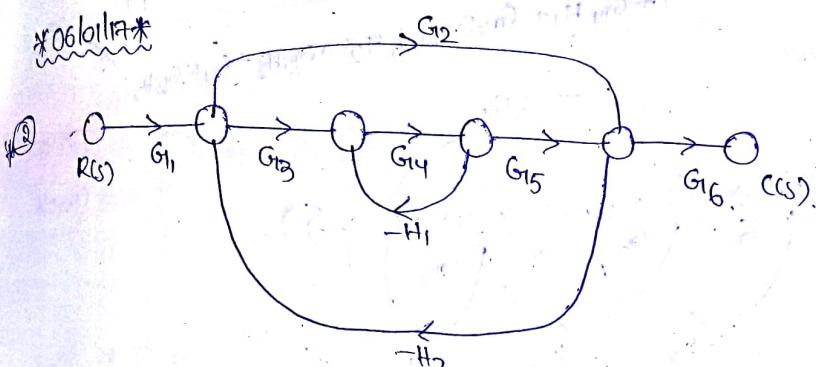
$$\Delta = 1 - [L_1 + L_2 + L_3] + [L_1 L_2 + L_2 L_3]$$

$$A_1 = 1$$

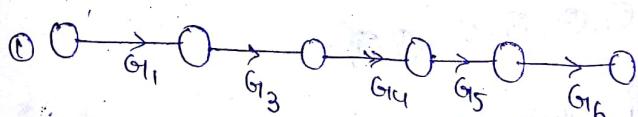
$$A_2 = 1 - [L_1]$$

$$\Rightarrow T.F = \frac{(G_1 G_2 G_3 G_4 G_5)(1) + (G_4 G_5 G_6)(1 + G_{12} H_1)}{1 + (G_2 H_1 + G_5 H_3 + G_2 H_2 G_3)}$$

$$1 + (G_2 H_1 + G_5 H_3 + G_2 H_2 G_3) = (G_5 H_3 G_2 H_1 + G_2 H_2 G_3 G_5 H_3)$$

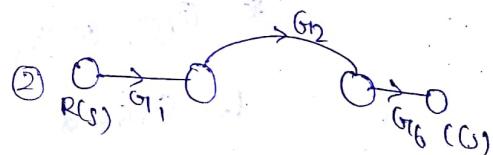


*Sel. *Fil.

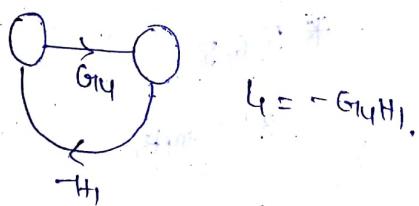


$$P_1 = G_1 G_2 G_3 G_4 G_5 G_6$$

$$P_2 = G_1 G_2 G_6$$

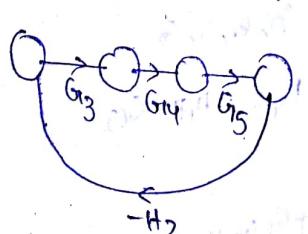


loop

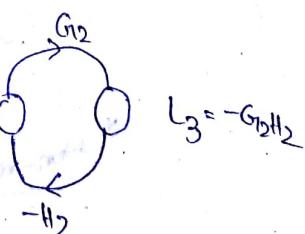


$$L_1 = -G_{14} H_1$$

$$T.F = \frac{P_1 A_1 + P_2 A_2}{\Delta}$$



$$L_2 = -G_3 G_4 G_5 H_2$$



$$L_3 = -G_{12} H_2$$

* Non-Touching loop = $L_1 L_3$

(common)

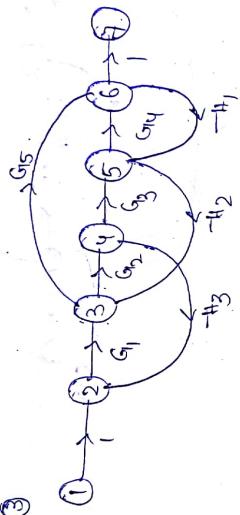
$$\Delta = 1 - \frac{[l_1 + l_2]}{l_1 l_2} = 1 + G_{14}H_1 + G_8G_{15}G_{12} + G_{24}H_{24} + G_{12}H_{24}G_{15}H_1$$

$$l_1 = 1 - 0 = 1$$

$$l_2 = 1 - l_1 = 1 + G_{14}H_1$$

$$\Rightarrow T.F = \frac{(G_1G_3G_5G_7G_6)(1) + (G_1G_5G_6)(1 + G_{14}H_1)}{1 + G_{14}H_1 + G_8G_{15}G_{12} + G_{24}H_{24} + G_{12}H_{24}G_{15}H_1}$$

*③



*⑧) *

* ①, ②, ③, ④, ⑤, ⑥, ⑦

$$P_1 = G_1G_2G_3G_4$$

* ①, ②, ④, ⑥, ⑦.

$$P_2 = G_1G_5$$

loop

* 2, 4, 2

$$L_1 = -G_1G_2H_3$$

$$L_3 = -G_{14}H_1$$

* 3, 4, 5, 8

$$L_2 = -G_2G_3H_2$$

$$L_4 = H_1H_2G_{15}$$

$$L_5 = +G_1G_2H_3G_4H_4$$

* Non-touching loops

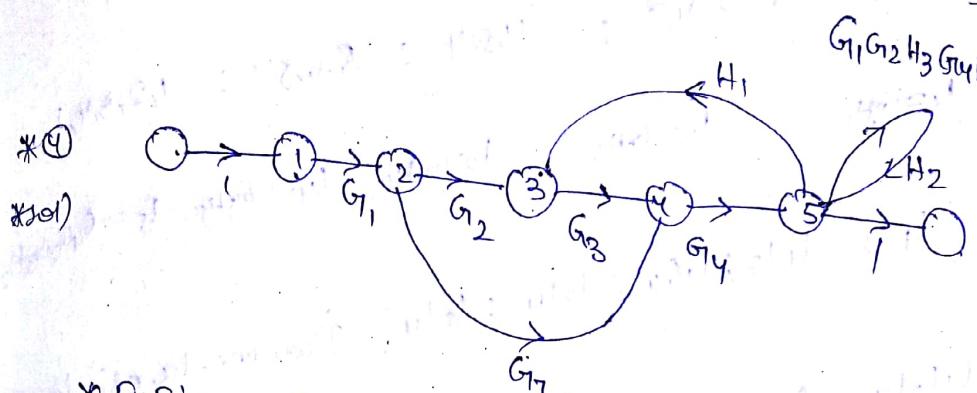
$$L_3 = +G_1G_2H_3G_4H_4$$

$$* \Delta = 1 - [l_1 + l_2 + l_3] + l_4 l_3$$

$$\Delta_1 = 1$$

$$\Delta_2 = 1$$

$$T.F = \frac{P_1 \Delta_1 + P_2 \Delta_2}{\Delta} = \frac{G_1 G_2 G_3 G_4 + G_1 G_5}{1 + [G_1 G_2 H_3 + G_4 H_1 + G_2 G_3 H_2 + H_1 H_2 G_5] + G_1 G_2 H_3 G_4 H_1}$$



* F-P-H

$$P_1 = G_1 G_2 G_3 G_4$$

$$P_2 = G_1 G_7 G_4$$

* loops,

$$l_1 = G_3 G_4 H_1, l_2 = H_2$$

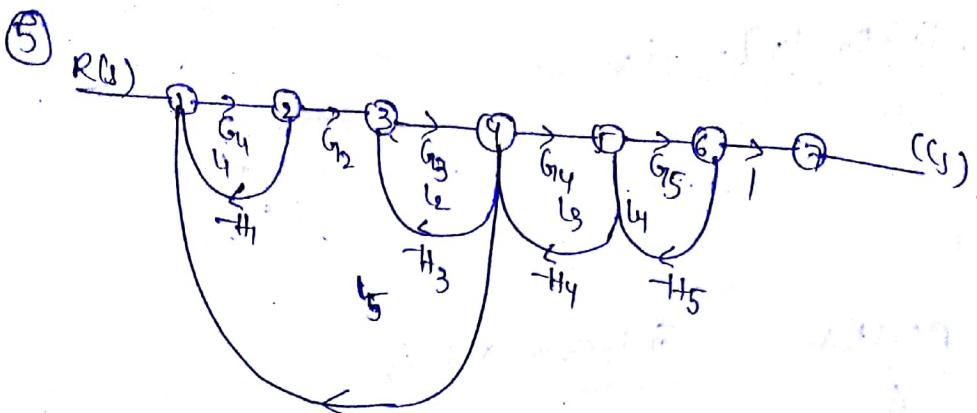
* Non-touching H

$$\Delta_1 = 1 - 0$$

$$\Delta_2 = 1 - 0$$

$$\Delta = 1 - [l_1 + l_2]$$

$$\Rightarrow T.F = \frac{G_1 G_2 G_3 G_4 + G_1 G_7 G_4}{1 - (G_3 G_4 H_1 + H_2)}$$



* (S)

$$l_1, l_2, l_1 ; l_3, l_4, l_5 ; l_6, l_7, l_8 ; l_9, l_{10}, l_{11}.$$

$$l_1 = G_1 H_1 \quad l_2 = G_2 H_2 \quad l_3 = G_3 H_3 \quad l_4 = G_4 H_4 \quad l_5 = G_5 H_5 \quad l_6 = G_6 H_6 \quad l_7 = G_7 H_7$$

* $l_1 l_2 ; l_1 l_3 ; l_1 l_4 ; l_2 l_3 ; l_2 l_4 ; l_3 l_4 ; l_4 l_5 ; l_5 l_6 ; l_6 l_7 ; l_7 l_8 ; l_8 l_9 ; l_9 l_{10} ; l_{10} l_{11}$. of P_{100} non-touching loops.

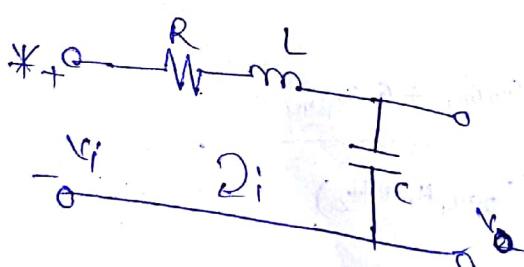
* $l_1 l_2 l_4$ {Three non-touching loops}.

$$\Delta = 1 - [l_1 + l_2 + l_3 + l_4 + l_5] + [l_1 l_2 + l_1 l_3 + l_1 l_4 + l_2 l_3 + l_2 l_4 + l_3 l_4]$$

$$A_1 = 1 - [l_1 l_2 l_4]$$

$$P_1 = G_1 G_2 G_3 G_4 G_5$$

$$T.F = \frac{P_1 A_1}{\Delta} \text{ Substitute } \Delta$$



* (S)

$$V_i = R_i + L \frac{di}{dt} + \frac{1}{C} \int idt$$

$$V_o = \frac{1}{C} \int idt$$

GT on B.T for 2 eqs.

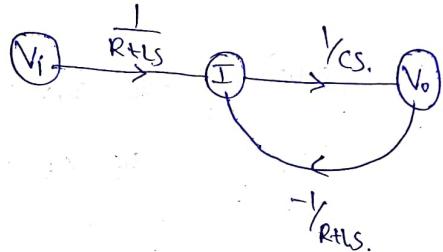
$$\Rightarrow V_i(s) = R I(s) + L I(s)s + \frac{1}{sC} I(s), \quad V_o(s) = \frac{1}{sC} I(s)$$

$$\Rightarrow V_i(s) = I(s) [R + sL + \frac{1}{sC}] ; \quad \boxed{V_o(s) = I(s) \left[\frac{1}{sC} \right]} \quad \begin{matrix} \downarrow ① \\ \downarrow ② \end{matrix}$$

* from ①, ③ we get

$$V(s) = I(s) [R + sL] + V_o(s).$$

$$\boxed{\frac{V_i(s) - V_o(s)}{R + sL} = I(s)}$$



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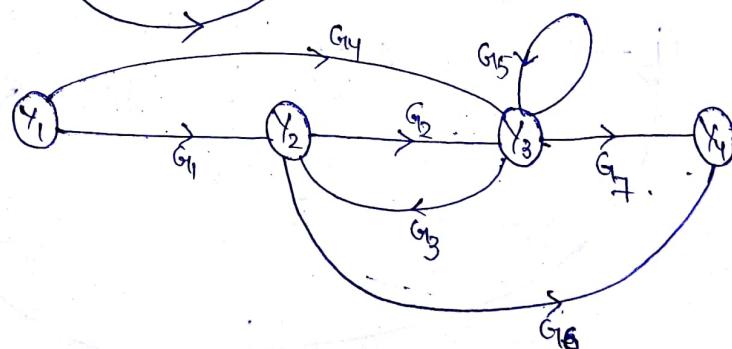
$$Y_2 = G_1 Y_1 + G_3 Y_3$$

$$Y_3 = G_4 Y_1 + G_2 Y_2 + G_5 Y_5 \quad \frac{Y_1}{Y_1} = ?$$

$$Y_4 = G_6 Y_2 + G_7 Y_3$$



Ans



* FP:

$$P_1 = G_1 G_2 G_{17} ; P_2 = G_1 G_{17} ; P_3 = G_{14} G_6 ; P_4 = G_{14} G_3 G_6$$

* loops:

$$L = G_2 G_3 ; L_2 = G_5$$

$$\Delta = 1 - [G_2 G_3 + G_5] - 0$$

$$\Delta_1 = 1 - 0$$

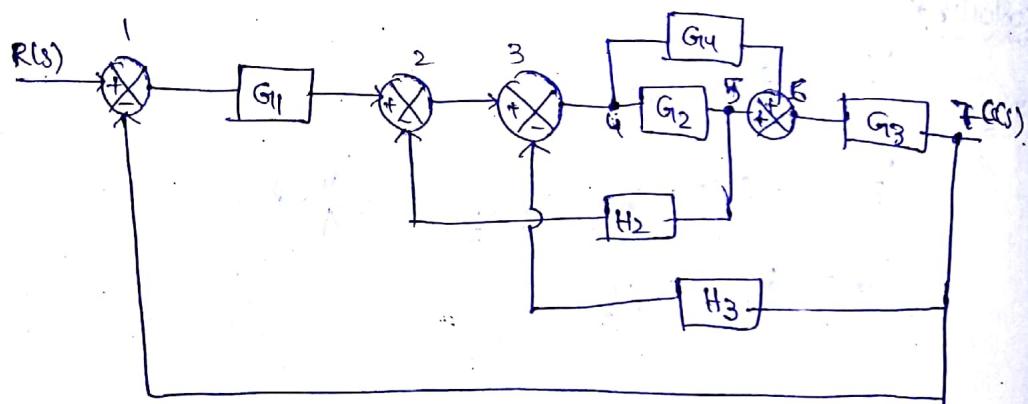
$$\Delta_2 = 1 - 0$$

$$\Delta_3 = 1 - 0$$

$$\Delta_4 = 1 - 0$$

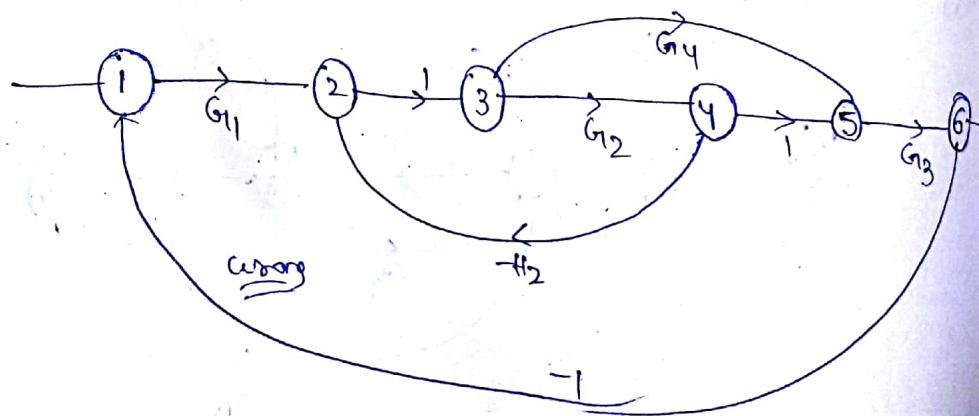
$$\frac{Y_4}{X} = \frac{\text{TF}}{\Delta} = \frac{P_1 \Delta_1 + P_2 \Delta_2 + P_3 \Delta_3 + P_4 \Delta_4}{\Delta} = \frac{G_1 G_2 G_{17} + G_{14} G_{17} + G_1 G_6 + G_{14} G_3 G_6}{1 - G_2 - G_3 - G_5}$$

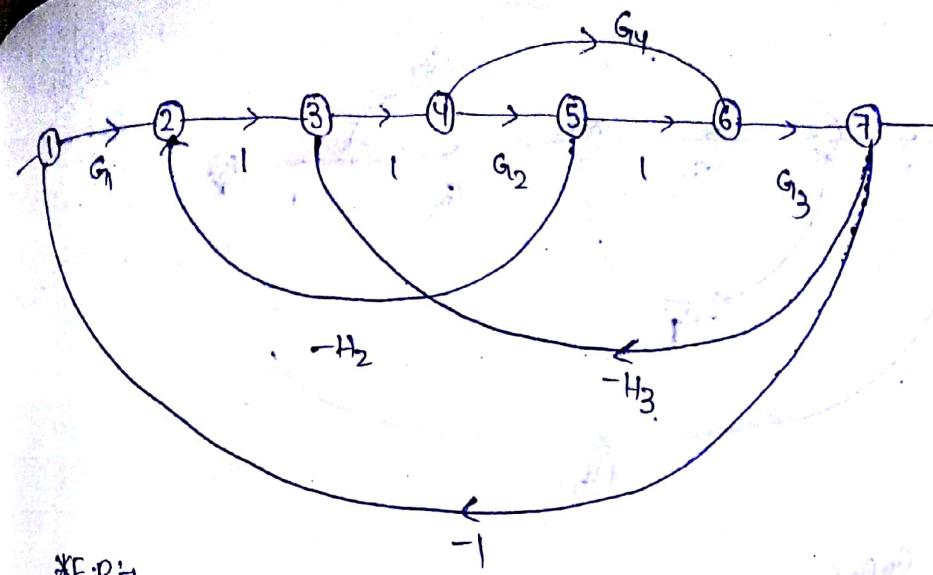
* SFG



* Draw the corresponding SFG & find $\frac{C(s)}{R(s)}$ {TF}

* SFG:





*F.P.H

$$P_1 = G_1 G_2 G_3; \quad P_2 = G_1 G_4 G_5$$

$$l_1 = -G_1 G_2 G_3 H_3$$

$$l_2 = -G_1 G_4 G_5 H_3$$

$$l_3 = -G_1 G_2 G_3$$

$$l_4 = -G_1 G_4 G_5$$

$$l_5 = -G_2 H_2$$

$$\Delta = 1 - (l_1 + l_2 + l_3 + l_4 + l_5) - 0$$

$$\frac{C(s)}{R(s)} = \frac{P_1 \Delta + P_2 \Delta_2}{\Delta}$$

$$\Delta_1 = 1 - 0$$

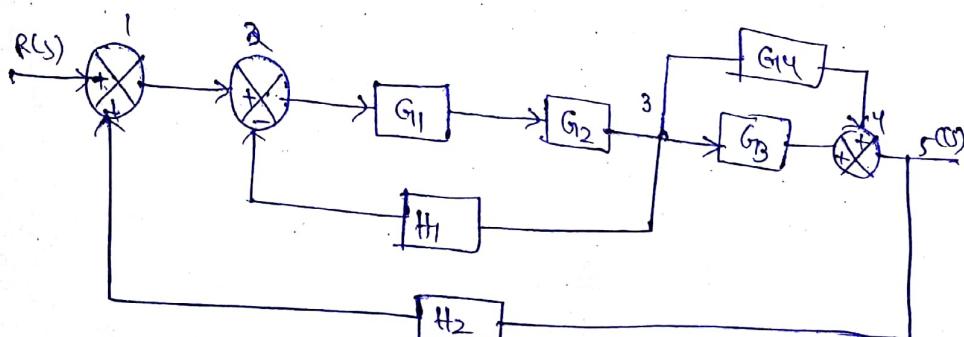
$$\Delta_2 = 1 - 0$$

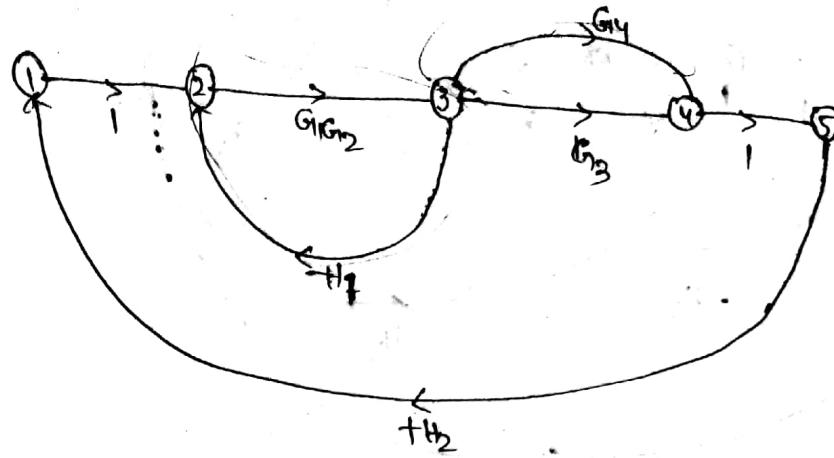
$$\Delta_3 = 1 - 0$$

$$\Delta_4 = 1 - 0$$

$$\Delta_5 = 1 - 0$$

*3.





$$P_1 = G_1 G_2 G_3$$

$$P_2 = G_1 G_2 G_4 \quad A = 1 - 0$$

$$L_1 = +G_1 G_2 G_3 H_2$$

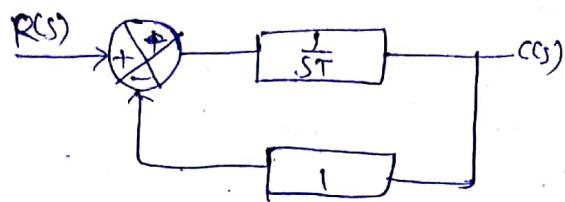
$$L_2 = +G_1 G_2 G_4 \cdot H_2$$

$$L_3 = -G_1 G_2 H_1$$

$$L_4 = G_3$$

* First order Time response Analysis :-

① Step response:-



$$\frac{C(s)}{R(s)} = \frac{\frac{1}{sT}}{1 + \frac{1}{sT}} = \frac{1}{1 + sT}$$

$$C(s) = \frac{1}{1 + sT} \cdot R(s)$$

if $i/p = u(t) \Rightarrow R(s) = 1/s$.

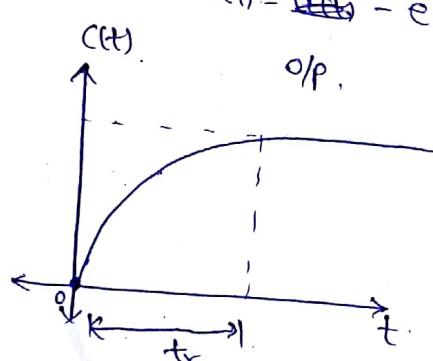
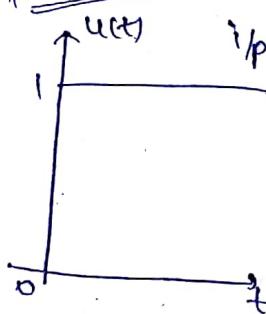
$$\therefore C(s) = \frac{1}{s} \cdot \frac{1}{1 + sT} = \frac{A}{s} + \frac{B}{1 + sT}$$

$$\therefore f/s C(s) = \frac{1}{s} - \frac{T}{1 + sT} = \frac{1}{s} - \frac{T}{s + T}$$

$$\therefore C(t) = \cancel{\frac{1}{s}} - T e^{-\frac{t}{T}}$$

$$\therefore C(t) = \cancel{\frac{1}{s}} - e^{-\frac{t}{T}} = 1 - e^{-\frac{t}{T}}$$

* Graphs *



② Ramp response:-

$$R(s) = \frac{1}{s^2}$$

$$\therefore C(s) = \frac{1}{s^2} \cdot \frac{1}{1 + sT} = \frac{A}{s} + \frac{B}{s^2} + \frac{C}{1 + sT}$$

$$A(1+sT) \cdot s + B(1+sT) + C(s^2) = 1$$

$$\text{at } s=0.$$

$$\Rightarrow B=1$$

$$\text{at } s=-\frac{1}{T}$$

$$C\left(\frac{1}{T}\right)=1$$

$$C=T^2$$

$$\text{at } A+B+C=0$$

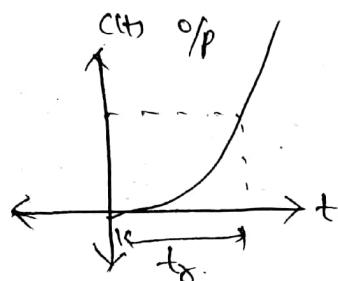
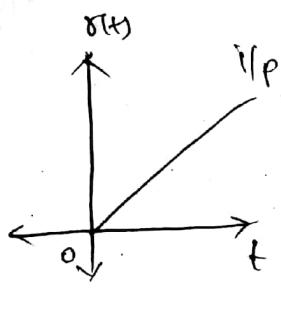
$$A=-C=-\frac{T^2}{T}=-T$$

$$\Rightarrow C(s) = -\frac{T}{s} + \frac{1}{s^2} + \frac{T^2}{1+sT}$$

$$C(t) = -T^2 t + t + T e^{t/T}$$

$$C(t) = t - T(1 - e^{-t/T})$$

$$C(t) = t - T[1 - e^{-t/T}]$$



③ Impulse \rightarrow
Response $r(s) = 1$

$$C(s) = \frac{1}{1+sT} \cdot J(s)$$

$$= \frac{1}{1+sT} = \frac{1}{T(s+\frac{1}{T})} = \frac{1}{T} e^{-t/T}$$

* Time response of 2nd order system is

$$* CLTF \text{ of a system} = \frac{\omega_n^2}{s^2 + 2\zeta\omega_n s + \omega_n^2}$$

$$\omega_n = \frac{\text{critical damping}}{\text{under damping}} = \text{damping ratio.}$$

$$\text{Char. Eq} = GE = s^2 + 2\zeta\omega_n s + \omega_n^2 \quad \{ \text{roots of } GE = \text{poles} \}$$

*1.	$\zeta = 0$	$s_1 s_2 = \pm j\omega_n$ Purely img	undamped system
*2	$\zeta = 1$	$s_1 = s_2 = -\omega_n$ Real Equal & -ve.	critically damped system
*3.	$0 < \zeta < 1$	Complex conjugate under damped system (-ve real part)	
*4.	$\zeta > 1$	Real & unequal	over damped system

* Unit step:

$$(C_s) = R(s) \cdot \frac{\omega_n^2}{s^2 + 2\zeta\omega_n s + \omega_n^2}$$

$\zeta = 0$ undamped system

$$R(s) = \frac{1}{s}$$

$$\Rightarrow (C_s) = \frac{1}{s} \cdot \frac{\omega_n^2}{s^2 + \omega_n^2}$$

$$\frac{\omega_n^2}{s(s^2 + \omega_n^2)} = \frac{A}{s} + \frac{Bs + C}{s^2 + \omega_n^2}$$

$$\Rightarrow A(s^2 + \omega_n^2) + Bs^2 + Cs = \omega_n^2$$

$$\text{at } s=0 \quad A=1, \quad \text{at } s=-\omega_n \quad A+B=0 \\ B=-1.$$

$$\text{at } s = -\omega_n L$$

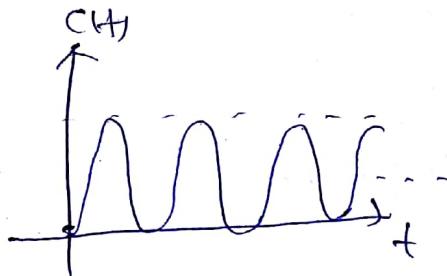
$$= -B \cdot \omega_n^2 + C(j\omega_n) = \omega_n^2$$

$$\Rightarrow C(j\omega_n) = \frac{\omega_n^2}{s+j\omega_n}$$

$$\Rightarrow C = 0.$$

$$\Rightarrow C(s) = \frac{1}{s} - \frac{1}{s^2 + \omega_n^2} = 1 - \frac{1}{\omega_n^2} \sin(\omega_n t)$$

$$\Rightarrow C(t) = 1 - \cos(\omega_n t)$$



* The response of undamped 2nd order system for the unit step input

Completely oscillatory

* Every practical system has some amount of damping only

hence, the undamped system doesn't exist in

Practise.

* 12/01/18 *

* time response of 2nd order system:-

* Case(ii),

underdamped system ($\omega_n > \xi > 1$), Roots are complex.

Conjugate.

$$\frac{C(s)}{R(s)} = \frac{\omega_n^2}{s^2 + 2\xi\omega_n s + \omega_n^2}$$

$$R(s) = Y_s.$$

$$C(s) = \frac{\omega_n^2}{s(s^2 + 2\xi\omega_n s + \omega_n^2)}$$

$$C(s) = \frac{A}{s} + \frac{Bs + C}{(s^2 + 2\xi\omega_n s + \omega_n^2)} = \frac{\omega_n^2}{s(s^2 + 2\xi\omega_n s + \omega_n^2)}$$

$$A(s^2 + Bs^2 + Cs) = \omega_n^2$$

$$A + B = 0; A = 1$$

$$B = -1$$

$$C = -2\xi\omega_n$$

$$C(s) = \frac{1}{s} - \frac{s + 2\xi\omega_n}{s^2 + 2\xi\omega_n s + \omega_n^2}$$

$$C(s) = \frac{1}{s} - \frac{s + 2\xi\omega_n}{s^2 + 2\xi\omega_n s + (\xi\omega_n)^2 - (\xi\omega_n)^2 + \omega_n^2}$$

$$C(s) = \frac{1}{s} - \frac{s + 2\xi\omega_n}{(s + \xi\omega_n)^2 + \omega_n^2(1 - \xi^2)}$$

$$\omega_d^2 = \omega_n^2(1 - \xi^2)$$

$$\omega_d = \omega_n \sqrt{1 - \xi^2} \approx \text{damping free}$$

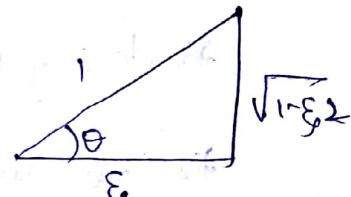
$$\Rightarrow CS = \frac{1}{s} - \frac{s + \xi \omega_n}{(s + \xi \omega_n)^2 + \omega_d^2}$$

$$\Rightarrow CS = \frac{1}{s} - \left[\frac{s + \xi \omega_n}{(s + \xi \omega_n)^2 + \omega_d^2} + \frac{\xi \omega_n}{(s + \xi \omega_n)^2 + \omega_d^2} \right]$$

$$\Rightarrow C_H = 1 - e^{-\xi \omega_n t} \cdot \cos(\omega_d t) + \frac{\xi \omega_n}{\omega_d} \cdot \sin(\omega_d t)$$

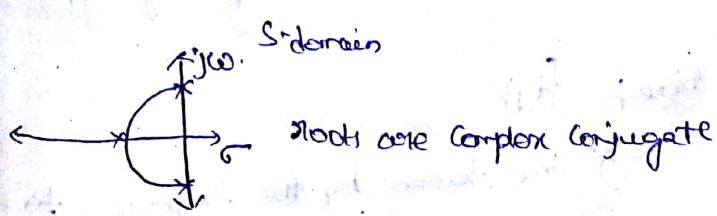
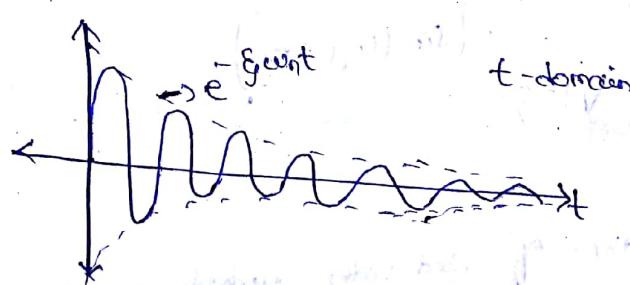
$$\Rightarrow C_H = 1 - e^{-\xi \omega_n t} \cdot \cos(\omega_d t) + \frac{\xi \omega_n}{\omega_d(1 - \xi^2)^{1/2}} \sin(\omega_d t) e^{-\xi \omega_n t}$$

$$\Rightarrow C_H = 1 - e^{-\xi \omega_n t} \sqrt{1 - \xi^2} \left[\sqrt{(1 - \xi^2)} \cdot \cos(\omega_d t) + \xi \sin(\omega_d t) \right]$$



$$\Rightarrow C_H = 1 - \frac{e^{-\xi \omega_n t}}{\sqrt{1 - \xi^2}} \left[\sin \theta \cos(\omega_d t) + \cos \theta \sin(\omega_d t) \right]$$

$$\Rightarrow C_H = 1 - \frac{e^{-\xi \omega_n t}}{\sqrt{1 - \xi^2}} \left[\sin(\omega_d t + \theta) \right]$$



Locus of the roots of the C-E follows a semi-circle then the response

* undamped response.

→ * When $\xi < 1$ output reaches its steady state in moderate time i.e., neither too quick nor too sluggish

The 90% of CS designs, type of damping is underdamping.

→ * If we plot all the poles & zeros of 2nd order system in underdamped system the locus of these poles gives semi-circle.

→ * Characteristics of 2nd order undamped response;

*1. damping coefficient (ξ_{wn})

*2. Time Constant of $\frac{1}{\xi_{wn}}$

*3. damping frequency $\{\omega_d\}$ [$\omega_d = \omega_n \sqrt{1-\xi^2}$]

*4. damping angle $\{\theta\}$ [$\omega_0 \theta = \xi$, $\theta = \cot^{-1}(\xi)$]

$$* C(t) = 1 - \frac{e^{-\xi \omega_n t}}{\sqrt{1-\xi^2}} \cdot [\sin(\omega_d t + \theta)]$$

→ * Specification of 2nd order undamped system;

*1. Delay time; $t_{1/2}$

* Time taken by the response to reach 50% of the final value of Steady state value.

*2. Rise time

of its

*3. peak

*4. Set

*5.

*6. S

*2 Rise time $\{t_r\}$

* Time taken by the response to reach 10 to 90%
of its final value

*3. peak time $\{t_p\}$

* Time taken by the response to reach its first
Peak. of the response.

*4. Settling time $\{t_s\}$

* The time taken by the response to reach its
Steady state & stay with in the 2% or 5% of
tolerance band is known as settling time.

*5. Peak overshoot \rightarrow

* The maximum amount of overshoot taken
by the response is known as Peak overshoot.

It is also defined as difference b/w max value
& desired o/p. of steady state value.

$$\% \text{ Max peak overshoot} (M_p) = \left[C(t) \right]_{t=t_p} - \left[C(t) \right]_{\text{desired o/p.}} / \left[C(t) \right]_{\text{desired o/p.}}$$

*6 Steady State Error \rightarrow

* It is defined as Error b/w actual o/p to
desired o/p.

$$E_{ss} = E_{eff} - \text{actual o/p} - \text{desired o/p.}$$

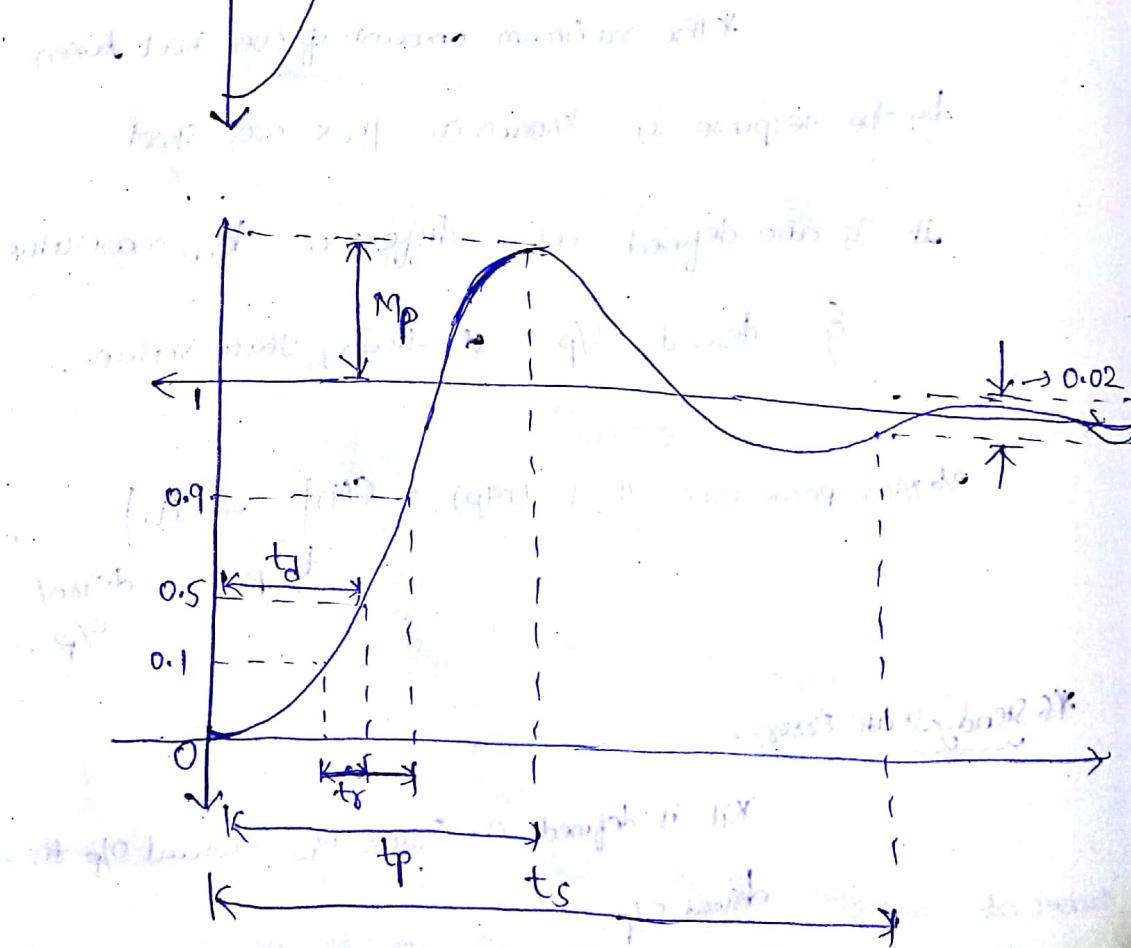
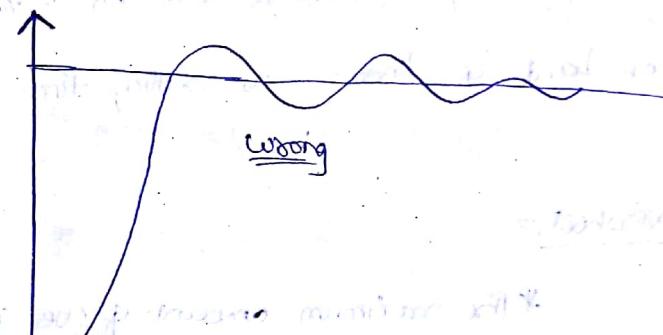
$$= (C(t))_a - (C(t))_d.$$

* 16 bits *

* TIME DOMAIN SPECIFICATIONS FOR UNDER DAMPED SYSTEM *

$$C(t) = 1 - \frac{e^{-\xi t \omega_n}}{\sqrt{1-\xi^2}} \sin(\omega_d t + \theta)$$

$$\% \text{M.P. peak overshoot time} (\% \text{M.P.}) = \frac{C(t_p) - C(\infty)}{C(\infty)}$$



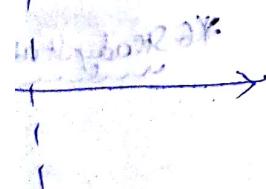
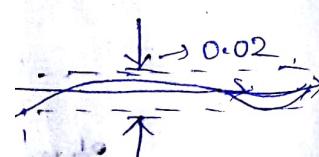
- * For undamped system $t_r = 10\% \text{ to } 90\%$
- * for underdamped system $t_r = 0\% \text{ to } 100\%$
- * For critically damped system $t_r = 5\% \text{ to } 95\%$.

DAMPED SYSTEM*

$$\frac{C(t_p) - C(\infty)}{C(\infty)}$$

Step response is characterised by the following performance indices :

- *1. How fast the system moves to follow its input.
- *2. How oscillatory it is (indicative of damping).
- *3. How long does it take practically to reach to its final value.



* Derivation of time domain specifications of 2nd order underdamped system

* (1) Rise time for underdamped system 0% to 100%

$$C(t) = 1 - e^{-\xi \omega_n t}$$

$$\sqrt{1-\xi^2} \cdot \sin(\omega_d t + \theta) \rightarrow \text{II}$$

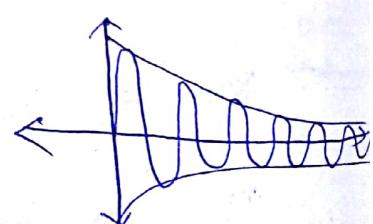
$$\text{at } t = t_r \Rightarrow C(t_r) = 1$$

$$1 = 1 - \frac{e^{-\xi \omega_n t_r}}{\sqrt{1-\xi^2}} \cdot \sin(\omega_d t_r + \theta)$$

$$\frac{1}{\sqrt{1-\xi^2}} \cdot \sin(\omega_d t_r + \theta) = 0$$

$$\omega_d t_r + \theta = n\pi = \theta$$

$$t_r = \frac{\pi - \theta}{\omega_d}$$



$$\text{Here } \pi - \theta = \cos^{-1}(\xi)$$

$$\pi = 3.14$$

$$\omega_d = \omega_n \sqrt{1-\xi^2}$$

* Peaktime: t_p

* Differentiate (I) on B.S we get.

$$\Rightarrow \left. \frac{dC(t)}{dt} \right|_{t=t_p} = 0.$$

$$\Rightarrow -\left[\frac{e^{-\xi \omega_n t}}{\sqrt{1-\xi^2}} \right] [-\xi \omega_n] [\sin(\omega_d t + \theta)] + \frac{e^{-\xi \omega_n t}}{\sqrt{1-\xi^2}} \cdot$$

$$[\cos(\omega_d t + \theta) \cdot \omega_d] = 0$$

$$\Rightarrow \frac{e^{-\xi \omega_n t}}{\sqrt{1-\xi^2}} [\omega_d \cos(\omega_d t + \theta) - \xi \omega_n \sin(\omega_d t + \theta)] = 0.$$

$$\Rightarrow \omega_d \cdot [\cos(\omega_d t + \theta) - \xi \omega_n \sin(\omega_d t + \theta)] = 0.$$

$$\Rightarrow \omega_n \sin \theta \cos(\omega_d t + \theta) - \cos \theta \cdot \omega_n \sin(\omega_d t + \theta) = 0.$$

$$\Rightarrow \sqrt{1-\xi^2} = \sin \theta; \quad \xi = \cos \theta; \quad \omega_d = \omega_n \sqrt{1-\xi^2}$$

$$\Rightarrow \sin \theta \cos(\omega_d t + \theta) - \cos \theta \sin(\omega_d t + \theta) = 0.$$

$$\Rightarrow \sin(\omega_d t + \theta - \theta) = 0$$

$$\Rightarrow \omega_d t_p = \theta \cdot n\pi = \pi \quad \text{(at } t = t_p\text{)}$$

$$\boxed{t_p = \frac{\pi}{\omega_d}}$$

$$*3 \% MP \Rightarrow \frac{C(t_p) - C(\infty)}{C(\infty)}$$

$$\cancel{\text{Ansatz}}: C(t_p) = 1 - \frac{e^{-\xi_{\text{cont}} t_p}}{\sqrt{1-\xi^2}} \sin(\omega_d t_p + \theta)$$

$$\cancel{C(\infty)} = 1 - 0 = 1$$

$$\cancel{t_p} = \frac{\pi}{\omega_d}$$

$$\Rightarrow C(t_p) = 1 - \frac{e^{-\xi_{\text{cont}} \cdot \frac{\pi}{\omega_d}}}{\sqrt{1-\xi^2}} \sin\left(\omega_d \cdot \frac{\pi}{\omega_d} + \theta\right)$$

$$\Rightarrow C(t_p) = 1 - \frac{e^{-\xi \frac{\pi}{\sqrt{1-\xi^2}}}}{\sqrt{1-\xi^2}} \sin(\pi + \theta)$$

$$\Rightarrow C(t_p) = 1 + \frac{e^{-\xi \frac{\pi}{\sqrt{1-\xi^2}}}}{\sqrt{1-\xi^2}} \sin(\theta) \quad \{ \sin\theta = \sqrt{1-\xi^2} \}$$

$$\Rightarrow \% MP = \frac{1 + e^{-\xi \frac{\pi}{\sqrt{1-\xi^2}}}}{1}$$

$$\boxed{\% MP = e^{-\xi \frac{\pi}{\sqrt{1-\xi^2}}}}$$

* 20/01/18
Setting Time (t_s)

* The response of 2nd order system by 2 components

$$(i) \text{ decaying exponential component: } \left\{ \frac{e^{-\xi \omega_n t}}{\sqrt{1-\xi^2}} \right\}$$

$$(ii) \text{ Sine component of } \sin(\omega_n t + \theta)$$

* In this the decaying Exp term reduces the oscillation by
 ↘ Produced by Sine component hence, settling time is
 ↗ decided by Exp term

* The settling time can be calculated by Equating Exp component to the % of tolerance

* For 2% tolerance band

$$e^{-\xi \omega_n t_s} = 0.02$$

$$\Rightarrow -\xi \omega_n t_s = \ln(0.02)$$

$$\Rightarrow t_s = \frac{4}{\xi \omega_n}$$

* For 5% tolerance band

$$\Rightarrow e^{-\xi \omega_n t_s} = 0.05$$

$$\Rightarrow t_s = \frac{3}{\xi \omega_n}$$

i.e., $t_s = \frac{1}{\xi \omega_n} \ln [\text{tolerance band}]$

$$\text{o.r. } \frac{\ln [\text{Expt.}]}{\xi \omega_n}$$

$$= \frac{\ln [\text{tolerance band}]}{\xi \omega_n} \times T ; T = \frac{1}{\xi \omega_n} \text{ Time constant of Order 2 System}$$

* Problem 1
 Q1 A unit step signal is applied to the unity F/B C.S. for which OLTF is $G(s) = \frac{16}{s(s+8)}$ find its CLTF, ξ , ω_n , ω_d .

* Sol. $CLTF = \frac{OLTF}{1+OLTF} = \frac{G(s)}{1+G(s).1} \quad \{1+G(s)=1\}$

$$\frac{\frac{16}{s(s+8)}}{1 + \frac{16}{s(s+8)}} = \frac{16}{s^2 + 8s + 16}$$

w.k.t. $CLTF = \frac{\omega_n^2}{s^2 + 2\xi\omega_n s + \omega_n^2}$

$$\Rightarrow \omega_n = 4$$

$$2\xi\omega_n = 8$$

$$2\xi = 2$$

$$\Rightarrow \xi = 1$$

$$\Rightarrow \omega_d = \omega_n \sqrt{1-\xi^2} = 4\sqrt{1-1} = 0.$$

* Q2 The CLTF of a 2nd order system is given by $T(s) = \frac{100}{s^2 + 10s + 100}$
 determine ξ , ω_n , t_r , t_s , $\% M.P.$

* Sol) $\Rightarrow \omega_n^2 = 100 \Rightarrow \omega_n = 10.$

$$\Rightarrow 2\xi\omega_n = 10 \quad ; \quad \omega_d = \omega_n \sqrt{1-\xi^2} \\ \Rightarrow \xi = \frac{1}{2} \quad ; \quad \omega_d = 10 \sqrt{1-\frac{1}{4}} = \frac{10\sqrt{3}}{2}$$

$$t_f = \frac{\pi - \theta}{\omega_d} = \frac{\pi - \cot^{-1}(g)}{5\sqrt{3}} = \frac{\pi - 60}{5\sqrt{3}} = 0.24 \text{ sec}$$

by F/B C.S. for

CLTF, ω_n ,

$$t_s = \frac{4}{g\omega_n} = \frac{4}{(k_2)(10)} = \frac{4}{5} = 0.8 \text{ sec.} \quad \text{10% d/g}$$

$$M_p = e^{-\frac{g\pi}{\sqrt{1-g^2}}} = e^{-\frac{\pi}{\sqrt{1-\frac{1}{4}}}} = e^{-\frac{\pi}{\sqrt{\frac{3}{4}}}} = e^{-\frac{\pi}{\sqrt{3}}} = e^{-\frac{\pi}{\sqrt{3}}}$$

$$= 0.16303$$

$$\Rightarrow \% M_p = 16.3\%$$

16

$s^2 + 8s + 16$

*③ obtain the response of unity F/B system whose CLTF is $G(s) = \frac{4}{s(s+5)}$ when the i/p is unit step.

(Q)

$$\text{CLTF} := \frac{4}{s(s+5)} = \frac{4}{s^2 + 5s + 9}$$

$$\Rightarrow \text{CLTF} = \frac{4}{s^2 + 5s + 9}$$

$$\Rightarrow \frac{C(s)}{R(s)} = \frac{4}{s^2 + 5s + 9}$$

$$\Rightarrow C(s) = \frac{R(s) \cdot 4}{s^2 + 5s + 9} \quad \left\{ R(s) = \frac{1}{s} \Rightarrow \text{unit step} \right. \\ \left. \Rightarrow \text{d(t)} \right\}$$

$$\Rightarrow C(s) = \frac{4}{s(s^2 + 5s + 9)} = \frac{4}{s(s^2 + 4s + s + 9)}$$

$$\Rightarrow C(s) = \frac{4}{s(s(s+4) + 1(s+4))} = \frac{4}{s(s+1)(s+4)} \quad \frac{K(s+2)}{12}$$

$$\text{by } T(s) = \frac{100}{s^2 + 10s + 100}$$

A.P.

$$= \omega_n \sqrt{1-g^2} \\ = 10 \sqrt{1-\frac{1}{4}} = \frac{10\sqrt{3}}{2}$$

$$C(s) = \frac{1}{s} + \frac{(-4y_3)}{s+1} + \frac{y_3}{s+4}$$

$$A(s+1)(s+4) + B(s)(s+4) + C(s)(s+1) = 4$$

at $s=0$

$$\Rightarrow A=1$$

at $s=-4$

$$\Rightarrow C(-4)y_3 = 4$$

$$\Rightarrow C = \frac{1}{y_3}$$

at $s=-1$

$$\Rightarrow B(-1)y_3 = 4 \Rightarrow -4y_3$$

$$\Rightarrow C(t) = u(t) - \frac{4}{3}e^t u(t) + \frac{1}{3} \cdot e^{-4t} u(t)$$

$$\Rightarrow C(t) = u(t) - \frac{1}{3} [4e^t u(t) - e^{-4t} u(t)]$$

* Steady state ~~and constraints~~

* Order & type of a system

origin

* Type number indicates no. of poles at

* Type of order indicates no. eff power of 's'

Main general if N is the no. of poles at origin then
the type number is N

* If $N=0$ then the system is type zero system.

* If $N=1$ then the system is type one system

* If $N=2$ then the system is type two system

* Steady state error is a value of Error signal "err"

When $t \rightarrow \infty$ Steady State is measure of system accuracy.

those errors are occur from the nature of ip's type

of system & non-linearity in system components.

* The steady state performance of stable control system is generally specified by its steady state error to unit step, ramp, parabola inputs.

* ~~adol118~~ \rightarrow Take Xesox.

~~monkey~~
~~2210118~~

* For a unity f/b System OLTF = $G(s) = \frac{K(s+2)}{s(s^2+7s+12)}$

① find order & type of a system

② Error Co-efficients.

③ Steady state error when ip to the system is $\frac{R}{2}t^2$.

$$* \underline{\text{Sol}} \quad \text{Type} = 1 \quad \{ \text{1 pole at origin}\}$$

$$\text{order} = 3 \quad \{ \text{Total 3 poles} \}$$

$$* ② \quad K_p = \lim_{s \rightarrow 0} G(s) \cdot H(s) = \infty$$

$$K_v = \lim_{s \rightarrow 0} s \cdot G(s) \cdot H(s) = \frac{K}{6}$$

$$K_a = \lim_{s \rightarrow 0} s^2 G(s) \cdot H(s) = 0$$

$$* ③ e_{ss} = \lim_{s \rightarrow 0} \frac{s R(s)}{1 + G(s) \cdot H(s)} = \frac{\frac{R \cdot R}{s^2}}{1 + \frac{K(s+2)}{s(s^2+7s+12)}} = \frac{\cancel{R/s}}{\cancel{(s^2+7s+12)+K(s+2)}} = \frac{\cancel{R/s}}{(s^2+7s+12)}$$

$$= \lim_{s \rightarrow 0} \frac{R(s^2+7s+12)}{s(s^2+7s+12)+K(s+2)} = \frac{\cancel{R/s}}{\cancel{2K}} = \frac{6R}{2K}$$

$$e_{ss} = \lim_{s \rightarrow 0} \frac{s \cdot R}{s^3} = \lim_{s \rightarrow 0} \frac{R}{s^2} \times \frac{s(s+7+12)}{s(s+7+12) + k(48)} \\ = \lim_{s \rightarrow 0} \frac{R}{s^2} \times \frac{s(s+19)}{s(s+19) + k(48)} \\ \Rightarrow e_{ss} = \infty \text{ at } s=0.$$

* Q) Loop TF of a FB CS is given by $G(s) \cdot H(s) = \frac{100}{s^2(s+4)(s+12)}$
 determine static error coefficient also determine
 steady state error for the i/p $\delta(t) = 2t^2 + 5t + 10$.

(*)

$$* K_p = \lim_{s \rightarrow 0} G(s) \cdot H(s) = \infty.$$

$$* K_v = \lim_{s \rightarrow 0} s \cdot G(s) \cdot H(s) = \infty.$$

$$* K_a = \lim_{s \rightarrow 0} s^2 G(s) \cdot H(s) = \frac{100}{48} = 2.0833 //$$

$$* e_{ss} = \lim_{s \rightarrow 0} \frac{s \cdot R(s)}{1 + G(s) \cdot H(s)} = \lim_{s \rightarrow 0} \frac{s \cdot \left[\frac{4}{s^3} + \frac{5}{s^2} + \frac{10}{s} \right]}{1 + \frac{100}{s^2(s+4)(s+12)}} \\ = \lim_{s \rightarrow 0} \frac{s^3 \left[\frac{4}{s^3} + \frac{5}{s^2} + \frac{10}{s} \right]}{(s+4)(s+12)} \\ = \lim_{s \rightarrow 0} \frac{(4 + 5s + 10s^2)(s+4)(s+12)}{s^2(s+4)(s+12) + 100} \\ = \frac{4(4)(12)}{100} = \frac{48 \times 12}{100 + 25} = \frac{48}{25} = 1.92$$

*③ For a system $G(s) \cdot H(s) = \frac{K}{s^2(s+2)(s+3)}$ find the value of K

To limit the steady state error is 10. when input

$$\text{to the system is } 1 + 10t + \frac{40}{2} t^2 \Rightarrow r(s) = \frac{1}{s} + \frac{10}{s^2} + \frac{40}{2} \frac{t^2}{s^3}$$

$$*(\text{sol}) \quad e_{ss} = \lim_{s \rightarrow 0} \frac{s \cdot R(s)}{1 + G(s) \cdot H(s)} = \lim_{s \rightarrow 0} \frac{s \left[\frac{1}{s} + \frac{10}{s^2} + \frac{40 \times \frac{t^2}{2}}{s^3} \right]}{1 + \frac{K}{s^2(s+2)(s+3)}}$$

$$\Rightarrow e_{ss} = \lim_{s \rightarrow 0} \frac{s^3 (s+2)(s+3) \left(\frac{1}{s} + \frac{10}{s^2} + \frac{40 \times \frac{t^2}{2}}{s^3} \right)}{s^2 (s+2)(s+3) + K}$$

$$\Rightarrow 10 = \lim_{s \rightarrow 0} \frac{(s+2)(s+3) (s^2 + 10s + 40)}{s^2 (s+2)(s+3) + K}$$

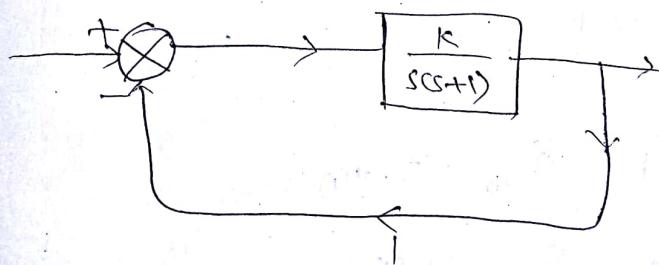
$$\Rightarrow 10 = \frac{(2)(3)(40)}{K}$$

$$\Rightarrow K = 24$$

*④ Assume $r(t) = 0.1t$ & it is desired that steady state

Error is ≤ 0.005 . If $e_{ss} \leq 0.005$, find the range of

values of "K" for error to be specified limit, for the given system is



$$*(\text{sol}) \quad e_{ss} = \lim_{s \rightarrow 0} \frac{s \cdot R(s)}{1 + G(s) \cdot H(s)} = \lim_{s \rightarrow 0} \frac{s \cdot 0.1}{1 + \frac{K}{s(s+1)}} = \frac{0.1}{s+1}$$

$$e_{ss} = \lim_{s \rightarrow 0} \frac{\frac{0.1}{s} \cdot s(s+1)}{(s+1)s+K} = \frac{0.1}{K}$$

$$\frac{48}{25} = 1.92$$

$$e_{ss} = \frac{0.1}{K}$$

to limit e_{ss} to 0.005

$$\Rightarrow 0.005 = \frac{0.1}{K} \Rightarrow K = \frac{1}{0.005}$$

$$= K = \frac{0.1}{0.005} = \frac{1}{0.005} \times \frac{1000}{5} = 200$$

$$\Rightarrow K = 200$$

i.e.. for any value of $K \geq 200$; $e_{ss} \leq 0.005$

i.e. $200 \leq K \leq \infty$

- * Q) For a unity f/B system having $G(s) = \frac{35(s+4)}{s(s+2)(s+5)}$
 find ① Type of the system
 ② all error co-eff (k_p, k_v, k_a)
 ③ Steady State Error (e_{ss}) for ramp input with magnitude of 5.

* Q) ① Type = I

$$\textcircled{2} \quad k_p = \lim_{s \rightarrow 0} G(s) \cdot H(s) = 0$$

$$\textcircled{3} \quad k_v = \lim_{s \rightarrow 0} s G(s) \cdot H(s) = \frac{7}{10} = 0.7$$

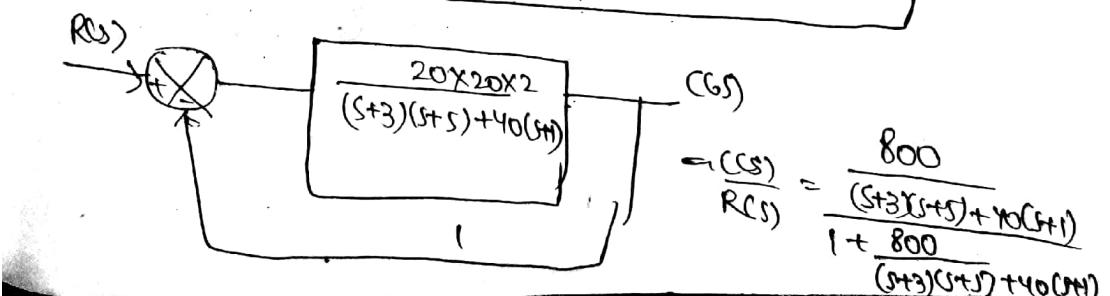
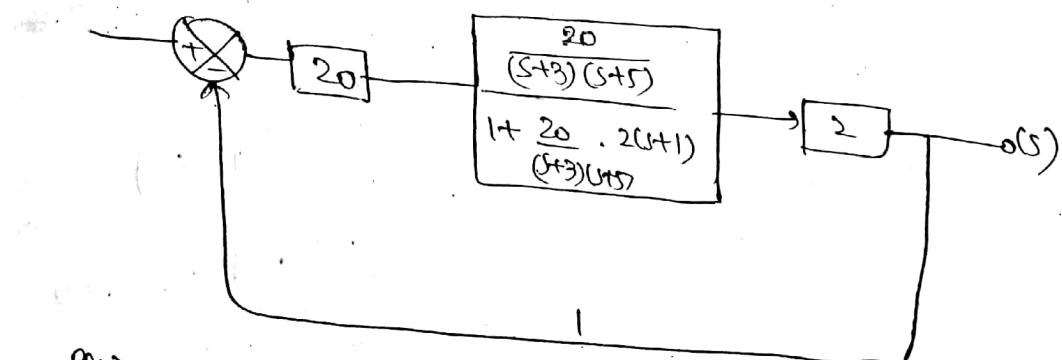
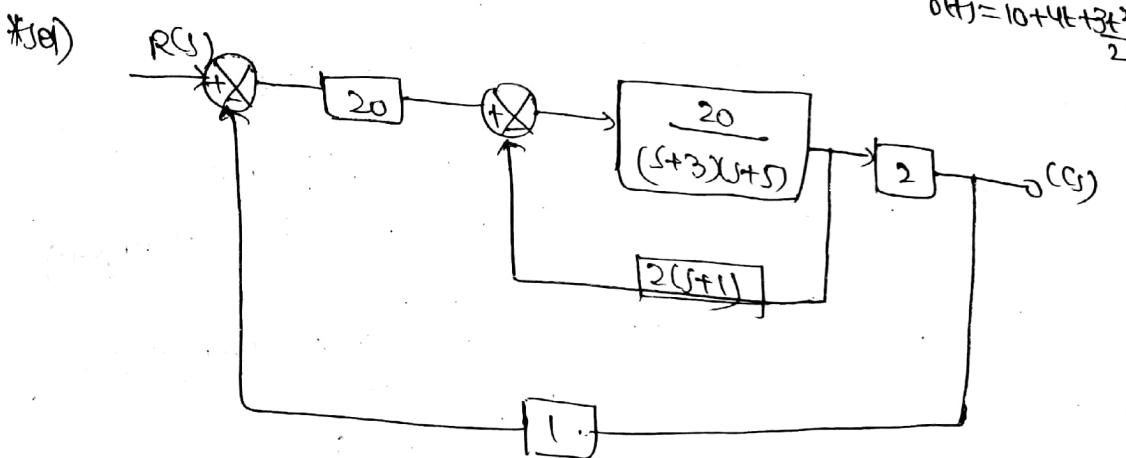
$$\textcircled{4} \quad k_a = \lim_{s \rightarrow 0} s^2 G(s) \cdot H(s) = 0$$

$$\textcircled{1} \quad e_{ss} = \lim_{s \rightarrow 0} \frac{s \cdot R(s)}{1 + G(s) \cdot H(s)} = \frac{s \cdot \frac{5}{s^2}}{1 + \frac{35(s+4)}{s(s+2)(s+5)}}$$

$$\Rightarrow e_{ss} = \lim_{s \rightarrow 0} \frac{\frac{5}{s}(s+2)(s+5)}{s(s+2)(s+5) + 35(s+4)}$$

$$\Rightarrow e_{ss} = \frac{\cancel{5} \times \cancel{2} \times \cancel{5}}{\cancel{35} \times \cancel{4} \cancel{2}} = \frac{5}{14}$$

- *⑥ For the following system determine
- *③ Type of the system
 - *⑤ Lead Co-eff
 - *⑦ e_{ss} for i/p's $\delta(t) = 6$
 - * $\delta(t) = 8t$
 - $\delta(t) = 10 + 4t + \frac{3t^2}{2}$



$$\frac{C(s)}{R(s)} = \frac{800}{(s+3)(s+5) + 40(s+1) + 800}$$

*@ type = 0.

*⑥

$$K_p = \lim_{s \rightarrow 0} G(s) \cdot H(s) = \lim_{s \rightarrow 0} \frac{800}{(s+3)(s+5) + 40(s+1)}$$

$$= \frac{800}{(3)(5)+40} = \frac{800}{15+40} = \frac{800}{55}$$

$$k_v = k_a = 0.$$

*⑦(i) $\delta(t) = 6$ $\therefore R(s) = \frac{6}{s}$

$$E_{ss} = \lim_{s \rightarrow 0} \frac{s \cdot R(s)}{(1 + G(s)) \cdot H(s)} = \frac{\frac{6}{s} \cdot \frac{s}{s}}{1 + \frac{800}{(s+3)(s+5) + 40(s+1)}}$$

$$= \lim_{s \rightarrow 0} \frac{6(s+3)(s+5) + 40(s+1)}{(s+3)(s+5) + 40(s+1) + 800}$$

$$\Rightarrow \frac{6[(3)(5)+(40)]}{(3)(5)+(40)+800} = \frac{90+60}{15+80+800} = \frac{150}{915} = 0.3859$$

$$*(\text{ii}) \quad R(s) = 8t \rightarrow R(s) = \frac{8}{s^2}$$

$$\bar{e}_{ss} = \lim_{s \rightarrow 0} \frac{\frac{8}{s^2}}{1 + \frac{800}{(s+3)(s+5)+40(s+1)}} = \frac{8}{s[1]} = \infty$$

$$s+40(s+1)$$

$$= \frac{800}{15+40} = \frac{800}{55}$$

$$*(\text{iii}) \quad R(s) = \frac{10s+4t+3t^2}{2} \Rightarrow R(s) = \frac{10}{s} + \frac{4}{s^2} + \frac{3}{s^3}$$

$$\bar{e}_{ss} = \lim_{s \rightarrow 0} \frac{s \left[\frac{10}{s} + \frac{4}{s^2} + \frac{3}{s^3} \right]}{1 + \frac{800}{(s+3)(s+5)+40(s+1)}} = \infty$$

$$\frac{s \cdot 6}{s} + \frac{800}{s}$$

$$(s+3)(s+5)+40(s+1)$$

$$3(s+5)+40(s+1)$$

~~$$1. (s+5)+40(s+1) + \frac{800}{s}$$~~

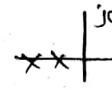
~~$$4(40) = \frac{90+40}{15+840}$$~~

~~$$5(40)+800$$~~

~~$$859 \quad 0.3859$$~~

20/11/8

* Stability analysis of a CS

* Absolute Stable System  Rely* Relative " "  System of varies with vibration in system parameters.* Conditionally " "  a & b are the roots.* Asymptotically " " If $a \neq -ve$ then unstable

↳ Neither relative nor conditional systems are Asymptotically stable systems.

 $a = -ve$ then stable $a = 0$ then stable.

* Stability :-

* If there is small change in input condition or system parameters "or" initial conditions then the opf ~~cannot~~ should not be changed ~~in the system~~.

* A system is said to be stable if & only if for bounded input (finite) if the system produces bounded output.

* Under the absence of u_p , the opf of the system should be zero.* Absolutely Stable System:* A system is said to be Absolutely stable for all the parameters over the entire range ($-\infty < t < \infty$) the system output is constant.

Conditionally Stable System:

* For a particular parameter, if for a particular range, the system output is constant.

Relative Stable System:

* If a system is said to be relative stable in accordance with another system with the relative parameters (settling time), the system output measured then it is called relative stability.

Asymptotically Stable System:

* A system is said to be asymptotically stable, when its output should tend to "0" under the absence of i/p or zero input.

Marginaly Stable System:

* If a system is neither absolutely stable nor unstable system, is known as marginally stable system.

Concept of location of poles & stability:-

① * If all the poles are lying on left half of s-plane, then the system is absolutely stable system.

② * If non-repeated poles are lying of Imag axis then the

System is marginally stable

*③ If repeated poles are lying on Imag axis then the system is unstable.

*④ If any one pole is lying on Right half of s-plane the system is unstable.