

CubeSat Survey

Miranda Straub, Ryan Watson, John A. Christian, and Jason N. Gross
Department of Mechanical and Aerospace Engineering
West Virginia University
Morgantown, WV 26505

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Abstract

In recent years, there has been a significant migration towards small satellite missions due to their reduced cost and development time. Specifically, CubeSats have become increasingly popular not only in university environments, but government and private industries as well. The standardization of these 10 cm CubeSats has also allowed for rideshare opportunities, thus finding a launch service provider is much simpler, allowing for more frequent access to space. CubeSat standards also allow for commercial off the shelf (COTS) hardware components to be made available at reasonable prices, however there is no standard software as of 2014. NASA Independent Verification and Validation (IV&V) facility has requested a survey of CubeSats to provide additional insight into all previous CubeSat missions. This information will assist them in developing a standard software to be used for all future IV&V CubeSat missions as well as missions at NASA Goddard Space Flight Center (GSFC).

1 Introduction

The CubeSat Project, developed in 1999 by Professor Jordi Puig-Suari of California Polytechnic State University (Cal Poly), San Luis Obispo and Professor Bob Twiggs of Stanford University's Space Systems Development Laboratory (SSDL), was an initial attempt at standardizing design parameters for small satellites in order to reduce cost and development time. As a result, these standards have allowed for more frequent launches and therefore easier access to space due to the low-budget nature of CubeSats [?]. The CubeSat Project outlines the standards required for CubeSat development while describing specific parameters (i.e. dimensions, mass, deployment mechanisms, etc.) that must be met for the spacecraft to qualify as a CubeSat.

When considering dimensions, a 1U (1 unit) CubeSat must have the dimensions of 10cm x 10cm x 10cm. Alternate configurations exist and include 2U and 3U CubeSats with dimensions of 10cm x 10cm x 20cm and 10cm x 10cm x 30cm, respectively (see Figure ??). In addition to dimensions, mass limitations prohibit CubeSats from exceeding 1.3 kg per unit (U), meaning a 2U CubeSat must not exceed 2.6 kg while a 3U must not exceed 4.0 kg. These limitations also limit the amount of power available for use on the spacecraft. The specified dimensions pose limitations regarding the surface area available for solar panels. Previous studies have shown that the average power of a CubeSat can vary from 1W (1U) and 5W (3U). However, assuming the spacecraft still meets the mass requirements, deployable solar panels can be used to significantly increase the amount of available power (i.e. a 3U CubeSat with deployable panels can have up to 20W of available power) [?].

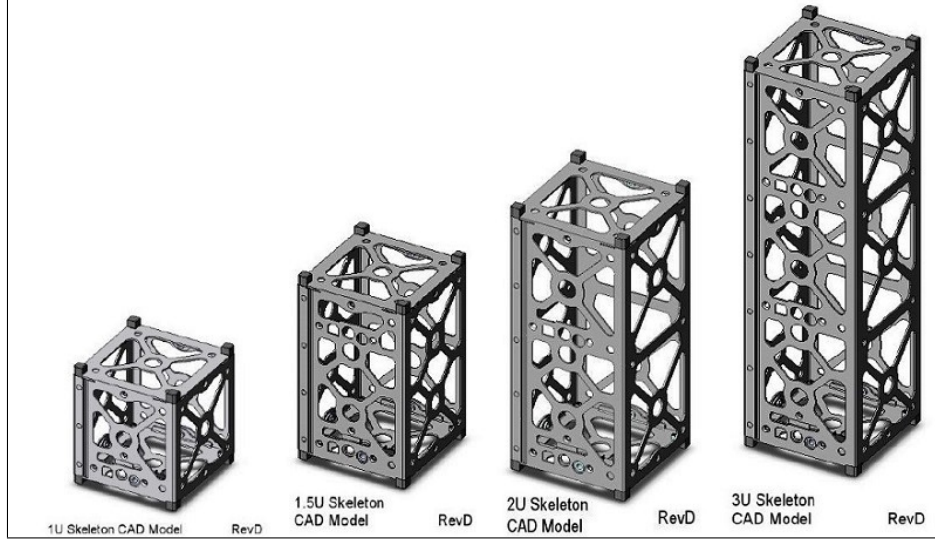


Figure 1: Example of CubeSat Configurations[?]

CubeSats that meet dimension and mass requirements can be launched from the Poly Picosatellite Orbital Deployer (P-POD) developed by Cal Poly. The P-POD serves as a standardized deployment system for up to three 1U CubeSats or any equivalent combination. Any CubeSat being deployed from a P-POD must adhere to additional specific guidelines in the CubeSat Design Specifications document in order to ensure compatibility for a successful deployment. Some recently proposed experiments, however, have been designed for 6U CubeSat structures, resulting in the need for an alternative deployer. Fortunately, NASA Wallops, NASA Ames, as well as other institutions have come up with designs for 6U deployers, all of which have been tested and have received a TRL of at least 6 [?].

The standards developed by Cal Poly have resulted in a significant increase in the number of small satellite missions launched over the past few years [?]. This is partially due to the development of commercial off-the-shelf (COTS) components specifically designed to meet all CubeSat requirements. Because of this, CubeSats have become relatively inexpensive, making them ideal educational tools in university and classroom settings. With the development of COTS components, there are a variety of options for various processors and operating systems; however, these options may not always meet the criteria required for various missions. It is for this reason that NASA IV&V has been asked to develop a portable, pure software build-and-test system that will provide the capability to develop, debug, and test software on a standard personal computer with no specialized hardware. Once this is developed, it is expected to be used in collaboration with NASA GSFC's SpaceCube development and testing.

2 Methodology

To determine possible options for a standardized NASA IV&V software, a comprehensive survey was conducted on past, present, and future CubeSat missions. This survey initially consisted of all CubeSat missions able to be found online and totaled slightly over 250 missions. For this particular report, it was decided that the focus would be on domestic missions which narrowed the search to approximately 150 missions, some of which had been launched while others were still only proposed concepts. This list was consolidated once more to focus on already launched missions as well as

missions with launch services confirmed in upcoming years. As a result, a comprehensive list of approximately 115 CubeSats has been developed.

To gather information, the first step involved creating a comprehensive list of mission details. This information can be categorized as shown in Table ???. A first pass of obtaining data was accomplished by thoroughly searching the internet, but a significant amount of data was still incomplete. As a second pass, the project managers of each mission were emailed and asked specifically for information regarding data of potential interest to NASA IV&V as well as any other applicable information or further references. This helped fill in a significant amount of missing data which was used to create a spreadsheet of all collected data. This spreadsheet will be provided electronically upon delivery of this report and includes all data collected in the categories described in Table ??.

It should be noted that despite all efforts made in this project, the results of this survey only provide estimates used to identify trends in the presented data. For example, many of the surveyed missions were conducted by universities with limited budgets while others were sponsored by private companies or government agencies with higher budgets, largely limiting the type of hardware chosen for the mission. This should be considered when analyzing numerical data obtained from this survey and will hopefully be made prominent through the discussion of results.

Table 1: Summary of Data Collected for CubeSat Survey

Mission Information	Launch Information	IV&V Information
Name of Satellite	Launch Date	Processor
Mission Objective	Launch Vehicle/Mission	Orbit
Sponsor	Mission Status	Uplink/Downlink
Manufacturer	Deployer	Attitude Determination
Domestic	Program	Attitude Control
Size	Orbit Parameters	Orbit Determination
Mass		Miscellaneous

3 Results of Survey

A market assessment conducted by SpaceWorks Enterprises, Inc. in 2014 showed that of the nanosatellites launched between 2009 and 2013, 47% can be categorized as 1U CubeSats [?]. Although not enough information is given to determine how many 2U and 3U CubeSats have been launched, there seems to be a significant trend towards 3U CubeSat missions based on the information in Figure ??. These results show the increase in CubeSat missions over recent years and gives reason to believe this trend will continue in the near future.

Figure ?? shows the number of U.S. CubeSats launched per year based on class (i.e. private, government, university, military, etc.). It is obvious that a majority of launches have been for university missions and educational purposes. However, there has been a significant increase seen in government missions as well as privately funded missions.

When considering verification and validation software, it is of most importance to determine the processor and operating system used for as many CubeSat missions as possible. This will provide insight into which processors are most effective and capable of ensuring successful mission operations. Based on the 55 missions where processor data was attainable, almost 38% of these missions utilized PIC processors. Of these processors, 52% used the PIC18. These results are summarized in Figure ??. In addition, it should be noted that of these processors, Pumpkin

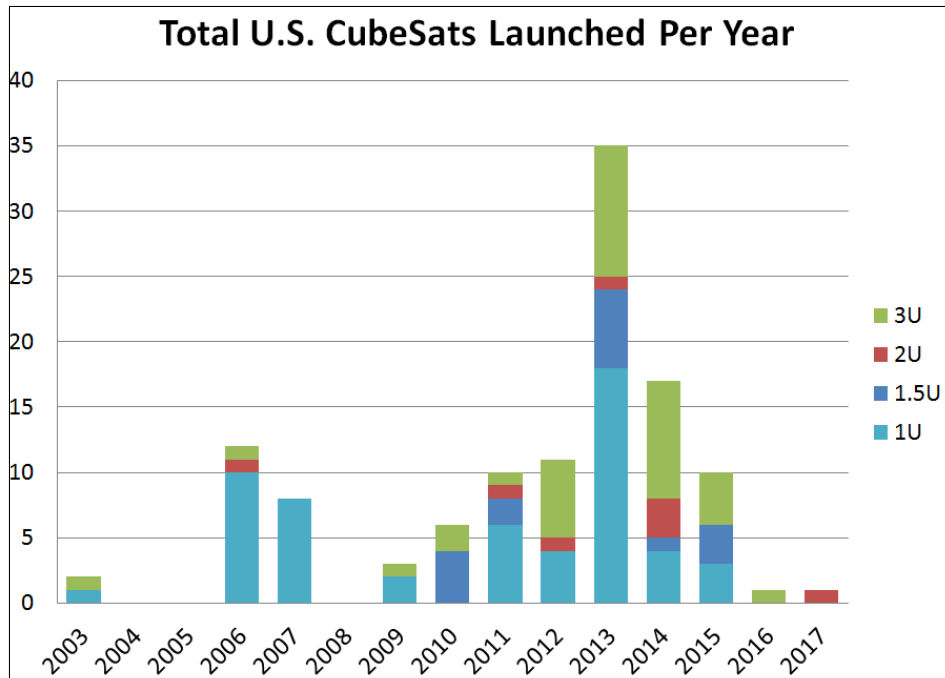


Figure 2: Results of survey showing the size and number of CubeSats launched per year as of August 2014 (based on 116 CubeSats with applicable data)

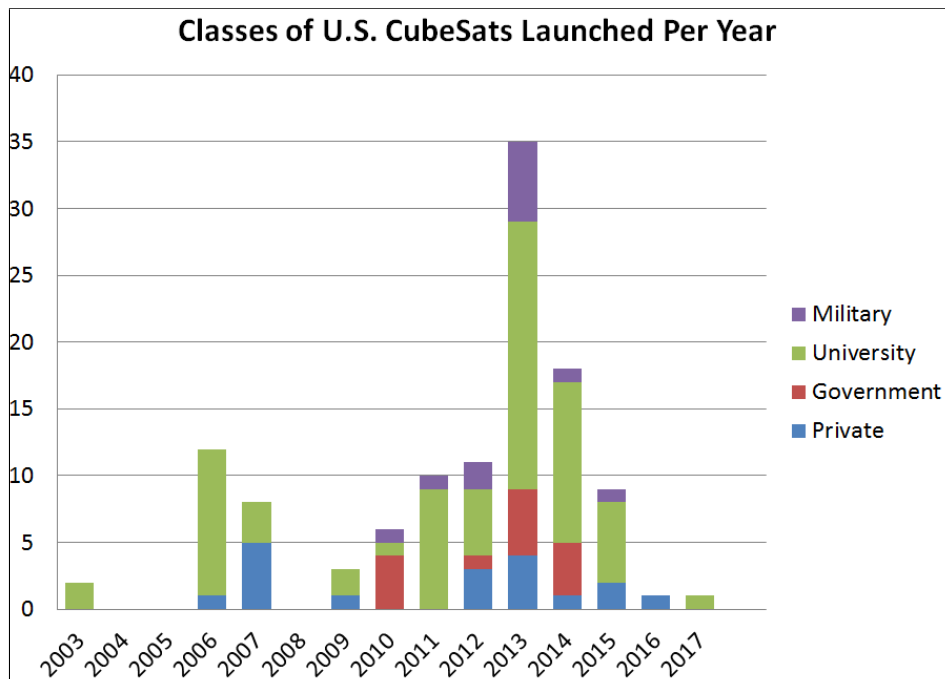


Figure 3: Results of survey showing the number of U.S. CubeSats launched per year, divided by class, as of August 2014 (based on 116 CubeSats with applicable data)

currently provides Pluggable Processor Modules (PPM) available with their CubeSat Kit which is capable of incorporating the following processors with the flight module: SiLabs C805, PIC24,

PIC33, and TI MSP430.

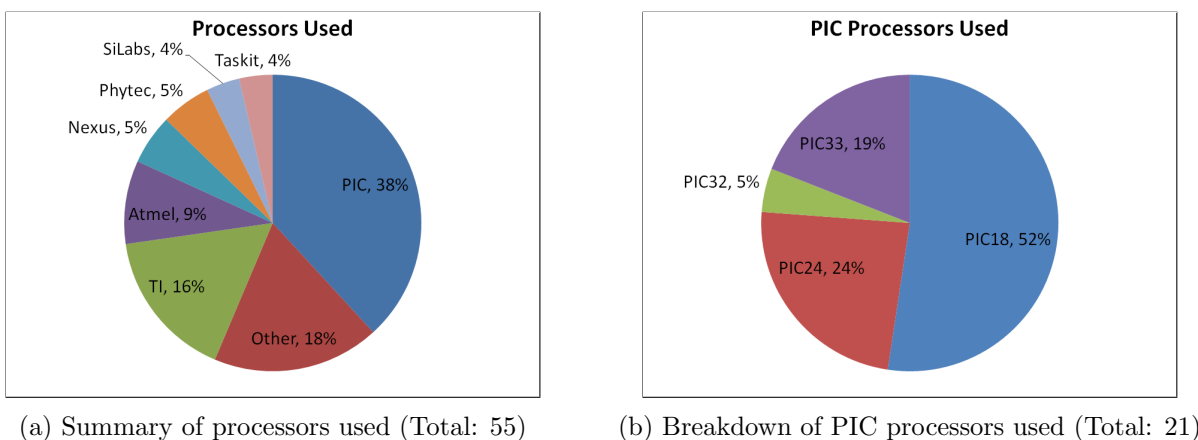


Figure 4: Summary of processors used in CubeSats

As far as operating systems are concerned, Salvo and Linux were the two most commonly used in the CubeSat missions surveyed. Salvo is available with the purchase of any Pumpkin PPM and is included with the purchase of a Pumpkin CubeSat Kit.

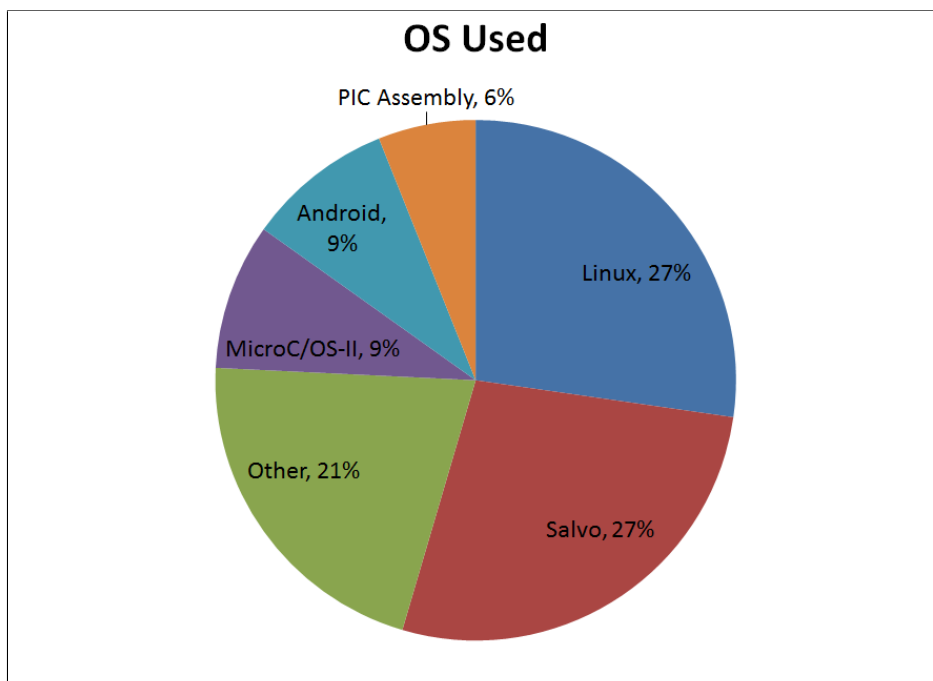


Figure 5: Results of survey showing the percentage of OS used in CubeSat missions (Total: 33)

4 Discussion of Results

Based on the information collected from this survey, a majority of CubeSat missions have used PIC processors (specifically PIC18) with either Linux or Salvo operating systems. The availability of Pumpkin CubeSat kits is likely a major contributor to these results, although this was not

confirmed in this survey. It is our recommendation that NASA IV&V use their previous knowledge and experience with processors and operating systems to create their own verification and validation software. Some aspects to consider are cost, user interface, and power requirements, each which will be determined by the nature of the mission. However, it should be noted that the results of this survey lead us to believe that successful missions can be achieved through the use of PIC processors and Linux or Salvo operating systems. As a reference, processor specifications are compared and summarized in Table ??.

Table 2: Processor Specifications for Consideration

	Operating Frequency	Clock Speed	Program Memory
PIC18	40 MHz	32 kHz	16-32 KB
PIC24	32 MHz	32 kHz	128-256 KB
PIC32	50 MHz	48 kHz	16-256 KB
PIC33	16 MHz	32 kHz	512 KB
TI MPS430	24 MHz		512 KB
Atmel AT91SAM9	210 MHz		
Phytec phyCORE-LPC3250	208 MHz		
SiLabs 8051	50 MHz	16 MHz	32 KB
Taskit Stamp9G20	132 MHz		

5 Future Work

Evident in the agency’s 2015 budget estimate, it is clear that the future of CubeSats within NASA is very promising, at least in the near-term. The 2015 NASA budget indicates a twenty five million dollar allocation to CubeSat projects over the next five years [?]. In light of this, there are currently multiple open or future planned solicitations accepting CubeSat based mission proposals. In particular, a review of the NSPIRES website (nspires.nasaprs.com), shows at least three specific solicitations that request CubeSat proposals, listed in Table ??.

In addition to the \$25M line-item, NASA Earth Science Technology Office (ESTO) funds missions to validate technology through its In-Space Validation of Earth Science Technologies (InVEST) program. Currently InVEST is planning to spend thirteen million over four years on four 3U CubeSat missions [?]. One mission currently in development is ICECUBE or Earth-1, which is being built at Goddard Space Flight Center.

Further, NASAs interest in the future of CubeSats is evidenced in its plans to host a CubeSat based Centennial Challenge competition in order to spur the advancement of propulsion and communication technologies for deep space applications. Primarily, the challenge is designed to develop innovative ways to return error free data from deep space without government assistance, and to demonstrate lunar orbital plane change from equatorial to polar [?]. This Centennial challenge will inevitably lead to technology and knowledge that will advance the field beyond its current state. NASAs investment in this event demonstrates the agencys commitment to the field.

NASA is showing a great commitment to the use of CubeSats for rigorous scientific endeavors as demonstrated by the availability of funding opportunities and the development of a CubeSat Centennial Challenge.

Table 3: CubeSat Proposal Outlines

CubeSat Proposal Outlines	
Name	Description
Remote Sensing Theory for Earth Science	Remote sensing science to establish a theoretical basis for measuring Earth surface properties using reflected, emitted, and scattered electromagnetic radiation and to develop the methodologies and technical approaches to analyze and interpret such measurements lies at the heart of NASAs mission.
Heliophysics Technology and Instrument Development for Science	The H-TIDEs program solicits proposals for investigations that are relevant to NASA’s programs in Heliophysics
Astrophysics Research and Analysis,Program	The Astrophysics Research and Analysis Program (APRA) program solicits basic research proposals for investigations that are relevant to NASA’s programs in astronomy and astrophysics and includes research over the entire range of photons, gravitational waves, and particle astrophysics.

References

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