```python
# LQR optimal controller

# Import libraries
import numpy as np
from base_controller import BaseController
from lqr_solver import dlqr, lqr
from scipy.linalg import solve_continuous_lyapunov, solve_lyapunov, solve_discrete_lyapunov
from math import cos, sin
import numpy as np
from scipy import signal

class LQRController(BaseController):
    """ The LQR controller class.

    """

    def __init__(self, robot, lossOfThurst=0):
        """ LQR controller __init__ method.

        Initialize parameters here.

        Args:
            robot (webots controller object): Controller for the drone.
            lossOfThrust (float): percent lost of thrust.

        """
        super().__init__(robot, lossOfThurst)

        # define integral error
        self.int_e1 = 0
        self.int_e2 = 0
        self.int_e3 = 0
        self.int_e4 = 0

        # define K matrix
        self.K = None

    def initializeGainMatrix(self):
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        """ Calculate the gain matrix.

        """

        # ---------------|LQR Controller|-------------------------
        # Use the results of linearization to create a state-space model

        # Given parameters
        n_p = 12 # number of states
        m = 4 # number of integral error terms

        # robot parameter
        self.m = 0.4
        self.d1x = 0.1122
        self.d1y = 0.1515
        self.d2x = 0.11709
        self.d2y = 0.128
        self.Ix = 0.000913855
        self.Iy = 0.00236242
        self.Iz = 0.00279965

        # constants
        self.g = 9.81
        self.ct = 0.00026
        self.ctau = 5.2e-06
        self.U1_max = 10
        self.pi = 3.1415926535

        # ----------------- Your Code Here ----------------- #
        # Compute the discretized A_d, B_d, C_d, D_d, for the computation of LQR gain

        # Matrix A logic
        # Initialize A matrix with zeros ( 16 x 16 )
        A = np.zeros((n_p+m, n_p+m))
        A[0, 6] = 1; A[1, 7] = 1; A[2, 8] = 1; A[3, 9] = 1; A[4, 10] = 1; A[5, 11] = 1
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        A[6, 4] = self.g; A[7, 3] = -self.g
        # A[12, 0] = 1; A[12, 12] = -1; A[13, 1] = 1; A[13, 13] = -1; A[14, 2] = 1; A[14, 14] = -1; A[15, 5] = 1; A[15, 15] = -1
        A[12, 0] = 1; A[13, 1] = 1; A[14, 2] = 1; A[15, 5] = 1

        # Matrix B logic
        # Initialize B matrix with zeros ( 16 x 4 )
        B = np.zeros((n_p+m, m))
        B[8, 0] = 1/self.m; B[9, 1] = 1/self.Ix; B[10, 2] = 1/self.Iy; B[11, 3] = 1/self.Iz

        # Matrix C logic
        # Initialize C matrix with zeros ( 4 x 16 )
        C = np.zeros ((m, n_p+m))
        C[0, 0] = 1; C[1, 1] = 1; C[2, 2] = 1; C[3, 3] = 1

        # Matrix D logic
        # Zero matrix ( 4 x 4 )
        D = np.zeros((m, m))

        # Discretize the system
        sys_discrete = signal.cont2discrete((A, B, C, D), self.delT, method='foh')

        # Extract A_d, B_d, C_d, D_d
        A_d = sys_discrete[0]
        B_d = sys_discrete[1]
        C_d = sys_discrete[2]
        D_d = sys_discrete[3]

        # ----------------- Your Code Ends Here ----------------- #

        # -----------------    Example code    ----------------- #
        # max_pos = 15.0
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        # max_ang = 0.2 * self.pi
        # max_vel = 6.0
        # max_rate = 0.015 * self.pi
        # max_eyI = 3.

        # max_states = np.array([0.1 * max_pos, 0.1 * max_pos, max_pos,
        #                      max_ang, max_ang, max_ang,
        #                      0.5 * max_vel, 0.5 * max_vel, max_vel,
        #                      max_rate, max_rate, max_rate,
        #                      0.1 * max_eyI, 0.1 * max_eyI, 1 * max_eyI, 0.1 * max_eyI])

        # max_inputs = np.array([0.2 * self.U1_max, self.U1_max, self.U1_max, self.U1_max])

        # Q = np.diag(1/max_states**2)
        # R = np.diag(1/max_inputs**2)
        # -----------------  Example code Ends ----------------- #
        # ----------------- Your Code Here ----------------- #
        # Come up with reasonable values for Q and R (state and control weights)
        # The example code above is a good starting point, feel free to use them or write you own.
        # Tune them to get the better performance

        # referencing the example code above
        max_pos = 15.0
        max_ang = 0.2 * self.pi
        max_vel = 6.0
        max_rate = 0.015 * self.pi
        # max_eyI = 0.75
        max_eyI = 3.0

        max_states = np.array([0.1 * max_pos, 0.1
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 * max_pos, max_pos,
                               max_ang, max_ang, max_ang,
                               0.5 * max_vel, 0.5 * max_vel, max_vel,
                               max_rate, max_rate, max_rate,
                               0.1 * max_eyI, 0.1 * max_eyI, 1 * max_eyI, 0.1 * max_eyI])

        max_inputs = np.array([0.2 * self.U1_max, self.U1_max, self.U1_max, self.U1_max])

        Q = np.diag(1 / max_states ** 2)
        R = np.diag(1 / max_inputs ** 2)

        # ----------------- Your Code Ends Here ----------------- #

        # solve for LQR gains
        [K, _, _] = dlqr(A_d, B_d, Q, R)

        self.K = -K

    def update(self, r):
        """ Get current states and calculate desired control input.

        Args:
            r (np.array): reference trajectory.

        Returns:
            np.array: states. information of the 16 states.
            np.array: U. desired control input.

        """

        # Fetch the states from the BaseController method
        x_t = super().getStates()
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        # update integral term
        self.int_e1 += float((x_t[0]-r[0])*(self.delT))
        self.int_e2 += float((x_t[1]-r[1])*(self.delT))
        self.int_e3 += float((x_t[2]-r[2])*(self.delT))
        self.int_e4 += float((x_t[5]-r[3])*(self.delT))

        # Assemble error-based states into array
        error_state = np.array([self.int_e1, self.int_e2, self.int_e3, self.int_e4]).reshape((-1,1))
        states = np.concatenate((x_t, error_state))

        # calculate control input
        U = np.matmul(self.K, states)
        U[0] += self.g * self.m

        # Return all states and calculated control inputs U
        return states, U
```