```
1 # LQR optimal controller
 2
 3 # Import libraries
 4 import numpy as np
 5 from base_controller import BaseController
 6 from lgr_solver import dlgr, lgr
 7 from scipy.linalg import solve_continuous_lyapunov
   , solve_lyapunov, solve_discrete_lyapunov
8 from math import cos, sin
9 import numpy as np
10 from scipy import signal
11
12 class LQRController(BaseController):
13
       """ The LOR controller class.
14
       11 11 11
15
16
17
       def __init__(self, robot, lossOfThurst=0):
           """ LQR controller __init__ method.
18
19
20
           Initialize parameters here.
21
22
           Args:
23
               robot (webots controller object):
   Controller for the drone.
24
               lossOfThrust (float): percent lost of
   thrust.
25
26
27
           super().__init__(robot, lossOfThurst)
28
29
           # define integral error
           self.int_e1 = 0
30
31
           self.int_e2 = 0
32
           self.int_e3 = 0
33
           self.int_e4 = 0
34
35
           # define K matrix
36
           self.K = None
37
38
       def initializeGainMatrix(self):
```

```
""" Calculate the gain matrix.
39
40
41
           11 11 11
42
43
           # -----|LQR Controller
44
           # Use the results of linearization to
   create a state-space model
45
46
           # Given parameters
47
           n_p = 12 \# number of states
48
           m = 4 # number of integral error terms
49
50
           # robot parameter
51
           self.m = 0.4
52
           self.d1x = 0.1122
53
           self.d1y = 0.1515
54
           self.d2x = 0.11709
55
           self.d2y = 0.128
56
           self.Ix = 0.000913855
57
           self.Iy = 0.00236242
58
           self.Iz = 0.00279965
59
60
           # constants
61
           self.q = 9.81
62
           self.ct = 0.00026
63
           self.ctau = 5.2e-06
64
           self.U1_max = 10
65
           self.pi = 3.1415926535
66
67
           # ----- Your Code Here
          -----#
           # Compute the discretized A_d, B_d, C_d,
68
  D_d, for the computation of LQR gain
69
70
           # Matrix A logic
71
           # Initialize A matrix with zeros ( 16 x 16
72
           A = np.zeros((n_p+m, n_p+m))
           A[0, 6] = 1; A[1, 7] = 1; A[2, 8] = 1; A[3]
73
   , 9] = 1; A[4, 10] = 1; A[5, 11] = 1
```

```
74
            A[6, 4] = self.g; A[7, 3] = -self.g
 75
            A[12, 0] = 1; A[12, 12] = -1; A[13, 1] = 1
    ; A[13, 13] = -1; A[14, 2] = 1; A[14, 14] = -1; A[
   15, 5] = 1; A[15, 15] = -1
         \# A[12, 0] = 1; A[13, 1] = 1; A[14, 2] = 1
 76
    ; A[15, 5] = 1;
 77
 78
 79
           # Matrix B logic
 80
           # Initialize B matrix with zeros ( 16 x 4
 81
            B = np.zeros((n_p+m, m))
            B[8, 0] = 1/self.m; B[9, 1] = 1/self.Ix; B
 82
    [10, 2] = 1/self.Iy; B[11, 3] = 1/self.Iz
 83
 84
           # Matrix C logic
 85
           # Initialize C matrix with zeros ( 4 x 16
     )
           C = np.zeros ((m, n_p+m))
 86
            C[0, 0] = 1; C[1, 1] = 1; C[2, 2] = 1; C[3]
 87
    , 3] = 1
 88
 89
          # Matrix D logic
 90
           # Zero matrix ( 4 x 4 )
 91
            D = np.zeros((m, m))
 92
 93
           # Discretize the system
 94
            sys_discrete = signal.cont2discrete((A, B
     C, D), self.delT, method='foh')
 95
 96
            # Extract A_d, B_d, C_d, D_d
 97
            A_d = sys_discrete[0]
 98
            B_d = sys_discrete[1]
 99
            C_d = sys_discrete[2]
100
            D_d = sys_discrete[3]
101
102
                      ----- Your Code Ends Here
103
            # ----- Example code
104
```

```
105
            \# \max_{pos} = 15.0
            # max_ang = 0.2 * self.pi
106
107
            \# \max_{vel} = 6.0
108
            # max_rate = 0.015 * self.pi
109
            \# \max_{eq} I = 3.
110
111
           # max_states = np.array([0.1 * max_pos, 0.
    1 * max_pos, max_pos,
112
            #
                                   max_ang, max_ang,
   max_ang,
113
                                   0.5 * max_vel, 0.5
     * max_vel, max_vel,
114
                                   max_rate, max_rate,
   max_rate,
115
                                   0.1 * max_{eyI}, 0.1
     * max_{eyI}, 1 * max_{eyI}, 0.1 * max_{eyI})
116
117
            # max_inputs = np.array([0.2 * self.U1_max
    , self.U1_max, self.U1_max, self.U1_max])
118
119
            \# Q = np.diag(1/max_states**2)
           # R = np.diag(1/max_inputs**2)
120
121
            # ----- Example code Ends
                        ----- Your Code Here
122
           # Come up with reasonable values for Q and
123
     R (state and control weights)
124
            # The example code above is a good
    starting point, feel free to use them or write you
     own.
125
            # Tune them to get the better performance
126
127
            # referencing the example code above
128
            max_pos = 15.0
129
            max_ang = 0.2 * self.pi
130
            max_vel = 6.0
131
            max_rate = 0.015 * self.pi
132
            max_eyI = 0.75
133
134
            max_states = np.array([0.1 * max_pos, 0.1])
```

```
134
     * max_pos, max_pos,
135
                                    max_ang, max_ang,
    max_ang,
136
                                    0.5 * max_vel, 0.5
     * max_vel, max_vel,
137
                                    max_rate, max_rate
    , max_rate,
138
                                    0.1 * max_eyI, 0.1
     * max_eyI, 1 * max_eyI, 0.1 * max_eyI])
139
140
            max_inputs = np.array([0.2 * self.U1_max,
    self.U1_max, self.U1_max, self.U1_max])
141
142
            Q = np.diag(1 / max_states ** 2)
            R = np.diag(1 / max_inputs ** 2)
143
144
145
                         ----- Your Code Ends Here
146
            # solve for LQR gains
147
148
            [K, _, _] = dlqr(A_d, B_d, Q, R)
149
150
            self.K = -K
151
152
        def update(self, r):
            """ Get current states and calculate
153
    desired control input.
154
155
            Args:
                r (np.array): reference trajectory.
156
157
158
            Returns:
159
                np.array: states. information of the
    16 states.
160
                np.array: U. desired control input.
161
            11 11 11
162
163
164
            # Fetch the states from the BaseController
     method
            x_t = super().getStates()
165
```

```
166
167
            # update integral term
168
            self.int_e1 += float((x_t[0]-r[0])*(self.
    delT))
            self.int_e2 += float((x_t[1]-r[1])*(self.
169
    delT))
            self.int_e3 += float((x_t[2]-r[2])*(self.
170
    delT))
            self.int_e4 += float((x_t[5]-r[3])*(self.
171
    delT))
172
173
            # Assemble error-based states into array
            error_state = np.array([self.int_e1, self.
174
    int_e2, self.int_e3, self.int_e4]).reshape((-1,1))
            states = np.concatenate((x_t, error_state
175
    ))
176
177
            # calculate control input
            U = np.matmul(self.K, states)
178
179
            U[0] += self.g * self.m
180
181
            # Return all states and calculated control
     inputs U
182
            return states, U
```